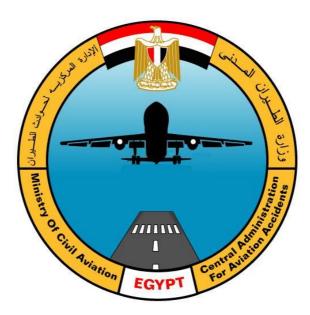
EGYPTIAN MINISTRY OF CIVIL AVIATION



ADDENDUM #1 TO THE FACTUAL REPORT¹ OF INVESTIGATION OF ACCIDENT

Flash Airlines flight 604

January 3, 2004

Boeing 737-300 SU-ZCF

Red Sea off Sharm El-Sheikh, Egypt

July 20, 2005

¹ The information in this addendum complement the information included in the factual report issued on November 2004

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 - Captain interpersonal characteristics, including perceptions of fellow pilots regarding their capability for assertiveness.
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 - Description of how well the flying crew got along.
 - Reported proficiency information. Outcome and comments from training records and proficiency check forms.
 - Spatial disorientation or upset recovery training received at Flash Air or in the military.

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² Items marked in yellow background are the newly added items

- Captain's flying proficiency and cockpit style from fellow pilots, instructors, and/or check pilots.
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 - FDR Parameter Review, B-H200-17884-ASI, 3 May 2004 (ATT For Decoding Grid.pdf)
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 - SimMatchaccidentflight 24-2-04.pdf (Simulation Match, FDR-Kincon-Simulation)
 - SimMatchpreviousflight 24-2-04.pdf (FDR-Kincon-Simulation match 24-2-04)
 - HEA_PQ294_prevfltSIM.pdf (26 Feb 2004, base lines, EDP, Kingon Sim previous flight)
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- HEA_PQ294_FDR_data.pdf (FDR Data accident flig)	nt
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	- Aileron PCU Control Valve.ppt
	- ControlWheelBias.pdf, CairoMarch04Slides (March
	Progress Meeting - Cairo).pdf
	- AileronFloat.pdf (PQ294 FDR Aileron Position,
	Aileron Float from Airload)
	- M-Cab Wheel (Flight Director Results Boeing.xls)
	- Force vs Wheel.ppt
	- Cor8tmp PCU correction.ppt
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	 Aileron PCU EQA Report (Aileron PCU EQA
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	- Cairo March 04 Autopilot Flash 737 March Progress
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	• Autopilot Engage Attempt- with Time
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	- AP Actuator description and Scenario 12 info b.pdf,
	AP Actuator description and Scenario 12 info 2.ppt
	- Scenario 12 ver 2.ppt (Rev - 3 Feb 05)
	- Honeywell SP-300 DFCS B737-300.ppt
	- Flash Airlines Presentation SP-300 DFCS Health
	Monitoring Honeywell.ppt
<mark>1.16.1.10</mark>	0. Flash Airlines AI236 RAM Simulator Configuration (Flash
	Airlines AI236 RAM Simulator Configuration.htm,
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- Exhibit D Airplane Performance Group Factual Report
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- Description of how well the flying crew got along. - Reported proficiency

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			A. FAA advisory Circular regarding SD
			B- MCA study regarding SD
			C- Medical records for the captain related to any
			of the conditions conducive to spatial
			disorientation.
			1.13.3. Most recent medical certification.
			1.13.4. General health information for each crew
			member
			1.13.5. Toxicological testing.
			1.13.6. Last civil medical check for Captain
1.14	Fire	1.14	Fire
1.15	Survival Aspects	1.15	Survival Aspects
1.16	Tests and Research	1.16	Tests and Research
	1.16.1 Performance Evaluation		1.16.1. Tests and researches conducted by Boeing and
	1.16.2 Baseline Simulation		Honeywell:
	1.16.3 Hypothetical Faults resulting in a rolling		
	moment		1.16.1.0. General Overview of Boeing Process_
	1.16.4 Multi-Purpose Engineering Cab Simulator		Kinematic Consistency:
	1.16.4.1 Tests conducted in the M-Cab		(CairoMarch04Slides March Progress Meeting

- Cairo.pdf)
(Kincon and Simulation.ppt)
1.16.1.1. Estimated accident flight path, calculated from
FDR data: (FlightPathMap.pdf)
1.16.1.2. FDR data plots (presented by Boeing)
- FDR Data accident flight - FDR plot.pdf
(some selected parameters 24 Feb 04)
Longitudinal axis
Lateral axis
- FIG_1_LATERAL_EVENT.pdf (Time
aligned FDR data 28 March 05)
Overbank during A/P turn- Lateral axis
- FIG_2_LONG_EVENT.pdf (Time
aligned FDR data 28 March 05)
Overbank during A/P turn- Longitudinal
axis
- FIG_3_LATERAL_GROUNDROLL_T
AKEOFF.pdf (Time aligned FDR data
28 March 05)
Overbank during A/P turn- Lateral axis,
Ground Roll
- FIG_4_LATERAL_AFTER_EVENT_T
RIM.pdf (Time aligned FDR data 28
March 05)
Overbank during A/P turn- Lateral axis,
Climbout after event
- FDR Parameter Review, B-H200-
17884-ASI, 3 May 2004 (ATT For
Decoding Grid.pdf)
1.16.1.3. Simulator Match accident flight:
1.10.1.5. Simulator Match accident fight.

 SimMatchaccidentTight 24-2-04.pdf Simulation SimMatchpreviousTight 24-2-04.pdf Simulation SimMatchpreviousTight 24-2-04.pdf (FDR-Kincon-Simulation match 24-2- 04) HEA_PQ294_prevIItSIM.pdf (26 Feb 2004, base lines, FDR-Kincon-Sim prvious Tight) HEA_PQ294_paselineSIM.pdf (26 Feb 2004, base lines, FDR-Kincon-Sim) HEA_PQ294_DaselineSIM.pdf (26 Feb 2004, base lines, FDR-Kincon-Sim) HEA_PQ294_Dase lines, FDR-Kincon-Sim) HEA_PQ294_PR data.pdf (FDR Data accident Tight - Boeing - 26 Feb 04 Figs 1, 2) HEA_PQ294_kincon (includes roll rate).pdf (FDR Data accident Tight - plotted by Boeing (some selected parameters)-26 Feb 04 Fig's 3, 4 HEA_PQ294_windsSIM29402to29442, pdf (26 Feb 04 Fig's 3, 4 HEA_PQ294_windsSIM29402to29442, pdf (26 Feb 04 Fig's 3, 2, 4<th></th>	
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17871-ASI 31 March 2004). Boeing plots - M Cab Recovery (Piloted Recovery.xls) - Simulation Scenario (Simulation Scenario Status20 Sep.,04.xls) - Simulation Scenario (Simulation Scenario (Simulation Scenario Status 27-30 Sep, 04.xls) - Simulation Scenario Status 27-30 Sep, 04.xls) - HEA_PQ294_Simulated_Failures	
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- HEA_PQ294_Simulated_Failures	
Spoilers, LE Slats.pdf (FDR-norm	
	Spoilers, LE Slats.pdf (FDR-norm

simulation-simulation with spoilers
failures)
 Right outboard flight spoilers (#7)
Hardover simulation (hardover
starts at 92391)
 Left outboard flight spoilers (#2)
Hardover simulation (hardover
starts at 92391)
 Right outboard flight spoilers (#7)
Float simulation (floats starts at
<mark>92391)</mark>
 Left outboard flight spoilers (#2)
Float simulation (floats starts at
<mark>92391)</mark>
 Critical right wing leading edge
slat # 6 extends
 Critical left wing leading edge slat
1 extends
- Fig 40-43 Lateral Control Jams.pdf
(FDR, normal simulation, simulation
with spoilers fault)
 Longitudinal Axis, simulated right
wing spoiler cable jam
Longitudinal Axis, simulated F/O's
wheel jam
- Hypothetical Scenario, Right Side Cable
Jam Induces Right Roll (Right Side
Cable Jam Effects.ppt)
1.16.1.5. FDR 25 Hour Data- Observations
(CairoMarch04Slides (March Progress
Meeting - Cairo).pdf)

1.16.1.6. FDR-CVR Overlay FDR-CVROverlay.pdf, FDR-CVR Overlay 3R2.pdf (21-June 2004)
CVR- FDR Correlation
(CVRFDRCorrelation-1 NTSB.xls)
1.16.1.7. Ailerons system
- IPC wheel posn xducer PW.pdf (Details
about the wheel post Addeer - Part
Catalog Maintenance)
- CairoMarch04Slides (March Progress
Meeting - Cairo).pdf
- Aileron PCU Control Valve.ppt
- ControlWheelBias.pdf,
CairoMarch04Slides (March Progress
Meeting - Cairo).pdf
- AileronFloat.pdf (PQ294 FDR Aileron
Position, Aileron Float from Airload)
- M-Cab Wheel (Flight Director Results
Boeing.xls)
- Force vs Wheel.ppt
- Cor8tmp PCU correction.ppt
- Aileron PCU EQA Field Note Summary
(Aileron PCU EQA Field Note
Summary.ppt)
- Aileron PCU EQA Report (Aileron
PCU EQA Report.pdf)
1.16.1.8. Master Caution:
CairoMarch04Slides (March Progress Meeting
- Cairo).pdf

	1.16.1.9. Auto Flight Systems
	- 17833 (B-H200-17833-ASI 12 Feb
	2004).pdf
	 CairoMarch04Slides (March Progress
	Meeting - Cairo).pdf
	Relevant Figures
	 737-300 (PQ294) Flight Director
	Control Law (see also
	FDControlLaw.pdf file)
	FD Display
	Times of Example Display Photos
	- M-Cab Flight Director Commands
	(Flight Director Results Boeing.xls)
	- Display Architecture (Display
	Architecture.ppt)
	- Cairo March 04 Autopilot Flash 737
	March Progress Meeting Flash 737
	March Progress Meeting - Cairo.pdf
	 Autopilot Engagement
	o Observation
	o Autopilot Engage Logic
	• Autopilot Engage Attempt-
	with Time Aligned Data
	o Autopilot Engage Attempt-
	with CVR Data
	o Estimated Autopilot
	Availability
	- AP Actuator description and Scenario
	12 info b.pdf, AP Actuator description
	and Scenario 12 info 2.ppt
	- Scenario 12 ver 2.ppt (Rev - 3 Feb 05)
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 Honeywell SP-300 DFCS B737-300.ppt Flash Airlines Presentation SP-300 DFCS Health Monitoring Honeywell.ppt
1.16.1.10. Flash Airlines AI236 RAM Simulator Configuration (Flash Airlines AI236 RAM Simulator Configuration.htm, Program_Pins.pdf)
1.16.1.11. Boeing response to raised questions.doc References 17833 (B-H200-17833-ASI 12 Feb 2004).pdf CairoMarch04Slides (March Progress Meeting - Cairo).pdf 17848 (B-H200-17848-ASI 04 March 2004).pdf Cairo March 04 Autopilot Flash 737 March Progress Meeting Flash 737 March Progress Flash Airlines Autopilot Answer to Questions - 31 Jan 2005.ppt Answers to question_cairo meeting05.ppt Action Item Response.ppt (Cairo meeting, 1- 30-05 to 2-2-05)
1.16.2. Tests and researches conducted by NTSB: c.wheel Dennis Grossi NTSB.ppt

(Trajecto_ma 1.16.4. Tests and res	earches conducted by MCA:
Spatial Disor	ientation Studies
1.16.5 Systems exan 1.16.5.1	Cause(s) for the autopilot disconnect (applied also to the
1.16.5.2	accident aircraft) Cause(s) for "Heading Select" disengage when the autopilot is engaged (applied also to the accident aircraft)
1.16.5.3	Availability of autopilot during the captain's requests "autopilot, autopilot" (accident aircraft)
1.16.5.4	operating the airplane with FD TO/GA mode inoperative (won't stay engaged)
1.16.5.5	Interlock logic for A/P with the definition of the likelyhood (ruled out, not likely, unknown) to the various interlocks regarding the role they may have played in the autopilot disengagement
<mark>1.16.5.6</mark>	

	<i>the command bars.</i> 1.16.5.7 <i>Examination of the selected</i> <i>course compared to the selected</i>
	heading (probability for having "dropouts"). 1.16.6 CVR examination: 1.16.6.1 Examination of the CVR recording for indications of A/P and heading select switch noises 1.16.6.2 Examination of CVR at 2.58.15 (when the MSR crew says that they heard a message from Flash on 121.5). 1.16.7 FDR examination: 1.16.7.1 Spatial disorientation study of the accident flight based on the recorded FDR data AI 081 1.16.8 Aileron PCU inspection and teardown (EQA report):
1.17	Organizational and Management Information 1.17.1 Flash Airlines 1.17.1.1 Flash Airlines Air Operator Certificat (AOC) 1.17.1.2 History 1.17.1.2 Personnels Training and Authorization 1.17.1.2.1. Maintenance Engineers 1.17.1.2.2. Cockpit Crews 1.17.2 Review of oversight by ECAA on 2003
	1.17

1 17 3 Relevant I	Elash Airlines procedures 1.18.3.
	lash Airlines procedures:
1.17.3.1	Flash Airlines procedures regarding
1.17.3.1	use of autopilot
	when recovering from unusual
	attitudes
1.17.3.2	Flash Airlines procedures regarding
1.17.3.2	Upset Recovery
	training
1.17.3.3	<u> </u>
1.1/.3.3	Flash Airlines procedures regarding
	"training about PNF assuming
	control when the PF is not responding
	to situations, callouts"
1.17.3.4	Flash Airlines training/operational
	information regarding intervention by
	the non-flying pilot when the flying
	pilot fails to respond to calls for
	correcting an unsafe situation.
1.17.3.5	Regularity (or irregularity) rules
	regarding sleeping schedules on and
	off-duty. Strategies for obtaining
	adequate rest and managing crew on-
	duty alertness
1.17.3.6	General description about Flash
	Airline.
1.17.3.7	Labor management issues, growth
	trends, and main competitors.
1.17.3.8	Egyptian requirements for the training
	of pilots at an airline such as Flash
	Airlines.
1.17.3.9	The training that was actually
	provided to all Flash Airlines
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			1.17.3.20 Previous violations, fines, or bans
			levied foreign aviation regulatory
			agencies.
			1.17.3.21 Selected additional information
			regarding Flash Airlines Organization.
			1.17.3.22 Airline Simulator program contract
			with RAM, ECAA letter of approval
			1.17.3.23 Simulator used by Flash Airlines at
			RAM,
			1.17.3.24 Flash Airlines procedures regarding
			which pilot (PF or PNF) engages the
			autopilot, Boeing recommended
			practice
			1.17.3.25 Additional information regarding
			dispatch from SSH
			A- All departures from SSH (accident
			aircraft)
			B- Extension of the outbound legs
			before beginning the turn
1.18	Additional Information	1.18	Additional Information
1.19	New Investigation Techniques	1.19	New Investigation Techniques
			1.19.1 Spatial disorientation :
			Definition
			The way the SD works
			Crew fatigue
			Human related factors
Exhibit A	Maintenance Records Group Factual Report	Exhibit A	Maintenance Records Group Factual Report
Exhibit B	FDR Group Factual Report	Exhibit B	FDR Group Factual Report
			Attachment 1, Tabular data of the accident flight.
			Attachment 2, FDR Plots
			Attachment 3, five plots represent FDR and CVR

			1 .*
			correlation
			Attachment 4, Summaries of previous flights of the
			accident aircraft
Exhibit C CVR	R Group Factual Report	Exhibit C	CVR Group Factual Report
			CVR Group Factual Report
			Accident flight plan (copy of the flight plan referred to
			by ATC at 02:38:05 in the CVR transcript)
Exhibit D Airp	blane Performance Group Factual Report	Exhibit D	Airplane Performance Group Factual Report
			Airplane Performance Group Factual Report
			Radar Spec formatted.doc (to complement the item C.2
			Radar Data, General specification)
Exhibit E Site	and Wreckage Group Factual Report	Exhibit E	Site and Wreckage Group Factual Report
Exhibit F Oper	ration Group Factual Report	Exhibit F	Operation Group Factual Report

1.5.1 The Captain

- 1.5.1.2. Background information.
 - i- Beginning of his flying career.Refer to captain CV, and his training records item 1.5.1.2 (vi)
 - ii- All airlines worked for prior to Flash Air
 - The captain joined the A.R.E. Military Aviation College on September 1968, and was graduated on May 1970
 - He continued working as military pilot at A.R.E. Air Force since that date flying the L29, MIG17, MIG21, Buffalo (Dash 5), C130 types until he retired from the A.R.E. Air Force at the beginning of 2000
 - He joined Scorpio Aviation working as a civil pilot on ATR 42 from March, 2000 up to December, 2001.
 - He joined Flash Airline working as a civil pilot on B737-300 from February 2003 until 4 January 2004 (accident date)

(All his flying hours were flown as PIC)

- iii- History of military and civilian employment as pilot The captain flew as a fighter pilot on L29, Mig17, Mig21 since his graduation until 1983. He then flew as a military transport pilot from that date on Buffalo and C130 until his retirement from the Air Force at the beginning of 2000. (Refer to previous item)
- iv- Retirement dates from A.R.E Air Force. Captain has retired from A.R.E. Air Force beginning of 2000
- v- History of position flown for specific aircraft, and dates of upgrades (i.e., copilot to captain)
 Refer to page 14 of the Factual Report
- vi- "All" captain's training records (including his last recurrent training).

C.V.

ersonal information:

Name: Nationality: Data of Birth: place of Birth:

Khedr abdalla saad said "Egyptian February 26th ,1950. Cairo

lot Qualifications & Certificates:

ESc. In aviation. Air Force Academy

AL.T by Egyptian Civil Aviation Organization

K T Communication License

lot Courses:

<u>Gro</u> <u>illitary</u> L29 Mig-17 Mig-21		<u>Courses:</u> <u>Civil</u> Gomhoria Dash-5 C-130 ATR-42	<u>Flight Courses</u> Gomhoria L-29 Mig-17 Mig-21 Dash-5 C-130 ATR-42	Experience Pilot Pilot Pilot Instructor Captain Captain and Instructor to all international route Captain
Total Flying Hours:			6967.05	
Total on	50. St. 19			1009 hrs
Total Civ				5958.05
Total Fly		ours as Instructor:		1967.54 hrs

All the documents are available upon request.

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Certificate, A.R.E. Air Force Head Quarter, Training Department

Number of Training Flying Hours for Captain/ Khedr Abdallah at Scorpio Aviation (15 June 2000)

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16 MAY2000	335	AST	CAi	01	10	1		1	GEO	
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Proficiency Checks at Scorpio Aviation: 17 June 2000

Egyptian Civil Aviation Anthority الهيئة للصرية الملمة الطيراق للدفم Flight Safety Standards Sector فراع السلامة الجوية "Operations Inspectorate ' , النفنيان الحـ 14 . PROFICIENCY/QUALIFICATION CHECKLIST REE. NO. DATROFFILMER KHEDR ABDALLA TYPEOFETIECK PROFICIENCY EMPLOYED BY SCORPIO AVIATION INSELECTOR OR CHECK AIRMAN BASE CAIRO MANIE GEORGESCU TYPE AIRCRAFT ATR-42 FLICHT HATE 01 has 35 mins TYPE SIMULATION ---SIMULATOR HATE ____ his ___ mins ____ -FLIGHT MANOEUVRES (S = Satisfactory, U = Unsatisfactory) PILOT HJOHT ENGINEER Air Sim TITEM Ù S ciaft Inter PREFLICHT S Equipment examination (oraal or written) Equipment exam (oraal or written) 🥏 5 pre flightcheck of aircraft pre-flight inspection Computation of fuel load and fuel loading proced 555 Taxiing > Completion of company approved forms Powerplant checks Starting, taxi, and mump TAKE-OFFS Powerplant control Normal 200000 Cruise control and computations Aircraft/powerplant operation analysis Instrument Cross-wind Fuel system management With signalated powerplant failure Rejected take-off Air comlition and pressurization dute Electrical system operation INSTRUMENT PROCEDURES Powerplant fire control 200000000 Area departure Emergency gear and flap extension Holding Heater fire and eargo compartment fire Area Arrival Smoke evacuation ILS approaches Other instrument approaches Emergency depress rization Fuel dumping procedure Powerplant shutdown and restart Circling approaches De leing auf unit leing Location and use of emergency equipment Missed approaches IN-FLIGHT MANOEUVRES Emergencies Hydraulie pressurization, etc Steep turns 201555 Approaches to stalls Crew to ordination and monitoring Specific flight characteristics RELARKS : Powerplant failure LANDINGS TYPE RATING AS 2000000000 Normal From an ILS PILOT IN COMMAND Cross-wind With simulated powerplant (s) failure ATR-42 Rejected landing From circling approach _____ Normal and abnormal procedures GEORGESCU Ho 17-JUIN . 2000 Hay Emergency procedures Judgement 5 ECAA - INSPECTION FORM (5/96) 1CAU - DOC 83357 C1 كورىيو -i of 1 7 161200 er.

8 December 2000

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SCORPIO AVIATION

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Holding	5		Heater fire and cargo compartment fire			
Area Arrival	S		Smoke evacuation			
H.S approaches	S		Emergency depressurization			
Other instrument approaches	S		Fuel dumping procedure			
Circling approaches	5	1	Powerplant shutdown and restart			
Missed approaches	S	1	De- icing and anti- icing			
IN- FLIGHT MANOEUVERS	S		Location and use of emergency equipment	_		
Steep turns	5		Emergencies - Hydraulic pressurization, etc.			
Approaches to stalls	S		Crew co- ordination and monitoring			
Specific flight characteristics	S		Remarks :			
Powerplant failure	S					
LANDINGS	+			-		
Normal	δ		PROFICIENCY Check			
From an H.S	S	-				
1 tom all the	S	-	PROFICIENCY Check Satisfactory /			
Cross - wind						
	5	+				
Cross - wind						
Cross - wind With simulated powerplant (s) failure	5					
Cross - wind With simulated powerplant (s) failure Rejected landing	5					
Cross - wind With simulated powerplant (s) failure Rejected landing From circuling approach	5		INSPECTOR OR CHECK AIRMAN SIGNATURE			

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12 December 2001

REF.NO. NAMÉ OF PILOTIFÉ. KHÊDRAB EMPLOYED BYSCORP.JOAV BASE TYPE AIRCRAFTAT.R4 TYPE SIMULATION	PR DAZ	OFIC	ciency Check/Qualification Form	
AME OF PILOTIFE KHEDR. AB EMPLOYED BYSCORPIOAV BASE	DAL			1.1.4
	2	10N	FLIGHT TIME	⊊H¢
A 1 YO M WANTED BY A 1 YO M WANTED BY A 1 YO MARKED BY A			Simulator time	1
· FLIGHT MANO	DEUVE	ERS (S= Satisfactory , U = Unsatisfactory) FLIGHT ENGINEER	<u> </u>
		Simu		S
L.J. MOHT	craft	lator	ITEM Equipment exam (oral or written)	FT
upment examination(oral or written)	S	12	Preflight check of aircraft	+
in fight inspection	S		Computation of fuel load and fuel loading procedure	++
	S	1	Completion of company approved forms	++
owerplant Checks	Ś	<u> </u>	Starting taxi and run up	++
AKE. OFFS	5		Powerplant control	++
omal	S		Cruise control and computations	Ħ
Infrument	S		Aircrft/powerplant operation analysis	++
cross - Wind	S		Fuel system management	
With simslated powerplant failure	2		Aircondition and pressurization control	Ħ
ejected take- off	S		Electricl system operation	++
NSTRUMENT PROCEDURES			Powerplant fire control	++
Tea departure	S	,	Emergency gear and flap extension	H
folding	S	12	Heater fire and cargo compartment fire	H+
rea Amval	S	4.5	Smoke evacuation	H
Sapproaches	S		Emergency depressurization	
Other instrument approaches	3	-	Fuel dumping procedure	
Circling approaches	8		Powerplant shutdown and restart	
Aissed approaches	S		De- icing and anti- icing	- İo
- FLIGHT MANOEUVERS			Location and use of emergency equipment	+
iteep turns	S		Emergencies - Hydraulic pressurization, etc.	
aproaches to stalls	S		Crew co- ordination and monitoring	
Decific flight characteristics	2	· · · ·	Remarks :	-
owerplant failure	2			
ANDINGS	-		00	,
lémal	2		PROFICIENCY CHECK	
tom an H.S	S			
lioss - wind			ON ATR 42-370	
Vith simulated powerplant (s) failure	8			<u> </u>
alacted landing	S		SATISFACTORY	
iem circuling approach	9		- ~	
armal and abnormal procedures	8			
mergency procedures	5.	- 1		
	2		INSPECTOR OR CHECK AIRMAN SIGNATURE	
			MARAU FUNICO	
			MAGDY KHALED	
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		- 1 19	Ataina	

Fixed Base Simulator Training:

JELASHIB	17
α μ	TRAINING RECORD FBS SSON 1
NAME: KHEDR ABDALAA	CREW POSITION: CAPTAIN
	A/C TYPE: 13 737-300/400/500
BRIFING NORMAL PROCEDURES S/ US	۲ Cruse Normal procedures S/پلا
PREFLICHT S / Ut tractice AFDS preflight S / Ut tractice FMC/CDU preflight S / Ut tractice IRS Full alignment S / Ut	5 Normal procedures 5708
ENGIN START Vormal procedures S / U	B Normal procedures S / US
Taxl-out & takeoff Jnal procedures S/J	18 Taxi - in & park Normal procedures S / US
Aormal procedures S/ VS	
CAPIAIN KHEDR	NEEDS TO TMPROVE
COCK PIT PEREPER	
(Nob	RMAL PROCEDURES)
NSTRUCTOR NAME: IHAB EL SONBATY	INSTRUCTOR SIGNATURE:
DATE: 28-04-03	TRENIY SIGNATURE:
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TRAINING MANUAL

		TRAINING RECORD FBS
	LESS	ON 2
AME: KHEDR ABDALAA		CREW POSITION: CAPTAIN
		NCTYPE: <u>B 737-300/400/500</u>
BRIFING NORMAL PROCEDURES Supplementary Normal procedures	s/µs	Cruse Normal procedures S / Y8'
FM alerting & advisory messages MCP controls and FMA FMC LNAV operation		Descent & Approach Normal procedures S / US
PREFLIGHT Normal procedures Supplementary Normal procedures	s/UK s/UK	Landing Normal procedures S / ŲS
ENGI START Normal procedures	s/125	Taxi - in & park Normal procedures S/US
Taxi-out & takcoff Normal procedures	5/95 5/95	
Climb Normal procedures	s/yø	
Demonstration flight	s/µs	
്		
REARKS: PROGRESSIN	Ģ P	UT STILL NEEDS TO
		DIT PREPERATION
		TO TAKEN BY TANK AN ARCT DOT.
INSTRUCTOR NAME: IHAB EL SONBA	TV	INSTRUCTOR SIGNATURE:
DATE: 29-04-03		TRENIY SIGNATURE BRAHORS
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TRAINING MANUAL

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J TLASH.		TRAINING RECORD FBS
والمراجع الواقع المراجع	LESS	ON 3
NE: <u>KHEDR ABDALAA</u>		CREW POSITION: CAPTAIN
		A/C TYPE: 13 737-300/400/500
FING IC LNAV OPERATION	s / Ųď	<u>Cruse</u> Normal procedures s / על
FLIGHT mal procedures	s / 125	Descent & Approach S / US Normal procedures S / US MISSED APPROACH S / US
elementary, Normal procedures	s/y⊄	te
		Landing Normal procedures S / US MISSED APPROACH S / US
GIN START n-Normal procedures	s/µ≴	Taxi - in & park Normal procedures S/J# Supplementary normal procedures S/J#
<u>ki-out & takcoff</u> C LNAV &VNAV OPERATION	s / µs	
imb rmal procedures	s/µs	
EMARKS: PROGRE	=<5/4	10-
NSTRUCTOR NAME:	55	INSTRUCTOR SIGNATURE:
ZHAB EL SONB DATE: 30-04-03	<u> </u>	TRENIY SIGNATURE:
na na manana na manini ni ni ni ni ni ni ni ni na na manana na manana manana manana manana manana manana manana Manana manana m	ne prome anomalisme	OPERATIONS
		TRAINING MANUAL

/ TRASS		TRAINING RECORD FBS	
• • • ,	LESSC	N 4	reconcernant
ME: <u>KHEOR ABDALAR</u>		CREW POSITION: <u>CAPIAI</u> A/ A/C TYPE: <u>13 737- 300/40</u>	0/500
RIFING ORMAL PROCEDURES	s / Ų5	Cruse Normal procedures Fix position	S/ປຽ S/ປຽ S/ປຽ
Non-normal procedures	s/µs	Fix position & abeam	5195
eview system & FMC / CDU	S / 145"	Descent & Approach Normal procedures Holding	s/vs s/vs
Iormal procedures <u>INGE</u> Iormal procedures Ion-normal procedures	s/ນຮ້	Landing Normal procedures Missed approach procedures Non - normal procedures	ร/ฟรี ร/มรี ร/บรี
faxi-out & takeoff formal procedures	S/ អូនី	Taxi - in & nark Normal procedures Non-normal procedures	S/US S/US
<u>Climb</u> Normal procedures Runaway stab. (demo) W/ W fire (demo)	5145 5145 5148		
c'			
REN RKS:	Good	PROGRESS	
REDY	For	FULL FLIGHT	
	SÌNU	LATOR	
INSTRUCTOR NAME:	SONBATY	INSTRUCTOR SIGNATURE:	
<u>IHAB EL</u> DATE: 1 1-05-03		TRENIY SIGNATURE: CTC	
ของรูสกุลสารทรงการเหตุลายการการการการการการการการการการการการการก	anan antar an	TRAINING MAI	JUAL

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Full Flight Simulator Training:

IAME <u>KHEDR ABDALAA</u>		CREW POSITION:CAPTAIN /	
MRCRAFT TYPE BOIEN 737-300.	1400/500	DATE OF COPELETION 03-	05-03
Briefing Fraining plan Operation philosophy	<u>s/vs</u>	Cruise 1. Normal procedures	<u>s/145</u>
Preflight Normal procedures Supplementary normal procedures	<u>s/us</u> <u>s/us</u>	Descent & approach Normal procedures	<u>s/u</u> s
Engine start Normal procedures Additional training item	<u>s/us</u> s/us	Landing Normal procedures	<u>s/u≾</u>
Taxi-out & Takeoff Normal procedures	<u>s / U</u> 8	Taxi -in & park Normal procedures	<u>s./ us</u>
Climb Normal procedures Demonstration flight	<u>s/US</u> <u>s/U8</u>		
REMARKES:	KHED	R	
		MOR" EFFORT	
	EHHGP R	R RECALL TTE	NS
INSTRUCTOR NAME:		SIGNATURE:	
INSTRUCTOR WAND, IHAIS	GL SONRI	179	
		TRAININ	A GRAS 2

TRADING DECORD FEE		A Shinking	ų.
. TRAINING RECORD FFS	LESSO		1
NAME KHEDR ABDALAA		CREW POSITION:CAPTAIN /-F+C	<u>r</u>
4			2
AIRCRAFT TYPE BOIEN 737-300	400/500	DATE OF COPELETION 4-05-	2
		Cruise , DESCENT	2.
Briefing Set up MCP, CDU		Hydraulic system A loss	s/us
Engine inoperative characteristics	S/US	Liydrauno system r 1888	
-		Annarch Londing	
Preflight Set up MCP , CDU	S/US	Approach , Landing One engine inop manual, F/D ILS	
After start checklist	S/JUS	Approach	S/U
		One engine inop visual traffic Patterns full stop.	s/v
Engine start	S/US	Wind shear training	
Normal procedures	51 45	Wind shear flight path control hold	S/US
Taxi-out & Takeoff		A/P, A/T, F/D VOR approach Full stop landing	s/v
Rejected T / O T/O engine failure after V 11	<u>s/us</u> s/us	Fur stop knows	1
T/O engine failure after V 1	S/US	Taxi -in & park	e (1)
Wind shear near VR.	S/US	Normal procedures	<u>s/</u>
Climb.			
Climb Normal procedures	S/WS		
r termini processare			
-			
		5	
REMARKES:			
	ROG-RESS	ING.	
	5 7.	GNSS NEED'S	
	0.4	1.4.4	
/ 0	RE	LAX	
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INSTRUCTOR NAME:	1	SIGNATURE: SITA	51
ZHAB E	- SONISTA	 A state of the sta	indies-
		TRAIMING MA	NYAS

NAME_KHEDR_ABDALAA CREW POSITION:CAPTAIN / F+O AIRCRAFT TYPE BOIEN 737-300/400/500 DATE OF COPELETION 6-05-03 Briefing Cruise , DESCENT Review item in phase of flight S / US Preflight Rapid depressurization Normal procedures S / US Engine start Approach to stall recovery Aborted engine starts S / US Normal procedures S / US Normal procedures S / US Maximum Aborted engine starts S / US Normal procedures S / US Maximum Aborted engine starts S / US Normal procedures S / US Missed approach Missed approach	TRAINING RECORD FFS		ATTEANSIL	4
AIRCRAFT TYPE BOIEN 737-300/400/500 DATE OF COPELETION 6-05-03 Briefing Cruise , DESCENT Review item in phase of flight S/US Preflight Rapid depressurization Normal procedures S/US Aborted engine starts S/US Aborted engine starts S/US Normal procedures S/US Preflight Approach to stall recovery Normal procedures S/US Procedures S/US Normal procedures S/US Rejected T/O S/US Normal T/O S/US Normal procedures S/US Normal T/O S/US Wheel well fre * * S/US REMARKES: S/TCL STTCL AGECD'S RELIA X SIGNATURE: RELIA X SIGNATURE: SIGNATURE: SIGNATURE: SIGNATURE: SIGNATURE: Rescue to the stall set of the stal		LESS	ON 3	1
Briefing Cruise , DESCENT Review item in phase of flight S/LX Preflight Rapid depressurization Normal procedures S/LX Approach to stall recovery S/ Engine start Approach to stall recovery Aborted engine starts S/LX Taxi-out & Takeoff Approach , Landing Normal procedures S/LX Rejected T/O S/LX Normal Procedures S/LX Normal T/O S/LX Climb S/LX Wheel well fire * Loss of both engine driven gen. S/LX RELIA X RELIA X RELIA X SIGNATURE: NSTRUCTOR NAME: EL Servisatry ZHARE SIGNATURE: Minil Linic EL <	NAME KHEDR ABDALAA		CREW POSITION: CAPTAIN / F+C	Ţ
Briefing Cruise , DESCENT Review item in phase of flight S/LMS Preflight Rapid depressurization Normal procedures S/LMS Engine start Approach to stall recovery Aborted engine starts S/LMS Taxi-out & Takeoff Approach , Landing Normal procedures S/LMS Rejected T/O S/LMS Normal procedures S/LMS Normal T/O S/LMS Wheel well fire * * S/LMS Wheel well fire * * S/LMS REMARKES: S/TEL S/TEL PRoGRESS */AC RELIA X Normal procedures Tak Normal procedures Tak REMARKES: S/TEL STELA <td>4</td> <td></td> <td></td> <td></td>	4			
Review item in phase of flight S/LS Preflight Rapid depressurization S/L Normal procedures S/LS Engine start Approach to stall recovery S/L Aborted engine starts S/LS Normal procedures S/LS Rejected T/O S/LS Normal procedures S/LS Rejected T/O S/LS Normal procedures S/LS Rejected T/O S/LS Normal procedures S/LS Normal T/O S/LS Climb S/LS Wheel well fire * * S/LS REMARKES: STELL AX STELLAX RELIAX RELIAX SIGNATURE: NSTRUCTOR NAME: SIGNATURE: TAMES EL SOM/RATY SIGNATURE: SIGNATURE: Manual procedures	AIRCRAFT TYPE BOIEN 737-300/4	400/500	DATE OF COPELETION 6-05-0	3
Review item in phase of flight S/LS Preflight Rapid depressurization S/L Normal procedures S/LS Engine start Approach to stall recovery S/L Aborted engine starts S/LS Normal procedures S/LS Rejected T/O S/LS T/O engine failure after V 1 S/LS Normal T/O S/LS Climb S/LS Wheel well fire * * S/LS RemARKES: S/LS STECL PROGRESSINCA S REMARKES: STECL PROGRESSINCA RELIAX S REMARKES: STECL PROGRESSINCA RELIAX S RELIAX SIGNATURE: RELIAX SIGNATURE: RELIAX SIGNATURE: RELIAX SIGNATURE:				
Preflight SIM Normal procedures SIM Engine start Approach to stall recovery Aborted engine starts SIM Taxi-out & Takeoff Approach , Landing Normal procedures SIM Rejected T/O SIM T/O engine failure after V1 SIM Normal T/O SIM Climb SIM Wheel well fire SIM Wheel well fire SIM STICL PRoGRESS in Gradiant REMARKES: STICL STICL PRoGRESS in Gradiant RELIA X STICL Normal procedures SIM Normal procedures </td <td></td> <td></td> <td></td> <td></td>				
Preflight Steep turns. S/LMS Normal procedures S/LMS Approach to stall recovery S/L Engine start Approach to stall recovery S/L Aborted engine starts S/LMS Approach to stall recovery S/L Taxi-out & Takeoff Approach to stall recovery S/L Normal procedures S/LMS Approach to stall recovery S/L Normal procedures S/LMS Missed approach S/L Normal T/O S/LMS Missed approach S/L Normal T/O S/LMS Missed approach S/L Wheel well fire S/LMS Mornal procedures S/L Remarkay stabilizer S/LMS Mornal procedures S/L Buss off Loss of both engine driven gen. S/L/L PRoGRESS in Gr. STICL MGE D'S To RE RELIA X SIGNATURE: Missed approach Signature Normal procedures SIGNATURE: Missed approach Signature STICL Missed approach Signature Signature Signature Normal procedures STICL	Review item in phase of flight	S/US	Rapid depressurization	<u>S/U</u>
Normal procedures S / US Approach to stall recovery S / I Engine start Aborted engine starts S / US Approach to stall recovery S / I Taki-out & Takeoff Approach to stall recovery Missed approach to stall recovery S / I Normal procedures S / US One engine inop A/P, F/D VOR Approach to stall recovery S / I Rejected T/O S / US Missed approach S / I T/O engine failure after V 1 S / US Missed approach S / I Normal T/O S / US Taxi - in & park Normal procedures S / Climb S / US Taxi - in & park Normal procedures S / Wheel well fire * S / US Taxi - in & park Normal procedures S / REMARKES: S / TC PRoGRESS in Co. S S S / TC VG < D'S	Preflight			S/1
Aborted engine starts S/US Taxi-out & Takeoff One engine inop A/P, F/D VOR Normal procedures S/US Rejected T/O S/US T/O engine failure after V 1 S/US Normal T/O S/US Wheel well fire S/US Wheel well fire S/US Wheel well fire S/US REMARKES: S/US STICL PRoGRESS in Gr. REMARKES: STICL REELIA X RELIA X RELIA X STICL RELIA X STICL RELIA X STICL STICL STICL STICL STICL STICL STICL STICL STICL Normal procedures Stick STICL STICL STICL STI		S/JUS		S/W
Aborted engine starts S/US One engine inop A/P, F/D VOR Approach, field to land, full S/L Taxi-out & Takeoff Approach, circle to land, full S/L Normal procedures S/L/S Rejected T/O S/L/S T/O engine failure after V 1 S/L/S Normal T/O S/L/S Climb S/L/S Wheel well fire S/L/S REMARKES: S/L/S STICL PRoGRESS in Gr. REMARKES: STICL RELIA X RELIA X RELIA X Stick of the second secon	Engine start		Approach , Landing	
Taxi-out & Takeoff One engine inop. ILS approach Normal procedures S/US Rejected T/O S/US Normal T/O S/US Wheel well fire S/US Rumaway stabilizer S/US Buss off S/US Loss of both engine driven gen. S/INS STICL PRoGRESS in Gr. REMARKES: STICL STICL PRoGRESS in Gr. RELIA X STICL RELIA X SIGNATURE: MISTRUCTOR NAME: SIGNATURE: THAB EL Servisation SIGNATURE:	Aborted engine starts	S/US	One engine inop A/P , F/D VOR	
Normal procedures S/LMS Rejected T/O S/LMS T/O engine failure after V 1 S/LMS Normal T/O S/LMS Normal T/O S/LMS Climb S/LMS Wheel well fire * * S/LMS Remarkant Structure S/LMS Remarkant Structure S/LMS Remarkant Structure S/LMS REMARKES: STICL PROGRESS in Gradient STICL REMARKES: STICL RELIA X STICL REMARKES STICL RELIA X STICL				S/V
Rejected T/O SIAKS T/O engine failure after V 1 S/LKS Normal T/O S/LKS T/O engine failure after V 1 S/LKS Wheel well fire * Wheel well fire * Wheel well fire * SITE SILKS Soft Loss of both engine driven gen. SITE SITE PRoGRESS in Orthogo SITE SITE SITE NEELIA X RELIA X SIGNATURE: BEL SIGNATURE: SIGNATURE: BEL		0/110		S/L
T/O engine failure after V 1 S/US Normal T/O S/US Climb S/US Wheel well fire * Runaway stabilizer S/US Buss off S/US Loss of both engine driven gen. S/US REMARKES: STELL STELL PRoGRESS in Gr. STELL RELIA X RELIA X SIGNATURE: DEL SONIBATY SIGNATURE: DEL SONIBATY SIGNATURE: DEL SONIBATY				S/L
Climb Normal procedures S. Wheel well fire * S./		S/WS		
Climb Wheel well fire Runaway stabilizer Buss off Loss of both engine driven gen. REMARKES: STILL PROGRESSING. STILL NEED'S TO IRE RELIAX. RELIAX. INSTRUCTOR NAME: THANS EL SONISATY SIGNATURE: DES MAILERS	Normal T/O	S/US		0/1
Wheel well fire SIMS Runaway stabilizer Buss off Loss of both engine driven gen. Image: State of the state o	Climb		Normal procedures	5/1
Rumaway stabilizer Buss off Loss of both engine driven gen. REMARKES: STILL PROGRESS in Cr. STILL NEED'S TO IRE RELIA X. INSTRUCTOR NAME: THATS EL SOMISATY SIGNATURE: DES AMALINE		S/HS		
INSTRUCTOR NAME: THAB EL SONIBATY REMARKES		0100		
REMARKES: STIL PROGRESSING. STIL NEED'S TO RE RELIAX NSTRUCTOR NAME: INSTRUCTOR				
STILL PROGRESSING. STILL NEED'S TO IRE RELAX NSTRUCTOR NAME: INSTRUCTOR NAME:	Loss of both engine driven gen.			
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STILL PROGRESSING. STILL NEED'S TO IRE RELAX NSTRUCTOR NAME: THAN EL SONIBATY SIGNATURE: DE MALLED				
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THAIS EL SONISATY	INSTRUCTOR NAME:			5A
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			LIVERING NUM	UNL

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NAME KHEDR ABDALAA		CREW POSITION: CAPTAIN / Pro	Į
AIRCRAFT TYPE BOIEN 737-300	1400/500	DATE OF COPELETION 8-05-0	3
Briefing		Cruise	
Full auto flight for precision app Review item in phase of light	<u>s///</u> s	Steep turns. Approach to stall recovery	<u>S/</u> # S/#
Preflight		Descent,	
Normal procedures	S/US	Normal procedures	S/W
Reduced thrust computation	<u>s / US</u>	Econ path descent Arrival procedures	s/u
Engine start			and the second sec
Aborted engine starts	S/US	Approach , Landing 5	
Taxi-out & Takeoff		A / ? , A/T , (no F/D) AUTOLAND	<i>a</i> (11)
Normal procedures	<u>s/µs</u>	ILS approach Touch & go landing	S/B
N0 autopilot & F/D Reduced thrust takeoff	<u>S/US</u> S/US	Row data F/D ILS, T & GO.	S/V
Flap retraction	S/ US	A/P , A/T , F/D VOR approach	S/U
•	ilining dis	Touch & go landing	S/U
Climb			
Normal procedures Max angle climb	<u>s / us</u>	Taxi -in & park	
Econ climb		Normal procedures	S/V
REMARKES:	Till	THPROVING.	
		NEED'S TO	
I.NI	PROVE	SiNG16 GNGA	
	HAND	LiNG.	
INSTRUCTOR NAME:	SONRATS	SIGNATURE:	
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TRAINING RECORD FFS	SONE
	CREW POSITION:CAPTAIN / FO
Anon Anon Anone	
AIRCRAFT TYPE BOIEN 737-300/400/50	DATE OF COPELETION 09-05-03
Briefing	Cruise
	Steep turns. S/US Approach to stall recovery S/US
,.	
Dualit	Normal procedures S/US
	0.1118
	* L
Engine start	Approach , Landing
	One engine inop A/P, F/D NO
Normal procedures	A/T ILS approach S/VS Missed approach S/XS
Taxi-out & Takeoff	One engine inop. Manual, F/D
T/O engine failure after V1 (1) S / US	
	Normal T/O .manual Row data
T/O engine failure after V1 (3) S/US	F/D ILS, T&GO S/US
	Loss of both engine driven gen <u>S/UP</u> A/P, A/T, F/D VOR approach <u>S/UP</u>
Climb	Circle to land rejected landing S/US
	VISUAL TRAFFIC PATTERNS S/145
	Taxi -in & park
Econ climb	Normal procedures S/US
REMARKES:	
PR	DGRESSING.
INSTRUCTOR NAME: INSTRUCTOR NAME:	SIGNATURE: Algorithm
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2-HAU DE SOIVISA	2
	LES IAME_ <u>KHEDR</u> _ <u>ABDALAA</u> IIRCRAFT TYPE <u>BOIEN 737-300/400/50</u> Briefing Set up MCP, CDU <u>S/125</u> Preflight Set up MCP, CDU <u>S/125</u> Preflight Set up MCP, CDU <u>S/125</u> Climb start Normal procedures <u>S/125</u> Climb Normal procedures <u>S/125</u> Climb REMARKES: PR

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INME KHEDR ABDALA	LESS &	CREW POSITION: CAPTAIN / J-+C	Ĩ
MRCRAFT TYPE BOIEN 737-30	0/400/500	DATE OF COPELETION 10-05-0	3
Briefing Set up MCP , CDU Engine inoperative flight characteris Preflight Set up MCP , CDU After start checklist Engine start Normal procedures Taxi-out & Takeoff Rejected T/O T/O engine failure after VI T/O engine failure after VI Wind shear near VR Climb Normal procedures	STAR	Full stop kinding Taxi -in & park Normal procedures	S/25 S/25 S/25 S/25 S/25 S/25 S/25
REMARKES:			
	GooD	PROGRESS	
		A BEST WINNER	0.000.01%
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INSTRUCTOR NAME:	SONRAT	TRAINING M	

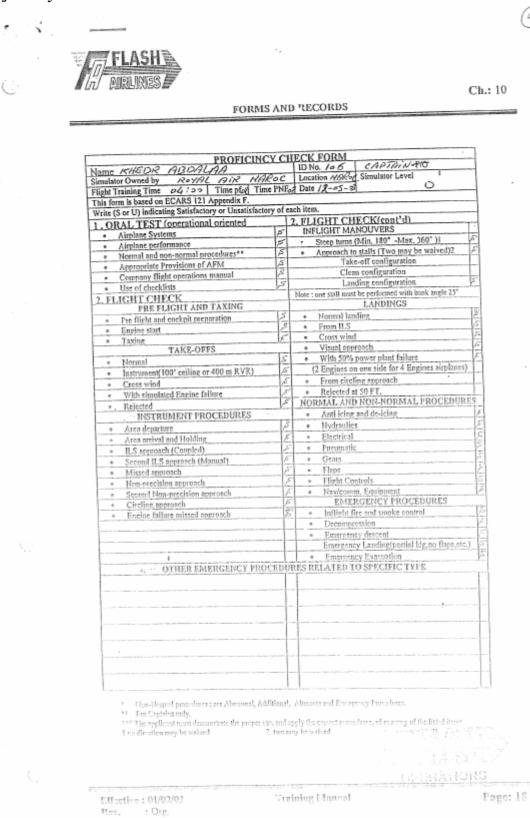
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FORMS AND RECORDS

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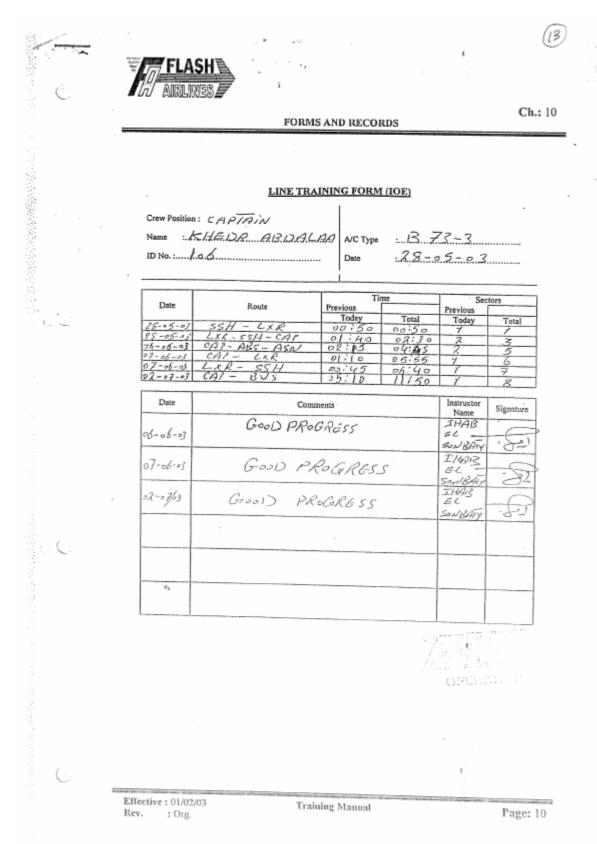
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FORMS AND RECORDS

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LINE TRAINING FORM (IOE) (cont'd)

Date	Route		Time		lors
	Route	Previous	32:45	Previous	14
09:7-03 130	A-CAI	Today	Total	Today	Total
and the second se	W-CAP	04:00	36:45	1	1.5
6-07-03 CA		02:05	38:50	1	16
7-07-05 5561	2311	01:00	39:50	1	17
9-07-1 SS14	- CAI	01:00	40:50	1	18
		00:45	41:35	1	10
Sorra LA	e - SSH	00:45	42:20	1	20

				1 20
Date	Commen	15	Instructor Name	Signature
10-07-03	Groold PROGRE	551	IHAI3 EC	R
17-07-05	STILL PROGRE	EssiNCI	SONIZOTY ZHAIZ EL	-0-
23-07/03	CLOOD PROGRE RGDY FOR CH		SONBATY IHAB GL SONBAK	-£1
				0-
			,	
taince's Signat	ше :	Training Manager:	1000	Pil
		D. souriers	CPL CPL	

Note : 2 Sectors must be conducted from right hand seat (RHS) for Captains (one sector PF and one Sector PNF)

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LINE CHECK FORM

THE FOLLOWING ITEMS MUST BE COVERD DURING LINE CHECK () Indicates that item has been checked

) Indicates that item has been checked 1. FLIGHT CHCK		DES	CENT AND APPROACH	
PRE FLIO	GHT			, SNOWTAM and braking action*	T
	Dispatch			Descent planing	- 6
• Re	parting for duty	5		Approach briefing, stars and	f
• Co	inputerised and ATC flight plan	3		Approaches:	-ř
	eather briefing, T.O. and landing min.	5		EPrecision & N.precision E Visual	
	emate planing Wx min	5		Destination and alternate weather minima	ť
	T. Operations Specifications*	5	LAN	DING AND TAXI IN	
	TAM briefing and "B" snags	5		Landing technique	T
	bin crew safety briefing	13		Use of auto breaks and reverse trust	-ŀ
	Cockplt			After landing and taxi in procedure	ť
• Te	clinical log and B sages	ø		2.KNOWLEDGE CHECK	
	EL-CDL and the effort on T.O/Landing	15	A) E	light operation manual	
	formance	13		IOE, Initial release, USV and Command	1
 Air 	craft library and documentation	5		Responsibility	
	ckpit preparation-FMS/FMGS/PMS	3	1	Crew luggage costent	÷
	KE OFF BRIEFING	51	4	The difference between planning and actual	ł
	ad, trim sheet and NOTOC	5		Weather rain, and Wa min, for new captain.	
	d Wx operation* Hot Wx operation	6		Fuel policy	-
	Performance, T.O speeds and C.G	3		Windshear, thunderstorms and turbulence	
* Em	cino start procedures	5		Fueling with PAX on board	-k
	KE-OFF AND INITIAL CLIMB	-Maria		Dangerous goods	-
and the second second second	h back procedures	3		Shoulder homess, seat belt policy and cockpit door	-
	ti speed and braking technique	5			÷
	Foll and VI cencept	B		First officer T.O. and landing ECARS 121	4
* No	ise abatement procedure and initial climb	51			4
* Be	angle, hest rate and turbulence speech	-1-1		Flight operations manuals & answers	
	a departure, SR) and holding	5		reraft performance and technical knowledge	
RUISE	is a second state of the second s	16.1		Operational system knowledge	_
	eht level selection, specific mage and OPT.ALT	10	8	T.O perference limits	_
r Ste	n climb and fuel savine	12	8	Wet and contaminated runways	į
	ise match no. and wanosuver capability	3	*	Reduced (flex) threat	ŀ
+ Use	of weather radar and weather avoidance	5		Annrouch and husling climb performance	-8
	PS and MORA (Special routes)		9.	Normal, non normal and emergency procedure	ŀ
	f down procedures	뉟	0	Flight patterns	14
	oute alternate and Emergency Proc. (NAT)*	2		cly procedure	
* AR	one anernate and chergency Proc. (NA) (*	12	ę	Communication between cockpit and eatin	1.
	ainsue wester mutima ainsue fuel for diversion(Alternate+Holdine)	55		Emergency evaluation procedure	
 Mili Cor 	nummication failure procedures	E.	*	Prepared/upprepared emergency	
		쾨		Bomh on bornd and least risk location	
 1.115 	ht control comm. Procedures (Stockholm radio)	3	4	Crew is INCAPACITATION	Ľ

' if applicable

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LINE CHECK FORM (cont'd)

Crew position : B Capt. Name : K.H.D.R	© F/O □ F/E (I.A.H.Code NoI.a.G	Алс Турс :Валала. 237-3-0
@Final line check	C Recurrent Line Check	Route check*
Route	No. of sectors	Flight Time
5517 - LXR - LXR - 5511	2.	01:30

	KNOWLEDGE	US	S
A/C Systems, Limi	tations and Performance		~
Normal Procedures		- 1	4
Operation manual a	and ECARS		2
Non-Normal Proce	dures*		6
	FLYING SKILLS	US	S
Attitude flying and	correct trim technique		-
Use of FMC, PMS,	FMGS, clc		1
Complying with SC	DP (Normal, Abnormal & Emerg.)		1
	ration, Attitude & Speed control		L
Flying accuracy &	Smoothness		L
	MANAGMENT	US	S
	use of FMC, PMS, FMGS, etc		L
	and use of available resources		1
	ances and safe heights		9
Situational awarene	155		0
45	(3021) STANI		
	SATESPACIO	RY	
	PARCed as capt in com		-ZAR
he Can			107-10
			107-10
Check Air man N	ame: .XHQB.A.G.SortBHY Check Airmon		107-10
	ame:	1's signature:	107-10
Check Air man N ID No. Check Result	ame:	ature Anager:	107-10 12-1 12-1
Check Air men N ID No. Check Result	ame: .XHQB., E.G. SordBHY Check Airman 	ature Anager:	107-10 12: 12:

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	FORM PILOT'S RECUR Name KHEOR ABIJALLA	S A	ND RECORDS
	PILOT'S RECUR		TATION COMPANY AND ADDRESS OF THE RESIDENCE AND ADDRESS AND ADDRESS ADDRE
	PILOT'S RECUR		
	PILOT'S RECUR		
	FILUTS RECOR	12121	NT TRAINING FORM
		100	IDNO. BOIENG 737-305/400
	Simulator Owned by RoyAL Aik HARoCo		Location CAZA Simulator Level
	Flight Training Time o 4 and Time PEAP Time P	NFa	Date 16-12-+3 10
	PART ONE ; GROU	UND	D TRAINING SEGMENT
	[) indicates that item has been covered.		
	 n) OPEN BOOK OUIZ (O&A)* 		b) Briefings
	 Airplane Systems 	2	
	 Aisplane performance 	E	
	Normal and non-normal procedures** Appropriate Provisions of AFM	2	
	 Appropriate Previsions of AFM Company flight operations and route 	15	
	 PharaohAir Operation Specifications 	12	
	PART TWO : FLI		T TRAINING SEGMENT
	Scenario :		
	PRE FLIGHT AND TAXING		LANDINGS
	 Pre flight and cockpit preparation 	12	 Noopal landing
3	 Engine start 	15	
	 Taxing 	15	
	TAKE-OFF5		Visual approach With \$0% never plant failure
	Normal	E	
	 fastrument(100' ceiling) Cross wind 	5	
	 With simulated engine failure 	1	a In Windshear conditions
	 Rejected 	15	 Rejected at 50 FT.
	 Windshear during take-off 	12	NORMAL AND NON-NORMAL PROCEDURES
	INSTRUMENT PROCEDURES		 Anti icing and de-icing
	 Area departure 	18	
	 Area acrival and Holding 	. 8	• Electrical
	 ILS approach (Coucked) 	12	• Pneumatic
	 Second ILS approach (Manual) Missed approach 	- 12	s e Gears s S e Flans s
	 Massed approach Non-precision approach 	-6	* Flight Controls
	 Second Non-precision approach 	N 12	* Nav/comm. Equipment
	+ Circling approach	13	S EMERGENCY PROCEDURES
	 Engine failure missed approach 	12	
	INFLIGHT MANEUVERS		Decompression
	 Steep turns (Min. 180° -Max, 360°) 	15	5 • Emergency descent
	 Approach to stalls 	14	 Emergency Landing(partial l/g,no flaps,etc.)
	 Specific flight characteristics 		
0	TRAC DECET OF Ind	KRG L-	ZENCY PROCEDURES
	RUSH		2
	1	-	

QAA spection and resovers
 Mon-Hammil procedures : are Alwarmal, Additional, Alexander and Energy and Procedures,
 For Captains only.

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PILOT'S RECURREN					
RHS TRAINING FOR INSTRUCTORS	RHS TRAINING	FOR CAPTAINS			
Error recovery	 Normal take Off 				
Lateral offsets	 Simulated Engine fail 	gine failure - Take off			
Vertical Offsets	 One Engine Out-Approach and landing 				
 Minimum 3 Touch and Go 	 Minimum 3 Touch an 	d Go's			
EV/	LUATION				
KNOWLEDGE	A STATE	JS S			
FLIGHT OPERATION MANUAL (FOM)and Rel	evant ECARs				
A/C systems Limitations and Performance		~			
Normal Non-Normal Procedures*		6			
PHARAOH AIR Operations Specifications		2			
FLYING SIGLLS	1	US S			
Compliance with SOP (Flight operations Manual &	FCOM)				
Attitude flying and correct trim technique					
Use of FMC, PMS, FMGS, etc					
Aeroplane configuration, Attitude &S speed control	d				
Flying accuracy & Smoothness					
MANAGMENT		US· S.			
Compliance with FLIGHT OPERATION MANUA	L (FOM)				
Planning ahead and use of FMC, PMS, FMGS, etc					
Crew coordination and use of available resources					
Adherence to clearances and safe heights					
Situational awareness					
Cabin crew safety briefing					
SATES	EACLORY				
	slid (through Last day of) ;	Next Event			
Base Month (through Last day of) : License V Month Year Month	alid (through Last day of) ;	and a second sec			
Base Month (through Last day of) : License V Month Year Month Date of last 3 take-offs &		Next Event Proficiency check			
Base Month (through Last day of) : License V Month Year Month Date of last 3 take-offs & 1, /	alid (through Last day of) : Year	and a second sec			
Base Month (through Last day of) : License V Month Vear Month Date of last 3 take-offs & 1, /	alid (through Last day of) : Year / 2. / /	Proficiency check			
Base Month (through Last day of) : License V. Month Year Month Date of last 3 take-offs & Laiddings** : 1. / Name*** CP. IP ID THAB EL SONBATY	alid (through Last day of) : Year / 2. / /	Proficiency check			
Base Month (through Last day of) : License V Month Year Month Date of last J take-offs & Lafidings** : 1. / Name*** QP IP ID THAB EL SONBATY	Alid (through Last day of) : Year 1 2. 1 1 No. 10.7	Proficiency check			

Nen-Normal procedures : see Abnoonal, Additional, Alternate and Emergency Procedures.
 Traines is responsible for the accuracy of this data, and he must sign the form,
 CP: CheckAirman, IP dispructor Pilot.

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vii- Personal situation

The captain was married and had 3 children ages 29, 25 and 18 years. The eldest son is married and is doing post graduate studies in USA. The second son is an engineer. The youngest daughter is still studying in university.

The captain has no known problems of any kind. He is known to be devoted to his family. He did not suffer from any abnormal health or social problem.

(Refer also to page 72 of the Factual Report (Interviews regarding Captain Kheider Abdullah)

1.5.1.3. 72-hour history of the captain:

Refer to interviews on page 73 of the FR. The captain and F/O left Cairo to SSH on January 1st, 2004 as passengers on Flash Airline flight departing Cairo at 15:00 GMT No more factual information could be obtained regarding the 72-hour history. 1.5.1.4. Interviewing the individuals who trained and flew with the captain (including ground and simulator instructors)

Interview with Captain/ Essam Eldin Brahmin Chief Pilot and instructor ATR 42 Scorpio Airlines during the period of employment of Captain/ Khedr in this Airline.

- How well did you know Captain/ Khedr?

He was a colleague during work at the Egyptian Air force and when he joined Scorpio, we worked together as I was Chief Pilot. I was in charge of organizing his flying schedule and monitoring his standard through line checks.

He was a well disciplined pilot, observed his flying schedule without any problems, was always careful to observe duty time limitation and rest periods, had good relations with his colleagues, was cheerful with his crew and always prepared his flight carefully.

During line check he performed well. He was attentive to his work, communicated well with his crew and was not tense. His previous experience on military air transport made him comfortable in flying commercial air transport with relation to route experience and airway flying requirements.

- What routes were flown at this time?

Mainly domestic flights.

- Was Sharm El Sheikh one of your common destinations? Yes.
- What was the common departure procedure Followed out of Sharm El Sheikh? The standard procedure followed was depending on the runway in use a turn was initiated towards the sea while climbing in a wide pattern to cross the VOR 11000 Ft to proceed on the 306 Radial to Cairo.
- Did you as chief pilot and instructor see or have any report of any kind about Captain/Khedr?

All comment and observations were good Captain and comfortable to work, always well prepared for his flight and kept his cockpit organized.

- Why did he leave Scorpio?

He left when the company stopped operations.

Interview with Captain/ Emad Sallam Instructor Pilot on C130 In the Egyptian Air force At the time Captain/ Khedr started to fly in the military air transport.

- How well did you know Captain/ Khedr?

As a pilot in the Air force we were colleagues although he was more senior than I, when he moved from the fighter squadrons to the air transport and when assigned to the C 130 I was an instructor and when he was assigned to training flights under my command was very willing and had no attitude about my being instructor with less seniority, he was always eager to learn and very attentive in the cockpit had no problem in asking for information from the crew with him and did not exercise unnecessary authority due to his rank, listened well to comments and observations of all the crew members without regard to rank and seniority was cheerful but well disciplined his training progress was standard.

Interview with Captain/ Essam Eldin Ibrahim Chief Pilot and instructor ATR 42 Scorpio Airlines during the period of employment of Captain/ Khedr in this airline.

- How well did you know Captain/Khedr?

He was a colleague during work at the Egyptian Air force and when he joined Scorpio we worked together as I was Chief Pilot I was in charge of organizing his flying schedule and monitoring his standard through line checks.

He was a well disciplined pilot observed his fighting schedule without any problems was always careful to observe duty time limitation and rest periods had good relations with his colleagues was cheerful with his crew and always prepared his flight carefully.

During line check he performed well was attentive to his work communicated well with his crew and was not tense his previous experience on military air transport made him comfortable in flying commercial air transport with relation to route experience and airway flying requirements.

- What routes were flown at this time?

Mainly domestic flights.

- Was Sharm El Sheikh one of your common destinations? Yes.

- What was the common departure procedure Followed out of Sharm El Sheikh?

The standard procedure followed was depending on the runway in use a turn was initiated towards the sea while climbing in a wide pattern to cross the VOR 11000 Ft to proceed on the 306Radial to Cairo.

Did you as chief pilot and instructor see or have any report of any kind about Captain/Khedr?

All comment and observations were good Captain and comfortable to work, always well prepared for his flight and kept his cockpit organized.

- Why did he leave Scorpio?

He left when the company stopped operations.

- 1.5.1.5. Interviewing CAA inspectors who flew with captain. Interviews to be carried out by OPS group
- 1.5.1.6. Interviewing former head of operations in Flash Airlines (No official former head of operation in Flash Airlines)

1.5.1.7. Additional factual documentation (Captain)

Number of days the captain had been working since his last day off.

Operation Department.

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Date	Route	Remark	Starting Time	Ending Time	
SAT.03/01/2004	SSH/CAI/CDG		03:00	XXX	
FRL 02/01/2004	SSH/TRN/SSH	1	04:30	12:45	
THU 1/01/2004	OFF			,	
WED.31/12/2003	ČAI/CDG/CAI	<u> </u>	12:00	23:00	
TUE.30/12/2003	OFF				
MON.29/12/2003	CAI/BCN/MAD/LXR	Heavy	06:25	19:15	
SUN.28/12/2003	OFF				
SAT.27/12/2003	LXR/CDG		11:00	16:00	
FRI.26/12/2003	CAI/BCN/MAD/ASW	Heavy	06:45	19:15	
THU. 25/12/2003	SSH/CAI		04:00	05:00	
WED.24/12/2003	SSH/LXR/SSH		13:45	16:15	
TUE.23/12/2003	SSH/AOI/BRI/SSH		15:30	01:00	
MON.22/12/2003	CAI/BCN/MAD/LXR	Heavy	06:45	19:15	
SUN.21/12/20036	LXR/SSH/NAP/BRI/SSH	Heavy	04:15	15:45	
SAT.20/12/2003	CDG/LXR		06:00	11:00	
FRL19/12/2003	OFF				
THU.18.12/2003	CAL/SSH/CAL		17:00	19:30	
WED.17/12/2093	-			19.50	
TUE.16/12/2003					
MON.15/12/2003		· · · · · · · · · · · · · · · · · · ·			
SUN.14/12/2003	OFF		· · ·		
SAT.13/12/2003		·		· · ·	
FRL12/12/20003					
THU.11/12/2003		· · · · · · · · · · · · · · · · · · ·	└── ·──		
WED.10/12/2003		· · · ·			

CAPT. KHIDR ABDULLAH

Note:

The captain and F/O left Cairo to SSH on January 1st, 2004 as passengers on Flash Airline flight departing Cairo at 15:00 GMT

Captain interpersonal characteristics, including perceptions of fellow pilots regarding their capability for assertiveness.

All available information is available in pages 72-73 Factual Report

Familiarity of the two flight crew members with each other. (Including number of legs flown together this trip, number of legs flown together in the last 30 days.

According to the available information, the accident flight was the 3rd sector in the last 24 hours.

Description of how well the flying crew got along. No information available

Reported proficiency information. Outcome and comments from training records and proficiency check forms. Refer to 1.5.1.2 (vi)

Spatial disorientation or upset recovery training received at Flash Air or in the military. AI196

According to CAA regulations, Spatial Disorientation training is not mandatory

No available documents from Flash Airline concerning SD training. Some verbal reports from the Egyptian Air Force are available concerning the captain SD training the time he was serving in the Egyptian Air Force as a military fighter pilot.

Inputs from different investigation partners are needed.

According to and CAA regulations, Upset Recovery training is not mandatory Upset Recovery Training recommendation should be included in the Recommendations Chapter.

Captain's flying proficiency and cockpit style from fellow pilots, instructors, and/or check pilots.

Refer to 1.5.1.4 and 1.5.1.2 (vi)

Flash Airlines chief pilot view regarding the departure procedure from SSH, based on company procedures

According to Chief Captain Flash Airline and all other pilots questioned about departure procedure from SSH, all agree that a turn towards the sea is initiated with a bank angle depending on available rate of climb and captain's discretion to cross the VOR on course radial 306 at or above 10500 ft.

Number of departures from SSH previously made by the captain (day and night)

Within the last month, the captain has made five departures from SSH including the accident flight.

(SAT 03-Jan-04 (night), FRI 02-Jan-04 (night), THU 25-Dec-03 (night), WED 24-Dec-03 (day) and TUE 23-Dec-03 (day))

The captain's time on Russian aircraft (MiG-21). Hercules transport aircrafts C130 (dates and number of hours). ADI display configuration in comparison with B737-300 ADI display. Refer to captain CV, and item 1.5.1.2 (vi)

For B737-300 ADI refer to 1.16.1.9 (reference CairoMarch04Slides (March Progress Meeting - Cairo).pdf file)

Comparison with ADI Displays for other airplanes types might be made by the OPS group if needed

1.5.2. The First Officer

- 1.5.2.2. Background information .
 - i- Beginning of his flying career.
 - The F/O began his ground training on the aircraft type 737-300 at Luxor Airway from 4 May 2002 to 16 May 2002
 - The F/O completed the Full Flight Simulator Training and the Flight Training at Flash Airline on 30 June 02
 - ii- All airlines worked for prior to Flash Air Refer to previous item
 - iii- "All" F/O training records at Flash (including his last recurrent training). All flying hours before Flash were different training phases

License Renewal Form (Boeing 737-500):

وزارة الطيران المدنى قطاع العمليات والنقل الجوي الإدارة المركزية للعمليات الجوية الإدارة العامة لإجازات الطيران إخطار تجديد إجازة طيار السيد الطيار / مدير عام العمليات مؤسكرة / شركة فل مرس للصر تحية طيبة وبعد .. بالإحالة إلى الطلب المقدم من السيد / محصر محصر محمد كليم مشا ضعر بخصوص تجديد إجازة / مُجْمِع مِتْبَ مِكْمَ رقم مَنْهُ هَا مَنْ الحاصل عليها نتشرف بالإفادة بأنه تم تجديدها من ٢٠ / ٢٢ / ٢٠.٣ إلى ٤ / ٥ / ٤ . ٢٠ على طراز : ______ علمًا بأن 4 200 / GM 3° / 6 / 200 / 1200 علمًا بأن 4 1200 وانتهاء اللياقة الطبية في ٢٠٠٤ م ٢٠٠٤ وتفضلوا بقبول فائق الإحترام.. مدير عام إجازات الطيران and delivers العربة المامة لشتون الطابع الأميرية ٣٤٥٠ من ٢٠٠٢ = ١٥٠٠٠

Certificate of Validity of a license:

. جمهورية مصر العربية وزارة الطيران المدنى قطاع العمليات والنقل الجوي شهادة سريان مفعول إجازة طيار ١ - حالة هذه الشمادة بالنسبة للإجازة -هذه الشهادة جزء من إجازة طيار – رقم _______ ويجب وجودها دائمًا بالإجازة . ٢ - سريان مفعول الإجازة -حامل الإجازة التي تعتبر هذه الشهادة جزء منها كشف عليه طبيًا بتاريخ ٥ / ٥ /٣~) وجد لاتتًا للعمل وقتًا للاشتراطات الموضحة بالإجازة كما إنه قد أتم جميع الإجراءات لتجديدها وعليه فهي سارية G. المفعول للمدة من ٢٠ / ٢٢ / ٢٠ الى ١١ ٥ /٤ ~ ٢ على طراز ٥ ٤ / ٥ /٤ على طراز إلى _/___ على طراز ~ فترةالسماح الشهرالأساسى 442 5-2 6-2 36

ARAB REPUBLIC OF EGYPT MINISTRY OF CIVIL AVIATION SECTOR OF OPERATIONS AND AIR TRANSPORT CERTIFICATE OF VALIDITY OF A LICENCE FOR PILOT's OF FLYING MACHINES

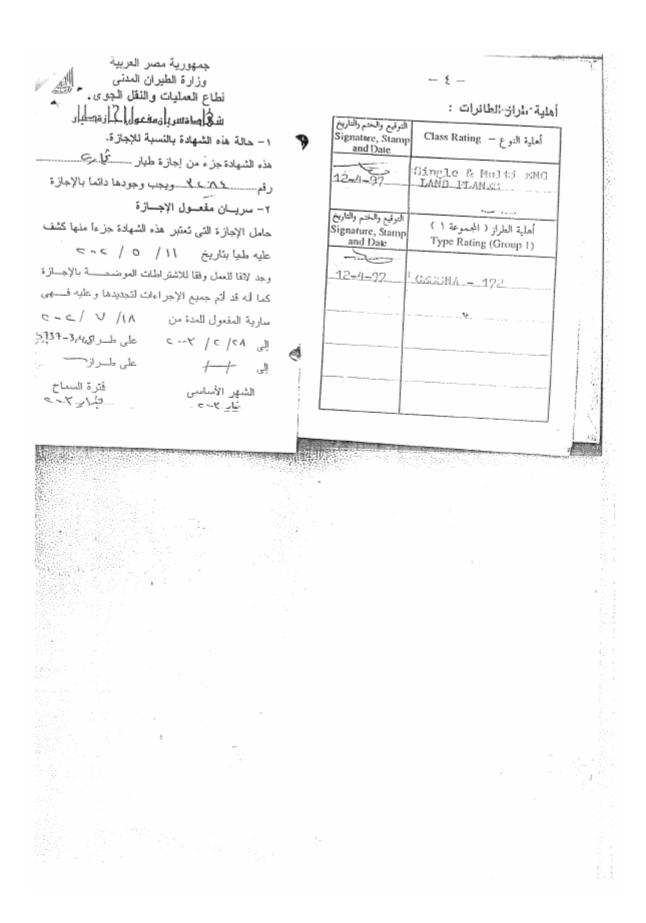
1 - Status of this certificate . This certificate forms part of -'o M pilot's licence flying machines number -32-8-1 and must always be carried with the licence . 2 - Validity of the licence

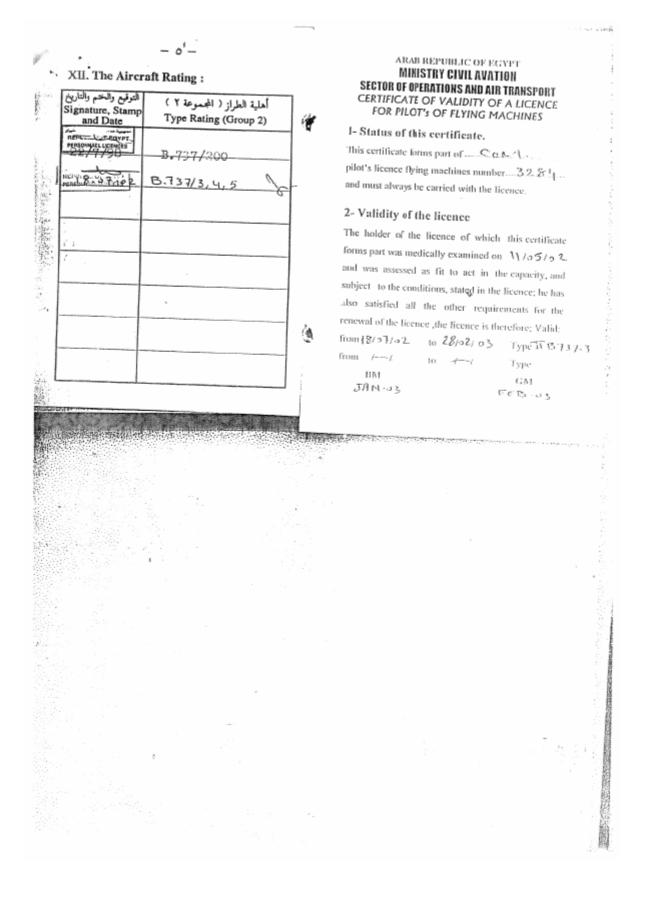
The holder of the licence of which this certificate forms part was medically examined on 515103 and was assessed as fit to act in the capacity, and subject to the conditions, stated in the licence; he has also satisfied all the other requirements for the renewal of the licence, the licence is therefore; Valid :

from 3 0 /1 2/03, to 415/04 Type IB from to 1----1 BMGM May 2-4 Junzay

Туре ____

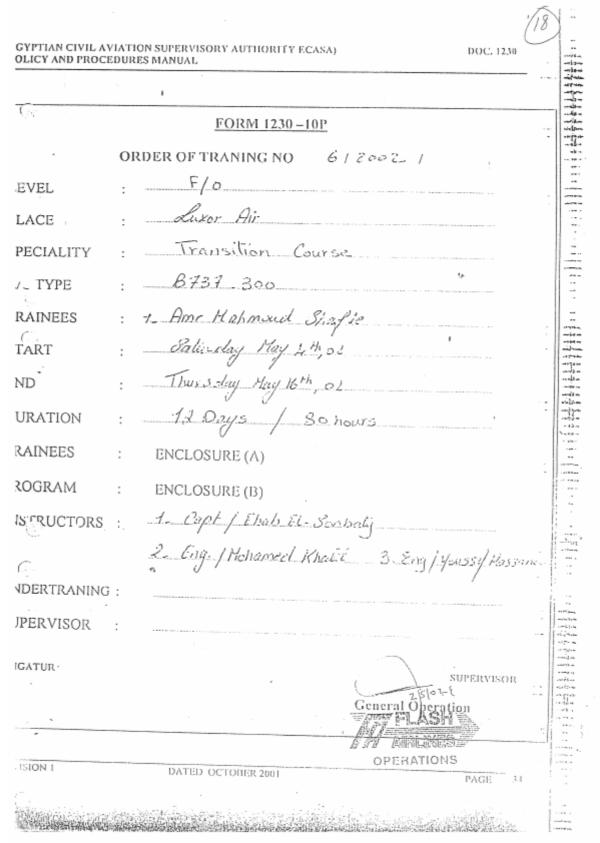
6 3. 635 (استمارة رقم ۳۹ ط.م.ر المكررا صدرت فأده الإجازة بموجب المرسوم بتبانون رقم ٢٨ يتاريخ ٢٢ مايو سنة ١٩٨١ واللحق إجازة طيار تجارى رقم المعاهدة الطيران المدنى الدولية الموقعية (ja COMMERCIAL 市民観察寺寺の合い في ٧ ديسمبر سنة ١٩٤٤ . ILOT'S LICENCE (7 4.5) الاسم عمرو محموف عبد الحليم DATE OF BIRTHS \ /o\ /1979 IV. Name Amr-Mahmoud_shal'o-الدوان- 7- 11 سوا 1 لمعلمين الموافع ملائل ٧. يصرح لحامل هذه الإجازة بقيبادة الطائرات 20-Belesraa-elmoslmeen-elmohandse VI. Nationality EGXEPLAN الآليسة الأفقل من الهسواء طبسقما للشمروط XII, Signature of 122 توتيع حامل الإجازة } والمواصبقيات المبينة بالإجبازة على أن يكون Holder . Signature of Issning Officer توقيع المرخص لهم حاصلا على استدارة رقم ٢٨ (ط.م.ر.) بإصدار الإجازة الختم والتاريخ سارية المفحول . بترخيص من رئيس مجلس إدارة⁽¹⁾ الهيئة المصربة المامة للطيران المدني Date and Stamp G) XI. By Authority of the C.A.A. . C.A.A. . 12-4-12 186





. Rating Contained in Licence is Valid ٣- أداية مدرب المعتمدة بمسهده الإجسازة مسارية ------4.7 Туре -----المفعول إلى _/__/ طراز ---The Privileges of an Instrument rating ٤ – أهلية الطير ان الألى المعتمدة بالإجسازة تخسرل contained in the licence may be exercised as لحاملها الحق في العمل كقائد طائرة أو كطينسار pilot in charge or as co-pilot (where one is مساعد (كما نقضى الحالة) على الطائرات الألية. required to be carried) of a flying machine. C -- C / V /IA From 18/07/02 مىن TO 31/07/03 c .. Y / YI إلىسى CERTIFICATE I the undersigned, a person fully author شهــادة أنا الموقع أدناء بمقتضى السلطة المخولة لي مسمن رئيس for this purpose by the chariman of قطاع العطبات والذقل الجوى بسوزارة الطميران المدنس SECTOR OF OPERATIONS AND بجمهورية مصر العربية أقر بصحة ما جاء بسالبنود TRANSPORT of the Arab Republic of Egypt ۲٫۲٫۲٫۱ من دذا الممستند. hereby certify the Facts stated in Paragraphs التوقيع : ----1 2:3,4 التاريخ : ٢٢ / ١ Signature 22,7-102 Date Stamp الختسم ا La sa set Contraction of the 頭印

B737-500 Transition Training:





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Boeing 737-300

Ground Training Syllabus

	SYSTEM	HOURS REQUIRED
	Weight& Balance	4 HRS
	Air conditioning,presurization	5 HRS
	Flight Controls	8 HRS
	Hydraulic	3 HRS
	Landing Gear	3 HRS
	Navigation	5 HRS
	Auto Flight	10 HRS
	F.M.C	10 HRS
	Pneumatic	3 HRS
	Electric	4 HRS
	Anti -ice	3 HRS
	Oxygen	3 HRS
	Engine	5 HRS
	Fuel & APU	3 HRS
	Performance	10 HRS
	Total	80 HRS
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eph T	to St., El-Nozha El-Godidab, Calva Essent	ODER ACLONS

58, Joseph Tito St., El-Nozha El-Gedidah, Cairo,Egypt. ol.: 202-2944700-800-550 Fax: 202-2941300 SITA: CAIHPCR E-mail:hpline@internetegypt.com

Proficiency Check (June 30, 02):





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Form No. 02 - 2/2

Error recovery Lateral offsets Manual Liks (CAT 1 minima) Vertical Offsets Manual Liks (CAT 1 minima) Vertical Offsets Manual Liks (CAT 1 minima) Vertical Offsets Month Cecking approach and landing. Minimum 3 Touch and Go Simulated Engine Gailane – Take off One Engine Out-Approach and landing. EVALUATION KNOWLEDGE EVALUATION KNOWLEDGE US S S S Compliance with SULL (FOM) and Relevant ECARs U W Verteral Diffections FLYING SKULLS US S S Compliance with SUP (Flight operations Manual & FCOM) Us TLYING SKULLS US S Compliance with SUP (Flight operations Manual & FCOM) Us TLYING SKULLS US S Compliance with SUP (Flight operations Manual & FCOM) Us Very MANAGMENT US S Compliance with FLIGHT OPERATION MANUAL (FOM) U landing ahead and use of FMC, PMS, FMGS, etc TL MANAGMENT US S Compliance with FLIGHT OPERATION MANUAL (FOM) U landing ahead and use of FMC, PMS, FMGS, etc TL MANAGMENT US S Compliance with FLIGHT OPERATION MANUAL (FOM) U landing ahead and use of FMC, PMS, FMGS, etc TL MANAGMENT US S Compliance with FLIGHT OPERATION MANUAL (FOM) U landing ahead and use of FMC, PMS, FMGS, etc TL MANAGMENT US S Compliance with FLIGHT OPERATION MANUAL (FOM) U l W Compliance with FLIGHT OPERATION MANUAL (FOM) U W Compliance with FLIGHT O	RHS TRAINING FOR INSTRUCTORS	uirement and should be cov			
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This form is based on ECARS 121 Appendix F.		AIRCRAFT TY	PE: B 73	7-300140015	30
Write (S or U) indicating Satisfactory or Unsatisfactor	ary of ca	ch item.		1000 CT	
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Airplane Systems	2	INFLIGHT MA			15
 Airplane performance 	SI			-Max, 360°)1	15
 Normal and non-normal procedures** 	121			may be waived)2	. 7
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2. FLIGHT CHECK		Note : one stall, must		with bank angle 25*	
PRE FLIGHT AND TAXING			LANDING	35	r.
 Pre flight and cockpit preparation 	12	 Normal lai 	aling		5
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 Area arrival and Holding 	5	 Electrical 			5
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Non-Nermal procedures : are Abnormal, Additional, Alternate and Emergency Procedures.
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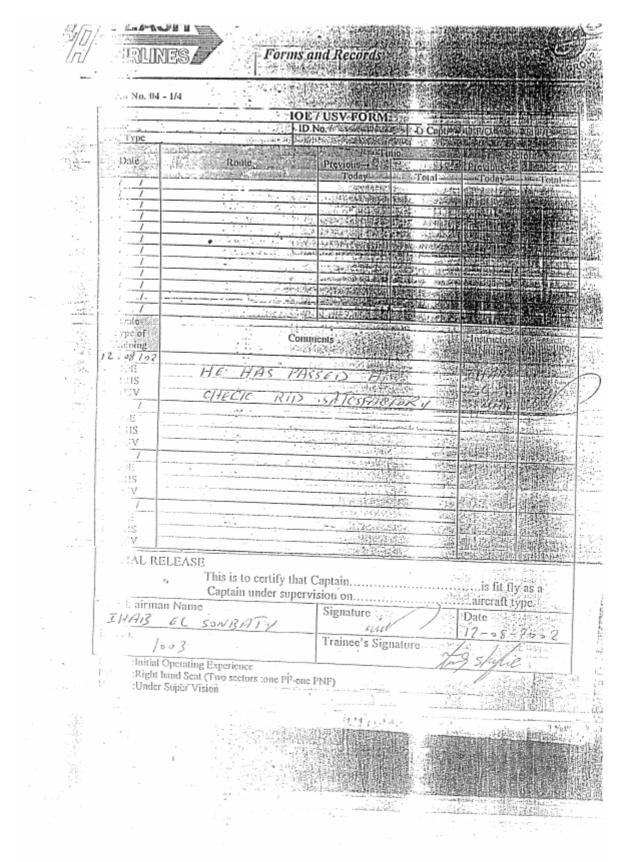
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Note:

Heliopolis Airline operation ceased operation and Flash Airline took over its traffic rights and operated under the name of Flash Airline

Flight Training (August 12, 02):



Form No. 04 -	1/4	•			
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IOE :Initial Operating Experience RHS :Right hand Sent (Two sectors tone PF-one PNF) USV ::Under Super Vision

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Form No. 04 - 4/4			
	IOE / USV FORM (Cont'd)		
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Flight Deck Ground Training/ Competency Check/ General Emergency (22-05-02):

زر فلز ش \bigcirc EGYPT AIR TRAINING DIVISION قطـــاع التدريب Gen. Dept. for Aviation Training الادارة العامه لتدريب الطيران E.T. C . مركز تدريب الطوارئ COMPETENCY CHECK. Flight deck Ground Training / COMPETENCY Check GENERAL EMERGENCY NAME. **Crew Position** Amr Mahmous shelie Code: X Cap D F/O , D ~ F/E D DATE : LOCATION INITIAL E.T.C 22-05-2002 RECURRENT RE-QUALIFICATION í ALL ITEMS MUST BE COMPLETED CHECK (/) INDICATING COMPLETION EACH ITEMS COMPETENCY CHECK ITEMS PART 1 EMERGENCY SITUATION 化基本因为 化合理 化合理 化合理 - Flight CREWMEMBER DUTIES AND RESPONSIBILITIES S - CREW COORDINATION AND COMPANY COMMUNICATION S - AIRCRAFT FIRES S - FIRST AID EQUIPMENT S - ILLNESS, INJURY, AND BASIC FIRST AID S - GROUND EVACUATION S - DITCHING S - RAPID DECOMPRESSION S - PREVIOUS AIRCRAFT ACCIDENTS/INCIDENTS S - CREWMEMBER INCAPACITATION S - HIJACK AND BOMB THREAT S PART2 EMERGENCY DRILL - HAND-HELD FIRE EXTINGUISHERS s - PORTABLE OXYGEN SYSTEM s - EMERGENCY EXITS AND SLIDES.* S - DITCHING EQUIPMENT. ** S INSTRUCTOR NAME CODE NO. 8028 INSTRUCTOR SIGNATURE AHMED HELMY RESULT E.T.C Manager Satisfactorily Completed TRAINEE SIGNATURE G.M. AVIATION TRAINING 1/06/ 6



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FORMS AND RECORDS

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Name AMR SHAFTE	CY	ID No. FO
Simulator Owned by ROYAL AIR MOK	000	Location of Oth Simulation I
	ne Ph	IF g Date 65, 16, 05, 0,3
This form is based on ECARS 121 Appendix P		1 3 10, 1 8 10h 013
Write (S or U) indicating Satisfactory or Unsatisfactor	ory of	each item.
1. ORAL TEST (operational oriented		
 Airplane Systems 	5	2.FLIGHT CHECK(cont'd) INFLIGHT MANOUVERS
 Airplane performance 	극중	 Steep turas (Min. 180* -Max. 360*)1
 Normal and non-normal procedures** 	-55	Anntasch to stalls (Ture and have a land
 Appropriate Provisions of AFM 	5	Approach to stalls (Two may be waived)2 Take-off configuration
 Company flight operations manual 	- Sicol	Clean configuration 5
 Use of checklists 	S	Landing configuration S
2. FLIGHT CHECK		Note : one stall must be performed with bank angle 25* 14
FRE FLIGHT AND TAXING		LANDINGS
 Pre flight and cockpit preparation 	5	Normal Isouing K:
 Engine start 	S	From ILS
 Taxing 	R	Cross wind
TAKE-OFFS		Visual approach
Normal	IS	With 50% power plant failure
 Instrument(100° ceiling or 400 m RVR) 	S	(2 Engines on one side for 4 Engines airplanes)
 Cross wind 	5	From circling approach Construction of a Congress airplanes)
 With simulated Engine failure 	SIS	Rejected at 50 FT.
 Rejected 	5	NORMAL AND NON NORMAL AND
INSTRUMENT PROCEDURES		NORMAL AND NON-NORMAL PROCEDURES Anti icing and de-icing
 Area departure 	15	a Hadevaller
 Area arrival and Holding 	North Har	Involutions Electrical S Precumatic
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 Engine failure missed approach 	2	EMERGENCY PROCEDURES
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	++	Emergency Landing(partial ldg,no flaps,etc.)
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PROFICIENCY	CHECK FORM (cont'd)		
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RHS TRAINING FOR INSTRUCTORS Error recovery	The second se		
Lateral offsets	RHS TRAINING FOR CAPTAINS Normal take Off		
Vertical Officer	Normal take Off Manual Line		
Minimum 3 Touch and Go	Manual ILS (CAT 1 minima) Non-Precision		
	Non-Precision approach and landing Simulated Engine Giles		
	Simulated Engine failure – Take off One Engine Out Action		
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FLIGHT OPED LINE AND WLEDGE	US		
FLIGHT OPERATION MANUAL (FOM)and Relev NC Systems, Limitations and Performance Normal, Non-Normal Performance	ant ECARs S		
Normal, Non-Normal Procedures*	S S S	-	
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FLYING SKILLS Compliance with SOP (Flight compliance)	2		
Autilude flying and the main operations Manual 4 mil	US		
Attitude flying and correct trim technique Use of FMC, PMS, FMGS, etc	COM) S	7	
Acroplane con G	S	-	,
Flying accuracy & Smoothness			
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Simulator Owned by NoyAL A	S.C. MARACO	Loc	ation HAROCO	Simulator Level	
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a) OPEN BOOK QUI	Z (Q&A)*			b) Briefings	
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 Airplane performance 	codures**		Review of nor	mal training Scenario;	12
 Normal and non-normal pro 			-Normal and	Non-normal procedures**	1
 Appropriate Provisions of A 			-LOFT		NAME AND AL
 Company flight operations : 			-Windshear		15
 PharaohAir Operation Speci 	ifications 2	•	CRM		1
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Scenario :					
PRE FLIGHT AND T	AXING			LANDINGS	
 Pre flight and cockpit prepa 	ration 🖉		Normal landin	11:	15
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 Taxing 	5		Cross wind		
TAKE-OFFS			Visual approx	ch	-6
 Normal 	2			ver plant failure	-
 Instrument(100° ceiling) 			(2 Engines on	one side for 4 Engines airplanes)	
 Cross wind 	18		From . ircling	approach	1
 With simulated engine failu 	7e 5		In Windshear		÷
 Rejected 	5		Rejected at 50		
 Windshear during take-off 	12	N	DRMAL AND D	NON-NORMAL PROCEDURES	e
INSTRUMENT PROC	EDURES	+	Anti icing and		ŤΤ:
 Area departure 	1.7		Hydraulics		ŤĒ
 Area arrival and Holding 	13		Electrical		-6
 ILS approach (Coupled) 	17		Presentie		-6
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 Missed approach 			Flaps		恃
 Non-precision approach 	2,		Flight Control	h	-6
 Second Non-precision appr 			Nawcomm, E		-6
 Circling approach 			EMERG	ENCY PROCEDURES	24
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INFLIGHT MANEL	IVERS		Decompressio		-ić
 Steep terms (Min. 180° -M 	ax, 360°)		Emergency de		-6
 Approach to stalls 	16		Emergency La	anding(partial l/g,no flaps,etc.)	官
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Q&A (question and answers)
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FORMS AND RECORDS

PILOT'S RECURREN RHS TRAINING FOR INSTRUCTORS	TRAINING BODY			
RHS TRAINING FOR INSTRUCTORS	TOURN (C	cont'd)		
Error recovery	Normal take Off	FORCAPTAINS		
Lateral offsets	Simulated Engine failure - Take off			
Vertical Offsets	One Engine fa	ilure - Take off		
 Minimum 3 Touch and Go 	One Engine Out-Ani	proach and landing		
	 Minimum 3 Touch a LUATION 	nd Go's		
KNOWLEDGE	1 m 1 m 1 m 1 m 1 m 1 m 1 m 1 m 1 m 1 m			
FLIGHT OPERATION MANUAL (FOM)		US S		
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Normal Non-Normal Procedures*		- L-		
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FLYING SKILLS		1 6		
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Compliance with SOP (Flight operations Manual &	FCOM)			
Attitude flying and correct trim technique Use of FMC, PMS, FMGS, etc				
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MANAGMENT		10		
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Non-Normal procedures : are Abnormal, Additional, Atternate and Emergency Procedures.
 Traince is responsible for the accuracy of this data, and he must sign the form.
 CP: CheckAirman, IP:Instructor Pilot.

Effective : 01/02/03 Rev. : Org.

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Training Manual

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FMRG TRAINING

قطىباع التدريب الإدارة العامة للتدريب الطيران مركز تدريب الطوارئ

نتيجة فرقة : تنشيطية للمسادة الطيساريسن العاملسين بشسر كسه فسلاش امر تدريب رقم :- ١٢٧ / ٢٠٠٢ (طوارئ) Aircraft Hight تاريخ بداية الفرقة :- ٢٢ / ٥ / ٢٠٠٢ First Aid Hazmat تاريخ لهاية الفرقة :-- ٢٣ / ٥ / ٢٠٠٢ Salatan. ملاحظات سلامة طانر ات اسعافات بضائع خطرة عملى الاستسم £ 97 ك / أيهاب السنباطي ناجح 1 . . 1 . . ۱.. ١ ناجح ك / اشرف زارع 97 ۱., ۱.. 1... ۲ _____ė راسب ب خطرة 1 . . ٩٦ 1 . . ك / ئـــور سعـــد ٣ ناجح 1 . . 90 1 . . ۱۰۰ ك / خريستو لوستانس ٤ ناجح ٩٥ 1 . . ۱.. ۱., ك / وانسل فكسرى ٥ ناجح ۱۰۰ 90 ٦ ۱۰۰ 1 . . ك / جمــال عــون ناجح 1 . . 90 ٩٢ ك / عمرو عبد الحميد ۱۰۰ Ŷ راسب <u>_____</u> 1 . . ٩٦ ك / علىي رئىساد ٨ ناجح 1 . . 90 ٩٦ 1 . . م،ك/على رشــاد ٩ ناجح 1 . . 90 1 . . 1 . . م ۵۰ ك / محمد حسنى ١. ناجح 90 1 . . ٩٢ 1 . . م ۵۰ ک /یاسر فکری 11 ناجح 97 1 . . 1 . . ۱۰۰ م ۱۰ ك / هبة درويش 11 ناجح 1 . . ۸١ 7 ۹ 1 . . م ۵۰ ک / شیریف ابو العز 11 ______ ر اسب ب خطر ۃ ٩٦ ۱۰۰ ۱۰۰ م ۰ ك / خالد كوثر ۱ : ناجح 90 ۱۰۰ 1 . . 1 . . م • ك / هاني المليجي ۱۰ 1... 100 90 95 ناجح do ۱., 1... ١٦ جم ٢٠ ٤ / عمر الشافعي 1.00 1... 100 90 95 ناجح 1 . . ۱۱ م ، ك /محمود حنفي ٢٠ ٩٦ 96 OBJERUEL ناجح 1 . . 90 مرحل / اشرف لملوم 1 . . ۱... 17

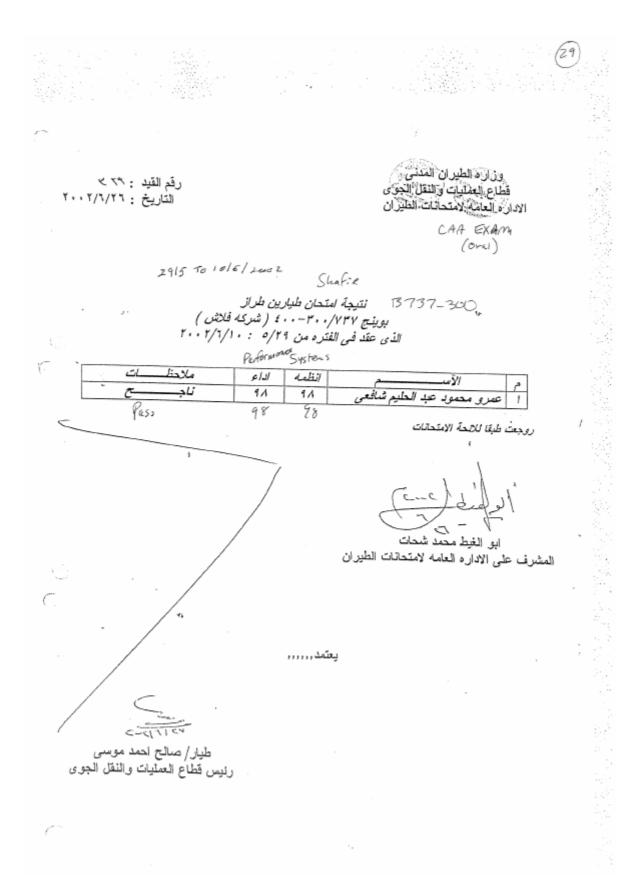
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كشف بنسبة حضور فرقة Basic indoctronation تاريخ بداية الفرقة 21 / 8 /2002 ، تاريخ انتهاء الفرقة 29 /8 / 2002 ملاحظات الاسم الدورة ô حضر الدورة Basic indoctronation رضا السيد مصطفى 1 محمود حنفي حضر الدورة Basic indoctronation 2 عمرو شافعي حضر الدورة Basic indoctronation 3 ېک : ٢/ رئيس تحدوايد / ٢/ يون ف : ٢/ رئيس تحدوايد / ٢/ يون : هديو ادني بعانه / هديو الجوده : ٢٥ / ٢ / ٥ - ٢ ٩٢ / ٨/٢ توقيع المدرب الإسم الوظيفة P2/A/2-2 تحريرا في 29/8/2002 تير الليا,سيم/يل - السيالي 1444 2--c / N /C9 OPERATIONS



Forms and Records



Form No. 04 - 3/4 IOE / USV FORM (Cont'd) THE FOLLOWING ITEMS MUST BE COVERD DURING LINE CHECK (v) Indicates that item has been checked DESCENT AND APPROACH PRE FLIGHT 5 A'I IS, SNOWTAM and braking action* Dispatch 5 Descent planing Computerized and ATC flight plan 1 ٠ ٠ 5 Approach briefing and stars Weather briefing, T.O. and landing min. A Approaches: Alternate planing Wx min <u>8</u> 11 ٠ . @ N.precision @ Visual \mathcal{S} Precision NAT. Operations Specifications* sí Destination and alternate weather minima s ٠ NOTAM briefing and "B" snags ٠ LANDING AND TAXLIN Cabin crew safety briefing ß . ۶Ÿ Landing technique Cockpit Use of auto breaks and reverse trust Ş, Technical log and B snags ٠ . After landing and taxi in procedure 51 ٠ f_{i}^{i} MEL-CDL and the effect on T.O/Landing DISCUSSION ITEMS .97 Performance A) Flight operation manual Aircraft library and documentation 19 IOE, Initial release, USV and Command M^{2} Cockpit preparation-FMS/FMGS/PMS \mathcal{F} ٠ Responsibility TAKE OFF BRIEFING ,51 Navigation Bag content ,57 Load, Irim sheet and NOTOC в The difference between planning and actual ,51 SNOWTAM (de-icing)* ø . Weather min, and Wx min, for new captain, ا م 31 Hot Wx operation 51 Fuel policy S T.O Performance, T.O speeds and C.G Windshear, thunderstornus and turbulence s' s Engine start procedures 0 51 TAXI, TAKE-OFF AND INITIAL CLIMB Fucting with PAX on board Dangerous goods Push back procedures 5 Shoulder harness, seat belt policy and coclipit dom 81 Aircraft geometry during turns First officer T.O. and landing 3 Taxi speed and braking technique ġ ECARS 121 T.O roll and V1 concept Flight operations manuals & answers Noise abatement procedure and initial climb 5 (b) Aircraft performance and technical knowledge Dest angle, best rate and turbulence speeds s Operational system knowledge 57 Area departure, SID and holding CO performance limits 51 CRUISE ø light level selection, specific range and OPT ALT of and contaminated runways -loced (flex) thrust اکر See climb and fuel saving \leq^{t} 'adde much no, and manoenvre capability crossed and haiding chiral performance cost, non-normal and emergency procedur or wention rolls and weather avoidance ,51 1919's and MORA (Special routes). Seed patternal c) Safety procedure bitt down precedence Enroute alternate and Emergency Proc. communication between cockpit and cabin regreency evaluation procedure Senate Weather minima mound/amprepared efficiency "Unineren feat for diversion(Alternate+Holding) rsh on board and least risk location Sourcessization foilure procedures en iscapacitation fight control contait. Procedures (Stockholm radio)

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0	TRAINING RE NAME :- ANR EL SHAFT AIRLINE: Flash Airlines	CORD., CI T	FFS - LESSON 1 REW POSITION: <i>F/-</i> (PE: B7373	ро- 400-Б20	
	Briefing Training plan Operation philosophy	51	Cruise Normal procedures	[Sr]	
	Preflight Normal procedures Supplementary Normal procedures	5	Descent, Approach Normal procedures	51	
	Engine start Normal procedures Additional training item	1 ⁵ 1 ² 1 ²	Landing Normal procedures	<i> </i>	
	Taxi- out & takeoff Normal procedures	54	Taxi - in & park Normal procedures	15"	
	Climb Normal procedures Demonstration fight	S. S.			
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TRAININ NAME :- <i>CHB. EL. SHA</i> AIRLINE: <u>Flash Airline</u>		
Briefing Set up MCP ,CDU Engine inoperative flight characterstics Preflight	Cruise , Descent الحر Hydraulic system A loss (حر)	a Linda ang Sina a
Set up MCP ,CDU After start checklist Engine start Normal procedures Taxi- out & takeoff Rejected T/O T/O engine failure after V I T/O engine failure after V I Wind shear near VR Climb Normal procedures	Approach , Landing One engine inop. manual , F/D ILS approach One ongine inop. Visual traffic Patterns full stop. One engine inop. Landing S' One engine inop. Landing S' Wind shear training S' Vind shear flight path control Hold S' A/P , A/T , F/D VOR approach Full stop landing Taxi - in & park Normal procedures	
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	TRAINING R NAME :- <i>QHC_EL_SHAFT</i> AIRLINE: <u>Flash Airlines</u>	C	, FFS - LESSON 3 REW POSITION:	io.
	Briefing Review item in phase of flight Preflight Normal procedures	5 ⁴	Cruise, Descent Rapid depressurization Emergency descent. Steep turns Approach to stall recovery Approach, Landing One engine inop. A/P , F/D ,VOR approach, circle to land, fr	181 181 181 181
	Engine start Aborted engine starts	,51	One engine inop. ILS approach missed approach	151
	Taxi- out & takeoff Normal procedures Rejected T/O T/O engine failure after V I Normal T/O Cliřftb Wheel well fire Runaway stabilizer Bus off Loss of both engine driven gen.	21 21 21 21 21 21 21 21 21 21 21 21 21 2	Hold * Taxi - in & park Normal procedures	ן זיבן ייבק
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TRAINING MANUAL TRAINING RECORD EFS - LESSON 4 NAME :- OHR EL SHOPL AIRLINE: Flash Airlines TYPE: B: 737- 30+(499/502					
Briefing Full auto flight for precision app. Review item in phase of flight	127) 724	Cruise Steep turns Approach to stall recovery	151 151		
Preflight Normal procedures Reduced thrust computation Engine start Aborted starts (1) Aborted starts (2) Taxi- out & takeoff Normal procedures No autopilot & F/D Reduced thrust takeoff Flap refraction Climb Normal procedures Max angle climb Econ climb	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	Descent , Normal procedures Economy path descent Arrival procedure Approach ,Landing Normal procedures A/P , A/T ,(no F/D) autoland ILS approach Touch & go landing Row data F/D /ILS , T&GO, A/P ,A/T ,3/D VOR approach Touch & go landing Taxi - in & park Normal procedures	27 27 28 28 28 28 28 28 28 28 28 28 28 28 28		
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	Briefing Set up MCP ,CDU Engine inoperative flight characterstics Preflight Set up MCP ,CDU After start checklist Engine start Normal procedures	12 12 22 22 22 22 22 22 22 22 22 22 22 2	Cruise, Descent Approach, Landing One engine inop. A/P , F/D No A/F ILS approach Missed approach One engine inop. manual , F/D No A/F ILS approach	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	
2.	Taxi- out & takeoff T/O engine failure after VI(1) T/O engine failure after VI(2) T/O engine failure after VI(3) Climb	57 57	Full stop landing Normal T/O, manual Row data F/D_ILS, T&GO Loss of both engine driven gen. Manual_ILS, T&GO, A/P_A/T_F/D_VOR approach , circle to land rejected landing A/P_A/T_F/D_ILS_approach Visual traffic patterns Taxi - in & park	12" 12" 12" 12" 12" 12" 12"	
- - -	Normal procedures	51	Normai procedures	157	
	REMARKS				
	INSTRUCTOR) 	ATE 26-06-3002		

37 TRAINING MANUAL TRAINING RECORD FFS - LESSON 6 NAME : AHR CL SHAFT CREW POSITION:---- F/a AIRLINE: Flash Airlines TYPE:---- 13. 737-300/402/500 Briefing Cruise, Descent Set up MCP ,CDU 137 Hydraulic system A loss 57 Engine inoperative flight characterstics 154 Preflight Set up MCP ,CDU ا کړ Approach , Landing After start checklist 15! One engine inop. manual , F/D ILS approach 15% Engine start One engine inop. Visual traffie Normal procedures 51 Patterns full stop. 61 Taxi- out & takeoff One engine inop. Landing SY. Rejected T/O Wind shear training 5 51 T/O engine failure after V II Wind shear flight path control اکہ T/O engine failure after V I 151 151 Hold 154 Wind shear near VR A/P .A/F .F/D VOR approach 51 , Full stop landing Climb 151 Normal procedures 15 Taxi - in & park Normal procedures 行手 151 ۲. է. Ť, REMARKS CHERLE RUDE 1. 19 INSTRUCTOR DATE 0.25-6

TR/	AINING	MANUAL		38
IRAINING NAME : <i>AHR EL- SHOE</i> AIRLINE: <u>Flash Airlines</u>	RECOF	RD FFS - LESSON 7 CREW POSITION:	0.0/500	
Briefing Review item in phase of flight Set up MCP ,CDU	15"	Cruise, Descent		
Preflight Set up MCP ,CDU After start checklist Engine start Fast start Taxi- out & takeoff Normal procedures	12 12 12 12 12 12 12 12 12 12 12 12 12 1	Approach, Landing A/P, A/F, no F/D VOR approach , full stop fanding. Hold. Jammed stabilizer visual traffic pattern full stop landing.(Capt) ILS approach full stop landing , ASS. Flaps. Hydraulic System A &B failure Manual rev. Visual traffic patterns all up Flapt capt.)	200 200 * 257 257	
Normal T/O Climb Normal procedures	L. L.	Taxi - in & park APU fire 470 Ene. fire on 400' (capt.) Passenger evacuation	127	
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34 TRAINING MANUAL TRAINING RECORD FFS - LESSON 8 NAME : AMR EL SHINEL CREW POSITION: F/2 AIRLINE: Flash Airlines TYPE: B 737-300/400/500 State Date Briefing Cruise, Descent Review item in phase of flight 151 Steep turns 151 Approach to stall recovery s:t Holding ß Preflight Engine fire L Normal procedures 5' Wing /body over heat اکم Bleed or pack trip 51 Engine start Rapid depressurization (capt) 51 18 Normal procedures 154 Emergency descent. D Approach , Landing Taxi- out & takeoff ٩e One engine inop. 17D, VOR 70 Rejected T/O 151 approach, circle to land (capt) 27 9 T/O engine failure after V I VI cut One engine inop. , ILS 61 Normal T/O 151 Approach, missed approach 2) 'C'imb Wheel well fire 51 Taxi - in & park Runaway stabilizer Normal procedures 51 A いの時間が数年のなどないと ŝ. 「「「「「「「「「「「「」」」」」 -65REMARKS HAS PERFORMED LANDING. AS 200 205% REGINIAG: AND HE ISREPY FOR DR. BISTRAISIAN mauras -AUTP-1112 INSTRUCTOR DATE delle 01-07-2007 å.,

- iv- Personal situation To be completed by the OPS Group
- 1.5.2.3. 72-hour history of the F/O: Refer to interviews included in pages 72-73 of the Factual Report
- 1.5.2.4. Interviewing the individuals who trained and flew with the F/O (including ground and simulator instructors) None available
- 1.5.2.5. Interviewing CAA inspectors who flew with F/O. Interviews to be carried out by OPS Group
- 1.5.2.6. Interviewing former head of operations at Flash Airlines (No official former head of operation in Flash Airlines)
- 1.5.2.7. Additional factual documentation (F/O)

Number of days the F/O had been working since his last day off. Refer to Factual Report

F/O interpersonal characteristics, including perceptions of fellow pilots regarding their capability for assertiveness. All available information is available in pages 72-73 Factual Report

Reported proficiency information. Outcome and comments from training records and proficiency check forms. Refer to 1.5.2.2 (iii)

Spatial disorientation or upset recovery training received at Flash Air AI196 According to CAA regulations, Spatial Disorientation training is not mandatory

No available documents from Flash Airline concerning SD training. Inputs from different investigation partners are needed.

According to and CAA regulations, Upset Recovery training is not mandatory Upset Recovery Training recommendation may be included in the Recommendations Chapter.

F/O's flying proficiency and cockpit style from fellow pilots, instructors, and/or check pilots. Not available

1.5.3 The Observer

Background:

The Observer "Ashraf Abdel Hamid" was completing his training as a first officer for Flash Airlines.

Beginning of his flying career:

Training at USA

ISIS Airman Report CAIS Information - Basic Information Cert No: 2440980 Cert Sfx: Soc.Sec.No: 620480104 Cert Pfx: Name: ABDELHAMID, ASHRAF Name Sfx: DOB: 1961 10 25 Sex: M Hair: BROWN Eyes: BROWN Ht: 68 Wt: 154 POB: CAIRO, EGYPT Status: Info: Name/Address Source: Airm Date of Address Update: 2004 03 10 Citizenship: USA Street: PO BOX 414 County: 065 City: PALM DESERT State: CA Zip: 92261-0414 Country: TOT CIVIL HOURS: 03750 TOT MIL HOURS: 00400

ISIS Airman Report CAIS Information - Medical Cert Pfx: Cert No: 2440980 Cert Sfx: Information Medical Information for: ABDELHAMID, ASHRAF Class: First Certificate Desc.: LIMITED Medical Date: 2003 01 28 Medical ID#: 200001408794 Restriction: MUST HAVE AVAILABLE GLASSES FORNEAR VISION.

ISIS Airman Report CAIS Information - Certificate Cert Pfx: Cert No: 2440980 Cert Sfx: Information Specl Purp Pilot Info ABDELHAMID ASHRAF Cert-Level: COMMERCIAL PILOT (FOREIGN BASED) Rating/Level: AIRPLANE SINGLE ENGINE LAND/COMMERCIAL PILOT (FOREIGN BASED) INSTRUMENT AIRPLANE/COMMERCIAL PILOT (FOREIGN BASED) Type Rating/Level: Date of Issue: 1991 10 17 OrgDOI: Update Date: 1991 10 17 Seal: Black Cert Status: Active

ISIS Airman Report CAIS Information - Certificate Cert Pfx: Cert No: 2440980 Cert Sfx: Information Specl Purp Pilot Info ABDELHAMID ASHRAF Certificate Limitations ISSUED ON BASIS OF AND VALID ONLY WHEN ACCOMPANIED BY CANADIAN PILOT LICENSE NO. C275467. ALL LIMITATIONS AND RESTRICTIONS ON THE CANADIAN PILOT LICENSE APPLY. NOT VALID FOR AGRICULTURAL AIRCRAFT OPERATIONS. INSTRUMENT AIRPLANE (U.S. TEST PASSED).

ISIS Airman Report CAIS Information - Certificate Cert Pfx: Cert No: 2635768 Cert Sfx: Information Pilot Information for: ABDELHAMID ASHRAF Cert-Level: AIRLINE TRANSPORT PILOT Rating/Level: AIRPLANE MULTIENGINE LAND/AIRLINE TRANSPORT PILOT Type Rating/Level: Date of Issue: 2000 06 15 OrgDOI: Update Date: 2001 06 21 Cert Status: Active Seal: Blue _____

ISIS Airman Report CAIS Information - Previous Certificate Cert Pfx: Certificate No: 2440980 Cert Sfx: Previous Certificate for: ABDELHAMID ASHRAF

Previous Certificate Information: Pfx Cert Num. Sfx Cert Date Cert Level/Type

NO PREVIOUS CERTIFICATE INFORMATION AVAILABLE

ISIS Accident/Incident (AID) Report Airman Accident/Incident Airman Name: ABDELHAMID, ASHRAF Cert #: 002440980 Accident Date: 02/15/2001 Air Agency Cert #: Accident Event: GENERAL AVIATION ACCIDENT Source: .4 Type of Accident: LOSS OF DIRECTIONAL CONTROL Accident Location------City: SAN DIEGO State: CA

Aircraft Involved------N-Number: N4922D Make: CESSNA Model: 172N

ISIS Accident Incident ReportFull AID TextPage No.: 1Case number: 4922D20010215115931of 3Jump to page: ____AID Text

ON FEBRUARY 15, 2001, ABOUT 1516 HOURS PST, A CESSNA 172N, N4922D, VEERED OFF THE RUNWAY AND COLLIDED WITH A TAXIWAY SIGN DURING LANDING ROLLOUT ON RUNWAY 28L AT THE MONTGOMERY FIELD, SAN DIEGO, CA. THE AIRPLANE WAS SUBSTANTIALLY DAMAGED. NEITHER THE AIRLINE TRANSPORT CERTIFICATED PILOT NOR PASSENGER WAS INJURED. PLUS ONE FLYERS, INC., IN SAN DIEGO, OPERATED THE AIRPLANE. VISUAL METEOROLOGICAL CONDITIONS PREVAILED AND AN INSTRUMENT FLIGHT RULES FLIGHT PLAN WAS FILED. THE PERSONAL FLIGHT WAS PERFORMED UNDER 14 CFR PART 91, AND IT ORIGINATED IN SCOTTSDALE, AZ. ABOUT 1135. AIRPORT PERSONNEL REPORTED THAT THE COLLISION OCCURRED ABOUT 1,000 FEET UPWIND OF THE RUNWAY'S THRESHOLD. THE AIRPLANE IMPACTED THE TAXIWAY "C" SIGN, AND VEERED OFF THE RUNWAY. THE AIRPLANE CAME TO A STOP ABOUT 200 FEET NORTH OF THE RUNWAY. THE PILOT STATED THAT DURING THE LANDING ROLLOUT, AS THE AIRPLANE WAS DECELERATING THROUGH ABOUT 50 KNOTS, THE LEFT WING SUDDENLY LIFTED UP. THEREAFTER HE LOST CONTROL OF THE AIRPLANE. HE ADDITIONALLY REPORTED THAT HE WAS UNAWARE OF THE REASON FOR THIS OCCURENCE. NO MECHANICAL MALFUNCTIONS WERE REPORTED WITH THE AIRPLANE.

ON FEBRUARY 15, 2001, ABOUT 1516 HOURS PACIFIC STANDARD TIME, A CESSNA 172N, N4922D, VEERED OFF THE RUNWAY AND COLLIDED WITH A TAXIWAY SIGN DURING LANDING ROLLOUT ON RUNWAY 28L AT THE MONTGOMERY FIELD, SAN DIEGO, CALIFORNIA. THE AIRPLANE WAS SUBSTANTIALLY DAMAGED. NEITHER THE AIRLINE TRANSPORT CERTIFICATED PILOT NOR PASSENGER WAS INJURED. PLUS ONE FLYERS. INC.. SAN DIEGO, OPERATED THE AIRPLANE. VISUAL METEOROLOGICAL CONDITIONS PREVAILED, ANDAN INSTRUMENT FLIGHT RULES FLIGHT PLAN WAS FILED. THE PERSONAL FLIGHT WAS PERFORMED UNDER 14 CFR PART 91, AND ORIGINATED IN SCOTTSDALE, ARIZONA, ABOUT 1235 MOUNTAIN STANDARD TIME. AIRPORT PERSONNEL REPORTED THAT THE COLLISION OCCURRED ABOUT 1,000 FEET UPWIND OF THE RUNWAY'S THRESHOLD. THE AIRPLANE IMPACTED THE TAXIWAY "C" SIGN AND VEERED OFF THE RUNWAY. THE AIRPLANE CAME TO A STOP ABOUT 550 FEET FARTHER UPWIND OF THE SIGN AND ABOUT 200 FEET NORTH OF THE RUNWAY. THE PILOT STATED TO THE NATIONAL TRANSPORTATION SAFETY BOARD INVESTIGATOR THAT DURING THE LANDING ROLLOUT, AS THE AIRPLANE WAS DECELERATING THROUGH ABOUT 50 KNOTS, THE LEFT WING SUDDENLY LIFTED UP. THEREAFTER, HE LOST CONTROL OF THE AIRPLANE. HE ADDITIONALLY REPORTED THAT HE WAS UNAWARE OF THE REASON FOR THIS OCCURRENCE. NO MECHANICAL MALFUNCTIONS WERE REPORTED WITH THE AIRPLANE. IN THE PILOT'S PARTIALLY COMPLETED ACCIDENT REPORT, HE INDICATED THAT WHEN THE AIRPLANE WAS "ALMOST HALF WAY DOWN THE RUNWAY" THE LEFT WING ROSE UP, AND THEREAFTER HE LOST CONTROL OF THE AIRPLANE AS IT "VIOLENTLY" VEERED OFF THE RUNWAY. THE PILOT ALSO REPORTED THAT WHEN HE WAS ON FINAL APPROACH THE TOWER CONTROLLER REPORTED THAT THE WIND WAS FROM 270 DEGREES AT 6 KNOTS.

Enforcement for Airman: ABDELHAMID, ASHRAFRecs: 0Using Certificate: 002440980(Specl Purp Pilot Inthru: 0A search of EIS data by LAST NAME found 0other matches, Press F5 to viewJump to VIOL. DATE_________Sort by column: 1 AViol.DateStatus RgnCase#Related case#

NO RECORDS FOUND

Enforcement for Airman: ABDELHAMID, ASHRAFRecs: 0Using Certificate: 002635768 (Pilot)thru: 0A search of EIS data by LAST NAME found 0 other matches, Press F5 to viewJump to VIOL. DATESort by column: 1 A of: 0Viol.DateStatus Rgn Case#Related case#

NO RECORDS FOUND

Inspection for Airman: ABDELHAMID, ASHRAF Recs: 1 Using Certificate: 002440980 (Specl Purp Pilot In thru: Jump to: RECORD ID ______ Sort by column: 1 A of: Record ID Activity Code FAR Status Start Date Completion

NO RECORDS FOUND

Inspection for Airman: ABDELHAMID, ASHRAF Recs: 1 Using Certificate: 002635768 (Pilot) thru: Jump to: RECORD ID ______ Sort by column: 1 A of: Record ID Activity Code FAR Status Start Date Completion NO RECORDS FOUND

Interview with Brother of observer Pilot/Ashraf Abdel Hamid:

Captain/Alaa El Saadany Training Captain with EgyptAir was interviewed by Dr. Adel Fouad and Captain Shaker Kelada who said that Ashraf Abdel Hamid was a lively person sociable and easy to get along with, was friendly confident and out spoken.

Asked about his career as a pilot he said that he started his initial training in Cairo than went to Canada and obtained Canadian citizenship and Canadian pilot license and flew single engine planes. He then went to the USA and also obtained USA citizenship and flew there on single engine and Lear jets had a total of around 4000 hrs.

On a family visit to Egypt, he was persuaded by Captain\ Sombaty (Operations Manager of Flash Airline), a colleague and personal friend to stay in Egypt and fly for Flash. He had attended B737 ground school course and was due for examination two days after the accident. He flew as an observer with Captain Sombaty who was assisting him to complete his B737 qualification.

Correction:

The following statement included in page 15 of the factual report should be deleted: Airline training procedures require a certain amount of observation time prior to serving as an active crew member. The observer was assigned to this flight to observe as a part of that training requirement.

The following statement should replace it:

Ashraf Abdel Hamid was flying as an observer as it is common practice for operators in Egypt is to assign pilots joining an airline or upgrading to a new type to fly as an observer on the type to be flown to get acquainted with company routes and procedures of the operator and type

CAA regulations regarding observation time: N/A

Flash Airline policy regarding observation time: As required

1.6.4.2. DFCS Mode

Operation of the FD vertical bar with "Heading Select" disengagement as the AP engages. Refer to Boeing AMM 22-11-00 Page 38

- 1.6.6.4. The maintenance log sheets for the flights after 12/31/03Lost on board and no copies prior to departures from SHH which is a violation of ECAA regulations. Necessary measures are taken by ECAA to ensure adherence.
- 1.6.6.5. The lack of write-ups on the TOGA problem and slat indication that existed on the entire 25-hours of FDR. Status of the technical log is not known due to being lost on board

1.9.1 ATC communications/ Transcript

Information about the conversation between ATC and MSR 227 (p 44, factual report) translated from Arabic into English.

2:58:15	C>P	
	P>C	
	P>C	Sharm MSR227
	C>P	Go Ahead Sir
	P>C	We heard on frequency 121.5 some one from Flash speaking, I do not know if it is 604 or it is another Flash Aircraft
	C>P	It is 604, there is no other aircrafts
	P>C	He was speaking on 121.5, so it is O.K.
	C>P	Thank you very much Sir
	P>C	You're welcome
	C>P	Ground 121.9 for company information, God willing
	P>C	Peace be with you 121.9
	C>P	And with you

N.B. Frequency 121.5 was checked no transmission was recorded at the time of the accident with any traffic

1.13.2. Medical factors related to SD (Spatial Disorientation): A. FAA advisory Circular regarding SD





Subject: FILOT'S SPATIAL DISORIENTATION	Date:	2/9/83	AC No: 60-44
	Initiated by:	AFO840	Change:

1. FURPOSE. To acquaint pilots with the hazards of disorientation caused by loss of visual reference with the Surface.

2. <u>CANCELLATION</u>. Advisory Circular 60-4, Pilot's Spatial Disorientation, dated February 9, 1965, is canceled.

DISCUSSION.

a. The attitude of an aircraft is generally determined by reference to the natural horizon or other visual references with the surface. If neither horizon nor surface references exist, the attitude of an aircraft must be determined by artificial means from the flight instruments. Sight, supported by other senses, allows the pilot to meintain crientation. However, during periods of low visibility, the supporting senses sometimes conflict with what is seen. When this happens, a pilot is particularly vulnerable to disorientation. The degree of disorientation may vary considerably with individual pilots. Spatial disorientation to a pilot means simply the inability to tell which way is "up."

b. During a recent 5-year period, there were almost 500 spatial disorientation accidents in the United States. Tragically, such accidents resulted in fatalities over 90 percent of the time.

c. Tests conducted with qualified instrument pilots indicate that it can take as much as 35 seconds to establish full control by instruments after the loss of visual reference with the surface. When another large group of pilots were asked to identify what types of spatial disorientation incidents they had personally experienced, the five most common illusions reported were: 60 percent had a sensation that one wing was low although Wings were level; 45 percent had, on leveling after banking, tended to bank in opposite direction; 39 percent had falt as if straight and level when in a turn; 34 percent had become confused in attempting to mix "contact" and instrument cues; and 29 percent had, on recovery from steep climbing turn, felt to be turning in opposite direction.

d. Surface references and the natural horizon may at times become obscured, although visibility may be above visual flight rule minimums. Lack of natural horizon or surface reference is common on overwater flights, at night, and especially at night in extremely sparsely populated areas, or in low visibility conditions. A sloping cloud formation, an obscured horizon, a dark scene Spread with ground lights and stars, and certain geometric patterns of ground lights con provide innecurate visual information for aligning the aircraft correctly with the actual horizon. The discrimined pilot may place the aircraft in a dangerous attitude. Other factors which contribute to discrimination are

AC 60-4A

reflections from outside lights, sunlight shining through clouds, and reflected light from the anticollision rotating beacon.

e. Another condition creating restrictions to both horizontal and vertical visibility is commonly called "white-out." "White-out" is generally caused by fog, have, or falling snow blending with the snow-covered earth surface which may obscure all outside references. Therefore, the use of flight instruments is concential to maintain proper attitude when encountering any of the elements which may result in spatial disorientation.

4. RECOMMENDED ACTION.

a. You, the pilot, should understand the elements contributing to spatial discrimination so as to prevent loss of sireraft control if these conditions are inadvertently encountered.

b. The following are certain basic steps which should assist materially in preventing spatial discrimination.

(1) Before you fly with less than 3 miles visibility, obtain training and maintain proficiency in aircraft control by reference to instruments.

(2) When flying at night or in reduced visibility, use your flight instruments, in conjunction with visual references.

(3) Maintain night currency if you intend to fly at night. Include crosscountry and local operations at different airports.

(4) Study and become familiar with unique geographical conditions in areas in which you intend to operate.

(5) Check weather forecasts before departure, on route, and at destination. Be alart for weather deterioration.

(A) Do not attempt visual flight rule flight when there is a possibility of getting trapped in deteriorating weather.

(7) Rely on instrument indications unless the natural horizon or surface reference is clearly visible.

5. <u>CONCLUSION</u>. You and only you have full knowledge of your limitations. Know these limitations and be guided by them.

KENNETH S. HUNT Director of Flight Operations

Par 3

2

- B- MCA study regarding SD Refer to Factual Report, page 55 (Dr. Marawan report) and item 1.16.4. Tests and researches conducted by MCA:
- C- Medical records for the captain related to any of the conditions conducive to spatial disorientation. No report found
- 1.13.3. Most recent medical certification
 - A- Date, type Refer to page 14 of the Factual Report
 - B- Limitations (if applicable) None (Refer to page 14 of the Factual Report)
- 1.13.4. General health information for each crew member. No Factual information available
- 1.13.5. Toxicological testing. No toxicological testing was possible because the bodies were not recovered.
- 1.13.6. Last civil medical check for Captain Refer to page 14 of the Factual Report

1.16 Tests and Researches:

1.16.1. Tests and researches conducted by Boeing and Honeywell:

1.16.1.0. General Overview of Boeing Process_ Kinematic Consistency: (CairoMarch04Slides March Progress Meeting - Cairo.pdf) (Kincon and Simulation (public release).ppt)

FDR Data

• Accelerations and Euler angles recorded on the FDR uniquely determine the path of the airplane

 Accelerations

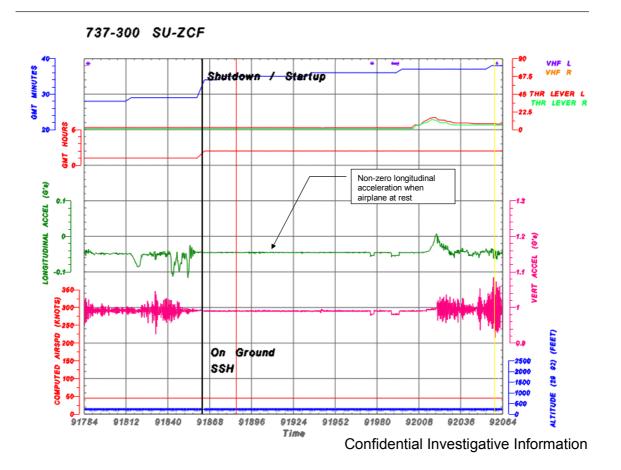
- Vertical
- Longitudinal
- Lateral

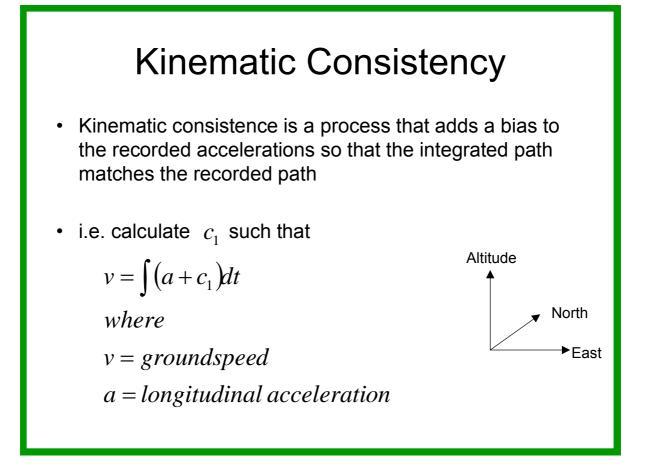
• Euler angles

- Pitch
- Roll
- Heading
- Additional parameters describe path
 - e.g. altitude, ground speed, drift angle

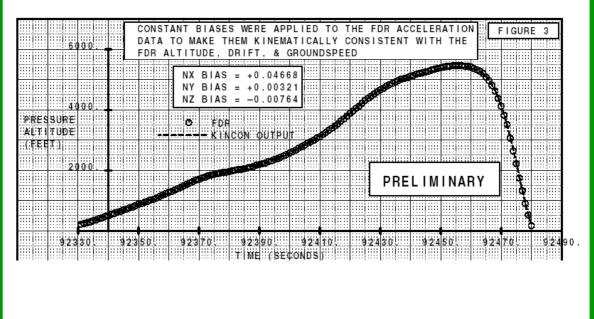
Problem

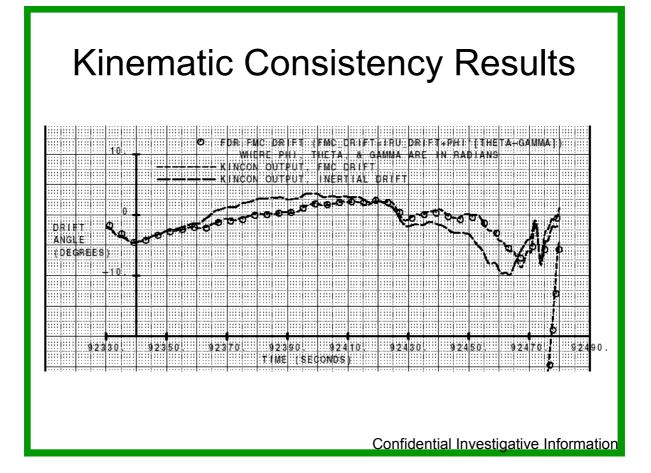
- Some FDR data may be inconsistent with other FDR data
- Example:
 - Integrating longitudinal acceleration during a takeoff roll results in groundspeed. The calculated value may differ from the recorded value.
- Solution:
 - Add an offset to the acceleration such that the calculated groundspeed matches the recorded groundspeed.



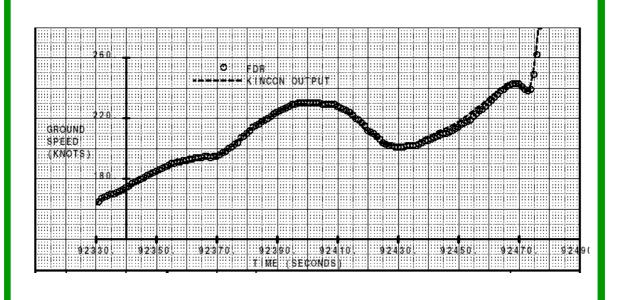


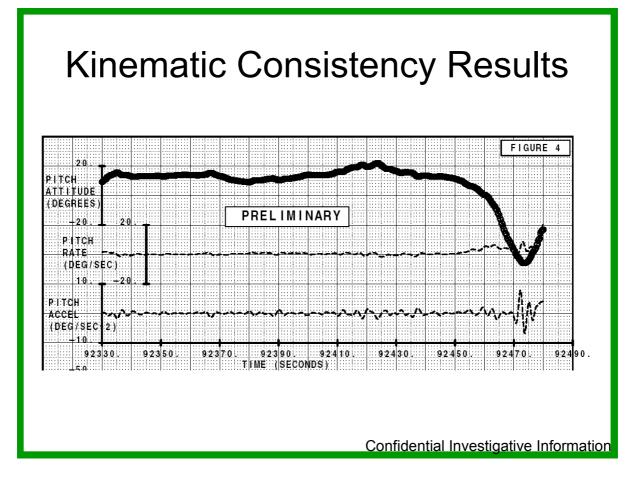
Kinematic Consistency Results



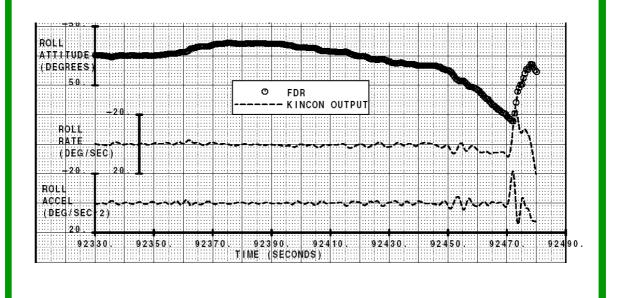


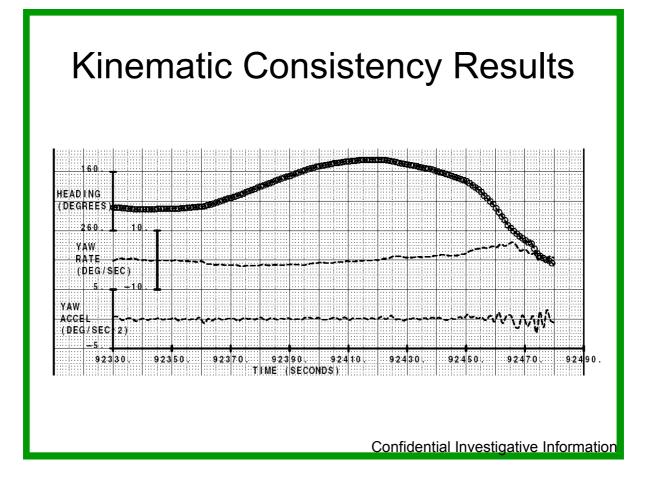












Kinematic Consistency

- Note:
- The kinematic consistency process does not make any assumptions about the aerodynamic properties of the airplane
- In fact, the process can be applied to any moving object

Simulation

- Once the kinematically consistent accelerations and Euler angles have been calculated, an aerodynamic simulation of the airplane is used to reconstruct the flight path
- Time-step integration is used to calculate the motion of the airplane from one step to the next

$$v_{t1} = v_{t0} + a_{t0}\Delta t \qquad x_{t1} = x_{t0} + v_{t0}\Delta t$$
$$Lift = \frac{1}{2}\rho v^2 SC_L$$
$$C_L = f(\alpha, v, flaps, gear, control surfaces, ...)$$

Sensitivity Example

•Accident flight is approximately 147 seconds long

•Simulator match of altitude differs by approximately 200 feet

•Sensitivity analysis for straight and level flight 147 seconds long

$$F = MA \text{ or } A = \frac{F}{M}$$

For vertical axis $\ddot{z} = \frac{L - W}{W} \longrightarrow z = \iint \frac{L - W}{W} dt^2$
For constant weight $z = g \frac{L - W}{W} \frac{t^2}{2} \Big|_{t_2}^{t_2}$

For constant weight

$$\left.\frac{W}{W}\right|_{t_1}$$

Sensitivity Example

For constant weight

$$z = g \frac{L - W}{W} \frac{t^2}{2} \Big|_{t_1}^{t_2}$$

Assume altitude error is result of incorrect lift

$$\Delta z = g\Delta \frac{L - W}{W} \frac{t^2}{2}$$

Solve for ΔL

$$\Delta \frac{L - W}{W} = \frac{2\Delta z}{g t^2} \qquad \Delta L = \frac{2W\Delta z}{g t^2}$$

$$\Delta L = \frac{2(113630 lb)(200 ft)}{32.2 \frac{ft}{\sec^2} (147 \sec)^2} = 65 \ lbs$$

Therefore-

A 65 lb error in calculated lift will result in a altitude error of 200 ft after 147 seconds.

Simulation Differences

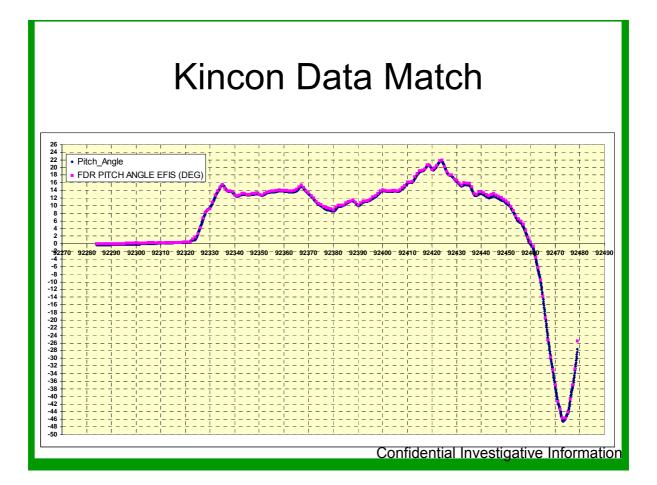
The 737-300 simulation model represents a nominal airplane with nominal engines.

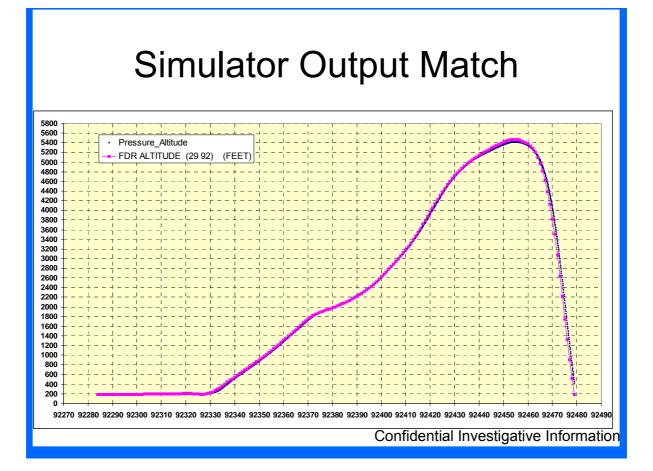
Small offsets between the nominal simulation airplane and an individual airplane in the fleet are common due to differences in rigging, engine wear, etc.

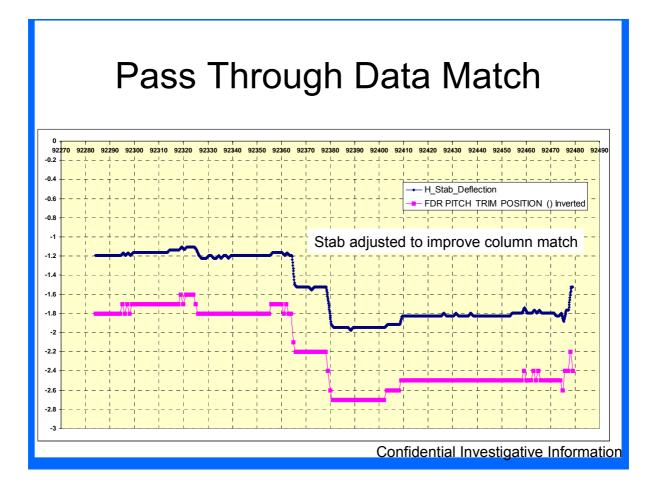
Pass Through Data

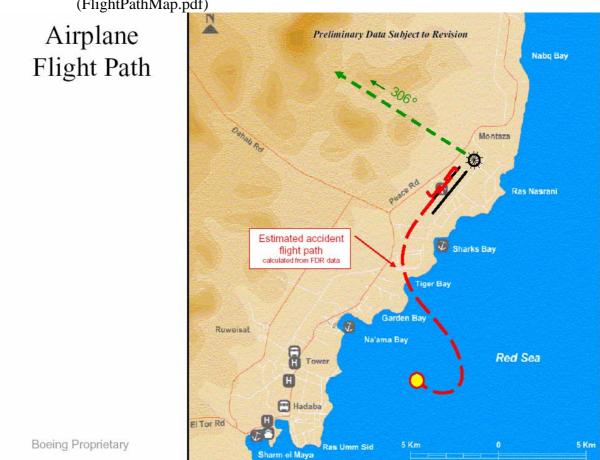
For Flash Airlines simulation –

- Stabilizer was adjusted to account for control column bias (2.9° offset)
- Throttle level position was adjusted to improve match of airspeed and altitude





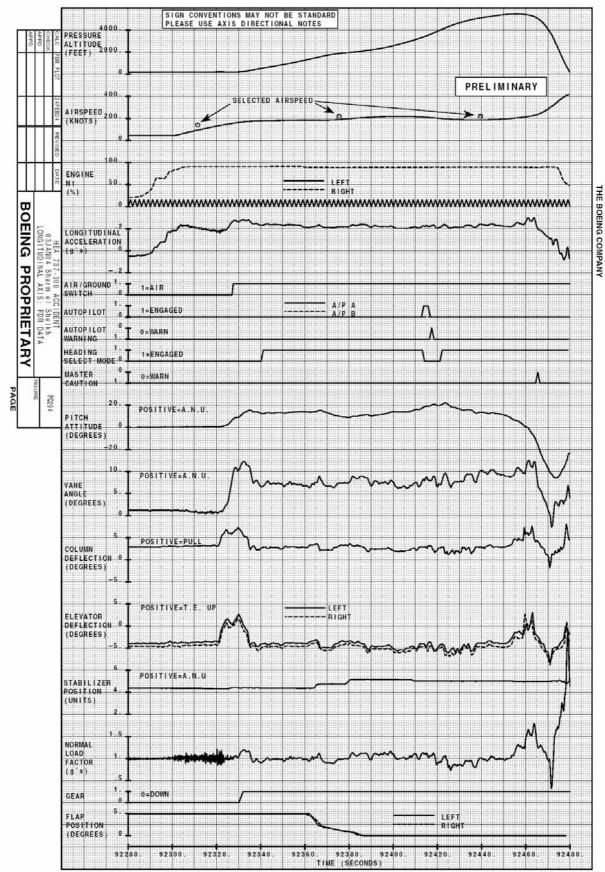




1.16.1.1. Estimated accident flight path, calculated from FDR data: (FlightPathMap.pdf)

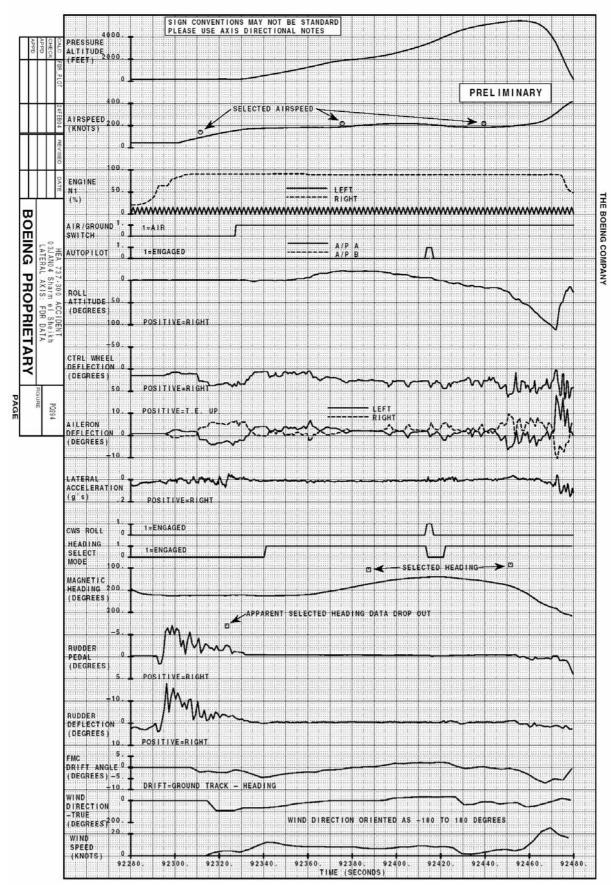
Boeing Proprietary

1.16.1.2. FDR data plots (presented by Boeing)



FDR Data accident flight - FDR plot.pdf (some selected parameters 24 Feb 04) (Longitudinal axis)

(Lateral axis)



FDR Parameter Review, B-H200-17884-ASI, 3 May 2004 (ATT For Decoding Grid.pdf)

1.16.1.3. Simulator Match accident flight: SimMatchaccidentflight 24-2-04.pdf (Simulation Match, FDR-Kincon-Simulation)

SimMatchpreviousflight 24-2-04.pdf (FDR-Kincon-Simulation match 24-2-04)

HEA_PQ294_prevfltSIM.pdf (26 Feb 2004, base lines, FDR-Kincon-Sim prvious flight) Boeing Proprietary information and will not be available for public use HEA_PQ294_baselineSIM.pdf (26 Feb 2004, base lines, FDR-Kincon-Sim) Boeing Proprietary information and will not be available for public use HEA_PQ294_FDR_data.pdf (FDR Data accident flight - Boeing -26 Feb 04 Fig's 1, 2)

HEA_PQ294_kincon (includes roll rate).pdf (FDR Data accident flight - plotted by Boeing (some selected parameters)-26 Feb 04 Fig's 3, 4

HEA_PQ294_WindsSIM29402to29442.pdf (26 Feb 04 Fig's 23- 25

17871 encl 4 (B-H200-17871-ASI 31 March 2004).pdf (enclosure 4 (B-H200-17871-ASI 31 March 2004). Boeing plots

M Cab Recovery (Piloted Recovery.xls)

Simulation Scenario (Simulation Scenario Status20 Sep.,04.xls)

Simulation Scenario (Simulation Scenario Status 27-30 Sep, 04.xls)

1.16.1.4. Simulated Failures:

HEA_PQ294_Simulated_Failures Spoilers, LE Slats.pdf (FDR-norm simulation-simulation with spoilers failures)

Right outboard flight spoilers (#7) Hardover simulation (hardover starts at 92391)

Left outboard flight spoilers (#2) Hardover simulation (hardover starts at 92391)

Right outboard flight spoilers (#7) Float simulation (floats starts at 92391)

Left outboard flight spoilers (#2) Float simulation (floats starts at 92391)

Critical right wing leading edge slat # 6 extends

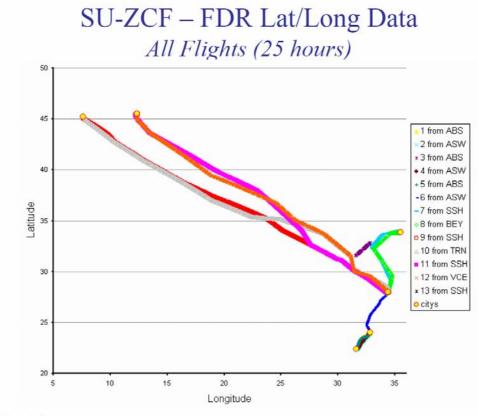
Critical left wing leading edge slat # 1 extends

Fig 40-43 Lateral Control Jams.pdf (FDR, normal simulation, simulation with spoilers fault)

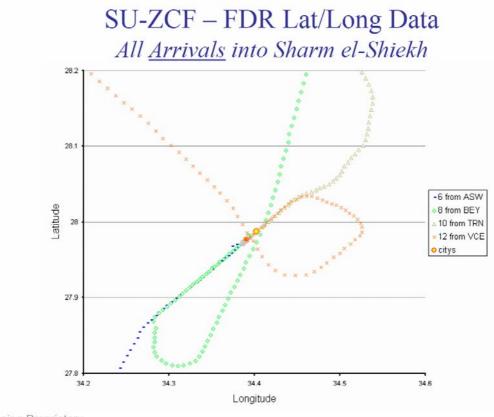
Longitudinal Axis, simulated right wing spoiler cable jam Boeing Proprietary information and will not be available for public use Longitudinal Axis, simulated F/O's wheel jam:

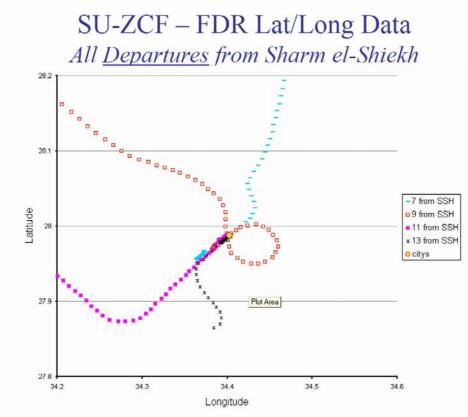
Hypothetical Scenario, Right Side Cable Jam Induces Right Roll (Right Side Cable Jam Effects.ppt)

1.16.1.5. FDR 25 Hour Data- Observations (CairoMarch04Slides (March Progress Meeting - Cairo).pdf, 040301 Flash 737 Cairo Mtg (public release version).pdf)



Boeing Proprietary





FDR 25 Hour Data Observations

SU-SCF Flight 9 departure from SSH

- Departed Rwy 4
- Circling departure to over-fly VOR

Use of TOGA on takeoff

SU-ZCF: TOGA typically engaged for ~2 sec SU-ZCD: TOGA typically engaged for 1-2 minutes

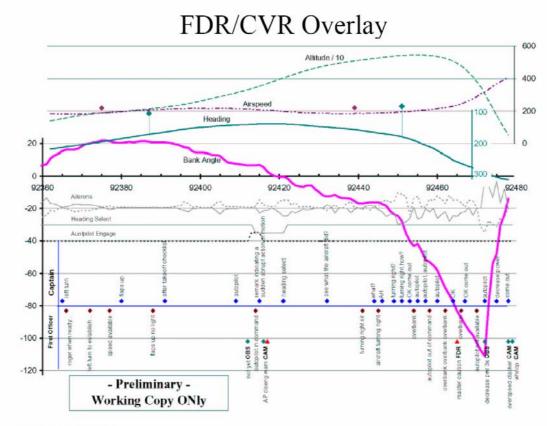
SU-ZCF – FDR 25 Hour Data TOGA Observations

Filght	Both F/D ON?	Normal looking A/T Takeoff	First TOGA Push (1)	If Second TOGA Push (1)
1	YES	YES	1	2
2	YES	YES	0	
3	YES	YES	2	
4	NO	YES	0	
5	YES	YES	2	
6	YES	YES	1	
7	YES	YES	1	
8	YES	YES	2	
9	YES	YES	2	1
10	YES	YES	0	
11	YES	YES	2	
12	YES	YES	2	
13	YES	YES	2	

(1) Number of samples recorded for TOGA_FCC (sample intvi-1 sec)

1.16.1.6. FDR-CVR Overlay

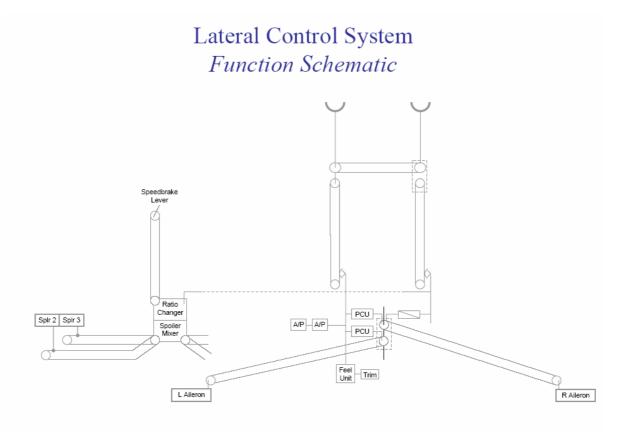
FDR-CVROverlay.pdf, FDR-CVR Overlay 3R2.pdf (21-June 2004, 040301 Flash 737 Cairo Mtg (public release version).pdf)

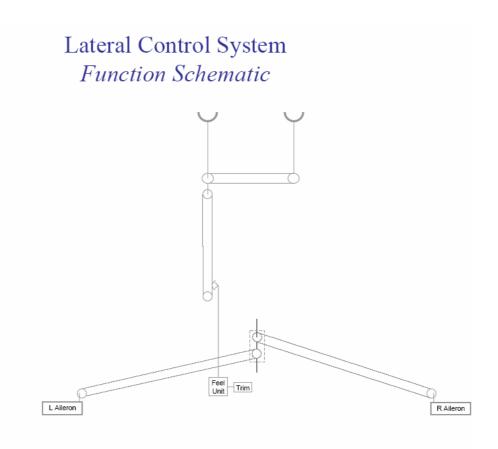


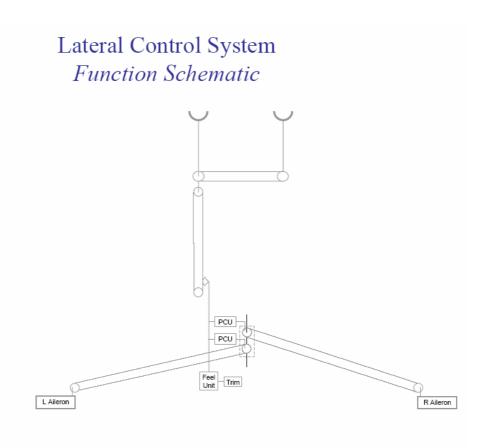
1.16.1.7. Ailerons system

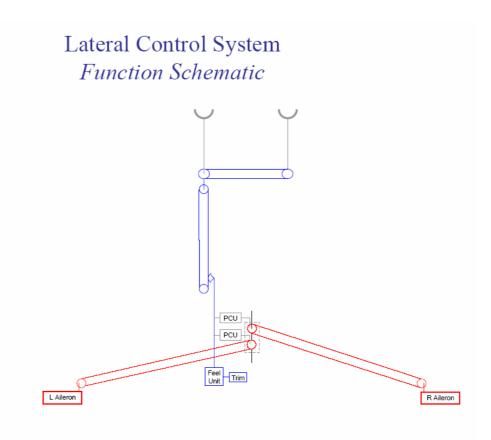
IPC wheel posn xducer PW.pdf (Details about the wheel posn xducer- Part Catalog Maintenance)

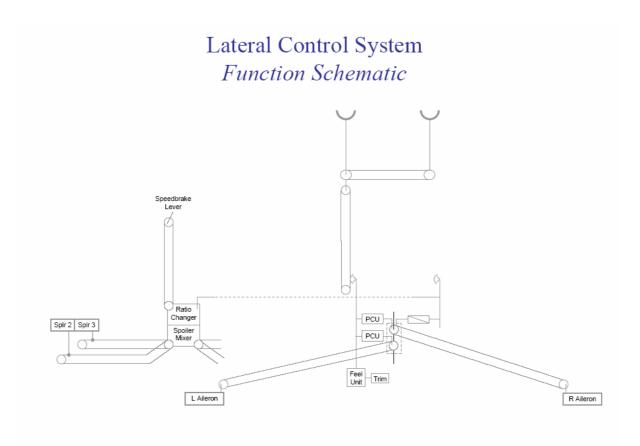
CairoMarch04Slides (March Progress Meeting - Cairo).pdf

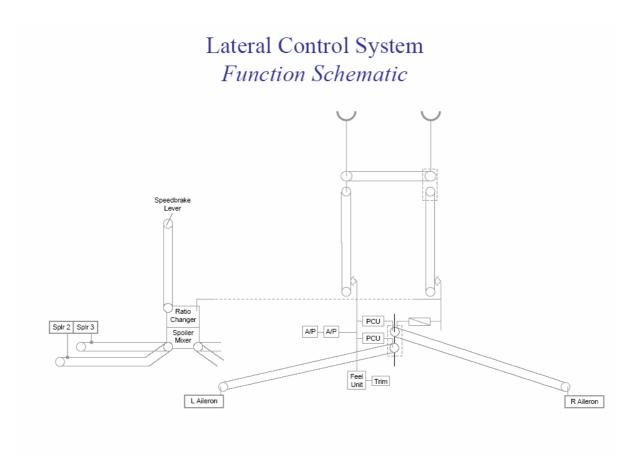




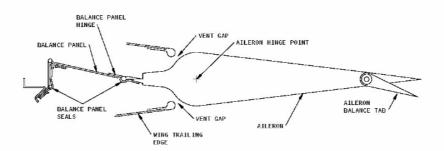








Aileron

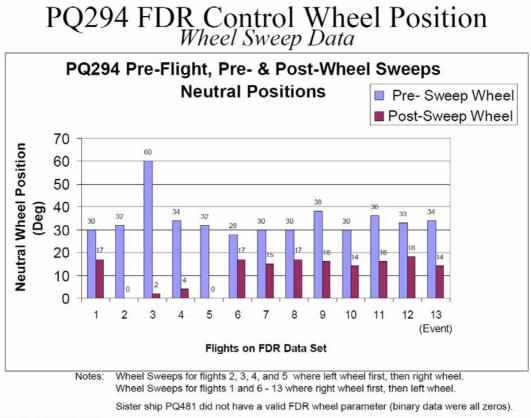


Note

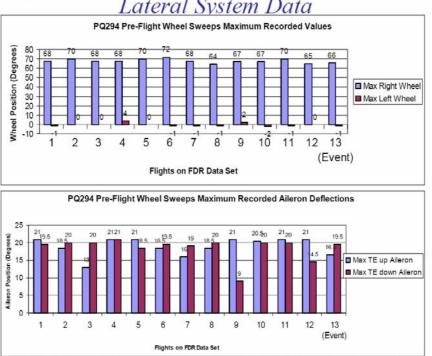
Remaining information is Boeing proprietary information and will not be available for public use

Aileron PCU Control Valve.ppt

ControlWheelBias.pdf, CairoMarch04Slides (March Progress Meeting - Cairo).pdf



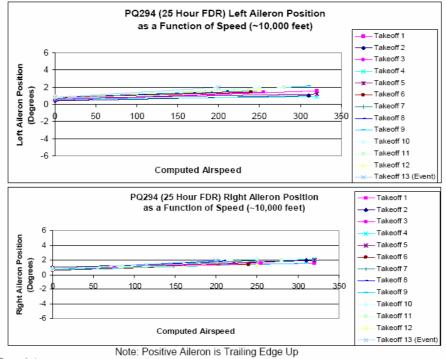
PQ294 FDR Control Wheel Position



Notes: Maximum wheel deflection is +/- 87.5 degrees, 107.5 degrees with cable stretch Maximum aileron deflection is +/- 20 degrees

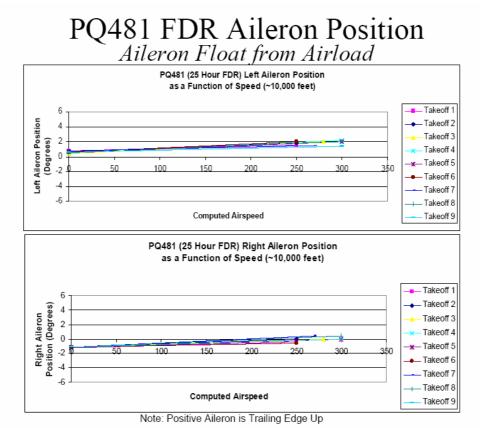
AileronFloat.pdf (PQ294 FDR Aileron Position, Aileron Float from Airload)

PQ294 FDR Aileron Position Aileron Float from Airload



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Boeing Proprietary

M-Cab Wheel (Flight Director Results Boeing.xls)

Force vs Wheel.ppt

Cor8tmp PCU correction.ppt

Aileron PCU EQA (Aileron PCU EQA Field Note Summary.ppt)

Aileron PCU EQA Report (Aileron PCU EQA Report.pdf)

1.16.1.8. Master Caution:

CairoMarch04Slides (March Progress Meeting - Cairo).pdf

Electrical

Master Caution Discrete at Time 92465

Flight Controls Low Quantity Low Pressure Feel Diff Press Speed Trim Fail 1	2 2 2	
Mach Trim Fail 1 Yaw Damper Autoslat Fail	2	3
<u>Hydraulics</u> Low Press – Elec Pump Overheat – Elec Pump Low Press – Eng Pump	2	3 3
IRS Fault On DC DC Fail	2 2 2	
Fuel Low Pressure 1 Filter Bypass.		3
APU Low Oil Pressure Fault	2	

Low Oil Pressure High Oil Temp Standby Power Off Transfer Bus Off Bus Off		2 2 2	3 3
Overheat Detection Engine1 overheat Engine 2 overheat APU Detection Inop	1	2 2	
Anti-Ice Window overheat Pitot heat Cowl Anti-Ice		2 2	3
Doors Fwd/Aft Entry Equipment Fwd/Aft Cargo Fwd/Aft Service Airstairs (not installed	1 1 1 0n	PQ	294)

Engine Reverser PMC-Inop Low Idle	1 1		3
Overhead Equipment Cooling - Of Emer Exit Lts-Not Armer Flight Recorder - Off Pass Oxy - On		2 2	3 3
Air Cond Flt Deck Duct Ovht Pax Duct Ovht Dual Bleed Wing-Body Overheat Bleed Trip Off Auto Fail Off Sched Descent Pack Trip Off	1	2 2 2 2 2 2 2 2 2 2 2 2	

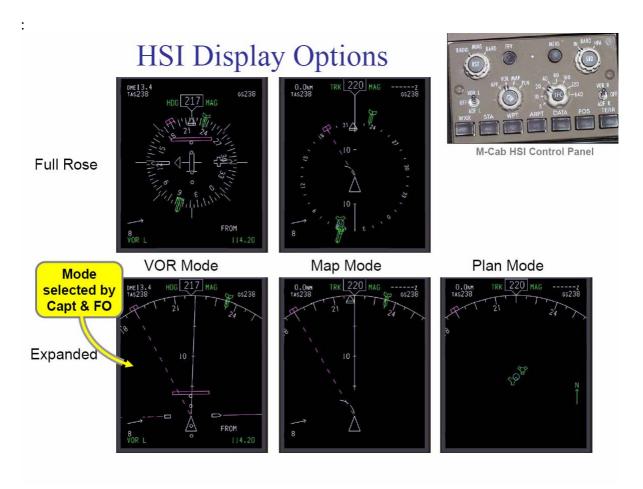
Legend 1 = unknown 2 = unlikely 3 = ruled out

1.16.1.9. Auto Flight Systems

CairoMarch04Slides (March Progress Meeting - Cairo).pdf, 040301 Flash 737 Cairo Mtg (public release version).pdf *Relevant Figures*

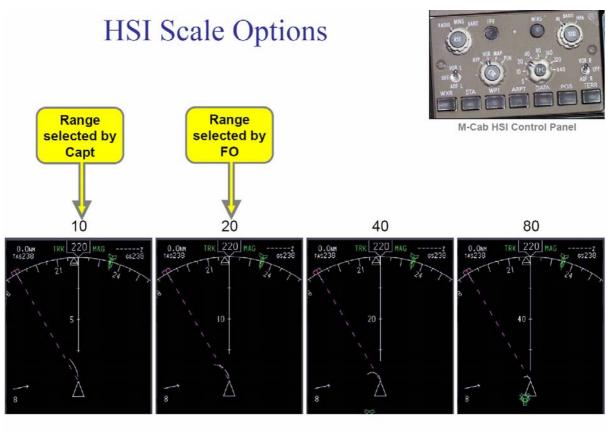
737-300 (PQ294) Flight Director Control Law: (see also FDControlLaw.pdf file)

HSI Display



Signal Name	Bit True	Bit False	Capt	FO)
FULL COMPASS ROSE	SELECT	NOT SEL	0	0
AIRPORTS	SELECT	NOT SEL	0	0
RTE DATA	SELECTED	NOT SEL	0	0
WPT	SELECT	NOT SEL	0	0
NAV AIDS	SELECT	NOT SEL	0	0
SPARE	SELECTED	NOT SEL		
NAV MODE SELECTED	SELECT	NOT SEL	0	0
ILS (STD) MODE SEL	ILS (STD)	NOT SEL	0	0
VOR (STD) MODE SEL	VOR (STD)	NOT SEL	0	0
PLAN MODE SEL	PLAN MODE	NOT SEL	0	0
ILS (MOD) MODE SEL	ILS (MOD)	NOT SEL	0	0
VOR (MOD) MODE SEL	VOR (MOD)	NOT SEL	1	1
MAP MODE SELECT	MAP MODE	NOT SEL	0	0
160 MI RANGE SEL	SET	NOT SET	0	0
80 MI RANGE SEL	SET	NOT SET	0	0
40 MI RANGE SEL	SET	NOT SET	0	0
20 MI RANGE SEL	SET	NOT SET	1	0
10 MI RANGE SEL	SET	NOT SET	0	1
WXR DATA	WXR SEL	NOT SEL	0	0 to 1 @ 530-534

Display Settings from FDR

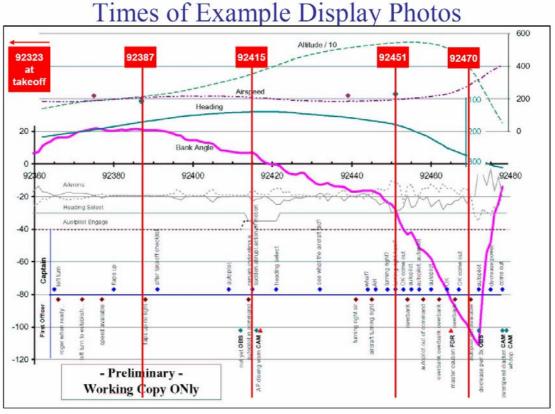


Boeing Proprietary

Note:

Remaining information is Boeing Proprietary information and will not be available for public use

Times of Example Display Photos:



Times of Example Display Photos

Boeing Proprietary

-











M-Cab Flight Director Commands (Flight Director Results Boeing.xls)

Display Architecture (Display Architecture.ppt)

Cairo March 04 Autopilot Flash 737 March Progress Meeting Flash 737 March Progress, 040301 Flash 737 Cairo Mtg (public release version).pdf

Autopilot Engagement Observations

Autopilot Engagement Observations

• Engage Hold Interlocks

- essentially the same as pre-engage interlocks, see table
- would need to have failed within the 3 seconds since engagement

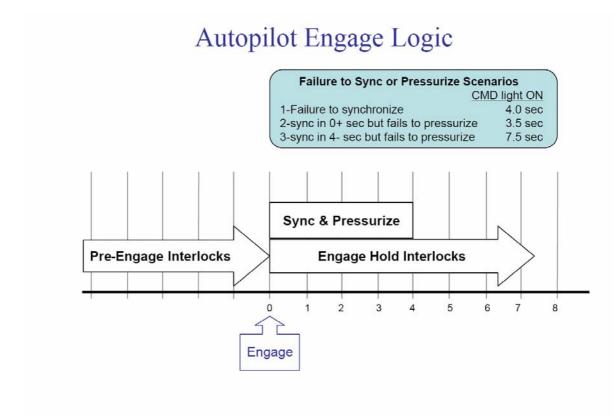
Engage Synchronization

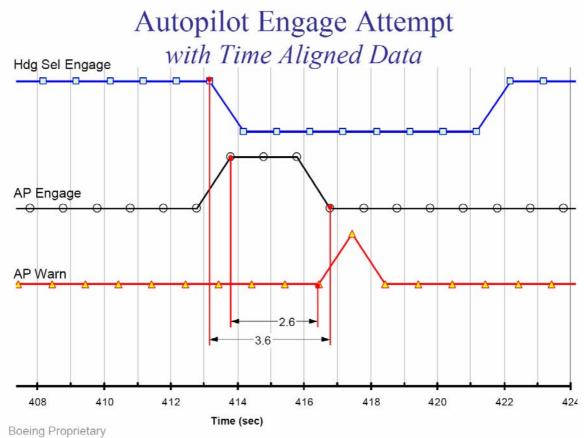
- syncs AP servo to aft quadrant
- FCC allows 4.0 seconds to complete
- Manually Disconnected

Autopilot Engage Logic

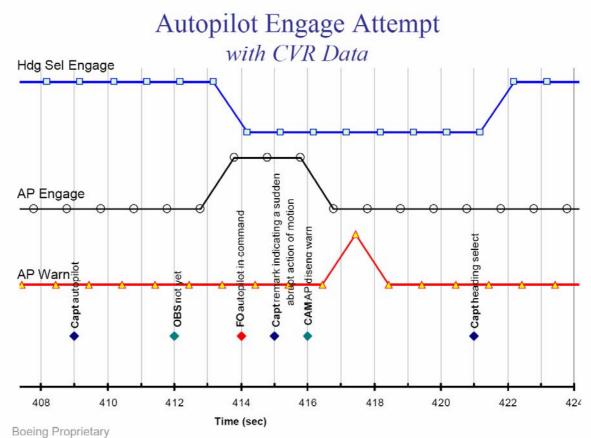
Autopilot Engage & Engage Hold Interlocks

	Pre- Engage	Engage Hold
Condition	Prevent Engage	Cause Disengage
Pitch CWS force greater than 5 lbs	Х	
Roll CWS force greater than 2.25 lbs	Х	
Elevator Detent Pressure Switch Indicates Pressurized	х	
Aileron Detent Pressure Switch Indicates Pressurized	X	
Auto Stab Trim Cutout Switch in Cutout	Х	х
Both Flap Switches and Stab Trim Motor don't agree as Flaps Up or as Flaps Down	х	х
Main Electric Trim Switch Activated	Х	××
Aileron Force Limiter position does not agree with Flaps UP or Flaps Down	х	Х
CAS Invalid	Х	Х
Uncorrected Altitude Invalid	х	×
26 VAC 400 Hz Invalid	Х	х
MCP to FCC Bus Invalid	Х	× × ×
Pitch Angle Invalid	Х	х
Pitch Rate Invalid	Х	х
Roll Angle Invalid	× ×	× × ×
Roll Rate Invalid	Х	Х
Baro Altitude Invalid (Prevents CMD only)	Х	Х
Elevator Detent Pressure Switch Indicates Non-Pressurized		Х
Aileron Detent Pressure Switch Indicates Non-Pressurized		х
(Magnetic Heading OR TAS Invalid) AND		
(Roll CWS) AND	х	х
(Bank Angle <8 degrees)		



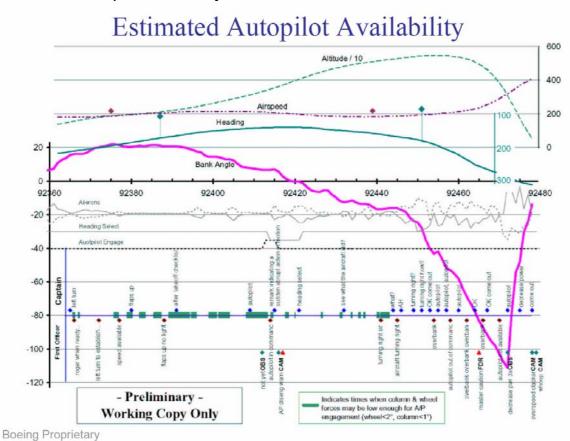


Autopilot Engage Attempt- with Time Aligned Data



Autopilot Engage Attempt- with CVR Data

Estimated Autopilot Availability



AP Actuator description and Scenario 12 info b.pdf, AP Actuator description and Scenario 12 info 2.ppt

Scenario 12 ver 2.ppt (Rev - 3 Feb 05)

Honeywell SP-300 DFCS B737-300.ppt

Flash Airlines Presentation SP-300 DFCS Health Monitoring Honeywell.ppt

1.16.1.9. Flash Airlines AI236 RAM Simulator Configuration (Flash Airlines AI236 RAM Simulator Configuration.htm, Program_Pins.pdf)

1.16.1.10. Boeing response to raised questions.doc

References
17833 (B-H200-17833-ASI 12 Feb 2004).pdf
CairoMarch04Slides (March Progress Meeting - Cairo).pdf
17848 (B-H200-17848-ASI 04 March 2004).pdf
Cairo March 04 Autopilot Flash 737 March Progress Meeting Flash 737 March Progress
Flash Airlines Autopilot Answer to Questions - 31 Jan 2005.ppt
Answers to question_cairo meeting05.ppt
Action Item Response.ppt (Cairo meeting, 1-30-05 to 2-2-05)

17833 (B-H200-17833-ASI 12 Feb 2004).pdf

CairoMarch04Slides (March Progress Meeting - Cairo).pdf, 040301 Flash 737 Cairo Mtg (public release version).pdf

Lateral System-answers to questions

- A2) What is the effect of hydraulic systems failures on the flight controls? *Answer: The hydraulic system arrangement for the 737-300 flight controls is provided in the attached figure. This figure shows which functions would be lost in the event of either an A or B hydraulic system failure.*
- A7) What are the aileron travel rates with various hydraulic system availability?
 - Answer: The aileron PCUs are significantly oversized. Because of this, aileron travel rates are not a function of hydraulic system availability. i.e. aileron travel rates are not significantly different whether either or both hydraulic system is pressurized. For reference, the no load rate is approximately 54 degrees per second of aileron.
- B1) Correlation between control inputs and flight control surface deflections, with special emphasis on the inconsistency of control wheel and aileron surface deflection as indicated by the FDR.

Answer: A kinematic consistency check and a simulator proof of match is being accomplished on the accident data at Boeing. This work is still in progress, however, we have been able to make a few observations on the bias in control wheel position. There is a bias in control wheel position that shifts over time, and possibly a scaling issue. Both issues are being further analyzed for possible explanations. (1)

- B2) Investigate the changes in aileron deflection bias.
 Answer: The changes in aileron position bias are caused by the airload on the aileron reacting against the wing cable run between the aileron and aileron PCU. Therefore, the bias in aileron position is due to aileron hinge moment which varies as a function of airspeed. ⁽¹⁾
- B7) Investigate the effect of flight control surface failures for surfaces like spoiler deflections that are not recorded on the FDR.

Answer: The effects of various spoiler failures are being examined using the Boeing simulation. These results are expected to be available for the next progress meeting in Cairo.

Autopilot - Answers To Questions

A1) Why did the autopilot disengage?

Answer: There are three possible reasons why the autopilot disengaged: the engage synchronization (actuator to surface) failed to complete; the engage hold interlocks were not satisfied; or it was manually disconnected. Based on the data recorded on the FDR, we are not able to pinpoint which of these caused the autopilot to disengage. Additional information on the details of the interlocks and their operation are provided in the Airplane Maintenance Manual section 22-11-01. \bigcirc

- B3) Investigate the cause(s) for the autopilot disconnect. Answer: See response to question A1.
- B6) Investigate availability of autopilot during the captain's requests for "autopilot, autopilot". Answer: The autopilot will not initiate the engage sequence if the A/P engage interlocks are not satisfied (ref AMM 22-11-01 page 54). If the engage interlocks are not satisfied, the attempt to engage (A/P button push) will not be recorded on the FDR. In the case of the accident flight it's possible that forces on the column or wheel prevented the engage logic from being satisfied. Additional information on the details of the interlocks and their operation are provided in the Airplane Maintenance Manual section 22-11-01. ①

FD-answers to questions

- A3) What does the FD command? Roll rate? Bank angle? *Answer: The Flight Director (FD) produces a roll and roll rate command to zero the error between the selected heading and the magnetic heading.* ①
- A5) What does the flight director do when the airplane bank angle exceeds the selected bank angle limit?
 Answer: It will produce a command to fly back to the desired bank angle. ①
- A6) What does the flight director do when the airplane roll rate exceeds the intended roll rate? *Answer: It will produce a command to fly back to the desired bank angle.* ①
- A8) How is Selected Heading recorded on the FDR if it is being turned while the knob is being moved)

Answer: The FCC transmits the Hdg Sel value to the DFDAU at a rate of 20 times per second. The DFDAU then takes the latest value once each 64 seconds and sends it to the DFDR for recording. Thus, if Hdg Sel is dynamically changing when the once-per-64-seconds sample is taken, it will record the Hdg Sel value at the time the sample was taken.

B4) Investigate the cause for Hdg Sel disengage when the autopilot was engaged. *Answer: If the FD command is greater than 7 degrees at the time autopilot engagement is attempted, the Heading Select mode will be reset and the roll mode will default to CWS. According to the FDR data, this seems consistent with the probable flight director command which existed when A/P engagement was initiated.* ①

Other-answers to questions

- A4) Please provide the FMEA for the 737-300 autopilot and flight controls related to the roll axis. *Answer: The following documents were mailed to the NTSB, MCA and BEA: D6-14070 737-300 Lateral Failure Analysis (7MB) D6-37432 737-300 Autopilot Failure Analysis (20MB)*
- A9) Is the hydraulic pump capable of outputting 5000 psi of pressure?

Answer: The following two failures are required In order to reach 5000 psi: /1/pump compensator failed open (full flow), and /2/ system relief valve failed closed. For the hydraulic system pressure display, in-range is considered to be from -100 to 4,100 psi, so 5000 psi would be out of range. If the system were to actually go to 5000 psi, the affected hydraulic pressure display (on the EIS) would slew to it's lower stop; hold for 2 seconds then the pointer would disappear and dashes would appear in the display.

A10) What caused the Master Caution discrete late in the flight?

Status: The Master Caution discrete occurs at time 92465 in the FDR data file received by Boeing. There are over 40 inputs that could have caused this discrete to be set. We are still evaluating the possible causes of the setting of this discrete, and expect to have an update for the next progress meeting in Cairo. We did notice that the Master Caution discrete was set several times on previous flights. Airplane records, such as technical log entries, may record the reason for previous Master Caution events. These records may help isolate why the Master Caution was set at time 92465 in the accident flight.

Displays-answers to questions

B5) Investigate the possible failure modes of the Flight Director indicator.
 Status: This is being researched. We will have some preliminary data available to discuss during the next progress meeting in Cairo.

17848 (B-H200-17848-ASI 04 March 2004).pdf, 17848 (public release).pdf

Enclosure to B-H200-17848-ASI

Responses to Queries Flash Airlines 737 SU-ZCF Accident at Sharm el Skeikh – 3 Jan 04

Questions from 1 March 04

- How is drift angle matched in KINCON with corrected accelerations? Response: Wheel-well based accelerometer data recorded on the FDR are integrated and converted into a ground speed vectors and altitude. Using IRU information, the ground speed vectors are converted into a drift angle and ground speed. The calculated altitude, drift angle and ground speed are then compared to the recorded altitude and the FMC's recorded drift angle and ground speed. Differences between the two sets of data are minimized by calculating a unique but constant acceleration bias for each axis. The biases are then applied to the recorded accelerometer data. The biases were calculated based on minimizing the error over the entire accident flight.
- 2) With the simulator match data vs FDR data, at the end of the flight when rolling back towards wings level, time 92470 thru the end of data, why does the FDR data show the oscillatory motion, but the simulator match does not?
 - Response: The simulator match is an iterative process in which the difference between the simulator behavior and the recorded FDR data is used as a feedback (with a specific gain) to revise the simulator control inputs. In general, a lower gain produces smoother control inputs (lower frequency content) while a higher gain is required to match highly dynamic maneuvers, but can produce significant noise. The gain used in this iteration was chosen to best match the behavior in the time period from 92337 to 92470. Increasing the gain to match the highly dynamic portion of the flight after time 92470 would have introduced significant noise into the earlier portion of the simulation.
- 3) From FDR time 92470 thru the end of data, are the aileron rates seen on the FDR within the capability of the system (i.e. is it real)? *Response: Yes, the aileron rates seen at the end of the FDR data are within the capability of the system.*
- 4) With respect to the FDR recorded wheel position data, the wheel bias in the air, just after takeoff, is different on the accident flight than the previous flight, Why? *Response: The bias in the recorded control wheel signal appears to change on numerous occasions. As noted in the earlier presentation material, the bias changes during the control wheel sweep prior to every takeoff. In addition, the bias appears to change during every climb out, typically between takeoff and flaps up. Furthermore, the bias of appears to change just prior to landing, either during descent or approach. See attached slides that show the changing wheel bias for the accident flight and the previous flight. Similar behavior is noted in all flights, including the first recorded landing, control sweep and takeoff from Abu Simbel. The behavior of the recorded FDR wheel signal appears consistent with a slipping synchro body.*

Boeing Proprietary

Responses to Queries Flash Airlines 737 SU-ZCF Accident at Sharm el Skeikh – 3 Jan 04

- 5) What equation is Boeing using to convert raw data into EU for Wheel Position? *Response: The conversion steps are as follows:*
 - 1. The raw data is first converted to a signed quantity using two's complement.
 - 2. The signed counts (C) are converted to synchro degrees (S) using the formula: S = C * 360 / 1024
 - 3. The synchro degrees (S) are converted to degrees of wheel (W) using the formula: W = S * 150.7663958 / 180

Additional Information: The control wheel sensor on this airplane is a synchro. The synchro signal is interpreted by the FDAU and passed to the flight recorded as counts. Different FDAUs interpret the synchro signal differently. SU-ZCF was equipped with a Sundstrand FDAU which interprets the synchro linearly. Other FDAU's (e.g. Teledyne) use a non-linear interpretation of synchro data. For Sundstrand FDAU's (and any other that interprets synchros linearly), the correct conversion for wheel data is a linear one such as the one shown above in step 2. For a Teledyne FDAU, a non-linear conversion is required. This conversion is built into the RAPS program and is called "dc_TELEDYNE_SYNCHRO". It would not be appropriate to use this function for converting data from a Sundstrand FDAU, such as the SU-ZCF data. In examining the FFD file provided, it appears that this function is being used to convert control wheel data. This conversion will introduce some errors as shown in the attached plots.

The MCA also provide a sheet of paper titled "Analog Signal Description" dated 24 May 1991, with the notation "Project BS7372". The data in this sheet appears to match the D6-55333 data for the 737-2 data frame with 2 exceptions:

D6-55333 defines control wheel as a 10 bit signal. BS7372 lists the signal as a 12 bit signal. The lower two bits of the actual dataframe are used to discrete bits. If both these bits are set, than a wheel position error of \sim 0.22 degrees will result.

The scaling of the BS7372 differs by a small amount from that of D6-55333. Note: The BS7372 sheet lists separate "Breakpoints" in the data. These "break points" exist to account for the signed nature of the signal (it wraps around from maximum counts back to zero). The function of the "break point" in the BS7372 data is accomplished by the two's complement function listed above and that also exists in the RAPS conversion listed in the FFD file provided.

Boeing Proprietary

Responses to Queries Flash Airlines 737 SU-ZCF Accident at Sharm el Skeikh – 3 Jan 04

- Please provide a schematic showing the dual concentric control valve in the aileron PCU, and how it attaches to the PCU input rod. *Response: Schematics provided.*
- What bias springs are present on the PCU valve, and which direction are they biased? *Response: Schematics provided.*
 - response. Senemanes provided.
- Is there any delay between the time the autopilot is disconnected and when the disconnect warning is issued.
 Response: The MCP monitors the CMD and CWS discretes from the FCC and

immediately sets the warning (light and aural) when an autopilot disconnect is detected.

- 9) What method does Boeing use to perform differentiation on flight data? Is there software available for purchase, or what is our algorithm?
 - Response: Without knowing the specifics of the differentiation in question, we can provide a very general answer. Because of the inherent noise associated with differentiation, Boeing tends to avoid differentiation of recorded signals where possible. In some cases, when differentiation is required, we have first modeled the recorded data with a curve fit known to have continuous derivatives and then performed the differentiation on the fitted curve. In other cases, it is possible to take advantage of the known behavior of specific physical quantities and required relationships between different recorded signals when differentiation is required.

Questions from 2 March 04

- Relative to the photo at time 92415, does the "CMD" and "CWS R" text appear on the EADI immediately when the cmd button is pushed or does it wait until the FCC has completed sync & pressurize (i.e. connected to system)? *Response: Immediately when CMD is received from the MCP (button push or paddle lift) the FCC retransmits it to the EFIS processor for display on the EADI.*
- 2) Would the roll FD bar really disappear when Hdg Sel was re-set during AP engage. The photo shows the bar gone because Hdg Sel had reset. *Response: Yes, the FD bar will be biased out of view in this situation.*
- 3) How does CWS R mode work? Response: In CWS R, the autopilot will enter Heading Hold if the bank angle is less than or equal to 8 degrees or Bank Angle Hold if bank angle is greater than 8 degrees (if bank angle is greater than 30 it will return the airplane to 30).

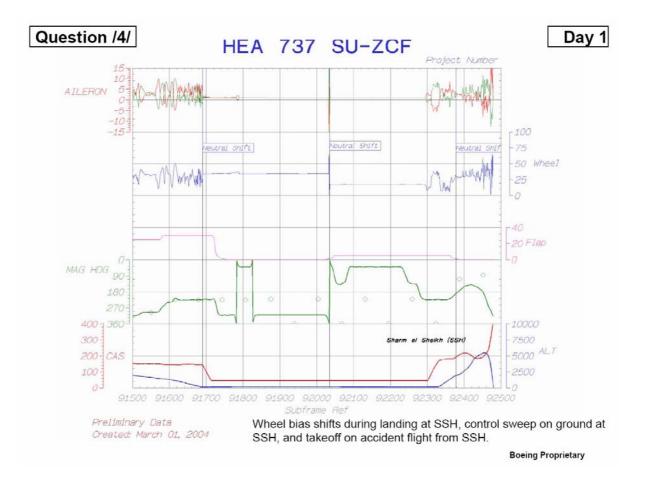
Boeing Proprietary

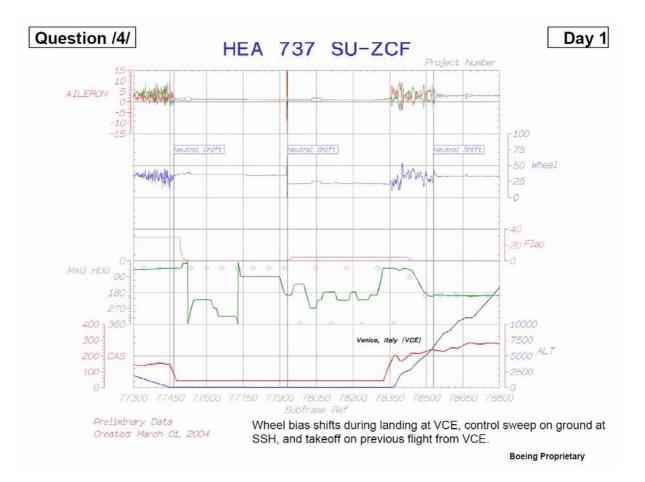
Enclosure to B-H200-17848-ASI

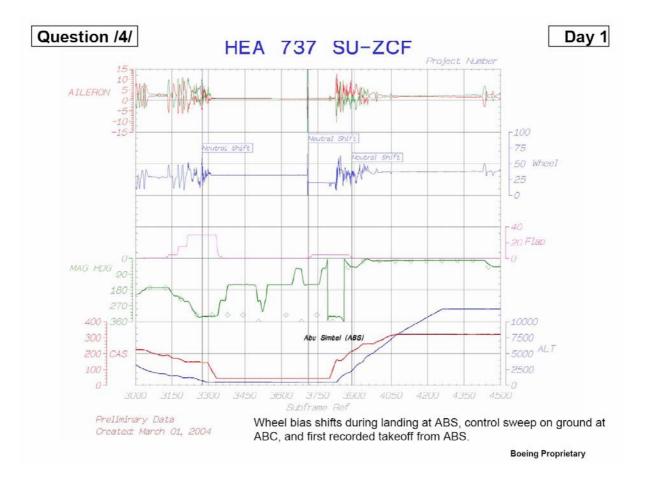
Responses to Queries Flash Airlines 737 SU-ZCF Accident at Sharm el Skeikh – 3 Jan 04

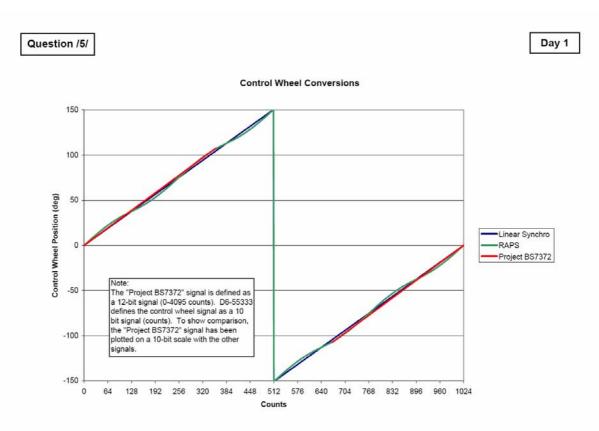
Relative to the photo at time 92470, does the EADI have the feature that forces the blue/brown line to always be present, even in unusual attitudes?
 Response: Yes, the forced blue/brown interface is present unless pitch attitude exceeds 85 degrees (up or down), at which point it is removed.

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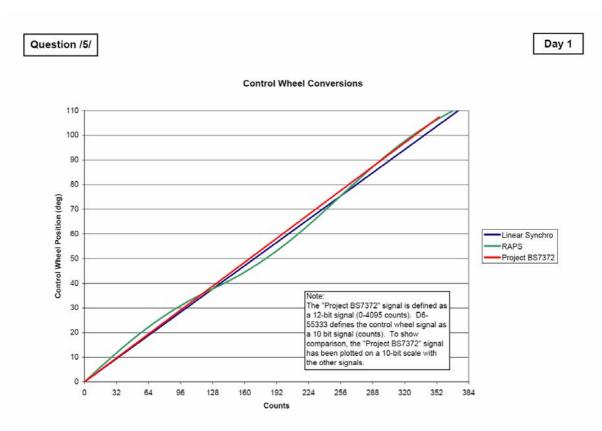




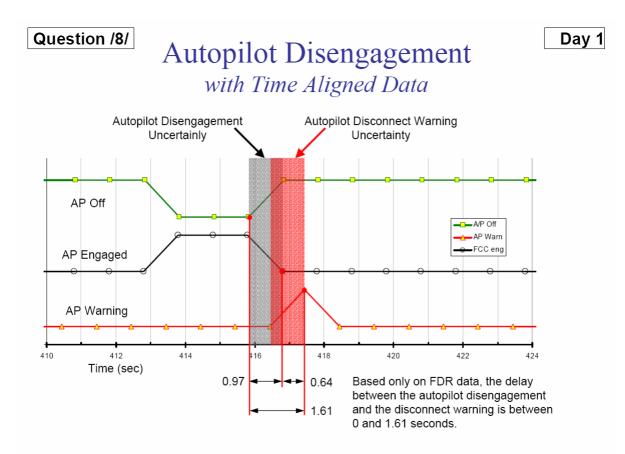




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Cairo March 04 Autopilot Flash 737 March Progress Meeting Flash 737 March Progress

Boeing proprietary information and will not be available for public use

Flash Airlines Autopilot Answer to Questions - 31 Jan 2005.ppt

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Answers to question_cairo meeting05.ppt

Boeing/ Honeywell proprietary information and will not be available for public use

Action Item Response.ppt (Cairo meeting, 1-30-05 to 2-2-05), Boeing Action Items of 30 January (public release).ppt

Question 1

Does the aileron PCU bypass valve interconnect the extend and retract side of the main ram when no hydraulic pressure is available?

What is the correct hydraulic schematic for the PCU?

Question 2

Q) Reference Scenario 9 - What will happen to lateral trim capability after the 12 degrees of lost motion is taken up?

A) Lateral trim capability will be limited to \pm 12 degrees of wheel. The force required to break out the transfer mechanism (50 Lb) is in excess of the feel and centering force (~20 Lb peak).

Question 3

What is the airplane level effect of lateral control scenario #9 (spoiler control drum jammed at neutral)?

Boeing to run desktop simulation

Question 4

Provide proposed corrections to scenario #10 write up See rewrite.

Question 5

Q) Reference Scenario 9-10 – What is breakout force of the aileron spring cartridge?
A) Breakout force of the aileron spring cartridge (reflected at the control wheel) is approximately 16 Lb.

Question 6

Q) Reference Scenario 16 – What is the effect of a failure in the PCA input rod (A or B)?
A) There is no functional effect of a single failure in the PCA input rod. The entire input rod and fasteners are dual load path. The effect of a multiple failure depends on the position of the primary slide at the time of the failure. Worst case effect is a rate jam of the affected PCU, causing a force fight with the other PCU and stalling of both PCUs. Control of spoilers is available from the FO side if the transfer mechanism is broken out. Lateral trim will not be available. Depressurizing the affected PCU will restore normal control.

Question 7

Q) Reference Scenario 17 – What is the effect of a jam between the primary and secondary slide in the aileron PCA?

1. If the primary slide and secondary slide jam together near neutral, the effect is a minor reduction in rate capability.

2. If the jam occurs away from neutral, the feedback motion of the PCU will cause the primary and secondary slides to counter each other (crossflow condition). At a full crossflow condition, the PCU will lose rate capability and be backdriven by the unaffected PCU. Question 8

Q) Reference Scenario 18 – What is the effect of a jam between the secondary slide and the sleeve in the aileron PCA?

If the secondary slide jams near neutral, the effect is a minor reduction in rate capability.
 If the jam occurs away from neutral, the feedback motion of the PCU will cause the primary and secondary slides to counter each other (crossflow condition). At a full crossflow condition, the PCU will lose rate capability and be backdriven by the unaffected PCU.

Question 9, 10

Q) Reference Scenarios 20, 21 – What is the effect of a piston to cylinder jam in the aileron PCA?

The effect is same as a jam elsewhere in the captain's side aileron control path. The FO must break out the transfer mechanism and aileron spring rod to move the spoilers. Aileron control is limited to deflections within the valve stops.

Question 11 Provide proposed corrections to scenario #34 write up See rewrite.

Question 12 Provide proposed corrections to scenario #36 write up See rewrite.

Question 13 Provide proposed corrections to scenario #47 write up See rewrite 1.16.2. Tests and researches conducted by NTSB:

c.wheel Dennis Grossi NTSB.ppt



Introduction

Define Sensor Malfunction Evaluate Data Quality Validate Control Wheel Adjustments

6/28/2005

Discussion Points

- Fact Control Wheel Sensor Maximum Minimum Values Recorded on 25-Hours of FDR data (-2.237deg to 81.5 deg)
- Theory Control Wheel Sensor Moved Freely Within Active Range (–2.237 and 81.5 degrees.), But due to Internal Binding of Rotating Components will not Exceed this Range.
- Theory Control Wheel Inputs Outside of Active Range Cause Sensor to Rotate in Mounting Bracket and Reposition Control Wheel Sensor/Cockpit Control Wheel Offset.
- Theory Rapid Control Wheel Inputs Will Also Cause Sensor to Shift in Mounting Bracket.

3

 Theory - Control Wheel Sensor Values Can Be Used to Evaluate Crew Inputs When Sensor Offset can be Derived From Known Control Wheel Position (i.e. Before and After Preflight Control Checks, 0 - Aileron Deflection.)

6/28/2005

Discussion Points (cont.)

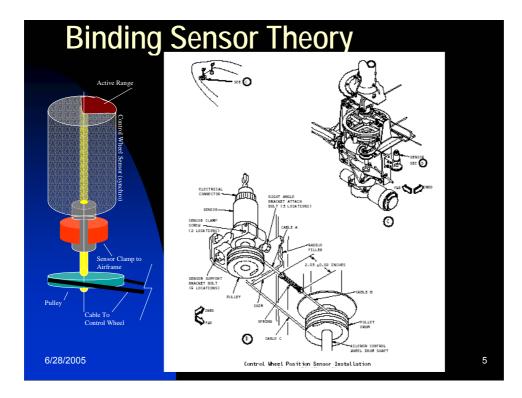
- Control Wheel Position Sensor is a synchro with a range of 0 to 360 degrees or +- 180 degrees.
- Full Range of Control Wheel as expressed in sensor units (synchro angles) is +- 128 degrees.
- Full Range of Control Wheel Travel as measured in cockpit is
 +- 107 degrees.
- The following discussion will reference sensor units only (ie, synchro angles +- 128 degrees)
- Theory Control Wheel Position (Cockpit) values recorded during accident flight can be corrected to actual by applying the following offsets:

From Frame 92250 to 92361.92 subtract 17.5444 deg.

From Frame 92362.42 to 92445 subtract 28.9 deg.

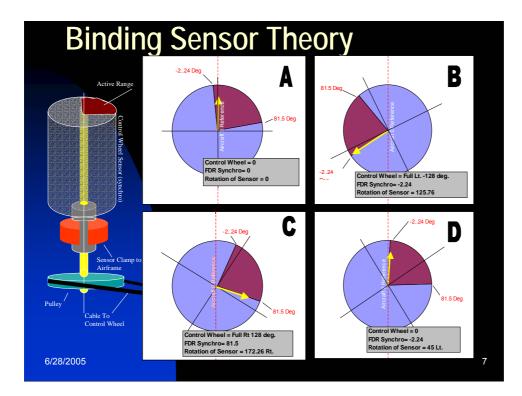
From Frame 92446 to end of data 28.9 deg sensor offset may not apply due to rapid control wheel inputs.

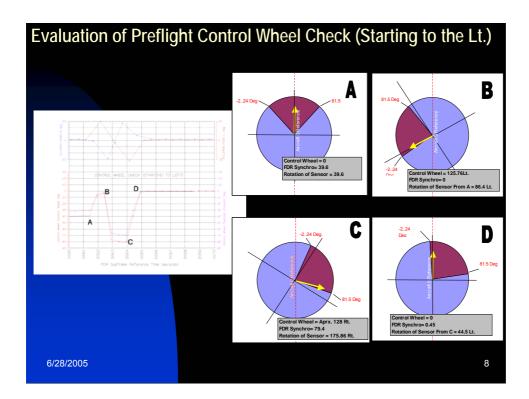
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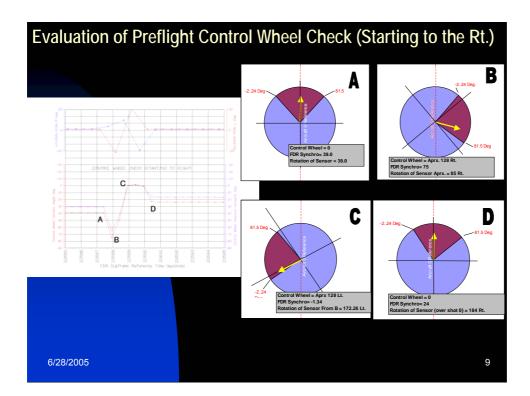


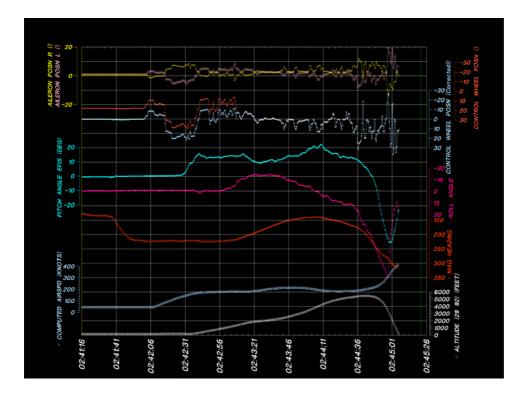
Control Wheel Position Sensor Values for Neutral Aileron Before & After Preflight Control Checks

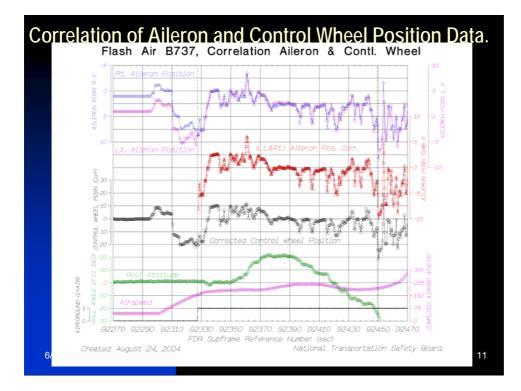
	Time in Seconds	Control W	/heel Position	Control Check
	(FDR Sub Frame)	Before Check	After Check	Direction
1	3713	29.4466	16.7846	Rt. To LT.
2	5568	31.2134	0	Lt. To Rt.
3	7801	58.8932	2.35573	Lt. To Rt.
4	9789	33.8636	3.23913	Lt. To Rt.
5	12124	31.8023	0.294466	Lt. To Rt.
6	14134	28.5632	16.4901	Rt. To LT.
7	17431	29.1521	14.7233	Rt. To LT.
8	22682	30.6245	16.7846	Rt. To LT.
9	30419	37.6915	15.012	Rt. To LT.
10	46964	30.6245	14.1344	Rt. To LT.
11	62156	35.6304	15.6067	Rt. To LT.
12	77924	32.9802	17.668	Rt. To LT.
13	92030	33.5691	14.4288	Rt. To LT.
6/28/2005	5			6

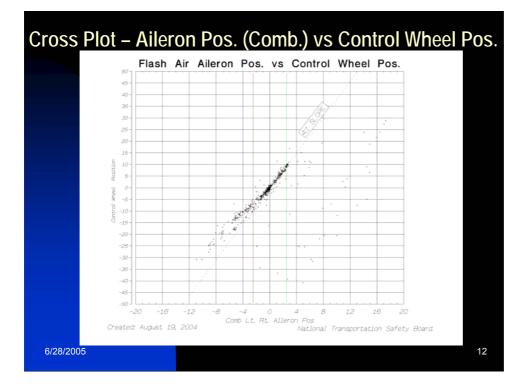




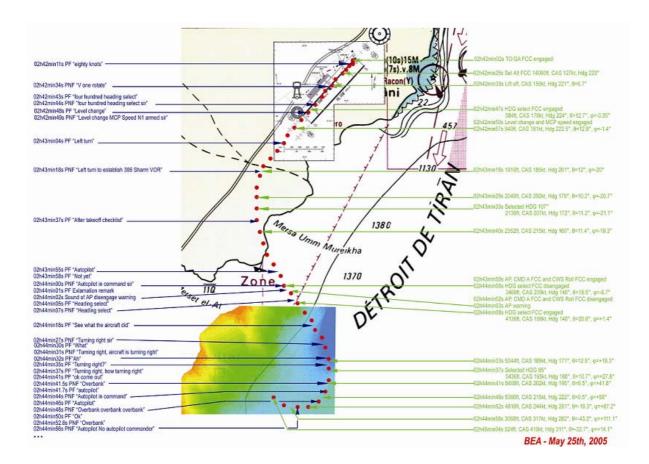








1.16.3. Tests and researches conducted by BEA (Trajecto_may05.jpg)



1.16.4. Tests and researches conducted by MCA:

Spatial Disorientation³

³ All studies are compiled and extracted from the "World Wide Web"

Spatial Disorientation

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PREVENTION OF SPATIAL DISORIENTATION

TREATMENT OF SPATIAL DISORIENTATION

Spatial Disorientation

Spatial disorientation contributes more to causing aircraft accidents than any other physiological problem in flight. Regardless of their flight-time experience, all aircrew members are subject to disorientation. The human body is structured to perceive changes in movement on land in relation to the surface of the earth. In an aircraft, the human sensory systems—the visual, vestibular, and proprioceptive systems—may give the brain erroneous orientation information. This information can cause sensory illusions, which may lead to spatial disorientation.

COMMON TERMS OF SPATIAL DISORIENTATION

SPATIAL DISORIENTION

9-1. Spatial disorientation is an individual's inability to determine his or her position, attitude, and motion relative to the surface of the earth or significant objects; for example, trees, poles, or buildings during hover. When it occurs, pilots are unable to see, believe, interpret, or prove the information derived from their flight instruments. Instead, they rely on the false information that their senses provide.

SENSORY ILLUSION

9-2. A sensory illusion is a false perception of reality caused by the conflict of orientation information from one or more mechanisms of equilibrium. Sensory illusions are a major cause of spatial disorientation.

VERTIGO

9-3. Vertigo is a spinning sensation usually caused by a peripheral vestibular abnormality in the middle ear. Aircrew members often misuse the term vertigo, applying it generically to all forms of spatial disorientation or dizziness.

TYPES OF SPATIAL DISORIENTATION

TYPE I (UNRECOGNIZED)

9-4. A disoriented aviator does not perceive any indication of spatial disorientation. In other words, he does not think anything is wrong. What he sees—or thinks he sees—is corroborated by his other senses. Type I disorientation is the most dangerous type of disorientation. The pilot—unaware of a problem—fails to recognize or correct the disorientation, usually resulting in a fatal aircraft mishap:

- The pilot may see the instruments functioning properly. There is no suspicion of an instrument malfunction.
- There may be no indication of aircraft-control malfunction. The aircraft is performing normally.

• An example of this type of SD would be the height-/depth-perception illusion when the pilot descends into the ground or some obstacle above the ground because of a lack of situational awareness.

TYPE II (RECOGNIZED)

9-5. In Type II spatial disorientation, the pilot perceives a problem (resulting from spatial disorientation). The pilot, however, may fail to recognize it as spatial disorientation:

- The pilot may feel that a control is malfunctioning.
- The pilot may perceive an instrument failure as in the graveyard spiral, a classic example of Type II disorientation. The pilot does not correct the aircraft roll, as indicated by the attitude indicator, because his vestibular indications of straight-and-level flight are so strong.

TYPE III (INCAPACITATING)

9-6. In Type III spatial disorientation, the pilot experiences such an overwhelming sensation of movement that he or she cannot orient himself or herself by using visual cues or the aircraft instruments. Type III spatial disorientation is not fatal if the copilot can gain control of the aircraft.

EQUILIBRIUM MAINTENANCE

9-7. Three sensory systems—the visual, vestibular, and proprioceptive systems—are especially important in maintaining equilibrium and balance. Figure 9-1 shows these systems. Normally, the combined functioning of these senses maintains equilibrium and prevents spatial disorientation. During flight, the visual system is the most reliable. In the absence of the visual system, the vestibular and proprioceptive systems are unreliable in flight.

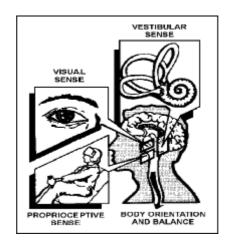


Figure 9-1. The Three Equilibrium Systems

VISUAL SYSTEM

9-8. Of the three sensory systems, the visual system is the most important in maintaining equilibrium and orientation. To some extent, the eyes can help determine the speed and direction of flight by comparing the position of the aircraft relative to some fixed point of

reference. Eighty percent of our orientation information comes from the visual system. (Chapter 8 contains information about the eye).

9-9. On flights under IMC, crew members lose fixed points of reference outside of the aircraft. Under IMC, the pilot must rely on visual sensory input from the instruments for spatial orientation. The decision to rely on the visual sense—and to believe the instruments rather than the input of the other senses—demands disciplined training.

9-10. The eyes allow the pilot to scan sensitive flight instruments that give accurate spatialorientation information. These instruments indicate unusual aircraft attitudes resulting from turbulence, distraction, inattention, mechanical failure, or spatial disorientation.

VESTIBULAR SYSTEM

9-11. The inner ear contains the vestibular system, which contains the motion- and gravitydetecting sense organs. This system is located in the temporal bone on each side of the head. Each vestibular apparatus consists of two distinct structures: the semicircular canals and the vestibule proper, which contain the otolith organs. Figure 9-2 depicts the vestibular system. Both the semicircular canals and the otolith organs sense changes in aircraft attitude. The semicircular canals of the inner ear sense changes in angular acceleration and deceleration.

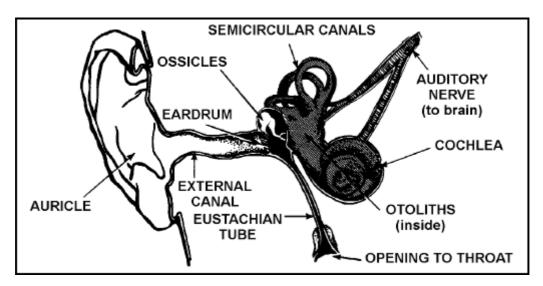


Figure 9-2. The Vestibular System

Otolith Organs

9-12. The otolith organs are small sacs located in the vestibule. Sensory hairs project from each macula into the otolithic membrane, an overlaying gelatinous membrane that contains chalklike crystals, called otoliths. The otolith organs, shown in Figure 9-3, respond to gravity and linear accelerations/decelerations. Changes in the position of the head, relative to the gravitational force, cause the otolithic membrane to shift position on the macula. The sensory hairs bend, signaling a change in the head position.

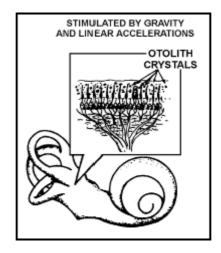


Figure 9-3. The Otolith Organs

9-13. When the head is upright, a "resting" frequency of nerve impulses is generated by the hair cells. Figure 9-4 shows the position of the hair cells when the head is upright.

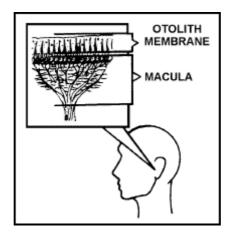


Figure 9-4. Position of the Hair Cells When the Head Is Upright

9-14. When the head is tilted, the "resting" frequency is altered. The brain is informed of the new position. The positions of the hair cells when the head is tilted forward and backward are shown in Figure 9-5.

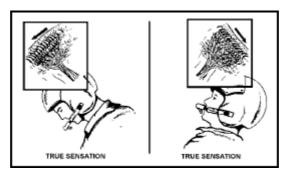


Figure 9-5. Position of the Hair Cells When the Head Is Tilted Forward and Backward

9-15. Linear accelerations/decelerations also stimulate the otolith organs. The body cannot physically distinguish between the inertial forces resulting from linear accelerations and the force of gravity. A forward acceleration results in backward displacement of the otolithic membranes. When an adequate visual reference is not available, aircrew members may experience an illusion of backward tilt. Figure 9-6 shows this false sensation of backward tilt.

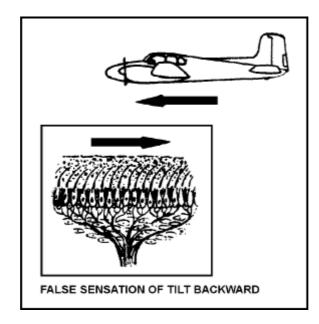


Figure 9-6. False Sensation During Backward Tilt

SEMICIRCULAR CANALS

9-16. The semicircular canals of the inner ear sense changes in angular acceleration. The canals will react to any changes in roll, pitch, or yaw attitude. <u>Figure 9-7</u> shows where these changes are registered in the semicircular canals.

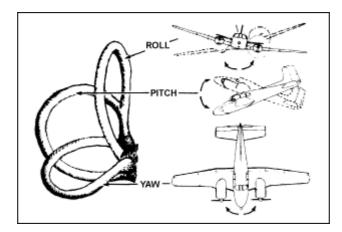


Figure 9-7. Reaction of the Semicircular Canals to Changes in Angular Acceleration

9-17. The semicircular canals are situated in three planes, perpendicular to each other. They are filled with a fluid called endolymph. The inertial torque resulting from angular acceleration in the plane of the canal puts this fluid into motion. The motion of the fluid bends the cupula, a gelatinous structure located in the ampulla of the canal. This, in turn, moves the hairs of the hair cells situated beneath the cupula. This movement stimulates the

vestibular nerve. These nerve impulses are then transmitted to the brain, where they are interpreted as rotation of the head. Figure 9-8 shows a cutaway section of the semicircular canal.

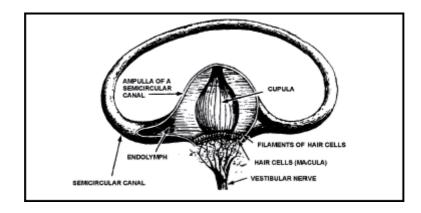


Figure 9-8. Cutaway View of the Semicircular Canals

9-18. When no acceleration takes place, the hair cells are upright. The body senses that no turn has occurred. The position of the hair cells and the actual sensation correspond, as shown in Figure 9-9.

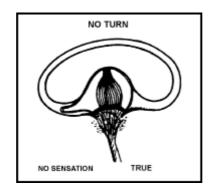


Figure 9-9. Position of Hair Cells During No Acceleration

9-19. When a semicircular canal is put into motion during clockwise acceleration, the fluid within the semicircular canal lags behind the accelerated canal walls. This lag creates a relative counterclockwise movement of the fluid within the canal. The canal wall and the cupula move in the opposite direction from the motion of the fluid. The brain interprets the movement of the hairs to be a turn in the same direction as the canal wall. The body correctly senses that a clockwise turn is being made. Figure 9-10 shows the position of the hair cells and the resulting true sensation during a clockwise turn.

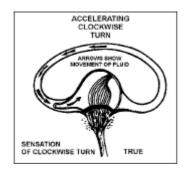


Figure 9-10. Sensation During a Clockwise Turn

9-20. If the clockwise turn then continues at a constant rate for several seconds or longer, the motion of the fluid in the canals catches up with the canal walls. The hairs are no longer bent, and the brain receives the false impression that turning has stopped. The position of the hair cells and the resulting false sensation during a prolonged, constant clockwise turn is shown in Figure 9-11. A prolonged constant turn in either direction will result in the false sensation of no turn.

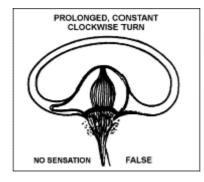


Figure 9-11. Sensation During a Prolonged Clockwise Turn

9-21. When the clockwise rotation of the aircraft slows or stops, the fluid in the canal moves briefly in a clockwise direction. This sends a signal to the brain that is falsely interpreted as body movement in the opposite direction. In an attempt to correct the falsely perceived counterclockwise turn, the pilot may turn the aircraft in the original clockwise direction. Figure 9-12 shows the position of the hair cells—and the resulting false sensation when a clockwise turn is suddenly slowed or stopped.

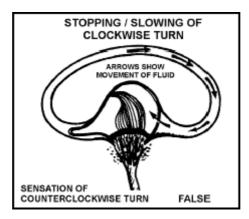


Figure 9-12. Sensation During Slowing or Stopping of a Clockwise Turn

PROPRIOCEPTIVE SYSTEM

9-22. This system reacts to the sensation resulting from pressures on joints, muscles, and skin and from slight changes in the position of internal organs. It is closely associated with the vestibular system and, to a lesser degree, the visual system. Forces act upon the seated pilot in flight. With training and experience, the pilot can easily distinguish the most distinct movements of the aircraft by the pressures of the aircraft seat against the body. The recognition of these movements has led to the term "seat-of-the-pants" flying.

VISUAL ILLUSIONS

9-23. Illusions give false impressions or misconceptions of actual conditions; therefore, aircrew members must understand the type of illusions that can occur and the resulting disorientation. Although the visual system is the most reliable of the senses, some illusions can result from misinterpreting what is seen; what is perceived is not always accurate. Even with the references outside the cockpit and the display of instruments inside, aircrew members must be on guard to interpret information correctly.

RELATIVE-MOTION ILLUSION

9-24. Relative motion is the falsely perceived self-motion in relation to the motion of another object. The most common example is when an individual in a car is stopped at a traffic light and another car pulls alongside. The individual that was stopped at the light perceives the forward motion of the second car as his own motion rearward. This results in the individual applying more pressure to the brakes unnecessarily. This illusion can be encountered during flight in situations such as formation flight, hover taxi, or hovering over water or tall grass.

CONFUSION WITH GROUND LIGHTS

9-25. Confusion with ground lights occurs when an aviator mistakes ground lights for stars. This illusion prompts the aviator to place the aircraft in an unusual attitude to keep the misperceived ground lights above them. Isolated ground lights can appear as stars and this could lead to the illusion that the aircraft is in a nose high or one wing low attitude (Part A of Figure 9-13). When no stars are visible because of overcast conditions, unlighted areas of terrain can blend with the dark overcast to create the illusion that the unlighted terrain is part of the sky (Part B of Figure 9-13). This illusion can be avoided by referencing the flight instruments and establishing a true horizon and attitude.

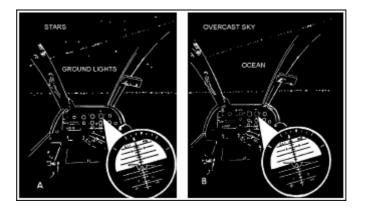


Figure 9-13. Confusion of Ground Lights and Stars at Night

FALSE HORIZON ILLUSION

The false horizon illusion (Figure 9-14) occurs when the aviator confuses cloud formations with the horizon or the ground. This illusion occurs when an aviator subconsciously chooses the only reference point available for orientation. A sloping cloud deck may be difficult to perceive as anything but horizontal if it extends for any great distance in the pilot's peripheral vision. An aviator may perceive the cloudbank below to be horizontal although it may not be

horizontal to the ground; thus, the pilot may fly the aircraft in a banked attitude. This condition is often insidious and goes undetected until the aviator recognizes it and makes the transition to the instruments and corrects it. This illusion can also occur if an aviator looks outside after having given prolonged attention to a task inside the cockpit. The confusion may result in the aviator placing the aircraft parallel to the cloudbank.

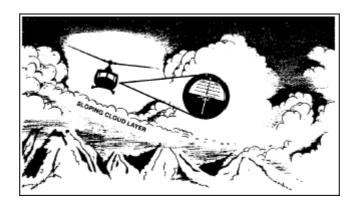


Figure 9-14. False Horizon Illusion

HEIGHT-DEPTH PERCEPTION ILLUSION

2-27. The height-depth perception illusion is due to a lack of sufficient visual cues and causes an aircrew member to lose depth perception. Flying over an area devoid of visual references—such as desert, snow, or water—will deprive the aircrew member of his perception of height. The aviator, misjudging the aircraft's true altitude, may fly the aircraft dangerously low in reference to the ground or other obstacles above the ground. Flight in an area where visibility is restricted by fog, smoke, or haze can produce the same illusion.

CRATER ILLUSION

9-28. The crater illusion occurs when aircrew members land at night, under NVG conditions, and the IR searchlight is directed too far under the nose of the aircraft. This will cause the illusion of landing with up-sloping terrain in all directions. This misperceived up-sloping terrain will give the aviator the perception of landing into a crater. This illusionary depression lulls the pilot into continuing to lower the collective. This can result in the aircraft prematurely impacting the ground, causing damage to both aircraft and crew. If observing another aircraft during hover taxi, the aviator may perceive that the crater actually appears to move with the aircraft being observed.

STRUCTURAL ILLUSIONS

9-29. Structural illusions are caused by the effects of heat waves, rain, snow, sleet, or other visual obscurants. A straight line may appear curved when it is viewed through the heat waves of the desert. A single wing-tip light may appear as a double light or in a different location when it is viewed during a rain shower. The curvature of the aircraft windscreen can also cause structural illusions, as illustrated in Figure 9-15. This illusion is due to the refraction of light rays as they pass through the windscreen. When encountering environments that contain these visual obscurants, the aviator must remain aware that these obscurants may present a false perception.



Figure 9-15. Structural Illusion

SIZE-DISTANCE ILLUSION

9-30. The size-distance illusion (Figure 9-16) is the false perception of distance from an object or the ground, created when a crew member misinterprets an unfamiliar object's size to be the same as an object that he is accustomed to viewing. This illusion can occur if the visual cues, such as a runway or trees, are of a different size than expected. An aviator making an approach to a larger, wider runway may perceive that the aircraft is too low. Conversely, an aviator—making an approach to a smaller, narrower runway—may perceive that the aircraft is too high. A pilot making an approach 25 feet above the trees in the State of Washington, where the average tree is 100 feet tall, may fly the aircraft dangerously low if trying to make the same approach at Fort Rucker, Alabama, where the average tree height is 30 feet. This illusion may also occur when an individual is viewing the position lights of another aircraft at night. If the aircraft being observed suddenly flies into smoke or haze, the aircraft will appear to be farther away than before.

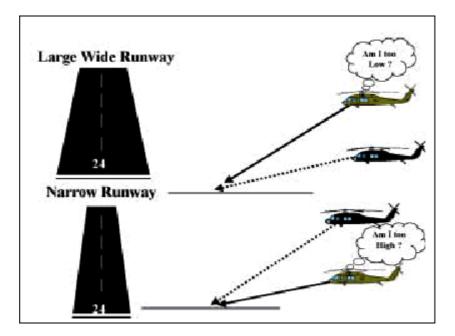


Figure 9-16. Size-Distance Illusion

FASCINATION (FIXATION) IN FLYING

9-31. Fascination, or fixation, flying can be separated into two categories: task saturation and target fixation. Task saturation may occur during the accomplishment of simple tasks within the cockpit. Crew members may become so engrossed with a problem or task within the cockpit that they fail to properly scan outside the aircraft. Target fixation, commonly referred to as target hypnosis, occurs when an aircrew member ignores orientation cues and focuses his attention on his object or goal; for example, an attack pilot on a gunnery range becomes so intent on hitting the target that he forgets to fly the aircraft, resulting in the aircraft striking the ground, the target, or the shrapnel created by hitting the target.

REVERSIBLE PERSPECTIVE ILLUSION

9-32. At night, an aircraft may appear to be moving away when it is actually approaching. If the pilot of each aircraft has the same assumption, and the rate of closure is significant, by the time each pilot realizes the misassumption, it may be too late to avoid a mishap. This illusion is termed reversible perspective and is often experienced when an aircrew member observes an aircraft flying a parallel course. In this situation, aircrew coordination is paramount. To determine the direction of flight, the aircrew member should observe the other aircraft's position lights. Remember the following: red on right returning; that is, if you see an aircraft with the red position light on the right and the green position light on the left, the observed aircraft is traveling in the opposite direction of your flight path.

ALTERED PLANES OF REFERENCE

9-33. In altered planes of reference(Figure 9-17), the pilot has an inaccurate sense of altitude, attitude, or flight-path position in relation to an object so great in size that the object becomes the new plane of reference rather than the correct plane of reference, the horizon. A pilot approaching a line of mountains may feel the need to climb although the altitude of the aircraft is adequate. This is because the horizon, which helps the pilot maintain orientation, is

subconsciously moved to the top of the ridgeline. Without an adequate horizon, the brain attempts to fix a new horizon. Conversely, an aircraft entering a valley that contains a slowly increasing up-slope condition may become trapped because the slope may quickly increase and exceed the ability of the aircraft to climb above the hill, causing the aircraft to crash into the surrounding hills.



Figure 9-17. Altered Planes of Reference

AUTOKINESIS

9-34. Autokinesis primarily occurs at night when ambient visual cues are minimal and a small, dim light is seen against a dark background. After about 6 to 12 seconds of visually fixating on the light, one perceives movement at up to 20 degrees in any particular direction or in several directions in succession, although there is no actual displacement of the object. This illusion may allow an aviator to mistake the object fixated as another aircraft. In addition, a pilot flying at night may perceive a relatively stable lead aircraft to be moving erratically, when in fact, it is not. The unnecessary and undesirable control inputs that the pilot makes to compensate for the illusory movement of the aircraft represent increased work and wasted motion, at best, and an operational hazard at worst.

FLICKER VERTIGO

9-35. Flicker vertigo (Figure 9-18) is technically not an illusion; however, as most people are aware from personal experience, viewing a flickering light can be both distracting and annoying. Flicker vertigo may be created by helicopter rotor blades or airplane propellers interrupting direct sunlight at a rate of 4 to 20 cycles per second. Flashing anticollision strobe lights, especially while the aircraft is in the clouds, can also produce this effect. One should also be aware that photic stimuli at certain frequencies could produce seizures in those rare individuals who are susceptible to flicker-induced epilepsy.

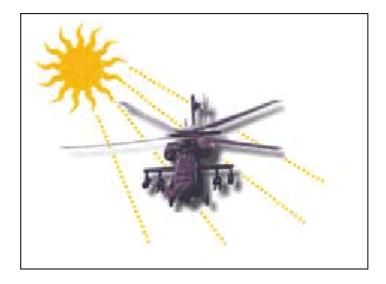


Figure 9-18. Flicker Vertigo

VESTIBULAR ILLUSIONS

9-36. The vestibular system provides accurate information as long as an individual is on the ground. Once the individual is airborne, however, the system may function incorrectly and cause illusions. These illusions pose the greatest problem with spatial disorientation. Aircrew members must understand vestibular illusions and the conditions under which they occur. They must be able to distinguish between the inputs of the vestibular system that are accurate and those that cause illusion.

SOMATOGYRAL ILLUSIONS

9-37. Somatogyral illusions are caused when angular accelerations and decelerations stimulate the semicircular canals. Those that may be encountered in flight are the leans, graveyard spin, and Coriolis illusions.

Leans

9-38. The most common form of spatial disorientation is the leans. This illusion occurs when the pilot fails to perceive angular motion. During continuous straight-and-level flight, the pilot will correctly perceive that he is straight and level (part A, Figure 9-19). However, a pilot rolling into or out of a bank may experience perceptions that disagree with the reading on the attitude indicator. In a slow roll, for instance, the pilot may fail to perceive that the aircraft is no longer vertical. He may feel that his aircraft is still flying straight and level although the attitude indicator shows that the aircraft is in a bank (part B, Figure 9-19). Once the pilot detects the slow roll, he makes a quick recovery. He rolls out of the bank and resumes straight-and-level flight. The pilot may now perceive that the aircraft is banking in the opposite direction. However, the attitude indicator shows the aircraft flying straight and level (part C, Figure 9-19). The pilot may then feel the need to turn the aircraft so that it aligns with the falsely perceived vertical position. Instead, the pilot should maintain straight-and-level flight as shown by the attitude indicator. To counter the falsely perceived vertical position, the pilot will lean his body in the original direction of the subthreshold roll until the false sensation leaves (part D, Figure 9-19).

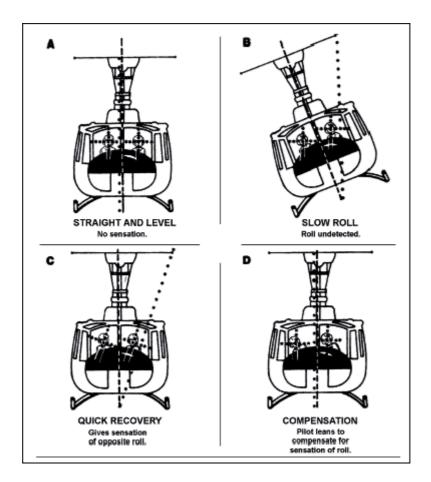


Figure 9-19. Leans

Graveyard Spin

9-39. This illusion, shown in Figure 9-20, usually occurs in fixed-wing aircraft. For example, a pilot enters a spin and remains in it for several seconds. The pilot's semicircular canals reach equilibrium; no motion is perceived. Upon recovering from the spin, the pilot undergoes deceleration, which is sensed by the semicircular canals. The pilot has a strong sensation of being in a spin in the opposite direction even if the flight instruments contradict that perception. If deprived of external visual references, the pilot may disregard the instrumentation and make control corrections against the falsely perceived spin. The aircraft will then reenter a spin in the original direction.

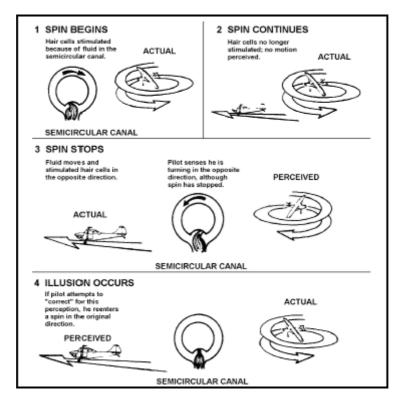


Figure 9-20. Graveyard Spin

9-40. To compound the action of the semicircular canals under these conditions, a pilot, noting a loss of altitude as the spin develops, may apply back pressure on the controls and add power in an attempt to gain altitude. This maneuver tightens the spin and may cause the pilot to lose control of the aircraft.

Coriolis Illusion

9-41. Regardless of the type of aircraft flown, the Coriolis illusion is the most dangerous of all vestibular illusions. It causes overwhelming disorientation.

9-42. This illusion occurs whenever a prolonged turn is initiated and the pilot makes a head motion in a different geometrical plane. When a pilot enters a turn and then remains in the turn, the semicircular canal corresponding to the yaw axis is equalized. The endolymph fluid no longer deviates, or bends, the cupula. Figure 9-21 shows the movement of the fluid in a semicircular canal when a pilot enters a turn.

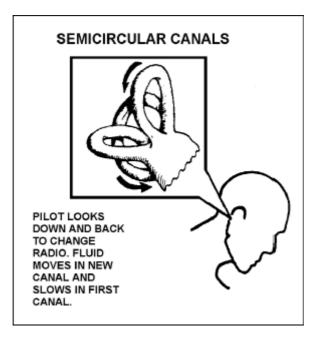


Figure 9-21. Movement of Fluid in the Semicircular Canals During a Turn

9-43. If the pilot initiates a head movement in a geometrical plane other than that of the turn, the yaw axis semicircular canal is moved from the plane of rotation to a new plane of nonrotation. The fluid then slows in that canal, resulting in a sensation of a turn in the direction opposite that of the original turn.

9-44. Simultaneously, the two other canals are brought within a plane of rotation. The fluid stimulates the two other cupulas. The combined effect of the coupler deflection in all three canals creates the new perception of motion in three different planes of rotation: yaw, pitch, and roll. The pilot experiences an overwhelming head-over-heels tumbling sensation.

SOMATOGRAVIC ILLUSIONS

9-45. Somatogravic illusions are caused by changes in linear accelerations and decelerations or gravity that stimulate the otolith organs. The three types of somatogravic illusions that can be encountered in flight are oculogravic, elevator, and oculoagravic.

Oculogravic Illusion

9-46. This type of illusion occurs when an aircraft accelerates and decelerates. Inertia from linear accelerations and decelerations cause the otolith organ to sense a nose-high or nose-low attitude. In a linear acceleration, the gelatinous layer, which contains the otolith organ, is shifted aft. The aviator falsely perceives that the aircraft is in a nose-high attitude. A pilot correcting for this illusion without cross-checking the instruments would most likely dive the aircraft. This illusion does not occur if adequate outside references are available. If making an instrument approach in inclement weather or in darkness, the pilot would be considerably more susceptible to the oculogravic illusion. An intuitive reaction to the sensed nose-high attitude could have catastrophic results

Elevator Illusion

9-47. This illusion occurs during upward acceleration. Because of the inertia encountered, the pilot's eyes will track downward as his body tries, through inputs supplied by the inner ear, to maintain visual fixation on the environment or instrument panel. With the eyes downward, the pilot will sense that the nose of the aircraft is rising. This illusion is common for aviators flying aircraft that encounter updrafts.

Oculoagravic Illusion

9-48. This illusion is the opposite of the elevator illusion and results from the downward movement of the aircraft. Because of the inertia encountered, the pilot's eyes will track upward. The pilot's senses then usually indicate that the aircraft is in a nose-low attitude. This illusion is commonly encountered as a helicopter enters autorotation. The pilot's usual intuitive response is to add aft cyclic, which decreases airspeed below the desired level.

PROPRIOCEPTIVE ILLUSIONS

9-49. Proprioceptive illusions rarely occur alone. They are closely associated with the vestibular system and, to a lesser degree, with the visual system. The proprioceptive information input to the brain may also lead to a false perception of true vertical. During turns, banks, climbs, and descending maneuvers, proprioceptive information is fed into the central nervous system. A properly executed turn vectors gravity and centrifugal force through the vertical axis of the aircraft. Without visual reference, the body only senses being pressed firmly into the seat. Because this sensation is normally associated with climbs, the pilot may falsely interpret it as such. Recovering from turns lightens pressure on the seat and creates an illusion of descending. This false perception of descent may cause the pilot to pull back on the stick, which would reduce airspeed. Figure 9-22 shows proprioceptive illusions.

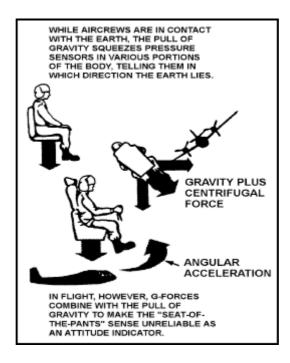


Figure 9-22. Proprioceptive Illusions

PREVENTION OF SPATIAL DISORIENTATION

9-50. Spatial disorientation cannot be totally eliminated. However, aircrew members need to remember that misleading sensations from sensory systems are predictable. These sensations can happen to anyone because they are due to the normal functions and limitations of the senses. Training, instrument proficiency, good health, and aircraft design minimize spatial disorientation. Spatial disorientation becomes dangerous when pilots become incapable of making their instruments read right. All pilots, regardless of experience level, can experience spatial disorientation. For that reason, they should be aware of the potential hazards, understand their significance, and learn to overcome them. To prevent disorientation, aviators should—

- Never fly without visual reference points (either the actual horizon or the artificial horizon provided by the instruments).
- Trust the instruments.
- Avoid fatigue, smoking, hypoglycemia, hypoxia, and anxiety, which all heighten illusions.
- Never try to fly VMC and IMC at the same time.

TREATMENT OF SPATIAL DISORIENTATION

9-51. Spatial disorientation can easily occur in the aviation environment. If disorientation occurs, aviators should—

- Refer to the instruments and develop a good cross-check.
- Delay intuitive actions long enough to check both visual references and instruments.
- Transfer control to the other pilot if two pilots are in the aircraft. Rarely will both experience disorientation at the same time.

Note:

The following references are made available for the specialized investigation group to assist in the studies.

- Surviving Spatial Disorientation
- Spatial Disorientation, From Wikipedia, the free encyclopedia.
- Spatial Disorientation Why you shouldn't fly by the seat of your pants
- Spatial Disorientation Deaths of Visual Flight Rules Pilots: J. F. Kennedy, Jr., et. al.
- Spatial Disorientation Stories, From AVWEB Question Of The Week

1.16.5 Systems examination:

- 1.16.5.1 *Cause(s) for the autopilot disconnect* (Refer to 1.16.1. (Tests and Researches), Cairo March 04 Autopilot Flash 737 March Progress Meeting Flash 737 March Progress, Autopilot Engagement)
- 1.16.5.2 Cause(s) for "Heading Select" disengage when the autopilot is engaged (applied also to the accident aircraft) (Refer to 1.16.1. (Tests and Researches), Boeing response to the raised questions, enclosure to B-H200-17833-ASI Question B4)
- 1.16.5.3 Availability of autopilot during the captain's requests "autopilot, autopilot" (accident aircraft)
 (Refer to 1.16.1. (Tests and Researches), Cairo March 04 Autopilot Flash 737 March Progress Meeting Flash 737 March Progress, Estimated Autopilot Availability, Boeing response to the raised questions, enclosure to B-H200-17833-ASI Question B6)
- 1.16.5.4 MMEL issues associated with operating the airplane with FD TO/GA mode inoperative (won't stay engaged)
 Relevant information to be added upon Human Factors Group discretion
- 1.16.5.5 Interlock logic for A/P with the definition of the likelyhood (ruled out, not likely, unknown) to the various interlocks regarding the role they may have played in the autopilot disengagement (Refer to 1.16.1. (Tests and Researches), Honeywell SP-300 DFCS B737-300.ppt file, and Flash Airlines Presentation SP-300 DFCS Health Monitoring Honeywell.ppt file)
- 1.16.5.6 The effects of the TOGA bit dropping out and way it affects the command bars.
 (Refer to 1.16.1. (Tests and Researches), Boeing AMM 22-03-00, 22-04-00)
- 1.16.5.7 *Examination of the selected course compared to the selected heading (probability for having "dropouts").*

1.16.6 CVR examination:

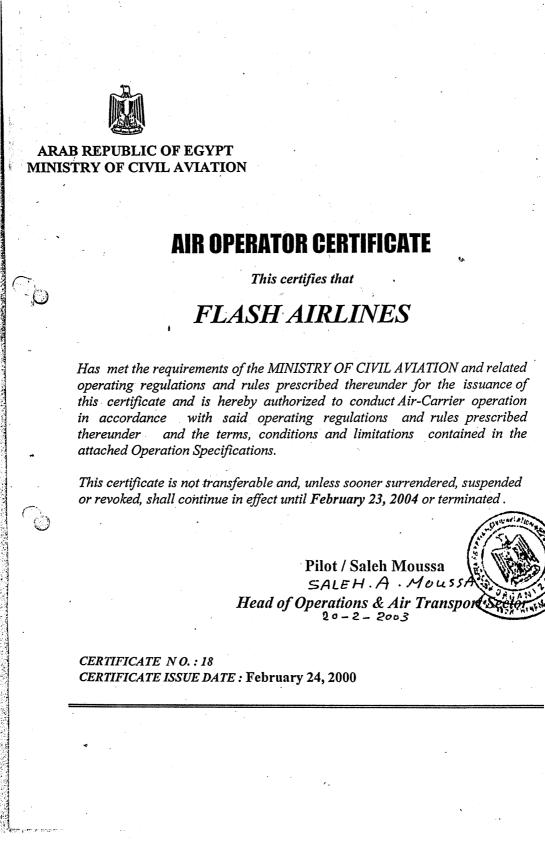
- 1.16.6.1 *Examination of the CVR recording for indications of A/P and heading select switch noises* (Could not be identified)
- 1.16.6.2 Examination of CVR at 2.58.15 (when the MSR crew says that they heard a message from Flash on 121.5).
 121.5 recording has been checked, no such message was recorded
- 1.16.7 FDR examination:

- 1.16.7.1 Spatial disorientation study of the accident flight based on the recorded FDR data TBC (CBS group)
- 1.16.8 *PCU inspection and teardown (EQA report):* (Refer to 1.16.1.7. Aileron system)

1.17 Organizational and Management Information

1.17.1. Flash Airlines

1.17.1.1. Flash Airlines Air Operator Certificate (AOC)



- 1.17.3. Relevant Flash Airlines procedures:
 - 1.17.3.1 Flash Airlines procedures regarding use of autopilot when recovering from unusual attitudes Refer to Flash Airline FOM (Ops Group)
 - 1.17.3.2 Flash Airlines procedures regarding Upset Recovery training

MCA requirements regarding Upset Recovery are not mandatory. Refer to Flash Airline FOM (Ops Group)

1.17.3.3 Flash Airlines procedures regarding "training about PNF assuming control when the PF is not responding to situations, callouts"



CHAPTER 4

FSH - 4.1.1

CREW HEALTH PRECAUTIONS

4. CREW HEALTH PRECAUTIONS

A crew member's

sickness/illness, his feeling unwell/indisposed or the impairment of his senses and reflexes by narcotics, drugs or pharmaceutical preparations/medicaments have

quite often contributed to incidents and accidents. Therefore, crew health is of the

highest importance and has a direct impact upon flight safety. This is reflected in very stringent requirements for regular medical examinations and medical certificates. It hardly needs to be mentioned that living health consciously is in the self-interest of every crew member.

Note: For incapacitation of crew members

crew member shall not perform duties on an aeroplane if he is in any doubt of being able to accomplish his assigned duties, or if he knows or suspects that he is suffering from fatigue, or feels unfit to the extent that the flight may be endangered.

4.1 Incapacitation of Crew Members

4.1.1 Definition

30 Jan 03

Incapacitation of a crew member is defined as any condition which affects the health of a crew member during the performance of duties associated with the duty/position assigned to him - which renders him incapable of performing the assigned duties. The definition includes either total or partial incapacitation which does not allow the fulfilment of duties in the "normal" way.

4.1.2 General

In-flight pilot incapacitation is a valid safety hazard and has already caused many accidents. Incapacities have occurred more frequently than other emergencies which are the subject of extensive training (such as engine failure, cabin fire etc). Aviation history and statistics indicate that incapacities may occur in all age groups and during all phases of the flight. There are many forms of incapacitation ranging from obvious sudden death to a lingering and difficult to detect partial loss of functions.

4.1.3 Types of Incapacitation

Obvious incapacitation; means total functional failure and loss of capabilities. This generally will be easily detectable and will be a prolonged condition. Among the possible causes are heart disorders, severe brain disorders, severe internal bleeding, etc. Subtle incapacitation; this may be considered a more significant operational hazard, because it is difficult to detect and the effects can range from partial loss of functions to a complete unconsciousness. Possible causes might be minor brain seizures, hypoglycemia (low blood sugar), other various medical

Flight Operations Manual



CHAPTER 4

CREW HEALTH PRECAUTIONS

disorders or preoccupation with personal problems. Since the crew member concerned may not be aware of, or capable of rationally evaluating his situation, this type of incapacitation is more dangerous!

4.1.4 Causes and Effects

As explained before, incapacitation may range from minor cases of <u>physiological up-</u> <u>sets</u> associated with intercurrent <u>mild disease or mental stress</u> which may result in <u>reduced</u> <u>levels of judgement or physical</u> <u>coordination</u> up to a complete <u>collapse</u>.

Among the causes for a mild incapacitation one may list: Body pains such as toothache, headache, gastroentehtis, the delayed effects of alcohol, drugs or medication, common disorders such as a cold, etc. Heart troubles, an acute infection thrombosis, epilepsy, hypoglycemia (extremely low level sugar) and others belong to the more serious causes of a sudden collapse. At least one incident is known, where a crew member had a heart attack right after his aviation medical examination, so a passed medical exam is not a guarantee!

It is obvious that living more health consciously may reduce the number of occurrences of also the avoidance of stress in your business and private life. Chapter **4.1** covers the subject of health precautions.

4.1.5 Recognition of an Incapacity

An early recognition of a incapacity is of outmost importance. A silent collapse will hardly be detected during normal activities (for instance during the cruise phase of a flight), as communications may sometimes be reduced to a minimum. This requires that all crew members monitor each other very closely.

"Closely" means, observing the other crew members for any "abnormal" reaction/action or behavior. One good method is to use the so called "TWO COMMUNICATION RULE". This simply means, that one crew member's comment must be answered by the other crew member(s).

If - for instance - the PNF reports the aeroplane being left of course, it is essential, that the PF not only corrects this problem but also <u>confirms this</u> <u>verbally</u>. If a crew member doesn't answer any question or checklist item

in the normal way, there is reason to believe that there might be the beginning of a subtle incapacitation.

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crew member incapacitation. This includes avoidance of drugs, moderate consumption of alcohol, adequate rest time -and its proper use for recreation adequate sleep and nutrition but

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of the Two Communication Rule:

1. the PNF, for example, notices the airplane is left of course,

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 any manner (verbally or by correcting the flight path), 4. the PNF repeats the abnormal condition to the PF (the second communication), 5. the PF again fails to respond, 6. after the PF fails to respond to the second communication, the PNF should assume the PF is incapacitated and should take action as described in Section 4.1.6 At the worst he may simply have fallen asleep. Other symptoms of the beginning of an incapacitation are: incoherent speech; strange behaviour; irregular breathing; pale fixed facial expression; jerky motions that are either delayed or too rapid. 	declare an urgency or emergency -whichever is applicable -, have an incapacitated cockpit crew member removed from his seat. In any case his seat should be moved fully back to prevent obstruction of flight controls, switches, levers, etc. The help of other crew members or passengers might be required, if necessary, reset COM and NAV to your side <u>econd Step</u> take care of the incapacitated crew member by trying to provide first aid (ask if doctors or other medical persons are aboard),
 the abnormal condition (the first communication), but the PF does not respond in any manner (verbally or by correcting the flight path), the PNF repeats the abnormal condition to the PF (the second communication), the PF again fails to respond to the second communication, the PNF should assume the PF is incapacitated and should take action as described in Section 4.1.6 At the worst he may simply have fallen asleep. Other symptoms of the beginning of an incapacitation are: incoherent speech; strange behaviour; irregular breathing; pale fixed facial expression; jerky motions that are either delayed or too rapid. 	emergency -whichever is applicable -, have an incapacitated cockpit crew member removed from his seat. In any case his seat should be moved fully back to prevent obstruction of flight controls, switches, levers, etc. The help of other crew members or passengers might be required, if necessary, reset COM and NAV to your side <u>econd Step</u> take care of the incapacitated crew member by trying to provide first aid (ask if doctors or other medical persons are
 correcting the flight path), 4. the PNF repeats the abnormal condition to the PF (the second communication), 5. the PF again fails to respond, 6. after the PF fails to respond to the second communication, the PNF should assume the PF is incapacitated and should take action as described in Section 4.1.6 At the worst he may simply have fallen asleep. Other symptoms of the beginning of an incapacitation are: incoherent speech; strange behaviour; irregular breathing; pale fixed facial expression; jerky motions that are either delayed or too rapid. 	cockpit crew member removed from his seat. In any case his seat should be moved fully back to prevent obstruction of flight controls, switches, levers, etc. The help of other crew members or passengers might be required, if necessary, reset COM and NAV to your side <u>econd Step</u> take care of the incapacitated crew member by trying to provide first aid (ask if doctors or other medical persons are
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 strange behaviour; irregular breathing; pale fixed facial expression; jerky motions that are either delayed or too rapid. If any of these are present, incapacitation must be 	arrange a landing as soon as practicable after considering all pertinent factors,
 pale fixed facial expression; jerky motions that are either delayed or too rapid. If any of these are present, incapacitation must be 	arrange medical assistance after landing
delayed or too rapid. If any of these are present, incapacitation must be	giving as many details about the condition of the affected crew member as possible.
If any of these are present, incapacitation must be	hird Step
incapacitation must be	prepare for landing (cockpit
	and cabin), but do not press
suspected and action taken to	for a hasty approach
check the state of the crew member.	perform approach checklist earlier than normal (request assistance from other crew
4.1.6 Actions to be taken when an incapacity is	members or "capable" persons),
recognised.	request radar vectoring and
First Step	make an extended approach where possible - to reduce
take over control of the	workload,
aeroplane by announcing "I have control",	for landing do not change seats - fly the aeroplane from
 engage autopilot, 	

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CREW HEALTH PRECAUTIONS

that position you initially were assigned to.

organise work after landing; this shall include

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- depending on the situation, a change of seats for taxiing in, but only after the aeroplane has come to a complete stop;
- having the incapacitated crew member offloaded and to the ambulance as quickly as possible;
- arrangements for the parking of the aeroplane.

NOTE:

- The company operations department must be kept informed at all times regarding the above circumstances for immediate relay to the Manager Flight Operations.
- In case of incapacitation of the system panel operator, pilots shall refer to procedures as published in the AOM.

4.1.7 Summary

The problems involved with incapacitation of crew members may be summarised as follows:

- If you do not feel well, say "NO" before the flight.
- Remember, that the best medical examination as well as a health conscious life still do not guarantee that an incapacitation during flight will not happen to you or to your other crew members.

- The "TWO COMMUNICATION RULE" must be used in order to
- have a chance of detecting any incapacitation in time. Take notice of any abnormal or unusual action of another crew member, as this might also be an indication of onset of incapacitation.
- Once an incapacitation is identified, remember the three basic steps:
- Step 1) Take over the aeroplane and bring it under YOUR control.
- <u>Step 2)</u> Take care of the incapacitated pilot (either have him removed from his seat or fixed so that he will not interfere the controls).
- Step 3) Prepare for landing.

Finally, it is emphasised that incapacitation requires special actions using the good judgement of the crew member left in command of the aeroplane.

4.2 ALCOHOLIC BERVERAGES

The use of intoxicating beverages by **FLASH AIR** flight crew members must of necessity be strictly regulated.

The following rules must be strictly observed by all flight crew members at all times:

 No alcoholic beverage shall be consumed on the same calendar day that a crew

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- 1.17.3.4 Flash Airlines training/operational information regarding intervention by the non-flying pilot when the flying pilot fails to respond to calls for correcting an unsafe situation. Refer to previous item
- 1.17.3.5 Regularity (or irregularity) rules regarding sleeping schedules on and off-duty. Strategies for obtaining adequate rest and managing crew on-duty alertness Refer to Flash Airline FOM (Ops Group)
- 1.17.3.6 General description about Flash Airline.
 (Date of foundation or transition, location of offices and bases, number of aircrafts operated, number of pilots and other personnel, annual flights, passengers carried, revenues, routes flown, and financial health)
 (All relevant information are already included in the Factual Report)
- 1.17.3.7 Labor management issues, growth trends, and main competitors. Closed
- 1.17.3.8 Egyptian requirements for the training of pilots at an airline such as Flash Airlines.



GENERAL. The following outline is intended to clarify the six categories of training used by operators and defined in Part 121, Subpart N. This clarification is intended to both define the type of training and describe for the Operator when each category of training is applicable.

APPLICABILITY OF TRAINING CATEGORIES. Usually, operators will need to conduct training in all six categories of training. Recurrent training applies to all operators. Initial equipment training, transition training, upgrade training, and requalification training apply in most situations. However, transition training is not applicable for an operator who operates only one aircraft type. Initial new hire training applies to operators who train and qualify newly hired personnel or personnel who have not been previously qualified as a crewmember by that operator.

CATEGORIES OF TRAINING. There are six basic categories of training applicable to Part 121 operators. The primary factors which determine the appropriate category of training are the student's previous experience with the operator and previous duty position. Each category of training consists of one or more curriculums, each one of which is specific to an aircraft type and a duty position (for example: A-320 SIC, and A-320 PIC). Training should be identified with and organized according to specific categories of training. When discussing training requirements, MoCA inspectors should be specific regarding the category of training discussed and use the same references as are stated in Part 121 Subpart N. Inspectors should encourage operators to use this nomenclature when developing new training curriculums or revising existing training curriculums. Use of this common nomenclature improves standardization and mutual understanding. The six categories of training are briefly discussed in the following subparagraphs:

A. Initial New Hire Training. This training category is for personnel who have not had previous experience with the operator (newly hired personnel). It also applies, however, to personnel employed by the operator who have not previously held a cockpit crewmember duty position with that operator. Initial new hire training includes basic indoctrination training and training for a specific duty position and aircraft type. Except for a basic indoctrination curriculum segment, the regulatory requirements for "initial new hire" and "initial equipment" training are the same. Since initial new hire training is usually the employee's first exposure to specific company methods, systems, and procedures, it must be the most comprehensive of the six categories of training. For this reason, initial new hire training is a distinct separate category of training.

B. Initial Equipment Training (PIC and SIC). This category of training is for personnel who have been previously trained and qualified for a duty position by the operator (not new hires) and who are being reassigned for any of the following reasons:

(a) Reassignment is to any duty position on an airplane of a different group (Group IIIP is reciprocating and turbopropeller powered and Group IIIJ is turbojet powered).

(b) Reassignment is to a different duty position on a different airplane type when the cockpit crewmember has not been previously trained and qualified by the operator for that duty position and airplane type.

C. Transition Training. This category of training is for an employee who has been previously trained and qualified for a specific duty position by the operator and who is being assigned to the same duty position on a different aircraft type and the different type aircraft must be in the same group. If it is not in the same group, initial equipment training is the applicable category of training.

D. Upgrade Training. This category of training is for an employee who has been previously trained and qualified as SIC or PIC (not eligible for requalification training) by the operator and is being assigned as PIC to the same aircraft type for which the employee was previously trained and qualified as SIC or PIC on the same type.

E. Recurrent Training. This category of training is for an employee who has been trained and qualified by the operator, who will continue to serve in the same duty position and aircraft type, and who must receive recurring training and/or checking within an appropriate eligibility period to maintain currency.

F. Requalification Training. This category of training is for an employee who has been trained and qualified by the operator, but has become unqualified to serve in a particular duty position and/or aircraft due to not having received recurrent training and/or a required flight or competency check within the appropriate eligibility period. Requalification training is also applicable in the following situations:

* PICs who are being reassigned as SICs on the same aircraft type when seat dependent training is required

* PICs and SICs who are being reassigned as FEs on the same aircraft type, provided they were previously qualified as FEs on that aircraft type

G. Summary of Categories of Training. The categories of training are summarized in general terms as follows: (a) All personnel not previously employed by the operator must complete initial new hire training.

Issue 2, Rev. 0

Dated July, 2002

Ministry of Civil Aviation

Training Standards Handbook



g Standards Handbook Pilots Experience and Training Standards وزارة الطيران المدني (b) All personnel must complete recurrent training for the duty position and aircraft type for which they are

(b) All personner must complete recurrent training for the day position and alrenal type for miner any are currently assigned within the appropriate eligibility period.
 (c) All personnel who have become unqualified for a duty position on an aircraft type with the operator must complete requalification training to reestablish qualification for that duty position and aircraft type.

(d) All personnel who are being assigned by the operator to a different duty position and/or aircraft type must complete either initial equipment, transition, upgrade, or requalification training depending on the aircraft type and duty position for which they were previously qualified.

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Dated July, 2002

Ministry of Civil Aviation





ECAR Part 121.400 Groups of aircraft	Demission of East	Unarada	Initial New Equipment		Initial New Hire	
(A) 121 - Air Taxi. Not exceed 5700 kg's	Requirements For	Upgrade	SIC PIC		SIC	PIC
<u>Group (I):</u> Single Engine Airplane	 Total Flight Experience. Flight Experience on Aeroplane Group. Flight Experience on Aeroplane Type. 	2150 300 100	500 300	2150 300	200	2150 300
<u>Group (II)</u> : Multi -Engines Airplane	: 1.Total Flight Experience. 2.Flight Experience on Aeroplane Group.		500 300	2500 500	200	2500 500
(B) 121 – Air Carriers & Air Taxi						
Group (IIIP) >5700 kg						
Reciprocating power	 Total Flight Experience. Flight Experience on Aeroplane Group. Flight Experience on Aeroplane Type. 	3000 750 300	500 300	3000 750	200	3000 750
Turbopropeller powered 3.Flight Experience on Aeroplane Group. 3.Flight Experience on Aeroplane Type.		3500 1500 500	700 500	3500 1500	200	3000 1500
<u>Group (IIIJ</u>) >5700 kg	1.Total Flight Experience. 2.Flight Experience on Aeroplane Group.	4000 2500	1200 1000	4000 2500	300	4000
Turbo- Jet Powered	3.Flight Experience on Aeroplane Type.	300				
(C) 121 – Air Carriers & Air Taxi Helicopter	 Total Flight Experience. Flight Experience on Aircraft Category. Flight Experience on Aircraft Type. 	1000 300 120	450 300	1000 300	150	100

Experience Hours Pre-Requisites for Different Training

Issue 2, Rev. 0

Dated July, 2002



Pilots Experience and Training Standards وزارة الطيران المدين

ECAR Part 121.400 Groups of aircraft	Upgrade SIC to PIC	Transition		Initial New Equipment		Initial New Hire	
(A) 121 - Air Taxi. Not exceed 5700 kg's	write	SIC	PIC	SIC	PIC	SIC	PIC
Group (I): Single Engine	2	4	4	4	4	8	8
Group (I) & (II): VFR only	4	4	4	4	4	4	4
Group (II) & (II): IFR/VFR	4	8	8	12	12	16	16
(B) 121 - Air Carriers & Air Taxi		1		12	12	10	10
Group (IIIP) : Exceeds 5700 kg	1						
 Reciprocating power 	12	20	20	20	20	24	24
 Turbopropeller powered 	12	20	20	20	20	24	24
Group (IIIJ) : Turbo- Jet Powered	12	24	24	24	24	28	28
(C) 121 – Air Carriers & Air Taxi Helicopter							
• VFR only	4	4	4	4	4	4	4
• IFR/VFR	4	8	8	12	12	16	16

Two Pilots Flight Training Minimum Hours Required

Dated July, 2002



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One Pilot Flight Training Minimum Hours Required
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ECAR Part 121.400 Groups of aircraft	Upgrade SIC	Transition		Initial New Equipment		Initial New Hire	
(A) 121 - Air Taxi. Not exceed 5700 kg's	to PIC	SIC	PIC	SIC	PIC	SIC	PIC
Group (I): Single Engine	4	4	4	4	4	6	6
Group (I) & (II): VFR only	2	3	3	3	3	4	4
Group (II) & (II): IFR/VFR	4	6	6	6	6	8	8
(B) 121 – Air Carriers & Air Taxi							
Group (IIIP) : Exceeds 5700 kg	1						
Reciprocating power	6	12	12	14	14	14	14
 Turbopropeller powered 	6	12	12	15	15	15	15
Group (IIIJ) : Turbo- Jet Powered	6	12	12	16	20	16	20
(C) 121 – Air Carriers & Air Taxi Helicopter							
VFR only	2	3	3	3	3	4	4
IFR/VFR	4	6	6	8	8	10	10

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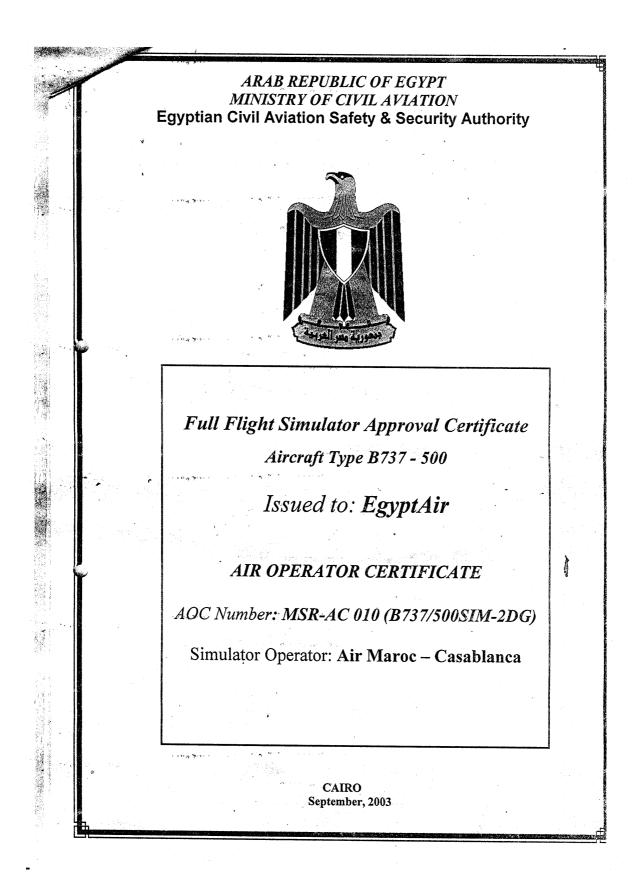
See also Pilots training documents included in items 1.5.1 and 1.5.2

- 1.17.3.9 The training that was actually provided to all Flash Airlines pilots Pilots training documents are included (refer to 1.5.1 and 1.5.2)
- 1.17.3.10 Flash Airlines procedures regarding pilots training and checking on operation of the auto flight system. . No specific form is available (refer to 1.5.1 and 1.5.2)
- 1.17.3.11 Flash Airlines program for training and checking pilots in the field of CRM and human factors (as contained in the company training manual)
 No mandatory training was required by ECAR at the time of the accident. However, CRM course is outlined in Flash Airline Training Manual 4.10
- 1.17.3.12 Flash Airlines pilots procedures for training and checking pilots on spatial disorientation countermeasures and upset recovery Spatial Disorientation training is not a requirement by Civil Aviation Authorities. However, some literature about this subject is included in Flash Airline Training Manual.
- 1.17.3.13 Flash Airlines policies regarding use of CRM. Refer to 1.17.3.11.
- 1.17.3.14 Flash Airlines policies relating to assertiveness and company guidelines as to when a first officer should take control of an aircraft from a captain. Refer to 1.17.3.3.
- 1.17.3.15 Flash Air general company policies related to crew communication, assertiveness, and other CRM-related behaviors Refer to 1.17.3.3.
- 1.17.3.16 Flash Airlines policies regarding use of the auto flight system (To be referred to the OPS group)
- 1.17.3.17 Regulations governing operators (like Flash Airlines) regarding Oversight audits by ECAA.
 ECAA regulations require every operator to undergo an oversight audit once every 12 month
- 1.17.3.18 Details about the ECAA oversight audit on Flash Airlines Is already included in the Factual Report
- 1.17.3.19 Outcomes of Oversight audits (previous violations, fines, or bans levied by ECAA)Is already included in the Factual Report

- 1.17.3.20 Previous violations, fines, or bans levied foreign aviation regulatory agencies. None identified
- 1.17.3.21 Selected additional information regarding Flash Airlines Organization including:
 - Organization and responsibilities Chapter 1 FSH 1.5.1/ 1.5.2
 - Organization and responsibilities Chapter 1 FSH 1.8.7
 - Qualification requirements Chapter 3 FSH 3.3.1/ 3.3.2
 - Crew Health Precautions Chapter 4 FSH-4.1.1- 4.1.4
 - Operating Procedures Chapter 6 FSH 6.3.44/ 6.3.45/ 6.3.46
 - Training details Flash Training Manual Chapt 05 Page 7

All pertinent information are included in the Factual Report

1.17.3.22 Airline Simulator program contract with RAM, ECAA letter of approval



جمهورية مصر العربية Arab Republic of Egypt Ministry of Civil Aviation Egyptian Civil Aviation Safety وزارة الطيران المدنى & Security Authority سلطة الطيران المدنى المصري . Qur Ref. MSR - AC010 - B737-500 FLT SIM-2/D Date: 24, September 2003 The General Manager Flight Training (GMFT) Flight Operations, EgyptAir, Cairo International Airport, Cairo, Egypt. + Stand Branch and a state of the second To: GMFT, EgyptAir APPROVAL TO USE THE FLIGHT SIMULATOR SPECIFIED IN THE ENCLOSED DOCUMENTATION Please find enclosed the required Approval Certificate and Licensing Considerations for the use of the requested Flight Simulator. A DATA TO THE A CONTRACT OF A DATA THE and the second second Yours sincerely, Signature: SALEH.A. MoussA Issued at: Cairo, Egypt Head of, Egyptian Civil Aviation Safety & Security Authority Date: 24, September 2003 and the second Enclosure. 1. B737-500.FLT. SIM Approval to EgyptAir. 2. Approval Certificate to Air Maroc, Casablanca 3. Licensing Considerations 4. Terms of Approval CAA -Flt. Sim September 2003 Pages 1/5



& Security Authority



جمهورية مصر العربية وزارة الطيران المدنى سلطة الطيران المدنى المصري

CERTIFICATE OF APPROVAL FLIGHT SIMULATOR

Number: MSR-AC010-B737-500 FLT. SIM-2D

This Certificate is issued to:

EgyptAir

Whose Business Address is:

Cairo International Airport Cairo, Egypt.

On behalf of the Egyptian Civil Aviation, It is hereby certified that the Flight Simulator for

B737-500

Located at Royal Air Maroc, Casablanca Airport Anfa

Has Satisfied the Qualification Requirements Prescribed In

Egyptian Civil Aviation Regulations (ECARs) Part 121 Section 121-407 Approval of Aircraft Simulators, and Appendices "E" and "F" Flight Training, Proficiency Check Requirements Respectively, and Appendix H to Part 121- Advanced Simulation. The Simulator must Maintain French DGCA, Approval and Qualification Level with JAR STD 1A as Reference

Subject to the conditions of the attached Specifications.

This Certificate is not transferable, and unless cancelled, revoked, suspended or varied shall continue in effect from September 24th 2003 until the end of September, 2004

Issued at: Cairo, Egypt Date: 24, September 2003

CAA -Flt. Sim*

Signature: SALEH.A. MoussA Head of, Egyptian Civil Aviation Safety & Security Authority

September 2003

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	Arab Republic of Egypt Ministry of Civil Aviation Egyptian Civil Aviation Safety & Security Authority سلطة الطيران المدنى المصري	
ſ	APPROVAL CERTIFICATE FLIGHT SIMULATOR	
	This Certificate is issued to:	
	Air Maroc	
	Whose Business Address is:	
	Air Maroc, Casablanca RAM	
	Upon finding that its organization complies in all respects with the requirements of the Egyptian Civil Aviation Regulations relating to the establishment of a Flight Simulator as described below for the approved Training and Testing for <i>EgyptAir-Cairo</i> . This certificate, unless cancelled suspended or revoked, shall continue in effect until <u>end of September 2004</u>	,
	Simulator Specifications:	
	Aeroplane/Type/Class Simulated B737/400-500 Category : Full Flight Simulator Data Package : Boeing STD	
م	Manufactured by : CAE Electronics LTD - 1993. Approval and Level : JAR - STD 1A Level "GD" Engines Type : CFM - 56 - C1	
	Engines Type : CFM - 56 - C1 Engine Instrumentation : Boeing Standard AFCS / EFIS : Honywell / Collins	
	Flight Management System : Smith Industires Visual System Manufacturer ;and : Vital VII , Day / Bright Day / Dusk / Night .	
į.	Type : 180 *40 Motion System/ and control loading : CAE/Hydraulic actuator with digital control electron.	ics
1.1°	Manufacturer : 6 Degrees of Freedom CAE series 500 6 DOF Other Equipment : TCAS-ATIS & RT Chatter-SATCOM-EGPWS-GPS	
	Simulated Computer Manufacturer; and Type (Host Computer)	
	Instructor's Station : Dual Indigo Touchscreen	
	<u>Note:</u> (1) A satisfactory assessment of one simulator session is required before use. <u>Note:</u> (2) A satisfactory assessment of flight Simulator Operators is required by ECASSA Flight Inspect.	tor
.,,	No. and Date of Issue:Signature: SALEH.A.MoussAMSR-B737/500 2D24, September 2003Egyptian Civil Aviation Safety & Security Authority	

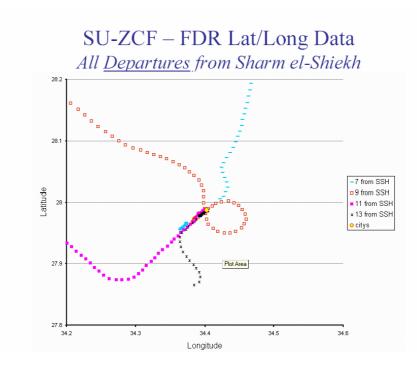
جمهورية مصر العربية Arab Republic of Egypt Ministry of Civil Aviation Egyptian Civil Aviation Safety وزارة الطيران المدنى & Security Authority سلطة الطيران المدني المصري TERMS OF APPROVAL Issued To: Royal Air Maroc - Casablanca Number: MSR-AOC-AC 010 -B737/500 FLT SIM - Issue 1 Date of Issue: 24/09/2003 and the second second The following terms of approval have been granted to Royal Air Maroc - Casablanca in respect of their organization at: Royal Air Maroc, Casablanca Airport Anfa 1. B737/500 Simulator to maintain French DGCA Approval. The Simulator maintains Qualification Level "D" with JAR -STD 1A as reference 2. Document until the end of September 2004, unless sooner refused, revoked, suspend or varied. Signature: SALEH.A. MoussA Issued at: Cairo, Egypt Head of, Egyptian Civil Aviation Safety & Security Authority Date: 24th of September 2003 1 and the second CAA-Flt.Sim September 2003 Pages 5/5 1 7 19 7 19 19 1

Fra REPUBLIQUE FRANÇAISE ministère de l'Equipement des Transports du Loge du Tourisme at do in Ne Certificat de Qualification STD (STD QUALIFICATION CERTIFICATE) Nr F-173Z Pour le compte de la Direction Générale de l'Aviation Civile, de l'Aviztio civile membre des Autorités Conjointes de l'Aviation (JAA), Il est déclaré par ce document que le simulateur de vol sarvice de la Formation aéronautique (on behalf of the French DGAC, a member of the Joint Aviation Authorities it is et du Contrôle hereby certified that the under mentioned flight simulator) technique B 737-500 Situé à (located at) MASABEANCATMAROCAL A satisfait les exigences de qualification du JAR-STD 1A et est qualifié pour le niveau DG (has satisfied the qualification requirements prescribed in JAR-STD 1A and is qualified for level DG) Ce certificat n'est pas transmissible et, à moins qu'il ne soit suspendu, retiré ou modifié, reste valable jusqu'au : (this certificate is not transferable and unless sooner suspended revoked or varied, shall continue in effect until) Ctober 2005 (31st October 2005) L'adjoint au Chef du Bureau des Equipages et des Procédures Paris, le 27 Octobre 2004 (Paris, on 27th October 2004) B. Ingénieur et de l'Exploita

- 1.17.3.23 Simulator used by Flash Airlines at RAM). Including
 - FCC options
 - Ground proximity
 - Bank angle options
 - Display type installed
 - FD type (split or integrated cue)

Pending Boeing response, see also 1.16.1.10.

- 1.17.3.24 Flash Airlines procedures regarding which pilot (PF or PNF) engages the autopilot, Boeing recommended practice No written procedure was found in Flash Airline FOM regarding this issue. Boeing procedures and common practices are for PF to connect the autopilot.
- 1.17.3.25 Additional information regarding dispatch from SSH A. All departures from SSH (accident aircraft)



-7 Departure from SSH-9 Departure from SSH-11 Departure from SSH-13 Departure from SSH

Same crew did flight no13 "Accident flight" and flight no 9 "SSH /TRN", following a comparison between the two flights.

FDR SSH Departure	Flight 13	Flight no.9
no.	Accident Flight	
Date	3 rd Jan, 04	2^{nd} Jan, 04
Take off Time	2.42 GMT	4.37 GMT
Runway	22R	04L
Captain	Khedr Aabdalla Saad	Khedr Aabdalla Saad
First officer	Amr Mahmoud Shafe	Amr Mahmoud Shafe
Autopilot in	A	А
Command		
Autopilot engaged at	3392 ft	2836 ft
Autopilot Mode	CMD /Heading Select	CMD /Heading Select

B- Extension of the outbound legs before beginning the turn

Interviewing Flash Airlines chief pilot:

Flash Airlines chief pilot stated that during the departure from SSH, Flash Airline pilots might extend the circuit as the situations need whether day or night departures (departure over water is mandatory)

Actual pattern flown depends on airplane performance (weight, OAT, etc). Most airplanes widen the pattern to gain additional altitude as a pilot technique. VOR crossing altitude restriction is shown on charts. This information should be added to Operations Group Notes.

- 1.19 New Investigation Techniques
 - 1.19.1 Spatial disorientation :
 - Definition
 - The way the SD works
 - Crew fatigue
 - Human related factors

Refer to (tests and researches), 1.16.4. Tests and researches conducted by MCA, Spatial Disorientation Studies

Additional work can be done through adding the report of the CBS group meeting)

Exhibit B FDR Group Factual Report

Attachment 4: Summaries of previous flight(s) by accident crew Refer to 1.17.3.25, all departures from SSH (accident aircraft)

Exhibit C CVR Group Factual Report Accident flight plan (copy of the flight plan referred to by ATC at 02:38:05 in the CVR transcript)

Aerodrome	KLINE3
FLIGHT PLAN PLAN DE VOL	مطار
PRIORITY ADDRESSEE(S) Priorite Destinutative(S)	برنامج رحلة
<=,FF→	
FILING TIME	
PILING TIME ORIGINATOR Heuro de depot Espedileur	
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Identification Freelse du(des) destinataire (s) et/ou de l'	RIOGINATOR espediteur
3.MESSAGE TYPE 7 AIRCRAFT IDENTIFICATIO	
FPL	Regies de vel TYPE OF FLICHT
9 NUMBER Nembre TYPE OF AIRCRAFT WAR	
	LE TURBULENCE CAT
13 DEPARTURE AERODROME TIME Aerodrome de depart Herro	
HESH GIZION	
15 CRUBIBING SPEED LEVEL ROUTE Viewe crosslere Nivean Route	
MOIGION FIZIGIO DOT SHI	1 AULINDIS DET EVG
16 DESTINA TION AERODROME TOTAL EET	
Acrodrome de destination HR. MIN	ALTN AERODROME 2 ND.ALTN AERODROME
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Competignettende divers	
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	a and a second sec
SUPPLEMENTARY INFORMATION (NO	T TO BE TRANSMITTED IN FFL MESSAGES
Renaelgnoments complementalres (a NE PAB TRA	T TO BE TRANSMITTED IN FFL MESSAGES) NISMETTRE DANS LES MESSAGES DE FLAN DE VOL DEPOSE) EMERGENCY RADIO
19 ENDURANCE PERSONS ON BOARD	Radio do socoare
HR. MIN Personaes a bord	
SURVIVAL EQUIPMENT/Equipment de surive	
FOLAR DESERT MARITIME JUNGLE Poisure Drawi Marijjane Gangle	LIGHT PLUORES UHF VHP
DINGHIES/Camets NUMBER CAPACTTY COVER CO	LOUR
AIRCRAFT COLOUR AND MARKINGS	
Content of margines de l'aeronof NWHITE	
REMARKS Remarques	
> N	K
PILOT-IN-COMMAND Filete commandent de bord	Signature of Pilot-in command or
-> CI KHEDR)
FILED BY/Depose por	rs Playely
SPACE RESERVED FOR ADDITIONAL REQIREMENT	

b. Spelling corrections

Two spelling corrections should be made:

- The phrase "02:34:25 Attendant: "on behalf of Captain Kheder" should read "02:34:25 Attendant: "on behalf of Captain Khedr"
- The phrase "advice ready for departure" should read " advise ready for departure "
- Exhibit D Airplane Performance Group Factual Report Radar Spec formatted.doc (to complement the item C.2 Radar Data, General specification)

ASR 12 Radar (Aircraft Surveillance Radar)

Secondary 250 nm Primary 60 nm 15 Revolution Per minutes approximately (Scan time = 4.13 sec)

<u>Field</u> 1	<u>Valid Field Variables</u> A-Z, 0-9	Data Field Description Aircraft flight identifier or callsign
2	#, *, +, or blank	<pre>Special processing indicator: # = track is inhibited from CA processing, either with another specified track or with all other tracks * = track is inhibited from MSAW processing + = track is inhibited from both CA and MSAW processing blank = track is subject to both CA and MSAW processing</pre>
3	H, M, or L	Aircraft wake indicator: H = heavy M = medium L = light
4	000-999 or ••••	Cleared level: NNN= assigned altitude in hundreds of feet •••• = altitude unavailable or less than sea level
5	T, ↑, ↓, or blank	Cleared level qualifier: T = temporary altitude ↑ = vertical movement of track - climbing ↓ = vertical movement of track - descending blank= permanent cleared level

6	000-999 or ••••	Reported altitude: NNN= reported altitude in hundreds of feet 999 = altitude greater than 99,900 feet •••• = altitude unavailable, altitude less than sea level or altitude has not been updated for approximately 15 seconds
7	a, C, E, e, N, n, or blank	 Altitude transition indicator: a = indicates altitude source is mode C, aircraft is below adapted transition level and altitude is in hundreds of feet above mean sea level C = indicates altitude source is mode C, aircraft is above adapted transition level and altitude is in flight levels E = indicates altitude source is manually entered, aircraft is above adapted transition levels e = indicates altitude source is manually entered, aircraft is below adapted transition levels e = indicates altitude source is manually entered, aircraft is below adapted transition level and altitude is in flight levels

<u>Field</u> 7 (Cont.)	<u>Valid Field Variables</u> a, C, E, e, N, n, or blank	 <u>Data Field Description</u> Altitude transition indicator: N = indicates mode C altitude has not been updated for approximately 7.5 seconds and is considered unreliable, aircraft is above adapted transition level and altitude is in flight levels n = indicates mode C altitude has not been updated for approximately 7.5 seconds and is considered unreliable, aircraft is below adapted transition level and altitude is in hundreds of feet above mean sea level blank= no data is available or altitude data has not been manually entered
8	0000-7777 (octal)	Reported code
9	0000-9999	Track ground speed in knots
10	0000-7777 (octal)	Assigned code
11 sim tracks	A-Z, 0-9	Aircraft type (field is blank for manually created
12	A-Z, 0-9	Destination aerodrome or last adapted point on flight plan route (XXXX)
13 A-2	Z, 0-9 Scratch pad note en	tered by controlling operator (XXXXXX)