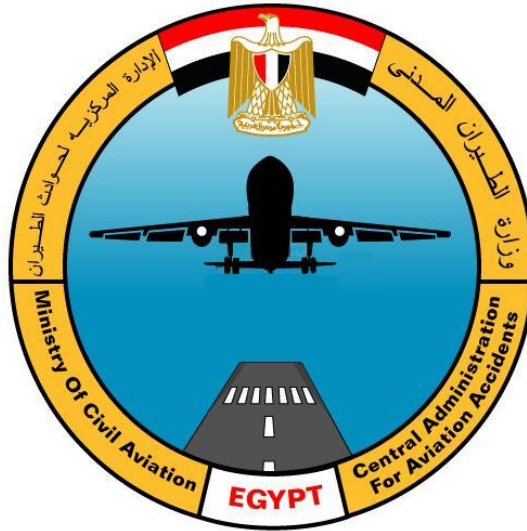


# EGYPTIAN MINISTRY OF CIVIL AVIATION



## FINAL REPORT OF THE ACCIDENT INVESTIGATION

Flash Airlines flight 604

January 3, 2004

Boeing 737-300 SU-ZCF

Red Sea off Sharm El-Sheikh, Egypt

**Occurrence Summary:**

On January 3, 2004, about 02:45:06 UTC, 04:45:06 Local time, Flash Airlines flight FSH604, a Boeing 737-300, Egyptian registration SU-ZCF, crashed into the Red Sea shortly after takeoff from Sharm el-Sheikh International Airport (SSH) in South Sinai, Egypt. The flight was a passenger charter flight to Charles de Gaulle Airport (CDG), France with a stopover in Cairo international Airport (CAI) for refueling. Flight 604 departed from Sharm el-Sheikh airport with 2 pilots (Captain and First Officer), 1 observer, 4 cabin crew, 6 off-duty crew members and 135 passengers on board. The airplane was destroyed due to impact forces with the Red Sea with no survivals.

The airplane had departed from Sharm el-Sheikh runway 22R and was air born at 02:42:33 UTC, approximately 2½ minutes prior to the crash, and had been cleared for a climbing left turn intercept the 306 radial from the Sharm el-Sheikh VOR station located just north of runway 22R. This climbing turn allows departing flights to gain sufficient altitude before proceeding over higher terrain located along the flight path to Cairo. Flight 604 was operating in Egyptian airspace as a charter flight operating under the provisions of Egyptian Civil Aviation Regulations Part 121

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- B.E.A. Comments on the Draft Final Report and MCA response
- Flash Airline Comments on the Draft Final Report and MCA response
- ECAA Comments on the Draft Final Report and MCA response

## 1. Factual Information

### 1.1. History of Flight

#### Summary

On January 3, 2004, about 02:45:06 UTC, 04:45:06 Local time, Flash Airlines flight FSH604, a Boeing 737-300, Egyptian registration SU-ZCF, crashed into the Red Sea shortly after takeoff from Sharm el-Sheikh International Airport (SSH) in South Sinai, Egypt. The flight was a passenger charter flight to Charles de Gaulle Airport (CDG), France with a stopover in Cairo international Airport (CAI) for refueling. Flight 604 departed from Sharm el-Sheikh airport with 2 pilots (Captain and First Officer), 1 observer, 4 cabin crew, 6 off-duty crew members and 135 passengers on board. The airplane was destroyed due to impact forces with the Red Sea with no survivals.

The airplane had departed from Sharm el-Sheikh runway 22R and was air born at 02:42:33 UTC, approximately 2½ minutes prior to the crash, and had been cleared for a climbing left turn intercept the 306 radial from the Sharm el-Sheikh VOR station located just north of runway 22R. This climbing turn allows departing flights to gain sufficient altitude before proceeding over higher terrain located along the flight path to Cairo. Flight 604 was operating in Egyptian airspace as a charter flight operating under the provisions of Egyptian Civil Aviation Regulations Part 121

## History of Flight

In the following history, comments originally in Arabic are translated in to English and appear in *italics*. A complete transcription of the CVR is contained in Exhibit C, CVR Group Factual Report

- Flash Airlines flight 604 Boeing 737-300 scheduling to depart Sharm El Sheikh at 0230 GMT 0430 local time.
- From Cockpit Voice Recorder information the first officer and observer were in the Cockpit at 02:14:30 the Captain was in the cockpit at 02:18:14.
- Load information and flight information were exchanged between the Flight Deck and Cabin Attendants.
- At 02:18:58 before start check list was requested by the Captain and was read by the F/O and responded by Captain and F/O completed at 02:20:17.
- The Cleared to Start checklist was carried out at 02:32:19, the After Start checklist at 02:35:36, and the Taxi checklist at 02:39:55.
- The ATC clearance was delivered at 02:38:15 and read back by F/O as follows:
- ATC Flash 604 destination Cairo as filed climb initially flight level 140 1673 on the squawk.
- F/O Our clear to destination via flight plan route 140 initially 1673 on the squawk Flash 604 we have total pax135 *God willing*.
- 02 h 39 min 54 s, A/T engaged (through the whole flight),
- The Take Off checklist was completed at 02:40:05.
- 02 h 40 min 38 s, F/O : "Flash 604 ready for departure",
- 02 h 40 min 46 s, TWR : "Flash 604 surface wind 280/13 kts left turn to intercept radial 306 clear for take off 22R",
- 02 h 40 min 55 s, F/O : "Clear for take off runway 22R with left turn to establish 306 Sharm VOR, our Flash 604 clear for take off",
- 02 h 41 min 19 s, F/O : "Left turn to establish radial 306",
- 02 h 41 min 30 s, Captain : "Initially 140",
- 02 h 41 min 34 s, Captain : "Confirm initially 140",
- 02 h 41 min 35 s, F/O : "And Flash 604 confirm to the left to establish 306",
- 02 h 41 min 40 s, Captain : "Initial 140",
- 02 h 41 min 43 s, TWR : "Inch Allah",
- 02 h 41 min 44 s, F/O : "And initially 140",
- Take off was initiated at 02:41:59 with standard call outs.
- At time 02:42:02 TOGA mode engaged and then disengaged at 02:42:04.
- Aileron movements during T/O roll and lift off were consistent with crosswind.
- 02 h 42 min 10 s, F/O : "Take off power set speed building up 80 kts throttle hold",
- 02 h 42 min 26 s to 02 h 42 min 33 s, Take off phase, Co-pilot : "V1 rotate, positive rate",
- 02 h 42 min 36 s, Captain : "Gears up",
- 02 h 42 min 38 s, gears are up (FDR), CAS 169,5 kts
- 02 h 42 min 43 s, Captain : "400 heading select",
- 02 h 42 min 44 s, F/O : "400 heading select" (FDR heading select engaged),
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- At time 02:42:49 the F/O announced "Level Change, MCP speed, N1 armed Sir".
- At time 02:42:59 the F/O announced "one thousand". At the same time, ATC reported the departure time and confirmed left turn clearance. The clearance was acknowledged by the F/O. This was the last ATC transmission from the flight crew. The aircraft rolled to 20° left bank and began a climbing turn.

- 02 h 43 min 00 s, Captain : "N1 speed 210 flaps 1",
- 02 h 43 min 04 s, Captain : "Left turn",
- 02 h 43 min 05 s, TWR : "Flash 604 airborne time 44 when you ready to the left to intercept 306 radial report on course", (Aircraft at 1268 ft),
- 02 h 43 min 11 s, Captain : "Left turn", (1528 ft, beft)
- 02 h 43 min 12 s, F/O : "Roger when ready inch Allah",
- 02 h 43 min 18 s, F/O : "left turn to establish 306 Sharm VOR", (maximum recorded left roll is 21,8° within that phase at 02:43:21),
- The turn continued as the magnetic heading approached 140° (at an altitude of 3600 ft), at which point the bank angle decreased to approximately 5° left bank.
- At time 02:43:19, EgyptAir Flight (MSR 227), a flight from Hurgada inbound to Sharm el-Sheikh called ATC. Conversations between ATC and MSR 227 continue for approximately 60 seconds.
- 02 h 43 min 21 s, MCP selected speed recorded 219 kts,
- 02 h 43 min 23 s, Captain : "Flaps up",
- 02 h 43 min 33 s, Selected heading recorded 106,8°,
- 02 h 43 min 35 s, Co-pilot : "Flaps up no light", (2196 ft, CAS 209 kts, Hdg 168, Pitch 10.9°, Roll 20,74° left),
- At time 02:43:37, the Captain called for the After Takeoff checklist. There was not audible response from the F/O.
- 02 h 43 min 53 s, CAS 216,5 kts decreasing (reached a minimum value of 184.5 Kts at 2:44:23 and then started increasing),
- At time 02:43:55, the Captain called "Autopilot". There was no immediate response from any crew member. (3124 ft, CAS 216 kts, Hdg 142.7, Pitch 15.3°, Roll 7.7° left)
- At time 02:43:58, the Captain stated "Not yet". (3320 ft, CAS 213.5 kts, Hdg 141.3°, Pitch 16.3°, Roll 6.6° left)
- At time 02:43:59, the FDR recorded the autopilot was engaged, and that the roll mode transition to CWS-R mode. This transition would have resulted in loss of Heading Select Mode (3392 ft, CAS 212 kts, Hdg 140.6°, Pitch 17.5°, Roll 6.6° left)
- At time 02:44:00, the F/O stated "Autopilot in command sir". (3468 ft, CAS 209.5 kts, Hdg 140.2°, Pitch 18.4°, Roll 6.6° left)
- At time 02:44:01, the captain stated "EDEELO", (an Arabic exclamation expressing a sharp response of some kind). At the same time, the FDR records momentary aileron surfaces movements. The right aileron deflected to 7.2 degree TEU for one second
- At time 02:44:02, the CVR records the autopilot disconnect warning and the FDR recorded the autopilot disengaged. The aural warning lasted for 2.136 seconds. (3624 ft, CAS 207 kts, Hdg 139.9°, Pitch 19.3°, Roll 5.6° left)
- During this time, an increase in pitch and decay in airspeed were observed
- At time 02:44:05, the Captain requested heading select. (3880 ft, CAS 203 kts, Hdg 139.5°, Pitch 20.5°, Roll 0.0° left)
- At time 02:44:07, the F/O states "heading select" and the FDR records heading select mode engaging. This mode transition would have resulted in the reappearance of the flight director roll command bar. During this sequence, the aircraft' left-bank continued to decrease at a slow rate until the airplane was briefly wings level. (4056 ft, CAS 199 kts, Hdg 139.5°, Pitch 19.8°, Roll 0.35° right)
- Beginning at this time, the FDR records a series of aileron motions that command a right bank and subsequent right turn.
- At time 02:44:18, the captain states "See what the aircraft did". At this point the aircraft bank angle was approximately 12° to the right. (4824 ft, CAS 186.5 kts, Hdg 149.4°, Pitch 15.4°, Roll 12.6° right)



- 02 h 44 min 23 s, CAS 184,5 kts and will increase to the end of the flight,
- 02 h 44 min 25 s, last recorded speed selected 220 kts,
- At time 02:44:27, the F/O states "Turning right, sir". Three seconds later, the captain responds "*What*". At the same time, bank angle is 17° to the right and the FDR records the aileron motions to increase the right bank (5172 ft, CAS 186 kts, Hdg 160.6°, Pitch 13.3°, Roll 16.8° right)
- At time 02:44:31, the F/O states "*Aircraft is turning right*". One second later, the captain response "*Ah*"
- At time 02:44:35, the Captain states "Turning right", at this point, the bank angle was 23.6° to the right (5396 ft, CAS 192 kts, Hdg 174.7°, Pitch 11,7° Roll 23,5° right), last selected heading 84,9°)
- At time 02:44:37, the Captain states – "*how turning right*" (5436 ft, CAS 195 kts, Hdg 179.6, Pitch 10.7°, Roll 27.7°)
- At time 02:44:41, the Captain states "OK come out". (5468 ft, CAS 202.5 kts, Hdg 194.7°, Pitch 6.5°, Roll 41.8° right) At this point, the bank angle was slightly more than 40° right bank and the FDR records the ailerons returning to just beyond neutral, the high right roll rate stopped and a momentary left roll rate occurred resulting in a slight decrease in the right bank from 43.2° at 2:44:40 to 41.8° at 2:44:41 before additional aileron movements command an increase in the right bank.
- At time 02:44:41.5, the F/O states "Overbank. The bank angle at this time was just beyond 50° right bank. The airplane reaches its maximum altitude of just over 5460 feet.
- At time 02:44:41.7, the Captain states "Autopilot". He repeats the statement at 02:44:43.4.
- At time 02:44:44, the F/O states "Autopilot in command". No autopilot engagement was recorded on the FDR.(5432 ft, CAS 209.5 kts, Hdg 210.5°, Pitch 3.5°, Roll 53.0° right)
- At time 02:44:46, the Captain again states "Autopilot".
- At time 02:44:48, the F/O states "Overbank, Overbank, Overbank".(5276 ft, CAS 222 kts, Hdg 235.9°, Pitch 3.5° nose down, Roll 68.9° right).
- 02 h 44 min 51 s, Master caution recorded,
- At time 02:44:52.8, the F/O again states "Overbank". (At 02:44:53, 4628 ft, CAS 254 kts, Hdg 265°, Pitch 25.14° nose down, Roll 91.4° right)
- At time 02:44:53.4, the Captain responds "OK, come out".
- 02 h 44 min 54 s, aileron motion to the left during 9 s (4388 ft, CAS 264.5 kts, Hdg 270°, Pitch 29.7° nose down, Roll 95.2° right)
- At time 02:44:56, the F/O states "No autopilot commander".(3820 ft, CAS 289.5 kts, Hdg 277°, Pitch 37° nose down, Roll 103.0° right)
- At time 02:44:58, the captain states "Autopilot". At the same time, the FDR records a large aileron motion to the left and the airplane begins rolling back towards wings level.(3068 ft, CAS 317.5 kts, Hdg 281°, Pitch 43.2° nose down, Roll 111° right)
- At time 02:44:58.8, the observer states "Retard power, retard power, retard power"
- At time 02:45:01.5, the captain states "Retard power", and the FDR records both engine throttles being moved to idle.(Pitch 42.4° nose down, Roll 39.2° right)
- At time 02:45:02, the CVR records the sound of the overspeed warning.(1320 ft, CAS 382.5 kts, Hdg 306.9°, Pitch 40.6° nose down, Roll 30.2° right)
- Recovery from severe Right Bank and nose down pitch continued
- At time 02:45:04.3, the captain states "Come out". Bank angle was 15.6° right, pitch attitude was 30.5° nose down, altitude was 421 ft, and airspeed was 411.8 KIAS

- At time 02:45:05, the CVR records a sound similar to ground proximity warning (180 ft, CAS 416 kts, heading 315.7°, pitch 25.4° nose down, right roll 19.3°),
- A/C impacted the water at about 02:45:06 with last recorded data:
  - Bank Angle 19.3° to the right
  - Pitch Angle 25.4° Nose down
  - Vertical G. Load 3.96 (2.7)
  - Speed 416 Kts

**Correlated FDR- CVR Data:**

**Boeing 737-300**  
**SU - ZCF**  
**Flash Airline**

Captain ●  
 First Officer ●  
 Roll Angle ●  
 Right Aileron ●  
 Left Aileron ●

**Sharm El Sheikh**  
**Egypt**  
**January 3,2004**

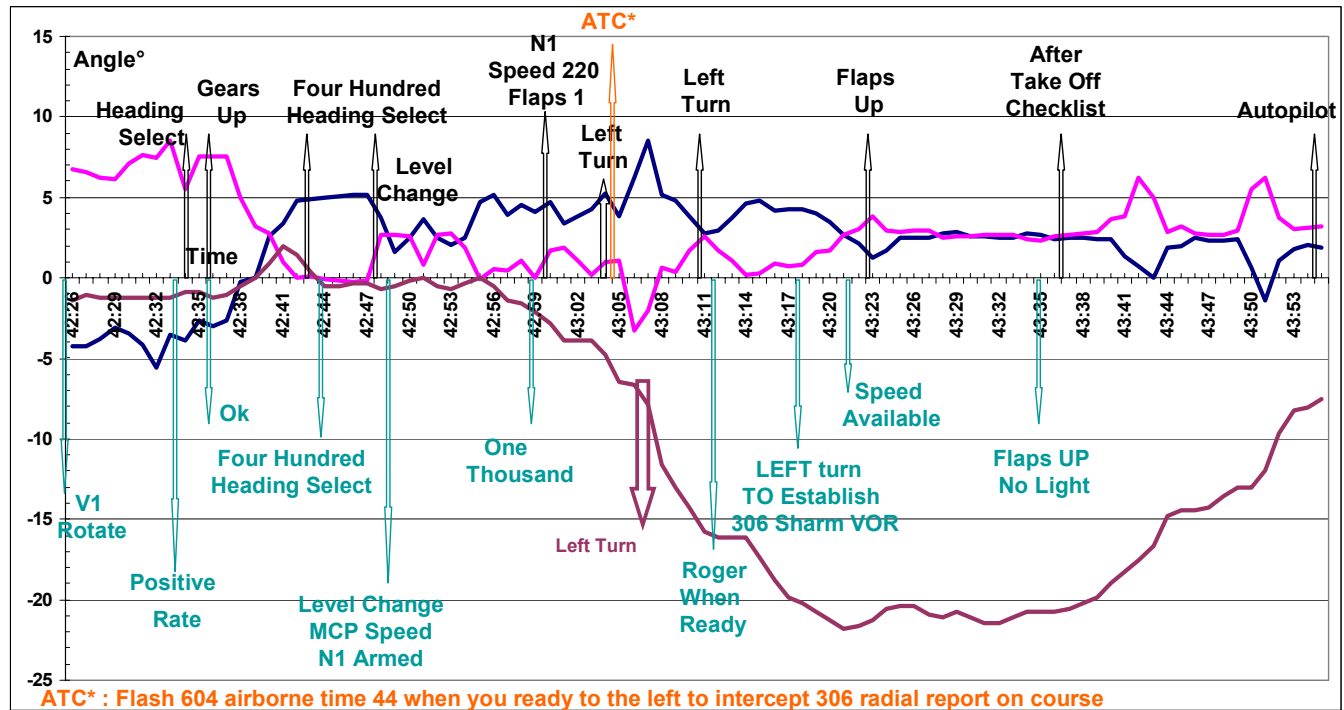


Figure 1.1-1 Correlated FDR- CVR Data

**Correlated FDR- CVR Data:**

**Boeing 737-300  
SU - ZCF  
Flash Airline**

Captain ●  
First Officer ●  
Roll Angle ●  
Right Aileron ●  
Left Aileron ●

**Sharm El Sheikh  
Egypt  
January 3, 2004**

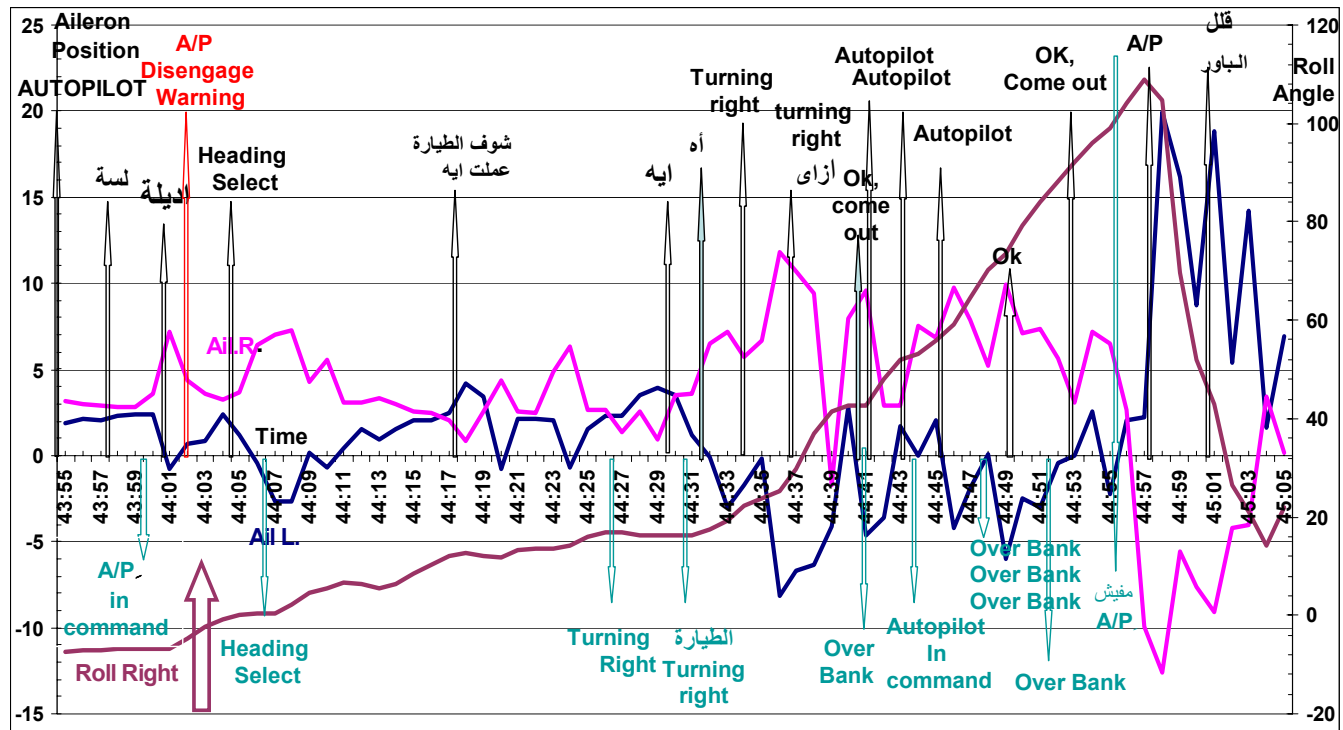


Figure 1.1-2 Correlated FDR- CVR Data

## 1.2. Injuries to Persons

There were no survivors.

Injuries	Flight Crew	Cabin Crew	Passengers	Off-Duty Crew	Total
Fatal	3	4	135	6	148
Serious	0	0	0	0	0
Minor	0	0	0	0	0
None	0	0	0	0	0
Total	3	4	135	6	148

Table 1: Injury chart.

### **1.3. Damage to Airplane**

The airplane was destroyed by impact with the water.

#### **1.4. Other Damage**

There was no other damage. Most of the wreckage remains on the floor of the Red Sea at a depth of approximately 1000 meters.

## 1.5. Personnel Information

Both the Captain and the First Officer were certified under Egyptian Civil Aviation Authority (ECAA).

### 1.5.1 The Captain

#### 1.5.1.1. Summary (personal and training information)

Date of birth:	February 26, 1950
Date of hire with Flash Airlines:	February 16, 2003
Airline Transport Pilot Egyptian Certificate Number 561(issued December 15, 1984)	
Airplane Multi-Engine Land	
Airplane Single Engine Land/Commercial Pilot	
Limitations:	None
Type Ratings:	ATR-42, B-737/300/400/500 (issued May 27, 2003), DHC-5 Buffalo, C-130 and Gomhoria
Medical:	First Class (issued November 19, 2003)
Limitations:	None
Initial Ground School Training:	Written Test April 9, 2003 Oral Test May 22, 2003
Initial Simulator Training	B-737-300/400/500: April 28- May 12, 2003
Initial Proficiency Check	B-737-300/400/500: May 12, 2003
Last Proficiency Check	B-737-300/400/500: May 12, 2003
Last Line Check:	July 23, 2003
Last Recurrent Training:	December 16, 2003
FLIGHT TIMES:	

Total flight time (hrs/min) <sup>1</sup> :	7,443:45
Total flight time on B-737:	474:15
Total flight time PIC:	5,473:35
Military Instructor Flight time:	1,967:55
Total flight time last 24 hours <sup>2</sup> :	7:15
Total flying time last 30 days:	83:51
Total flying Time 90 days:	244:43

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<sup>1</sup> Times are calculated for the captain up until December 31, 2003.

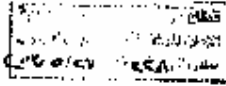
<sup>2</sup> Times do not include the accident flight.



1.5.1.2. Background information.

- i- Beginning of his flying career.  
Refer to captain CV, and his training records item 1.5.1.2 (vi)
  
- ii- All airlines worked for prior to Flash Air
  - The captain joined the A.R.E. Military Aviation College on September 1968, and was graduated on May 1970
  - He continued working as military pilot at A.R.E. Air Force since that date flying the L29, MIG17, MIG21, Buffalo (Dash 5), C130 types until he retired from the A.R.E. Air Force at the beginning of 2000
  - He joined Scorpio Aviation working as a civil pilot on ATR 42 from March, 2000 up to December, 2001.
  
  - He joined Flash Airline working as a civil pilot on B737-300 from February 2003 until *3 January 2004 (accident date)*

(All his flying hours were flown as PIC)
  
- iii- History of military and civilian employment as pilot  
The captain flew as a fighter pilot on L29, Mig17, Mig21 since his graduation until 1983. He then flew as a military transport pilot from that date on Buffalo and C130 until his retirement from the Air Force at the beginning of 2000.  
(Refer to previous item)
  
- iv- Retirement dates from A.R.E Air Force.  
Captain has retired from A.R.E. Air Force beginning of 2000
  
- v- History of position flown for specific aircraft, and dates of upgrades (i.e., copilot to captain)  
Refer to page 14 of the Factual Report  
(All his flying hours were flown as PIC)
  
- vi- "All" captain's training records (including his last recurrent training).



وزارة الطيران المدني  
سلطة الطيران المدني المصري  
الإدارة المركزية للتفتيش الجوي

السيد الملاح/ رئيس الإدارة المركزية للعمليات الجوية

تحية طيبة... وبعد

بالإحالة كغالب شركة فلاش الطيران بتاريخ ٢٠٠٣/٥/٢٢ ومرفقه أمر الترخيب  
رقم ٣ لسنة ٢٠٠٢ بشأن عقد ابرقة BASIC INDOCTRINATION بواقع (٢١ ساعة) في الفترة من  
٢٠٠٣/٥/٢٤ إلى ٢٠٠٣/٥/٢٦ لكفى أسلوهم بعد:-

- ١- الطيار/ خضر عبد الله سعد
  - ٢- الطيار/ عارف بالله محمد السيد
  - ٣- الطيار/ محمد جهاد شعراوي
  - ٤- منضيف جوي/ حامد محمد القورشاني
- يرجى التكرم بالإحالة بأنه تم متابعة الفرفة بواسطة الإدارة المركزية للتفتيش الجوي ولا يسع  
لدينا من اعضاء الفرفة لسماحتهم.

وتفضلوا بالقبول فائق الاحترام ،،

التوقيع /  
عبد الله محمد السيد  
رئيس الإدارة المركزية للتفتيش الجوي

مسيرة السيد كاتيار/ مدير عام عمليات شركة فلاش الطيران

التعليق : ٢١٧

رقم الطيران/ كفى : ٢٦٨١٢٤٩

Letter issued by ECAA approving Flash Airline Basic Indoctrination Course for 4 trainees including Captain/ Khedr Abdallah Lasting 21 hrs from 24 May 2003 to 26 May 2003

Curriculum Vitae:

(7)

C.V.

Personal information:

Name: Khedr abdalla saad said  
Nationality: Egyptian  
Data of Birth: February 26<sup>th</sup>, 1950.  
place of Birth: Cairo

Qualifications & Certificates:

BSc. In aviation: Air Force Academy  
AL.T by Egyptian Civil Aviation Organization  
R/T Communication License

Flight Courses:

<u>Ground Courses:</u>		<u>Flight Courses</u>	<u>Experience</u>
<u>Military</u>	<u>Civil</u>		
L-29	Gomhoria	Gomhoria	Pilot
Mig-17	Dash-5	L-29	Pilot
Mig-21	C-130	Mig-17	Pilot
	ATR-42	Mig-21	Instructor
		Dash-5	Captain
		C-130	Captain and Instructor
			to all international route
		ATR-42	Captain

Flying Hours:

Total Flying Hours:	6967.05
Total on jet A / C:	1009 hrs
Total Civil Time:	5958.05
Total Flying Hours as Instructor:	1967.54 hrs

All the documents are available upon request.

**Certificate, A.R.E. Air Force Head Quarter, Training Department**

16

A. R. E.  
AIR FORCE H. Q.  
TRAINING DEPARTMENT

**CERTIFICATE**

Case: 14-12-1994

Air training Department Certifies That: A/c / Rheda Alid Alloh Saad H. A. P. Saad  
in That Types of Aircraft: Comharic (L24) (Regis) (14) (21) (Boflo) (C130)

Single Engine Aircraft								Total hours of the single engine	Multi Engine Aircraft						Total hours of mult Engines .	Total Time	Instrument Flying	Liuk Simulate												
Day				Night					Day			Night																		
Eual		Solo		Eual		Solo			Dual	Solo	2 and pilot	Dual	Solo	2 and pilot																
HR.	Min	HR.	Min	HR.	Min	HR.	Min	HR.	Min	HR.	Min	HR.	Min	HR.	Min	HR.	Min	HR.	Min	HR.	Min	HR.	Min							
27	10	8	40	39	10	28	25	10	25	83	-	36	4	75	655	15	29	20	438	15	45	55	37	20	645	45	39	05	23	30

	HR.	Min
Total Flying Hours On Jet Aircraft .	1009	-
" " " " Prop Engine .	5425	45
Total Flying Hourse .	6434	45
" " " instruction .	2417	45

Chief of Registering Branch  
A/c. Rana Alid Eloud  
G.A.M.H.

Chief of Training Dept.  
Air Vice Marshal: Mohammed K.

Number of Training Flying Hours for Captain/ Khedr Abdallah at Scorpio Aviation  
(15 June 2000)

عدد ساعات الطيران التدريبي للطيار/

Date	FLT. .No	From	To	Block Time				No. of landings	Instructor
				Day		Night			
				H	M	H	M		
13 APR. 2000	335	AST	AST	03	00	-	-	8	GEO
3 APR 2000	336	AST	CAI	00	20	00	50	1	GEO
18 APR 2000	112	ASW	CAI	01	55	-	-	1	GEO
16 MAY 2000	333	CAI	AST	01	35	-	-	1	GEO
16 MAY 2000	335	AST	CAI	01	10	-	-	1	GEO
13 JUN 2000	112	ASW	CAI	02	05	-	-	1	GEO
28 APR 2000	131	CAI	ARG	01	15	-	-	1	POP
28 APR 2000	101	CAI	LXR	01	25	-	-	1	POP
15 JUN 2000	301	HRG	LXR	00	50	-	-	1	GEO
15 JUN 2000	301	LXR	LXR	00	30	-	-	2	GEO
				14	05	0	50		
TOTAL		14.55		14.55				18	GEO

شركة الطيران الدولية  
مسكوريو  
١٧ سابع

15. Juni. 2000  
CAPT GEOR GEORGE  
th

**Proficiency Checks at Scorpio Aviation:**  
17 June 2000

Egyptian Civil Aviation Authority  
Flight Safety Standards Sector  
"Operations Inspectorate"

الهيئة العامة للغواصة الجوية  
مركز التفتيش  
القاهرة

PROFICIENCY/QUALIFICATION CHECKLIST				
REF. NO. _____	DATE OF CHECK: <b>17 JUN. 2000</b>		TYPE OF CHECK: <b>PROFICIENCY</b>	
NAME OF PILOT: <b>KHEOR ABOALLA</b>	EMPLOYED BY: <b>SCORPIO AVIATION</b>		INSPECTOR OR CHECK AIRMAN NAME: <b>GEORGESCU</b>	
BASE: <b>CAIRO</b>	TYPE AIRCRAFT: <b>ATR-42</b>		FLIGHT TIME: <b>01</b> hrs <b>35</b> mins	
TYPE SIMULATION: _____	SIMULATOR TIME: _____ hrs _____ mins			
FLIGHT MANOEUVRES (S = Satisfactory, U = Unsatisfactory)				
PILOT	FLIGHT ENGINEER		S	U
	Air craft	Simulator		
<b>PRE-FLIGHT</b>				
Equipment examination (oral or written)	S			
pre-flight inspection	S			
Taxiing	S			
Powerplant checks	S			
<b>TAKE-OFFS</b>				
Normal	S			
Instrument	S			
Cross-wind	S			
With simulated powerplant failure	S			
Rejected take-off	S			
<b>INSTRUMENT PROCEDURES</b>				
Area departure	S			
Holding	S			
Area Arrival	S			
I.S. approaches	S			
Other instrument approaches	S			
Circling approaches	S			
Mixed approaches	S			
<b>IN-FLIGHT MANOEUVRES</b>				
Steep turns	S			
Approaches to stalls	S			
Specific flight characteristics	S			
Powerplant failure	S			
<b>LANDINGS</b>				
Normal	S			
From an I.S.	S			
Cross-wind	S			
With simulated powerplant (s) failure	S			
Rejected landing	S			
From circling approach	S			
Normal and abnormal procedures	S			
Emergency procedures	S			
Judgement	S			
REMARKS:				
TYPE RATING AS				
PILOT IN COMMAND				
ATR-42				
GEORGESCU				
17-JUN-2000				
INSPECTOR OR CHECK AIRMAN SIGNATURE				

ECAA - INSPECTION FORM (5/96)

ICAO - DOC 8357/C1

شركة الطيران الدولية  
مكوريو  
17 يونيو 2000

Approved  
ECAA Capt  
A del Adel  
inspector  
17/6/2000

17  
18  
19

8 December 2000

## SCORPIO AVIATION

FLIGHT OPERATION DEPARTMENT				PROFICIENCY CHECK/QUALIFICATION FORM					
AIRMAN <b>KIDR ABD</b>		A/C REGISTER MARK <b>SUBM</b>	SEAT POSITION CMI <input checked="" type="checkbox"/> CM2 <input type="checkbox"/>	SIMULAT		LOCAL FLIGHT		LINE FLIGHT	
CHECK PILOT <b>MAGDY Khaled ALLA</b>		SPECIFIC ITEMS: S-SATISFACTORY U-UNSATISFACTORY		S	U	S	U	S	U
DATE <b>8-12-2000</b>	AIRPORT <b>CAI</b>	TECHNICAL KNOWLEDGE							
		FLIGHT PREPARATION & FUEL PLANNING							
SIMULAT	SIMULATOR TIME.....	AIRCRAFT AND COCKPIT PREPARATION							
		NORMAL ENGINE START PROCEDURE							
LOCAL FLIGHT	AC <input type="checkbox"/> STICK TIME.....	ENGINE START MALFUNCTIONS							
	T/O.....LND.....GA.....	GROUND OPERATION AND TAXING							
	VIS.SIM. <input type="checkbox"/> SIM TIME.....	TAKE OFF PROCEDURES							
LINE FLIGHT	NUMBER OF LEGS..... AP.....	REJECTED TAKE OFF							
		ENGINE FAILURE AT V VI							
		ENGINE OUT APPROACH AND LANDING							
		ENGINE OUT GO AROUND							
<b>RECURRENT CHECK</b>		ABNORMAL CONFIGURATION APP. & LAN.							
1' CK SEMEST <input type="checkbox"/>	2' CK SEMEST <input type="checkbox"/>	ABNORMAL PROCEDURES							
LICENCE(IFR) <input type="checkbox"/>	NIL	CONDITIONAL PROCEDURES							
TYPE(ABILIT) <input type="checkbox"/>	NIL	EMERGENCY PROCEDURES							
THEOR.EXAMIN <input type="checkbox"/>	NIL	EMERGENCY EVALUATION							
EMERG.EQUIPM <input checked="" type="checkbox"/>	NIL	DEPARTURE AREA COORDINATION							
PROFICIENCY <input checked="" type="checkbox"/>	PROFICIENCY <input type="checkbox"/>	USE OF NAVIGATION SYSTEM							
CAT II QUALIF. <input type="checkbox"/>	CAT II QUALIF. <input type="checkbox"/>	USE OF RADIO-AIDS ROUTE DOCUM.							
<b>QUALIFICATION CHECK</b>		RADIO COMMUNICATIONS							
TYPE RATING <input type="checkbox"/>		IN FLIGHT NORMAL PROCEDURES							
LINE QUALIFICATION <input type="checkbox"/>		ENGINES AND FUEL MANAGEMENT							
CAT II (DH.....RVR.....)	<input type="checkbox"/>	DESCENT PLANNING							
IFR QUALIFICATION <input type="checkbox"/>		MINIMUM SAFE ALTITUDE AWARENESS							
<b>AIRMAN SELF-CERTIFICATION:</b>		TERMINAL AREA COORDINATION							
		INSTRUMENT APPROACH							
Nr. OF T/O and LDN last 3 months <input type="checkbox"/>		VISUAL APPROACH							
Nr. OF IFR APP. last 6 months <input type="checkbox"/>		LANDING TECHNIQUE							
Nr. OF Cat. II APP. last 6 months <input type="checkbox"/>		GO-AROUND TECHNIQUE							
Airman signature _____		TIME AND POSITIVENESS OF REACTION							
		CREW COORDINATION							
		SUPER VISION OF CABIN ACTIVITY							
		PUBLIC RELATIONS AND INFORMATION							
		ECONOMIC AND COMMERCIAL ASPECTS							
<b>REMARKS</b>									
<b>Proficiency Check on ATR42.320</b>									
<b>Satisfactory</b>									
SATISFACTORY <input checked="" type="checkbox"/>		CHECK PILOT SIGNATURE: <b>MAGDY Khaled No.724</b>							
UNSATISFACTORY <input type="checkbox"/>		شركة الطيران الدولية مستورينو 11 شارع الامارات - القاهرة							

12/12
12/12

17 June 2001

**SCORPIO AVIATION**

**Flight Operation Department Proficiency Check/Qualification Form**

PROFICIENCY CHECKLIST				
REF. NO.....		DATE OF CHECK..... 17-06-01.....		
NAME OF PILOT/F.E. KHEDR SAID		TYPE OF CHECK..... PROFICIENCY CHECK.....		
EMPLOYED BY... S. Co. Pilot		INSPECTOR OR CHECK AIRMAN		
BASE... CAIRO		NAME..... ESSAM S. B. HEGEM		
TYPE AIRCRAFT... ATR 42		FLIGHT TIME... 1.40 hrs... 1 mins... 40		
TYPE SIMULATION.....		Simulator time..... hrs..... mins.....		
FLIGHT MANOEUVERS (S= Satisfactory , U = Unsatisfactory )				
PILOT		FLIGHT ENGINEER		
	Air craft	Simu lator	ITEM	S/U
PRELIGHT				
Equipment examination (oral or written)	S		Equipment exam ( oral or written )	
Preflight inspection	S		Preflight check of aircraft	
Taxiing	S		Computation of fuel load and fuel loading procedure	
Powerplant Checks	S		Completion of company approved forms	
TAKE-OFFS				
Normal	S		Starting taxi and run up	
Instrument	S		Powerplant control	
Cross - Wind	S		Cruise control and computations	
With simulated powerplant failure	S		Aircraft/powerplant operation analysis	
Rejected take-off	S		Fuel system management	
INSTRUMENT PROCEDURES				
Area departure	S		Aircondition and pressurization control	
Holding	S		Electrical system operation	
Area Arrival	S		Powerplant fire control	
H.S approaches	S		Emergency gear and flap extension	
Other instrument approaches	S		Heater fire and cargo compartment fire	
Circling approaches	S		Smoke evacuation	
Missed approaches	S		Emergency depressurization	
IN- FLIGHT MANOEUVERS				
Steep turns	S		Fuel dumping procedure	
Approaches to stalls	S		Powerplant shutdown and restart	
Specific flight characteristics	S		De- icing and anti- icing	
Powerplant failure	S		Location and use of emergency equipment	
LANDINGS				
Normal	S		Emergencies - Hydraulic pressurization, etc.	
From an H.S	S		Crew co- ordination and monitoring	
Cross - wind	S		Remarks :	
With simulated powerplant ( s ) failure	S			
Rejected landing	S			
From circling approach	S			
Normal and abnormal procedures	S			
Emergency procedures	S			
Judgement	S			

PROFICIENCY CHECK

Satisfactory ✓

ESSAM S. B. HEGEM  
شركة الطيران الدولية  
مستورب  
17 يونيو 2001

17  
2001  
Lic 655

بالتفصيل، كالتالي  
أ. ج. ج. ج.  
أ. ج. ج.  
ع. ج. ج.

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12 December 2001

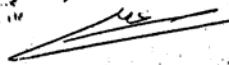
**SCORPIO AVIATION**

**Flight Operation Department Proficiency Check/Qualification Form**

**PROFICIENCY CHECKLIST**

REF. NO. ....  
 NAME OF PILOT FE. KHEDR. ABDALL DATE OF CHECK 12-12-2001  
 EMPLOYED BY SCORPIO AVIATION TYPE OF CHECK PROFICIENCY CHECK  
 BASE ..... INSPECTOR OR CHECK AIRMAN  
 TYPE AIRCRAFT ATR 42 NAME MAGDY KHALED  
 TYPE SIMULATION ..... FLIGHT TIME 1.20 hrs. 1 mins. 20  
 Simulator time ..... hrs. .... mins.

FLIGHT MANOEUVERS (S= Satisfactory , U = Unsatisfactory)			
PILOT		FLIGHT ENGINEER	
	Air craft	Simu lator	ITEM
			S U
FLIGHT			Equipment exam ( oral or written )
Equipment examination(oral or written)	S		Preflight check of aircraft
Flight Inspection	S		Computation of fuel load and fuel loading procedure
Weighting	S		Completion of company approved forms
Powerplant Checks	S		Starting taxi and run up
TAKE-OFFS	S		Powerplant control
Normal	S		Cruise control and computations
Instrument	S		Aircrft/powerplant operation analysis
Cross - Wind	S		Fuel system management
With simulated powerplant failure	S		Aircondition and pressurization control
Rejected take-off	S		Electrical system operation
<b>INSTRUMENT PROCEDURES</b>			
Area departure	S		Powerplant fire control
Holding	S		Emergency gear and flap extension
Area Arrival	S		Heater fire and cargo compartment fire
H.S approaches	S		Smoke evacuation
Other instrument approaches	S		Emergency depressurization
Circling approaches	S		Fuel dumping procedure
Missed approaches	S		Powerplant shutdown and restart
<b>IN- FLIGHT MANOEUVERS</b>			
Steep turns	S		De- icing and anti- icing
Approaches to stalls	S		Location and use of emergency equipment
Specific flight characteristics	S		Emergencies - Hydraulic pressurization, etc.
Powerplant failure	S		Crew co- ordination and monitoring
Remarks :			
<b>LANDINGS</b>			
Normal	S		<b>PROFICIENCY CHECK</b> <b>ON ATR 42-390</b> <b>SATISFACTORY</b>
From an H.S	S		
Cross - wind	S		
With simulated powerplant ( s) failure	S		
Rejected landing	S		
From circling approach	S		
Normal and abnormal procedures	S		
Emergency procedures	S		
Judgement	S		

MAGDY KHALED  
 No 724  


Fixed Base Simulator Training:



14

TRAINING RECORD FBS

LESSON 1

NAME: <u>KHEDR ABDALAA</u>		CREW POSITION: <u>CAPTAIN</u>	
		A/C TYPE: <u>B 737-800/400/500</u>	
<b>BRIFING</b>		<b>Cruise</b>	
NORMAL PROCEDURES	S/VS	Normal procedures	S/VS
<b>PREFLIGHT</b>		<b>Descent &amp; Approach</b>	
Practice AFDS preflight	S/VS	Normal procedures	S/VS
Practice FMC/CDU preflight	S/VS		
Practice IRS Full alignment	S/VS		
<b>ENGINE START</b>		<b>Landing</b>	
Normal procedures	S/VS	Normal procedures	S/VS
<b>Taxi-out &amp; takeoff</b>		<b>Taxi-in &amp; park</b>	
Normal procedures	S/VS	Normal procedures	S/VS
<b>Climb</b>			
Normal procedures	S/VS		
REMARKS:			
<u>CAPTAIN KHEDR NEEDS TO IMPROVE</u>			
<u>COCKPIT PREPERATION</u>			
<u>(NORMAL PROCEDURES)</u>			
INSTRUCTOR NAME:		INSTRUCTOR SIGNATURE:	
<u>IHAB EL SONBATY</u>			
DATE:		TRAINING SIGNATURE:	
<u>28-04-03</u>			
		OPERATIONS	

TRAINING RECORD FBS

LESSON 2

NAME: <u>KHEDR ABDALAA</u>		CREW POSITION: <u>CAPTAIN</u>	
		A/C TYPE: <u>B 737-300/400/500</u>	
<b>BRIFING</b>		<b>Cruse</b>	
NORMAL PROCEDURES	S / <del>US</del>	Normal procedures	S / <del>US</del>
Supplementary Normal procedures		<b>Descent &amp; Approach</b>	
FM alerting & advisory messages		Normal procedures	S / <del>US</del>
MCP controls and FMA			
FMC LNAV operation			
<b>PREFLIGHT</b>		<b>Landing</b>	
Normal procedures	S / <del>US</del>	Normal procedures	S / <del>US</del>
Supplementary Normal procedures	S / <del>US</del>		
		<b>Taxi - in &amp; park</b>	
		Normal procedures	S / <del>US</del>
<b>ENG! START</b>			
Normal procedures	S / <del>US</del>		
<b>Taxi-out &amp; takeoff</b>			
Normal procedures	S / <del>US</del>		
<b>Climb</b>			
Normal procedures	S / <del>US</del>		
Demonstration flight	S / <del>US</del>		
REMARKS:			
<u>PROGRESSING BUT STILL NEEDS TO</u>			
<u>IMPROVE COCKPIT PREPERATION</u>			
INSTRUCTOR NAME:		INSTRUCTOR SIGNATURE:	
<u>IHAB EL SONBATY</u>		<u>[Signature]</u>	
DATE:		TRENIV SIGNATURE:	
<u>29-04-03</u>		<u>[Signature]</u>	



TRAINING RECORD FBS

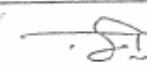
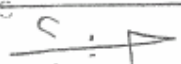
LESSON 3

NAME: <u>KHEDR ABDALAA</u>		CREW POSITION: <u>CAPTAIN</u>	
		A/C TYPE: <u>B 737-300/400/500</u>	
<b>FLIGHT</b>		<b>Cruise</b>	
IC LNAV OPERATION	S / <input checked="" type="checkbox"/> S	Normal procedures	S / <input checked="" type="checkbox"/> S
<b>FLIGHT</b>		<b>Descent &amp; Approach</b>	
Normal procedures	S / <input checked="" type="checkbox"/> S	Normal procedures	S / <input checked="" type="checkbox"/> S
Supplementary Normal procedures	S / <input checked="" type="checkbox"/> S	MISSED APPROACH	S / <input checked="" type="checkbox"/> S
		<b>Landing</b>	
<b>GIN START</b>		Normal procedures	S / <input checked="" type="checkbox"/> S
Normal procedures	S / <input checked="" type="checkbox"/> S	MISSED APPROACH	S / <input checked="" type="checkbox"/> S
<b>xi-ont &amp; takeoff</b>		<b>Taxi - in &amp; park</b>	
IC LNAV & VNAV OPERATION	S / <input checked="" type="checkbox"/> S	Normal procedures	S / <input checked="" type="checkbox"/> S
<b>imb</b>		Supplementary normal procedures	S / <input checked="" type="checkbox"/> S
Normal procedures	S / <input checked="" type="checkbox"/> S		
REMARKS: <u>PROGRESSING</u>			
INSTRUCTOR NAME: <u>ZHAR EL SONBATY</u>		INSTRUCTOR SIGNATURE:	
DATE: <u>30-04-03</u>		TRAINING SIGNATURE:	

OPERATIONS  
TRAINING MANUAL

TRAINING RECORD FBS

LESSON 4

NAME: <u>KHEOR ABDALAA</u>		CREW POSITION: <u>CAPTAIN</u>	
		A/C TYPE: <u>B 737-300/400/500</u>	
<b><u>RIFING</u></b>		<b><u>Cruise</u></b>	
Normal procedures	S/US	Normal procedures	S/US
Non-normal procedures	S/US	Fix position	S/US
Review system & FMC/CDU	S/US	Fix position & abeam	S/US
<b><u>REFLIGHT</u></b>		<b><u>Descent &amp; Approach</u></b>	
Normal procedures	S/US	Normal procedures	S/US
		Holding	S/US
<b><u>NGH PART</u></b>		<b><u>Landing</u></b>	
Normal procedures	S/US	Normal procedures	S/US
Non-normal procedures	S/US	Missed approach procedures	S/US
		Non-normal procedures	S/US
<b><u>Taxi-out &amp; takeoff</u></b>		<b><u>Taxi-in &amp; park</u></b>	
Normal procedures	S/US	Normal procedures	S/US
		Non-normal procedures	S/US
<b><u>Climb</u></b>			
Normal procedures	S/US		
Runaway stab. (demo)	S/US		
W/W fire (demo)	S/US		
REMARKS:			
<u>Good PROGRESS</u>			
<u>READY FOR FULL FLIGHT</u>			
<u>SIMULATOR</u>			
INSTRUCTOR NAME:		INSTRUCTOR SIGNATURE:	
<u>IHAB EL SONBATY</u>			
DATE:		TRAINING SIGNATURE:	
<u>1-05-03</u>			

**Full Flight Simulator Training:**

15

TRAINING RECORD FFS

LESSON-1




NAME <u>KHEDR ABDALAA</u>		CREW POSITION: <u>CAPTAIN / PTO</u>	
AIRCRAFT TYPE <u>BOIEN 737-300/400/500</u>		DATE OF COPELETION <u>03-05-03</u>	
Briefing Training plan Operation philosophy <u>S/US</u>		Cruise 1. Normal procedures <u>S/US</u>	
Preflight Normal procedures <u>S/US</u> Supplementary normal procedures <u>S/US</u>		Descent & approach Normal procedures <u>S/US</u>	
Engine start Normal procedures <u>S/US</u> Additional training item <u>S/US</u>		Landing Normal procedures <u>S/US</u>	
Taxi-out & Takeoff Normal procedures <u>S/US</u>		Taxi-in & park Normal procedures <u>S/US</u>	
Climb Normal procedures <u>S/US</u> Demonstration flight <u>S/US</u>			
REMARKES:			
<u>CAPTAIN KHEDR</u>			
<u>NEEDS TO MORE EFFORT</u>			
<u>TO REMEMBER RECALL ITEM'S</u>			
INSTRUCTOR NAME: <u>EMAR EL SOURAT</u>		SIGNATURE:	

TRAINING RECORD FFS

LESSON-2

1

NAME <u>KHEDR ABDALAA</u>		CREW POSITION: <u>CAPTAIN LPTO</u>	
AIRCRAFT TYPE <u>BOEING 737-300/400/500</u>		DATE OF COPELETION <u>4-05-03</u>	
<b>Briefing</b> Set up MCP, CDU Engine inoperative characteristics <u>S/US</u>		Cruise, DESCENT 2. Hydraulic system A loss <u>S/US</u>	
<b>Preflight</b> Set up MCP, CDU <u>S/US</u> After start checklist <u>S/US</u>		<b>Approach, Landing</b> One engine inop manual, F/D ILS Approach <u>S/US</u> One engine inop visual traffic Patterns full stop. <u>S/US</u> Wind shear training Wind shear flight path control hold <u>S/US</u> A/P, A/T, F/D VOR approach Full stop landing <u>S/US</u>	
<b>Engine start</b> Normal procedures <u>S/US</u>		Taxi -in & park Normal procedures <u>S/US</u>	
<b>Taxi-out &amp; Takeoff</b> Rejected T/O <u>S/US</u> T/O engine failure after V 11 <u>S/US</u> T/O engine failure after V 1 <u>S/US</u> Wind shear near VR <u>S/US</u>			
<b>Climb</b> Normal procedures <u>S/US</u>			
REMARKES:			
<u>PROGRESSING.</u>			
<u>HE IS TENSE NEED'S</u>			
<u>TO RELAX</u>			
INSTRUCTOR NAME: <u>ZHAB EL SONRABY</u>		SIGNATURE: 	

TRAINING MANUAL

TRAINING RECORD FFS

LESSON 3

1

NAME <u>KHEDR ABDALAA</u>		CREW POSITION: <u>CAPTAIN L-FTO</u>	
AIRCRAFT TYPE <u>BOIEN 737-300/400/500</u>		DATE OF COPELETION <u>6-05-03</u>	
<p><b>Briefing</b></p> <p>Review item in phase of flight <u>S/WS</u></p>		<p><b>Cruise , DESCENT</b></p> <p>Rapid depressurization <u>S/WS</u></p> <p>Emergency descent <u>S/WS</u></p> <p>Steep turns. <u>S/WS</u></p> <p>Approach to stall recovery <u>S/WS</u></p>	
<p><b>Preflight</b></p> <p>Normal procedures <u>S/WS</u></p>		<p><b>Approach , Landing</b></p> <p>One engine inop A/P , F/D VOR <u>S/WS</u></p> <p>Approach , circle to land , full <u>S/WS</u></p> <p>One engine inop . ILS approach <u>S/WS</u></p> <p>Missed approach <u>S/WS</u></p> <p>Hold <u>S/WS</u></p>	
<p><b>Engine start</b></p> <p>Aborted engine starts <u>S/WS</u></p>		<p><b>Taxi -in &amp; park</b></p> <p>Normal procedures <u>S/WS</u></p>	
<p><b>Taxi-out &amp; Takeoff</b></p> <p>Normal procedures <u>S/WS</u></p> <p>Rejected T/O <u>S/WS</u></p> <p>T/O engine failure after V 1 <u>S/WS</u></p> <p>Normal T/O <u>S/WS</u></p>		<p><b>Climb</b></p> <p>Wheel well fire <u>S/WS</u></p> <p>Runaway stabilizer</p> <p>Buss off</p> <p>Loss of both engine driven gen.</p>	
REMARKES:			
<u>STILL PROGRESSING.</u>			
<u>STILL NEED'S TO RG</u>			
<u>RELAX.</u>			
INSTRUCTOR NAME: <u>IHAB EL SONIBATY</u>		SIGNATURE: <u>[Signature]</u>	


OPERATIONS TRAINING MANUAL



TRAINING RECORD FFS

LESSON 4

1

NAME <u>KHEDR ABDALAA</u>		CREW POSITION: <u>CAPTAIN / FTO</u>	
AIRCRAFT TYPE <u>BOIEN 737-300/400/500</u>		DATE OF COPELETION <u>8-05-03</u>	
<b>Briefing</b> Full auto flight for precision app <u>S/US</u> Review item in phase of light		<b>Cruise</b> Steep turns. <u>S/US</u> Approach to stall recovery <u>S/US</u>	
<b>Preflight</b> Normal procedures <u>S/US</u> Reduced thrust computation <u>S/US</u>		<b>Descent,</b> Normal procedures <u>S/US</u> Econ path descent <u>S/US</u> Arrival procedures <u>S/US</u>	
<b>Engine start</b> Aborted engine starts <u>S/US</u>		<b>Approach, Landing</b> Normal procedures A/P, A/T, (no F/D) AUTO LAND	
<b>Taxi-out &amp; Takeoff</b> Normal procedures <u>S/US</u> NO autopilot & F/D <u>S/US</u> Reduced thrust takeoff <u>S/US</u> Flap retraction <u>S/US</u>		ILS approach <u>S/US</u> Touch & go landing Row data F/D ILS, T & GO. <u>S/US</u> A/P, A/T, F/D VOR approach <u>S/US</u> Touch & go landing <u>S/US</u>	
<b>Climb</b> Normal procedures <u>S/US</u> Max angle climb Econ climb		<b>Taxi-in &amp; park</b> Normal procedures <u>S/US</u>	
REMARKIES:			
<u>STILL IMPROVING.</u>			
<u>PUT HE NEED'S TO</u>			
<u>IMPROVE SINGLE ENG</u>			
<u>HANDLING.</u>			
INSTRUCTOR NAME: <u>THAIR ELSONRATY</u>		SIGNATURE: 	

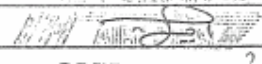
TRAINING MANUAL  
OPERATIONS

M. MORA  
497  
8-5-03

TRAINING RECORD FFS

LESSON 5


1

NAME <u>KHEDR ABDALAA</u>		CREW POSITION: <u>CAPTAIN / F/O</u>	
AIRCRAFT TYPE <u>BOEING 737-300/400/500</u>		DATE OF COMPLETION <u>09-05-03</u>	
<b>Briefing</b>		<b>Cruise</b>	
Set up MCP, CDU	<u>S/US</u>	Steep turns.	<u>S/US</u>
Engine inoperative flight characteristic	<u>S/US</u>	Approach to stall recovery	<u>S/US</u>
<b>Preflight</b>		<b>Descent,</b>	
Set up MCP, CDU	<u>S/US</u>	Normal procedures	<u>S/US</u>
After start checklist	<u>S/US</u>	Econ path descent	<u>S/US</u>
		Arrival procedures	<u>S/US</u>
<b>Engine start</b>		<b>Approach, Landing</b>	
Normal procedures	<u>S/US</u>	One engine inop. A/P, F/D NO	
<b>Taxi-out &amp; Takeoff</b>		A/T ILS approach	<u>S/US</u>
T/O engine failure after V1 (1)	<u>S/US</u>	Missed approach	<u>S/US</u>
T/O engine failure after V1(2)	<u>S/US</u>	One engine inop. Manual. F/D	
T/O engine failure after V1 (3)	<u>S/US</u>	NOA/T ILS approach	<u>S/US</u>
		Full stop landing	<u>S/US</u>
		Normal T/O manual Row data	<u>S/US</u>
		F/D ILS, T&GO	<u>S/US</u>
		Loss of both engine driven gen	<u>S/US</u>
		A/P, A/T, F/D VOR approach	<u>S/US</u>
<b>Climb</b>		Circle to land rejected landing	<u>S/US</u>
Normal procedures	<u>S/US</u>	A/P, A/T, F/D ILS approach	<u>S/US</u>
		VISUAL TRAFFIC PATTERNS	<u>S/US</u>
<b>Econ climb</b>		Taxi -in & park	
		Normal procedures	<u>S/US</u>
REMARKES:			
<u>PROGRESSING.</u>			
INSTRUCTOR NAME: <u>ZHAB EL SONBATY</u>		SIGNATURE: 	

TRAINING RECORD FFS

LESSON 6

1

NAME <u>KHEDR ABDALAA</u>		CREW POSITION: <u>CAPTAIN / PIC</u>	
AIRCRAFT TYPE <u>BOEING 737-300/400/500</u>		DATE OF COMPLETION <u>10-05-03</u>	
<b>Briefing</b> Set up MCP, CDU <u>S/VS</u> Engine inoperative flight characteristic <u>S/VS</u>		<b>Cruise &amp; Descent</b> Hydraulic system A loss <u>S/VS</u>	
<b>Preflight</b> Set up MCP, CDU <u>S/VS</u> After start checklist <u>S/VS</u>		<b>Approach, Landing</b> One engine inop manual, F/D <u>S/VS</u> ILS approach <u>S/VS</u> One engine inop. Visual traffic <u>S/VS</u> Pattern full stop <u>S/VS</u> One engine inop, landing <u>S/VS</u> Wind shear training <u>S/VS</u> Wind shear flight path control hold + <u>S/VS</u> A/P, A/T, F/D VOR APPROACH <u>S/VS</u> Full stop landing <u>S/VS</u> Taxi-in & park <u>S/VS</u> Normal procedures <u>S/VS</u>	
<b>Engine start</b> Normal procedures <u>S/VS</u>			
<b>Taxi-out &amp; Takeoff</b> Rejected T/O <u>S/VS</u> T/O engine failure after V1 <u>S/VS</u> T/O engine failure after V1 <u>S/VS</u> Wind shear near VR <u>S/VS</u>			
<b>Climb</b> Normal procedures <u>S/VS</u>			
REMARKS:			
<u>GOOD PROGRESS</u>			
INSTRUCTOR NAME: <u>THAR EL SONRATY</u>		SIGNATURE: 	

TRAINING MANUAL

*M. Mone*  
442  
 10.5.03

TRAINING RECORD FFS

LESSON 7

1

NAME <u>KHEOR ABDALAA</u>		CREW POSITION: <u>CAPTAIN / PLO</u>	
AIRCRAFT TYPE <u>BOEING 737-300/400/500</u>		DATE OF COMPLETION <u>11-05-03</u>	
<b>Briefing</b> Review item in phase of flight <u>S/VS</u> Set up MCP, CDU <u>S/VS</u>		<b>Cruise &amp; Descent</b> Hydraulic system A loss <u>S/VS</u>	
<b>Preflight</b> Set up MCP, CDU <u>S/VS</u> After start checklist <u>S/VS</u>		<b>Approach, Landing</b> AP, A/T, no F/D VOR approach <u>S/VS</u> Full stop landing <u>S/VS</u> HOLD <u>S/VS</u> Jammed stabilizer visual traffic <u>S/VS</u> Pattern full stop landing <u>S/VS</u> ASS. FLAPS <u>S/VS</u>	
<b>Engine start</b> FAST START <u>S/VS</u>		Hydraulic system A & B FAILURE <u>S/VS</u> MANUAL REVERGIN <u>S/VS</u> Visual traffic pattern all flap up <u>S/VS</u>	
<b>Taxi-out &amp; Takeoff</b> Normal procedures <u>S/VS</u> Normal T/O <u>S/VS</u>		Taxi-in & park <u>S/VS</u> APH fire <u>S/VS</u> Engine fire on 400 <u>S/VS</u> PASSENGER EVACUATION <u>S/VS</u>	
<b>Climb</b> Normal procedures <u>S/VS</u>			

REMARKS:

GOOD PROGRESS

READY FOR CHECK

RID

INSTRUCTOR NAME:

THAIR EL SONBATH

SIGNATURE:



OPERATIONS

TRAINING MANUAL

m. mohamed  
442  
11-5-03

TRAINING RECORD FFS

LESSON 8

NAME <u>KHEIR ABDALAA</u>		CREW POSITION: <u>CAPTAIN / PTO</u>	
AIRCRAFT TYPE <u>BOIEN 737-300/400/500</u>		DATE OF COPELETION <u>12-05-03</u>	
<b>Briefing</b>		<b>Cruise &amp; Descent</b>	
Review item in phase of flight	<u>S/US</u>	Steep turns	<u>S/US</u>
		Approach to stall recovery	<u>S/US</u>
		Holding	<u>S/US</u>
<b>Preflight</b>		Engine fire	<u>S/US</u>
Normal procedures	<u>S/US</u>	Wing / body over heat	<u>S/US</u>
		Bleed trip of	<u>S/US</u>
		Rapid depressurization	<u>S/US</u>
		Emergency descent	<u>S/US</u>
<b>Engine start</b>			
Normal procedures	<u>S/US</u>	<b>Approach, Landing</b>	
		One engine inop F/D, VOR	<u>S/US</u>
<b>Taxi-out &amp; Takeoff</b>		Approach, circuit to land	<u>S/US</u>
Rejected T/O	<u>S/US</u>	V1 cut one engine inop, ILS approach	<u>S/US</u>
1 engine failure after V1	<u>S/US</u>	Missed approach.	<u>S/US</u>
Normal T/O	<u>S/US</u>		
<b>Climb</b>		<b>Taxi-in &amp; park</b>	
Wheel well fire	<u>S/US</u>	Normal procedures	<u>S/US</u>
Runaway stabilizer	<u>S/US</u>		

REMARKS:

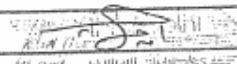
HE HAS PASSED

HES CHECK

SATISFACTORY

REDY FOR BASE TRAINING.

INSTRUCTOR NAME: THAIR EL SONBAY

SIGNATURE: 

OPERATIONS  
TRAINING MANUAL

m. no. 492  
12.05.02

**Proficiency Check:**

9

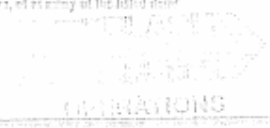


Ch.: 10

**FORMS AND RECORDS**

PROFICIENCY CHECK FORM			
Name <b>KHEDIR ABDALAA</b>	ID No. <b>106</b>	<b>CAPTAIN-PILOT</b>	
Simulator Owned by <b>ROYAL AIR MAROC</b>	Location <b>MAROC</b>	Simulator Level	
Flight Training Time <b>06:00</b>	Time PLO	Time PNF	Date <b>12-05-98</b>
This form is based on ECARS 121 Appendix F.			
Write (S or U) indicating Satisfactory or Unsatisfactory of each item.			
<b>1. ORAL TEST (operational oriented)</b>		<b>2. FLIGHT CHECK (cont'd)</b>	
<ul style="list-style-type: none"> <li>• Airplane Systems</li> <li>• Airplane performance</li> <li>• Normal and non-normal procedures**</li> <li>• Appropriate Provisions of AFM</li> <li>• Company flight operations manual</li> <li>• Use of checklists</li> </ul>		<p><b>INFLIGHT MANOEUVERS</b></p> <ul style="list-style-type: none"> <li>• Steep turns (Min. 120° -Max. 360°) 1</li> <li>• Approach to stalls (Two may be waived) 2</li> <li>• Take-off configuration</li> <li>• Clean configuration</li> <li>• Landing configuration</li> </ul> <p>Note : one stall must be performed with bank angle 25°</p>	
<b>2. FLIGHT CHECK</b>		<b>LANDINGS</b>	
<b>FRE FLIGHT AND TAXING</b>		<ul style="list-style-type: none"> <li>• Normal landing</li> <li>• From ILS</li> <li>• Cross wind</li> <li>• Visual approach</li> </ul>	
<ul style="list-style-type: none"> <li>• Pre flight and cockpit preparation</li> <li>• Engine start</li> <li>• Taxiing</li> </ul>		<ul style="list-style-type: none"> <li>• With 50% power plant failure (2 Engines on one side for 4 Engines airplanes)</li> <li>• From circling approach</li> <li>• Rejected at 50 FF.</li> </ul>	
<b>TAKE-OFFS</b>		<b>NORMAL AND NON-NORMAL PROCEDURES</b>	
<ul style="list-style-type: none"> <li>• Normal</li> <li>• Instrument (100' ceiling or 400 m RVR)</li> <li>• Cross wind</li> <li>• With simulated Engine failure</li> <li>• Rejected</li> </ul>		<ul style="list-style-type: none"> <li>• Anti icing and de-icing</li> <li>• Hydraulics</li> <li>• Electrical</li> <li>• Pneumatic</li> <li>• Gears</li> <li>• Flaps</li> <li>• Flight Controls</li> <li>• Nav/Comm. Equipment</li> </ul>	
<b>INSTRUMENT PROCEDURES</b>		<b>EMERGENCY PROCEDURES</b>	
<ul style="list-style-type: none"> <li>• Area departure</li> <li>• Area arrival and Holding</li> <li>• ILS approach (Coupled)</li> <li>• Second ILS approach (Manual)</li> <li>• Missed approach</li> <li>• Non-precision approach</li> <li>• Second Non-precision approach</li> <li>• Circling approach</li> <li>• Engine failure missed approach</li> </ul>		<ul style="list-style-type: none"> <li>• Inflight fire and smoke control</li> <li>• Decompression</li> <li>• Emergency descent</li> <li>• Emergency Landing (partial ldg. no flaps, etc.)</li> <li>• Emergency Evacuation</li> </ul>	
<b>OTHER EMERGENCY PROCEDURES RELATED TO SPECIFIC TYPE</b>			

\* This legend procedures are Abnormal, AABNormal, Abnormal and Emergency Procedures.  
 \*\* For Captains only.  
 \*\*\* The applicant must demonstrate the proper use, and apply the correct procedures, of emergency of the listed item.  
 1. one direction may be waived 2. two may be waived





FORMS AND RECORDS

PROFICIENCY CHECK FORM (cont'd)			
RHS TRAINING FOR INSTRUCTORS		RHS TRAINING FOR CAPTAINS	
• Error recovery		• Normal take Off	✓
• Lateral offsets		• Manual ILS (CAT I minima)	✓
• Vertical Offsets		• Non-Precision approach and landing	✓
• Minimum 3 Touch and Go		• Simulated Engine failure - Take off	✓
		• One Engine Out-Approach and landing	✓
EVALUATION			
KNOWLEDGE		US	S
FLIGHT OPERATION MANUAL (FOM) and Relevant ECARs			
A/C Systems, Limitations and Performance			✓
Normal, Non-Normal Procedures*			
PHARAOH AIR Operations Specifications			
FLYING SKILLS		US	S
Compliance with SOP (Flight operations Manual & FCOM)			
Attitude flying and correct trim technique			✓
Use of FMC, FMS, FMGS, etc...			
Aeroplane configuration, Attitude & Speed control			
Flying accuracy & Smoothness			
MANAGEMENT		US	S
Compliance with FLIGHT OPERATION MANUAL (FOM)			
Planning ahead and use of FMC, FMS, FMGS, etc...			
Crew co-ordination and use of available resources			✓
Adherence to clearances and safe heights			
Situational awareness			
Cabin crew safety briefing			
COMMENTS :			
<p><i>Good HANDLING</i>  <i>Good STANDARDIZATION</i></p> <p><i>Satisfactory and Ready for Base Training</i></p> <p><i>mm.mokaa 4/22</i></p>			
Base Month (through Last day of):		Licence Valid (through Last day of):	
Month	Year	Month	Year
Date of last 3 take-offs & Landings** :		Next Event <i>725/02</i>	
1. / /	2. / /	3. / /	
Name*** CP IP		ID No.	Check Airman's Signature
<i>JNAB ELSONBATHY</i>		<i>100</i>	<i>[Signature]</i>
Training Result		Trainee's Signature	Training Manager
Previous	US	<i>[Signature]</i>	<i>[Signature]</i>
Current	<i>B</i>		

**Base Flight Training:**



**FORMS AND RECORDS**

Ch.: 10

BASE FLIGHT TRAINING FORM						
Name <b>KHEOR ABDALAA</b>		ID No. <b>106</b>	Crew position <b>CAPTAIN</b>			
A/C Type <b>B 737/502</b>		A/C Registration <b>SIMULATOR</b>	Sim. Level (ZFT) <b>Level D</b>	Location <b>MAROC</b>	Date <b>13-05-08</b>	
R/W	GA	TG	FS	Weather <b>CAVOK</b>		
Flight type		Hours <b>00</b>	Minutes <b>00</b>			
Flight Maneuvers						
	Sim	A/C		Sim	A/C	
• Exterior inspection	<b>E</b>		• Visual approach (ILS supported)-T/Go	<b>E</b>		
• Cockpit preparation	<b>E</b>		• Visual approach (No ILS)-T/Go	<b>E</b>		
• Engines start	<b>E</b>		• Visual approach - (ILS supported)-T/Go	<b>E</b>		
• Taxi	<b>E</b>		• Visual approach - No ILS-No/ATIS-T/Go	<b>E</b>		
• Flex/reduced Thrust Take-Off	<b>E</b>		• Simulated engine failure after Take-Off	<b>E</b>		
• ILS pattern	<b>E</b>		• One Engine out visual approach	<b>E</b>		
• Automatic approach	<b>E</b>		• Engine(s) out landing	<b>E</b>		
Remarks						
<b>HE HAS PERFORMED</b>						
<b>REJECTED LANDING AT 50°</b>						
<b>WITH ONE ENGINE INOP</b>						
<b>Satisfactory, Ready for line Training 20 sectors</b>						
<b>with not less than 40 H and up to proficiency</b>						
<b>13-05-08</b>						
I hereby certify that: <b>CAPT / BO</b> is ready for A/C type rating						
CF Name <b>IMAB ELSONRAG</b>			CF Signature			
ID No. <b>106</b>	Inspector Name <b>M. MOUTRAC</b>		Training Manager			
OPERATIONS						



Company Oral Test

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Ch.: 10

FORMS AND RECORDS

COMPANY ORAL TEST			
Name	KHEDIR ABDALAA	ID No.	106
A/C Type	B 737-300/400/500	Crew position	CAPTAIN
Date	02-05-03	Location	MAROCO
The Company Oral is Oriented to Check the outcomes of all the practical phases of the training program and the knowledge of the operational aspects of the aircraft systems. The trainee must demonstrate knowledge of, but not limited to the items listed below.			
Enter :	S Satisfactory	U Unsatisfactory	N/A Not Applicable
Aircraft Limitations		Non-Normal and Emergency Procedures	
• All A/C systems limitation	S	• Ability to perform or state immediate action items	S
• Weight limitation	S	• Ability to locate Non-Normal Check list	S
• Performance	S	• Communication Between Cockpit and Cabin	S
• Knowledge of, and ability to compute	S	• Emergency Evacuation Procedures	S
• Takeoff data card	S	• Prepared / Unprepared emergency	S
• Landing data card	S	Aircraft systems	
• Cruise performance	S	• Electrical - Hydraulic - Pneumatic - Fuel	S
• Effect of MEL on Performance	S	• Powerplant EFIS Air conditioning Pressurization	S
• High speed Vs low speed Phases of takeoff	S	• Autopilot, FID- FMS, FMGS Navigation systems	S
• Wet and Contaminated Run Ways	S	• Flight controls-Flight instruments-Landing gear	S
• Flight Level selection, Specific Range and OPT.ALT	S	Flight operation Manual	
• Step Climb and Fuel Saving	S	• Weather minima Limitation(operations Manual)	S
• Cruise mach No. and manoeuvre capability	S	• Fuel policy	S
Normal Procedures		• Windshear, thunderstorms and turbulence	S
• Flight Crew operations Manual (FCOM) SOP	S	• Fueling with PAX on board	S
• Flight operations Manual (FCOM) SOP	S	• Dangerous goods	S
• Flight patterns	S	• Shoulder harness, seat belt policy and cockpit door	S
• Flight Control comm. Procedures (Stockholm radio)	S	• First officer T.O. and landing	S
Remarks			
GOOD KNOWLEDGE.			
Instructor Name	Code#	Result	Instructor Signature
IHAB EL SONBATY	100	S	[Signature]
Average		Passing Grade 70%	
Test Result	Trainee Signature	Training Manager Signature	
S	[Signature]	[Signature]	

M. Mohamed  
12/05/03  
OPERATIONS



FORMS AND RECORDS

**CERTIFICATION ORAL**

Crew position : <input checked="" type="checkbox"/> Capt. <input type="checkbox"/> F/O	<input type="checkbox"/> CPT <input type="checkbox"/> CSS <input type="checkbox"/> FBS
Name : <u>KHEDR ABDALAA</u>	<input checked="" type="checkbox"/> Sim
Code No. : <u>106</u>	AC Type : <u>B 737-300/400/500</u>
Date : <u>12-05-03</u>	Location : <u>MAROC</u>

The Certification Oral may be conducted at the end of CPT-CSS-FBS or before the Sim.Type Rating Check Ride.

	U		S		
	NA	US	S-	S	S+
<p>The Certification Oral is oriented to the knowledge of the operational aspects of the systems. The trainee must demonstrate a knowledge of the items listed below :</p>					
1. Knowledge of, and ability to compute :					
- Takeoff Data Card.				<input checked="" type="checkbox"/>	<input type="checkbox"/>
- Landing Data Card.				<input checked="" type="checkbox"/>	<input type="checkbox"/>
- Cruise Performance.				<input checked="" type="checkbox"/>	<input type="checkbox"/>
2. Ability to compute or validate weight and balance.				<input checked="" type="checkbox"/>	<input type="checkbox"/>
3. Preliminary Cockpit preparation :				<input checked="" type="checkbox"/>	<input type="checkbox"/>
- Emergency equipment check - Cockpit safety check				<input checked="" type="checkbox"/>	<input type="checkbox"/>
- APU start - Before start Cockpit preparation.				<input checked="" type="checkbox"/>	<input type="checkbox"/>
4. Knowledge of flight Engineer Station : <u>NA</u>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>
Sufficient for safe operation of airplane if the F/E is incapacitated or absent from the flight deck.				<input checked="" type="checkbox"/>	<input type="checkbox"/>
5. Ability to perform or state immediate action items.				<input checked="" type="checkbox"/>	<input type="checkbox"/>
6. Knowledge of, and ability to, state operating limitations.				<input checked="" type="checkbox"/>	<input type="checkbox"/>
7. Knowledge of MEL.				<input checked="" type="checkbox"/>	<input type="checkbox"/>
8. Knowledge of the following aircraft systems :				<input checked="" type="checkbox"/>	<input type="checkbox"/>
- Hydraulic				<input checked="" type="checkbox"/>	<input type="checkbox"/>
- Pneumatic				<input checked="" type="checkbox"/>	<input type="checkbox"/>
- Flight Instruments				<input checked="" type="checkbox"/>	<input type="checkbox"/>
- Landing gears				<input checked="" type="checkbox"/>	<input type="checkbox"/>
- EFIS, FMS, FMGS.				<input checked="" type="checkbox"/>	<input type="checkbox"/>
- Fuel				<input checked="" type="checkbox"/>	<input type="checkbox"/>
- Electrical				<input checked="" type="checkbox"/>	<input type="checkbox"/>
- Powerplants				<input checked="" type="checkbox"/>	<input type="checkbox"/>
- Flight controls				<input checked="" type="checkbox"/>	<input type="checkbox"/>
- Autopilot, F/D				<input checked="" type="checkbox"/>	<input type="checkbox"/>
- Navigation systems				<input checked="" type="checkbox"/>	<input type="checkbox"/>
- Air conditioning and pressurization				<input checked="" type="checkbox"/>	<input type="checkbox"/>

Result :	US	S-	S	S+	Trainee Signature : <u>[Signature]</u>
	<input type="checkbox"/>	<input type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>	Examiner Signature : <u>[Signature]</u>
Examiner Name : <u>JHAB EL SANISATI</u>					GMFT : <u>[Signature]</u>
Examiner Code : <u>106</u>					

Line Training:

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FORMS AND RECORDS

Ch.: 10

LINE TRAINING FORM (IOE)

Crew Position: CAPTAIN  
 Name: KHEIDR ABIDALMA A/C Type: B 73-3  
 ID No.: 106 Date: 28-05-03

Date	Route	Time		Sectors	
		Previous	Total	Previous	Total
26-06-02	SSH - LXR	00:50	00:50	1	1
28-06-02	LXR - SSH - CAP	01:40	02:30	2	3
26-06-02	CAP - ABE - ASN	02:15	04:45	2	5
27-06-02	CAP - LXR	01:10	05:55	1	6
02-06-03	LXR - SSH	00:45	06:40	1	7
02-07-03	CAP - BUS	05:10	11:50	1	8

Date	Comments	Instructor Name	Signature
06-06-03	GOOD PROGRESS	IMAB EC SAMBATI	[Signature]
07-06-03	GOOD PROGRESS	IMAB EC SAMBATI	[Signature]
02-07-03	GOOD PROGRESS	IMAB EC SAMBATI	[Signature]

OPERATOR



FORMS AND RECORDS

Ch.: 10

LINE TRAINING FORM (IOE)

Crew Position : CAPTAIN

Name : K. HEDR ABDALAA

A/C Type : B 73-3

ID No. : 106

Date : .....

Date	Route	Time		Sectors	
		Previous	11:50 Total	Previous	S
02-07-03	BVA - CAT	04:35	16:25	7	9
03-07-03	CAT - HRG	07:10	17:35	7	10
04-07-03	HRG - WAW	06:25	22:00	7	11
04-07-03	WAW - HRG	04:20	26:20	7	12
05-07-03	HRG - CAT	00:55	27:15	7	13
04-07-03	CAT - BVA	04:30	32:45	7	14

Date	Comments	Instructor Name	Signature
04-07-03	PROGRESSING PUT STILL NEED TO IMPROVE P.N.F. DOTO'S.	IHAIS GC SOMBATI	[Signature]
09-07-03	GOOD PROGRESS PUT STILL HE HAS ALOT TO DO AT HOME	IHAIS GC SOMBATI	[Signature]



LINE TRAINING FORM (IOE) (cont'd)

Date	Route	Time		Sectors	
		Previous	38:45 Today	Previous	14 Today
09-07-03	BVA - CAP		38:45		14
10-07-03	SAW - CAP	04:00	38:45	1	15
16-07-03	CAP - SSH	02:05	38:50	1	16
17-07-03	CAP - SSH	01:00	39:50	1	17
17-07-03	SSH - CAP	07:00	40:50	1	18
29-07-03	SSH - LXR	00:45	41:35	1	19
23-07-03	LXR - SSH	00:45	42:20	1	20

Date	Comments	Instructor Name	Signature
10-07-03	GOOD PROGRESS.	IHAIB GL SONBARI	[Signature]
17-07-03	STILL PROGRESSING	IHAIB GL SONBARI	[Signature]
23-07-03	GOOD PROGRESS READY FOR CHECK RID	IHAIB GL SONBARI	[Signature]

Trainee's Signature: ..... Training Manager: [Signature]

Note : 2 Sectors must be conducted from right hand seat (RHS) for Captains (one sector PF and one Sector PNF)

**Line Check:**

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**FORMS AND RECORDS**

**LINE CHECK FORM**

**THE FOLLOWING ITEMS MUST BE COVERED DURING LINE CHECK**

( ) Indicates that item has been checked

1. FLIGHT CHECK		DESCENT AND APPROACH	
<b>PRE FLIGHT</b>		ATIS, SNOWTAM and braking action*	
Dispatch		Descent planning	
• Reporting for duty	/	• Approach briefings, stars and	/
• Computerised and ATC flight plan	/	• Approaches:	/
• Weather briefing, T.O. and landing min.	/	• Precision <input checked="" type="checkbox"/> Non-precision <input checked="" type="checkbox"/> Visual <input checked="" type="checkbox"/>	/
• Alternate planning Wx min	/	• Destination and alternate weather minima	/
• NAT, Operations Specifications*	/	<b>LANDING AND TAXI IN</b>	
• NOTAM briefing and "B" snags	/	• Landing technique	/
• Cabin crew safety briefing	/	• Use of auto breaks and reverse thrust	/
Cockpit		• After landing and taxi in procedure	/
• Technical log and B snags	/	<b>2. KNOWLEDGE CHECK</b>	
• MEL-CDL and the effect on T.O./Landing	/	A) Flight operation manual	
• Performance	/	• IOE, initial release, USV and Command Responsibility	/
• Aircraft library and documentation	/	• Crew licence content	/
• Cockpit preparation-FMS/FMGS/PMS	/	• The difference between planning and actual Weather min. and Wx min. for new captain.	/
• TAKE OFF BRIEFING	/	• Fuel policy	/
• Load, trim sheet and NOTOC	/	• Windshear, thunderstorms and turbulence	/
• Cold Wx operation* Hot Wx operation	/	• Fueling with PAX on board	/
• T.O Performance, T.O speeds and C.G	/	• Dangerous goods	/
• Engine start procedures	/	• Shoulder harness, seat belt policy and cockpit door	/
<b>TAXI, TAKE-OFF AND INITIAL CLIMB</b>		• First officer T.O. and landing	/
• Push back procedures	/	• ECARS 121	/
• Taxi speed and braking technique	/	• Flight operations manuals & answers	/
• T.O roll and VI concept	/	B) Aircraft performance and technical knowledge	
• Noise abatement procedure and initial climb	/	• Operational system knowledge	/
• Best angle, best rate and turbulence speeds	/	• T.O performance limits	/
• Area departure, SID and holding	/	• Wet and contaminated runways	/
<b>CRUISE</b>		• Reduced (flex) thrust	/
• Flight level selection, specific range and OPT.ALT	/	• Approach and holding climb performance	/
• Step climb and fuel saving	/	• Normal, non normal and emergency procedure	/
• Cruise Mach no. and manoeuvre capability	/	• Flight patterns	/
• Use of weather radar and weather avoidance	/	c) Safety procedure	
• MNPS and MORA (Special routes)	/	• Communication between cockpit and cabin	/
• Drill down procedures	/	• Emergency evaluation procedure	/
• Enroute alternate and Emergency Proc. (NAV)*	/	• Perceived/imperceived emergency	/
• Alternate Weather minima	/	• Bomb on board and least risk location	/
• Minimum fuel for diversion(Alternate+Holdline)	/	• Crew in INCAPACITATION	/
• Communication failure procedures	/		
• Flight control comm. Procedures (Stockholm radio)	/		

\* if applicable



FORMS AND RECORDS

**LINE CHECK FORM** (cont'd)

Crew position: <input checked="" type="checkbox"/> Capt. <input type="checkbox"/> F/O <input type="checkbox"/> F/E		A/c Type: <u>B737-300</u>
Name: <u>K.HADR</u> <u>B.DALLAH</u> Code No. <u>126</u>		
<input checked="" type="checkbox"/> Final line check		<input type="checkbox"/> Recurrent Line Check
<input type="checkbox"/> Route check*		
Route	No. of sectors	Flight Time
<u>SSH - LXR - LXR - SSH</u>	<u>2</u>	<u>01:30</u>

PERFORMANCE EVALUATION		
KNOWLEDGE	US	S
A/C Systems, Limitations and Performance		✓
Normal Procedures		✓
Operation manual and ECARS		✓
Non-Normal Procedures*		✓
FLYING SKILLS	US	S
Attitude flying and correct trim technique		✓
Use of FMC, PMS, FMGS, etc...		✓
Complying with SOP (Normal, Abnormal & Emerg.)		✓
Aeroplane configuration, Altitude & Speed control		✓
Flying accuracy & Smoothness		✓
MANAGEMENT	US	S
Planning ahead and use of FMC, PMS, FMGS, etc...		✓
Crew co-ordination and use of available resources		✓
Adherence to clearances and safe heights		✓
Situational awareness		✓
COMMENTS:		
<u>(Good) STANDARD</u>		
<u>SATISFACTORY</u>		
<u>he can PANCE as CAPT in command Y-ZAKH</u>		
<u>23/07/03</u>		

Check Airman Name: <u>J.H.A.S. A.L. SADRATY</u>	Check Airman's signature: <u>[Signature]</u>
ID No. <u>[Blank]</u>	Trainee's signature: <u>[Signature]</u>
Check Result: <input checked="" type="checkbox"/> Satisfactory <input type="checkbox"/> Unsatisfactory	Training Manager: <u>[Signature]</u>

\* Route qualification is mandatory before conducting a route check  
 \*\* Non-Normal Procedure: Are Abnormal, Additional, Alternate and Emergency Procedures.

*[Handwritten signatures and stamps]*  
 OPERATIONS

**Recurrent Training:**

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Ch.: 10

FORMS AND RECORDS

PILOT'S RECURRENT TRAINING FORM			
Name <b>KHEOR ARDALLA</b>	ID No.	<b>321ENG 737-300/600</b>	
Simulator Owned by <b>ROYAL AIR MARCO</b>	Location <b>CAEP</b>	Simulator Level	
Flight Training Time <b>04:00</b>	Time PNF <sub>1</sub>	Time PNF <sub>2</sub>	Date <b>16-12-03</b> Simulator Level <b>1</b>
PART ONE : GROUND TRAINING SEGMENT			
[ ] indicates that item has been covered.			
<b>a) OPEN BOOK QUIZ (O&amp;A)*</b>		<b>b) Briefings</b>	
• Airplane Systems	<input checked="" type="checkbox"/>	• Use of checklists	<input checked="" type="checkbox"/>
• Airplane performance	<input checked="" type="checkbox"/>	• Review of normal training Scenario:	<input checked="" type="checkbox"/>
• Normal and non-normal procedures**	<input checked="" type="checkbox"/>	-Normal and Non-normal procedures**	<input checked="" type="checkbox"/>
• Appropriate Provisions of APM	<input checked="" type="checkbox"/>	-LOFT	<input checked="" type="checkbox"/>
• Company flight operations and route	<input checked="" type="checkbox"/>	-Windshear	<input checked="" type="checkbox"/>
• PharaohAir Operation Specifications	<input checked="" type="checkbox"/>	• CRM	<input checked="" type="checkbox"/>
PART TWO : FLIGHT TRAINING SEGMENT			
Scenario :			
<b>PRE FLIGHT AND TAXING</b>		<b>LANDINGS</b>	
• Pre flight and cockpit preparation	<input checked="" type="checkbox"/>	• Normal landing	<input checked="" type="checkbox"/>
• Engine start	<input checked="" type="checkbox"/>	• From ILS	<input checked="" type="checkbox"/>
• Taxiing	<input checked="" type="checkbox"/>	• Cross wind	<input checked="" type="checkbox"/>
<b>TAKE-OFFS</b>		• Visual approach	<input checked="" type="checkbox"/>
• Normal	<input checked="" type="checkbox"/>	• With 50% power plant failure	<input checked="" type="checkbox"/>
• Instrument(100' ceiling)	<input checked="" type="checkbox"/>	• (2 Engines on one side for 4 Engines airplanes)	<input checked="" type="checkbox"/>
• Cross wind	<input checked="" type="checkbox"/>	• From circling approach	<input checked="" type="checkbox"/>
• With simulated engine failure	<input checked="" type="checkbox"/>	• In Windshear conditions	<input checked="" type="checkbox"/>
• Rejected	<input checked="" type="checkbox"/>	• Rejected at 50 FT.	<input checked="" type="checkbox"/>
• Windshear during take-off	<input checked="" type="checkbox"/>	<b>NORMAL AND NON-NORMAL PROCEDURES</b>	<input checked="" type="checkbox"/>
<b>INSTRUMENT PROCEDURES</b>		• Anti-icing and de-icing	<input checked="" type="checkbox"/>
• Area departure	<input checked="" type="checkbox"/>	• Hydraulics	<input checked="" type="checkbox"/>
• Area arrival and Holding	<input checked="" type="checkbox"/>	• Electrical	<input checked="" type="checkbox"/>
• ILS approach (Coupled)	<input checked="" type="checkbox"/>	• Pneumatic	<input checked="" type="checkbox"/>
• Second ILS approach (Manual)	<input checked="" type="checkbox"/>	• Gears	<input checked="" type="checkbox"/>
• Missed approach	<input checked="" type="checkbox"/>	• Flaps	<input checked="" type="checkbox"/>
• Non-precision approach	<input checked="" type="checkbox"/>	• Flight Controls	<input checked="" type="checkbox"/>
• Second Non-precision approach	<input checked="" type="checkbox"/>	• Navicom Equipment	<input checked="" type="checkbox"/>
• Circling approach	<input checked="" type="checkbox"/>	<b>EMERGENCY PROCEDURES</b>	<input checked="" type="checkbox"/>
• Engine failure missed approach	<input checked="" type="checkbox"/>	• Inflight fire and smoke control	<input checked="" type="checkbox"/>
<b>INFLIGHT MANEUVERS</b>		• Decompression	<input checked="" type="checkbox"/>
• Steep turns (Min. 180° -Max. 360°)	<input checked="" type="checkbox"/>	• Emergency descent	<input checked="" type="checkbox"/>
• Approach to stalls	<input checked="" type="checkbox"/>	• Emergency Landing (partial Up, no flaps etc.)	<input checked="" type="checkbox"/>
• Specific flight characteristics	<input checked="" type="checkbox"/>	• Emergency Evacuation	<input checked="" type="checkbox"/>
<b>OTHER EMERGENCY PROCEDURES</b>			
<b>TOPS RESOLUTION</b>	<input checked="" type="checkbox"/>		
<b>RUSM</b>	<input checked="" type="checkbox"/>		

\* Q&A question and answers  
 \*\* Non-Manual procedures (i.e. Abnormal, Additional, Abnormal and Emergency Procedures)  
 \*\*\* For Captain only.





FORMS AND RECORDS

PILOT'S RECURRENT TRAINING FORM (cont'd)			
RHS TRAINING FOR INSTRUCTORS		RHS TRAINING FOR CAPTAINS	
• Error recovery		• Normal take Off	
Lateral offsets		• Simulated Engine failure – Take off	
Vertical Offsets		• One Engine Out-Approach and landing	
• Minimum 3 Touch and Go		• Minimum 3 Touch and Go's	
EVALUATION			
KNOWLEDGE		US	S
FLIGHT OPERATION MANUAL (FOM) and Relevant ECARs			
A/C systems Limitations and Performance			
Normal Non-Normal Procedures*			
PHARAOH AIR Operations Specifications			
FLYING SKILLS		US	S
Compliance with SOP (Flight operations Manual & FCOM)			
Attitude flying and correct trim technique			
Use of FMC, PMS, FMGS, etc...			
Aeroplane configuration, Attitude & S speed control			
Flying accuracy & Smoothness			
MANAGEMENT		US	S
Compliance with FLIGHT OPERATION MANUAL (FOM)			
Planning ahead and use of FMC, PMS, FMGS, etc...			
Crew coordination and use of available resources			
Adherence to clearances and safe heights			
Situational awareness			
Cabin crew safety briefing			
COMMENTS :			
SATISFACTORY			
Base Month (through Last day of) :		License Valid (through Last day of) :	
Month	Year	Month	Year
Date of last 3 take-offs & Landings** :		Next Event	
1. / /		Proficiency check	
2. / /			
3. / /			
Name*** CP IP		ID No.	Check Airman's Signature
IHAB EL SONBATI		107	
Training Result		Trainer's Signature	Safety & Training Manager
Previous	US (S)		
Current	US (S)		

\* Non-Normal procedures : see Abnormal, Additional, Alternate and Emergency Procedures.

\*\* Trainee is responsible for the accuracy of this data, and he must sign the form.

\*\*\* CP: Check Airman, IP :Instructor Pilot.

vii- Personal situation

The captain was married and had 3 children ages 29, 25 and 18 years. The eldest son is married and is doing post graduate studies in USA. The second son is an engineer. The youngest daughter is still studying in university.

The captain has no known problems of any kind. He is known to be devoted to his family. He did not suffer from any abnormal health or social problem.

(Refer also to page 72 of the Factual Report (Interviews regarding Captain Kheider Abdullah)

1.5.1.3. 72-hour history of the captain:

Refer to interviews on page 73 of the FR.

The captain and F/O left Cairo to SSH on January 1<sup>st</sup>, 2004 as passengers on Flash Airline flight departing Cairo at 15:00 GMT. No more factual information could be obtained regarding the 72-hour history.

- 1.5.1.4. Interviewing the individuals who trained and flew with the captain  
(including ground and simulator instructors)

**Interview with Captain/ Essam Eldin Brahmin Chief Pilot and  
instructor ATR 42 Scorpio Airlines during the period of  
employment of Captain/ Khedr in this Airline.**

- ***How well did you know Captain/ Khedr?***  
He was a colleague during work at the Egyptian Air force and when he joined Scorpio, we worked together as I was Chief Pilot. I was in charge of organizing his flying schedule and monitoring his standard through line checks.  
He was a well disciplined pilot, observed his flying schedule without any problems, was always careful to observe duty time limitation and rest periods, had good relations with his colleagues, was cheerful with his crew and always prepared his flight carefully.  
During line check he performed well. He was attentive to his work, communicated well with his crew and was not tense. His previous experience on military air transport made him comfortable in flying commercial air transport with relation to route experience and airway flying requirements.
- ***What routes were flown at this time?***  
Mainly domestic flights.
- ***Was Sharm El Sheikh one of your common destinations?***  
Yes.
- ***What was the common departure procedure Followed out of Sharm El Sheikh?***  
The standard procedure followed was depending on the runway in use a turn was initiated towards the sea while climbing in a wide pattern to cross the VOR 11000 Ft to proceed on the 306 Radial to Cairo.
- ***Did you as chief pilot and instructor see or have any report of any kind about Captain/ Khedr?***  
All comment and observations were good Captain and comfortable to work, always well prepared for his flight and kept his cockpit organized.
- ***Why did he leave Scorpio?***  
He left when the company stopped operations.

**Interview with Captain/ Emad Sallam Instructor Pilot on C130  
In the Egyptian Air force  
At the time Captain/ Khedr started to fly in the military air transport.**

- ***How well did you know Captain/ Khedr?***

As a pilot in the Air force we were colleagues although he was more senior than I, when he moved from the fighter squadrons to the air transport and when assigned to the C 130 I was an instructor and when he was assigned to training flights under my command was very willing and had no attitude about my being instructor with less seniority, he was always eager to learn and very attentive in the cockpit had no problem in asking for information from the crew with him and did not exercise unnecessary authority due to his rank, listened well to comments and observations of all the crew members without regard to rank and seniority was cheerful but well disciplined his training progress was standard.

---

**Interview with Captain/ Essam Eldin Ibrahim Chief Pilot and instructor ATR 42 Scorpio Airlines during the period of employment of Captain/ Khedr in this airline.**

**- *How well did you know Captain/ Khedr?***

He was a colleague during work at the Egyptian Air force and when he joined Scorpio we worked together as I was Chief Pilot I was in charge of organizing his flying schedule and monitoring his standard through line checks.

He was a well disciplined pilot observed his fighting schedule without any problems was always careful to observe duty time limitation and rest periods had good relations with his colleagues was cheerful with his crew and always prepared his flight carefully.

During line check he performed well was attentive to his work communicated well with his crew and was not tense his previous experience on military air transport made him comfortable in flying commercial air transport with relation to route experience and airway flying requirements.

**- *What routes were flown at this time?***

Mainly domestic flights.

**- *Was Sharm El Sheikh one of your common destinations?***

Yes.

**- *What was the common departure procedure Followed out of Sharm El Sheikh?***

The standard procedure followed was depending on the runway in use a turn was initiated towards the sea while climbing in a wide pattern to cross the VOR 11000 Ft to proceed on the 306Radial to Cairo.

**- *Did you as chief pilot and instructor see or have any report of any kind about Captain/ Khedr?***

All comment and observations were good Captain and comfortable to work, always well prepared for his flight and kept his cockpit organized.

**- *Why did he leave Scorpio?***

He left when the company stopped operations.

1.5.1.5. Interviewing CAA inspectors who flew with captain.  
Interviews to be carried out by OPS group

1.5.1.6. Interviewing former head of operations in Flash Airlines  
(No official former head of operation in Flash Airlines)

1.5.1.7. Additional factual documentation (Captain)

Number of days the captain had been working since his last day off.

**1.0 CAPT: KHIDR**

<b>DATE</b>	<b>A/C</b>	<b>FLT</b>	<b>CAPT</b>	<b>REMARKS</b>
1/12/03	ZCD	CAI/BCN BCN/MAD MAD/LXR	PIC D.H D.H	HE RETURNED TO CAI AS A PAX ON FSH 8883 LXR/CAI T/O
2/12/03		OFF		
3/12/03	1.2 ZCF	CAI/LYS LYS/CHG CHG/HRG	D.H  IC PIC	
4/12/03	1.3	OFF		
5/12/03	1.4	HRG/LXR	PIC	
6/12/03	1.5	LXR/CAI	PIC	
7/12/03	1.6	CAI/SSH SSH/NAP NAP/BRI BRI/SSH	D.H D.H PIC PIC	
8/12	1.7	OFF		
9/12	1.8	SSH/CAI	PIC	
10/12 TO 17/12	1.9	OFF		
18/12	1.10	CAI/SSH SSH/CAI	PIC PIC	
19/12	1.11	OFF		HE TRAVELLED AS A PAX FROM CAI TO HRG
20/12	1.12	CDG/LXR	PIC	HE WAS PAX ON FSH 606 HRG/CDG
21/12	1.13	LXR/SSH SSH/NAP NAP/BRI BRI/SSH	PIC PIC H.D H.D	HE RETURNED TO CAI AS A PAX ON MSR FLT
22/12	1.14	CAI/BCN BCN/MAD MAD/LXR	H.D PIC PIC	HE RETURNED TO CAI ON FSH 8883 AS A PAX

23/12	1.15	SSH/AOI AOI/BRI BRI/SSH	PIC PIC PIC	HE TRAVELLED FROM LXR TO SSH ON FSH 313 AS A PAX
24/12	1.16	SSH/LXR LXR/SSH	PIC PIC	
25/12	1.17	SSH/CAI	PIC	
26/12	1.18	BCN/MAD MAD/ASW	PIC PIC	HE TRAVELLED AS A PAX ON FSH884 CAI/BCN& RETURNED AS A PAX ON FSH 8885 ASW/CAI
27/12	1.19	LXR/CDG	PIC	HE TRAVLLED ON MSR TO LXR& RETURNED AS A PAX ON FSH 603 LXR/CAI
28/12	1.20	OFF		
29/12	D	CAI/BCN BCN/MAD MAD/LXR	H.D PIC PIC	HE RETURNED AS A PAX ON FSH 8883 LXR/CAI

## 2.0

D.H: DEAD HEADING

PIC: PILOT IN- COMMAND

DATE	A/C	FLT	CAPT	REMARKS
30/12		OFF		
31/12	2.2	CAI/ CDG CDG/CAI	IC PIC	
1/1/04	2.3	OFF		HE TRAVELLED TO SSH AS A PAX ON FSH 314 CAI/SSH
2/1/04	2.4	SSH/TRN TRN/SSH	PIC PIC	
3/1/04	2.5	SSH/CAI	PIC	CRASH

### Note:

The captain and F/O left Cairo to SSH on January 1<sup>st</sup>, 2004 as passengers on Flash Airline flight departing Cairo at 15:00 GMT



Captain interpersonal characteristics, including perceptions of fellow pilots regarding their capability for assertiveness.

All available information is available in pages 72-73 Factual Report

Familiarity of the two flight crew members with each other. (Including number of legs flown together this trip, number of legs flown together in the last 30 days.

According to the available information, the accident flight was the 3<sup>rd</sup> sector in the last 24 hours.

Description of how well the flying crew got along. No information available

Reported proficiency information. Outcome and comments from training records and proficiency check forms.

Refer to 1.5.1.2 (vi)

Spatial disorientation or upset recovery training received at Flash Air or in the military. *A1196*

According to CAA regulations, Spatial Disorientation training is not mandatory

No available documents from Flash Airline concerning SD training. Some verbal reports from the Egyptian Air Force are available concerning the captain SD training the time he was serving in the Egyptian Air Force as a military fighter pilot.

Inputs from different investigation partners are needed.

According to and CAA regulations, Upset Recovery training is not mandatory

Upset Recovery Training recommendation should be included in the Recommendations Chapter.

Captain's flying proficiency and cockpit style from fellow pilots, instructors, and/or check pilots.

Refer to 1.5.1.4 and 1.5.1.2 (vi)

*Flash Airlines chief pilot view regarding the departure procedure from SSH, based on company procedures*

According to Chief Captain Flash Airline and all other pilots questioned about departure procedure from SSH, all agree that a turn towards the sea is initiated with a bank angle depending on available rate of climb and captain's discretion to cross the VOR on course radial 306 at or above 10500 ft.

*Number of departures from SSH previously made by the captain (day and night)*

Within the last month, the captain has made five departures from SSH including the accident flight.

(SAT 03-Jan-04 (night), FRI 02-Jan-04 (night), THU 25-Dec-03 (night), WED 24-Dec-03 (day) and TUE 23-Dec-03 (day))

*The captain's time on Russian aircraft (MiG-21). Hercules transport aircrafts C130 (dates and number of hours). ADI display configuration in comparison with B737-300 ADI display.*

Refer to captain CV, and item 1.5.1.2 (vi)

Captain flew approximately:

Russian Mig: 1000 flying hours (Russian ADI display)

C130: 5000 hours (Conventional ADI display)

ATR: 700 hours (Conventional ADI display)

Boeing 737: 700 hours (Conventional ADI display)

For B737-300 ADI refer to 1.16.1.9 (reference CairoMarch04Slides (March Progress Meeting - Cairo).pdf file)

Comparison with ADI Displays for other airplanes types might be made by the OPS group if needed

## 1.5.2 The First Officer

### 1.5.2.1. Summary: (Personal, training information)

Date of birth: January 1, 1979

Date of hire with Flash Airlines: May 22, 2002

Egyptian Commercial Pilot License Number 3284 (issued April 12, 1997)

TYPE RATINGS: CESSNA (ISSUED April, 12, 1997) I

B737-200 (ISSUED July, 22, 1998) II

B737-300/400/500 (ISSUED July, 18, 2002) II

Commercial Pilot License issued by the Federal Aviation Administration (FAA)

Certificate Number 2546582 (issued July 31, 1996)

Airplane Multi-Engine Land Instrument Airplane

Private Privileges

Airplane Single Engine Land

Limitations: None

Medical: First Class last check (May 5, 2003)

Limitations: None, valid till May 4, 2004

Initial Ground School Training: Written Test June 10, 2002

Oral Test May 22, 2002

Initial Simulator Training  
2002

B-737-300/400/500: June 22–June 30,

Initial Proficiency Check

B-737-300/400/500: June 30, 2002

Line Check:

July 11, 2002

Last Proficiency Check:

May 15, 2003

Last Recurrent Training:

December 12, 2003

### FLIGHT TIMES:

Total flight time (hrs/min)<sup>3</sup>: 788:53

Total flight time B-737: 242:28

Total flying time last 24 hours<sup>4</sup>: 7:15

Total flying time last 30 days: 43:45

Total flying Time 90 days: 61:10

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<sup>3</sup> Times are calculated for the first officer up until December 31, 2003.

<sup>4</sup> Times do not include the accident flight.

#### 1.5.2.2. Background information.

- i- Beginning of his flying career.
  - The F/O began his ground training on the aircraft type 737-300 at Luxor Airway from 4 May 2002 to 16 May 2002
  - The F/O completed the Full Flight Simulator Training and the Flight Training at Flash Airline on 30 June 02

**Note:**

Luxor Air training forms are approved training syllabus by ECAA. The audit of Flash Airline carried on January 2003 comment that Flash was still using training forms under the name of the previous operator who was also ECAA approved but they should change the forms to the name of Flash.

- ii- All airlines worked for prior to Flash Air  
Refer to previous item
- iii- "All" F/O training records at Flash (including his last recurrent training).  
All flying hours before Flash were different training phases

License Renewal Form (Boeing 737-500):

(17)



وزارة الطيران المدني  
قطاع العمليات والنقل الجوي  
الإدارة المركزية للعمليات الجوية  
الإدارة العامة لإجازات الطيران

إخطار تجديد إجازة طيار

السيد الطيار / مدير عام العمليات

مؤسسية / شركة ملاءم للصحة

تحيةة وطنية وبصحة ..

بالإحالة إلى الطلب المقدم من السيد / محمود محمود عبد الكريم شامي

بمخصص تجديد إجازة / طيار (تج) رقم ٣٢٨٤ الحاصل عليها

نتشرف بالإفادة بأنه تم تجديدها من ٢٠٠٣/١٢/٣٠ إلى ٢٠٠٤/١٠/١٤

على طراز : TT B 737-500

علماً بأن 31 15 / 2004 GM 30 16 / 2004

وانتهاء اللياقة الطبية في ٢٠٠٤/١٠/١٤

وتفجناوا بقبول فائق الاحترام ..

محمد  
١٤٢٠

مدير عام إجازات الطيران

Signature of the Director General of Licenses

الهيئة العامة لتجديد الطاقم الجوية ٢٤٥٠ من ٢٠٠٢ - ٢٥٠٠٠

**Certificate of Validity of a license:**



جمهورية مصر العربية  
وزارة الطيران المدني  
قطاع العمليات والنقل الجوي

**شهادة سريان مفعول إجازة طيار**

١ - حالة هذه الشهادة بالنسبة للإجازة .  
هذه الشهادة جزء من إجازة طيار تبارك  
رقم ٣٢٨٤ ويجب وجودها دائماً بالإجازة .

٢ - سريان مفعول الإجازة .  
حامل الإجازة التي تعتبر هذه الشهادة جزءاً منها كشف  
عليه طبيباً بتاريخ ٥ / ٥ / ٣٠

وجد لائقاً للعمل وفقاً للاشتراطات الموضحة بالإجازة كما  
إنه قد أتم جميع الإجراءات لتجديدها وعليه فهي سارية  
المفعول للمدة من ٣٠ / ١٢ / ٣٠

إلى ٤ / ٥ / ٤٠ على طراز ٥٥٠ B737 II  
إلى ١ / ١ / ٤٠ على طراز ١ / ١ / ٤٠

الشهر الأساسي مايو ٣٠  
فترة السماح نوفمبر ٢٠٣٠

ARAB REPUBLIC OF EGYPT  
MINISTRY OF CIVIL AVIATION  
SECTOR OF OPERATIONS AND AIR TRANSPORT  
CERTIFICATE OF VALIDITY OF A LICENCE  
FOR PILOT'S OF FLYING MACHINES

1 - Status of this certificate .  
This certificate forms part of C01  
pilot's licence flying machines number 3284  
and must always be carried with the licence .

2 - Validity of the licence  
The holder of the licence of which this  
certificate forms part was medically examined  
on 5/5/30 and was assessed as fit to act  
in the capacity, and subject to the conditions,  
stated in the licence; he has also satisfied all the  
other requirements for the renewal of the  
licence , the licence is therefore; Valid :


from 30/12/30, to 4/5/40 Type II B  
from 1/1/40 to 1/1/40 Type ---

BM 1/1/40 GM Jan 2-4

Copy of the Commercial Pilot license:

- ٣ -

( استمارة رقم ٢٩ ط.م.ر.مكررا )



**إجازة طيار تجارى**  
**COMMERCIAL**  
**PILOT'S LICENCE**  
( تنفة )

الاسم عمرو محمود عبد الحليم شافعى  
IV. Name Amr-Mahmoud-Jhal'ou

العنوان ٢٠ بحارة المعلمين المحندية  
V. Belouraa-almoalmoon-almohandse

الجنسية EGYPTIAN  
VI. Nationality

توقيع حامل الإجازة  
XII. Signature of Holder

توقيع الرخص له بإصدار الإجازة  
X. Signature of Issuing Officer

الختم والتاريخ  
Date and Stamp

X1. By Authority of the  
C.A.A.  
12-6-77

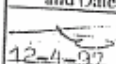
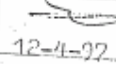
- ٢ -

صدرت هذه الإجازة بموجب المرسوم بشارون  
رقم ٢٨ بتاريخ ٢٣ مايو سنة ١٩٨١ والملاحق  
رقم ١ لمعاهدة الطيران المدني الدولية الموقعة  
في ٧ ديسمبر سنة ١٩٤٤ .



DATE OF BIRTH ١٠ / ١٠ / ١٩٦٩

يصرح الحامل هذه الإجازة بقيادة الطائرات  
الآلية الأثقل من الهواء طبقا للشروط  
والمواصفات المبينة بالإجازة على أن يكون  
حاصلا على استمارة رقم ٢٨ ( ط.م.و. )  
سارية المفعول .

أهلية - طراز الطائرات :

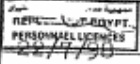

التوقيع والختم والتاريخ Signature, Stamp and Date	أهلية النوع - Class Rating
 12-4-97	Single & Multi ENG LAND PLANS
التوقيع والختم والتاريخ Signature, Stamp and Date	أهلية الطراز ( المجموعة ١ ) Type Rating (Group 1)
 12-4-97	172 - GORRA

جمهورية مصر العربية  
وزارة الطيران المدني  
نطاق العمليات والنقل الجوي  
شهادة لمفعول إجازة طيار

١- حالة هذه الشهادة بالنسبة للإجازة.  
هذه الشهادة جزء من إجازة طيار   
رقم ٤٤٨٤. ويجب وجودها دائما بالإجازة  
٢- سريان مفعول الإجازة  
حامل الإجازة التي تعتبر هذه الشهادة جزءا منها كشف  
عليه طيا بتاريخ ١١ / ٥ / ٢٠٠٢  
وجد لائقا للعمل وفقا للاشتراطات الموضحة بالإجازة  
كما أنه قد أتم جميع الإجراءات لتجديدها وعليه فهي  
سارية المفعول للمدة من ١٨ / ٧ / ٢٠٠٢  
إلى ٢٨ / ٢ / ٢٠٠٣ على طراز 172-3,4  
إلى  على طراز  
الشهر الأساسي ٢٠٠٢  
فترة السماح ٢٠٠٢



XII. The Aircraft Rating :

التوقيع والختم والتاريخ Signature, Stamp and Date	أهلية الطراز ( المجموعة ٢ ) Type Rating (Group 2)
 PERSONAL LICENCES 17/7/52	B.737/200
 17/7/52	B.737/3, 4, 5

ARAB REPUBLIC OF EGYPT  
 MINISTRY CIVIL AVIATION  
 SECTOR OF OPERATIONS AND AIR TRANSPORT  
 CERTIFICATE OF VALIDITY OF A LICENCE  
 FOR PILOT'S OF FLYING MACHINES

1- Status of this certificate.

This certificate forms part of CA 1  
 pilot's licence flying machines number 3281  
 and must always be carried with the licence.

2- Validity of the licence

The holder of the licence of which this certificate forms part was medically examined on 11/05/52 and was assessed as fit to act in the capacity, and subject to the conditions, stated in the licence; he has also satisfied all the other requirements for the renewal of the licence, the licence is therefore: Valid:  
 from 18/07/52 to 28/02/53 Type B.737-3  
 from 1-1 to 1-1 Type  
 DM GM  
JAN 53 FEB 53

Rating Contained in Licence is Valid

Type

The Privileges of an Instrument rating contained in the licence may be exercised as pilot in charge or as co-pilot (where one is required to be carried) of a flying machine.

From 18/07/02

TO 31/07/03

### CERTIFICATE

I the undersigned, a person fully authorised for this purpose by the Chairman of the SECTOR OF OPERATIONS AND TRANSPORT of the Arab Republic of Egypt hereby certify the Facts stated in Paragraphs 2.3.4

Signature

Date

Stamp

22/7/02

٣- أهلية مدرب المعتمدة بهذه الإجازة سارية

المنفعل إلى / طراز

٤- أهلية الطيران الآلي المعتمدة بالإجازة تخسر

لحامها الحق في العمل كقائد طائرة أو كطيار

مساعد (كما تقتضى الحالة) على الطائرات الآلية.

من ١٨ / ٧ / ٠٢

إلى ٣١ / ٧ / ٠٣

### شهادة

أنا الموقع أدناه بمقتضى السلامة المخولة لي من رئيس

قطاع العمليات والنقل الجوي بوزارة الطيران المدني

بجمهورية مصر العربية لقر بصحة ما جاء بالبنود

٢،٣،٤ من هذا المستند.

التوقيع :

التاريخ : ٢٢ / ٧ / ٠٢

الختم :

**B737-500 Transition Training:**

18

**FORM 1230-10P**

ORDER OF TRAINING NO 612002-1

LEVEL : F/O

PLACE : Luxor Air

SPECIALITY : Transition Course

AIRCRAFT TYPE : B737-300

TRAINING OFFICERS : 1. Amr Mahmoud Shafie

START DATE : Saturday May 4<sup>th</sup>, 02

END DATE : Thursday May 16<sup>th</sup>, 02

DURATION : 12 Days / 30 hours

ENCLOSURE (A) :

ENCLOSURE (B) :

INSTRUCTORS : 1. Capt / Ehab El-Sorbaty

2. Eng. / Mohamed Khalil

3. Eng. / Youssef Hassan

UNDERTRAINING :

SUPERVISOR :

SIGNATURE :

SUPERVISOR  
251021  
General Operation  
**FLASH**  
OPERATIONS



Boeing 737-300



Ground Training Syllabus

SYSTEM	HOURS REQUIRED
Weight& Balance	4 HRS
Air conditioning,presurization	5 HRS
Flight Controls	8 HRS
Hydraulic	3 HRS
Landing Gear	3 HRS
Navigation	5 HRS
Auto Flight	10 HRS
F.M.C	10 HRS
Pneumatic	3 HRS
Electric	4 HRS
Anti -ice	3 HRS
Oxygen	3 HRS
Engine	5 HRS
Fuel & APU	3 HRS
Performance	10 HRS
Total	80 HRS



58, Joseph Tito St., El-Nozha El-Gedidah, Cairo,Egypt.

tel. : 202-2944700-800-550 Fax : 202-2941300

SITA : CAIHPCR

OPERATIONS

58 شارع جوزيف تيتو النزهة الجديدة - القاهرة  
تليفون : 202-2944700 - 800-550 فاكس : 202-2941300

E-mail: hpline@internegypt.com

Proficiency Check (June 30, 02):



Form No. 02 - 2/2

PROFICIENCY CHECK FORM (cont'd)			
This Training is an		AIR T.M requirement and should be covered during Training day	
RHS TRAINING FOR INSTRUCTORS		RHS TRAINING FOR CAPTAINS	
• Error recovery		• Normal take Off	
Lateral offsets		• Manual ILS (CAT I minima)	
Vertical Offsets		• Non-Precision approach and landing	
• Minimum 3 Touch and Go		• Simulated Engine failure - Take off	
		• One Engine Out-Approach and landing	
EVALUATION			
KNOWLEDGE		US	S
FLIGHT OPERATION MANUAL (FOM) and Relevant ECARs		U	S
A/C Systems, Limitations and Performance		U	S
Normal, Non-Normal Procedures*		U	S
PHARAOH AIR Operations Specifications		U	S
FLYING SKILLS		US	S
Compliance with SOP (Flight operations Manual & FCOM)		U	S
Attitude flying and correct trim technique		U	S
Use of FMC, PMS, FMGS, etc...		U	S
Aeroplane configuration, Attitude & Speed control		U	S
Flying accuracy & Smoothness		U	S
MANAGEMENT		US	S
Compliance with FLIGHT OPERATION MANUAL (FOM)		U	S
Planning ahead and use of FMC, PMS, FMGS, etc...		U	S
Crew co-ordination and use of available resources		U	S
Adherence to clearances and safe heights		U	S
Situational awareness		U	S
Cabin crew safety briefing		U	S
COMMENTS :			
HE HAS PASSED HIS FINAL			
CHECK SATISFACTORILY WITH GOOD			
PERFORMANCE.			
SIGNED: <i>[Signature]</i>			
Date Month (through Last day of) :		Licence Valid (through Last day of) :	
Month	Year	Month	Year
Date of last 3 take-offs & Landings** :		Next Event	
1	2	3	4
1	1	1	1
Name***		ID No.	Check Airman's Signature
IZABEL SONBATHY		1004	<i>[Signature]</i>
Training Result		Trainee's Signature	Safety & Training Manager
Previous	Current	<i>[Signature]</i>	
OUS	OUS		
OS	MS		



Form No. 02 - 1/2

**PROFICIENCY CHECK FORM**

Name <u>AMR EL SHAFI</u>		ID No.	ICapt. <u>WFO</u>	II/III
Simulator Owned by <u>OLIMB/C</u>		Location <u>CI/BA</u>	Simulator Level	
Flight Training Time <u>32 hrs</u>	Time p/pt	Time PNF/pt	Date <u>30-08-02</u>	II A II B HC II
This form is based on ECARS 121 Appendix F.			AIRCRAFT TYPE: <u>B 737-300/400/500</u>	
Write (S or U) indicating Satisfactory or Unsatisfactory of each item.				
<b>1. ORAL TEST (operational oriented)</b>		<b>2. FLIGHT CHECK (cont'd)</b>		
<ul style="list-style-type: none"> <li>Airplane Systems</li> <li>Airplane performance</li> <li>Normal and non-normal procedures**</li> <li>Appropriate Provisions of AFM</li> <li>Company flight operations manual</li> <li>Use of checklists</li> </ul>		<b>INFLIGHT MANOEUVERS</b> <ul style="list-style-type: none"> <li>Steep turns (Min. 180° -Max. 360°) 1</li> <li>Approach to stalls (Two may be waived) 2</li> <li>Take-off configuration</li> <li>Clean configuration</li> <li>Landing configuration</li> </ul>		
<b>2. FLIGHT CHECK</b>		Note: one stall must be performed with bank angle 25°		
<b>PRE FLIGHT AND TAXING</b>		<b>LANDINGS</b>		
<ul style="list-style-type: none"> <li>Pre flight and cockpit preparation</li> <li>Engine start</li> <li>Taxing</li> </ul>		<ul style="list-style-type: none"> <li>Normal landing</li> <li>From ILS</li> <li>Cross wind</li> <li>Visual approach</li> <li>With 50% power plant failure (2 Engines on one side for 4 Engines airplanes)</li> <li>From circling approach</li> <li>Rejected at 50 FT.</li> </ul>		
<b>TAKE-OFFS</b>		<b>NORMAL AND NON-NORMAL PROCEDURES</b>		
<ul style="list-style-type: none"> <li>Normal</li> <li>Instrument (100' ceiling or 400 m RVR)</li> <li>Cross wind</li> <li>With simulated Engine failure</li> <li>Rejected</li> </ul>		<ul style="list-style-type: none"> <li>Anti icing and de-icing</li> <li>Hydraulics</li> <li>Electrical</li> <li>Pneumatic</li> <li>Gears</li> <li>Flaps</li> <li>Flight Controls</li> <li>Nav/comm. Equipment</li> </ul>		
<b>INSTRUMENT PROCEDURES</b>		<b>EMERGENCY PROCEDURES</b>		
<ul style="list-style-type: none"> <li>Area departure</li> <li>Area arrival and Holding</li> <li>ILS approach (Coupled)</li> <li>Second ILS approach (Manual)</li> <li>Missed approach</li> <li>Non-precision approach</li> <li>Second Non-precision approach</li> <li>Circling approach</li> <li>Engine failure missed approach</li> </ul>		<ul style="list-style-type: none"> <li>Inflight fire and smoke control</li> <li>Decompression</li> <li>Emergency descent</li> <li>Emergency Landing (partial ldg, no flaps, etc.)</li> <li>Emergency Evacuation</li> </ul>		
<b>OTHER EMERGENCY PROCEDURES RELATED TO SPECIFIC TYPE</b>				

\* Non-Normal procedures : are Abnormal, Additional, Alternate and Emergency Procedures.  
 \*\* For Captains only.  
 \*\*\* The applicant must demonstrate the proper use, and apply the correct procedures, of as many of the listed items. 1 on direction may be waived      2 two may be waived

Form No. C30-0369

PROFICIENCY CHECK FORM					
Name <i>ANDERSON SHAFER</i>		Code No.		ECapt	PIAO
Simulator Owned By <i>A/C FLASH AIRLINES</i>		Location <i>PHOENIX</i>		Aircraft Type <i>A320-233</i>	
Time PF <i>08:00</i>		Time PFD <i>08:00</i>		Date <i>11-27-07</i>	
Flight Training Time <i>00:00</i>		Date		11A	11B
This form is based on ECARS 121 Appendix F.					
Enter (S/U or NA) indicating Satisfactory/Unsatisfactory completion of each item or Not Applicable					
1. ORAL TEST (Operational Oriented Questions)			2. FLIGHT CHECK (cont'd)		
1. Aeroplane systems • Aeroplane performance • Normal and non-normal procedures 2. ETOPS, North Atlantic or special routes • Company flight operations manual • 3 of checklists			INFLIGHT MANEUVERS • Sleep time (Min 150" - Max 260") • Approach to stalls (Two may be waived) - Take-Off configuration - Clean configuration - Landing configuration Note: One stall must be performed with bank angle 25°.		
2. FLIGHT CHECK			LANDINGS		
PRE FLIGHT AND TAXIING			• Normal Landing • From ILS • Cross Wind • Visual approaches • With 50% power plant failure • (2 Eng.'s on one side for 4 Eng.'s aeroplanes) • From circling approach • Rejected at 50'		
1. Pre-flight and cockpit preparation			NORMAL AND ABNORMAL PROCEDURES		
2. Engine start			• Anti icing and De-icing • Hydraulics • Electrical • Pneumatic • Gears • Flaps • Flight Controls • Nav/Comm. Equipment		
3. Low visibility taxiing (150/200m RVR) <sup>1</sup>			EMERGENCY PROCEDURES		
TAKE-OFFS			• In-flight Fire and Smoke Control • Decompression • Emergency Descent • Emergency Landing (Partial L/C, No Flaps, etc.) • Emergency Evacuation		
1. Normal			EMERGENCY PROCEDURES RELATED TO SPECIFIC TYPE		
2. Low visibility takeoffs (150/200m RVR) <sup>2</sup>					
3. - X- Wind with loss of visual cues at 100 Kt					
4. - Rejected T.O with an engine failure before V <sub>1</sub>					
5. - With simulated engine failure at V <sub>1</sub>					
INSTRUMENT PROCEDURES					
1. Area departure					
2. Area arrival and Holding					
3. ILS approach (Coupled)					
4. Second ILS approach (Manual)					
5. Missed approach					
6. Non-precision approach					
7. Second Non-precision approach					
8. Circling approach					
9. Engine failure missed approach					
CAT II Approaches					
1. A min. of 3 CAT II approaches are required for					
2. CAT II recurrent					
SPECIAL TRAINING					
1. ETOPS					
2. North Atlantic En-route diversion scenario					
3. MNPS					

1. Non-Normal Procedures; Are Abnormal, Additional, Alternate and Emergency Procedures  
 2. 150/200m RVR for category C/D aircrafts respectively.  
 3. One direction may be waived.  
 4. For Captains Only.

Note:  
Heliopolis Airline operation ceased operation and Flash Airline took over its traffic rights and operated under the name of Flash Airline  
**Flight Training (August 12, 02):**

PROFICIENCY CHECK FORM (cont'd)				
This Training is an Egyptian T.M. requirement and should be covered during TRAINING DAY				
RHS TRAINING FOR INSTRUCTORS		RH: TRAINING FOR CAPTAINS		
<ul style="list-style-type: none"> <li>• Error recovery</li> <li>• Lateral effects</li> <li>• Vertical effects</li> <li>• Minimum 3 Touch and Go</li> </ul>		<ul style="list-style-type: none"> <li>• Normal Take Off</li> <li>• Minimum 3 CAT I descent</li> <li>• Minimum 3 Touch &amp; Go</li> </ul>		
Note: These minimums are not required per instructor				
EVALUATION				
Knowledge	US	S-	S	S+
Flight Operations Manual (FOM) and Relevant ECARs	<input type="checkbox"/>	<input type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>
A/C Systems, Limitations and Performance	<input type="checkbox"/>	<input type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>
Normal, Non-Normal Procedures	<input type="checkbox"/>	<input type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>
Fresh Air Operations Specifications	<input type="checkbox"/>	<input type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>
Flying Skills	US	S-	S	S+
Compliance with SOP (Flight Operations Manual & FCOM)	<input type="checkbox"/>	<input type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>
Attitude flying and correct trim technique	<input type="checkbox"/>	<input type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>
Use of FMC, PMS, FMGS, etc.	<input type="checkbox"/>	<input type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>
Aeroplane configuration, Altitude & Speed Control	<input type="checkbox"/>	<input type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>
Flying accuracy & Smoothness	<input type="checkbox"/>	<input type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>
Management	US	S-	S	S+
Compliance with Flight Operations Manual (FOM)	<input type="checkbox"/>	<input type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>
Planning ahead and use of FMC, PMS, FMGS, etc.	<input type="checkbox"/>	<input type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>
Crew co-ordination and use of available resources	<input type="checkbox"/>	<input type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>
Adherence to clearances and safe heights	<input type="checkbox"/>	<input type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>
Situational awareness	<input type="checkbox"/>	<input type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>
Cabin crew safety briefing	<input type="checkbox"/>	<input type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>
Comments				
Note: Please write positive or negative comments only, remedial actions are strictly a training management business.				
HE HAS PERFORMED AIRCRAFT GMS TRAINING				
ON IA A/C AT ADDISABABA FOR				
3 T.O AND 2 LANDINGS WITH GOOD				
REFERENCE TO THE SIGNATURE				
Base Month (Through Last Day of): Month: Year:	License Valid (Through Last Day of): Month: Year:	Next Event <input type="checkbox"/> PC <input type="checkbox"/> Recurrent Training		
Date of Last 3 Takeoffs and Landings: <sup>2</sup>	1. / /	2. / /	3. / /	
Check Airman Name IMAB EL SONBATHY	Code No. 1007	Check Airman's signature 		
Checking Result Previous <input type="checkbox"/> US <input type="checkbox"/> S Current <input type="checkbox"/> US <input checked="" type="checkbox"/> S	Trainee's signature	GM Flight Training		

1. Non-Normal Procedures: Are Abnormal, Additional, Alternate and Emergency Procedures.  
2. Trainee is responsible for the accuracy of this data, and he must sign the form.

IMAB





### Forms and Records



No. 04 - 1/4

IOE / USV FORM					
Type		ID No.	Captain's I/R/O		
Date	Route	Ratio		Sector	
		Previous	Today	Previous	Today
/					
/					
/					
/					
/					
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/					
/					
/					
/					
/					
/					
/					
/					
12-08-70					
12					
08					
70					
2					

**FINAL RELEASE**

This is to certify that Captain..... is fit fly as a  
 Captain under supervision on..... aircraft type.

Pilot Name <b>1203 IHAIB EL SONBATHY</b>	Signature <i>[Handwritten Signature]</i>	Date <b>12-08-70-2</b>
	Trainee's Signature <i>[Handwritten Signature]</i>	

Initial Operating Experience  
 Right hand Seat (Two sectors :one PF-one PNF)  
 Under Super Vision



Forms and Records



Form No. 04 - 1/4

IOE / USV FORM					
Name <i>AMR EL SHAFI</i>		ID No.	Jr. Capt.	<i>WFO</i>	LI F/E
A/C Type <i>B737-300</i>					
Date	Route	Time		Sectors	
		Previous → Today	Total	Previous → Today	Total
<i>27/07/02</i>	<i>SSH - CDG</i>	<i>05:15</i>	<i>05:15</i>	<i>1</i>	<i>1</i>
<i>27/07/02</i>	<i>CDG - FAR</i>	<i>05:00</i>	<i>10:15</i>	<i>1</i>	<i>2</i>
<i>30/07/02</i>	<i>HRG - CAT</i>	<i>07:00</i>	<i>11:15</i>	<i>1</i>	<i>3</i>
<i>01/08/02</i>	<i>CAT - BEE - SSH</i>	<i>02:50</i>	<i>14:05</i>	<i>2</i>	<i>5</i>
<i>07/08/02</i>	<i>SSH - CAT</i>	<i>20:50</i>	<i>14:55</i>	<i>1</i>	<i>6</i>
<i>10/08/02</i>	<i>CAT - CDG</i>	<i>04:30</i>	<i>14:20</i>	<i>1</i>	<i>7</i>
<i>11/08/02</i>	<i>CDG - HRG</i>	<i>05:15</i>	<i>24:35</i>	<i>1</i>	<i>8</i>
<i>12/08/02</i>	<i>HRG - BLC</i>	<i>03:55</i>	<i>28:30</i>	<i>6</i>	<i>14</i>
<i>12/08/02</i>	<i>BLC - HRG</i>	<i>03:45</i>	<i>32:15</i>	<i>1</i>	<i>15</i>
<i>1/1</i>					
<i>1/1</i>					
<i>1/1</i>					
Date	Type of Training	Comments		Instructor Name	Signature
<i>27/07/02</i>	<input type="checkbox"/> IOE <input type="checkbox"/> RHS <input type="checkbox"/> USV	<i>NORMAL FLIGHT</i>		<i>IHAB EL</i>	<i>[Signature]</i>
<i>30/07/02</i>	<input type="checkbox"/> IOE <input type="checkbox"/> RHS <input type="checkbox"/> USV	<i>PROGRESSING</i>		<i>SANBAY</i>	<i>[Signature]</i>
<i>01/08/02</i>	<input type="checkbox"/> IOE <input type="checkbox"/> RHS <input type="checkbox"/> USV	<i>NORMAL FLIGHT</i>		<i>IHAB EL</i>	<i>[Signature]</i>
<i>07/08/02</i>	<input type="checkbox"/> IOE <input type="checkbox"/> RHS <input type="checkbox"/> USV	<i>GOOD FLIGHT</i>		<i>IHAB EL</i>	<i>[Signature]</i>
<i>11/08/02</i>	<input type="checkbox"/> IOE <input type="checkbox"/> RHS <input type="checkbox"/> USV	<i>GOOD FLIGHT</i>		<i>IHAB EL</i>	<i>[Signature]</i>
		<i>READY FOR CHECK RIDE</i>		<i>SANBAY</i>	<i>[Signature]</i>

IOE : Initial Operating Experience  
 RHS : Right hand Seat (Two sectors: one PF-one PNF)  
 USV : Under Super Vision



Form No. 04 - 4/4

IOE / USV FORM (Cont'd)

EVALUATION

KNOWLEDGE	US	S
FLIGHT OPERATION MANUAL (FOM) and Relevant ECARs	?	2 ✓
A/C Systems, Limitations and Performance	?	2 ✓
Normal, Non-Normal Procedures*	?	2 ✓
LUXOR AIR Operations Specifications	?	2 ✓
FLYING SKILLS	US	S
Compliance with SOP (Flight operations Manual & FCOM)	?	2 ✓
Attitude flying and correct trim technique	?	2 ✓
Use of FMC, PMS, FMGS, etc...	?	2 ✓
Aeroplane configuration, Attitude & Speed control	?	2 ✓
Flying accuracy & Smoothness	?	2 ✓
MANAGEMENT	US	S
Compliance with FLIGHT OPERATION MANUAL (FOM)	?	2 ✓
Planning ahead and use of FMC, PMS, FMGS, etc...	?	2 ✓
Crew co-ordination and use of available resources	?	2 ✓
Adherence to clearances and safe heights	?	2 ✓
Situational awareness	?	2 ✓
Cabin crew safety briefing	?	2 ✓

Remarks

HE HAS PASSED WITH GOOD KNOWLEDGE AND GOOD PERFORMANCE

Date 12-08-02	ID No. 1003	Signature <i>[Signature]</i>
------------------	----------------	---------------------------------

This is to certify that all applicable Flight Training and Discussion items on this form have been completed and trainee is Ready For final line check and company oral

✓ ID Name IHAIBEL SONBATI	Signature <i>[Signature]</i>	Date 12-08-02
1003	Trainer's Signature <i>[Signature]</i>	

\*Normal procedures are Abnormal, Additional, Alternate and Emergency Procedures.

**Flight Deck Ground Training/ Competency Check/ General Emergency (22-05-02):**

شركة فلايس

(25)

EGYPT AIR TRAINING DIVISION  
Gen. Dept. for Aviation Training  
E.T. C.  
COMPETENCY CHECK.

قطاع التدريب  
الإدارة العامة لتدريب الطيران  
مركز تدريب الطوارئ

Flight deck Ground Training / COMPETENCY Check GENERAL EMERGENCY		
NAME: <i>Amr Mahmoud shakie</i>		Crew Position
Code: <i>X</i>	Cap <input type="checkbox"/> F/O <input checked="" type="checkbox"/> F/E <input type="checkbox"/>	
DATE: <i>22-05-2002</i>	LOCATION: E.T.C	<input type="checkbox"/> INITIAL <input checked="" type="checkbox"/> RECURRENT <input type="checkbox"/> RE-QUALIFICATION
ALL ITEMS MUST BE COMPLETED CHECK ( / ) INDICATING COMPLETION EACH ITEMS		
COMPETENCY CHECK ITEMS		
<b>PART 1: EMERGENCY SITUATION</b>		
- Flight CREWMEMBER DUTIES AND RESPONSIBILITIES	S	
- CREW COORDINATION AND COMPANY COMMUNICATION	S	
- AIRCRAFT FIRES	S	
- FIRST AID EQUIPMENT	S	
- ILLNESS, INJURY, AND BASIC FIRST AID	S	
- GROUND EVACUATION	S	
- DITCHING	S	
- RAPID DECOMPRESSION	S	
- PREVIOUS AIRCRAFT ACCIDENTS/INCIDENTS	S	
- CREWMEMBER INCAPACITATION	S	
- HIJACK AND BOMB THREAT	S	
<b>PART 2: EMERGENCY DRILL</b>		
- HAND-HELD FIRE EXTINGUISHERS	S	
- PORTABLE OXYGEN SYSTEM	S	
- EMERGENCY EXITS AND SLIDES. *	S	
- DITCHING EQUIPMENT. **	S	
INSTRUCTOR NAME AHMED HELMY	CODE NO. 8028	INSTRUCTOR SIGNATURE <i>[Signature]</i>
RESULT <input checked="" type="checkbox"/> Satisfactorily Completed	TRAINEE SIGNATURE <i>[Signature]</i>	E.T.C Manager <i>[Signature]</i>
G.M. AVIATION TRAINING		





PROFICIENCY CHECK FORM (cont'd)

RIIS TRAINING FOR INSTRUCTORS		RIIS TRAINING FOR CAPTAINS	
• Error recovery		• Normal take off	
• Lateral offsets		• Manual ILS (CAT I minima)	
• Vertical Offsets		• Non-Precision approach and landing	
• Minimum 3 Touch and Go		• Simulated Engine failure - Take off	
		• One Engine Out-Approach and landing	

EVALUATION		
KNOWLEDGE	US	S
FLIGHT OPERATION MANUAL (FOM) and Relevant ECARs		
A/C Systems, Limitations and Performance		
Normal, Non-Normal Procedures*		
PHARAOH AIR Operations Specifications		
FLYING SKILLS	US	S
Compliance with SOP (Flight operations Manual & FCOM)		
Attitude flying and correct trim technique		
Use of FMC, PMS, FMGS, etc...		
Aeroplane configuration, Attitude & Speed control		
Flying accuracy & Smoothness		
MANAGEMENT	US	S
Compliance with FLIGHT OPERATION MANUAL (FOM)		
Planning ahead and use of FMC, PMS, FMGS, etc...		
Crew co-ordination and use of available resources		
Adherence to clearances and safe heights		
Situational awareness		
Cabin crew safety briefing		

COMMENTS:

Satisfactory  
 Good handling  
 16.05.02

Base Month (through Last day of):  
 Month: 1, Year: 11  
 Licensee-Valid (through Last day of):  
 Month: 2, Year: 11  
 Next Event: 3, 11

Date of last 3 take-offs & Landings\*\*:

Name\*\*\*: YORDAN DIMITROV  
 CP: IP: ID No. 464

Training Result:  
 Previous: US S  
 Current: US S

Trainee's Signature: [Signature]  
 Check Airman's Signature: [Signature]  
 Training Manager: [Signature]



FORMS AND RECORDS

PILOT'S RECURRENT TRAINING FORM			
Name <i>AMR EL SHOFIE</i>		ID No.	
Simulator Owned by <i>ROYAL O.C. MARSA MATRUH</i>		Location <i>MARSA MATRUH</i>	Simulator Level
Flight Training Time <i>04:15</i>	Time PFT <i>1</i>	Time PNF	Date <i>11-12-03</i>
<b>PART ONE : GROUND TRAINING SEGMENT</b>			
[ ] indicates that item has been covered.			
<b>a) OPEN BOOK QUIZ (Q&amp;A)*</b>		<b>b) Briefings</b>	
• Airplane Systems	<input checked="" type="checkbox"/>	• Use of checklist	<input checked="" type="checkbox"/>
• Airplane performance	<input checked="" type="checkbox"/>	• Review of normal training Scenario:	<input checked="" type="checkbox"/>
• Normal and non-normal procedures**	<input checked="" type="checkbox"/>	-Normal and Non-normal procedures**	<input checked="" type="checkbox"/>
• Appropriate Provisions of A/P	<input checked="" type="checkbox"/>	-LOFT	<input checked="" type="checkbox"/>
• Company flight operations and route	<input checked="" type="checkbox"/>	-Windshear	<input checked="" type="checkbox"/>
• Pharaoh/Air Operation Specifications	<input checked="" type="checkbox"/>	• CRM	<input checked="" type="checkbox"/>
<b>PART TWO : FLIGHT TRAINING SEGMENT</b>			
Scenario			
<b>PRE FLIGHT AND TAXING</b>		<b>LANDINGS</b>	
• Pre flight and cockpit preparation	<input checked="" type="checkbox"/>	• Normal landing	<input checked="" type="checkbox"/>
• Engine start	<input checked="" type="checkbox"/>	• From ILS	<input checked="" type="checkbox"/>
• Taxiing	<input checked="" type="checkbox"/>	• Cross wind	<input checked="" type="checkbox"/>
<b>TAKE-OFFS</b>		• Visual approach	<input checked="" type="checkbox"/>
• Normal	<input checked="" type="checkbox"/>	• With 50% power plant failure	<input checked="" type="checkbox"/>
• Instrument (100' ceiling)	<input checked="" type="checkbox"/>	• (2 Engines on one side for 4 Engines airplanes)	<input checked="" type="checkbox"/>
• Cross wind	<input checked="" type="checkbox"/>	• From <u>        </u> icling approach	<input checked="" type="checkbox"/>
• With simulated engine failure	<input checked="" type="checkbox"/>	• In Windshear conditions	<input checked="" type="checkbox"/>
• Rejected	<input checked="" type="checkbox"/>	• Rejected at 50 FT.	<input checked="" type="checkbox"/>
• Windshear during take-off	<input checked="" type="checkbox"/>	<b>NORMAL AND NON-NORMAL PROCEDURES</b>	
<b>INSTRUMENT PROCEDURES</b>		• Anti icling and de-icing	<input checked="" type="checkbox"/>
• Area departure	<input checked="" type="checkbox"/>	• Hydraulics	<input checked="" type="checkbox"/>
• Area arrival and Holding	<input checked="" type="checkbox"/>	• Electrical	<input checked="" type="checkbox"/>
• ILS approach (Coupled)	<input checked="" type="checkbox"/>	• Pneumatic	<input checked="" type="checkbox"/>
• Second ILS approach (Manual)	<input checked="" type="checkbox"/>	• Gear	<input checked="" type="checkbox"/>
• Missed approach	<input checked="" type="checkbox"/>	• Flaps	<input checked="" type="checkbox"/>
• Non-precision approach	<input checked="" type="checkbox"/>	• Flight Controls	<input checked="" type="checkbox"/>
• Second Non-precision approach	<input checked="" type="checkbox"/>	• Nav/comm. Equipment	<input checked="" type="checkbox"/>
• Circling approach	<input checked="" type="checkbox"/>	<b>EMERGENCY PROCEDURES</b>	
• Engine failure missed approach	<input checked="" type="checkbox"/>	• Inflight fire and smoke control	<input checked="" type="checkbox"/>
<b>INFLIGHT MANEUVERS</b>		• Decompression	<input checked="" type="checkbox"/>
• Step turns (Min. 180° -Max. 360°)	<input checked="" type="checkbox"/>	• Emergency descent	<input checked="" type="checkbox"/>
• Approach to stalls	<input checked="" type="checkbox"/>	• Emergency Landing (partial lg, no flaps etc.)	<input checked="" type="checkbox"/>
• Specific flight characteristics	<input checked="" type="checkbox"/>	• Emergency Evacuation	<input checked="" type="checkbox"/>
<b>OTHER EMERGENCY PROCEDURES</b>			
<i>R.F.F.M</i>	<input checked="" type="checkbox"/>		
<i>T.O.P.P</i>	<input checked="" type="checkbox"/>		

\* Q&A :question and answers  
 \*\* Non-Nomnal procedures : are Abnormal, Additional, Alternate and Emergency Procedures.  
 \*\*\* For Captains only.



PILOT'S RECURRENT TRAINING FORM (cont'd)		
RHS TRAINING FOR INSTRUCTORS		RHS TRAINING FOR CAPTAINS
<ul style="list-style-type: none"> <li>Error recovery</li> <li>Lateral offsets</li> <li>Vertical Offsets</li> <li>Minimum 3 Touch and Go</li> </ul>		<ul style="list-style-type: none"> <li>Normal take off</li> <li>Simulated Engine failure - Take off</li> <li>One Engine Out-Approach and landing</li> <li>Minimum 3 Touch and Go's</li> </ul>
EVALUATION		
KNOWLEDGE		
FLIGHT OPERATION MANUAL (FOM) and Relevant ECARs	US	S
A/C systems Limitations and Performance		✓
Normal Non-Normal Procedures*		✓
PHARAOH AIR Operations Specifications		✓
FLYING SKILLS		
Compliance with SOP (Flight operations Manual & FCOM)	US	S
Attitude flying and correct trim technique		✓
Use of FMC, PMS, FMGS, etc...		✓
Aeroplane configuration, Attitude & S speed control		✓
Flying accuracy & Smoothness		✓
MANAGEMENT		
Compliance with FLIGHT OPERATION MANUAL (FOM)	US	S
Planning ahead and use of FMC, PMS, FMGS, etc...		✓
Crew coordination and use of available resources		✓
Adherence to clearances and safe heights		✓
Situational awareness		✓
Cabin crew safety briefing		✓
COMMENTS:		
SALES FACTORY		
GOOD KNOWLEDGE		
Base Month (through Last day of):		
Month _____ Year _____	License Valid (through Last day of):	Next Event
Month _____ Year _____	Month _____ Year _____	<input checked="" type="checkbox"/> Proficiency check
Duty of last 3 take-offs & Landings**:	1. / /	2. / /
Name*** CP _____ IP _____	ID No.	Check Airman's Signature
_____	107	_____
Training Result	Trainee's Signature	Safety & Training Manager
Previous <input checked="" type="checkbox"/>	_____	_____
Current <input checked="" type="checkbox"/>		

\* Non-Normal procedures : see Abnormal, Additional, Alternate and Emergency Procedures.  
 \*\* Trainee is responsible for the accuracy of this data, and he must sign the form.  
 \*\*\* CP: Check Airman, IP: Instructor Pilot.



نتيجة فرقة : تشيطة للسادة الطيارين العاملين بشركه فلاش

امر تدريب رقم :- ١٢٧ / ٢٠٠٢ ( طوارئ )

تاريخ بداية الفرقة :- ٢٠٠٢ / ٥ / ٢٢

تاريخ نهاية الفرقة :- ٢٠٠٢ / ٥ / ٢٣

ملاحظات	اسعافات	بضائع خطرة	سلامة طائرات	عملي	الاسم	م
ناجح	١٠٠	١٠٠	٩٦	١٠٠	ك / ايهاب السنباطي	١
ناجح	١٠٠	٩٣	١٠٠	١٠٠	ك / اشرف زارع	٢
راسب ب خطرة	١٠٠	غـ	٩٦	١٠٠	ك / نور سعد	٣
ناجح	١٠٠	٩٥	١٠٠	١٠٠	ك / خريستو لوستانس	٤
ناجح	١٠٠	٩٥	١٠٠	١٠٠	ك / وائل فكري	٥
ناجح	١٠٠	٩٥	١٠٠	١٠٠	ك / جمال عون	٦
ناجح	١٠٠	٩٥	٩٢	١٠٠	ك / عمرو عبد الحميد	٧
راسب	١٠٠	غـ	٩٦	غـ	ك / علي رشاد	٨
ناجح	١٠٠	٩٥	٩٦	١٠٠	م . ك / علي رشاد	٩
ناجح	١٠٠	٩٥	١٠٠	١٠٠	م . ك / محمد حسني	١٠
ناجح	١٠٠	٩٥	٩٢	١٠٠	م . ك / ياسر فكري	١١
ناجح	١٠٠	١٠٠	٩٦	١٠٠	م . ك / هبة درويش	١٢
ناجح	١٠٠	٨١	٩٦	١٠٠	م . ك / شيريف ابو العزم	١٣
راسب ب خطرة	١٠٠	غـ	٩٦	١٠٠	م . ك / خالد كوثر	١٤
ناجح	١٠٠	٩٥	١٠٠	١٠٠	م . ك / هاني المليجي	١٥
ناجح	١٠٠	٩٥	١٠٠	١٠٠	م . ك / عمر الشافعي	١٦
ناجح	١٠٠	٩٥	٩٦	١٠٠	م . ك / محمود حنفي	١٧
ناجح	١٠٠	٩٥	١٠٠	١٠٠	مرحل / اشرف لعلوم	١٨

Observer

Practical Test (Procedures)

التوقيع :-

التوقيع :-

الاسم :- ا / كلوديا يحيى وفا  
الوظيفة :- مدير ادارة مركز تدريب الطوارئ  
الوظيفة :- مدير عام الادارة العامة لتدريب الطيران

تحرير في ٢٠٠٢ / ٦ / ٢

رقم القيد : ٤٢٦  
التاريخ : ٢٠٠٢/٦/٢٦

وزارة الطيران المدني  
قطاع العمليات والنقل الجوي  
الادارة العامة لامتحانات الطيران  
CAA EXAM  
(oral)

2915 To 1016/2002 Shafie

نتيجة امتحان طيارين طراز B737-300  
بوينج ٣٠٠-٤٠٠ (شركة فلاش)  
الذي عقد في الفترة من ٥/٢٩ : ٢٠٠٢/٦/١٠

Performance Systems

م	الاسم	انظمه	اداء	ملاحظات
١	عسوي محمود عبد الحليم شافعي	٩٨	٩٨	تاجح
		٩٨	٩٨	Pass

روجعت طبقا لائحة الامتحانات

ابو الغيث محمد شحات  
المشرف على الاداره العامه لامتحانات الطيران

يعتمد.....

طيار/ صالح احمد موسى  
رئيس قطاع العمليات والنقل الجوي

Sleafra

كشف بنسبة حضور فرقة Basic indoctrination  
تاريخ بداية الفرقة 2002/ 8 / 21 ، تاريخ انتهاء الفرقة 2002 / 8 / 29

ملاحظات	الدورة	الاسم	م
حضر الدورة	Basic indoctrination	رضا السيد مصطفى	1
حضر الدورة	Basic indoctrination	محمود حنفي	2
حضر الدورة	Basic indoctrination	عمرو شافعي	3

توقيع المدرب : / /  
الاسم : / /  
الوظيفة : مدير الخطوط / مدير الجودة  
تحريرا في 29/8/2002 : / /

بدر العياريه / ابراهيم السبالي

/ /





Forms and Records



Form No. 04 - 3/4

IOE / USV FORM (Cont'd)

THE FOLLOWING ITEMS MUST BE COVERED DURING LINE CHECK

(N) Indicates that item has been checked

PRE FLIGHT	DESCENT AND APPROACH
<b>Dispatch</b>	ATIS, SNOWTAM and braking action*
• Computerized and ATC flight plan	• Descent planing
• Weather briefing, T.O. and landing min.	• Approach briefing and stars
• Alternate planing Wx min	• Approaches:
• NAT. Operations Specifications*	<input checked="" type="checkbox"/> Precision <input type="checkbox"/> N precision <input type="checkbox"/> Visual
• NOTAM briefing and "B" snags	• Destination and alternate weather minima
• Cabin crew safety briefing	<b>LANDING AND TAXI IN</b>
<b>Cockpit</b>	• Landing technique
• Technical log and B snags	• Use of auto breaks and reverse thrust
• MEL-CDL and the effect on T.O/Landing Performance	• After landing and taxi in procedure
• Aircraft library and documentation	<b>DISCUSSION ITEMS</b>
• Cockpit preparation-FMS/FMGS/PMS	<b>A) Flight operation manual</b>
• TAKE OFF BRIEFING	• IOE, Initial release, USV and Command Responsibility
• Load, trim sheet and NOTOC	• Navigation Bag content
• SNOWTAM (de-icing)*	• The difference between planning and actual Weather min. and Wx min. for new captain.
• Hot Wx operation	• Fuel policy
• T.O Performance, T.O speeds and C.G	• Windshear, thunderstorms and turbulence
• Engine start procedures	• Fueling with PAX on board
<b>TAXI, TAKE-OFF AND INITIAL CLIMB</b>	• Dangerous goods
• Push back procedures	• Shoulder harness, seat belt policy and cockpit door
• Aircraft geometry during turns	• First officer T.O. and landing
• Taxi speed and braking technique	• ECARS 121
• T.O roll and VI concept	• Flight operations manuals & answers
• Noise abatement procedure and initial climb	<b>B) Aircraft performance and technical knowledge</b>
• Best angle, best rate and turbulence speeds	• Operational system knowledge
• Area departure, SID and holding	• T.O performance limits
<b>CRUISE</b>	• Wet and contaminated runways
• Flight level selection, specific range and OPT. ALT	• Thrust (flex) thrust
• Step climb and fuel saving	• Approach and landing climb performance
• Cruise mach no. and manoeuvre capability	• Standard, non normal and emergency procedure
• Effect of weather on flight and weather avoidance	• Flight patterns
• CPDS and AD-PLA (Special routes)	<b>C) Safety procedure</b>
• Hold down procedures	• Communication between cockpit and cabin
• Enroute alternate and Emergency Proc.	• Emergency evaluation procedure
• Alternate Weather minima	• Crewed/uncrewed emergency
• Enroute fuel for diversion(Alternate+Holding)	• Crew on board and least risk location
• Communication holding procedures	• Crew incapacitation
• Flight control comm. Procedures (Stockholm radio)	

TRAINING RECORD, FFS - LESSON 1

NAME : AMR EL SHAEL CREW POSITION: F/O  
AIRLINE: Flash Airlines TYPE: B. 737-300-400-500

Briefing  
Training plan  SI Cruise  
Operation philosophy  SI Normal procedures  SI

Preflight  
Normal procedures  SI Descent , Approach  
Supplementary Normal procedures  SI Normal procedures  SI

Engine start  
Normal procedures  SI Landing  
Additional training item  SI Normal procedures  SI

Taxi- out & takeoff  
Normal procedures  SI Taxi - in & park  
Normal procedures  SI

Climb  
Normal procedures  SI  
Demonstration flight  SI

REMARKS  
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INSTRUCTOR amr DATE 22-6-2028

TRAINING RECORD FFS - LESSON 2

NAME : CHR. EL. SHAFI CREW POSITION: P/O  
AIRLINE: Flash Airlines TYPE: B-737-800-900-53

Briefing		Cruise , Descent	
Set up MCP ,CDU	<input checked="" type="checkbox"/>	Hydraulic system A loss	<input checked="" type="checkbox"/>
Engine inoperative flight characteristics	<input checked="" type="checkbox"/>		
Preflight		Approach , Landing	
Set up MCP ,CDU	<input checked="" type="checkbox"/>	One engine inop. manual , F/D	
After start checklist	<input checked="" type="checkbox"/>	ILS approach	<input checked="" type="checkbox"/>
Engine start		One engine inop. Visual traffic	
Normal procedures	<input checked="" type="checkbox"/>	Patterns full stop.	<input checked="" type="checkbox"/>
Taxi- out & takeoff		One engine inop. Landing	<input checked="" type="checkbox"/>
Rejected T/O	<input checked="" type="checkbox"/>	Wind shear training	<input checked="" type="checkbox"/>
T/O engine failure after V II	<input checked="" type="checkbox"/>	Wind shear flight path control	<input checked="" type="checkbox"/>
T/O engine failure after V I	<input checked="" type="checkbox"/>	Hold	<input checked="" type="checkbox"/>
Wind shear near VR	<input checked="" type="checkbox"/>	A/P ,AT ,F/D VOR approach , Full stop landing	<input checked="" type="checkbox"/>
Climb		Taxi - in & park	
Normal procedures	<input checked="" type="checkbox"/>	Normal procedures	<input checked="" type="checkbox"/>

REMARKS

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INSTRUCTOR: [Signature] DATE: 23-06-2007

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TRAINING RECORD, FFS - LESSON 3

NAME : OMR EL SHAEL CREW POSITION: F/O  
AIRLINE: Flash Airlines TYPE: B.737-300/400/500

Briefing  
Review item in phase of flight

Preflight  
Normal procedures

Engine start  
Aborted engine starts

Taxi-out & takeoff  
Normal procedures   
Rejected T/O   
T/O engine failure after V1   
Normal T/O

Climb  
Wheel well fire   
Runaway stabilizer   
Bus off   
Loss of both engine driven gen.

Cruise, Descent  
Rapid depressurization   
Emergency descent   
Steep turns   
Approach to stall recovery

Approach, Landing  
One engine inop. AP, F/D   
VOR approach, circle to land, full   
One engine inop. ILS approach   
missed approach   
Hold

Taxi-in & park

Normal procedure

REMARKS  
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INSTRUCTOR: aw DATE: 2002-10-28

TRAINING MANUAL

TRAINING RECORD FFS - LESSON 4

NAME: AHR EL SHARFI

CREW POSITION: F/O

AIRLINE: Flash Airlines

TYPE: B. 737-300/400/500

**Briefing**  
Full auto flight for precision app.  
Review item in phase of flight

**Cruise**  
Steep turns  
Approach to stall recovery

**Preflight**  
Normal procedures  
Reduced thrust computation

**Descent**  
Normal procedures  
Economy path descent  
Arrival procedure  
Approach Landing

**Engine start**  
Aborted starts (1)  
Aborted starts (2)

Normal procedures  
A/P, A/T (no F/D) autoland  
ILS approach  
Touch & go landing  
Row data F/D ILS, T&GO.  
A/P, A/T 3/D VOR approach  
Touch & go landing

**Taxi-out & takeoff**  
Normal procedures  
No autopilot & F/D  
Reduced thrust takeoff  
Flap retraction  
Climb  
Normal procedures  
Max angle climb  
Econ climb

**Taxi-in & park**  
Normal procedures

REMARKS  
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INSTRUCTOR: [Signature] DATE: 25-06-2002



TRAINING MANUAL

TRAINING RECORD - FFS - LESSON 5

30

NAME : AMR EL SHAFI CREW POSITION : P/O  
 AIRLINE : Flash Airlines TYPE : B. 737 - 300 / 400 / 600

<b>Briefing</b>		<b>Cruise, Descent</b>	
Set up MCP ,CDU	<input checked="" type="checkbox"/>		
Engine inoperative flight characteristics	<input checked="" type="checkbox"/>		
<b>Preflight</b>		<b>Approach, Landing</b>	
Set up MCP ,CDU	<input checked="" type="checkbox"/>	One engine inop. A/P , F/D	<input checked="" type="checkbox"/>
After start checklist	<input checked="" type="checkbox"/>	No A/T ILS approach	<input checked="" type="checkbox"/>
		Missed approach	<input checked="" type="checkbox"/>
<b>Engine start</b>		One engine inop. manual , F/D	<input checked="" type="checkbox"/>
Normal procedures	<input checked="" type="checkbox"/>	No A/T ILS approach	<input checked="" type="checkbox"/>
		Full stop landing	<input checked="" type="checkbox"/>
<b>Taxi- out &amp; takeoff</b>		Normal T/O , manual Row data	<input checked="" type="checkbox"/>
T/O engine failure after V1(1)	<input checked="" type="checkbox"/>	F/D ILS , T&GO	<input checked="" type="checkbox"/>
T/O engine failure after V1(2)	<input checked="" type="checkbox"/>	Loss of both engine driven gen.	<input checked="" type="checkbox"/>
T/O engine failure after V1(3)	<input checked="" type="checkbox"/>	Manual ILS , T&GO.	<input checked="" type="checkbox"/>
		A/P , A/T , F/D VOR approach	<input checked="" type="checkbox"/>
		circle to land rejected landing	<input checked="" type="checkbox"/>
		A/P , A/T , F/D ILS approach	<input checked="" type="checkbox"/>
		Visual traffic patterns	<input checked="" type="checkbox"/>
<b>Climb</b>		<b>Taxi - in &amp; park</b>	
Normal procedures	<input checked="" type="checkbox"/>	Normal procedures	<input checked="" type="checkbox"/>

REMARKS

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INSTRUCTOR: icd DATE: 16-06-2002

TRAINING MANUAL

TRAINING RECORD FFS - LESSON 6

NAME: AHR EL SHAFI

CREW POSITION: F/O

AIRLINE: Flash Airlines

TYPE: B. 737-300/400/500

<b>Briefing</b>		<b>Cruise, Descent</b>	
Set up MCP, CDU	<input checked="" type="checkbox"/>	Hydraulic system A loss	<input checked="" type="checkbox"/>
Engine inoperative flight characteristics	<input checked="" type="checkbox"/>		
<b>Preflight</b>		<b>Approach, Landing</b>	
Set up MCP, CDU	<input checked="" type="checkbox"/>	One engine inop. manual, F/D	
After start checklist	<input checked="" type="checkbox"/>	ILS approach	<input checked="" type="checkbox"/>
<b>Engine start</b>		One engine inop. Visual traffic	
Normal procedures	<input checked="" type="checkbox"/>	Patterns full stop	<input checked="" type="checkbox"/>
<b>Taxi-out &amp; takeoff</b>		One engine inop. Landing	<input checked="" type="checkbox"/>
Rejected T/O	<input checked="" type="checkbox"/>	Wind shear training	<input checked="" type="checkbox"/>
T/O engine failure after V II	<input checked="" type="checkbox"/>	Wind shear flight path control	<input checked="" type="checkbox"/>
T/O engine failure after V I	<input checked="" type="checkbox"/>	Hold	<input checked="" type="checkbox"/>
Wind shear near VR	<input checked="" type="checkbox"/>	A/P, A/T, F/D VOR approach	
<b>Climb</b>		Full stop landing	<input checked="" type="checkbox"/>
Normal procedures	<input checked="" type="checkbox"/>	<b>Taxi-in &amp; park</b>	
		Normal procedures	<input checked="" type="checkbox"/>

REMARKS

HE IS READY FOR CHECK RIDE.

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INSTRUCTOR: [Signature] DATE: 08-6-2007

TRAINING MANUAL

TRAINING RECORD FFS - LESSON 7

NAME : AMR EL SHAEL CREW POSITION: F/O  
AIRLINE: Flash Airlines TYPE: B-737-300/400/500

Briefing		Cruise, Descent	
Review item in phase of flight	<input checked="" type="checkbox"/>		
Set up MCP ,CDU	<input checked="" type="checkbox"/>		
Preflight		Approach, Landing	
Set up MCP ,CDU	<input checked="" type="checkbox"/>	MP ,MT ,no F/D VOR approach	
After start checklist	<input checked="" type="checkbox"/>	, full stop landing.	<input checked="" type="checkbox"/>
Engine start		Hold.	<input checked="" type="checkbox"/>
Fast start	<input checked="" type="checkbox"/>	Jammed stabilizer visual traffic	
		pattern full stop landing. (Capt)	<input checked="" type="checkbox"/>
Taxi- out & takeoff		ILS approach full stop landing .	<input checked="" type="checkbox"/>
Normal procedures	<input checked="" type="checkbox"/>	ASS. Flaps.	<input checked="" type="checkbox"/>
Normal T/O	<input checked="" type="checkbox"/>	Hydraulic System A & B failure	
Climb		Manual rev.	<input checked="" type="checkbox"/>
Normal procedures	<input checked="" type="checkbox"/>	Visual traffic patterns all up Flap + capt.	<input checked="" type="checkbox"/>
		Taxi - in & park	
		APU fire F/O	<input checked="" type="checkbox"/>
		Eng. fire on 400' ( capt.)	<input checked="" type="checkbox"/>
		Passenger evacuation	<input checked="" type="checkbox"/>

REMARKS

HE HAS PASSED THE CHECK RIDE

SATISFACTORY WITH (92%)

PERFORMANCE

ALP 442

INSTRUCTOR: [Signature] DATE: 30-06-2002

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TRAINING MANUAL

TRAINING RECORD FFS - LESSON 8

NAME: ADAR EL SHARAF

CREW POSITION: F/2

AIRLINE: Flash Airlines

TYPE: B.737-300/400/500

<b>Briefing</b>		<b>Cruise, Descent</b>	
Review item in phase of flight	<input checked="" type="checkbox"/>	Steep turns	<input checked="" type="checkbox"/>
		Approach to stall recovery	<input checked="" type="checkbox"/>
		Holding	<input checked="" type="checkbox"/>
<b>Preflight</b>		Engine fire	<input checked="" type="checkbox"/>
Normal procedures	<input checked="" type="checkbox"/>	Wing/body over heat	<input checked="" type="checkbox"/>
		Bleed or pack trip	<input checked="" type="checkbox"/>
<b>Engine start</b>		Rapid depressurization (capt)	<input checked="" type="checkbox"/>
<sup>18</sup> Normal procedures	<input checked="" type="checkbox"/>	Emergency descent	<input checked="" type="checkbox"/>
		<b>Approach, Landing</b>	
<b>Taxi-out &amp; takeoff</b>		One engine inop. F/D, VOR	
Rejected T/O	<input checked="" type="checkbox"/>	approach, circle to land (capt)	<input checked="" type="checkbox"/>
T/O engine failure after V1	<input checked="" type="checkbox"/>	V1 cut One engine inop. ILS	<input checked="" type="checkbox"/>
Normal T/O	<input checked="" type="checkbox"/>	Approach, missed approach	<input checked="" type="checkbox"/>
Climb			
Wheel well fire	<input checked="" type="checkbox"/>	<b>Taxi-in &amp; park</b>	
Runaway stabilizer	<input checked="" type="checkbox"/>	Normal procedures	<input checked="" type="checkbox"/>

REMARKS

HE HAS PERFORMED T.O AND  
 LANDING AS BASIC TRAINING AND  
 HE IS READY FOR THE DISTANCE  
 ON MAINTENANCE  
 ALP/MLZ

INSTRUCTOR ad DATE 07-07-2007

iv- Personal situation  
To be completed by the OPS Group

1.5.2.3. 72-hour history of the F/O:  
Refer to interviews included in pages 72-73 of the Factual Report

1.5.2.4. Interviewing the individuals who trained and flew with the F/O  
(including ground and simulator instructors)  
None available

1.5.2.5. Interviewing CAA inspectors who flew with F/O.  
Interviews to be carried out by OPS Group

1.5.2.6. Interviewing former head of operations at Flash Airlines  
(No official former head of operation in Flash Airlines)

1.5.2.7. Additional factual documentation (F/O)

Number of days the F/O had been working since his last day off.  
Refer to Factual Report

F/O interpersonal characteristics, including perceptions of fellow pilots regarding their capability for assertiveness.  
All available information is available in pages 72-73 Factual Report

Reported proficiency information. Outcome and comments from training records and proficiency check forms.  
Refer to 1.5.2.2 (iii)

Spatial disorientation or upset recovery training received at Flash Air AI196

According to CAA regulations, Spatial Disorientation training is not mandatory

No available documents from Flash Airline concerning SD training. Inputs from different investigation partners are needed.

According to and CAA regulations, Upset Recovery training is not mandatory

Upset Recovery Training recommendation may be included in the Recommendations Chapter.

F/O's flying proficiency and cockpit style from fellow pilots, instructors, and/or check pilots.  
Not available

### 1.5.3 The Observer

*Background:*

The Observer “Ashraf Abdel Hamid” was completing his training as a first officer for Flash Airlines.

*Beginning of his flying career:*

**Training at USA**

ISIS Airman Report                      CAIS Information - Basic Information  
Cert Pfx:    Cert No: 2440980    Cert Sfx:    Soc.Sec.No: 620480104  
Name: ABDELHAMID, ASHRAF                      Name Sfx:  
DOB: 1961 10 25 Sex: M Hair: BROWN Eyes: BROWN    Ht: 68 Wt: 154  
POB: CAIRO, EGYPT  
Status:                      Info:                      Name/Address Source: Air  
Date of Address Update: 2004 03 10    Citizenship: USA  
Street: PO BOX 414                      County: 065  
City: PALM DESERT                      State: CA    Zip: 92261-0414  
Country:

TOT CIVIL HOURS: 03750                      TOT MIL HOURS: 00400

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ISIS Airman Report                      CAIS Information - Medical  
Cert Pfx:    Cert No: 2440980    Cert Sfx:                      Information  
Medical Information for: ABDELHAMID, ASHRAF  
Class:                      First  
Certificate Desc.: LIMITED  
Medical Date: 2003 01 28    Medical ID#: 200001408794  
Restriction:  
MUST HAVE AVAILABLE GLASSES FOR NEAR VISION.

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ISIS Airman Report                      CAIS Information - Certificate  
Cert Pfx:    Cert No: 2440980    Cert Sfx:                      Information  
Spec Purp Pilot Info                      ABDELHAMID                      ASHRAF  
Cert-Level: COMMERCIAL PILOT (FOREIGN BASED)  
Rating/Level:  
AIRPLANE SINGLE ENGINE LAND/COMMERCIAL PILOT (FOREIGN BASED)  
INSTRUMENT AIRPLANE/COMMERCIAL PILOT (FOREIGN BASED)  
Type Rating/Level:  
Date of Issue: 1991 10 17    OrgDOI:                      Update Date: 1991 10 17  
Seal: Black                      Cert Status: Active

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ISIS Airman Report                      CAIS Information - Certificate  
Cert Pfx:    Cert No: 2440980    Cert Sfx:                      Information  
Spec Purp Pilot Info                      ABDELHAMID                      ASHRAF  
Certificate Limitations  
ISSUED ON BASIS OF AND VALID ONLY WHEN ACCOMPANIED BY CANADIAN  
PILOT LICENSE NO. C275467. ALL LIMITATIONS AND RESTRICTIONS ON THE  
CANADIAN PILOT LICENSE APPLY. NOT VALID FOR AGRICULTURAL  
AIRCRAFT OPERATIONS.  
INSTRUMENT AIRPLANE (U.S. TEST PASSED).

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ISIS Airman Report                      CAIS Information - Certificate  
Cert Pfx:    Cert No: 2635768    Cert Sfx:                      Information  
Pilot Information for:                      ABDELHAMID                      ASHRAF

Cert-Level: AIRLINE TRANSPORT PILOT  
Rating/Level:  
AIRPLANE MULTIENGINE LAND/AIRLINE TRANSPORT PILOT  
Type Rating/Level:  
Date of Issue: 2000 06 15 OrgDOI: Update Date: 2001 06 21  
Seal: Blue Cert Status: Active

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ISIS Airman Report CAIS Information - Previous Certificate  
Cert Pfx: Certificate No: 2440980 Cert Sfx:  
Previous Certificate for: ABDELHAMID ASHRAF

Previous Certificate Information:  
Pfx Cert Num. Sfx Cert Date Cert Level/Type

NO PREVIOUS CERTIFICATE INFORMATION AVAILABLE

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ISIS Accident/Incident (AID) Report Airman Accident/Incident  
Airman Name: ABDELHAMID, ASHRAF Cert #: 002440980  
Accident Date: 02/15/2001 Air Agency Cert #:  
Accident Event: GENERAL AVIATION ACCIDENT Source: .4  
Type of Accident: LOSS OF DIRECTIONAL CONTROL  
Accident Location-----  
City: SAN DIEGO State: CA

Aircraft Involved-----  
N-Number: N4922D  
Make: CESSNA Model: 172N

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ISIS Accident Incident Report Full AID Text Page No.: 1  
Case number: 4922D20010215115931 of 3  
Jump to page: \_\_ AID Text

ON FEBRUARY 15, 2001, ABOUT 1516 HOURS PST, A CESSNA 172N, N4922D, VEERED OFF THE RUNWAY AND COLLIDED WITH A TAXIWAY SIGN DURING LANDING ROLLOUT ON RUNWAY 28L AT THE MONTGOMERY FIELD, SAN DIEGO, CA. THE AIRPLANE WAS SUBSTANTIALLY DAMAGED. NEITHER THE AIRLINE TRANSPORT CERTIFICATED PILOT NOR PASSENGER WAS INJURED. PLUS ONE FLYERS, INC., IN SAN DIEGO, OPERATED THE AIRPLANE. VISUAL METEOROLOGICAL CONDITIONS PREVAILED AND AN INSTRUMENT FLIGHT RULES FLIGHT PLAN WAS FILED. THE PERSONAL FLIGHT WAS PERFORMED UNDER 14 CFR PART 91, AND IT ORIGINATED IN SCOTTSDALE, AZ. ABOUT 1135. AIRPORT PERSONNEL REPORTED THAT THE COLLISION OCCURRED ABOUT 1,000 FEET UPWIND OF THE RUNWAY'S THRESHOLD. THE AIRPLANE IMPACTED THE TAXIWAY "C" SIGN, AND VEERED OFF THE RUNWAY. THE AIRPLANE CAME TO A STOP ABOUT 200 FEET NORTH OF THE RUNWAY. THE PILOT STATED THAT DURING THE LANDING ROLLOUT, AS THE AIRPLANE WAS DECELERATING THROUGH ABOUT 50 KNOTS, THE LEFT WING SUDDENLY LIFTED UP. THEREAFTER HE LOST CONTROL OF THE AIRPLANE. HE ADDITIONALLY REPORTED THAT HE WAS UNAWARE OF THE REASON FOR THIS OCCURENCE. NO MECHANICAL MALFUNCTIONS WERE REPORTED WITH THE AIRPLANE.





NO RECORDS FOUND

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Inspection for Airman: ABDELHAMID, ASHRAF Recs: 1  
Using Certificate: 002635768 (Pilot) thru:  
Jump to: RECORD ID \_\_\_\_\_ Sort by column: 1 A of:  
Record ID Activity Code FAR Status Start Date Completion

NO RECORDS FOUND

***Interview with Brother of observer Pilot/ Ashraf Abdel Hamid:***

Captain/Alaa El Saadany Training Captain with EgyptAir was interviewed by Dr. Adel Fouad and Captain Shaker Kelada who said that Ashraf Abdel Hamid was a lively person sociable and easy to get along with, was friendly confident and out spoken. Asked about his career as a pilot he said that he started his initial training in Cairo than went to Canada and obtained Canadian citizenship and Canadian pilot license and flew single engine planes. He then went to the USA and also obtained USA citizenship and flew there on single engine and Lear jets had a total of around 4000 hrs.

On a family visit to Egypt, he was persuaded by Captain Sombaty (Operations Manager of Flash Airline), a colleague and personal friend to stay in Egypt and fly for Flash. He had attended B737 ground school course and was due for examination two days after the accident. He flew as an observer with Captain Sombaty who was assisting him to complete his B737 qualification.

***Correction:***

The following statement included in page 15 of the factual report should be deleted: Airline training procedures require a certain amount of observation time prior to serving as an active crew member. The observer was assigned to this flight to observe as a part of that training requirement.

The following statement should replace it:

Ashraf Abdel Hamid was flying as an observer as it is common practice for operators in Egypt is to assign pilots joining an airline or upgrading to a new type to fly as an observer on the type to be flown to get acquainted with company routes and procedures of the operator and type

CAA regulations regarding observation time:  
N/A

Flash Airline policy regarding observation time:  
As required

#### 1.5.4 Maintenance Engineer

Engineer Mostafa Erfan graduated from the National Civil Aviation Training Institute on September 6<sup>th</sup> 1972. He worked as a mechanic for the Kuwait Airways for twenty years during which he received the following training courses:

- 1- B 747-269B Mechanics Familiarization during the period from Feb 17<sup>th</sup> 1979 to March 3<sup>rd</sup> 1979. (Kuwait Airways).
- 2- Airbus Mechanics Familiarization Course during the period from October 6<sup>th</sup> to October 18<sup>th</sup> 1984 (Kuwait Airways).
- 3- B767 Mechanics Familiarization A& C Course during the period between February 7<sup>th</sup> to February 19<sup>th</sup>, 1987 (Kuwait Airways).

In 1991 he attended the Cessna 188 course at DEVCO training center, and then he got his Egyptian license without type rating (LWTR) No 1525 on August 1<sup>st</sup> 1992 which is valid until July 27<sup>th</sup>, 2004.

He joined Flash Airlines two years ago; during these two years he had the following training and exams:

- 1- B737-300 type course at EgyptAir approved training center during the period from December 22<sup>nd</sup>, 2002 to February 27<sup>th</sup>, 2003.
- 2- Basic Indoctrination Course during the period from 13-14 June 2003.
- 3- An On Job Training for 9 months on Flash Airlines B737-300 fleet.
- 4- An approval authorization exam for the engine on November 2<sup>nd</sup>, 2003 and for the airframe November 3<sup>rd</sup>, 2003.

His approval No: 014 Valid until: July 26<sup>th</sup>, 2004 Issued on: Nov 28<sup>th</sup>, 2003  
LWTR No: 1525 Valid until: July 27<sup>th</sup>, 2004 issued on: August 1<sup>st</sup>, 1992

## 1.6 Airplane Information

### 1.6.1 Airplane History

The accident airplane was a Boeing model 737-3Q8 airplane, serial number 26283, and was equipped with two CFM56-3 engines. The airplane was delivered on 22 October 1992 to an aircraft lessor. Since that time, it had been leased to several different operators and had carried US, UK, and Egyptian registration marks. The airplane had been operated by Flash Airlines since June 2001. At the time of the accident, the airplane carried Egyptian registration marks SU-ZCF and had accumulated 25603 flight hours and 17976 cycles.

Aircraft Type	: B737-3Q8
Minimum Crew	: 2 (Pilot and Copilot)
Registration Marks	: SU-ZCF
Serial Number	: 26283
Manufacture Date	: October 1992
Line Number	: 2383
Variable No	: PQ294
Interior Configuration	: Total 148 Economy Class
ECAA Minimum Number of Flight Attendant	: 3

## 1.6.2 Cockpit Instrumentation

The airplane was equipped with an electronic flight instrument system (EFIS) which provides displays for most of the airplane's navigational systems. The major displays provided by the EFIS are: color displays of pitch and roll; navigational maps; weather; radio altitude and decision height; and autopilot and flight path information. The EFIS also provides displays of: airspeed; ADF/VOR bearings; ILS data; and stall warning information. There are two separate display screens for each pilot, the electronic attitude direction indicator (EADI) and the electronic horizontal situation indicator (EHSI). The EADI is mounted just above the EHSI in front of each pilot. In addition to the EADI and EHSI, each pilot's panel includes an airspeed indicator, a radio digital distance magnetic indicator (RDDMI) which displays directions and distance to radio navigation aids, an altimeter, a vertical speed indicator (VSI), and a clock. See Figure 1.6.2-1 for a simulated view of the captain's panel showing these instruments.



Figure 1.6.2-1 Example Captain's Instrument Display

### 1.6.2.1 Electronic Attitude Direction Indicator (EADI)

The Electronic Attitude Director Indicator (EADI) provides a multicolor display of airplane attitude, airspeed, flight director commands and various other data. The primary display is an artificial horizon which depicts the pitch and roll of the airplane. The artificial horizon line which separates the upper blue portion of the display from the lower brown portion moves up and down as the airplane pitches and tilts *left and right as the airplane rolls*. The display is designed such that the artificial horizon line that appears on the display is always parallel with the real horizon. Pitch and roll data for the captain's and first officer's EADI are supplied by separate left and right inertial reference units. In independent standby attitude indicator is installed on the captain's panel inboard of the EADI. In addition to attitude information, the EADI displays a moving airspeed scale along the left side and ground speed in the lower left corner. The upper portion of the EADI is called Flight Mode Annunciator (FMA). This area is used to display the current operating modes of the autoflight system to the crew. The FMA is separated into four separate areas in which are displayed (from left to right), the autothrottle mode, pitch mode, roll mode, and autopilot mode. See section 1.6.4 for further information about the autopilot and flight director.

An example EADI screen is shown in Figure 1.6.2.1-1.



Figure 1.6.2-2 Example EADI Display – In this example, the airplane is pitch is 7.5 degrees above the horizon and the roll angle is 20 degrees to the left, airspeed is 220 knots, ground speed is 238 knots, the autopilot mode is "N1", the pitch mode is "MCP Speed", the roll mode is "heading select", and the autopilot mode is "Flight Director"

### 1.6.2.2 Electronic Horizontal Situation Indicator (EHSI)

The EHSI provides horizontal navigation information to the flight crew. There are a number of display formats available which can be separately selected by the flight crew. On the accident flight, both the captain and first officer were using the expanded VOR display which is described below



Figure 1.6.2-3 Example EHSI Display – Expanded VOR Mode – Flag notes denote various options



### 1.6.3 Lateral Flight Control System

Lateral control is provided by an aileron and two flight spoilers on each wing which are controlled by either control wheel in the flight deck. A pair of cables transfers motion of the control wheels to motion of an aft quadrant located near the main landing gear wheel well.

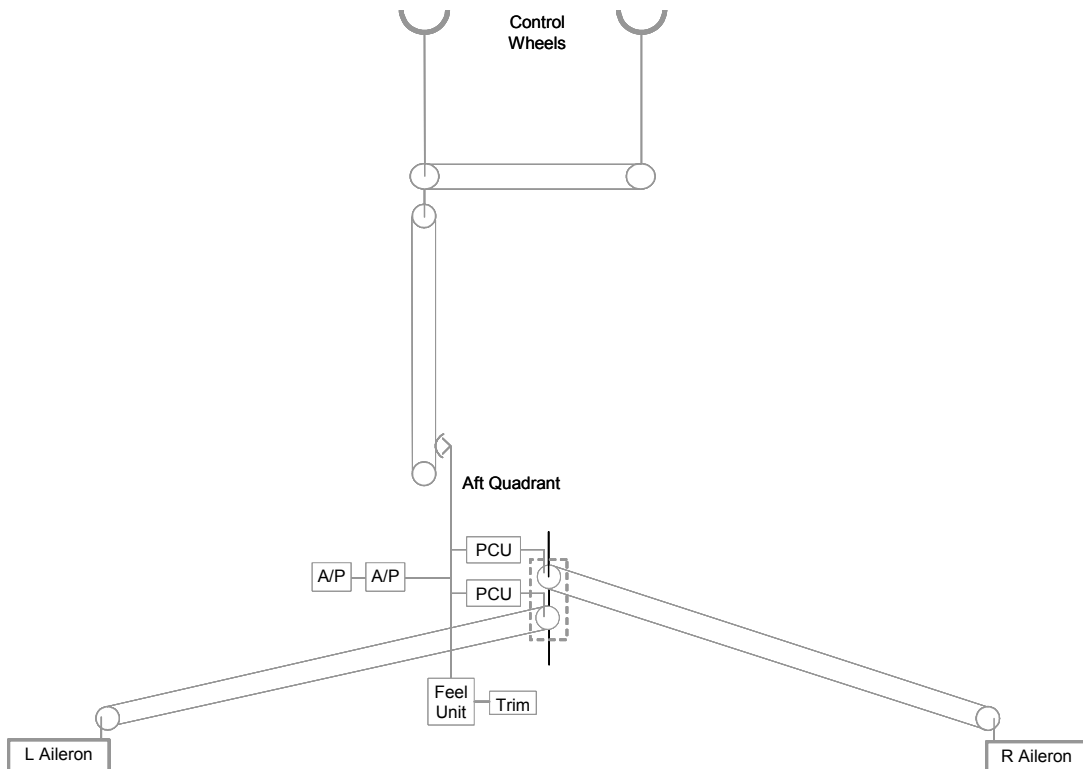


Figure 1.6.3-1 Simplified Lateral Control System Schematic – Additional cable runs, jam protection features, and spoilers not shown

The aft quadrant is connected to the control valves of two independent hydraulic power control units. Either unit alone is capable of providing full-range lateral control. Artificial feel and wheel centering for lateral control is provided by the feel unit which consists of a centering cam, roller, and spring. Aileron trim is accomplished with aileron trim switches on the aft end of the pilots' control stand. The trim switches command an electro-mechanical linear actuator which repositions the feel and centering mechanism.

Two flight spoilers on each wing operate in conjunction with the ailerons through a spoiler mixer mechanism connected to the aft quadrant.

Two autopilot actuators are connected to the aft quadrant. Either or both of the autopilot actuators can move the aft quadrant, resulting in movement of both the control wheels and the ailerons. One feature of the lateral control system is that the position of the ailerons always corresponds to the position of the wheel. Even if aileron trim or the autopilots are in use, the relationship between the position of the control wheels and the position of the aileron is unchanged.

#### 1.6.4 Autoflight System

The digital flight control system consists of a centrally located mode control panel (MCP), two independent flight control computers (FCCs), two aileron autopilot servo actuators, and two elevator autopilot servo actuators. Together, these components provide the functions of the autopilot and flight director. The MCP, located above the pilot's front panels and below the windows, provides a centralized location for all autopilot, flight director and autothrottle control selections. The FCCs receive flight crew requests and airplane sensor inputs which are used to generate flight director displays and, if the autopilot is engaged, command flight control surfaces.

##### 1.6.4.1 Autopilot System

Each of the two FCCs provides an independent autopilot and are designated A and B. Each FCC is connected to one aileron and one elevator servo actuator. The autopilot is engaged by selecting the appropriate push button on the MCP. If certain required conditions are met, the selected autopilot will synchronize the roll channel autopilot servo to the current position of the ailerons. Following synchronization, the autopilot servo will clamp onto the aft quadrant and begin moving the ailerons (and control wheel) in response to the flight path selected by the crew. A similar process occurs in the pitch channel. During cruise, only a single autopilot is used. If the second autopilot is selected, the first autopilot is disengaged when the second autopilot engages. During approach, both autopilots may be used together for two channel operation.

##### Engage Switches:

The pushbuttons are normally-open, momentary contact switches which control an engage relay by means of electronic circuitry. Either channel can be engaged in CWS or CMD by pressing the appropriate switch. A light illuminates on the switch to indicate that the autopilot has been engaged, and each switch may be disengaged by pressing the switch again. Loss of power (28v) or ground to the relay will cause it to de-energize and the pushbutton switch light will go out. If CWS or CMD is pressed while either power or ground for the relay is not provided, the relay will not energize and the pushbutton light will not illuminate.

##### Autopilot Actuators: (Figure 1.6.3-1)

A- Four autopilot actuators are installed, two in the main wheel well area for the aileron axis and two in the aft fuselage for the elevator axis. One set, aileron and elevator, is controlled by the A autopilot system and the other set by the B autopilot system. The units are mechanically linked to aileron and elevator power control units (PCU's) which drive the flight control surface

B- A pressure switch is installed on each actuator. The switch closes when normal hydraulic pressure is applied to the PCU. The engage interlock voltage is wired through the switches.

C- Autopilot system electrical signals operate valves which modulate hydraulic pressure to displace a hydraulic piston and provide a rotary output to the respective PCU. Control and position signals are provided by the following components which re

installed on each actuator: engage solenoids, transfer valve, linear variable displacement transducer (LVDT), and pressure regulator.

### 1- Engage Solenoids

Two engage solenoids are on each autopilot module. Each solenoid is an electrically operated valve (28 volts dc) which, when energized, applies hydraulic pressure within the module. The ACTUATOR solenoid provides hydraulic pressure to the TRANSFER VALVE and to the DETENT SOLENOID. The detent solenoid provides hydraulic pressure to the detent mechanism. Both solenoids are energized at A/P engagement. However, the detent solenoid is delayed slightly from the ACTUATOR solenoid. The solenoids are attached to the module with four bolts. Electrical pins mate with wiring within the module when the units are installed. Hydraulic pressure is powered into the units through ports which align when the solenoids are installed.

### 2- Linear variable displacement transducer (LVDT)

The linear variable displacement transducer provides positional information for the actuator piston and provides an ac output signal in proportion to piston position.

### 3- Pressure regulator

The pressure regulator is in line with the hydraulic passages between the detent solenoid and the detent piston (which locks the actuator piston to the output crank). The regulator bypasses hydraulic fluid to limit the output force (autopilot authority) of the actuator when the unit is backdriven or stalled

# Autopilot Servo Schematic

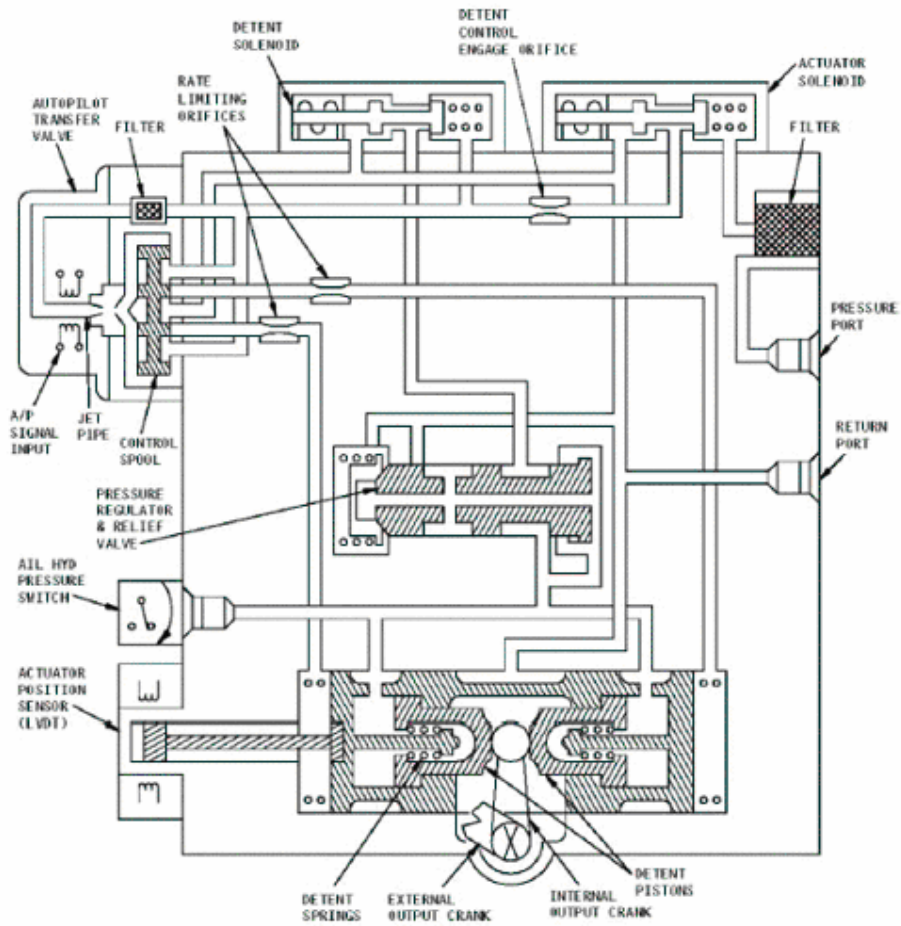


Figure 1.6.4-1 Autopilot Actuator

#### 1.6.4.2 DFCS Modes

Various pitch and roll modes are available and can be manually selected by the flight crew via the MCP. In some cases, automatic mode changes can occur in response to invalid sensor inputs, certain flight conditions, or selection of other compatible modes. During the accident flight, the following modes were used:

##### Take-Off

Flight director guidance during takeoff is initiated by pressing the take-off/go-around (TOGA) switches located on the throttles. In addition to selecting flight director TOGA mode, these switches also signal the autothrottle to advance the throttles to takeoff power. In TOGA mode, the flight director provides pitch and roll guidance to the crew. If TOGA is engaged, no other modes may be selected until an altitude of 400 ft AGL.

##### Level Change

Level Change is an autopilot and flight director pitch mode during climb or descent. In this mode, a fixed thrust level is selected and the autopilot will control the angle of climb or descent to hold the airplane's speed to the value selected in the speed window on the MCP. If the airplane is flying faster than the selected speed, the autopilot will command the airplane to pitch nose up to a steeper climb angle, thus lowering the speed. If the airplane's speed is slower than the selected speed, the autopilot will command the airplane to pitch nose down to a shallower climb angle, which will result in a speed increase. When Level Change mode is selected, "MCP SPD" appears in the pitch section of the flight mode annunciator (FMA) on the EADI. As the airplane nears the selected altitude, the autopilot will automatically transition to altitude acquire ("ALT ACQ" on the MCP) and then altitude hold ("ALT HOLD"). Level Change is available for both autopilot and flight director operation.

##### Heading Select

Heading select is an autopilot and flight director roll mode used to turn to and hold a specific heading. The MCP contains a selected heading window, as well as a bank angle limit selector. The window displays the selected heading, a number from 0 to 359, corresponding to the magnetic heading selected by the crew. The value can be changed by rotating the heading selector knob located immediately below the window. A bank angle limit selector is concentrically located on the same shaft. In Heading Select, the crew can select the bank angle of autopilot turns from 10° to 30° by 5° increments. When heading select mode is engaged, the autopilot will command a turn towards the selected heading. The airplane will bank to the selected bank angle limit and will remain at that limit until the current heading begins to approach the selected heading. As the turn nears completion, the bank angle is reduced until the airplane is flying wings level on the selected heading. The direction of turn is determined to be the shortest turn between the current heading and the selected heading. If the airplane is already in a turn and the selected heading is changed to pass through the reciprocal bearing (greater than 180°), the direction of turn will reverse and the autopilot will seek the shortest turn to reach the selected heading. Heading select is active when "HDG SEL" appears in the roll section of the FMA and is available during both flight director and autopilot operation.

##### Control Wheel Steering - Roll

Control wheel steering roll (CWS R) is a separate autopilot roll mode designed to reduce crew workload. CWS R mode may be manually selected via the CWS pushbutton on the MCP. In this case, flight director modes may be selected via the mode selection push buttons on the MCP. If certain conditions required for other

roll modes are not met or if a certain amount of force is applied to the control wheel, the autopilot mode will automatically change from CMD to CWS R. In CWS R, the autopilot commands the aileron servo to follow the motions of the control wheel. If the pilot releases the control wheel, the autopilot provides aileron commands to hold the current bank angle and thereby continue the commanded turn. However, if the bank angle when the wheel is released exceeds 30°, the autopilot will command a roll back to a bank angle of 30°. If the bank angle when the wheel is released is less than 6°, the autopilot will command wings level and maintain the current heading. CWS R is active when "CWS R" appears in the autopilot section of the FMA. When the autopilot enters CWS R mode, the roll section of the FMA will be blank and the flight director roll command bar disappears. However, other roll flight director modes may subsequently be engaged.

#### MCP Speed

MCP speed is a pitch mode of the autopilot that is used when climbing or descending. In this mode, a fixed thrust level is selected and the autopilot will control the angle of climb or descent in order to hold the airplane's speed to the value selected in the speed window on the MCP. If the airplane is flying faster than the selected speed, the autopilot will command the airplane to pitch nose up to a steeper climb angle, thus lowering the speed. If the airplane's speed is slower than the selected speed, the autopilot will command the airplane to pitch nose down to a shallower climb angle, which will result in a speed increase. MCP speed mode is active when "MCP SPD" appears in the pitch section of the flight mode annunciator (FMA) on the EADI.

Operation of the FD vertical bar with "Heading Select" disengagement as the AP engages.

Refer to Boeing AMM 22-11-00 Page 38

#### 1.6.4.3 Flight Director

The flight director is provided as an aid to the crew during manual flight and as a way for the crew to monitor the operation of the autopilot. The flight director consists of pitch and roll command bars which appears as horizontal and vertical magenta lines on the EADI respectively. When the airplane is following the flight path selected on the MCP, the flight director bars will be centered on the EADI display. If the airplane is flying below the selected path, the horizontal pitch bar will begin to rise on the display, indicating that a nose up command is required to regain the path. As the airplane regains the selected path, the command bar returns to the centered position. Similarly, if the airplane is following the selected roll path, then the vertical roll command bar will be centered. If the airplane deviates to the right of the selected path, the roll command bar will deviate to the left indicating that a bank to the left is required. It should be noted that the flight director roll command bar indicates the additional bank that is required to fly the selected path. For example, with the bank angle limit set to 20 degrees, if the airplane is in a 20 degree right bank as part of a 90 degree right turn, the flight director bar will be centered on the display because the airplane is on the desired path (in this case a 20 degree bank turn). As the turn continues and the airplane approaches the selected heading, the flight director bar will begin to move to the left indicating that the airplane should begin rolling left, out of the turn, and back towards wings level.

## 1.6.5 Engines:

### General:

The airplane is powered by two CFM56-3C1 engines (Serial numbers are: "engine #1" 857 352, "engine #2" 856 481. The engine is a dual rotor axial flow turbofan. The N1 rotor consists of a fan, a three stage booster section connected by a through shaft to a four stage low pressure turbine. The N2 rotor consists of a high pressure compressor and a high pressure turbine. The N1 and N2 rotors are mechanically independent.

The main engine control (MEC) schedules fuel to provide the thrust called for by the forward lever setting. The fuel flow is further refined electronically by the power management control. Thrust is set by positioning the thrust levers. The thrust levers are positioned automatically by the autothrottle system or manually by the flight crew. The forward thrust levers control forward from forward idle to maximum. The reverse thrust control thrust from reverse idle to maximum reverse

Engine indications are displayed on the center instrument panel by the Engine indication System (EIS). N1, EGT, N2, and FF/FU are the primary indications and are displayed as both digital readouts and round dial/ moving pointer indications. N1, EGT, N2 have operating and caution ranges and limits indicated by green and yellow bands and red dials. Oil Pressure and oil temperature indications are displayed with a round dial/moving pointer. Operating and caution ranges and limits are displayed with green and yellow bands and red dials. The oil quantity indicator displays a digital readout of quantity as a percentage of full

The low pressure spool (fan) rotating speed (N1) of the left engine (position 1) does not appear representative of the high pressure spool (core) rotating speed and fuel flow on the DFDR read out; however, the indicated core speed is working as well as the other parameters, which indicate most probably a data recording or read out problem for N1. (refer to Exhibit B FDR Group Factual Report)



## 1.6.6 Airplane Maintenance<sup>5</sup>

### 1.6.6.1 Maintenance Records

#### 1.6.6.1.1 Maintenance Program Summary- Flash Airlines B737-300

Flash Airlines has developed their customized Maintenance Program. The Maintenance Program last revision was issued on January 20, 2003 and approved by the (ECASSA), Airworthiness Central Administration under approval No MOCA/FLASH/737-300/MP/R2/03. This Maintenance Program incorporated guidance from Boeing Maintenance Planning Document (MPD) Revision July 2002.

The Periodic Service Check is accomplished on layover. The check is performed as a walk-around, visual inspection and servicing when necessary.

The Routine Inspection is performed every 250 flight-hours (A Checks). A Routine Inspection Procedures Index is used to assure the check is completed. The Inspection consists of a visual inspection of the aircraft's major components, servicing, operational and functional checks.

#### 1.6.6.1.2 Last Heavy Check

The last "A" check accomplished by Flash Airlines and the last "C" check and Structural inspection carried by Braathens Engineering and Maintenance for the SU-ZCF were as follows:

"8A" Check	:	December 12, 2003	at 25423:50 Flight Hours
"7C" Check	:	From Nov 3 - Dec 21, 2002	at 23531 Flight Hours
Last SI Check	:	From Nov 3 - Dec 21, 2002	at 23531 Flight Hours
Last 15 M Check:		From Nov 3 - Dec 21, 2002	
Last 45 M Check:		From Nov 3 - Dec 21, 2002	

#### 1.6.6.1.3 Repairs and Alterations

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<sup>5</sup> See the Maintenance Records Group Report for full details

#### 1.6.6.1.4 Aircraft Total Hours and Cycles

Total Hours at Time of Accident: 25603 Flight Hours  
Total Cycles at Time of Accident: 17976 Flight Cycles

#### 1.6.6.1.5 Weights and Balance Summary

According to the Egyptian Civil Aviation Regulations, ECAR 91 Appendix H attachment 1 the aircraft has to be reweighed every three years. Furthermore, aircraft must be reweighed if the effect of modifications on the mass and balance is not accurately known. Flash Airlines aircraft was weighed last time on December 19, 2002 in Braathens SAFE, Stavanger, Norway and recalculated by Flash Airlines after the reinforced cockpit door modification installation on November 1<sup>st</sup>, 2003, and the results were as follows.

Empty Weight : 70794 lbs  
Moment : 45921358.6 lb.in  
% AMC : 17.42%

#### 1.6.6.1.6 Engines: CFM56-3C-1

Engines are maintained in accordance with Flash Airlines Maintenance program and are based on the life cycle limits of the rotating components. CFMI Engine maintenance manual together with the applicable Service Bulletins and engine teardown data determine these limits. Overhauls are performed at the SNECMA MOROCCO Workshop or other authorized Certified Repair Station.

	<u>Engine Position 1</u> (Left Side)	<u>Engine Position 2</u> (Right Side)
Serial Number (ESN)	857352	856481
Time Since New (TSN)	25314 hours	26045 hours
Cycles Since New (CSN)	17815 Cycles	17523 Cycles
Date of Installation on SU-ZCF	August 1998	Jan 3, 2003
Time Since Last O/H	8741 Hours	1828 Hours
Cycles Since Last O/H	6188 Cycles	909 Cycles

Engine Disks and First Limiters Status as per attached (refer to exhibit A, Maintenance Records Group Factual Report- attachment 02)

#### 1.6.6.1.7 Engine Monitoring System

Flash Airlines engines are monitored as per the manufacturer (CFMI) engine condition monitoring program (Sage Trend Analysis program). Sage is a set of programs which collectively provide the functionality to perform standard condition monitoring of CFMI engines. Sage is designed to work in an interactive environment with the major analytical calculations performed at scheduled times throughout the day.

By reviewing the engine condition monitoring trend reports for both engines, they showed no deviation or important shift, the EGT margin is considerable ok. Engine

Condition Monitoring cruise trend sheet is attached (refer to exhibit A, Maintenance Records Group Factual Report- attachment 14)

1.6.6.1.8 Flight Data Recorder/ Cockpit Voice Recorder.

Description	P/N	S/N	Test Date	Workshop
Sundstrand FDR	980-4120-DXUN	10069	O/H 18/11/02	Air Transport
Avionic CVR	93A100-80	57994	Tested 12/11/02	Braathens

1.6.6.1.9 Aircraft Status

1.6.6.1.9.1 Minimum Equipment List (MEL)

Flash Airlines Customized Minimum Equipment List CMEL was approved by the ECAA on Feb 23<sup>rd</sup>, 2002

1.6.6.1.9.2 Aircraft Condition Report (A/C deferred defects)

No deferred items were recorded in the aircraft deferred snags log Book

1.6.6.1.9.3 Type Certificate Data Sheet

FAA "Type Certificate Data Sheet" number A16WE (revision 28, dated October 29, 1999) for B737-300 series airplanes was reviewed for compliance conditions and limitations. No discrepancies were noted. Type certificate Data Sheet attached (refer to exhibit A, Maintenance Records Group Factual Report- attachment 15)

1.6.6.1.9.4 Supplemental Type Certificates

Supplemental Type Certificates supplied by Flash Airlines were reviewed. One Supplemental Type Certificate was issued to install a Matsushita Audio Entertainment System in accordance with General Aerospace Engineering Order No GA-23-1042. STC attached (refer to exhibit A, Maintenance Records Group Factual Report- attachment 16)

1.6.6.1.9.5 Airworthiness Directives (AD) Summary and Service Bulletins (SB) Summary

The Airworthiness Directives compliance status list dated January 12<sup>th</sup>, 2004 (attachment 03) submitted by Flash Airlines was reviewed with special concentration on AD's carried out after the aircraft was leased by Flash Airlines.

The previous AD's Status which was forward to Flash Airlines during the aircraft delivery was reviewed with special attention to those AD's which had an open or repetitive status.

All listed Airworthiness Directives and Service Bulletins have been complied with no discrepancies noted.

Service Bulletins compliance status attached ((refer to exhibit A, Maintenance Records Group Factual Report- attachment 17)

1.6.6.1.9.6 Prior Discrepancies/Accidents Involving SU-ZCF

Per Flash Airlines records, no previous accidents were reported for the accident aircraft.

#### 1.6.6.1.9.7 Logbook Forms

- The original aircraft Technical Log Book sheets were reviewed for the last three months from September 27, 2003 through December 2003 for discrepancies, no trends or discrepancies noted.
- Copy of the technical log book sheets listing as well as a list of technical log book entries and relevant corrective actions are attached to “Exhibit A Maintenance Records Group Factual Report”

#### 1.6.6.2 Contracted Repair Station Listing

- EgyptAir Maintenance and Engineering
- Braathens Maintenance and Engineering
- Snecma Morocco Engine Services.

#### 1.6.6.3 Maintenance Performed on the A/C before the accident flight.

##### A Maintenance done by Flash Airlines Tech Staff at Cairo Base

The Last Check carried out on the accident aircraft was an 8A check. The check was performed by Flash Airlines Technical staff at Cairo base station. The check work package included visual inspection, servicing, and operational checks. A routine borescope inspection for the HPT nozzles guides vanes and the combustion chamber was performed on both engines by EgyptAir with no findings. The work package was reviewed with no discrepancies.

##### B Transient Check carried out for the Flight VCE/SSH

A transient check was carried out in VCE by engineer Motaz Awad on January 2<sup>nd</sup>, 2004 a copy of the interview with him is attached

##### C Last PDC carried out for the Accident Flight

On 3 *January 2004*, aircraft SU-ZCF, a daily check was performed in accordance with the approved checklist as per the company maintenance schedule at SSH station just before the flight. The check was carried out by the accident flight on board engineer.

This was reported by incoming engineer

##### D Aircraft refueling before the Accident Flight and investigations done after the accident.

The Refueling was done for the accident aircraft on January 3<sup>rd</sup>, 2004 between 03:50 and 04:00 local time (UTC +2) for the quantity of 3500Liters by truck

no 4432 belonging to Misr Petroleum Company (service invoice is attached) (refer to exhibit A, Maintenance Records Group Factual Report- attachment10)

The same truck had refueled the following airplanes on the same date:

- EgyptAir aircraft A320 SU-GBF at 02:05 LT before the accident aircraft.
- Taroum aircraft YR-GGX at 04:20 LT after the accident aircraft.
- EgyptAir aircraft SU-GCD at 05:10 LT after the accident aircraft.

After the aircraft accident, three fuel samples had been drawn from the Misr Petroleum fuel truck on January 3<sup>rd</sup>, 2004 at 12:45 local time. One of them was used for a dehydrated Copper Sulfate capsule field inspection for fuel water content, which was satisfactory (attachment 11). The two others samples were sent to the following laboratories for analysis:

- The Egyptian Petroleum Research Institute Nasr City, Cairo (refer to exhibit A, Maintenance Records Group Factual Report- attachment 12)
- Misr Petroleum Company, Ghamra Research Center Laboratory (refer to exhibit A, Maintenance Records Group Factual Report- attachment 13)

The Egyptian Petroleum Research Institute (EPRI) performed the Jet (A-1) fuel analysis, ASTM distillation and ASTM D-86. The results of these analyses show that all the values are within limits except for the water content, ppm, which is 48, and the max is 30.

The Misr Petroleum Co, Ghamra Research Center Laboratory performed the same analyses done by (EPRI), all the results comply with the requirements of DES-STAN 91-91 issue 4 (DERD 2494) and the joint fueling systems "Checklist" specifications for JET A-1 issue 19 Sept, 2002.

1.6.6.4. The maintenance log sheets for the flights after 12/31/03

Lost on board and no copies prior to departures from SHH which is a violation of ECAA regulations. Necessary measures are taken by ECAA to ensure adherence.

1.6.6.5. The lack of write-ups on the TOGA problem and slat indication that existed on the entire 25-hours of FDR.

Status of the technical log is not known due to being lost on board.

1.6.7 Weight and Balance:<sup>6</sup>

The Flash Airlines weight and balance calculations provided to the flight crew contained the following information<sup>7</sup>:

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<sup>6</sup> See attached Performance Factual Report

<sup>7</sup> See attached Flash Airlines Load and Trim Sheet.

	Weight (kilograms)	
Total Traffic Load	11,450 <sup>8</sup>	
Dry Operating Mass	33,200	
Actual Zero Fuel Mass	44,650	
Maximum Zero Fuel Mass	47,627	
Takeoff Fuel	7,000	
Actual Takeoff Mass	51,650	
Maximum Takeoff Mass (Certificate Limit)	63,276	
Landing Mass	49,650	
Maximum Landing Mass (Certificate Limit)	51,709	

Zero Fuel Mass Center of Gravity (CG)	20.0%	
Zero Fuel Mass CG Limits <sup>9</sup>	8.0% Forward	28.4% Aft
Takeoff Mass CG	18.0%	
Takeoff Mass CG Limits <sup>10</sup>	6.7% Forward	27.9% Aft

Stabilizer Trim settings for takeoff were:

Flaps 1 or 5	4 $\frac{3}{4}$ Units
Flaps 15	3 $\frac{3}{4}$ Units

According to the Flash Airlines Flight Operations Manual Chapter 6, Paragraph 6.1.8.3, Passenger and Baggage Masses, the following chart was published:

	Male	Female
All flights except	88kg	70kg
Holiday	83kg	69kg
Children	35kg	35kg

<sup>8</sup> A review of the Load and Trim Sheet indicated a low 100-kilogram error. The total cargo weight plus passenger mass (Total Traffic Load) should be 11,550 kilograms. Correspondingly, the Zero Fuel Mass, Takeoff Mass, and Landing Mass will be low in error by the same 100-kilogram Mass.

<sup>9</sup> Estimated Zero Fuel Mass CG limits were derived from Flash Airlines Load and Trim sheet index chart based upon a Zero Fuel Mass of 44,650 kilograms.

<sup>10</sup> Estimated Takeoff Mass CG limits were derived from Flash Airlines Load and Trim sheet index chart based upon a Takeoff Mass of 51,650 kilograms.

A review of the accident Load and Trim Sheet indicated a Passenger Mass of 9,450kg. If 350kg is removed for 10 children (10 x 35kg) the result is 9,100kg. Dividing the 125 adult passengers into the 9,100kg would give an average value of 72.8kg per adult passenger.

Using the table above, and assuming 50% Male and 50% Female adult passengers, the worst-case difference in weight calculation would be the following:

The average weight of male and female for all flights except would be  $88\text{kg} + 70\text{kg} / 2 = 79\text{kg}$  per adult passenger.

$$79\text{kg} \times 125 \text{ passengers} = 9,875\text{kg}$$

This represents an increase in weight of 775kg.

Using this value for Load and Trim calculations provided the following information:

Takeoff CG	18.2%MAC
Zero Fuel Mass CG	20% MAC
Takeoff Trim (flaps 5)	4 <sup>3</sup> / <sub>4</sub> Units

These worst-case differences in values for passenger weight still fall within structural and calculated limitations for the airplane.

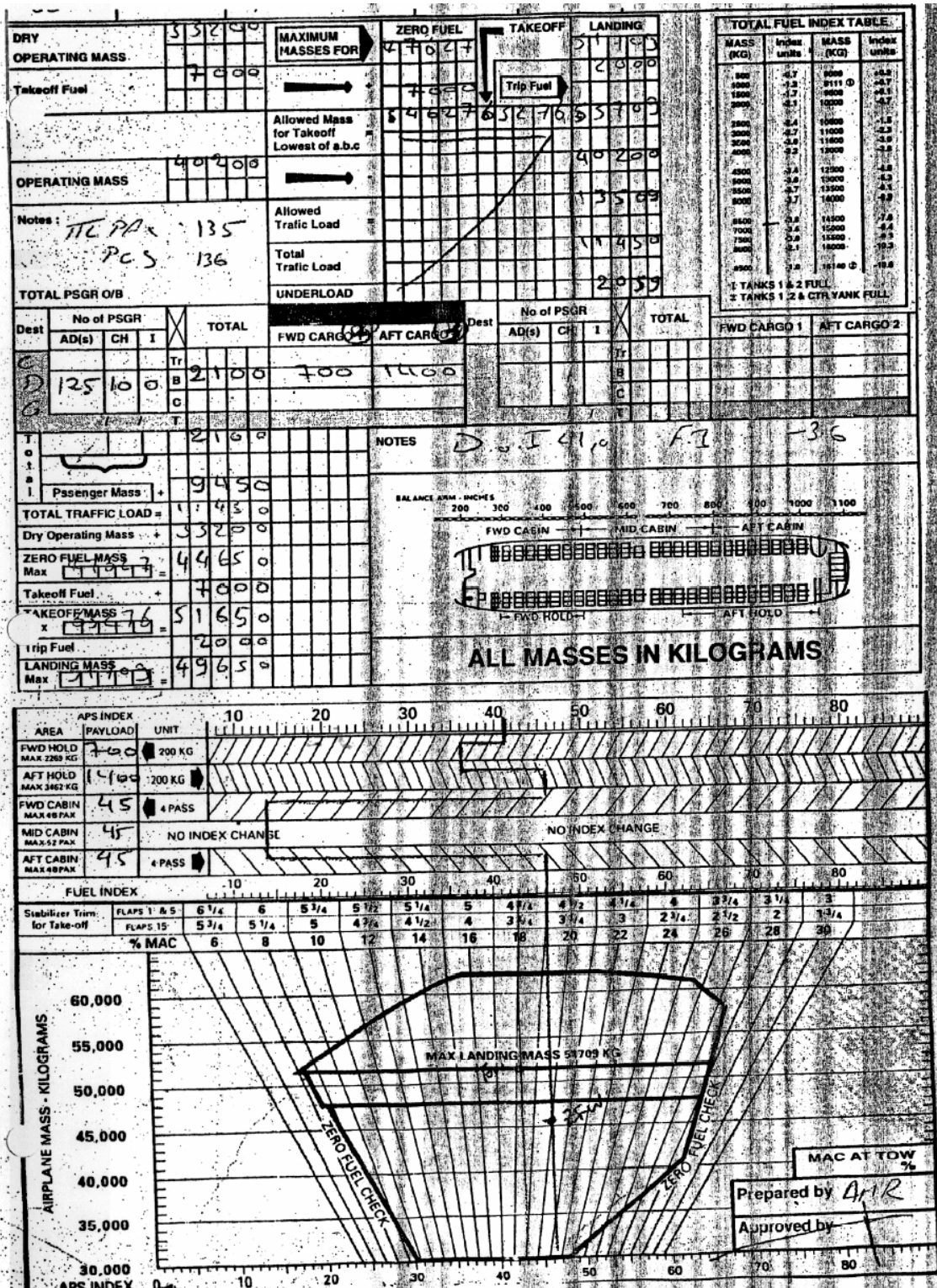


Fig 1.6.5-1 Copy of the Accident Flight Load Sheet



1.7 **Meteorological Information:** <sup>11</sup>

Sharm El Sheikh does not provide Automatic Terminal Information Service (ATIS).

The SSH weather at 0200Z was reported as:

270 degrees at 06 knots, ceiling and visibility OK (CAVOK)<sup>12</sup>, temperature 17 degrees Celsius, dew point minus 6 degree Celsius, altimeter 1011 HectoPascals (hPa), No significant change (NOSIG)<sup>13</sup>.

The SSH weather at 0300Z was reported as:

280 degrees at 08 knots, ceiling and visibility OK (CAVOK) temperature 17 degrees Celsius, dew point minus 6 degree Celsius, altimeter 1011 HectoPascals (hPa), No significant change (NOSIG).

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<sup>11</sup> Refer to exhibit D, Airplane performance Group Factual Report

<sup>12</sup> CAVOK, this terminology means ceiling above 5000 ft and visibility above 10 kilometers.

<sup>13</sup> NOSIG, this terminology means no significant change expected

## 1.8 Aids to Navigation:

### 1.8.1 Maps, charts, etc.

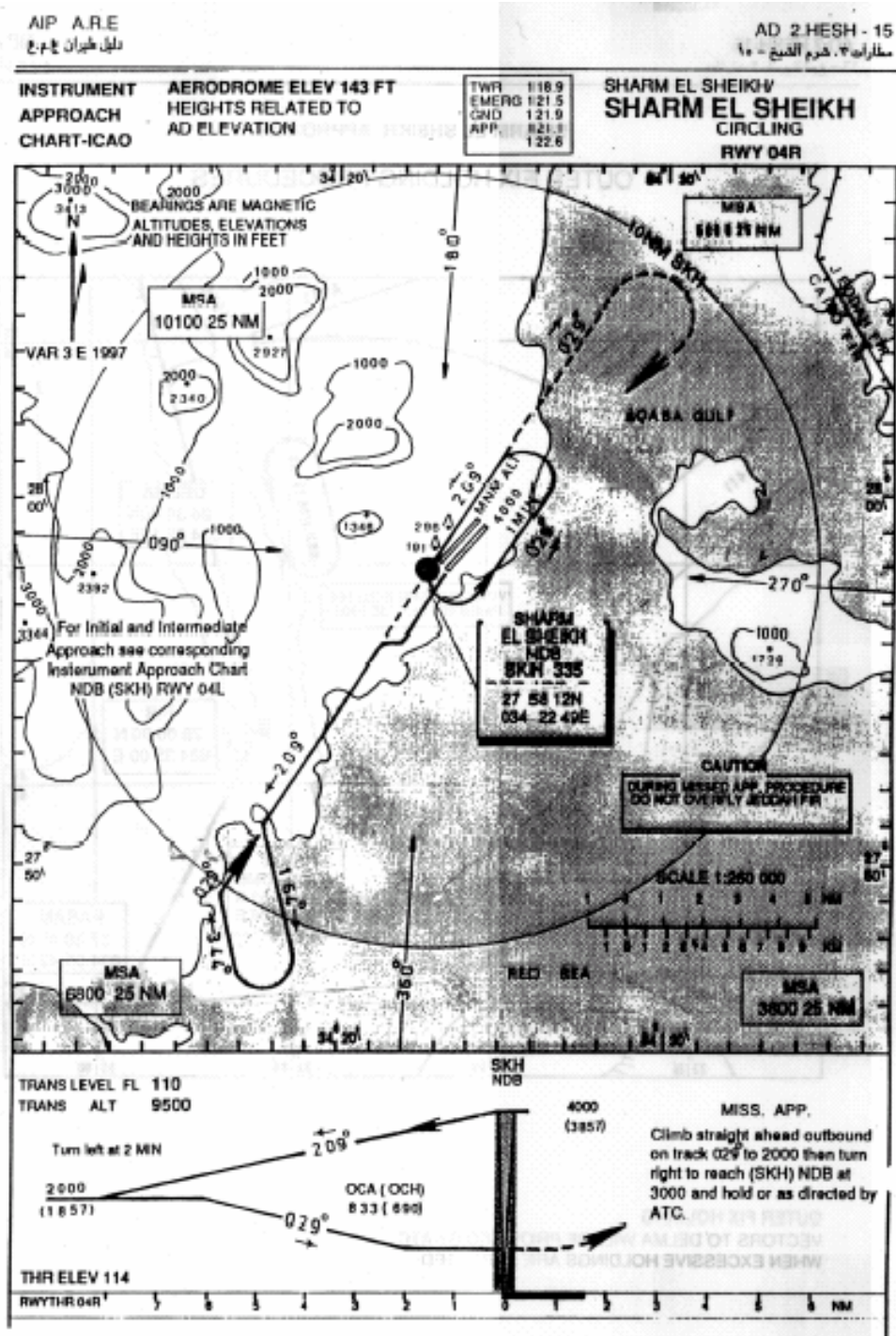


Fig. 1.8.1-1

### SHARM EL SHEIKH Minimum Radar Vectoring Altitude Chart

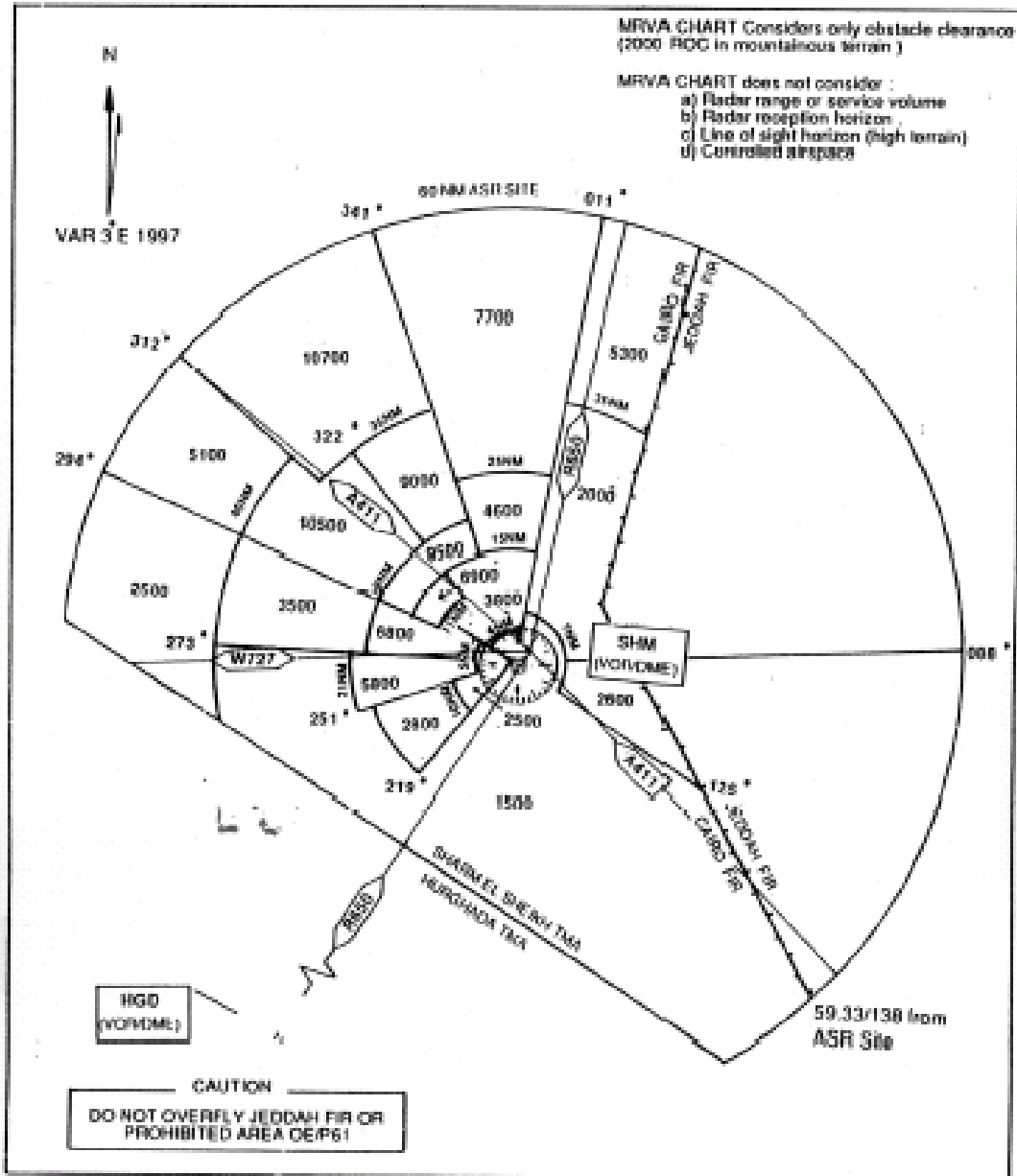


Fig. 1.8.1-2

## 1.8.2 Sharm el-Sheikh Radar<sup>14</sup>

### 1.8.2.1 General Specifications:

ASR 12 Radar (Aircraft Surveillance Radar)

Secondary 250 nm

Primary 60 nm

15 revolution per minute approximately (Scan time = 4.13 sec)

Radar site location: 2758.057n/ 03421.985e (Lat. 27.96762 Degree north, Long. 34.36642

Degree east)

Radar Elevation: 299.3 ft

### 1.8.2.2 Radar data

The radar data from Sharm were reviewed and compared with FDR data to produce flight path

## 1.8.3 Hurgada Radar

### 1.8.3.1 General Specifications:

Radar site location: 2711.546N/03346.814E (Lat. 27.19243333 Degree north, Long. 33.78023 Degree east)

Radar Elevation: 176.344 ft

### 1.8.3.2 Radar data

The radar data from Hurgada were reviewed and compared with FDR to produce flight path

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<sup>14</sup> See attached Performance Factual Report

## 1.9. Communications

### 1.9.1 ATC communications with FSH604 1-Frequency 118.9

Time	Speaker	Content	CVR/FDR time
02:30:00 FSH604	C > P	FSH604 Sharm el Sheikh	02:28:59
	P > C	Go ahead sir	
	C > P	FSH604 copy Cairo MET condition time 02:22(GMT) S/W 210/10 kt VIS 6 Km W Sky clear D 01 QNH 1013	
		Confirm due point please	
	P > C	D 01	
	C > P	Roger Copied next call when ready ان شاء الله يا كابتن	
02:33:43 FSH604	P > C	Check tower FSH604	02:31:55
	C > P	FSH604 go ahead	
	P > C	Our stand destination Cairo request startup clearance	
	C > P	Startup approved QNH 1011 RWY 22R	
	P > C	Startup approved RWY 22R . FSH604 thank you	
02:38:26 FSH604	P > C	Sharm el sheikh FSH604 ready to taxi out	02:36:39
	C > P	04 taxi right D_A hold short 22R	
	P > C	Roger to the right via D_A to holding point 22R. FSH604	
02:39:50 FSH604	C > P	604 ready to copy	02:38:01
	P > C	Go ahead sir	
	C > P	FSH604 destinations Cairo as filed climb initially FL 140 1673 on the squak	
	P > C	Ok destination Cairo via flight plan rout 140 initially 1673 on the squak FSH604 and we have total pax 135 ان شاء الله	
	C > P	135 and confirm SU-ZCF	
	P > C	I do confirm	
	C > P	ان شاء الله continue taxi via "A" , line up 22R . Advice ready for departure	
	P > C	Roger next call ready ان شاء الله	
02:42:25 FSH604	P > C	604ready to departure	02:42:38
	C > P	FSH604 S/W 280/13 Kts left turn to intercept R306 clear for take off 22R	
	P > C	Clear for take off RWY 22R with left turn to establish 306 Sharm VOR our FSH604 clear for take off	
Time	Speaker	Content	CVR/FDR time
02:43:22 FSH604	P > C	FSH604 confirm to the left to establish 306	02:41:35
	C > P	ان شاء الله	

	P > C	And initially 140	
	C > P	ان شاء الله	
	P > C	شكرا	
02:44:49 FSH604	C > P	FSH604 air born time 44 when ready to the left to intercept 306 radial report on course ان شاء الله	02:43:05
	P > C	Roger when ready ان شاء الله left turn to establish 306 Sharm VOR	
02:45:05 MSR227	P > C	Sharm MSR227 السلام عليكم	02:43:19
	C > P	MSR227 go ahead وعليكم السلام ورحمة الله وبركاته	
	P > C	Maintaining FL 120 43 DME inbound to sharm el sheikh and request descent	
	C > P	MSR227 clear SHM VOR visual approach RWY 22R pilot discretion descent 4000 ft. QNH 1011	
	P > C	دلوقتي اد ايه wind هو حضرتك الـ	
	C > P	Indicated 280/10 kts	
	P > C	Right 04 طيب حضرتك ما تشغل RWY 04 يا فندم	
	C > P	straights ILS approach RWY 04L report full establish QNH 1011 مافيش مشاكل يا فندم	
	P > C	Straights approach RWY 04L 1011 next call full establish MSR227	
			End of CVR recording 02:45:06
02:47:45 FSH604	C >	604 position	
02:47:54 FSH604	C >	FSH604 sharm el sheikh	
02:48:06 FSH604	C >	604 sharm el sheikh do you read?	
02:48:17 FSH604	C >	FSH604 sharm el sheikh do you read?	
02:48:28 FSH604	C >	FSH604 sharm el sheikh tower do you read?	
02:48:50 FSH604	C >	FSH604 sharm el sheikh tower do you read?	
02:49:00 FSH604	C >	FSH604 sharm el sheikh tower do you read?	
02:49:08 FSH604	C >	FSH604 sharm el sheikh tower do you read?	
02:50:12 MSR227	C > P	MSR227 could you please to attempt two- way communication with FSH604	
	P > C	حاضر يا فندم	
	C > P	شكرا	
Time	Speaker	Content	CVR/FDR time
	P > P	FSH604 from MSR227	
	P > P	FSH604 from MSR227 how do you read ?	
	P > C	negative contact with FSH604 MSR227 حضرتك	
	C > P	شكرا جزيلاً	
	P > C	عفوا	
02:50:36	C > P	MSR227 insight S/W 290/10 Kts clear to land RWY 04L	

	P > C	Clear to land RWY 04L MSR227	
02:51:02	C >	FSH604 sharm el sheikh do you read ?	
02:51:20	C >	FSH604 sharm el sheikh do you read ?	
02:51:37	C >	FSH604 sharm el sheikh do you read ?	
02:52:02	C >	FSH604 sharm el sheikh do you read ?	
02:52:30	C >	FSH604 sharm el sheikh do you read ?	
02:52:43	C >	FSH604 sharm el sheikh do you read ?	
02:54:23	C >	FSH604 sharm el sheikh do you read ?	
02:54:30	C >	FSH604 sharm el sheikh do you read ?	
02:54:40	C >	FSH604 sharm el sheikh do you read ?	
02:54:45 MSR227	P > C	الفلاش رايح فين ولا جاى منين يافندم ؟	
	C > P	يا كابتن الطائرة طلعت air born واخذت left turn علشان يكسب ارتفاع فوق الميه المفروض كان هو داخلى over head وداخلى على الـ route كنت وقتها حضرتك حوالى 30 ميل او 35 ميل ومن ساعتها مبيرضش عليه	
	P > C	ما تسأل كده نشوف على الرادار باين ولا لا ؟	
	C > P	مش باين فى الرادار فى القاهرة خالص مفيش اى Communication	
	P > C	دخل left turn على الجبال؟	
	C > P	يا كابتن 22R من Left turn	
	P > C	هو مش باين ومفيش اى حد خالص Ok	
	C > P	ان شاء الله Clear to land	
	P > C	Clear to land MSR227	
02:55:47	C >	FSH604 sharm el sheikh do you read ?	
02:56:37	C >	FSH604 sharm el sheikh do you read ?	
02:56:49	C >	FSH604 sharm el sheikh do you read ?	
02:58:15	C > P	MSR227 on ground time 58 to the left via F-A-E stand number 14 report marcheller insight	
	P > C	TO the left F-A-E next call marcheller insight MSR227	
	P > C	Sharm MSR227	
	C > P	اتفضل يا فندم	
	P > C	احنا سمعنا على 121,5 حد من فلاش بيتكلم يعنى مش عارف 604 ولا فيه طيارة ثانية فلاش	
	C > P	هيه 604 مفيش حاجة غيرها خالص	
	P > C	هو كان على 121,5 بيتكلم يعنى ok	
	C > P	شكرا جزيليا يا فندم	
	P > C	عفوا	
	C > P	ان شاء الله Ground 121.9 for company information	
Time	Speaker	Content	CVR/FDR time
	P > C	السلام عليكم 121.9	
	C > P	عليكم السلام	

Information about the conversation between ATC and MSR 227 translated from Arabic into English.

2:58:15 C>P  
P>C  
P>C  
C>P  
P>C

Sharm MSR227

**Go Ahead Sir**

***We heard on frequency 121.5 some one from Flash speaking, I do not know if it is 604 or it is another Flash Aircraft***

C>P

***It is 604, there is no other aircrafts***

P>C

***He was speaking on 121.5, so it is O.K.***

C>P

***Thank you very much Sir***

P>C

***You're welcome***

C>P

Ground 121.9 for company information, ***God willing***

P>C

***Peace be with you 121.9***

C>P

***And with you***

**N.B.** Frequency 121.5 was checked no transmission was recorded at the time of the accident with any traffic



## **1.10. Aerodrome Information**

According to the Aeronautical Information Publication (AIP), Sharm el-Sheikh International Airport is located 23 kilometers northeast of the city. The elevation of the airport is 143 feet mean sea level. The airport had two paved parallel runways; 04L-22R and 04R-22L. Both runways were 3081 meters in length and 45 meters in width. Runways 04R and 04L have CAT 1 Approach Lighting System and runways 22R and 22L had Simple Approach Lighting System. Neither runway had runway centerline lights.

According to the AIP Flight procedures, there were no standard departures and standard arrival routes or any other systematic procedures established within Sharm el-Sheikh approach airspace, heading, flight level, speed and or holding instructions shall be specified in approach control clearances to arriving and departing flights as appropriate to meet the requirements of traffic conditions.

Air Traffic Control Services for Sharm el-Sheikh

An Interview with the Director of Radar Airports, National Air Navigation Service Company indicated that at SSH, the local controller and the departure controller were the same person. The previous last flight departure before the accident flight departed about one hour earlier. An arrival flight landed less than 10 minutes after the accident flight departed. Radar was operating but no radar service was provided to the accident flight.

According to the Director, there were no Standard Instrument Departures (SIDs), or Standard Terminal Arrival Routes (STARs) in Egypt. Clearance was provided to the accident flight crew while on the ground and the departure included a left turn at pilot's discretion and to climb to Flight Level (FL) 140 and to intercept the 306 VOR radial. MEA for this sector is 10500 ft.

According to the Director, the prevailing winds at SSH require the use of runway 04L 70%-80% of the year. On the date of the accident, runway 04L was being used. However, sometime during the day prior to the accident, the runway was changed to 22R.

There was no inspection of the runway after notification of the accident, however, it was stated that the landing airplane after the accident did not report debris on the runway. There is a daily runway inspection performed at SSH.

For AIP information, see attachment

## 1.11. Flight Recorders

### 1.11.1. Flight Data Recorder<sup>15</sup>

The accident airplane's flight data recorder (SSFDR), part number 980-4120-DXUN S/N 10069, was retrieved from the Red Sea on January 16, 2004 by the French Navy. The FDR was immersed in water and sealed in an ice chest and transported to MCA, accident investigation laboratory at Cairo.

- Readout of the FDR was accomplished using the laboratory's playback hardware, Hand held Down Load unit manufactured by ALLIED SIGNAL Part No. 964-0446-001 and recovery/ analysis/ presentation system (RAPS) software.
- In spite of the damage that had occurred to the external case of SSFDR, the internal solid state memory was in good condition and all the available data was retrieved. RAPS considered the recorded signal and data quality to be very good.
- Data plots and tabular listings of each data parameter for the entire accident flight are included in this report as Appendix "exhibit B, FDR Group Factual Report". The entire 25-hour contents of the FDR were also transcribed,

After the cockpit voice recorder (CVR) timing had been compared to the SSFDR vhf microphone keying and Autopilot disengages warning, a time correlation was developed. (refer to exhibit B, FDR Group Factual Report)

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<sup>15</sup> See FDR Group Factual Report

### 1.11.2 Cockpit Voice Recorder<sup>16</sup>

- The accident airplane's Cockpit Voice data recorder (CVR), Fairchild, Part no. 93-A100 – 80, serial no. 57994 was retrieved from the Red Sea on January 17, 2004 by the French Navy. The CVR was immersed in water and sealed in an ice chest and transported to MOCA, accident investigation laboratory at Cairo.
- Readout of the CVR was accomplished using the laboratory's playback hardware and software as follow:

**Download Unit:**

A100 CVR play back Deck - Store 4DS

**Audio Analysis System:**

MPL 1024 , 12 Channel Microphone Mixer – Samson

Filter : PCAP II (Samson)

Amplifier : Samson - Servo-550 Studio Amplifier

**Software:**

Vegas 4 – Sound Forge 6 –PCAP II

- The recorder consisted of four channels of audio information.
  - Channel One: First officer hot mic.
  - Channel Two: Area Mic.
  - Channel Three: Observer hot Mic..
  - Channel Four: Captain hot Mic..
- After the initial retrieved sound task was completed another effort was undertaken with the assistance of BEA expert as follows:
  - The output signal from the tape deck playback machine was too low compared to the recording on the same conditions in BEA. This problem was solved by increasing the output level when the screw of the adjustable gain control was turned clockwise.
  - The sensitivity of the acquisition audio card of the PC was not good enough to capture correctly the audio signal coming from the tape deck player. This problem was solved by changing the value of the "Variable Signal Levels" on the hardware setting of the audio card, from the manufacture value +4 to -10. The gain was increased and the input signal amplified.
  - The speed of the tape was not correct with an interference of the power (115 V, 400 Hz) measured at 375 Hz. It was not possible to adjust properly the speed of the tape with the device installed. This problem is solved by reemulating the wave file with a correct ratio ( $400/375= 1.0665$ ).

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<sup>16</sup> (refer to exhibit C, CVR Group Factual Report)

- Some high frequencies were missing when doing the spectrum analysis. This problem was solved by using a sampling rate of 32000 kHz instead of 22000 kHz.
- The alignment of the head installed on tape deck player was checked, adjusted and was found satisfactory prior to playback the tape.

A new copy of the CVR was performed. This recorded copy is satisfactory.

## 1.12. Wreckage and Impact Information:<sup>17</sup>

### 1.12.1 Scope of Site and Wreckage Group Field Notes

The scope of this report is the recovery operations that took place from 3 January 2004 through 5 February 2004 in the Red Sea off Sharm el-Sheikh, Egypt and initial inspection for the recovered parts. Recovery operations initially consisted of the recovery of floating wreckage elements only. Recovery of the underwater wreckage (including FDR and CVR) began when the first ship equipped with a suitable Remote Operated Vehicle (ROV), arrived at the accident scene on 11 January 2004.

This report provides a summary of the recovery operations and documents the wreckage that was identified and recovered.

### 1.12.2 Recovery Operations

#### Survival aspects

The initial search for possible survivors and the recovery of bodies were priorities for the rescue and investigation teams. Rescue teams were on site minutes after the accident. They searched for survivors but due to the high energy impact of the aircraft with the sea surface, the depth of the water in this area, their efforts were unsuccessful in recovering any survivors.

Efforts were made to locate human remains by use of deep sea cameras and robots but were also not successful due to the location of the wreckage and the depth of more than 1000 meters.

#### Floating Wreckage

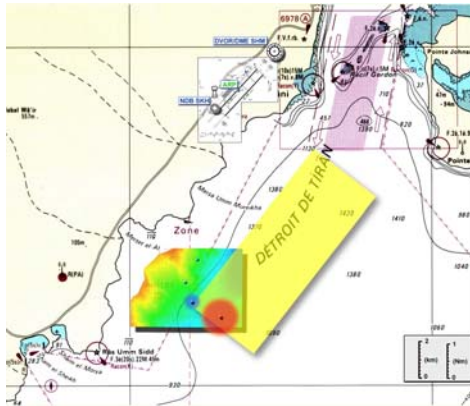


Figure 1.12.4-1 Water depth map

<sup>17</sup> Refer to Exhibit E Site and Wreckage Group Factual Report

The floating wreckage which was recovered shortly after the crash was stored in a hangar in Sharm el-Sheikh airport. On 11 January 2004, the Site and Recovery Group met in the hangar for wreckage inspection. The wreckage was then identified (as much as possible), inspected, segregated (aircraft parts or personal effects). Later, the personal effects were transferred to the Egyptian Legal Authority in Sharm el-Sheikh. A database for the floating wreckage was created (including wreckage pictures).

### **Underwater Wreckage**

Because of the depth of the Red Sea in the area where the accident occurred (approximately 1000 meters), specialized recovery resources were required for the submerged wreckage. The French vessels “Ile de Batz” and “Janus II” were contracted to conduct the underwater wreckage survey and recovery. Both vessels were equipped with deep water recovery capabilities consisting of submersible Remotely Operated Vehicles (ROV). The necessary support equipment to accurately locate and map the airplane wreckage was provided by the French Navy. An oceanographic vessel, the “Beautemps-Beaupré” was sent to the accident site to undertake a bathymetry (depth mapping) of the seabed and a survey of tidal currents.



Figure 1.12.4-2 ROV

### **FDR / CVR Recovery**

The initial focus of the underwater recovery operation was finding and retrieving the protected recorders, the Cockpit Voice Recorder (CVR) and Flight Data Recorder (FDR) and mapping the searched areas. Each recorder is equipped with an acoustic transmitter, called a “pinger” that transmits a detection signal that can be used to locate the box. Based on the initial determination of pinger locations, the ROV from Ile de- Batz, Scorpio, began a visual search using its cameras to find the recorders. To refine the location of the pingers, a network of sonobuoys (GIB, GPS Intelligent Buoys), (see Appendix 5 for detailed description of this operation), was employed in a cooperative effort between the French and Egyptian Navies. This method produced a new pinger position accurate to within 10 meters and the ROV was moved to the new location. A visual search of a grid created around the new pinger location resulted in discovery of the FDR on 16 January 2004.

The FDR was recovered by the ROV and taken onboard the Ile de Batz. Custody of the recorder was transferred to the Investigator in Charge, at the port of Sharm El Sheikh.

The pinger of the second recorder (CVR) was initially identified approximately 800 meters north of the first pinger. However, it was decided to continue the visual search using grids in the area where the first recorder was found. This search was successful and resulted in finding of the CVR on 17 January 2004 (approximately 24 hours after the FDR). It was also taken onboard the Ile de Batz and custody was transferred to the Investigator in Charge at the port of Sharm El Sheikh.

FDR underwater Location: N27 52.3605, E34 22.0165.

CVR underwater Location: N27 52.3467, E34 22.0207.

The recorders were both sent to Cairo for read out and analysis.

The focus of the recovery operation then changed to detailed mapping of the wreckage and recovery of selected airplane equipment. In addition, the recovery operation included recovery of any equipment deemed important to the investigation based on the review of the FDR and CVR in Cairo.

### **Wreckage Mapping**

During the structured search for the recorders, the position (latitude and longitude) and description of surveyed wreckage was recorded. Following recovery of the FDR and CVR, additional grids were defined for ROV operations. These grids were used to systematically survey and document the entire wreckage area. The positions of large pieces, such as the three landing gears and the cores of the two engines were identified.

Data from both ships involved in mapping and recovery were consolidated into a single listing of all surveyed wreckage, which is included herein as Appendix 2.

The distribution of wreckage is included within a rectangle of approximately 275 by 440 meters defined by the following corner point coordinates:

North corner:	N 27°52,559	E 34°21,933
East corner:	N 27°52,410	E 34°22,126
South corner:	N 27°52,294	E 34°22,022
West corner:	N 27°52,450	E 34°21,817

Multiple surveys of the area confirmed the containment of the wreckage within these established boundaries.

### **Recovered Wreckage**

The investigation team developed a strategy for wreckage recovery based on the review of the FDR and CVR undertaken in Cairo. Flight control actuation components and flight deck systems were considered as a priority.

A system was developed for recording the description, external dimensions and the location, in latitude and longitude coordinates, of all recovered wreckage pieces. A database of recovered floating wreckage is included herein as Appendix 3. Another database documenting all wreckage recovered by Ile de Batz and Janus II is included as Appendix 4. Both databases reference digital images of all floating and recovered wreckage.

Recovered wreckage was stored aboard the ships in sea water until taken ashore and loaded onto trucks. All of the recovered wreckage is stored in a hangar at Sharm El Sheikh Airport and is under the control of the investigative authorities.

#### 1.12.3 Partial list of the Recovered Wreckage

- Parts of the horizontal stabilizer central section structure (called "Texas Star"), elements of the elevator structure and components of the elevator control system, including both elevator PCU's (Power Control Unit), both autopilot actuators, the feel and centering unit including the feel actuator.
- Horizontal stabilizer jackscrew and actuator gearbox.
- Vertical stabilizer structure with rudder control system components, including the main rudder PCU and standby rudder PCU, the feel and centering mechanism and with the trim actuator.
- Aileron PCU, spoiler mixer and TBD spoiler actuators.

#### 1.12.4 Initial Observations

- The two engines were found approximately 24 meters apart
- The left and right main landing gear assemblies were found in between the two engines
- The recovered thrust reverser actuator was found retracted
- The recovered leading edge flap actuator was found retracted
- The recovered trailing edge flap jackscrew indicates that flaps were retracted
- The stabilizer jackscrew was measured at 7.5 inches between the flat of the ball nut and the flat of the end stop which corresponds to a stabilizer leading edge position between 2 and 3 degrees down or a trim unit setting between 5 and 6 pilot units.<sup>18</sup>

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<sup>18</sup> B737-300 Aircraft Maintenance Manual 27-41-00



#### 1.12.5 Wreckage Data bases and Photos

The full data base and photos of the wreckage are on a CD, which is available at the Egyptian Civil Aviation Ministry (MCA). This CD contains:

- a. A folder with three Excel files for wreckage complete data base.
  - i. Floating Wreckage data base.
  - ii. Recovered Wreckage data base.
  - iii. Underwater Surveyed Wreckage data base.
  
- b. A folder for photos with four sub-folders
  - i. Floating Wreckage Photos: 104 photos.
  - ii. Recovered Wreckage Photos: 98 photos.
  - iii. Underwater Surveyed Wreckage Photos: 330 photos.
  - iv. Wreckage Recovery Process Photos: 25 photos

### 1.13. Medical and Pathological Information

#### 1.13.1. Egyptian Air Force – Medical Board Report

**From** : Egyptian Air Force – Medical Board  
**To** : Chairman of Civil Aviation Medical Board  
**Subject:** Medical records of RET. AVM Kheider Abdullah Saad

##### 1. Sequence of medical records

- a) Medically fit for all flying duties as from his first medical examination dated 30/05/1970.
- b) Amend to be medically fit for all flying duties to be reexamined every six months as of 14/07/1982.
- c) Amend to be medically fit for all flying duties (remove six months restriction) as of 22/04/1985.
- d) Medically fit for all flying duties until his last medical examination dated 08/01/1997.

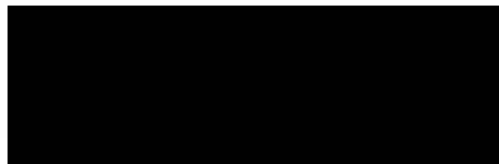
##### 2. Medical History<sup>19</sup>

- a) Admitted to hospital on 06/02/1988, diagnosed (cut wound on left hand) sick leave until 20/02/1988, return to normal duty.
- b) Admitted to hospital on 26/04/1999, released on the same day, diagnosed (effusion left knee).
- c) Examined on 03/11/1999, fit for all flying duties as per last medical exam.

##### During Service A.F. Pilots are subjected to the following:

- a) Tests for Spatial Disorientation as part of his routine periodic physical examination.
- b) Sessions of physiologic training which include:
  - Sudden Decompression.
  - Certificate.
  - Spatial Disorientation Training Chair.

No report was found of any medical factors related to Spatial Disorientation.



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<sup>19</sup> During the time from 1997 to 1999 the Captain held an administrative post (Chief of Staff of an Airforce base) with no flying duties.

1.13.2. Medical factors related to SD (Spatial Disorientation):  
A. FAA advisory Circular regarding SD



U.S. Department  
of Transportation  
Federal Aviation  
Administration

# Advisory Circular

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Subject: PILOT'S SPATIAL DISORIENTATION      Date: 2/9/83      AC No: 60-4A  
Initiated by: AFO-840      Change:

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1. PURPOSE. To acquaint pilots with the hazards of disorientation caused by loss of visual reference with the surface.

2. CANCELLATION. Advisory Circular 60-4, Pilot's Spatial Disorientation, dated February 9, 1965, is canceled.

3. DISCUSSION.

a. The attitude of an aircraft is generally determined by reference to the natural horizon or other visual references with the surface. If neither horizon nor surface references exist, the attitude of an aircraft must be determined by artificial means from the flight instruments. Sight, supported by other senses, allows the pilot to maintain orientation. However, during periods of low visibility, the supporting senses sometimes conflict with what is seen. When this happens, a pilot is particularly vulnerable to disorientation. The degree of disorientation may vary considerably with individual pilots. Spatial disorientation to a pilot means simply the inability to tell which way is "up."

b. During a recent 5-year period, there were almost 500 spatial disorientation accidents in the United States. Tragically, such accidents resulted in fatalities over 90 percent of the time.

c. Tests conducted with qualified instrument pilots indicate that it can take as much as 35 seconds to establish full control by instruments after the loss of visual reference with the surface. When another large group of pilots were asked to identify what types of spatial disorientation incidents they had personally experienced, the five most common illusions reported were: 60 percent had a sensation that one wing was low although wings were level; 45 percent had, on leveling after banking, tended to bank in opposite direction; 39 percent had felt as if straight and level when in a turn; 34 percent had become confused in attempting to mix "contact" and instrument cues; and 29 percent had, on recovery from steep climbing turn, felt to be turning in opposite direction.

d. Surface references and the natural horizon may at times become obscured, although visibility may be above visual flight rule minimums. Lack of natural horizon or surface reference is common on overwater flights, at night, and especially at night in extremely sparsely populated areas, or in low visibility conditions. A sloping cloud formation, an obscured horizon, a dark scene spread with ground lights and stars, and certain geometric patterns of ground lights can provide inaccurate visual information for aligning the aircraft correctly with the actual horizon. The disoriented pilot may place the aircraft in a dangerous attitude. Other factors which contribute to disorientation are

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reflections from outside lights, sunlight shining through clouds, and reflected light from the anticollision rotating beacon.

e. Another condition creating restrictions to both horizontal and vertical visibility is commonly called "white-out." "White-out" is generally caused by fog, haze, or falling snow blending with the snow-covered earth surface which may obscure all outside references. Therefore, the use of flight instruments is essential to maintain proper attitude when encountering any of the elements which may result in spatial disorientation.

#### 4. RECOMMENDED ACTION.

a. You, the pilot, should understand the elements contributing to spatial disorientation so as to prevent loss of aircraft control if these conditions are inadvertently encountered.

b. The following are certain basic steps which should assist materially in preventing spatial disorientation.

(1) Before you fly with less than 3 miles visibility, obtain training and maintain proficiency in aircraft control by reference to instruments.

(2) When flying at night or in reduced visibility, use your flight instruments, in conjunction with visual references.

(3) Maintain night currency if you intend to fly at night. Include cross-country and local operations at different airports.

(4) Study and become familiar with unique geographical conditions in areas in which you intend to operate.

(5) Check weather forecasts before departure, en route, and at destination. Be alert for weather deterioration.

(6) Do not attempt visual flight rule flight when there is a possibility of getting trapped in deteriorating weather.

(7) Rely on instrument indications unless the natural horizon or surface reference is clearly visible.

5. CONCLUSION. You and only you have full knowledge of your limitations. Know these limitations and be guided by them.



KENNETH S. HUNT  
Director of Flight Operations

B- MCA study regarding SD  
Refer to Factual Report, page 55 (Dr. Marawan report) and item 1.16.4.  
Tests and researches conducted by MCA:

C- Medical records for the captain related to any of the conditions conducive to spatial disorientation.  
No report found

1.13.3. Most recent medical certification

A- Date, type

Refer to page 14 of the Factual Report

B- Limitations (if applicable)

None (Refer to page 14 of the Factual Report)

1.13.4. General health information for each crew member.

No Factual information available

1.13.5. Toxicological testing.

No toxicological testing was possible because the bodies were not recovered.

1.13.6. Last civil medical check for Captain

Refer to page 14 of the Factual Report

**1.14. Fire**

N/A

**1.15. Survival Aspects**

Refer to 1.12 Wreckage and Impact Information

## 1.16 Tests and Research

### 1.16.1. Tests and researches conducted by Boeing and Honeywell:

#### General:

A. The FDR records the movements of the pilot's controls (e.g. control column, control wheel position and rudder pedals), the movement of the control surfaces (e.g. elevator, aileron and rudder) as well as motion of the airplane (e.g. pitch and roll attitude and heading angle). The performance evaluation was conducted to determine if the control surfaces were responding normally to the pilot's controls and if the airplane was responding normally to movement of the control surfaces.

In order to accomplish this work, Boeing's 737-300 aerodynamic simulation model was used to recreate the accident flight. The simulation calculates the response of the airplane to movement of the flight control surfaces – for example, it can calculate the roll rate resulting from a 10 degree deflection of the ailerons. The simulation has been verified by comparison against actual flight test data and was used for the design and certification of the 737-300 airplane. In addition, the simulation is the basis for 737-300 crew training simulators used around the world. It should be noted that the 737-300 simulation model is essentially a computer program that represents a nominal airplane with nominal engines. Small differences between the simulation and individual airplane's behavior are common and expected due to differences in control surface rigging, engine wear, and other normal tolerances.

#### *B. Performance Evaluation*

FDR data are recorded at relatively low sample rates and are recorded from different sources, some of which have inherent biases. Because of these issues, a kinematic consistency (KINCON) process was used to supplement the FDR data and calculate additional parameters to be used in the performance analysis. Kinematic consistency analysis is a general practice for processing flight data (either flight test data or FDR data) to ensure consistency of position, speed, and acceleration data.

#### *C. Baseline Simulation*

A baseline simulation recreation of the accident flight was started just as the airplane turned onto the runway and the throttles were advanced, and the simulation was stopped at the end of the FDR data. Because the simulation can calculate the response of the airplane to control inputs, a set of control input time histories (column, wheel, and rudder movements) can be determined that results in the simulation following the same path as the accident airplane. It is important to note that this process does not use the control or surface position data recorded on the FDR, only the path information (e.g. accelerations, attitude and altitude).

Comparisons between the recorded FDR data and the simulation time history data are provided for longitudinal and lateral/directional data in Figures Figure 1.16.2-1 and Figure 1.16.2-2 respectively.



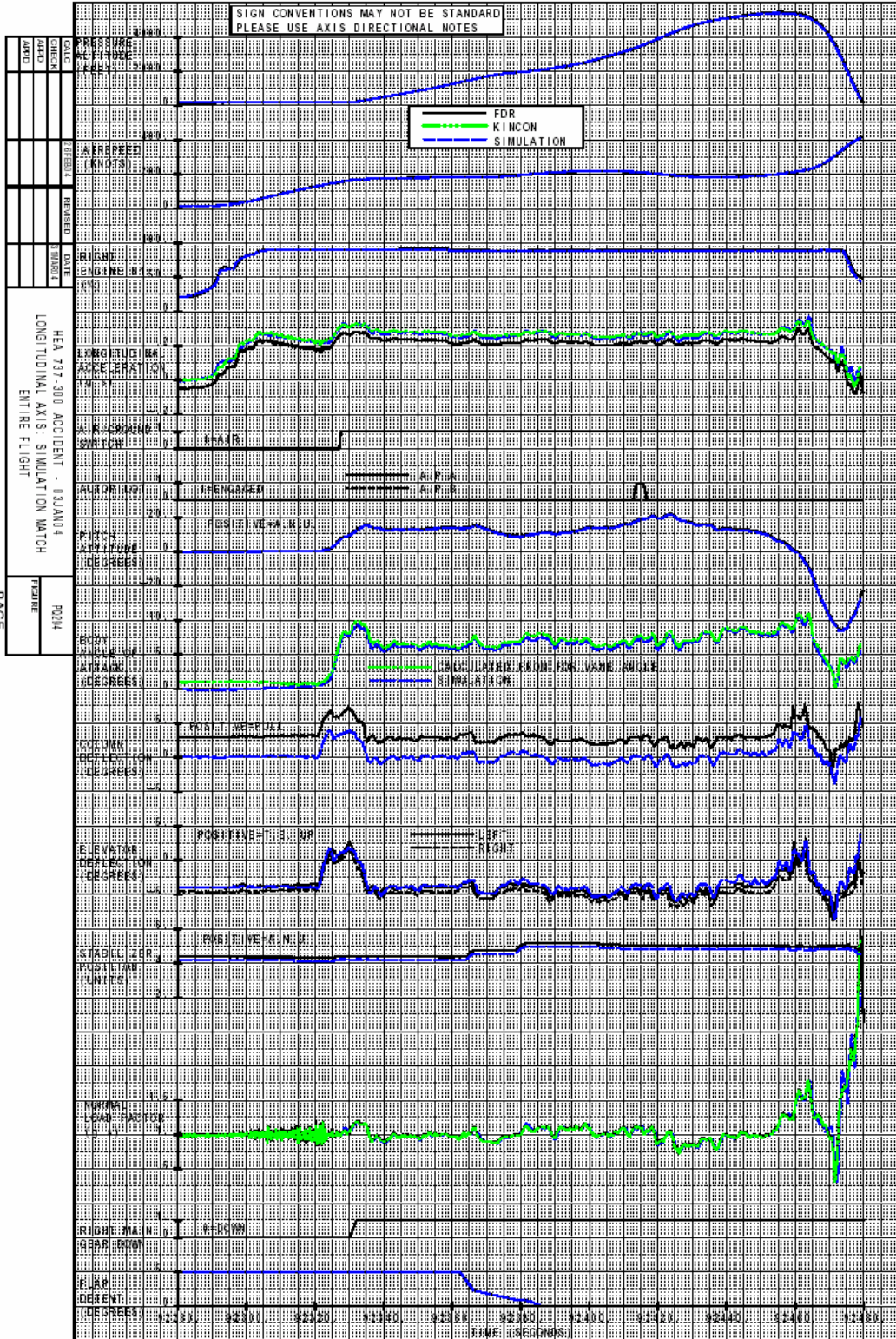


Figure 1.16.2-1 – FDR and Simulation Match Data – Longitudinal Axis

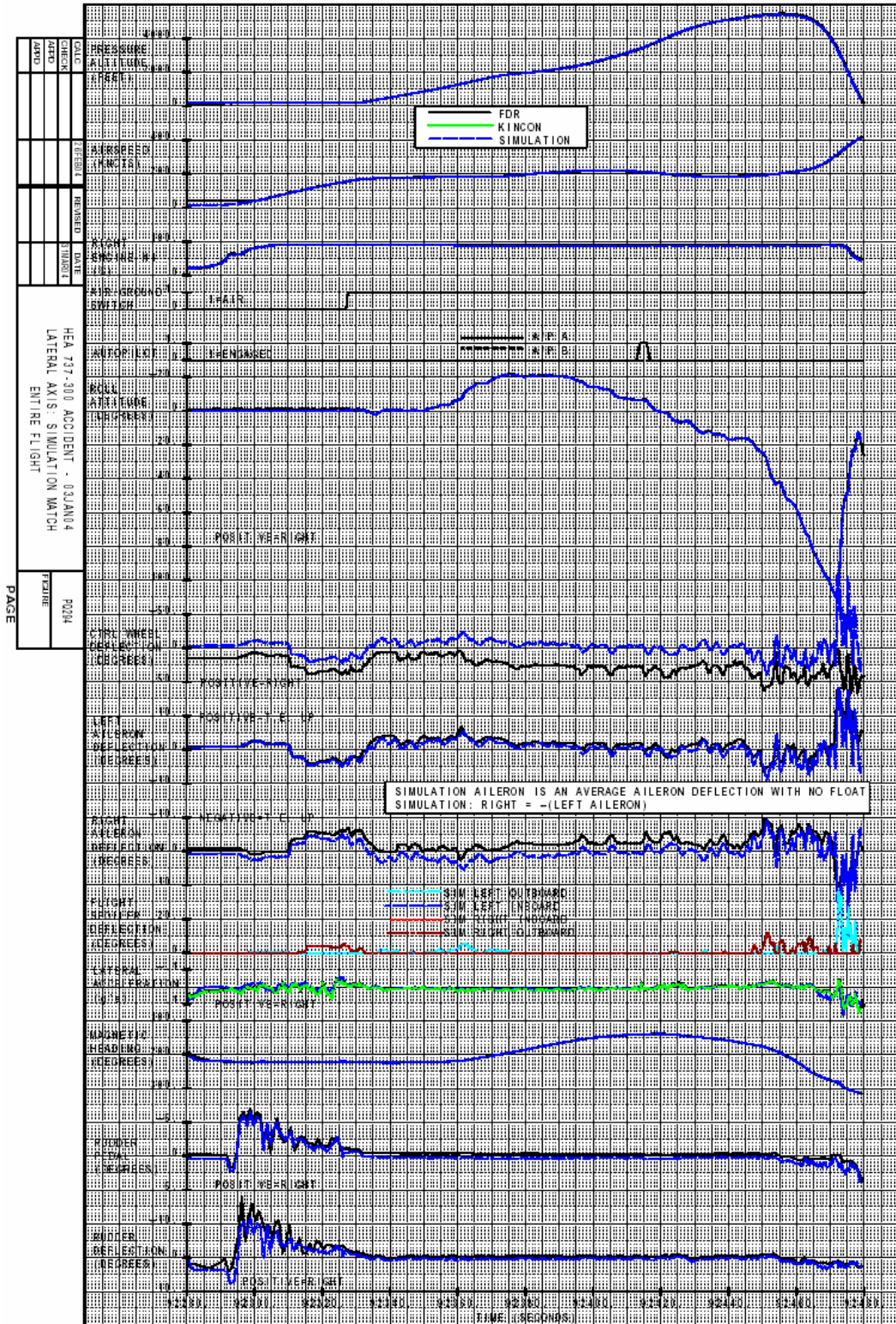


Figure 1.16.2-2 – FDR and Simulation Match Data – Lateral/Directional Axis

An examination of the baseline simulation revealed that the path of the accident airplane is consistent with the recorded motion of the control surfaces. Specifically, the extreme bank attitude that occurs towards the end of the flight is consistent with recorded motion of the ailerons.

The simulation also revealed that the motion of the control surfaces is consistent with the recorded motion of the control inputs, with the exception of control wheel

#### *D. Hypothetical Faults resulting in a rolling moment*

Several hypothetical airplane system faults were examined to determine if any could have resulted in the right roll behavior recorded on the FDR. These faults included:

- Uncommanded deployment of the #1 slat
- Uncommanded spoiler deflection to full travel (hardover)
- A spoiler disconnected from its actuator (spoiler float)
- Flap asymmetry
- Thrust asymmetry
- Unrecorded rudder motion

The hypothetical faults listed above are similar in that they each create a rolling moment unrelated to the position of the ailerons that will cause the airplane to bank. That is to say, if one of these faults had occurred, the path of the airplane would have differed from that predicted by the recorded position of the ailerons.

#### *E. Multi-Purpose Engineering Cab Simulator*

Additional tests were conducted at Boeing's multi-purpose engineering cab simulator or M-Cab. The M-Cab is similar to a flight crew training simulator in that it consists of a realistic flight deck mounted on a movable base. The M-Cab includes a visual system providing out-the-window views to the flight crew. Because the M-Cab is used to simulate the flight deck of many different Boeing models, actual flight instruments are not used. Instead, a large LCD display is programmed to simulate the flight instrument displays. Examples of the M-Cab's flight instrument displays for the 737-300 are shown in section 1.6.2.

Major differences between the M-Cab and a typical flight crew training simulator are listed below.

- The M-Cab can simulate different model airplanes including 707, 727, 737, 747, 757, 767, and 777.
- The M-Cab can be reprogrammed to simulate a wide variety of hypothetical aircraft system faults.
- The M-Cab can be "backdriven" to reproduce recorded data, such as the simulation match to the accident flight discussed in section 1.16.2. In addition, the backdrive can be interrupted at any point with a transition to normal simulator operation at the current flight conditions. This capability (known as "breakout" allows pilots in the simulator to attempt to recover the airplane from various points in the accident profile.
- The operation of the M-Cab is recorded at a high sample rate

The M-Cab was used to recreate the accident flight as well as to study a number of hypothetical airplane system faults.

#### *F. Tests conducted in the M-Cab*

The M-Cab was used to examine some of the faults mentioned above (item D), as well as a number of other hypothetical faults affecting the lateral control system or the autopilot system. M-Cab tests included:

- Backdrive of FDR data

- Backdrive with breakout at 02:44:44
- Backdrive with breakout at 02:44:56
- Spoiler float
- Uncommanded aileron trim to full authority
- Uncommanded aileron trim to half authority
- Autopilot servo actuator hardover without force limiter engaged
- Autopilot servo actuator hardover with force limiter engaged
- Autopilot servo actuator hardover with pressure regulator and relief valve inoperative

The spoiler control drum jam and control wheel shaft jam scenarios were accomplished by "background" simulation analysis.

The tests in the M-Cab were conducted with an out-the-window scene equivalent to that available to the accident pilots with the following exceptions:

- 1) The visibility conditions simulated (ceiling and visibility unlimited at night with no moon) were those reported at the airport at the time of the accident. Actual visibility conditions on the flight deck at the time of the accident are unknown.
- 2) The ground in the vicinity of Sharm el-Sheikh was depicted through the use of satellite photography taken during daylight hours. It did not represent the nighttime scene of street lights, building lights, etc. against an otherwise dark landscape.

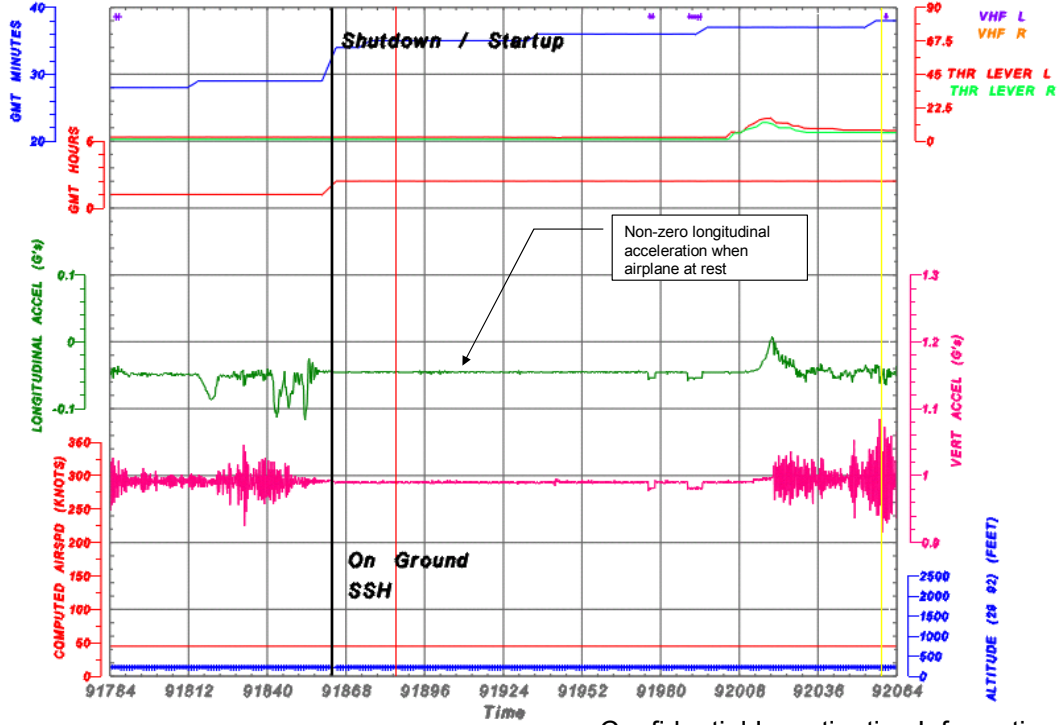
## FDR Data

- Accelerations and Euler angles recorded on the FDR uniquely determine the path of the airplane
- Accelerations
  - Vertical
  - Longitudinal
  - Lateral
- Euler angles
  - Pitch
  - Roll
  - Heading
- Additional parameters describe path
  - e.g. altitude, ground speed, drift angle

# Problem

- Some FDR data may be inconsistent with other FDR data
- Example:
  - Integrating longitudinal acceleration during a takeoff roll results in groundspeed. The calculated value may differ from the recorded value.
- Solution:
  - Add an offset to the acceleration such that the calculated groundspeed matches the recorded groundspeed.

737-300 SU-ZCF



Confidential Investigative Information

# Kinematic Consistency

- Kinematic consistency is a process that adds a bias to the recorded accelerations so that the integrated path matches the recorded path

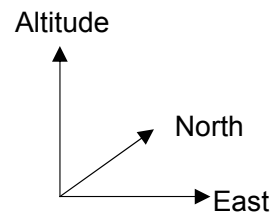
- i.e. calculate  $c_1$  such that

$$v = \int (a + c_1) dt$$

*where*

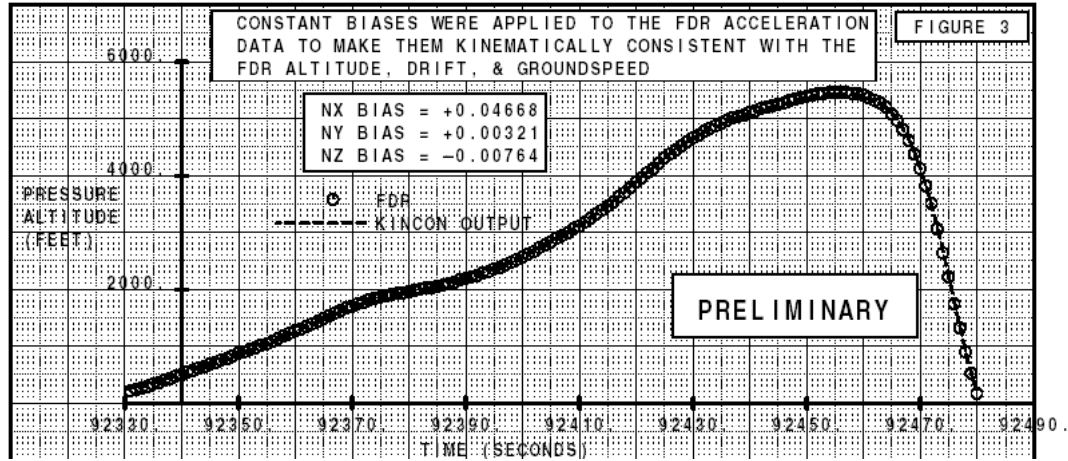
$v = \textit{groundspeed}$

$a = \textit{longitudinal acceleration}$



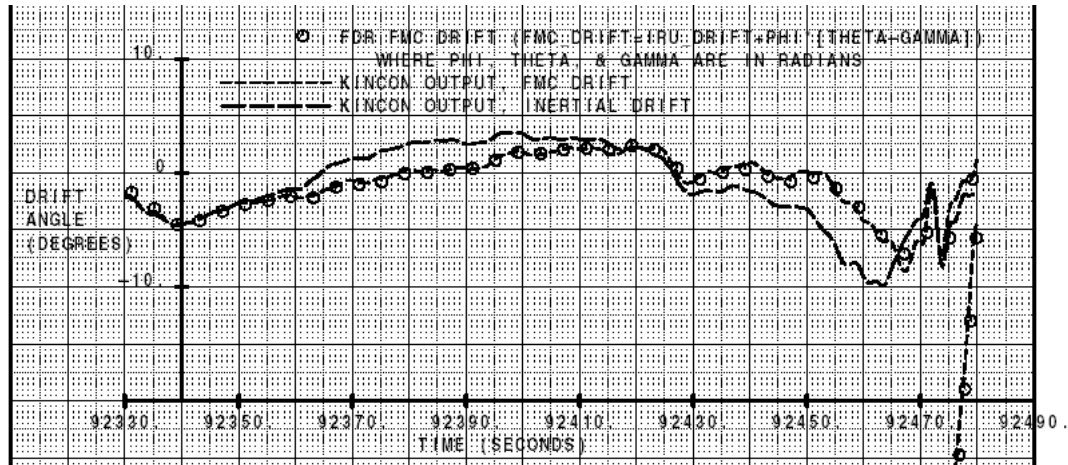


# Kinematic Consistency Results



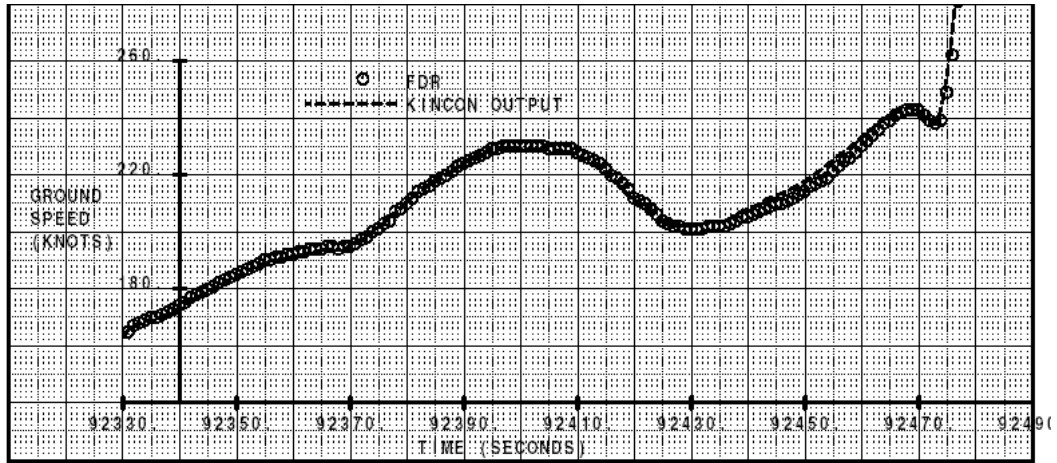
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# Kinematic Consistency Results



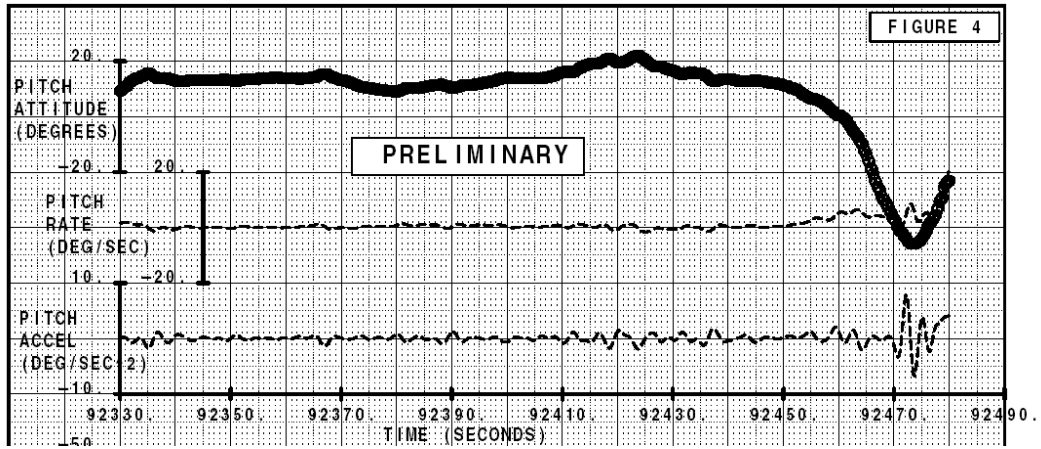
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# Kinematic Consistency Results



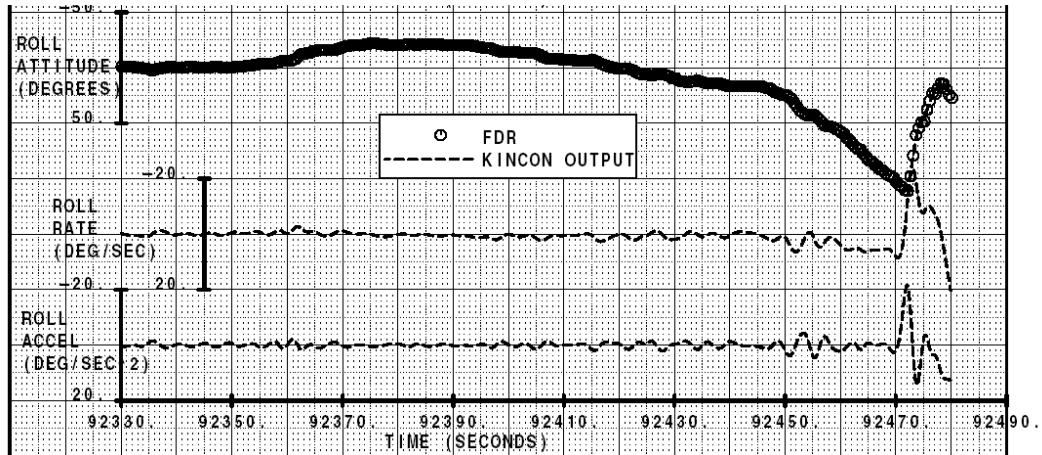
Confidential Investigative Information

# Kinematic Consistency Results



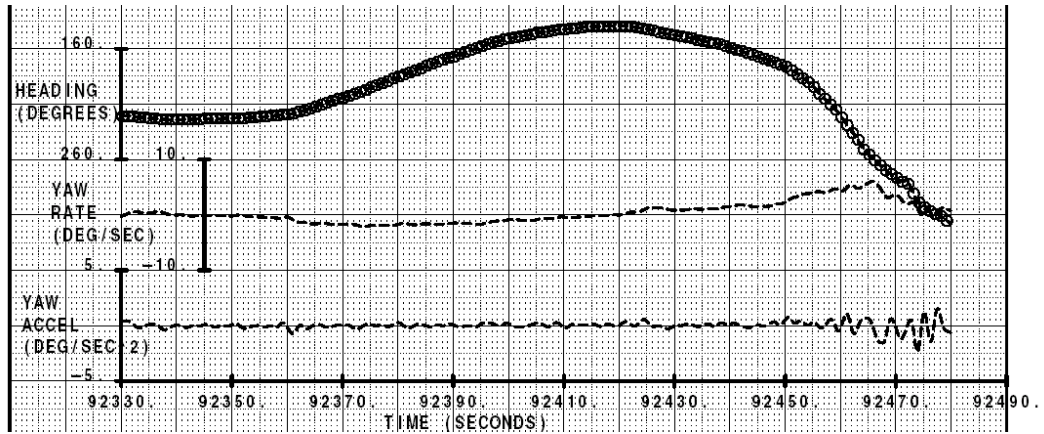
Confidential Investigative Information

# Kinematic Consistency Results



Confidential Investigative Information

# Kinematic Consistency Results



Confidential Investigative Information

# Kinematic Consistency

- Note:
- The kinematic consistency process does not make any assumptions about the aerodynamic properties of the airplane
- In fact, the process can be applied to any moving object

# Simulation

- Once the kinematically consistent accelerations and Euler angles have been calculated, an aerodynamic simulation of the airplane is used to reconstruct the flight path
- Time-step integration is used to calculate the motion of the airplane from one step to the next

$$v_{t1} = v_{t0} + a_{t0} \Delta t \quad x_{t1} = x_{t0} + v_{t0} \Delta t$$

$$Lift = \frac{1}{2} \rho v^2 S C_L$$

$$C_L = f(\alpha, v, flaps, gear, control surfaces, \dots)$$



# Sensitivity Example

- Accident flight is approximately 147 seconds long
- Simulator match of altitude differs by approximately 200 feet
- Sensitivity analysis for straight and level flight 147 seconds long

$$F = MA \text{ or } A = \frac{F}{M}$$

$$\text{For vertical axis } \ddot{z} = \frac{L-W}{W} \longrightarrow z = \iint \frac{L-W}{W} dt^2$$

$$\text{For constant weight } z = g \frac{L-W}{W} \frac{t^2}{2} \Big|_{t_1}^{t_2}$$

# Sensitivity Example

For constant weight  $z = g \frac{L-W}{W} \frac{t^2}{2} \Big|_{t_1}^{t_2}$

Assume altitude error is result of incorrect lift  $\Delta z = g \Delta \frac{L-W}{W} \frac{t^2}{2}$

Solve for  $\Delta L$   $\Delta \frac{L-W}{W} = \frac{2\Delta z}{g t^2}$   $\Delta L = \frac{2W\Delta z}{g t^2}$

$$\Delta L = \frac{2(113630 \text{ lb})(200 \text{ ft})}{32.2 \frac{\text{ft}}{\text{sec}^2} (147 \text{ sec})^2} = 65 \text{ lbs}$$

Therefore-

A 65 lb error in calculated lift will result in a altitude error of 200 ft after 147 seconds.

# Simulation Differences

The 737-300 simulation model represents a nominal airplane with nominal engines.

Small offsets between the nominal simulation airplane and an individual airplane in the fleet are common due to differences in rigging, engine wear, etc.

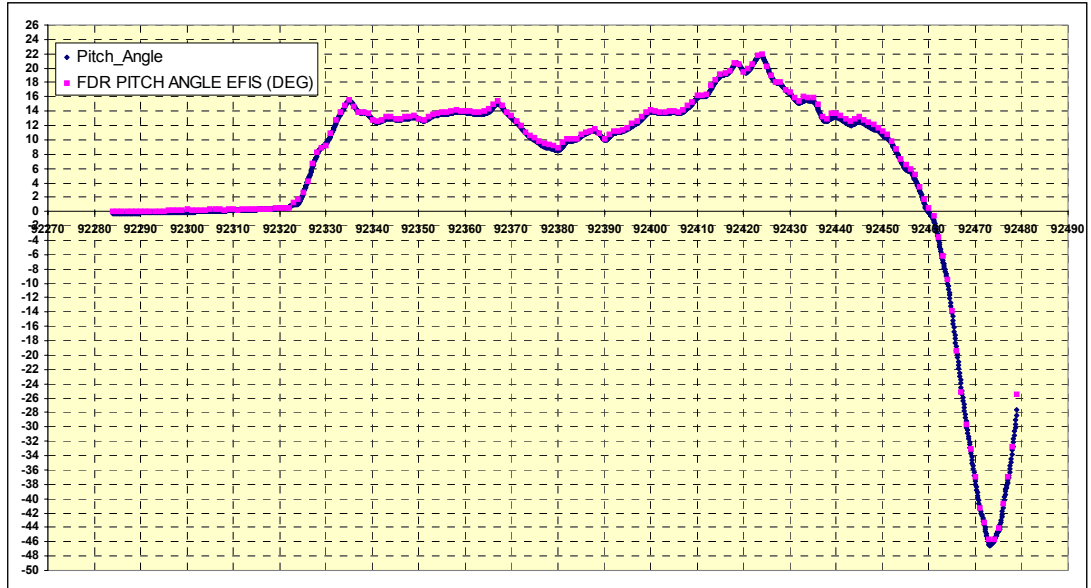
# Pass Through Data

For Flash Airlines simulation –

- Stabilizer was adjusted to account for control column bias (2.9° offset)
- Throttle level position was adjusted to improve match of airspeed and altitude

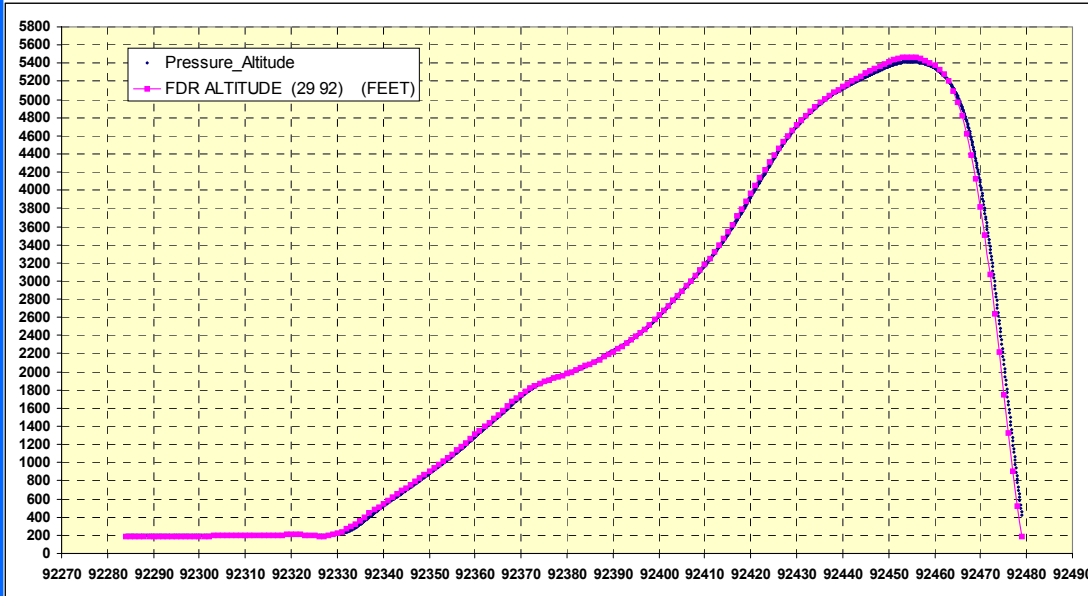
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# Kincon Data Match



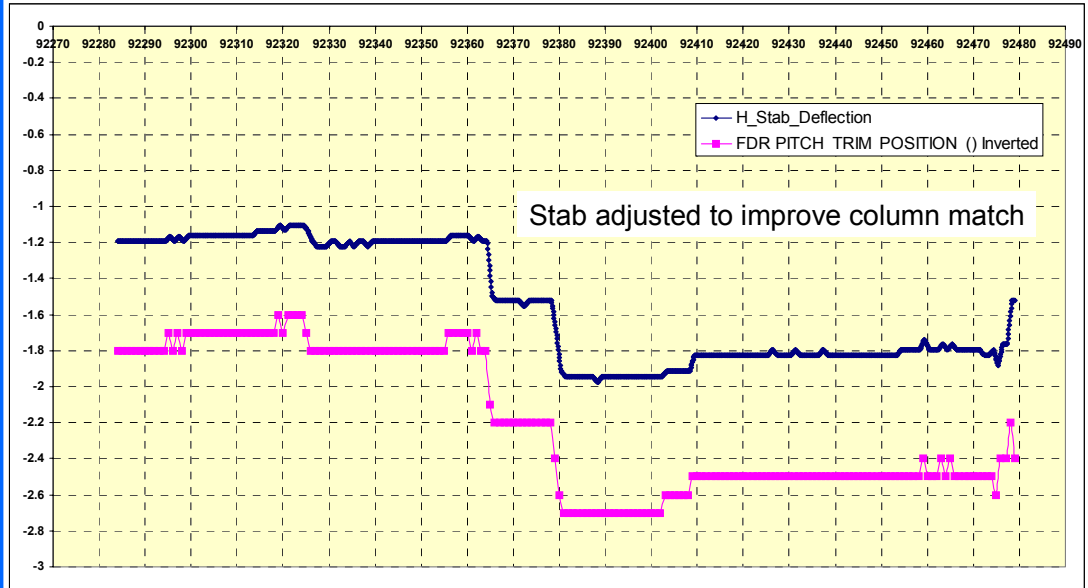
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# Simulator Output Match



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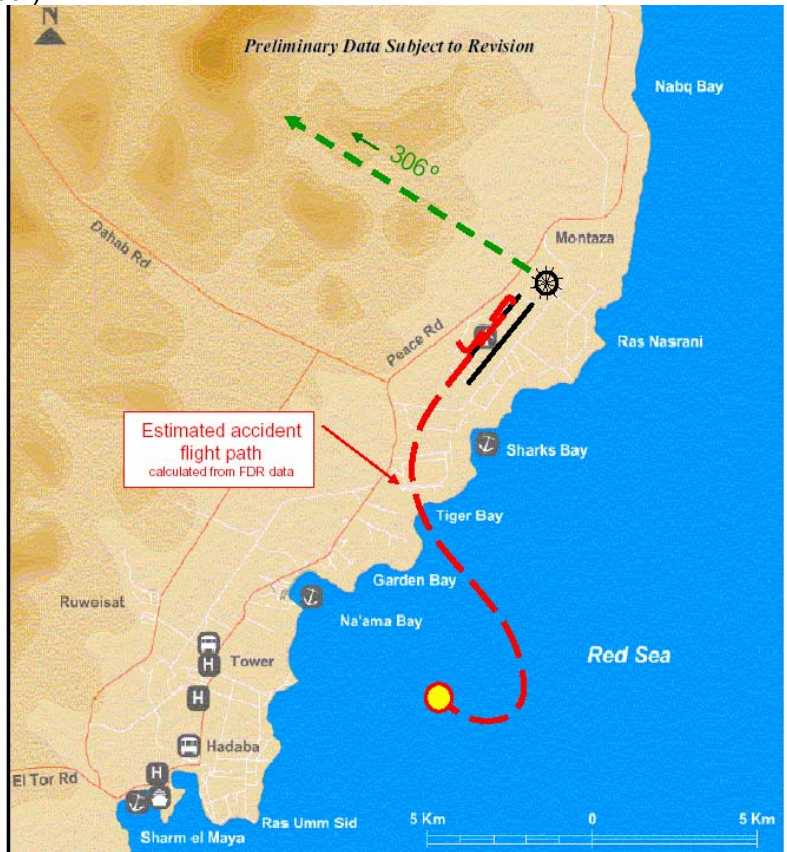
# Pass Through Data Match



Confidential Investigative Information

1.16.1.1. **Estimated accident flight path, calculated from FDR data:**  
(FlightPathMap.pdf)

# Airplane Flight Path

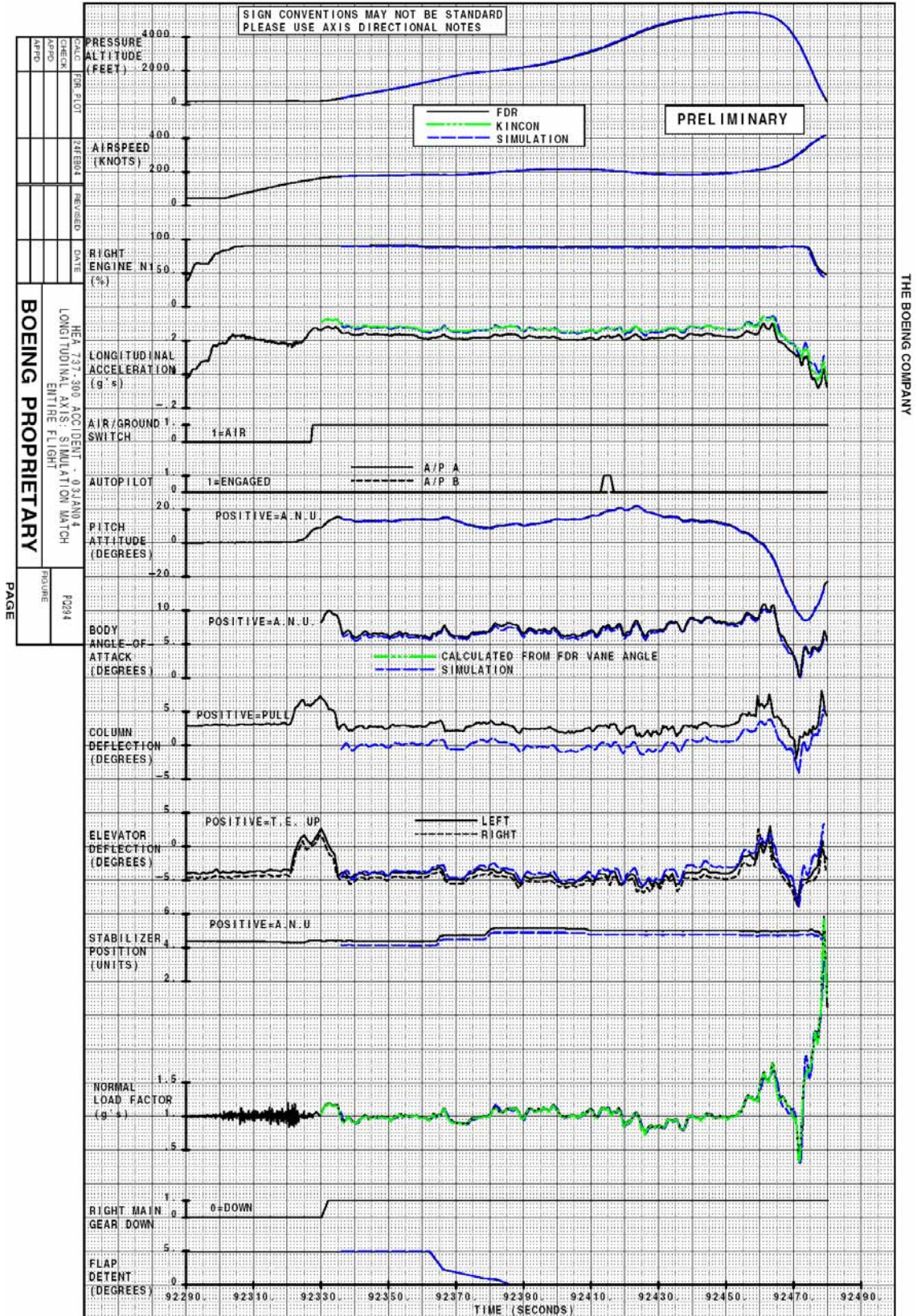


Boeing Proprietary

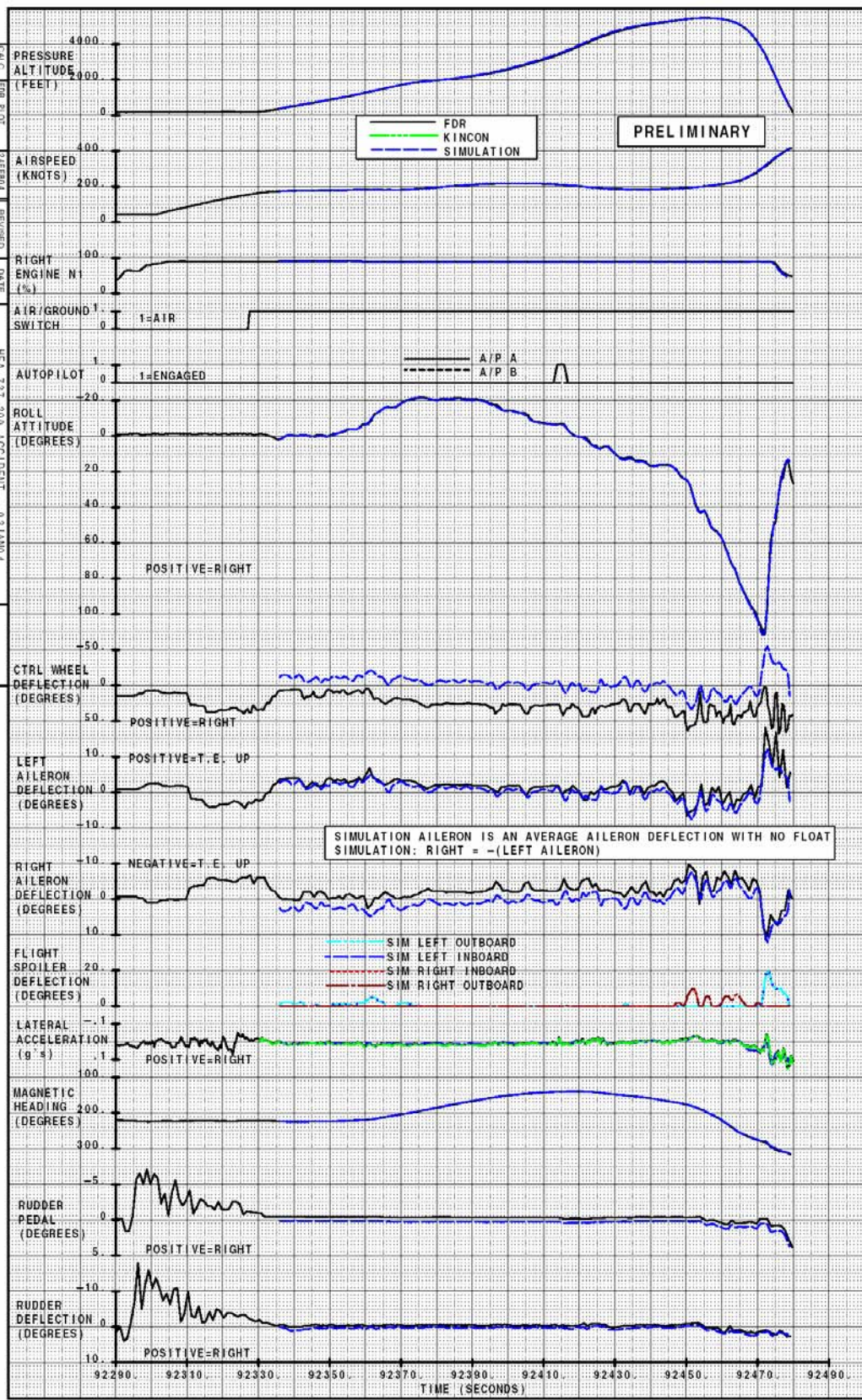


1.16.1.2. NA

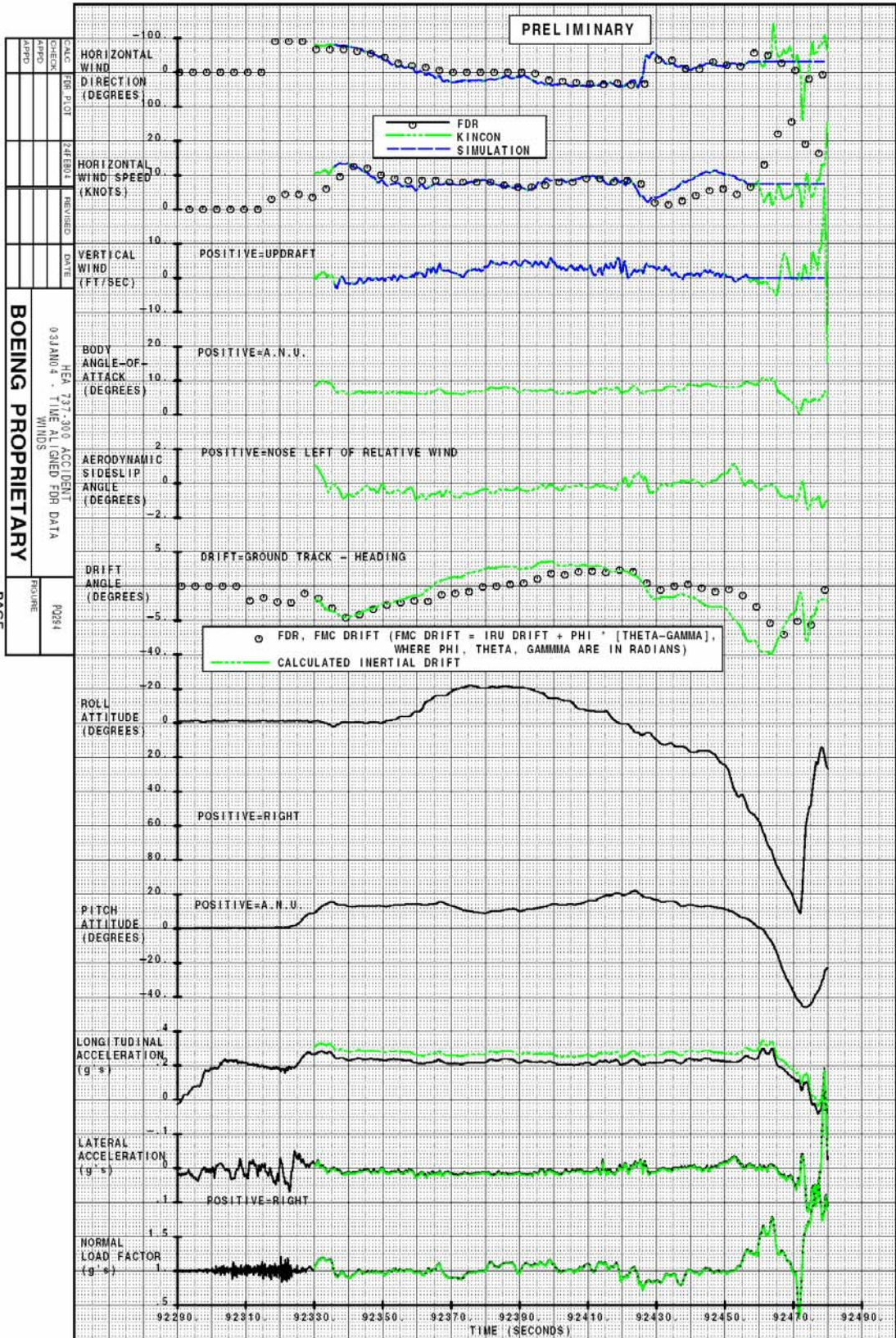
1.16.1.3. Simulator Match accident flight:  
 SimMatchaccidentflight 24-2-04.pdf (Simulation Match, FDR-Kincon-Simulation)



CALC	FOR PILOT	REQUIRED	DATE	HEA 737-300 ACCIDENT - 03JAN04	FIGURE
CHECK				LATERAL AXIS SIMULATION MATCH	10084
APPRO				ENTIRE FLIGHT	
<b>BOEING PROPRIETARY</b>					PAGE

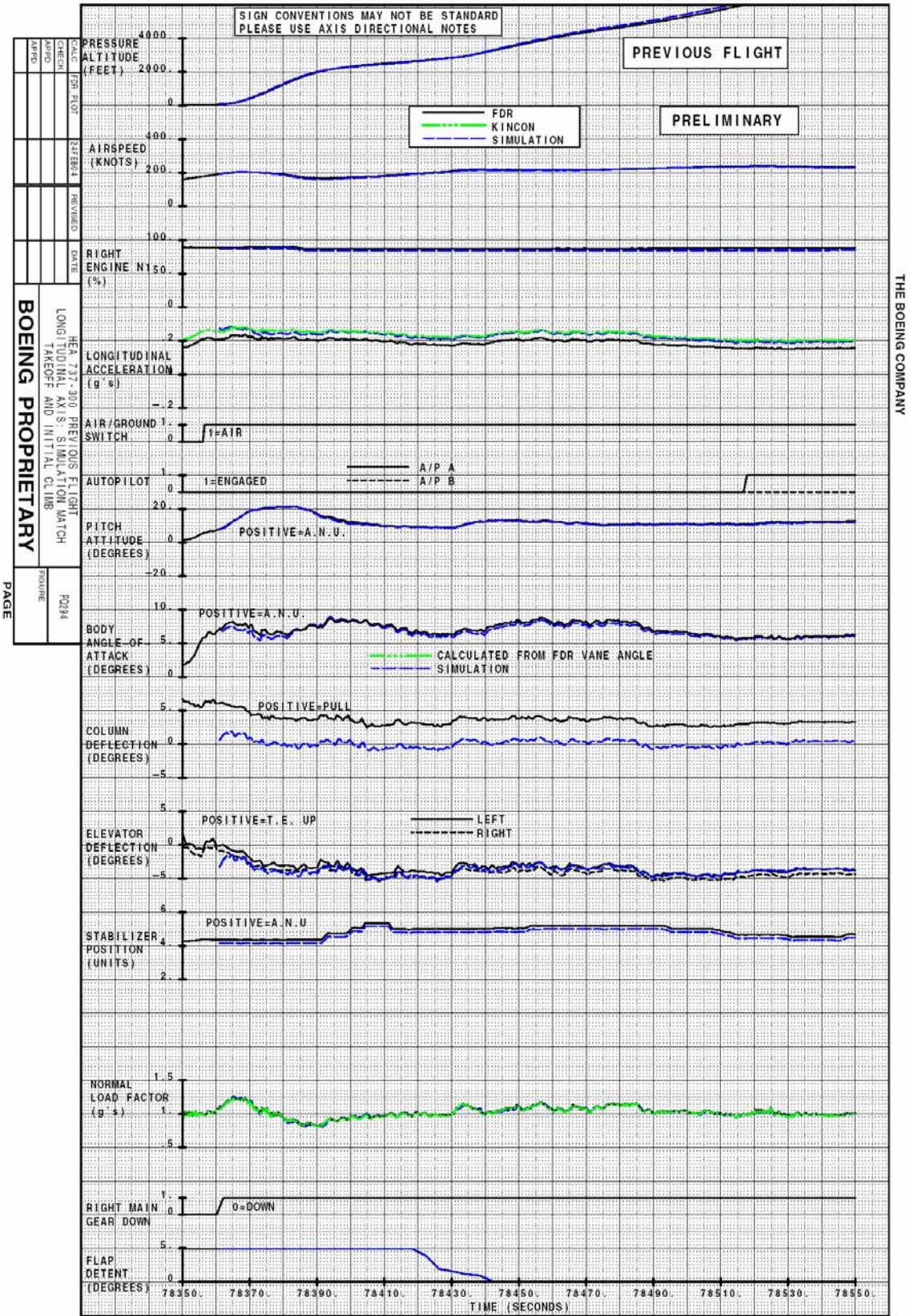


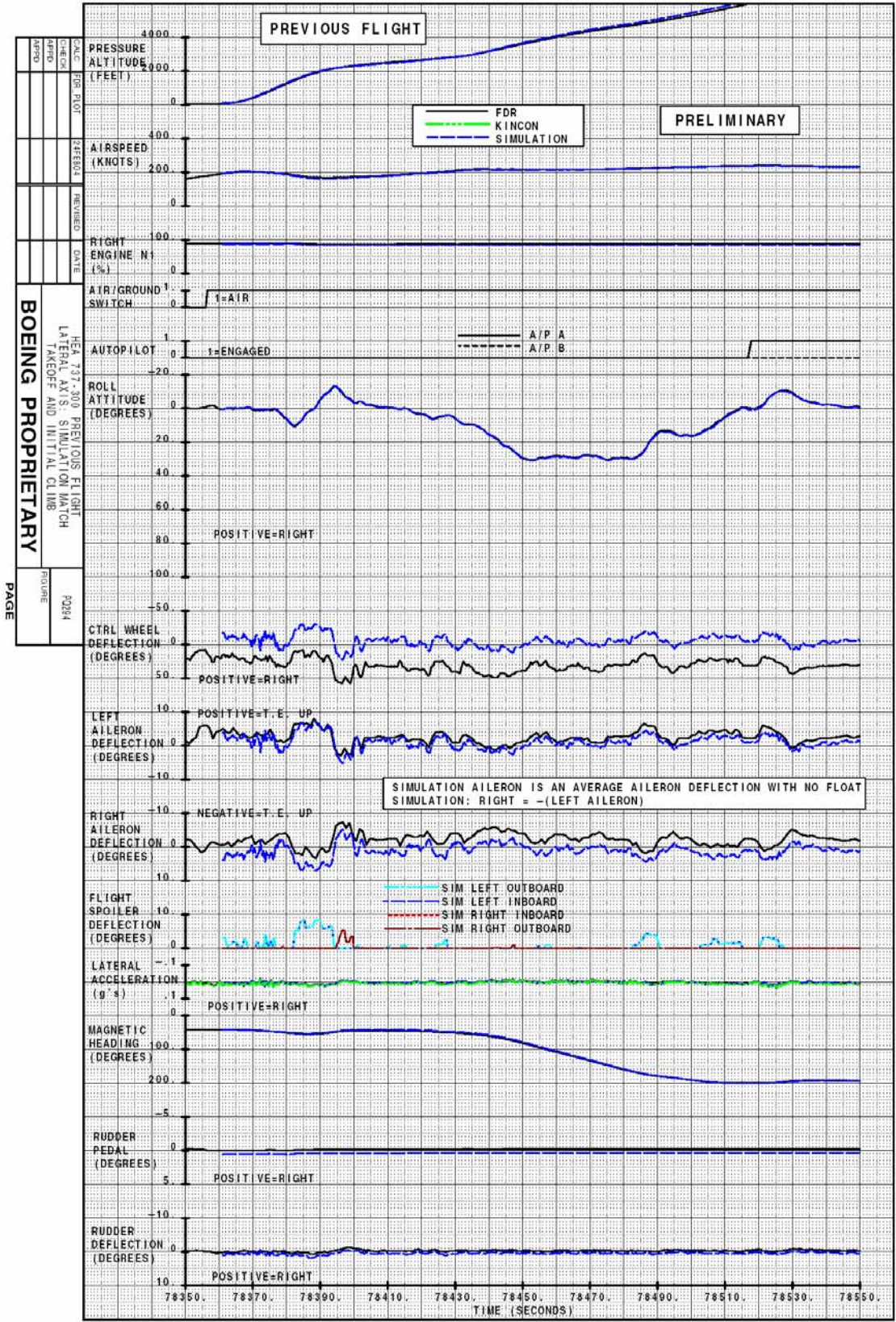
THE BOEING COMPANY



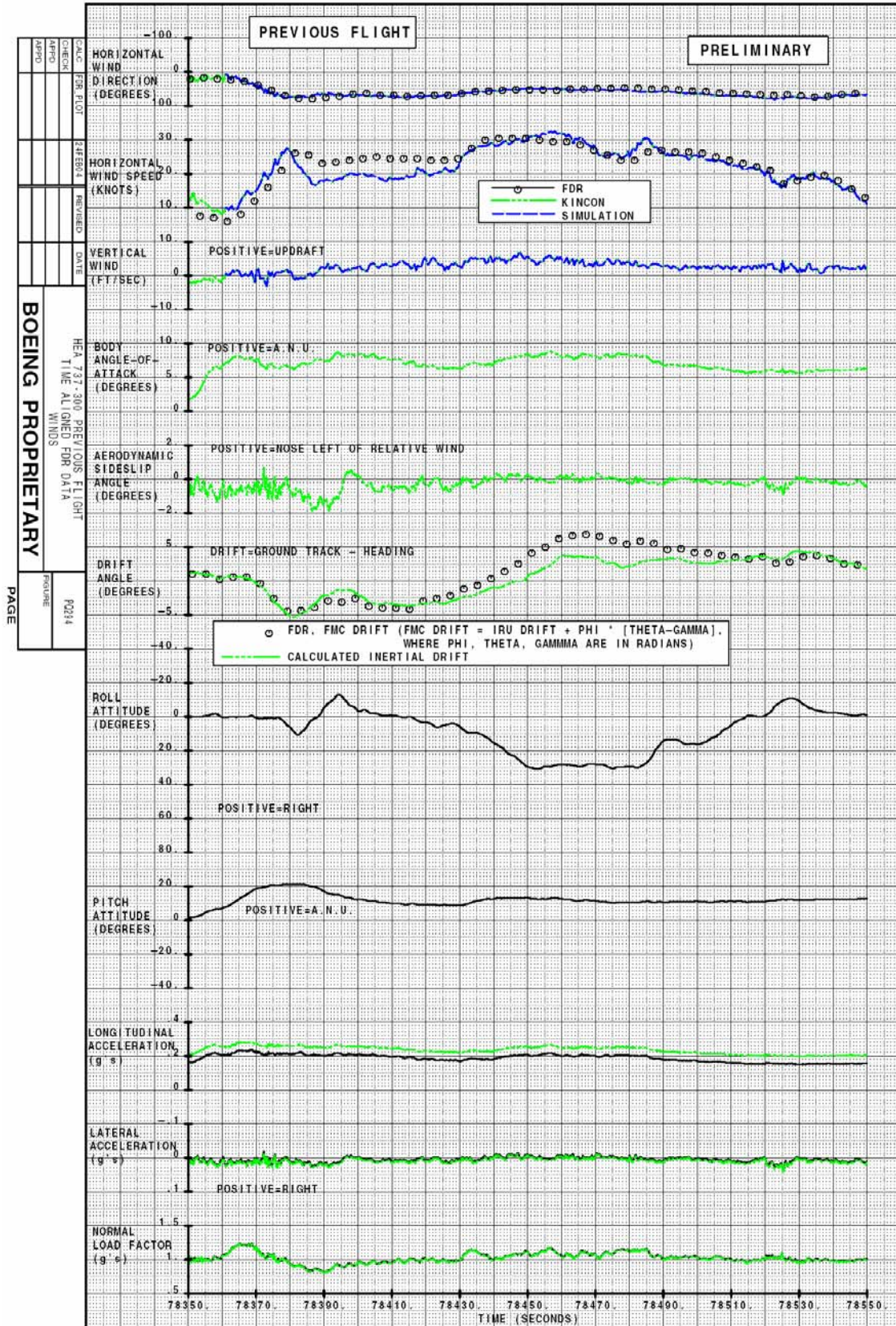
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 BOEING PROPRIETARY  
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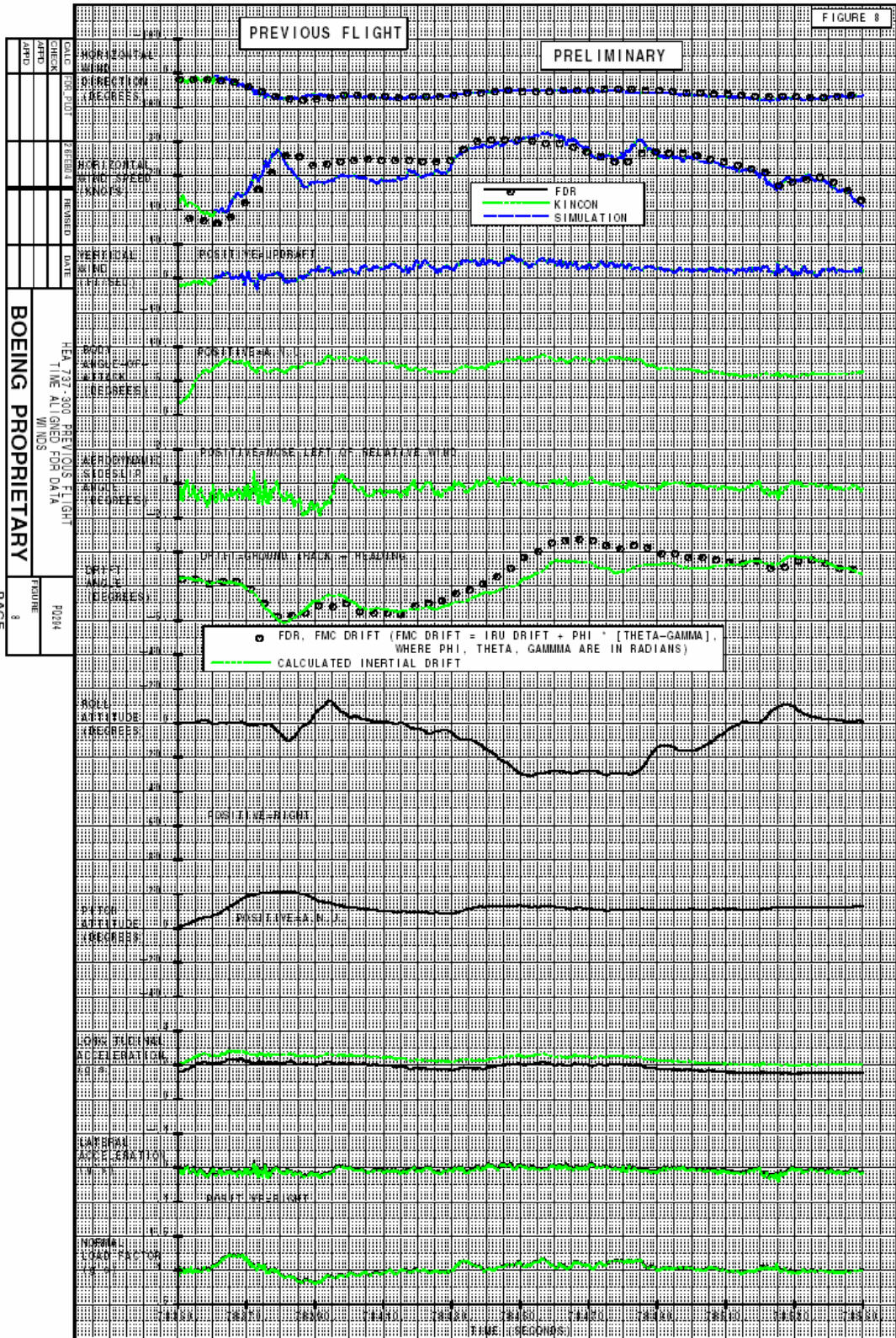
THE BOEING COMPANY



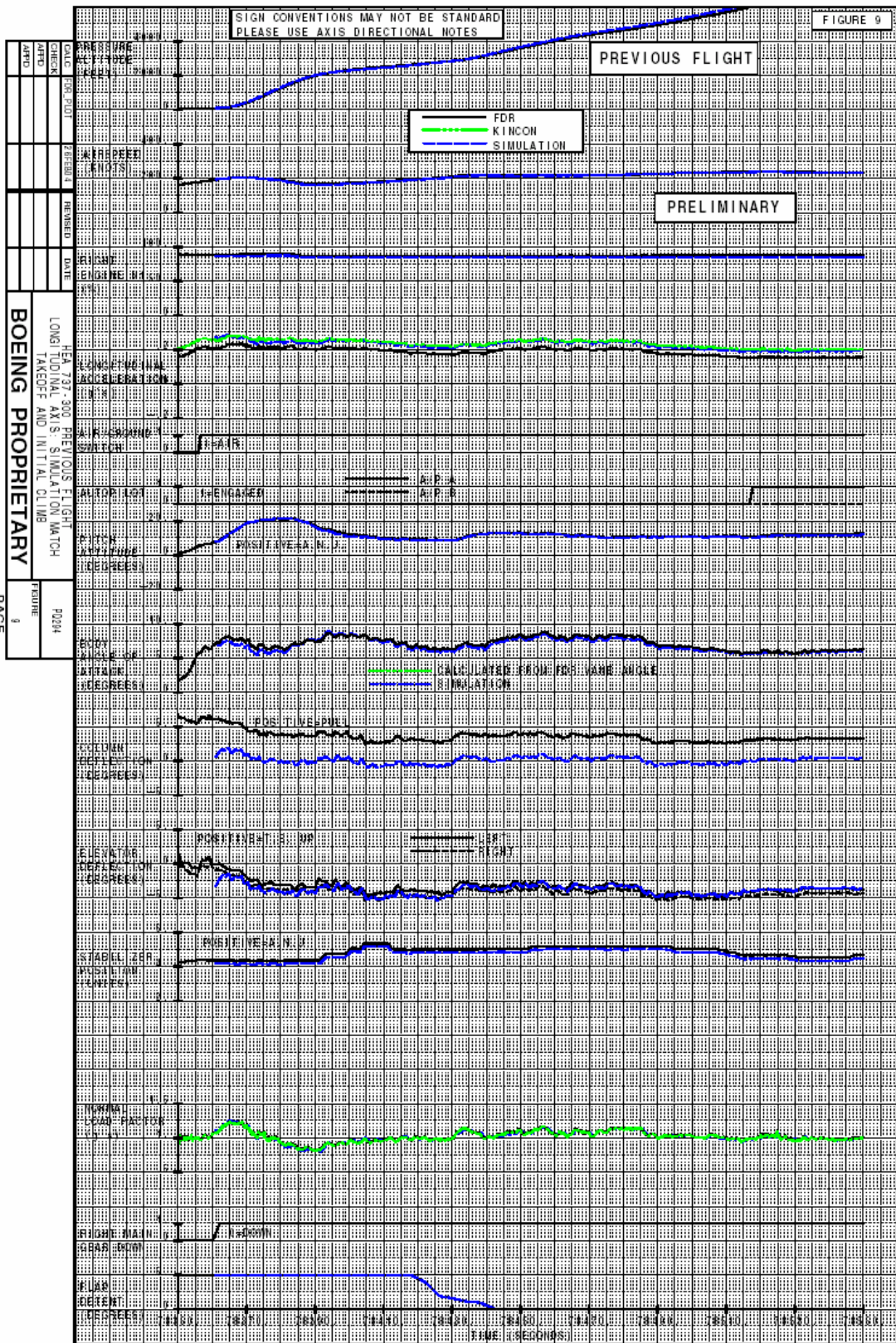


THE BOEING COMPANY







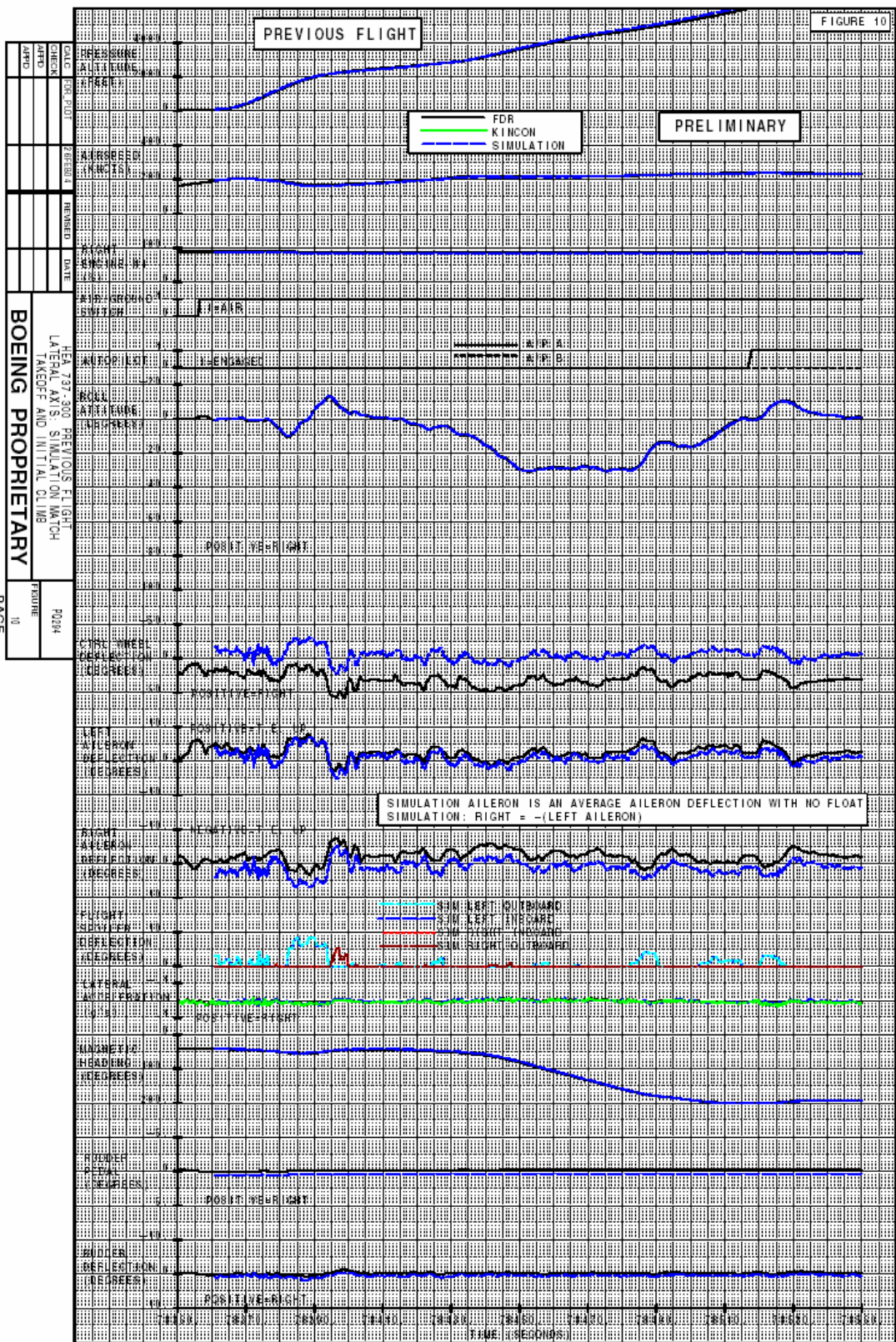


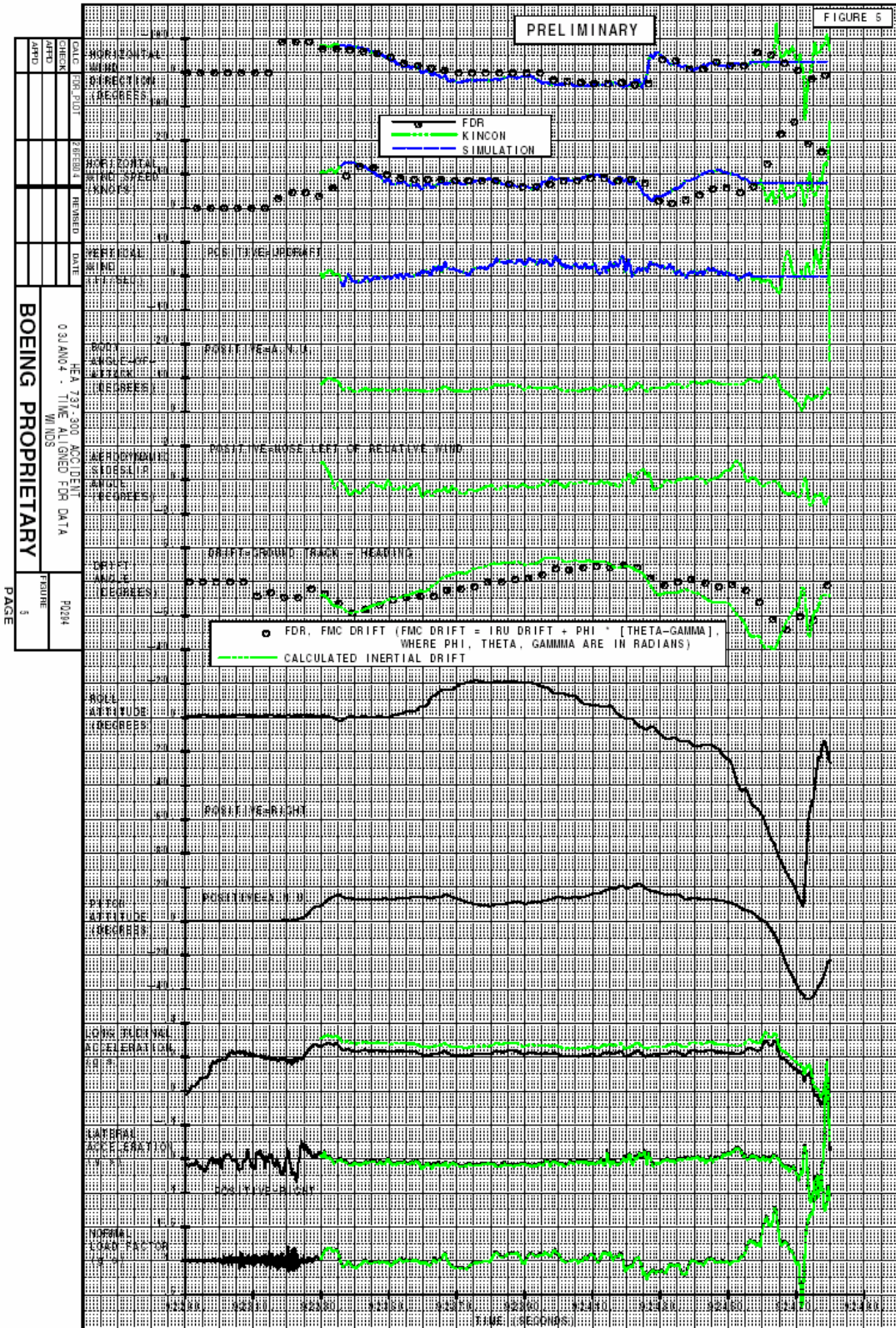
THE BOEING COMPANY

HEA 737-300 PREVIOUS FLIGHT  
LONGITUDINAL AXIS SIMULATION MATCH  
TAKEOFF AND INITIAL CLIMB

BOEING PROPRIETARY

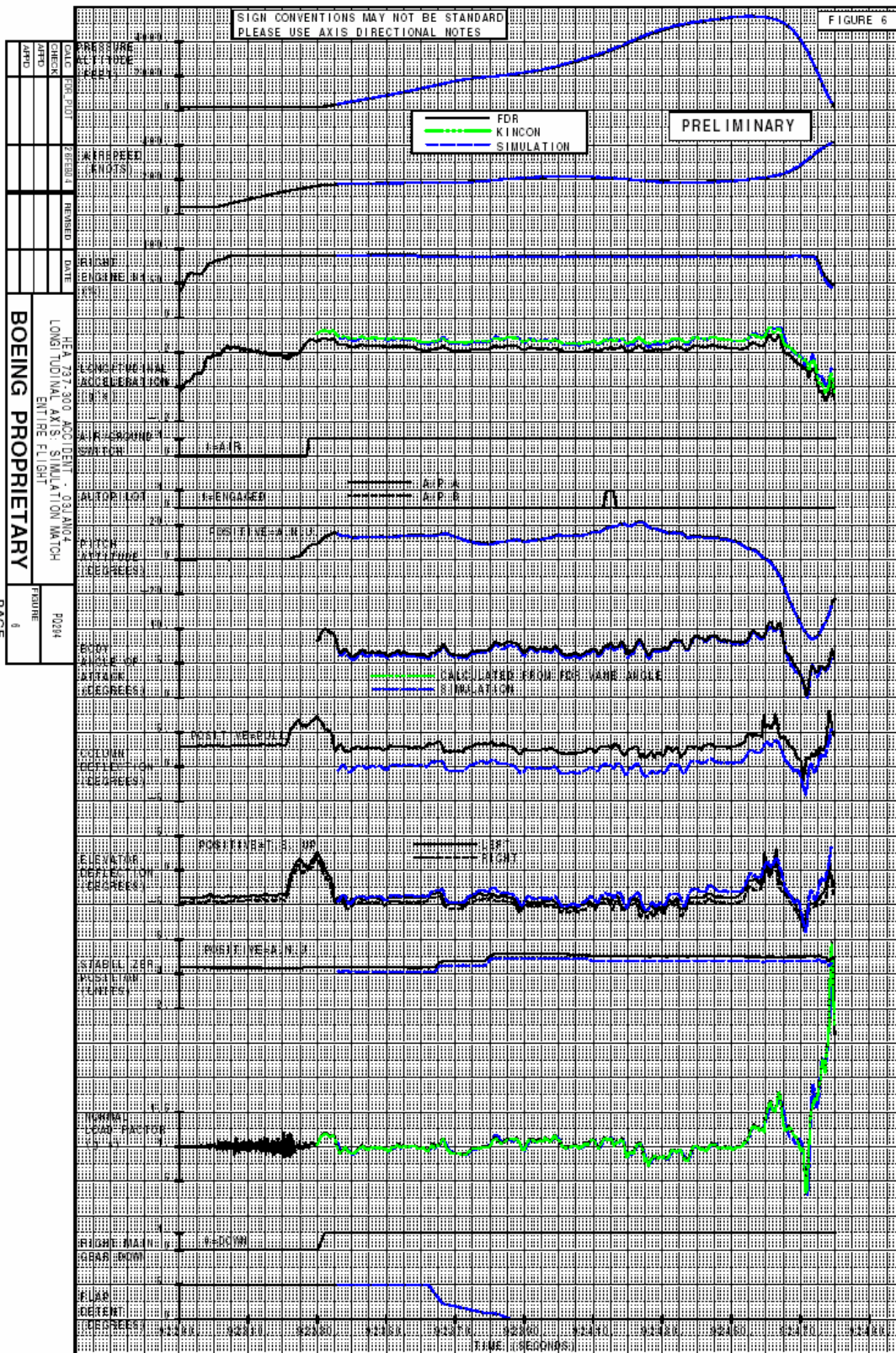
FIGURE 9  
PAGE 9

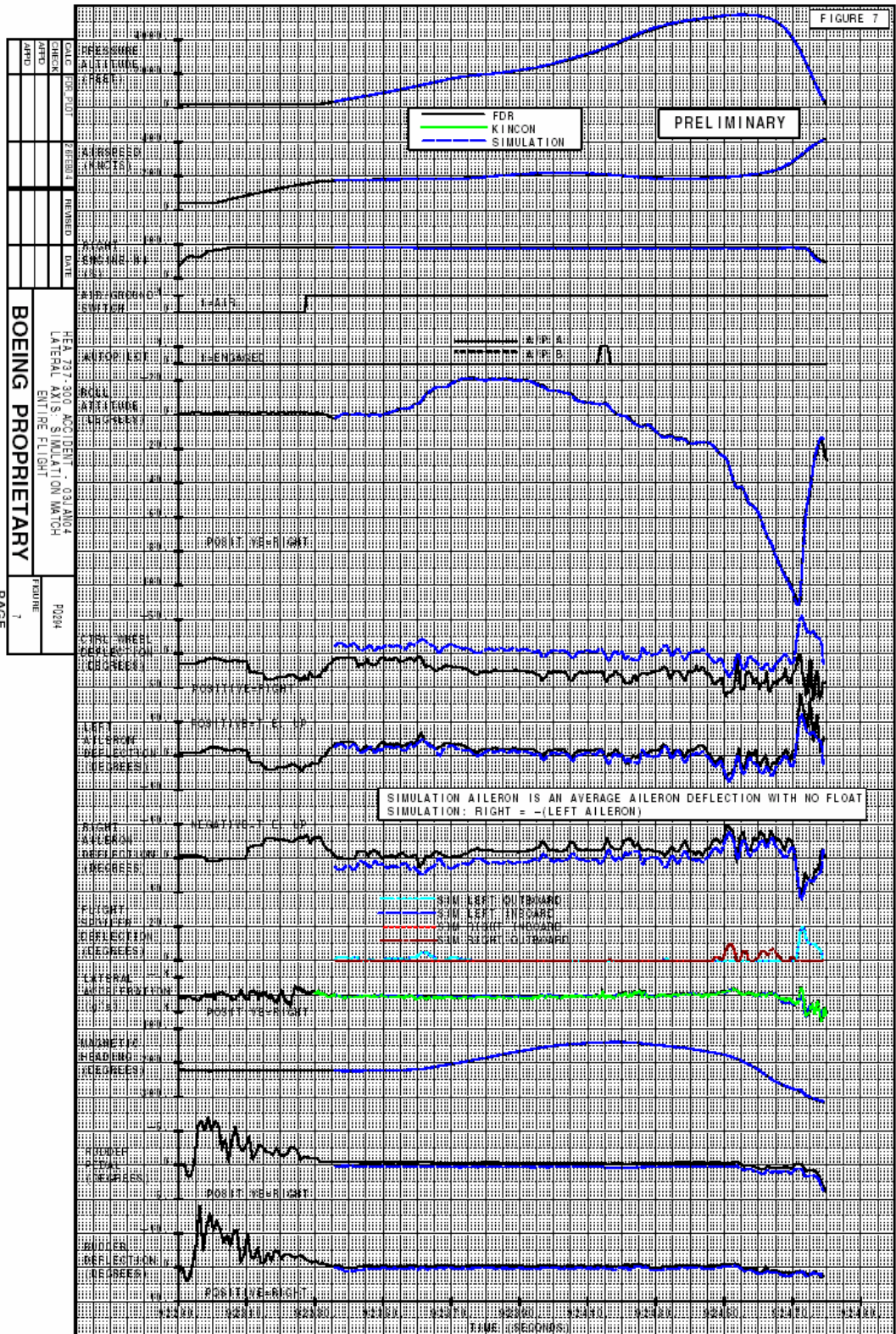




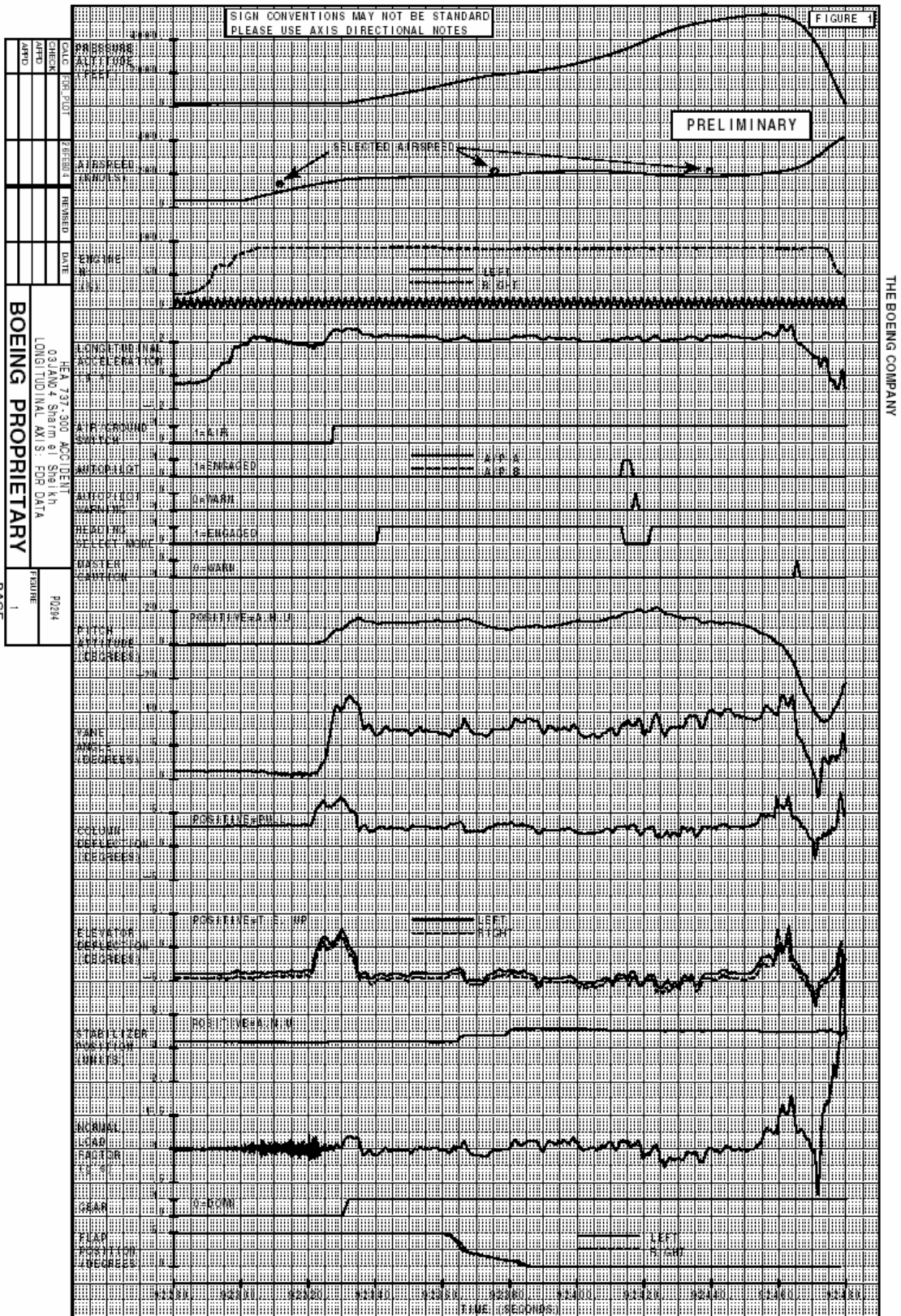
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 APPROVED: \_\_\_\_\_  
 PREPARED: \_\_\_\_\_  
 REVIEWED: \_\_\_\_\_  
 DATE: \_\_\_\_\_  
 HEA 737-300 ACCIDENT  
 03/JAN/04 - TIME ALIGNED FOR DATA  
 WINDS  
 BOEING PROPRIETARY  
 FIGURE 5  
 PAGE 3

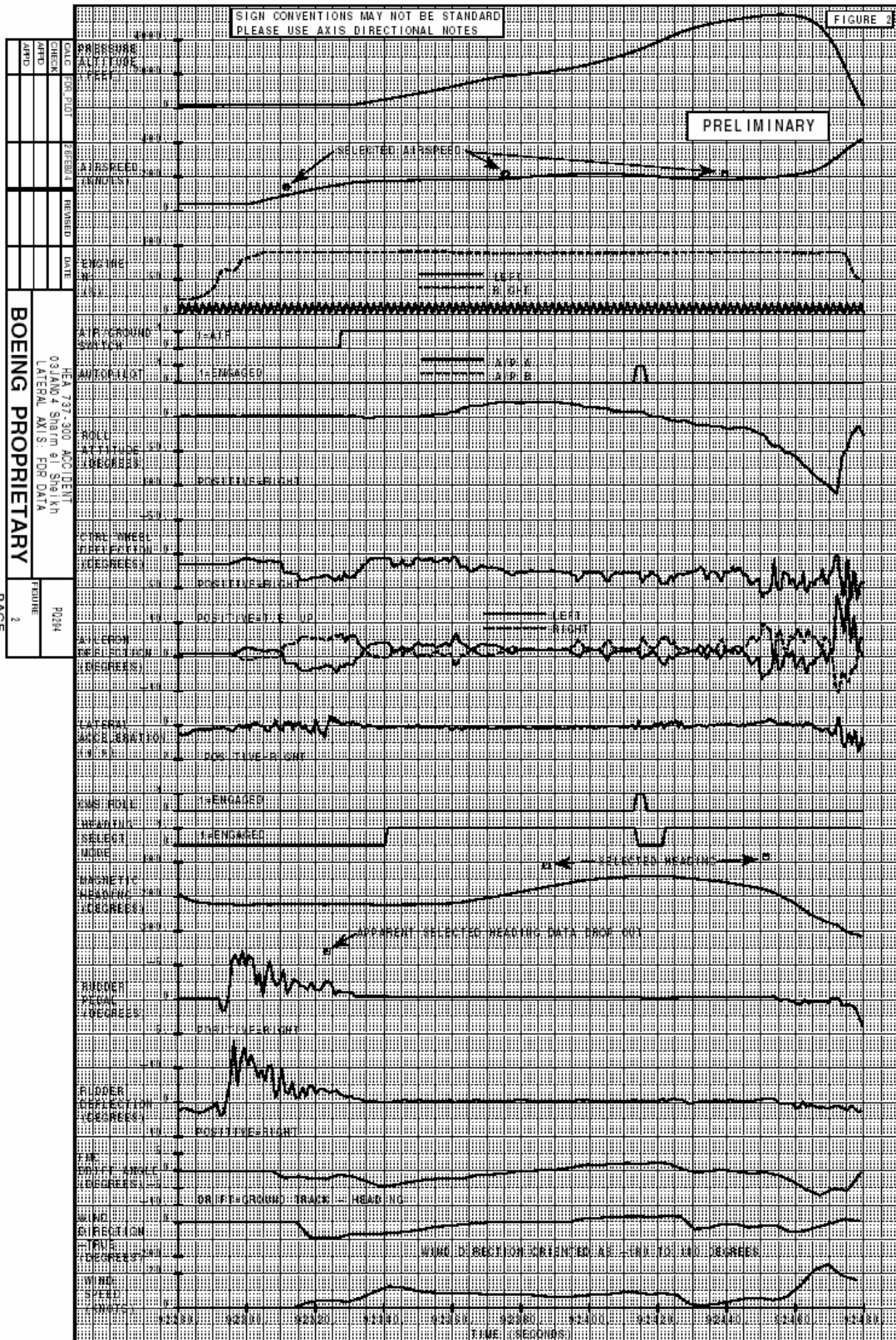
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CALC	FOR FLD	REMOVED	DATE
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APPRO			

HEA 737-300 ACCIDENT  
 03JAN4 Sharm el Sheikh  
 LATERAL AXIS: FDR DATA

**BOEING PROPRIETARY**

FORM  
 TIME  
 PAGE 2

HEA\_PQ294\_kincon (includes roll rate).pdf (FDR Data accident flight - plotted by Boeing (some selected parameters)-26 Feb 04 Fig's 3, 4

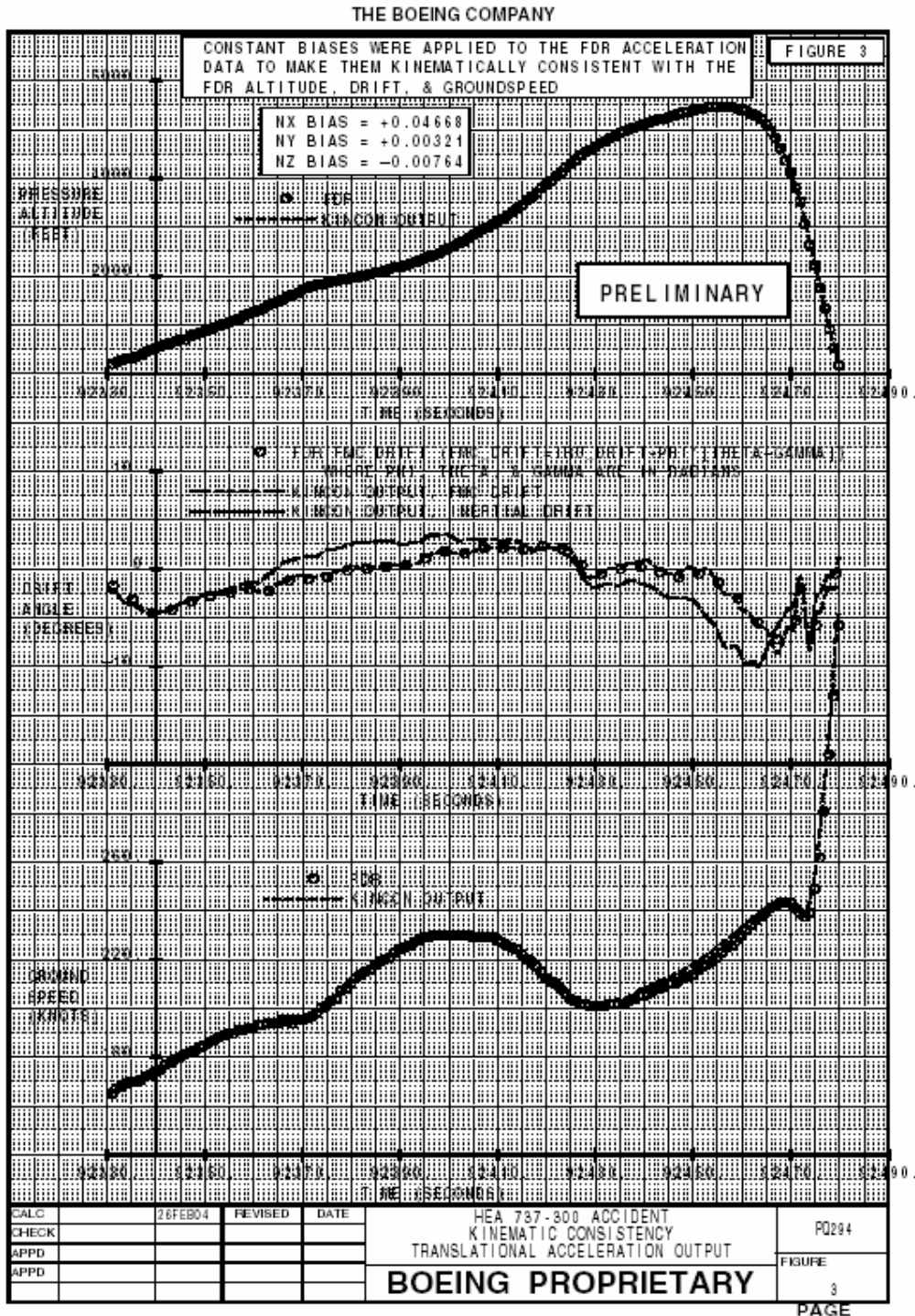
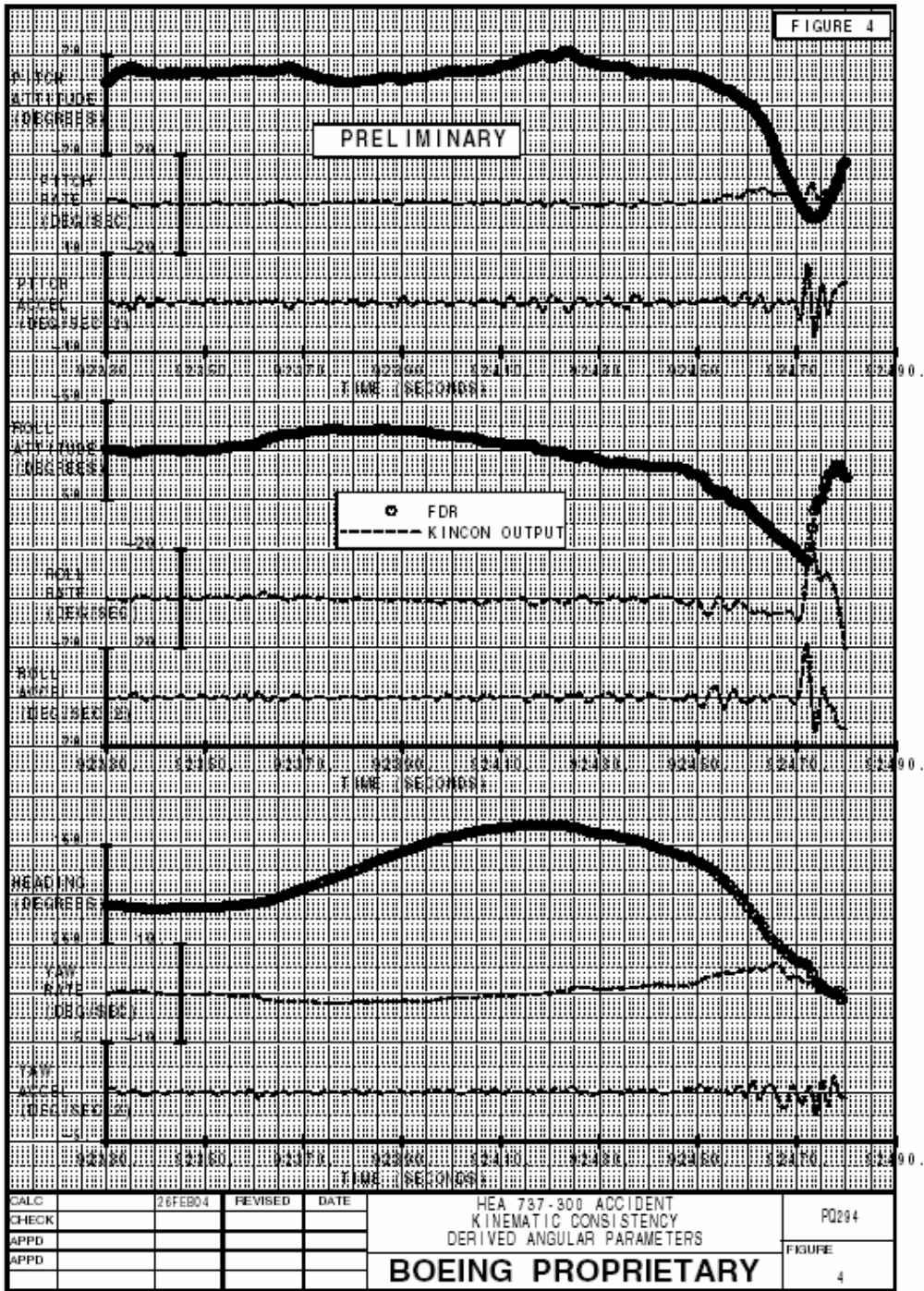
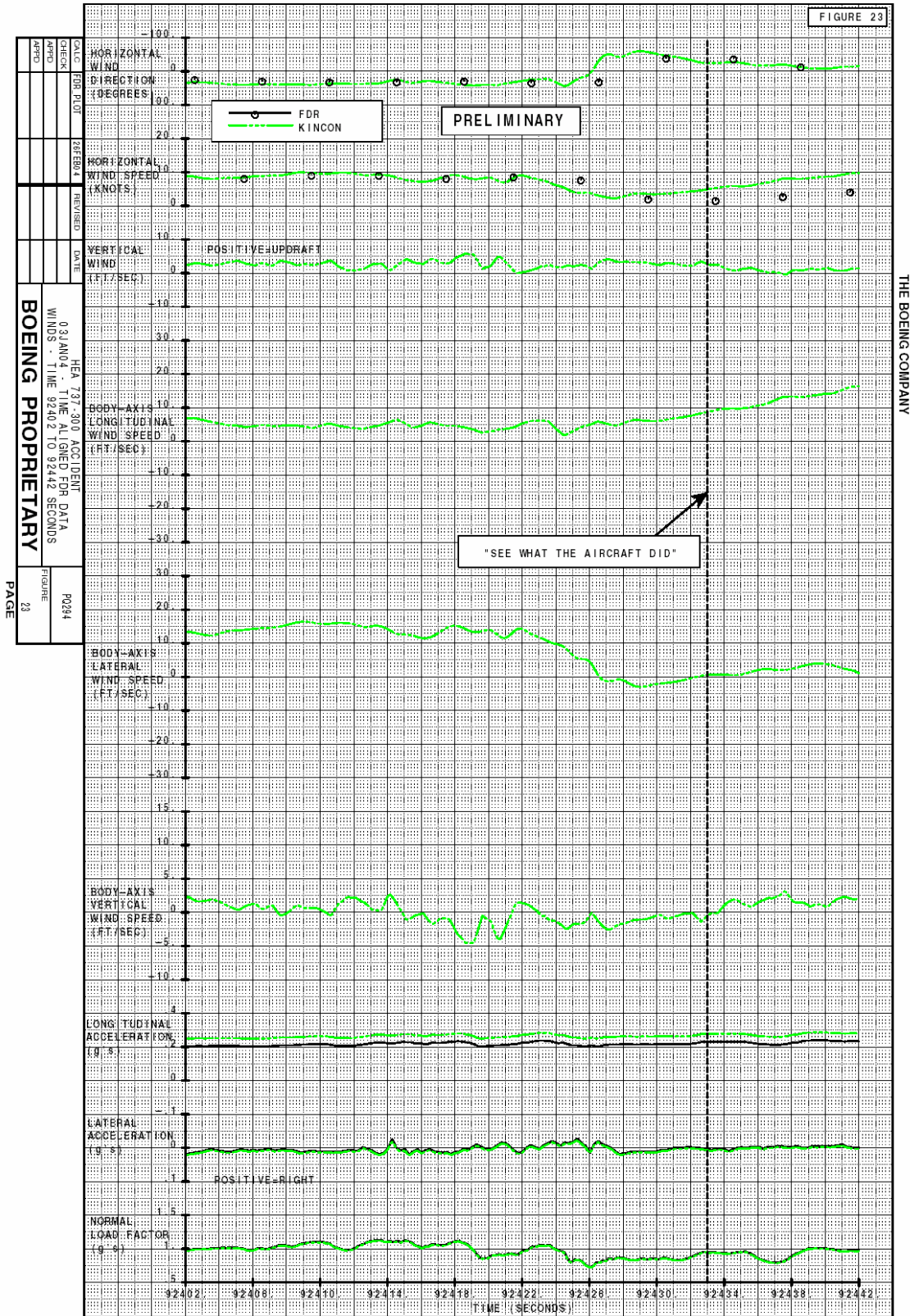


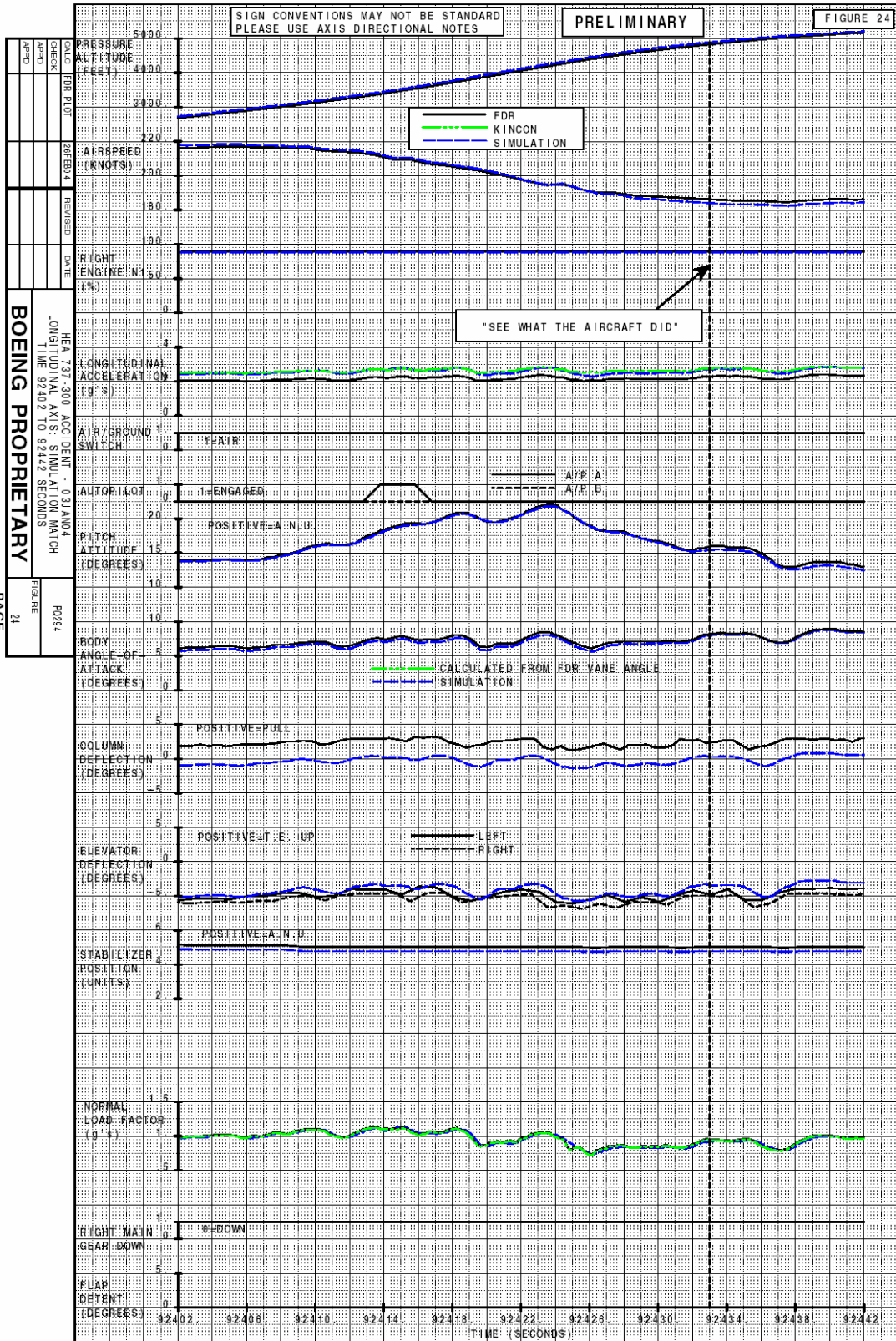


FIGURE 4



CALC	26FEB04	REVISED	DATE	HEA 737-300 ACCIDENT KINEMATIC CONSISTENCY DERIVED ANGULAR PARAMETERS	P0204
CHECK					FIGURE 4
APPD					
APPD				<b>BOEING PROPRIETARY</b>	PAGE





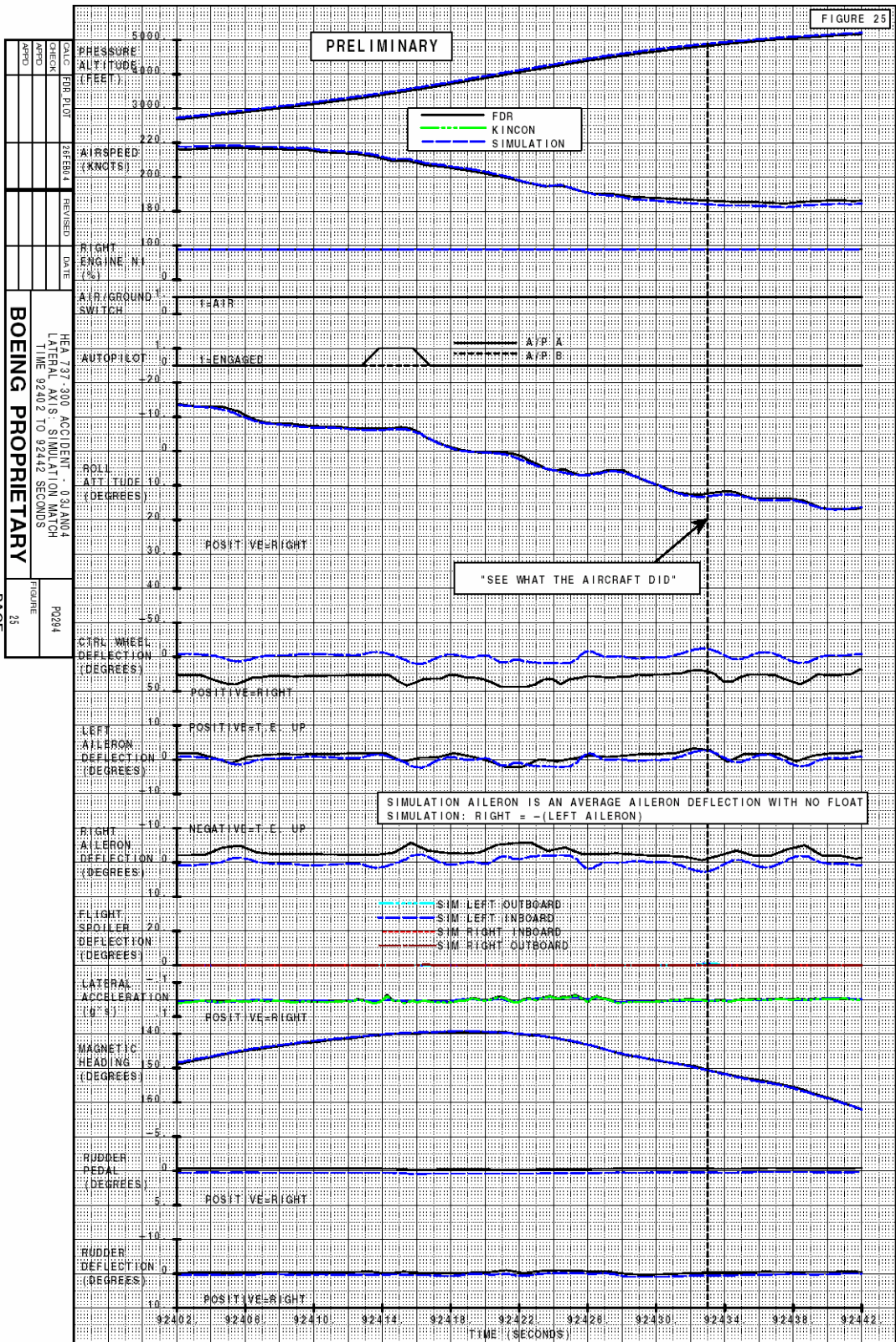
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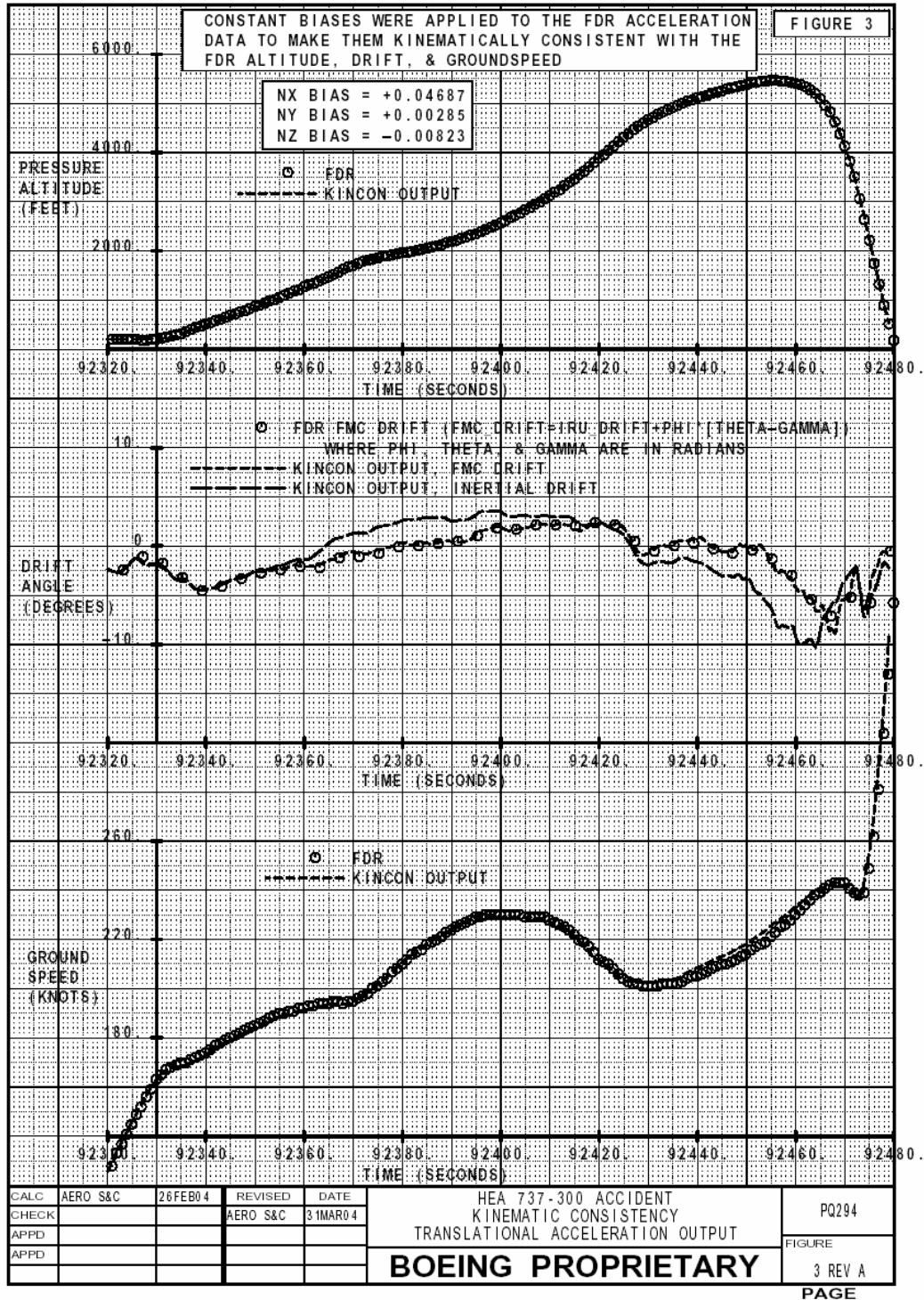
BOEING PROPRIETARY

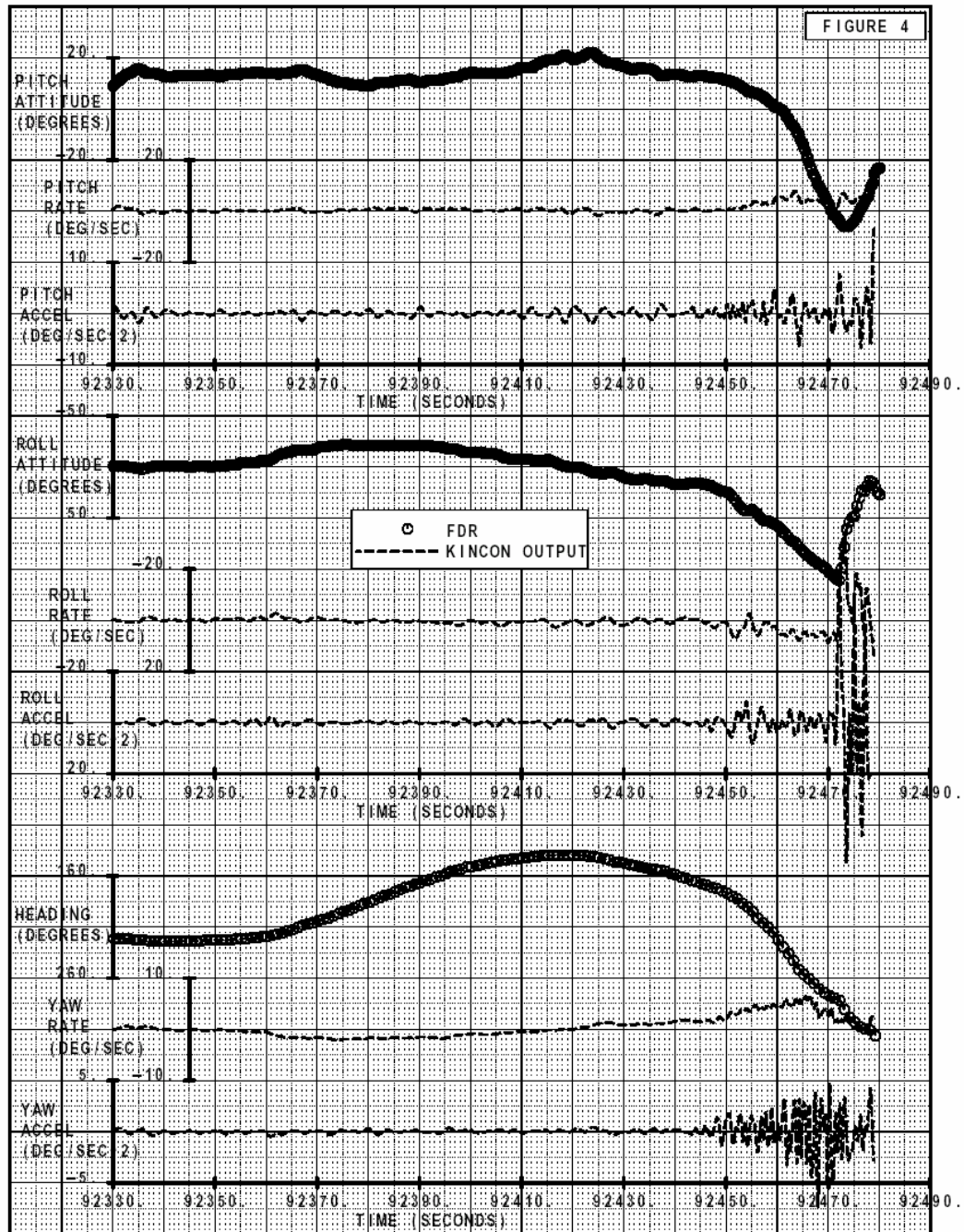
PAGE 24

CALC: FDR PLOT  
 CHECK: 28 FEB 04  
 APPD: 28 FEB 04  
 REVISED: 28 FEB 04  
 DATE: 28 FEB 04  
 HEA 737-300 ACCIDENT - 031AND4  
 LONGITUDINAL AXIS - SIMULATION MATCH  
 TIME 92402 TO 92442 SECONDS

FIGURE 24  
 P0294

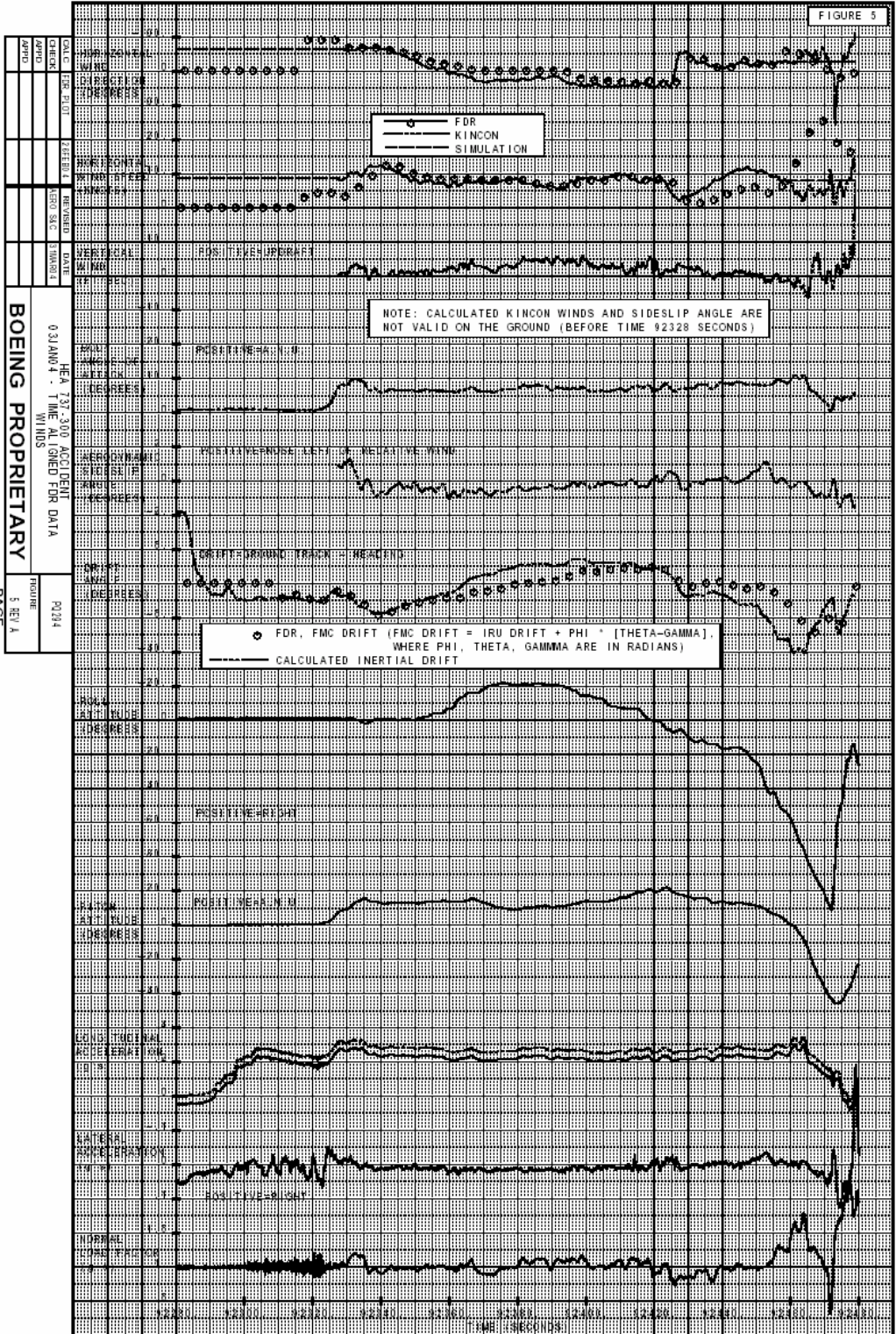




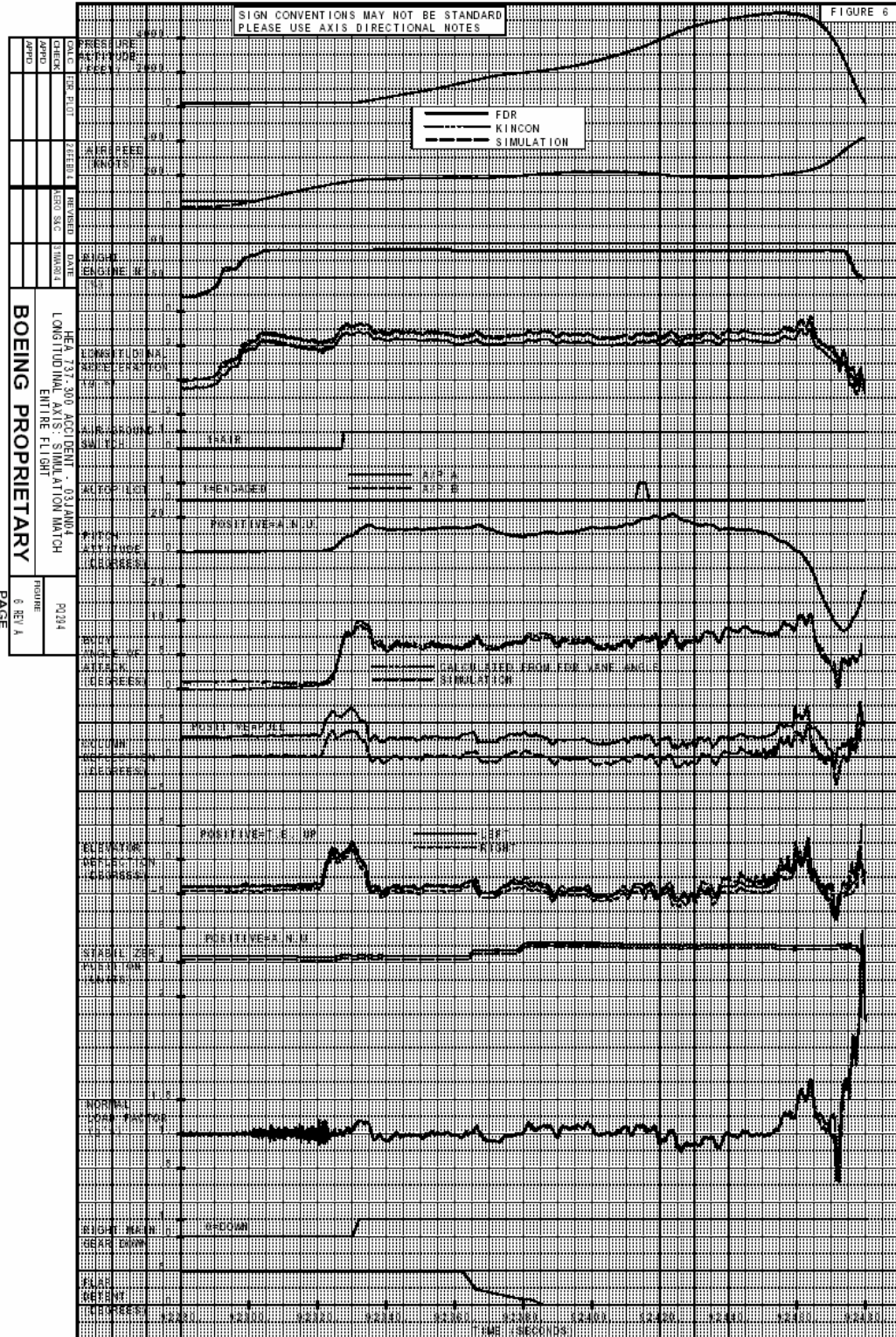


CALC	AERO S&C	26FEB04	REVISED	DATE	HEA 737-300 ACCIDENT KINEMATIC CONSISTENCY DERIVED ANGULAR PARAMETERS	PQ294
CHECK		AERO S&C	31MAR04	FIGURE		
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APPD				PAGE		

**BOEING PROPRIETARY**

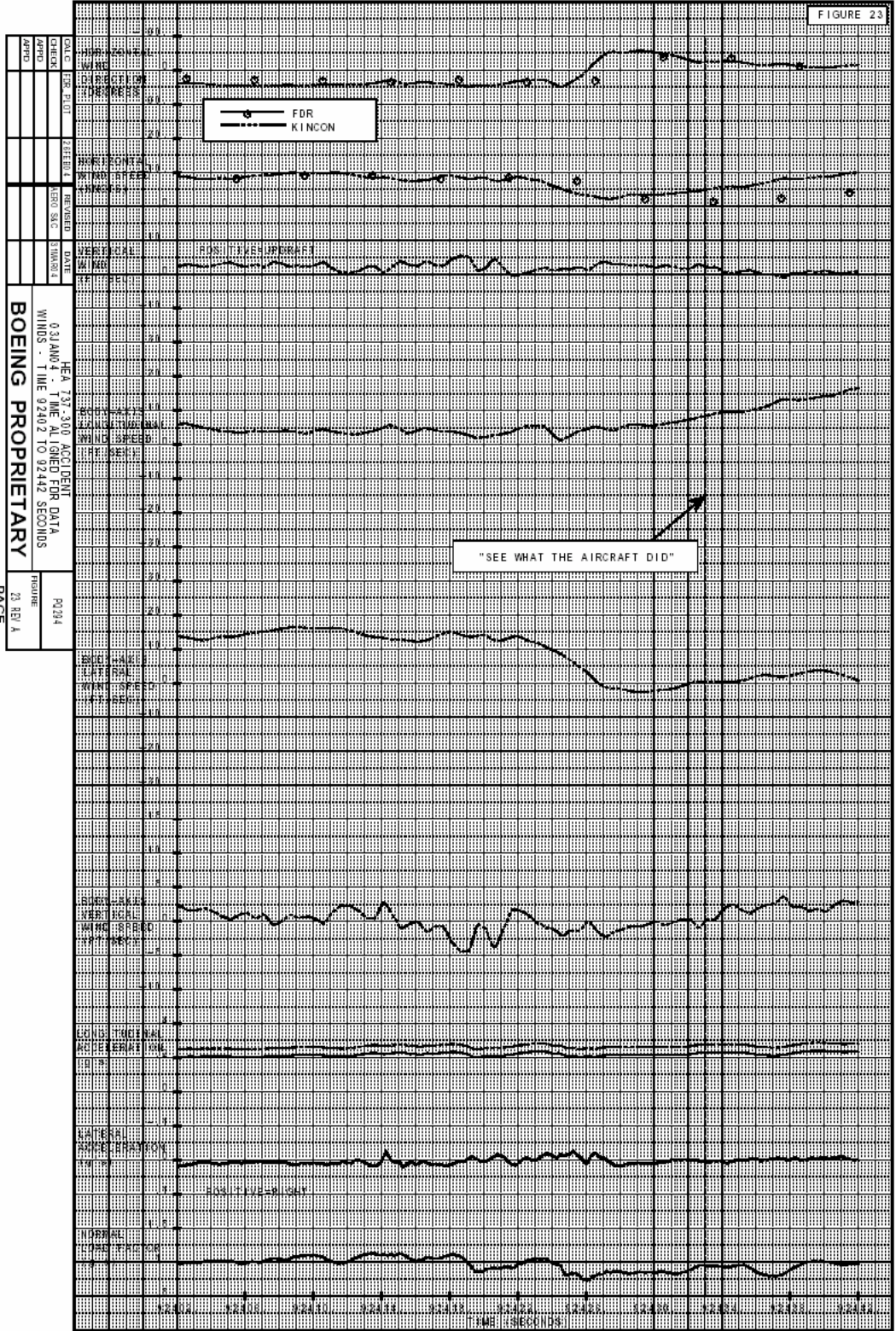


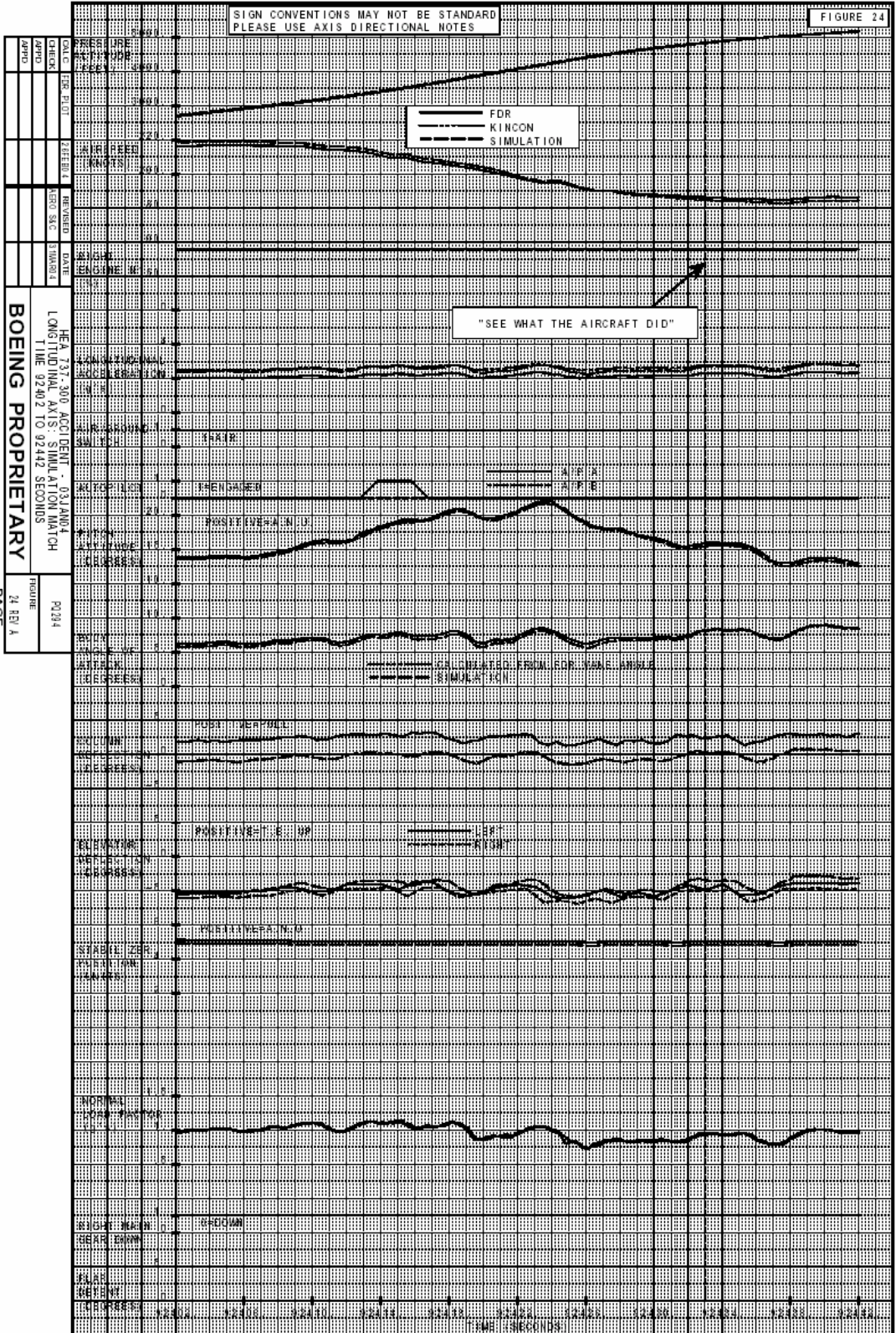
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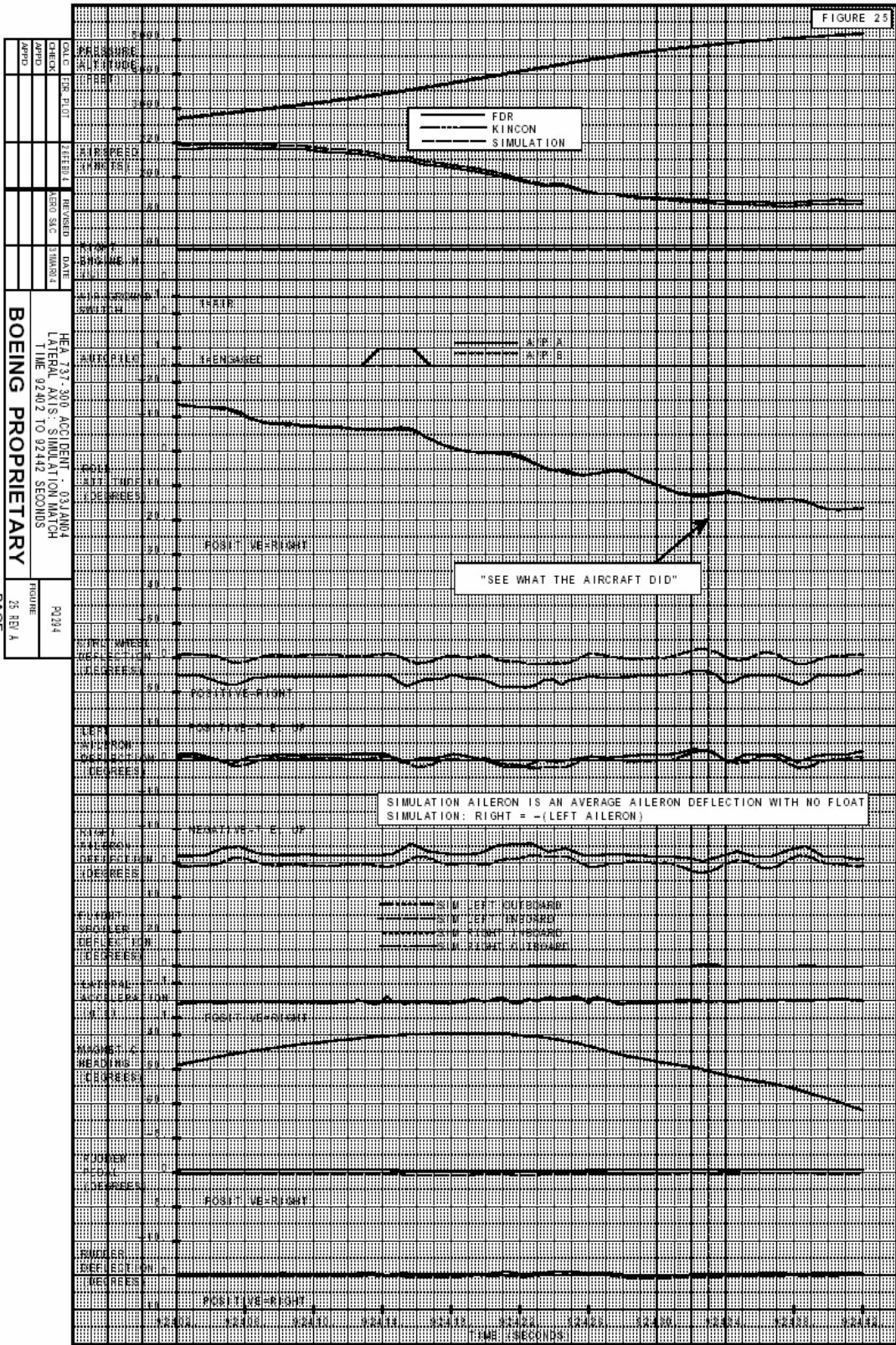








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DATE	TIME	REVISION	DATE

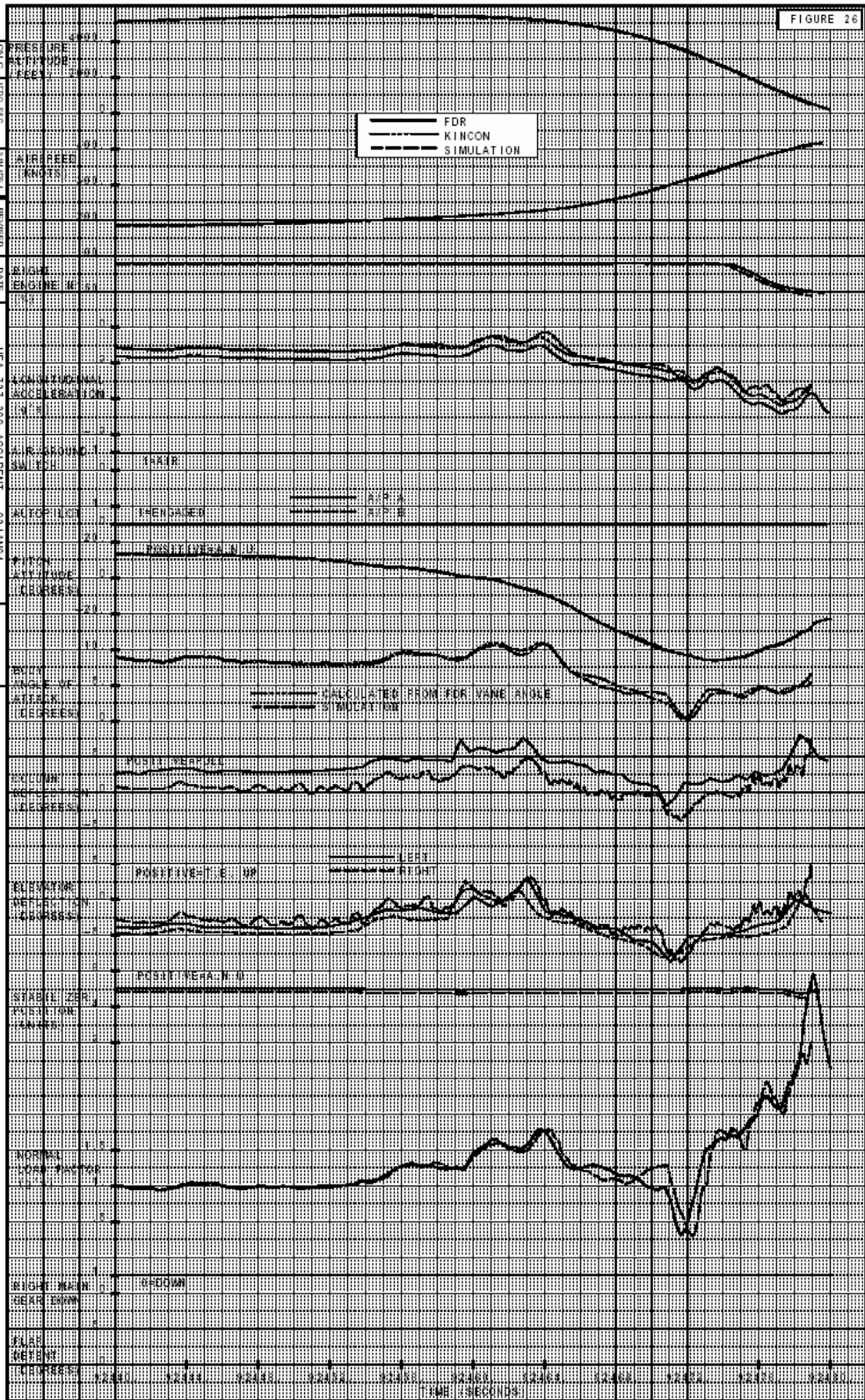
HEA 737-300 ACCIDENT - 031AND4  
 LATERAL AXIS - SIMULATION MATCH  
 TIME 92402 TO 92442 SECONDS

FIGURE 23 REV A

BOEING PROPRIETARY

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DATE	REVISED	DATE
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CHKD	CHKD	CHKD
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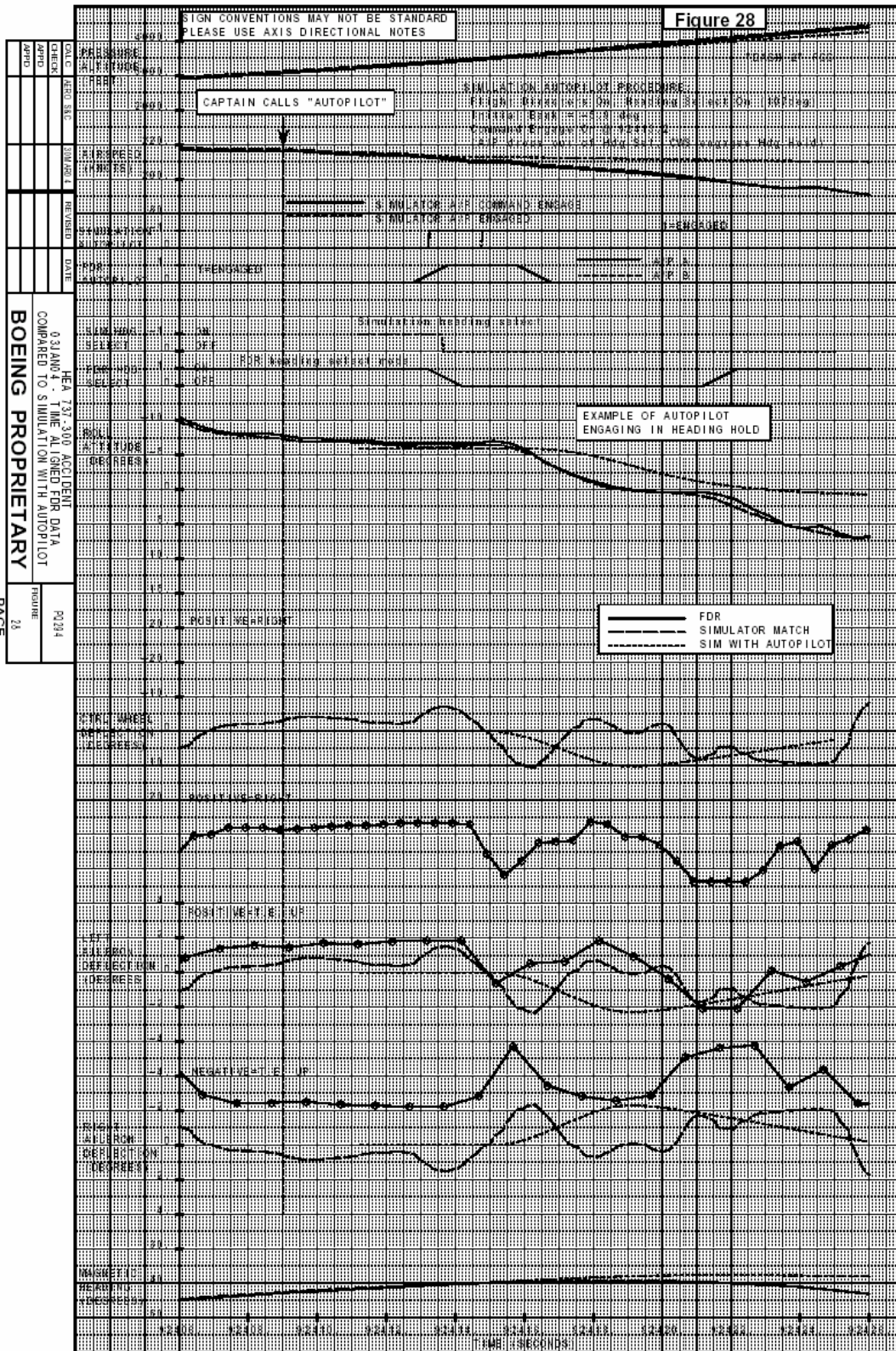
HEA 737-300 ACCIDENT - 03 JAN 04  
 LONGITUDINAL AXIS - SIMULATION MATCH  
 TIME 02:40 TO 02:480 SECONDS

P0334  
 FIGURE  
 26

BOEING PROPRIETARY

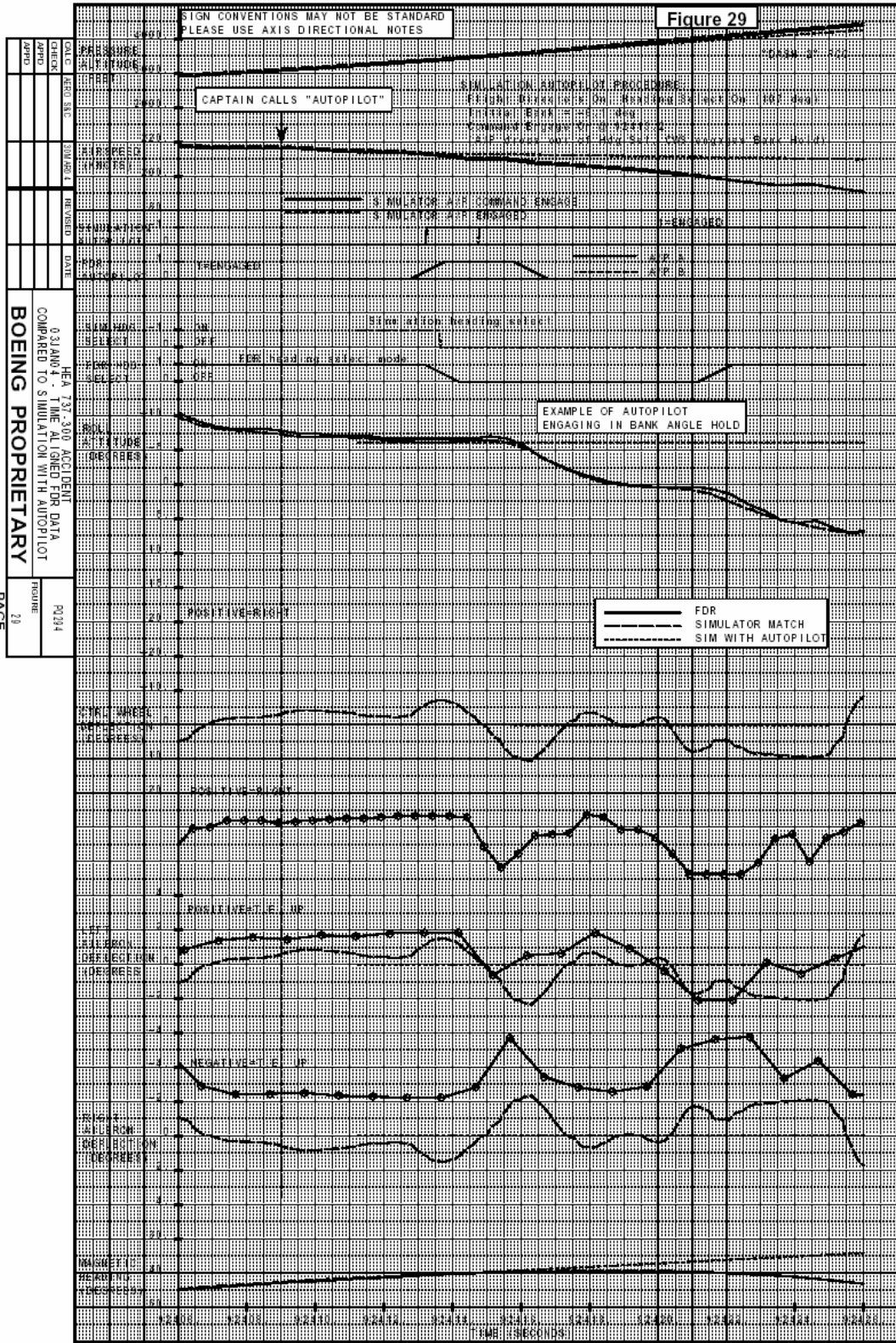
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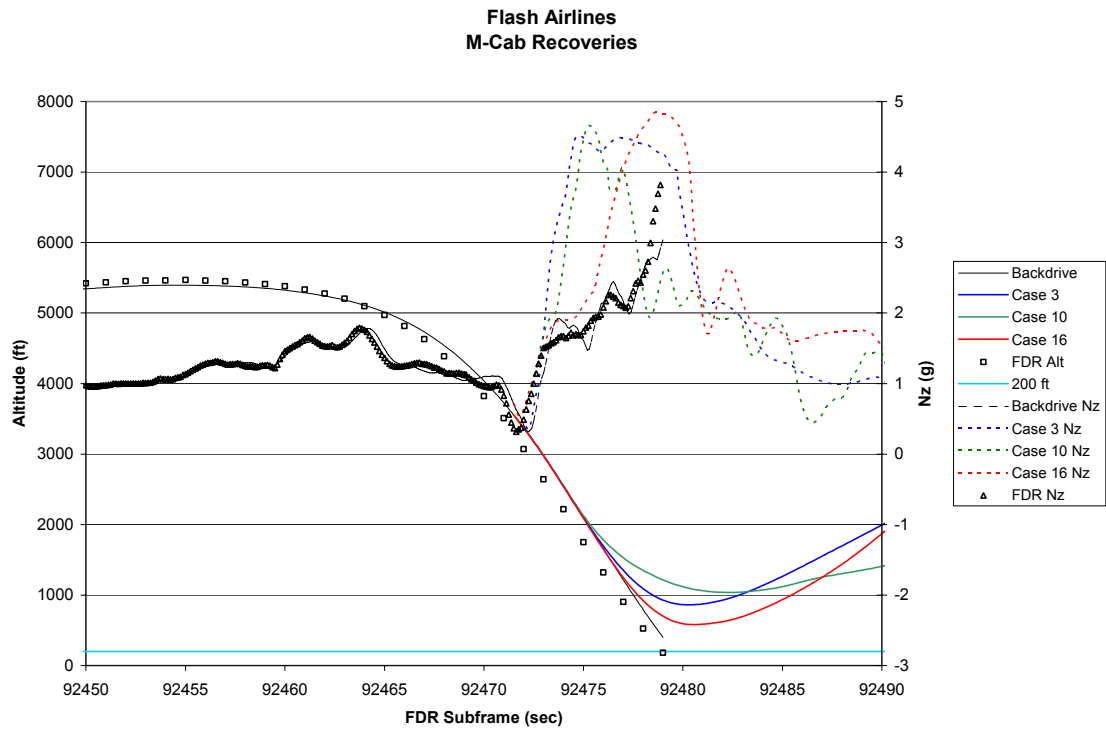
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DATE	03 JAN 4	HEA 131-300 ACCIDENT	PG 204
TIME	11:00	COMPARE TO SIMULATION WITH AUTOPILOT	FIGURE
APPRO		BOEING PROPRIETARY	PAGE
APPRO			21

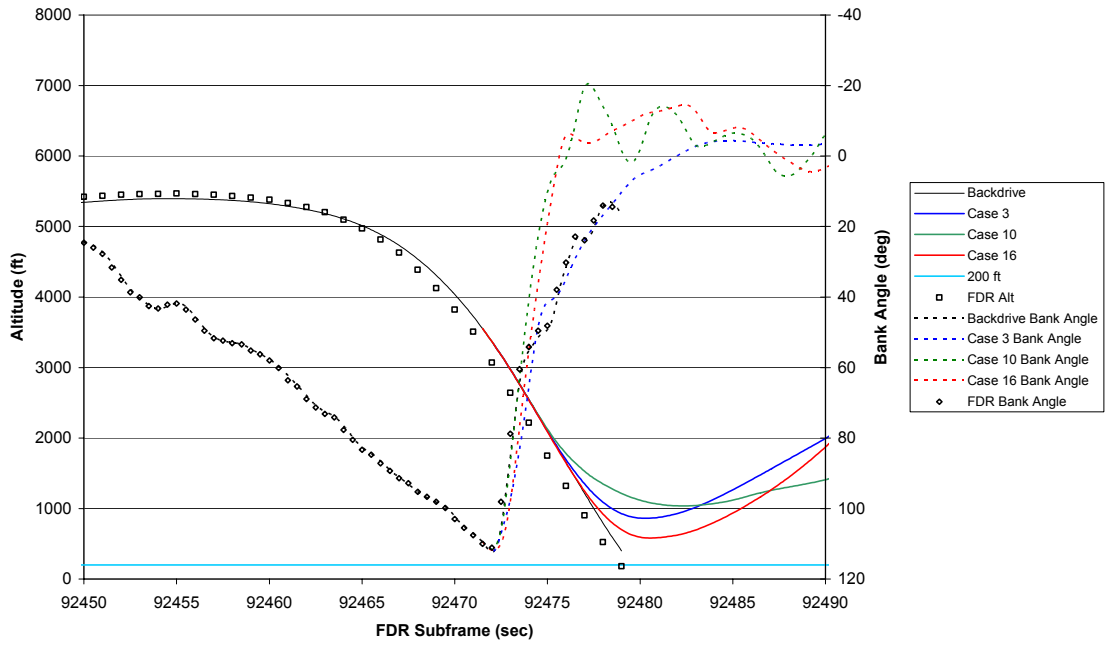




M Cab Recovery (Piloted Recovery.xls)



### Flash Airlines M-Cab Recoveries



Simulation Scenario (Simulation Scenario Status20 Sep.,04.xls)

Flash Airlines Requested Simulation Scenarios			Last Updated 7 Sept 04	29-Jul-04	20-Sep-04	
No.	Scenario	M-Cab Status	Comments	MCA Comments	MCA Comment	Presentation
1	Use M-cab like a training simulator (manual flight with no backdrive)	Available now	The M-cab is capable of performing like a training simulator. However, it does not have an "instructor's station" to insert pre-programmed malfunctions like many training simulators do. Therefore, if pre-programmed malfunctions are desired in the M cab, advance notice is required to ensure the correct routines can be loaded and available.	OK MCA will advise if any such pre-programmed malfunctions are desired.	OK	Boeing
2	Backdrive of accident flight (from FDR data)	Available now	The full backdrive from the FDR data is available. A "breakout" switch will be installed that will allow manual pilot inputs at any point in the scenario.	OK	OK	Boeing
3a	Slat extend (mid) fault	In work	No aero extend data			Boeing
3b	Slat extend (full) fault	In work	This scenario will be available in the cab. It is the same scenario for which plots were provided in March at the Cairo meeting, <b>except that we will insert the fault at flaps up.</b>		MCA requests to perform fault insertion simultaneously with breakout and then attempt to fly accident flight path. The intention is to compare FDR aileron to aileron required to fly accident profile with fault.	Boeing
4	Spoiler hardover fault	In work	Same as #3b except at time 92444		MCA requests that fault be inserted at A/P engage (92415)	Boeing
5	Spoiler float fault	In work	Same as #3b except at time 92444		MCA requests that fault be inserted at A/P engage (92415)	Boeing
6	Slat "float" (assumed actuator detached and/or jammed/cocked slat)	Not available	The position of a floating slat is determined by the airload on the slat and friction within the system. We do not currently have that data available for the accident flight airspeed and altitude conditions. The airloads will either extend the slat, retract the slat, or will be insufficient to overcome system friction. Therefore, we believe the airplane level roll response will be bounded by the response to a slat fully extended fault such as #3a above. We are currently searching for additional aero data as requested by the MCA. <b>We have not been able to locate any additional aero data requested by the MCA.</b>	Is there any additional aero data available for the effects of slats at other positions (i.e. between up and mid, between mid and full, or cocked)?	OK, Must be done or at least mid posn.	Boeing
7	Hardover on one aileron PCU	In work	A hardover of one aileron PCU will result in both aileron PCUs commanding full aileron, spoiler and control wheel hardover. We intend to demonstrate this scenario in the same manner as #3a above by inserting the fault at time 92444.	OK	OK	Boeing
8	Aileron trim runaway	Available now	Aileron trim runaway can be simulated by manually moving the aileron trim control in the cab during manual flight. This can be done as part of #1 above.	OK	OK	Boeing
9	A/P with MCP erroneous selected heading	In work	This scenario will result in the autopilot flying to the erroneous selected heading. This scenario can be simulated initializing the simulator at time 92395, then running open loop. At that point, the autopilot can be engaged and the desired "erroneous" selected heading can be entered on the MCP.	OK	OK	Honeywell
10a	A/P with MCP Selected Heading knob mechanically inoperative, such that it does not transfer pilot commands. (Selected heading window and output to FCC constant regardless of knob movement)	Not required	This scenario has the same effect as #9 above and can be simulated in the same way.		OK	Honeywell
10b	A/P with one or more segments in the MCP selected heading LCD window inoperative leading to improper indication (e.g. displaying 6 instead of 8)	Not required	The result of this fault will be that the apparent value in the heading window can be different than the value transmitted to the EADI for display of the heading bug and to the FCC for use in autopilot heading select mode. Although we will not be able to simulate a different value in the selected heading window, we believe that this fault can be simulated in the same way as #9 above.		OK	Honeywell

11	A/P Actuator hardover	In work	This scenario will result in a "hardover" to the autopilot actuator authority limit (60 deg with the autopilot force limited not engaged). We can simulate this scenario by introducing the fault and "breaking out" simultaneously at 92415 (A/P initial engage)	OK	OK	Boeing
12a	A/P Actuator ARM Solenoid valve failed open with A/P disconnected	Not required	With the arm solenoid open, the autopilot mod piston can move in response to FCC commands, but as the detent solenoid is not open, the mod piston is not coupled to the ailerons and the A/P actuator cannot command aileron motion. We do not believe it is necessary to simulate this scenario.		OK	Boeing
12b	A/P Actuator Detent Solenoid failed open with A/P disconnected	Not required	The arm and detent solenoids are in series. If the arm solenoid is closed, no hydraulic fluid is available to allow the detent pistons to couple the mod piston to the ailerons. The A/P actuator cannot command aileron motion. If this fault exists when the autopilot is trying to engage, the engagement may occur with a jolt as the mod piston would be coupled to the ailerons before the position synchronization is complete. We do not believe it is necessary to simulate this scenario.		OK	Boeing
12c	A/P Actuator both arm and detent solenoid open with A/P disconnected	Not required	This is the normal condition when the autopilot is engaged. The transfer valve spool moves the mod piston moves in response to commands from the FCC and the detent pistons are pressurized to couple the actuator to the ailerons. If the autopilot is not engaged, the FCC commands the transfer valve to hold the autopilot actuator in the neutral (aileron failed) position. Normal autopilot breakout is still available to override the autopilot. Without pilot intervention, the net result would be the same as letting go of the wheel and letting it center. We do not believe it is necessary to simulate this condition.		OK, Must be done	Boeing
12d	A/P Actuator triple fault (arm and detent solenoid open, transfer valve jam off center)	See #11	This triple fault will result in an A/P actuator hardover. The force limit of the actuator still operates normally. The hardover condition is the same as #11 above.		OK	Boeing
12e	A/P Actuator quadruple fault (arm and detent solenoid open, transfer valve jam, pressure regulator jam)	See #11	This quadruple fault will result in an A/P actuator hardover. Because the pressure regulator is jammed, the relief valve operates and limits detent piston pressure. The wheel force required to overcome the actuator increases from 16 lbs of wheel to approximately 20 lbs of wheel.		OK, transfer valve jamed at different posn	Boeing
12f	A/P Actuator quadruple fault (arm and detent solenoid open, transfer valve jam, relief valve jam)	See #11	This quadruple fault will result in an A/P actuator hardover. Although the relief valve is jammed (stuck to the pressure regulator slide), the pressure regulator limits detent piston pressure to the normal level. The wheel force required to overcome the actuator is the normal 16 lbs of wheel.			Boeing
12g	A/P Actuator quintuple fault (arm and detent solenoid open, transfer valve jam, pressure regulator and pressure relief valve)	In work	This quintuple fault will result in an A/P actuator hardover. In this scenario, neither the pressure regulator nor the relief valve can reduce the detent piston pressure which reaches hydraulic system pressure (3000 psi). Wheel force required to overcome the actuator increases from 16 lbs of wheel to approximately 80 lbs of wheel.		MCA requests to observe this fault (feel the forces) or the highest forces possible in the M-cab.	Boeing
13	A/P with IRU shutdown	Not required	The response of the autopilot to an IRU shutdown is to disconnect. We do not believe it is necessary to simulator this scenario.	OK	OK	Honeywell
14	A/P with Erroneous R IRU output of straight and level flight during bank (no NCD or fail warn transmitted)	In work	The autopilot will command aileron to its authority limit (20 deg with aileron force limiter). If the airplane heading crosses the selected heading the autopilot command will reverse. M-Cab simulation will not accurately reflect the wheel forces in this situation.	OK	OK	Honeywell
15a	A/P with Erroneous L IRU output of roll rate with all other parameters correct (separately and then see if possible to do at same time as above fault)	Not required	Autopilot A does not use L IRU roll rate as an input. This fault has no effect on the operation of autopilot A.		MCA requests this be changed to R IRU output of NCD for roll rate.	Honeywell
15b	A/P with R IRU output of NCD for roll rate	Not required	The response of the autopilot to R IRU output of NCD for roll rate is to disconnect. We do not believe it is necessary to simulate this scenario.		OK	Honeywell
16	Autopilot spoiler sensor fault (erroneous value)	Not applicable to M-Cab	The sensed value of spoiler angle is only used by the autopilot when the flaps at 30 or beyond. This fault would have no effect on the operation of the autopilot for the accident flight.	OK	OK	Honeywell

Simulation Scenario (Simulation Scenario Status 27-30 Sep, 04.xls)

Flash Airlines Requested Simulation Scenarios				Last Updated 21 Sept 04		
No.	Scenario	M-Cab Status	Motion	Comments	20-Sep-04 MCA Comment	Presentation
1	Use M-cab like a training simulator (manual flight with no backdrive)	Available now	Yes	The M-cab is capable of performing like a training simulator. However, it does not have an "instructor's station" to insert pre-programmed malfunctions like many training simulators do. Therefore, if pre-programmed malfunctions are desired in the M-cab, advance notice is required to ensure the correct routines can be loaded and available.	OK	Boeing
2	Backdrive of accident flight (from FDR data)	Available now	Yes	The full backdrive from the FDR data is available. A "breakout" switch is installed that will allow manual pilot inputs at any point in the scenario.	OK	Boeing
3a	Slat extend (mid) fault	Not available		No aero extend data		Boeing
3b	Slat extend (full) fault	In work	No	This scenario will be available in the cab. It is the same scenario for which plots were provided in March at the Cairo meeting, <b>except that we will insert the fault at flaps up.</b>	MCA requests to perform fault insertion simultaneously with breakout and then attempt to fly accident flight path. The intention is to compare FDR aileron to aileron required to fly accident profile with fault.	Boeing
4a	Spoiler hardover fault	In work	No	Same as #3b except at time 92444	MCA requests that fault be inserted at A/P engage (92415)	Boeing
4b	Spoiler mid extend jam	Requested	No			
5	Spoiler float fault	In work	No	Same as #3b except at time 92444	MCA requests that fault be inserted at A/P engage (92415)	Boeing
6	Slat "float" (assumed actuator detached and/or jammed/cocked slat)	Not available		The position of a floating slat is determined by the airload on the slat and friction within the system. We do not have aero data available for the accident flight airspeed and altitude conditions. The airloads will either extend the slat, retract the slat, or will be insufficient to overcome system friction. Therefore, we believe the airplane level roll response will be bounded by the response to a slat fully extended fault such as #3b above.	OK	Boeing
7	Hardover on one aileron PCU	In work		A hardover of one aileron PCU will result in both aileron PCUs commanding full aileron, spoiler and control wheel hardover. We intend to demonstrate this scenario in the same manner as #3b <b>above by inserting the fault at time 92444.</b>	OK	Boeing
8	Aileron trim runaway	Available now	Yes	Aileron trim runaway can be simulated by manually moving the aileron trim control in the cab during manual flight. This can be done by breaking out at 92444 and manually inputting aileron trim.	OK	Boeing
9	A/P with MCP erroneous selected heading	In work		This scenario will result in the autopilot flying to the erroneous selected heading. This scenario can be simulated initializing the simulator at time 92395, then running open loop. At that point, the autopilot can be engaged and the desired "erroneous" selected heading can be entered on the MCP.	OK	Honeywell
10a	A/P with MCP Selected Heading knob mechanically inoperative, such that it does not transfer pilot commands. (Selected heading window and output to FCC constant regardless of knob movement)	See #9		This scenario has the same effect as #9 above and can be simulated in the same way.	OK	Honeywell
10b	A/P with one or more segments in the MCP selected heading LCD window inoperative leading to improper indication (e.g. displaying 6 instead of 8)	See #9		The result of this fault will be that the apparent value in the heading window can be different than the value transmitted to the EADI for display of the heading bug and to the FCC for use in autopilot heading select mode. Although we will not be able to simulate a different value in the selected heading window, we believe that this fault can be simulated in the same way as #9 above.	OK	Honeywell
10c	A/P with MCP internal processor or MUX fault resulting in dissimilar values between the selected heading window and the selected heading command to the FCC	See #9		This scenario has the same effect as #10b and can be simulated in the same manner as #9.	OK	Honeywell

11	A/P Actuator hardover	In work		This scenario will result in a "hardover" to the autopilot actuator authority limit (60 deg with the autopilot force limited not engaged). We can simulate this scenario by introducing the fault and "breaking out" simultaneously at 92415 (A/P initial engage)	OK	Boeing
12a	A/P Actuator ARM Solenoid valve failed open with A/P disconnected	Not applicable to M-Cab		With the arm solenoid open, the autopilot mod piston can move in response to FCC commands, but as the detent solenoid is not open, the mod piston is not coupled to the ailerons and the A/P actuator cannot command aileron motion. We do not believe it is necessary to simulate this scenario.	OK	Boeing
12b	A/P Actuator Detent Solenoid failed open with A/P disconnected	Not applicable to M-Cab		The arm and detent solenoids are in series. If the arm solenoid is closed, no hydraulic fluid is available to allow the detent pistons to couple the mod piston to the ailerons. The A/P actuator cannot command aileron motion. If this fault exists when the autopilot is trying to engage, the engagement may occur with a jolt as the mod piston would be coupled to the ailerons before the position synchronization is complete. We do not believe it is necessary to simulate this scenario.	OK	Boeing
12c	A/P Actuator both arm and detent solenoid open with A/P disconnected	Not applicable to M-Cab		This is the normal condition when the autopilot is engaged. The transfer valve spool moves the mod piston moves in response to commands from the FCC and the detent pistons are pressurized to couple the actuator to the ailerons. If the autopilot is not engaged, the FCC commands the transfer valve to hold the autopilot actuator in the neutral (aileron failed) position. Normal autopilot breakout is still available to override the autopilot. Without pilot intervention, the net result would be the same as letting go of the wheel and letting it center. We do not believe it is necessary to simulate this condition.	OK	Boeing
12d	A/P Actuator triple fault (arm and detent solenoid open, transfer valve jam off center)	See #11		This triple fault will result in an A/P actuator hardover. The force limit of the actuator still operates normally. The hardover condition is the same as #11 above.	OK	Boeing
12e	A/P Actuator quadruple fault (arm and detent solenoid open, transfer valve jam, pressure regulator jam)	In work		This quadruple fault will result in an A/P actuator hardover. Because the pressure regulator is jammed, the relief valve operates and limits detent piston pressure. The wheel force required to overcome the actuator increases from 16 lbs of wheel to approximately 20 lbs of wheel.	OK	Boeing
12f	A/P Actuator quadruple fault (arm and detent solenoid open, transfer valve jam, relief valve jam)	See #11		This quadruple fault will result in an A/P actuator hardover. Although the relief valve is jammed (stuck to the pressure regulator slide), the pressure regulator limits detent piston pressure to the normal level. The wheel force required to overcome the actuator is the normal 16 lbs of wheel.		Boeing
12g	A/P Actuator quintuple fault (arm and detent solenoid open, transfer valve jam, pressure regulator and pressure relief valve)	In work		This quintuple fault will result in an A/P actuator hardover. In this scenario, neither the pressure regulator nor the relief valve can reduce the detent piston pressure which reaches hydraulic system pressure (3000 psi). Wheel force required to overcome the actuator increases from 16 lbs of wheel to approximately 80 lbs of wheel.	MCA requests to observe this fault (feel the forces) or the highest forces possible in the M-cab.	Boeing
13	A/P with IRU shutdown	Not applicable to M-Cab		The response of the autopilot to an IRU shutdown is to disconnect. We do not believe it is necessary to simulator this scenario.	OK	Honeywell
14	A/P with Erroneous R IRU output of straight and level flight during bank (no NCD or fail warn transmitted)	In work		The autopilot will command aileron to its authority limit (20 deg with aileron force limiter). If the airplane heading crosses the selected heading the autopilot command will reverse. M-Cab simulation will not accurately reflect the wheel forces in this situation.	OK	Honeywell
15a	A/P with Erroneous L IRU output of roll rate with all other parameters correct (separately and then see if possible to do at same time as above fault)	Not applicable to M-Cab		Autopilot A does not use L IRU roll rate as an input. This fault has no effect on the operation of autopilot A.	OK	Honeywell
15b	A/P with R IRU output of NCD for roll rate	Not applicable to M-Cab		The response of the autopilot to R IRU output of NCD for roll rate is to disconnect. We do not believe it is necessary to simulate this scenario.	OK	Honeywell
16	Autopilot spoiler sensor fault (erroneous value)	Not applicable to M-Cab		The sensed value of spoiler angle is only used by the autopilot when the flaps at 30 or beyond. This fault would have no effect on the operation of the autopilot for the accident flight.	OK	Honeywell
17	Failure of bank angle limit function in autopilot	See #14		No condition has been identified that could lead to this fault without causing an FCC shutdown. However, if it did occur, the extreme result would be an autopilot actuator hardover as the FCC seeks to achieve an excessive roll angle. As the aileron force limiter is engaged, the hardover would result in wheel offset to 20 degrees.	OK	Honeywell
18	Other FCC internal faults	See #11 or #14		No condition has been identified that could lead to this fault without causing an FCC shutdown. However, if it did occur, the extreme result would be an autopilot actuator hardover. As the aileron force limiter is engaged, the hardover would result in wheel offset to 20 degrees (AFL eng) or 60 deg (AFL not engaged).	OK	Honeywell
19	FD behavior with erroneous selected heading data from MCP	In work		We intend to implement this scenario as the part of #21 below. The desired "erroneous" selected heading can be entered using the MCP.	OK	Boeing
20	FD behavior with erroneous roll rate data from IRU	In work		The roll rate error will effectively reduce or increase the maximum bank angle for the maneuver (depending upon the sign of the roll rate error). It will also result in a steady state heading error once the turn was complete. In order for the aileron command to remain at zero the heading error and roll rate error will cancel.	OK	Honeywell

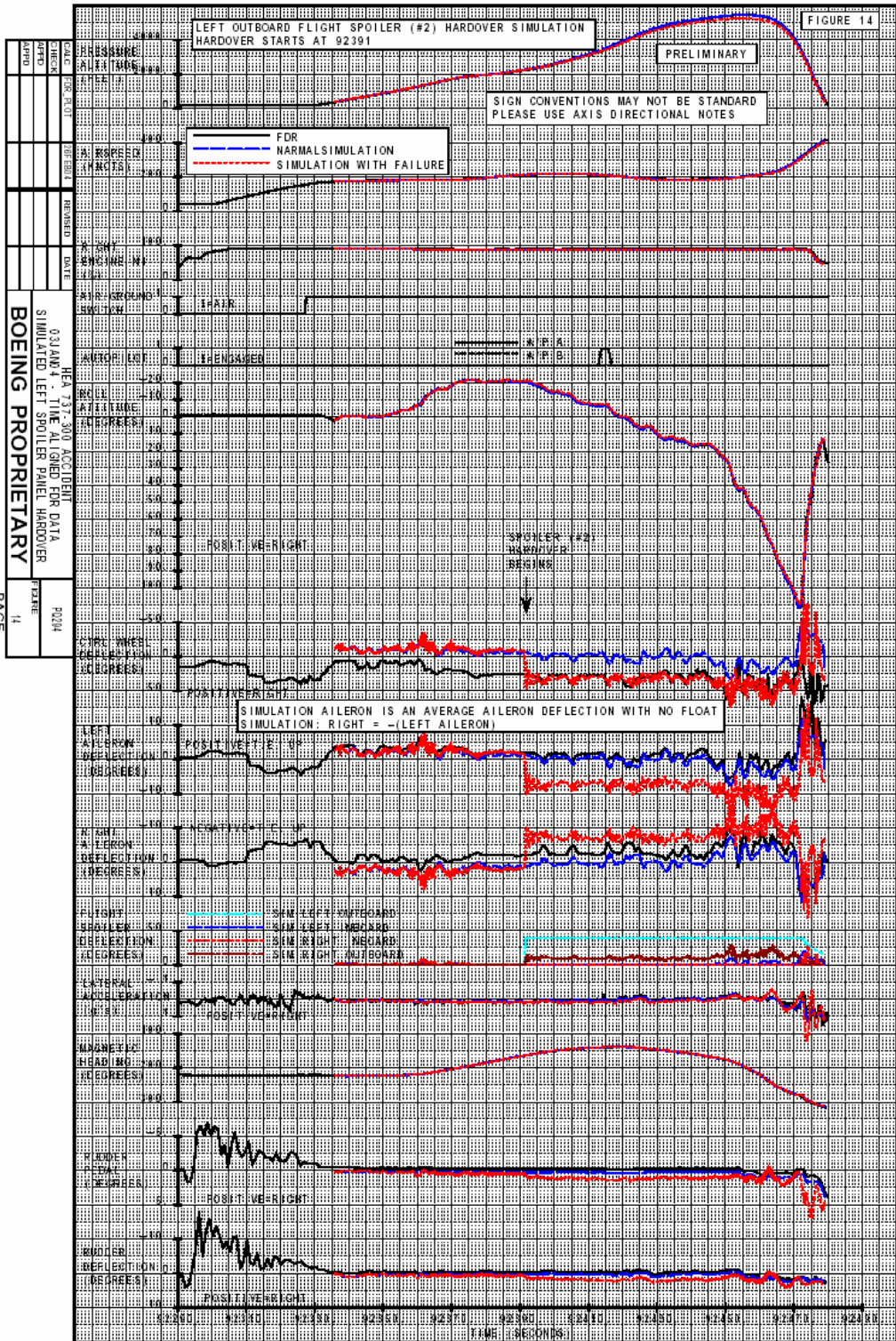
#### 1.16.1.4. Simulated Failures:











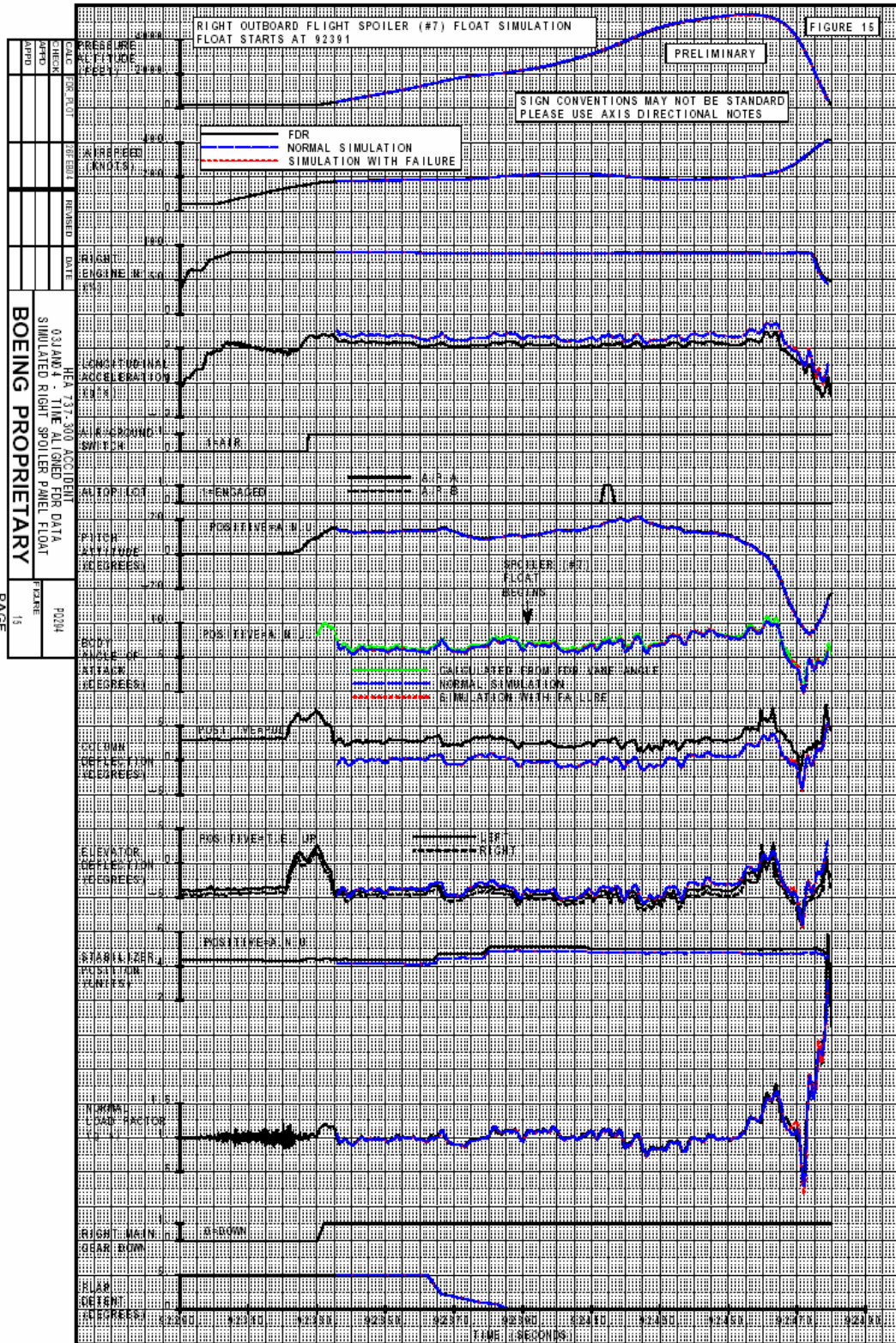
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TIME	11:00	TIME ALIGNED PER DATA	
FROM	0000	SIMULATED LEFT SPOILER PANEL HARDOVER	
TO	0000		
FILE	11		

BOEING PROPRIETARY

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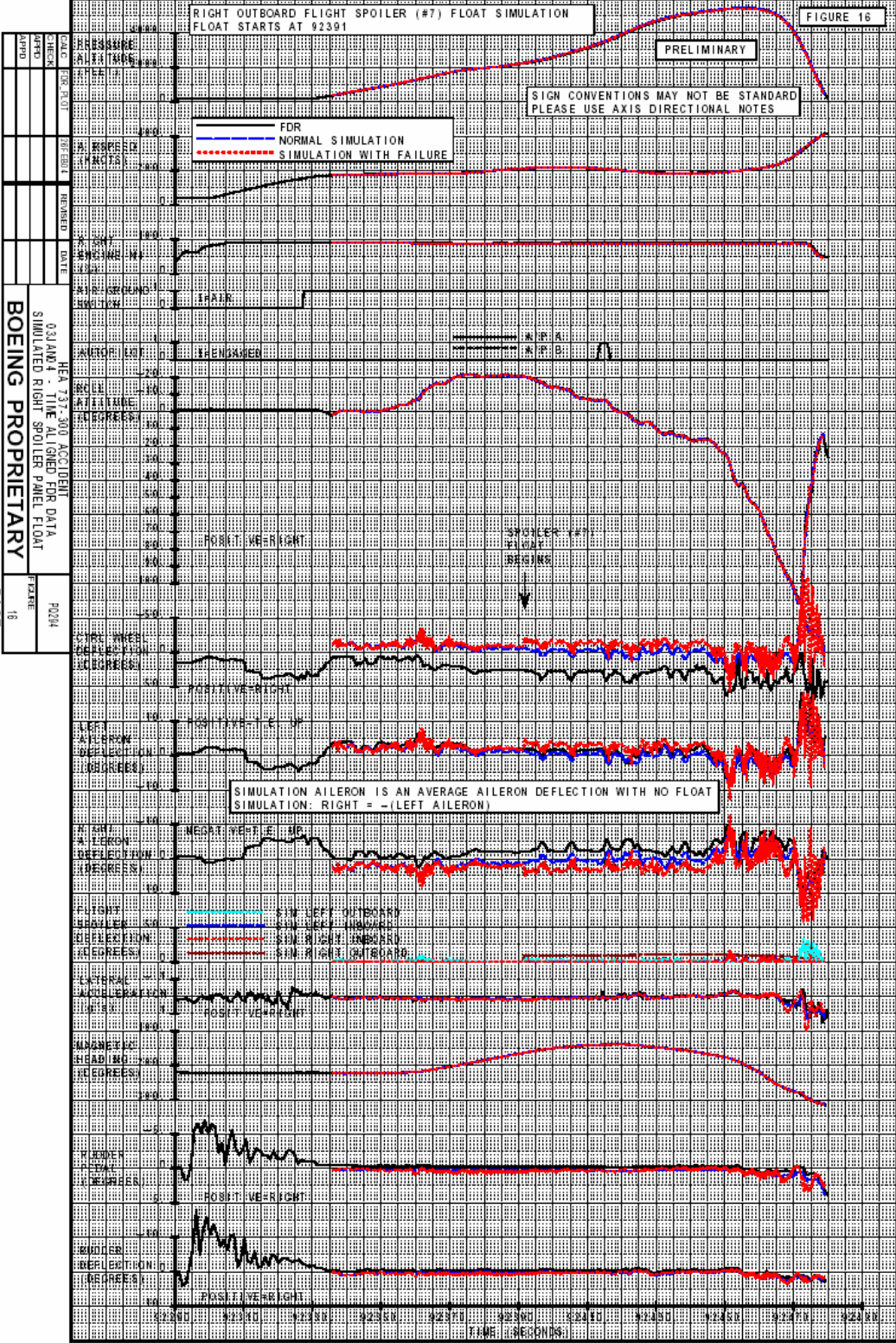
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Right outboard flight spoilers (#7) Float simulation (floats starts at 92391)

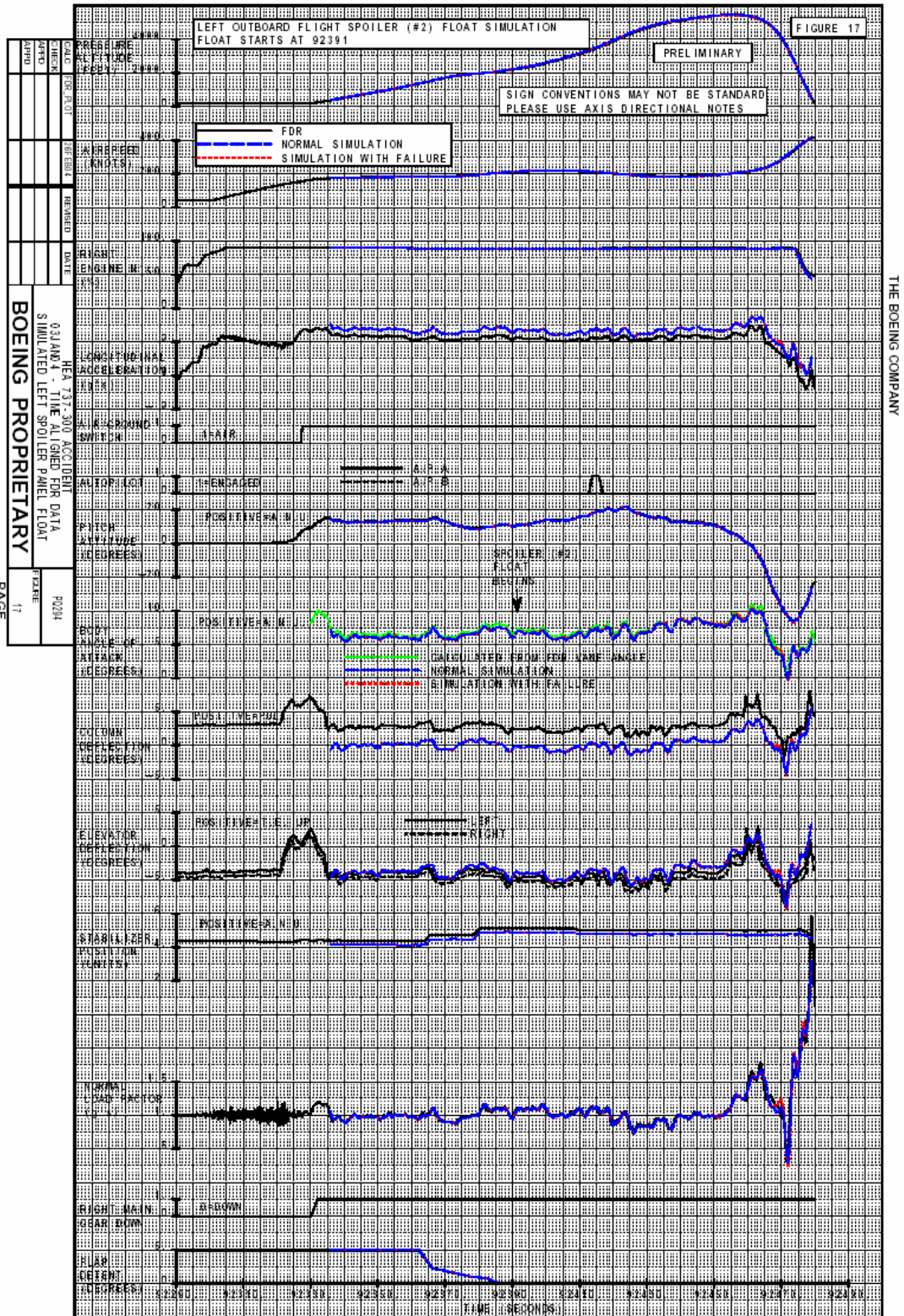


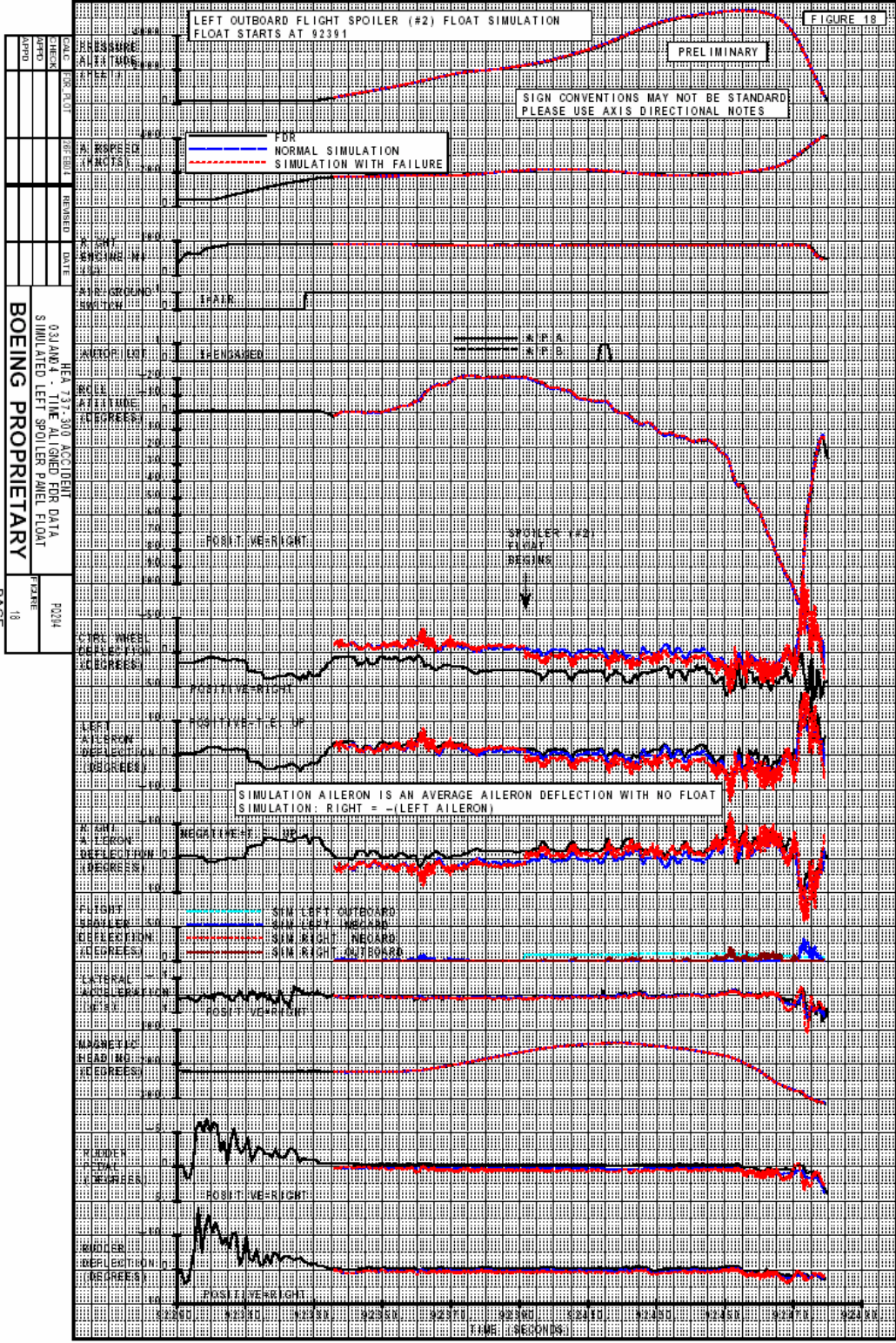
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HEAVY GUSTS  
 03 JAN 14 TIME ALIGNED FOR DATA  
 SIMULATED RIGHT SPOILER PANEL FLOAT  
 BOEING PROPRIETARY  
 PAGE 15



Left outboard flight spoilers (#2) Float simulation (floats starts at 92391)





THE BOEING COMPANY

DATE	TIME	INITIALS	REVISION

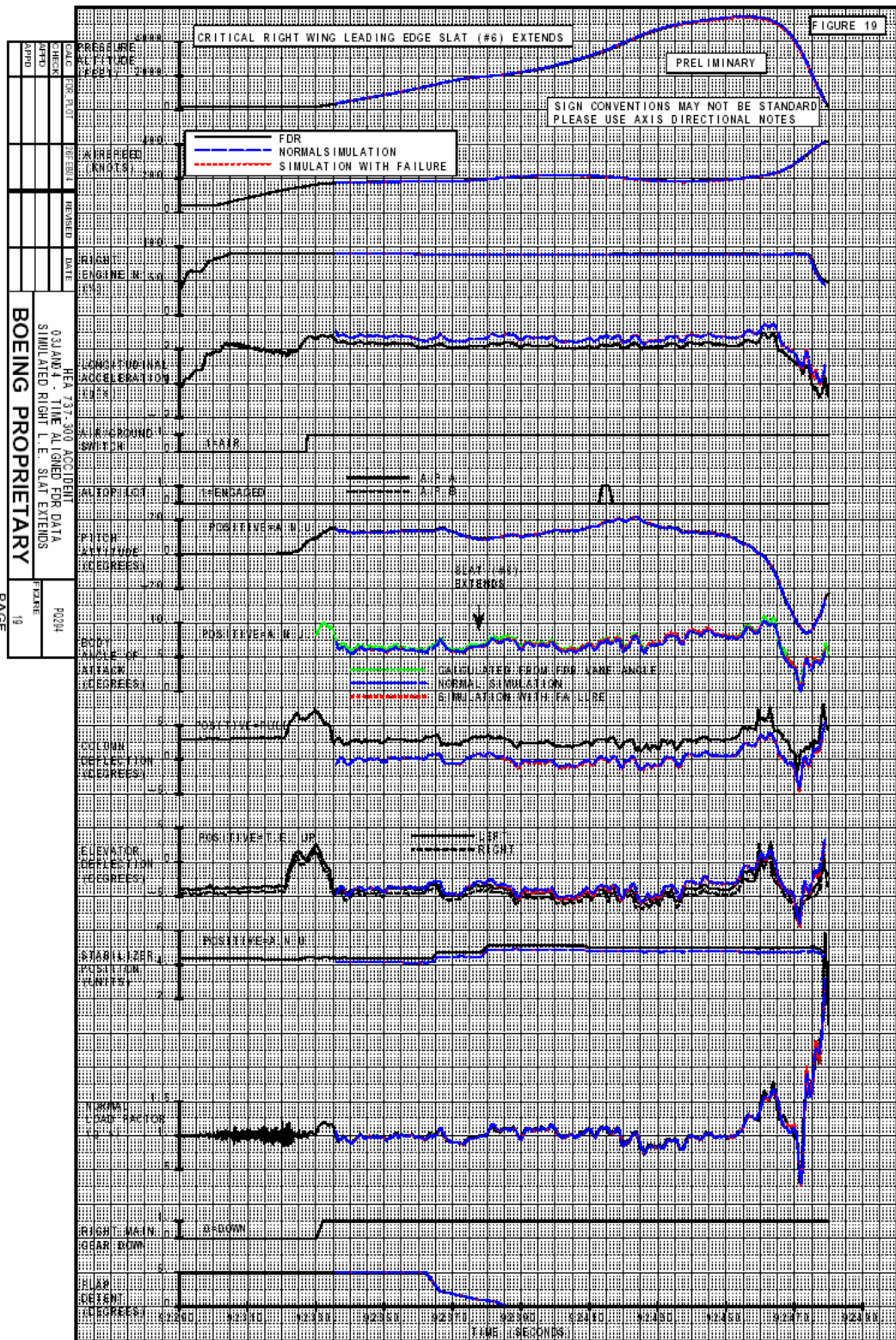
CHECKED: _____ APPROVED: _____ DRAWN: _____	DATE: _____ TIME: _____ INITIALS: _____
---	---

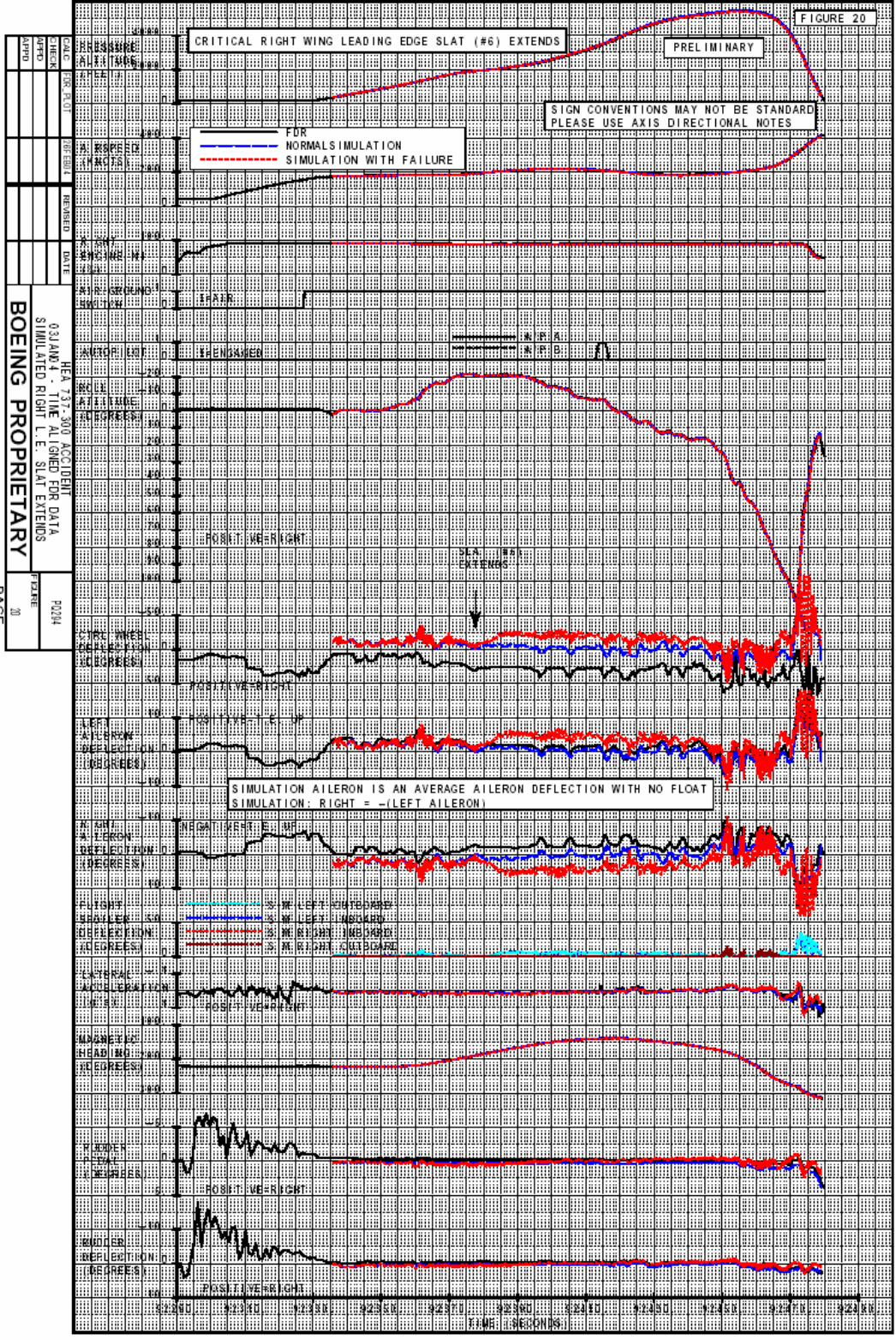
TITLE: MEA 737-300 ACCIDENT 03 JAN 4 - TIME ALIGNED FOR DATA SIMULATED LEFT SPOILER PANEL FLOAT	PAGE: 18 OF: 18
---	--------------------

BOEING PROPRIETARY

Critical right wing leading edge slat # 6 extends

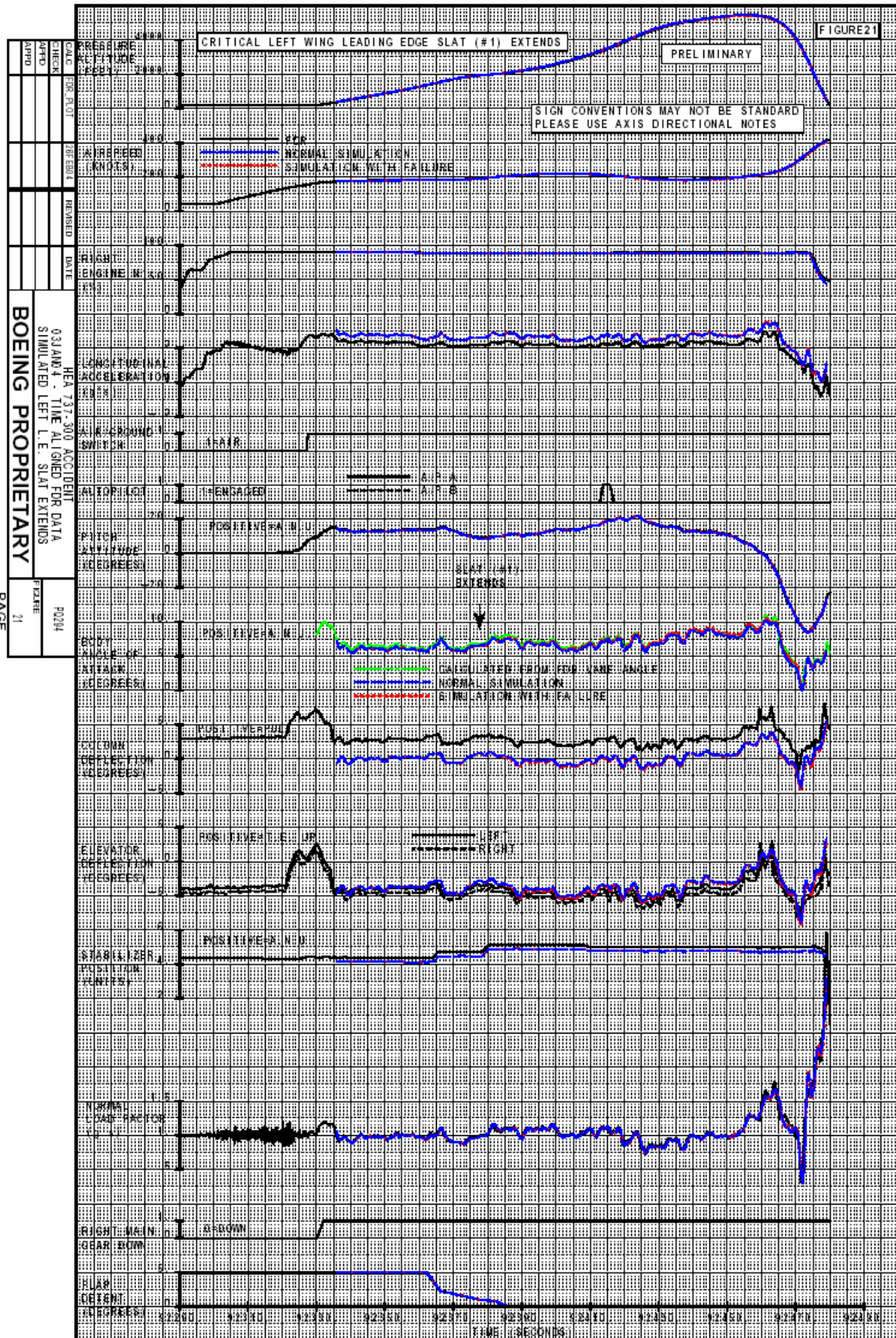


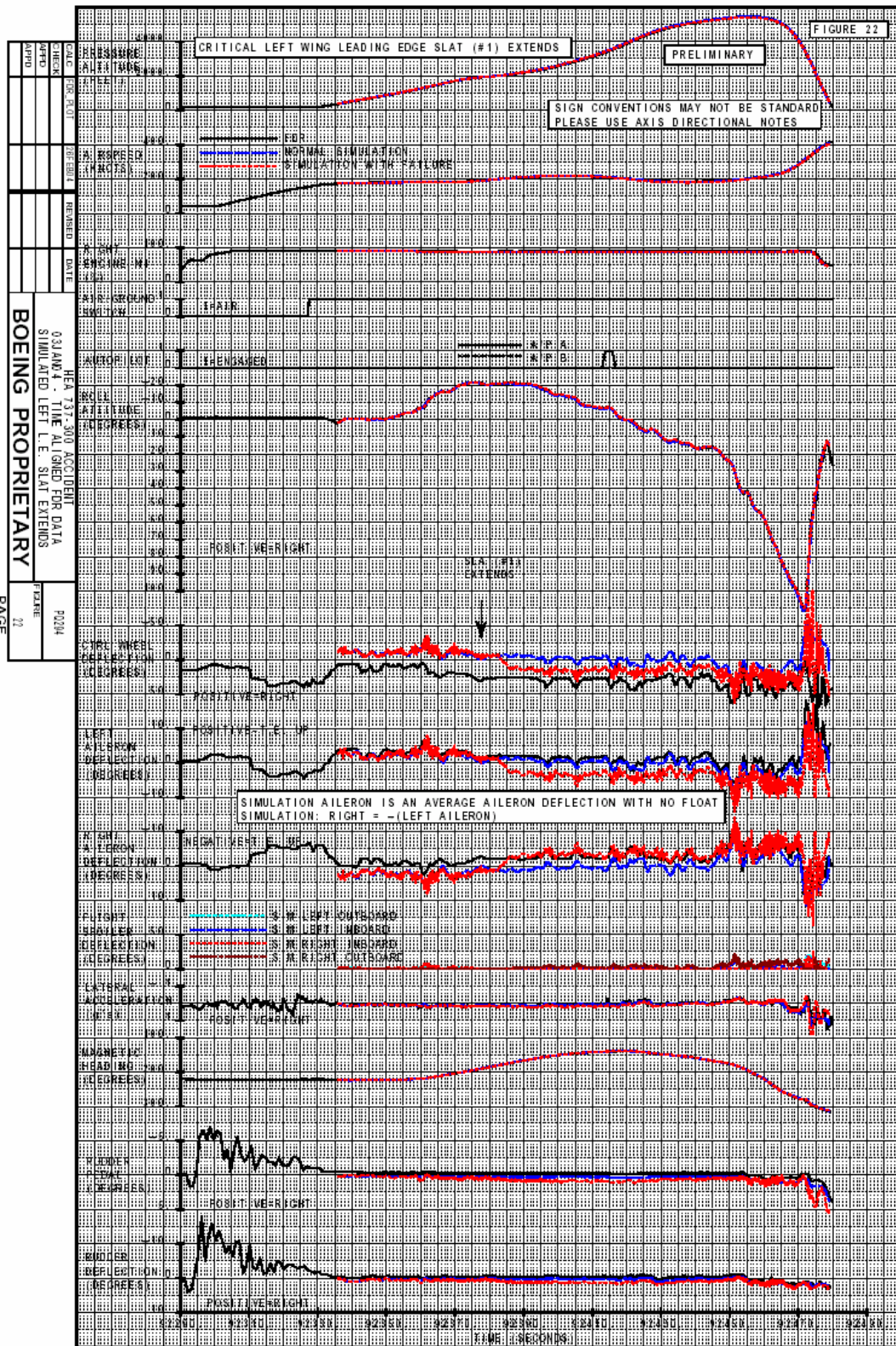




DATE	TIME	REVISION	DESCRIPTION
03JAN04	11:00	1	HEA 737-300 ACCIDENT
03JAN04	11:00	1	TIME ALIGNED FOR DATA
03JAN04	11:00	1	STIMULATED RIGHT L.E. SLAT EXTENDS
03JAN04	11:00	1	BOEING PROPRIETARY
03JAN04	11:00	1	PAGE 20

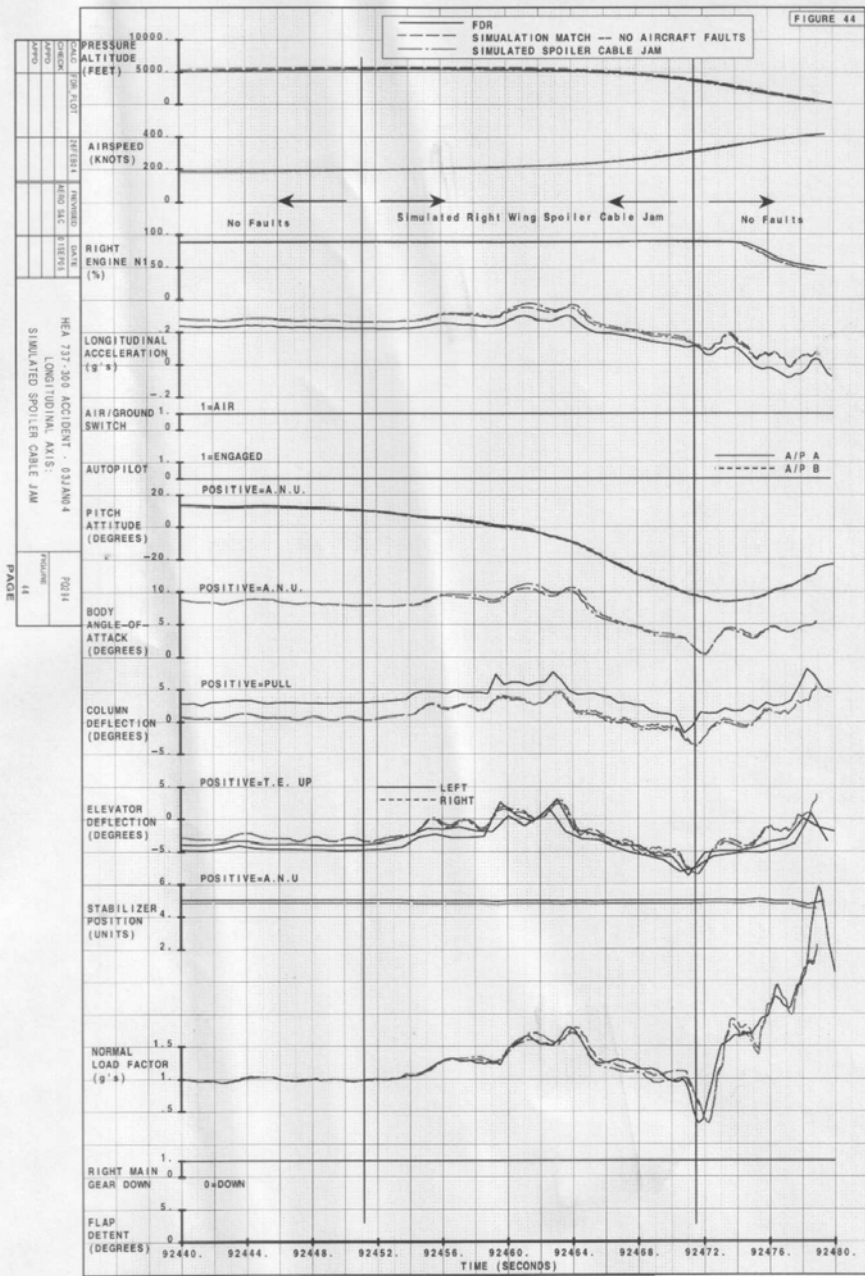
Critical left wing leading edge slat # 1 extends



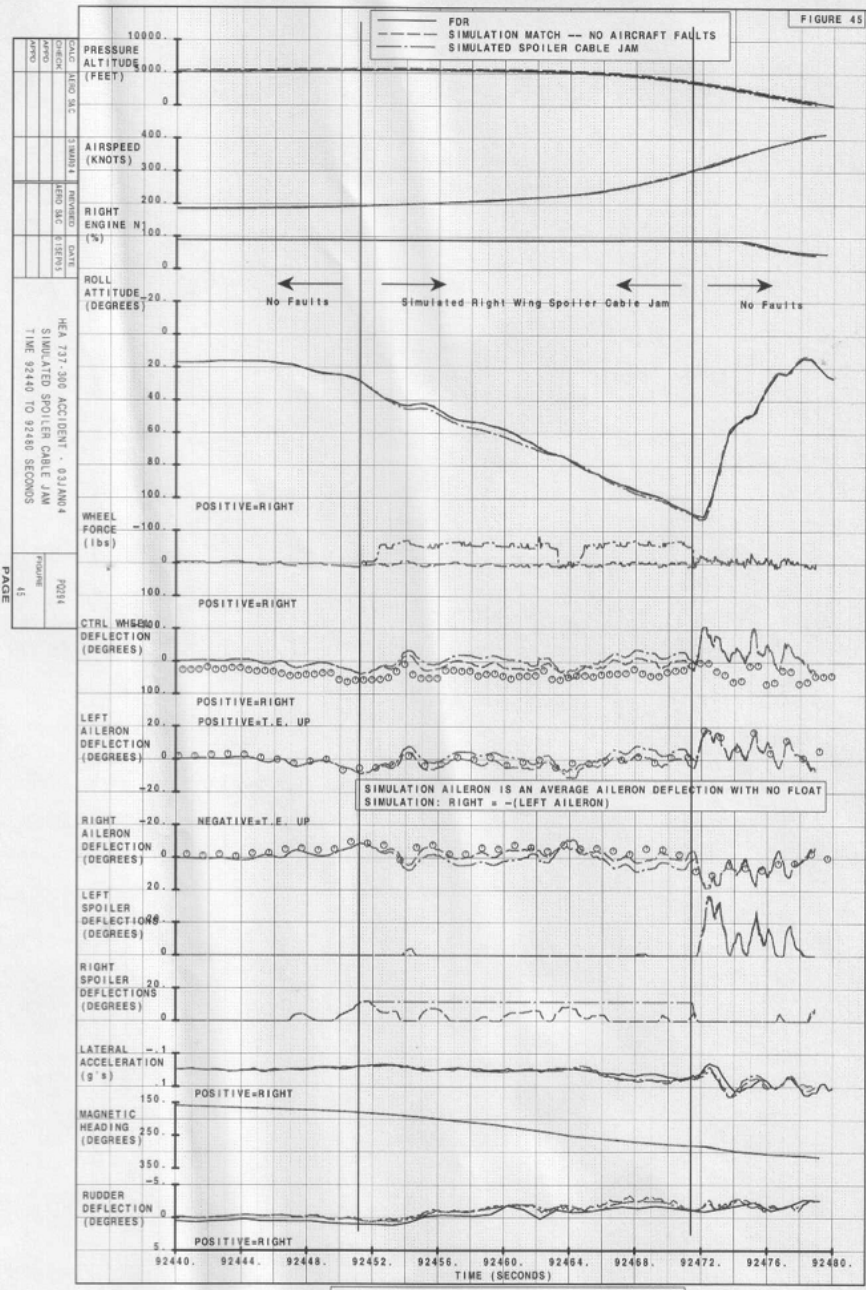


Scenario 10 - Spoiler wing cable jam (Spoiler wing cable jam) offset of the neutral position at time 92450 (maximum wheel deflection).and clears at 92472

Longitudinal Axis, simulated right wing spoiler cable jam

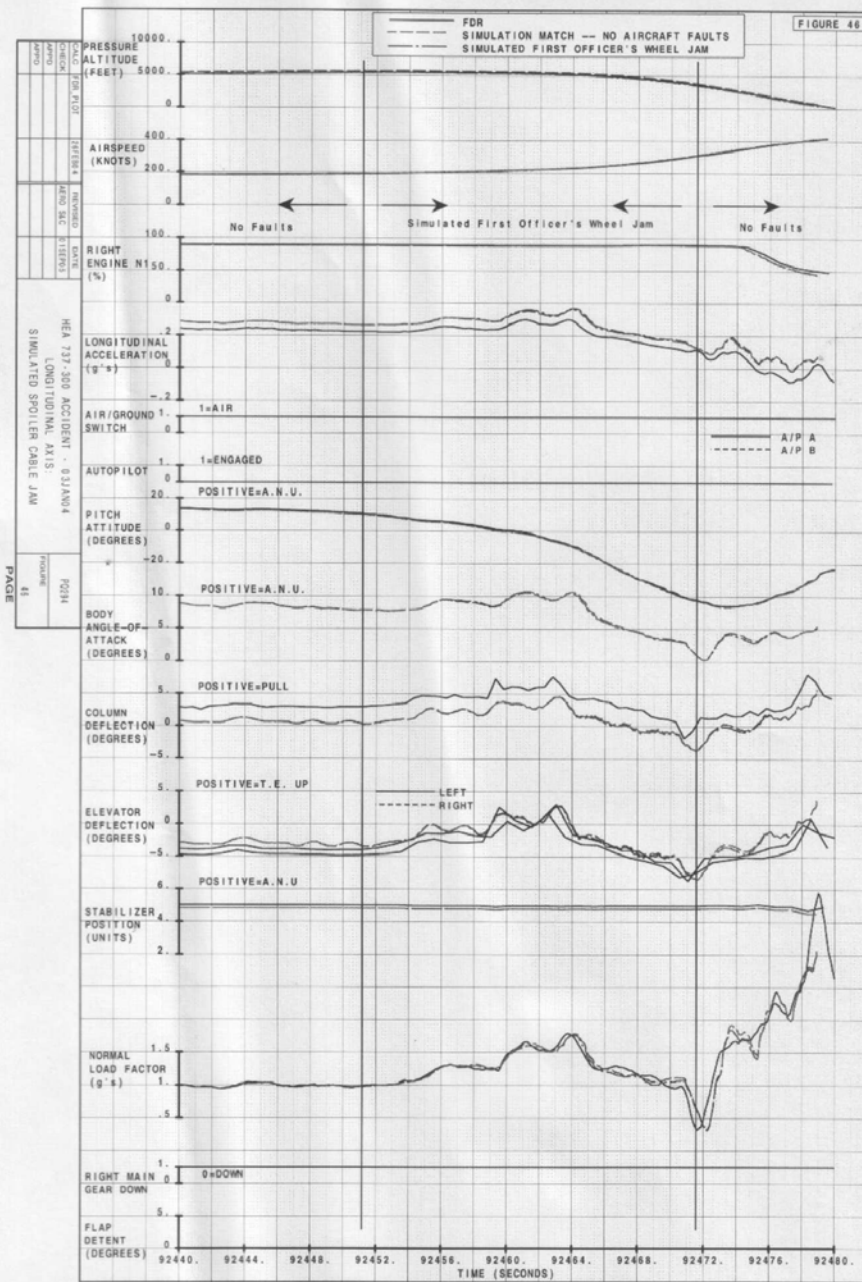


Lateral Axis, simulated right wing spoiler cable jam



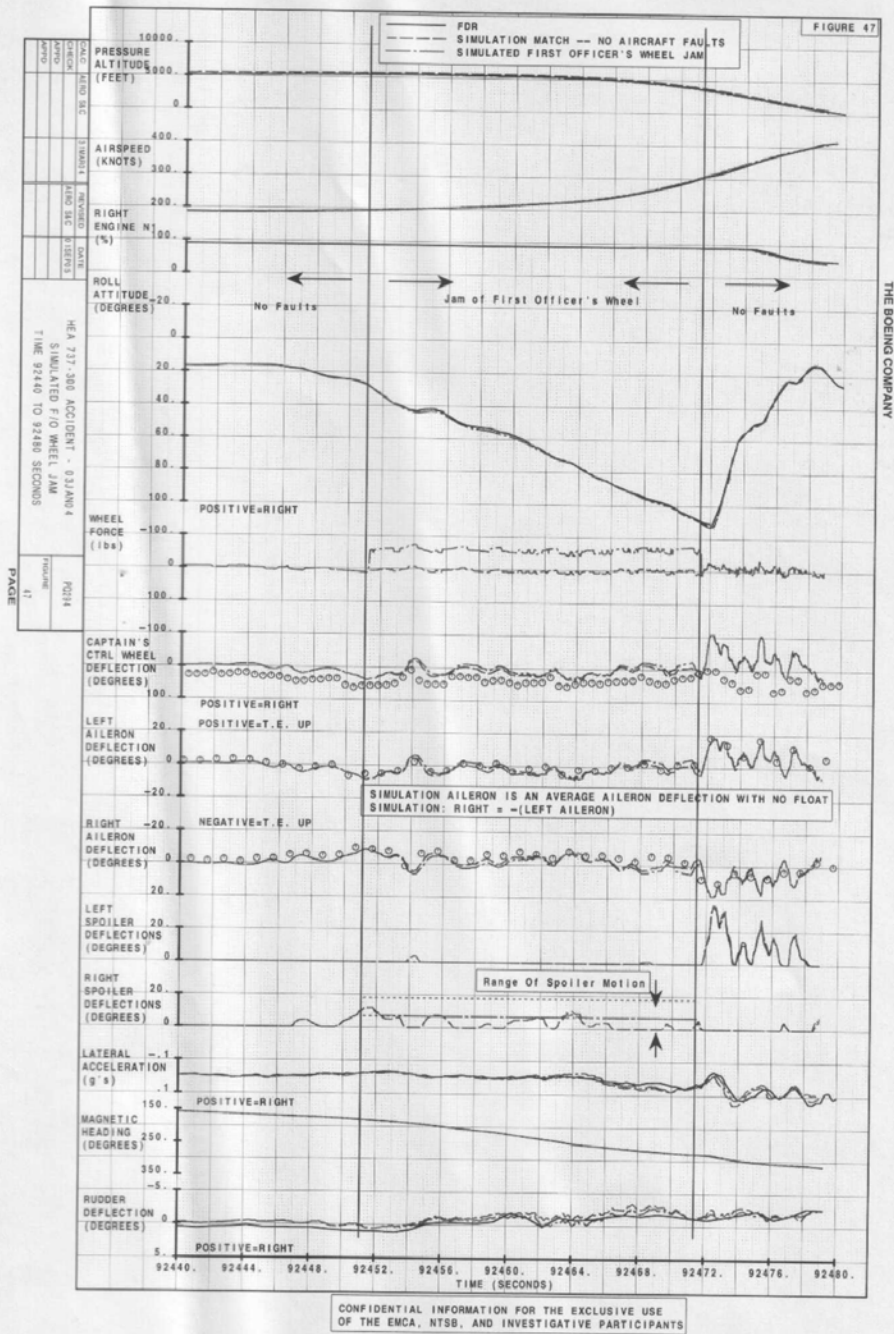
Scenario 10a - F/O wheel jam (F/O wheel jam) offset of the neutral position at time 92450 (maximum wheel deflection).and clears at 92472

Longitudinal Axis, simulated F/O's wheel jam:

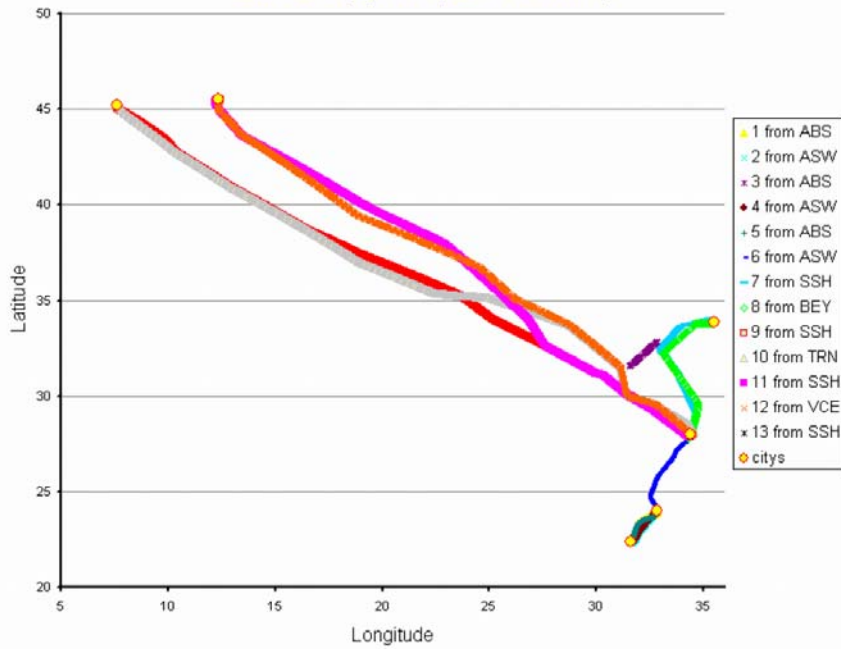




Lateral Axis, simulated F/O's wheel jam:



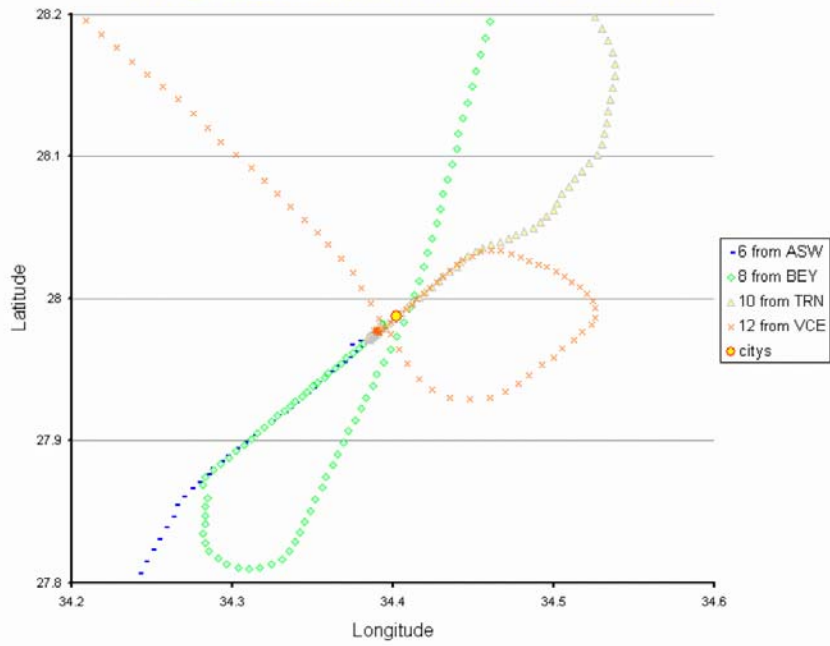
## SU-ZCF – FDR Lat/Long Data *All Flights (25 hours)*



Boeing Proprietary

# SU-ZCF – FDR Lat/Long Data

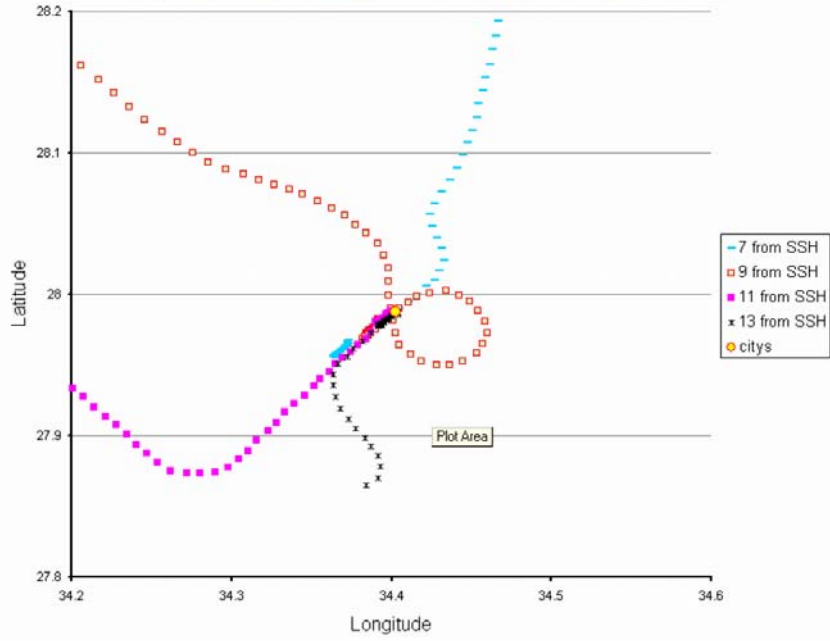
*All Arrivals into Sharm el-Shiekh*



Boeing Proprietary

# SU-ZCF – FDR Lat/Long Data

*All Departures from Sharm el-Shiekh*



Boeing Proprietary

## FDR 25 Hour Data

### *Observations*

- SU-SCF Flight 9 departure from SSH
  - *Departed Rwy 4*
  - *Circling departure to over-fly VOR*
  
- Use of TOGA on takeoff
  - SU-ZCF: TOGA typically engaged for ~2 sec*
  - SU-ZCD: TOGA typically engaged for 1-2 minutes*

Boeing Proprietary

## SU-ZCF – FDR 25 Hour Data

### *TOGA Observations*

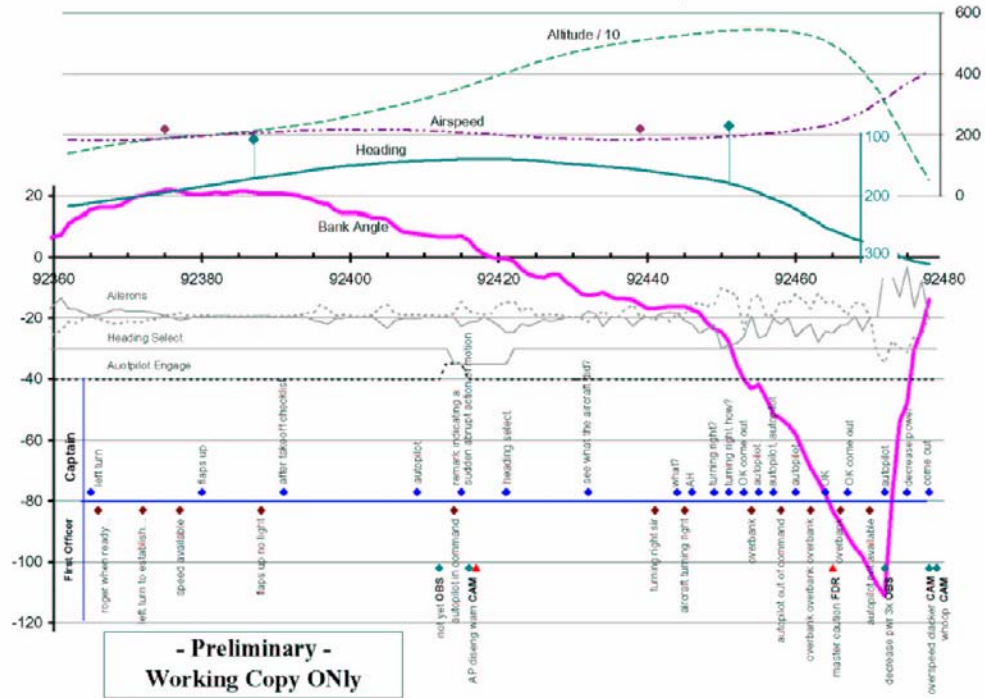
Flight	Both F/D ON?	Normal looking A/T Takeoff	First TOGA Push (1)	If Second TOGA Push (1)
1	YES	YES	1	2
2	YES	YES	0	
3	YES	YES	2	
4	NO	YES	0	
5	YES	YES	2	
6	YES	YES	1	
7	YES	YES	1	
8	YES	YES	2	
9	YES	YES	2	1
10	YES	YES	0	
11	YES	YES	2	
12	YES	YES	2	
13	YES	YES	2	

(1) Number of samples recorded for TOGA\_FCC (sample Intvl=1 sec)

Boeing Proprietary

1.16.1.6. FDR-CVR Overlay  
 FDR-CVROverlay.pdf, FDR-CVR Overlay 3R2.pdf (21-June 2004,  
 040301 Flash 737 Cairo Mtg (public release version).pdf)

### FDR/CVR Overlay



Boeing Proprietary

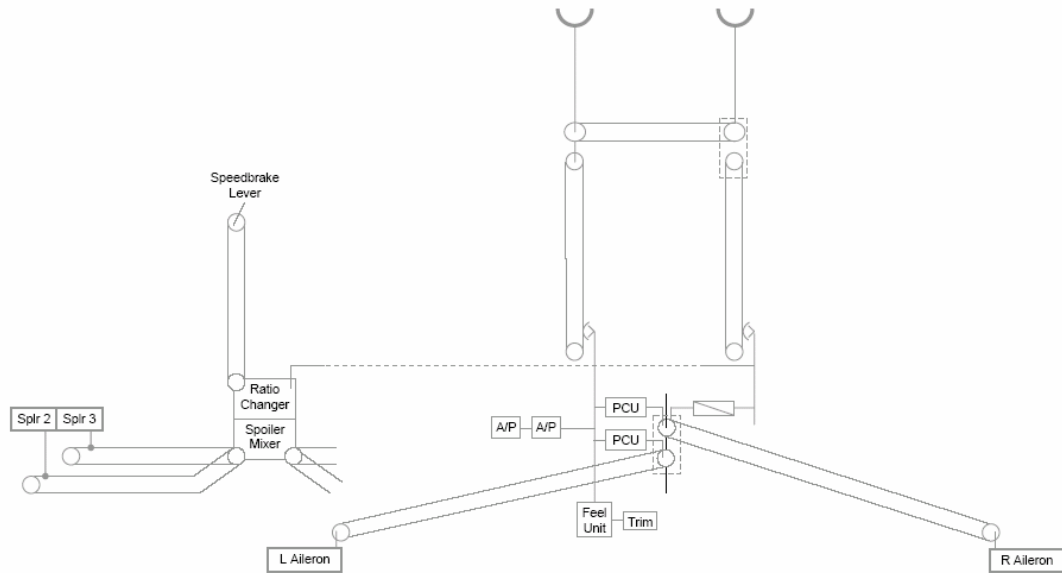
1.16.1.7. Ailerons system

IPC wheel posn xducer PW.pdf (Details about the wheel posn xducer- Part Catalog Maintenance)

**Boeing Proprietary information and will not be available for public use**

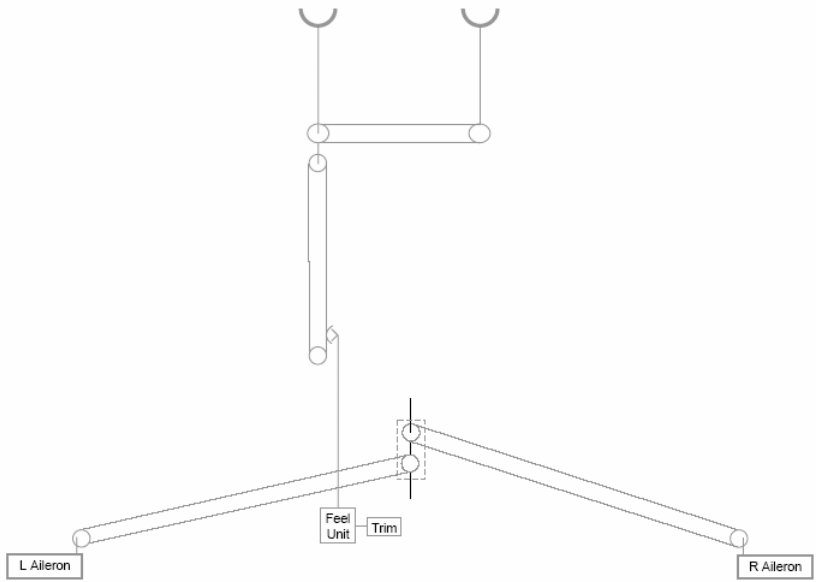


## Lateral Control System *Function Schematic*



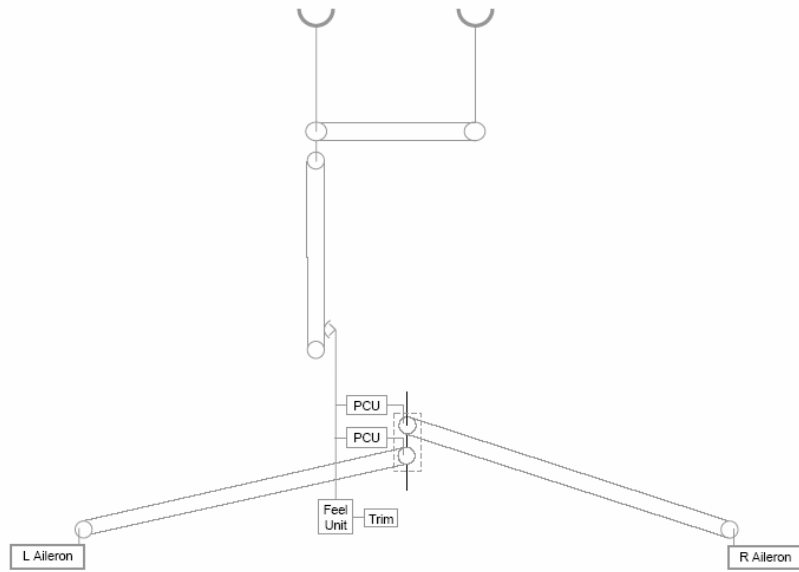
# Lateral Control System

## *Function Schematic*



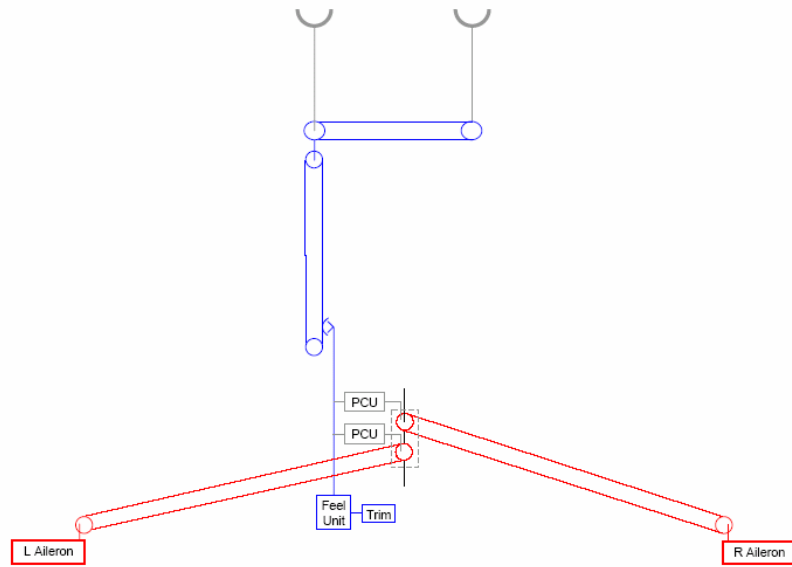
# Lateral Control System

## *Function Schematic*



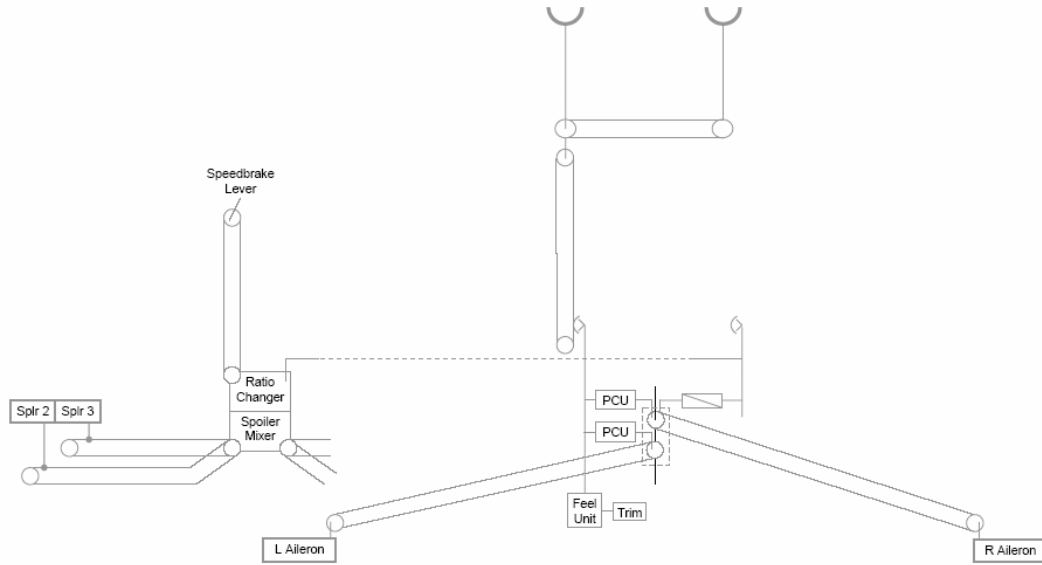
# Lateral Control System

## *Function Schematic*

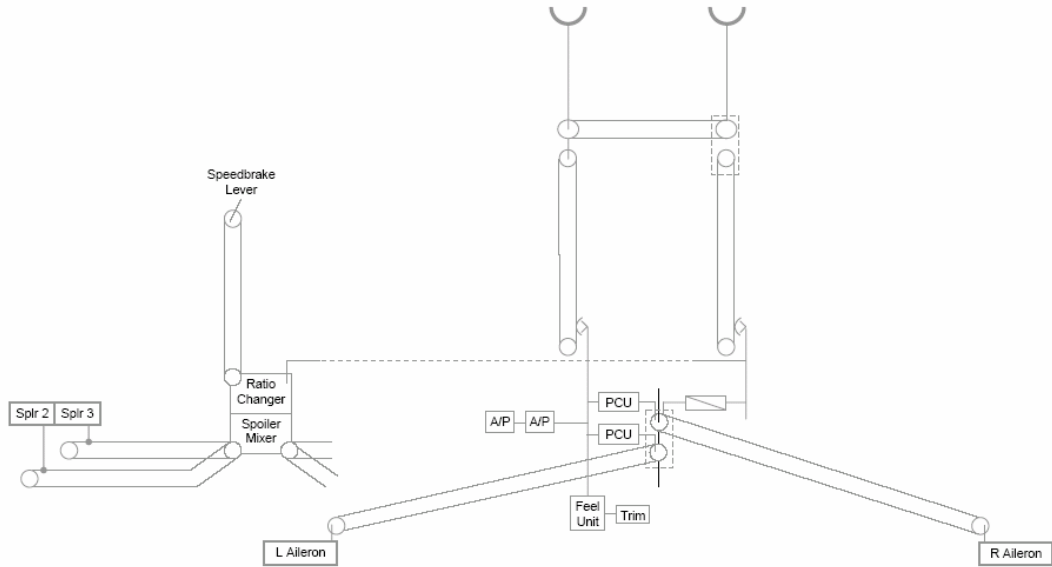


# Lateral Control System

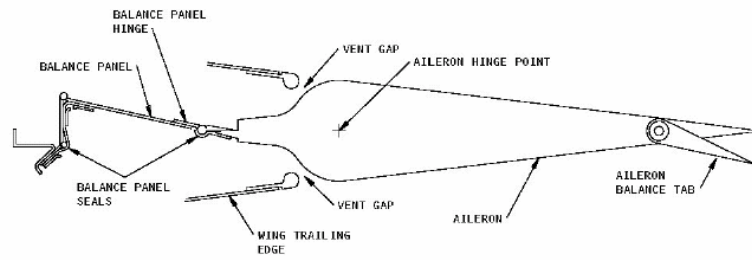
## *Function Schematic*



# Lateral Control System *Function Schematic*



# Aileron



Note

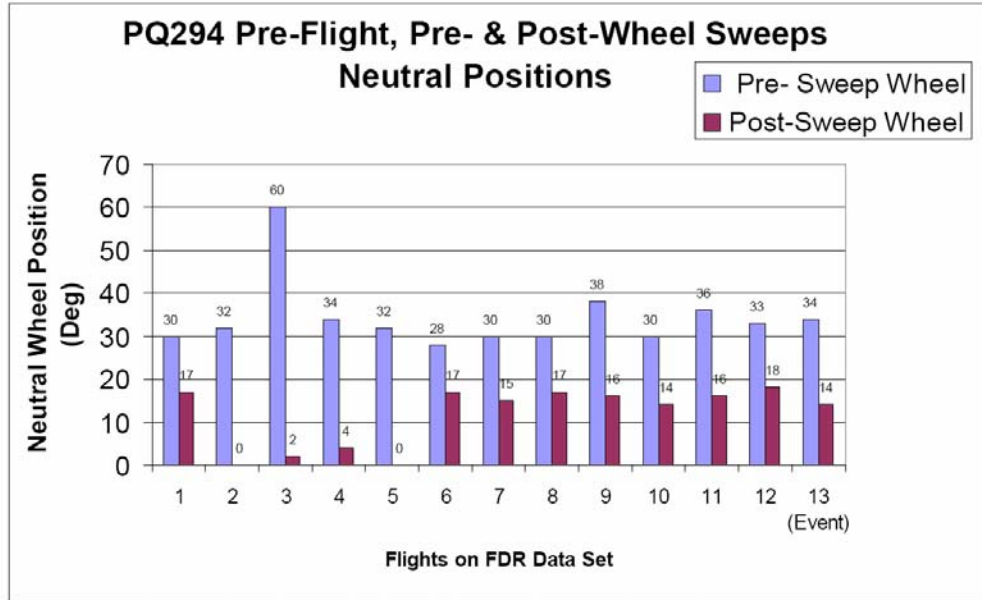
Remaining information is Boeing proprietary information and will not be available for public use

Aileron PCU Control Valve.ppt

**Boeing Proprietary information and will not be available for public use**



## PQ294 FDR Control Wheel Position *Wheel Sweep Data*

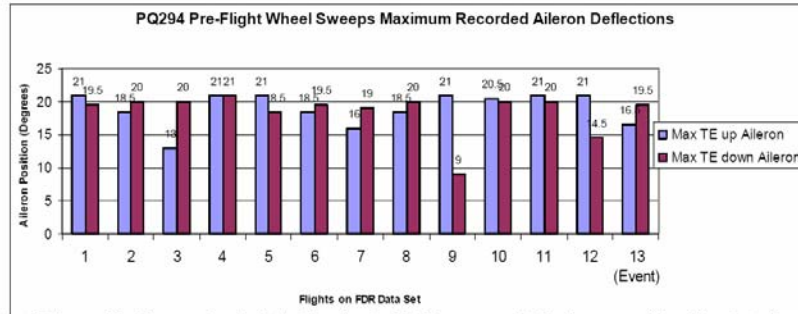
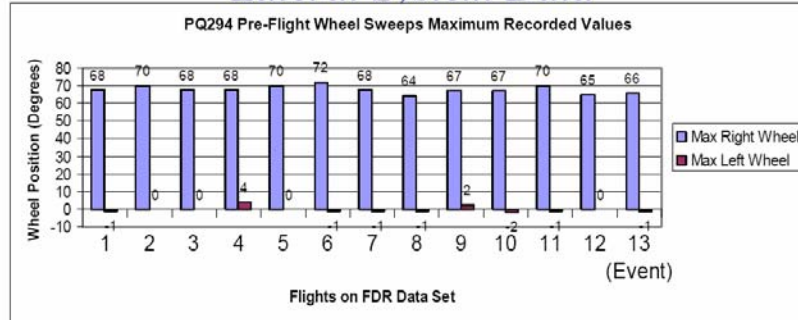


Notes: Wheel Sweeps for flights 2, 3, 4, and 5 where left wheel first, then right wheel.  
 Wheel Sweeps for flights 1 and 6 - 13 where right wheel first, then left wheel.  
 Sister ship PQ481 did not have a valid FDR wheel parameter (binary data were all zeros).

Boeing Proprietary

# PQ294 FDR Control Wheel Position

## *Lateral System Data*

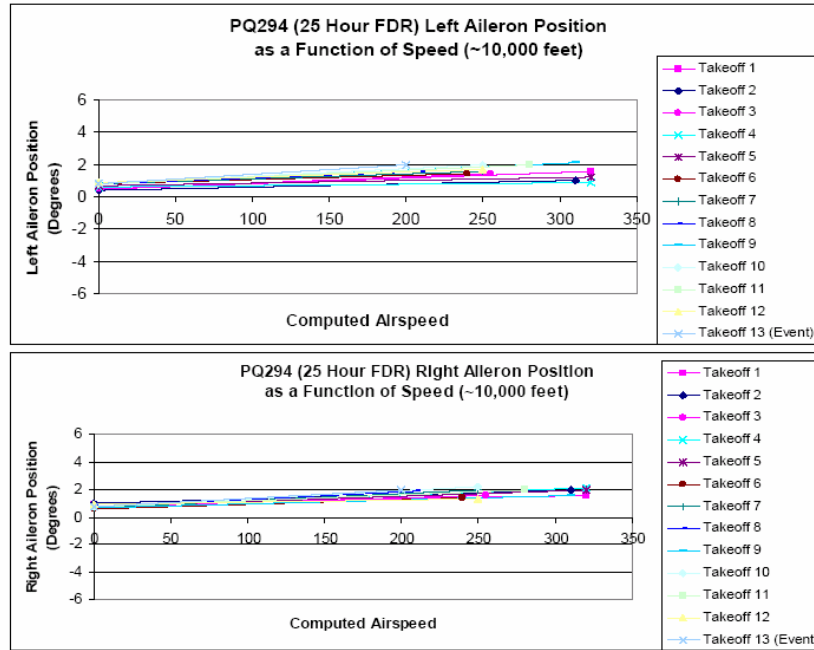


Notes: Maximum wheel deflection is +/- 87.5 degrees, 107.5 degrees with cable stretch  
 Maximum aileron deflection is +/- 20 degrees

Boeing Proprietary

# PQ294 FDR Aileron Position

## *Aileron Float from Airload*

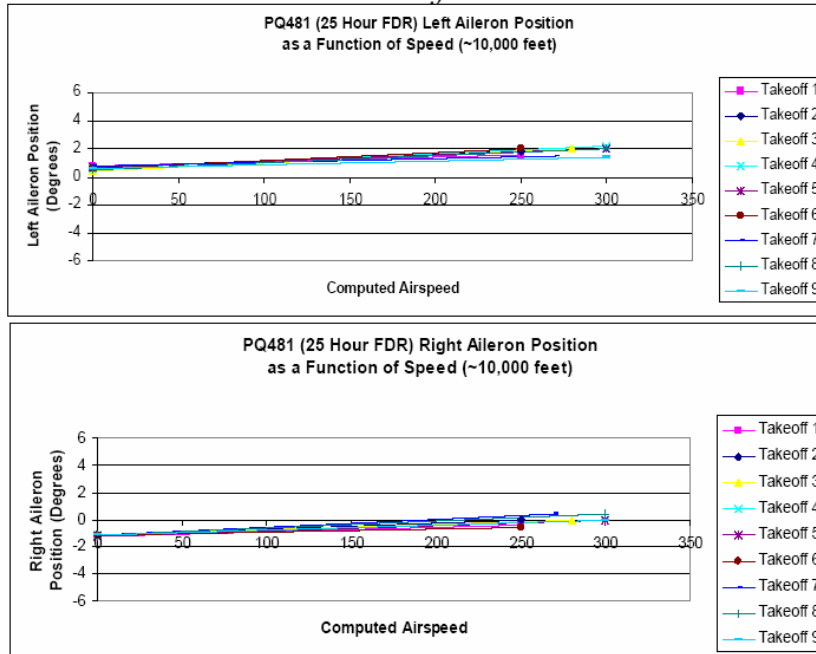


Note: Positive Aileron is Trailing Edge Up

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# PQ481 FDR Aileron Position

## *Aileron Float from Airload*



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M-Cab Wheel (Flight Director Results Boeing.xls)

**Boeing Proprietary information and will not be available for public use**

Force vs Wheel.ppt

**Boeing Proprietary information and will not be available for public use**

Cor8tmp PCU correction.ppt

**Boeing Proprietary information and will not be available for public use**

# Aileron PCU Field Note Summary

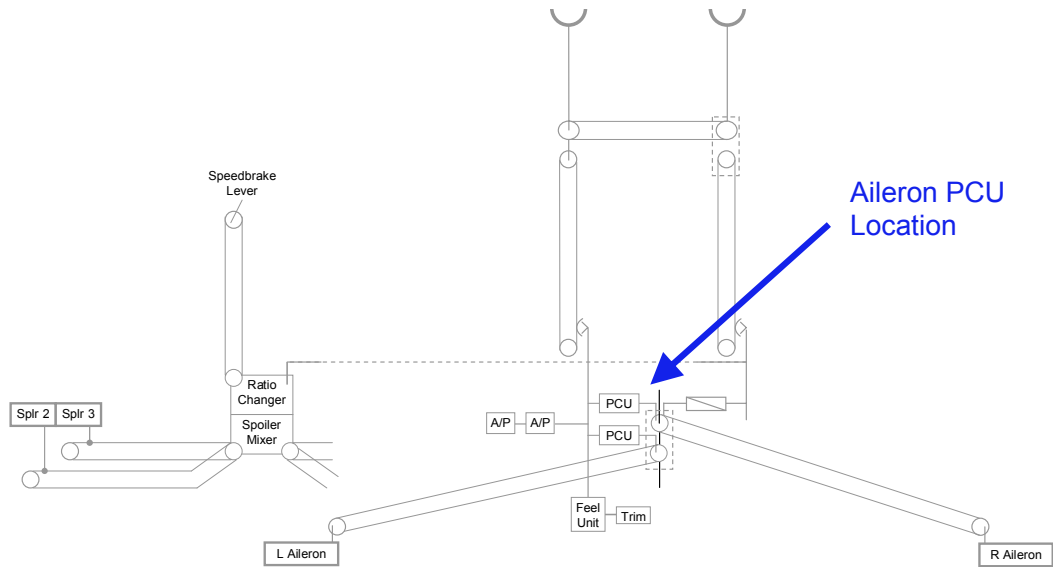
- Recovered 25 Jan 04 (day 23)
- Stored in seawater on board
- Rinsed in freshwater on shore
- Stored at Sharm el-Sheikh airport until shipped to Seattle
- EQA conducted 25-26 Jan 05



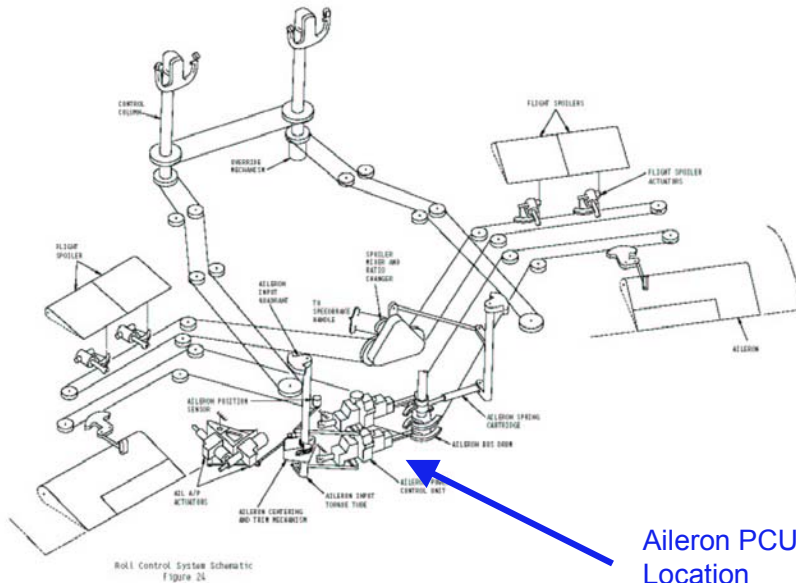
Photos taken Jan 04 onboard recovery ship



# Lateral Control System *Function Schematic*



# Lateral Control System



Aileron PCU Location

**BOEING**  
737-300/400/500  
MAINTENANCE MANUAL

EFFECTIVITY AIRPLANES WITH EFIS  
06  
22-11-01  
Page 74  
Nov 12/01  
ENGINE PROPRIETARY - Copy right (C) - UNPUBLISHED WORK - SEE TITLE PAGE FOR DETAILS.

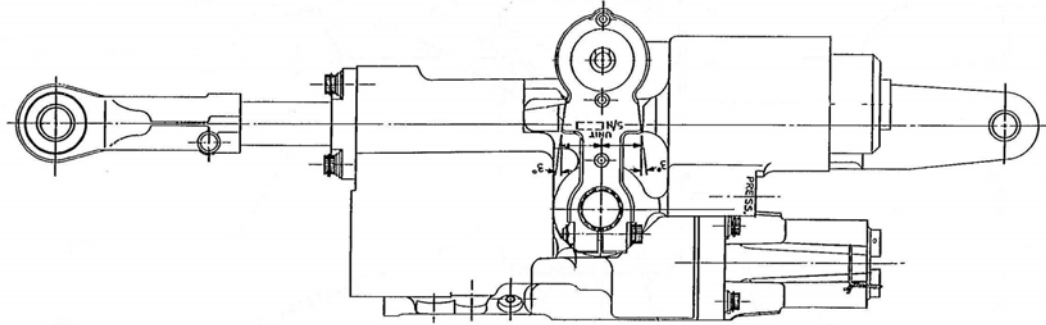
## Part Identification

Supplier: Parker Hannifin  
Boeing P/N: 65-44761-21\*  
S/N: 10748A\*  
Date Built: 1992\*

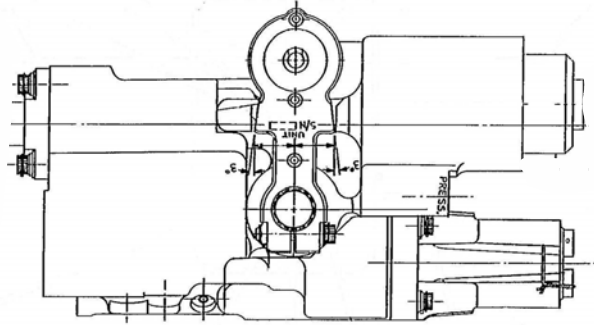
\*Data plate missing, information derived from Parker records based on manifold part number, serial number, and servo valve part number and serial number.



# 65-44761-21 Aileron PCU



## 65-44761-21 Aileron PCU



Rod end fitting missing  
Main ram fractured

Tailstock  
missing

65-44761-21 Aileron PCU

---



## 65-44761-21 Aileron PCU

---



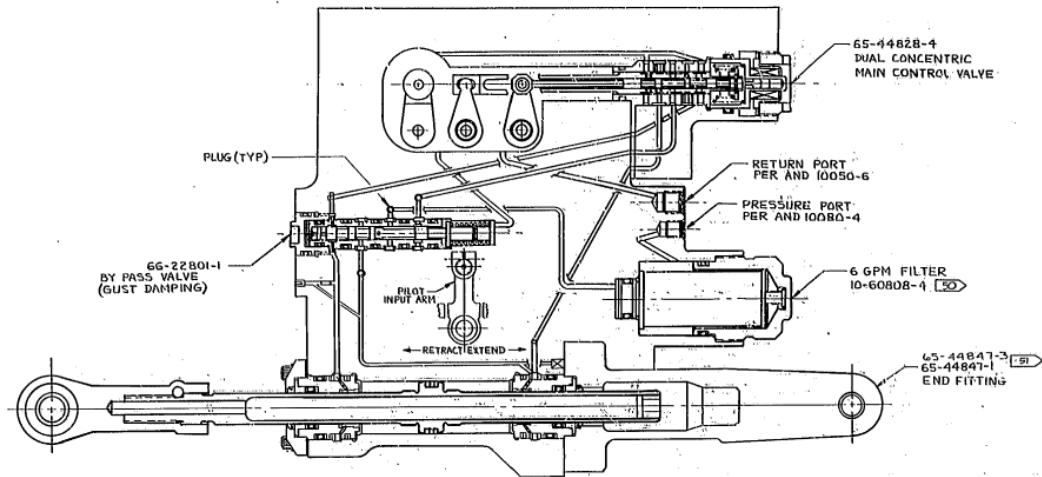
## Hydraulic Fittings

- Hydraulic fittings found broken
- Provides a path for sea water and other contaminants to enter the actuator





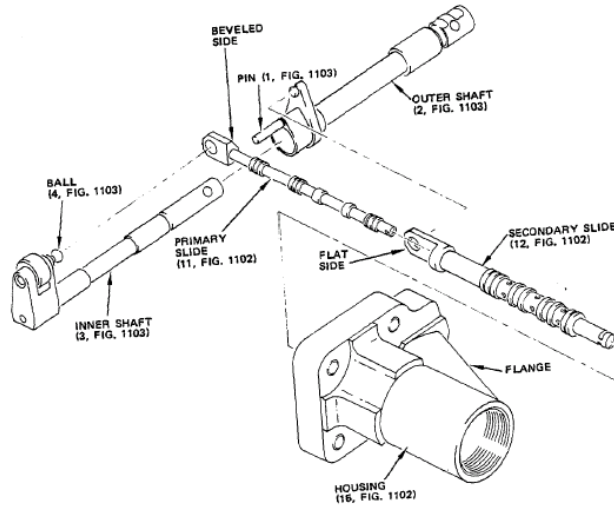
# Hydraulic Schematic



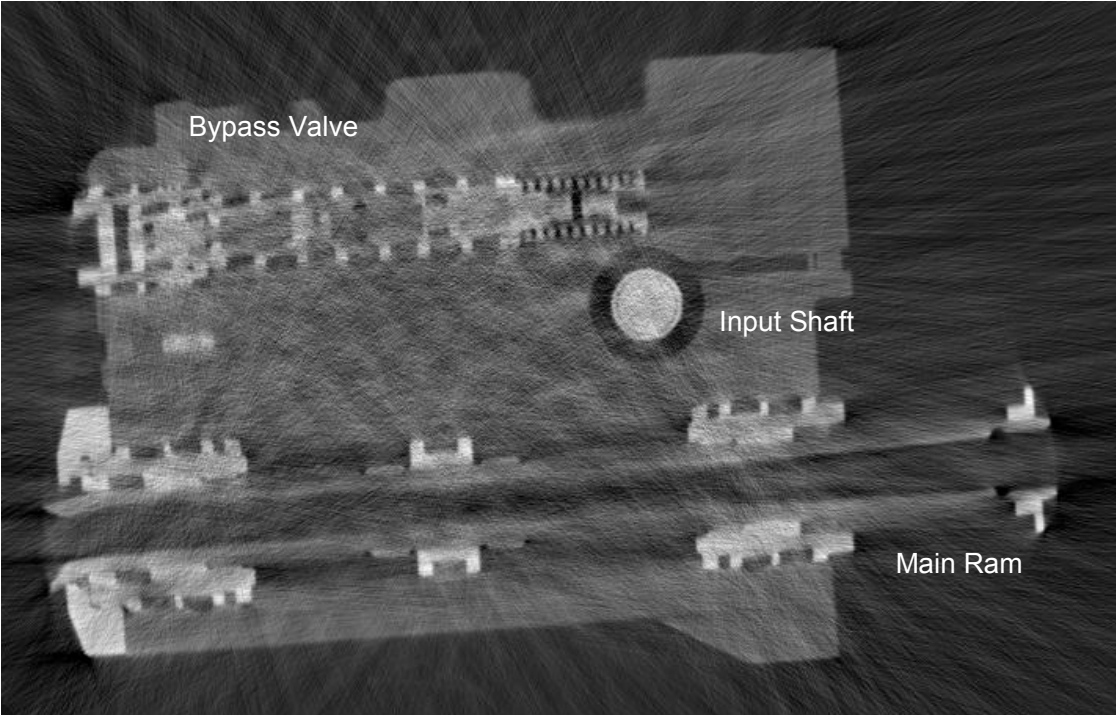
HYDRAULIC SCHEMATIC  
(SCALE: NONE)

# Servo Valve Components

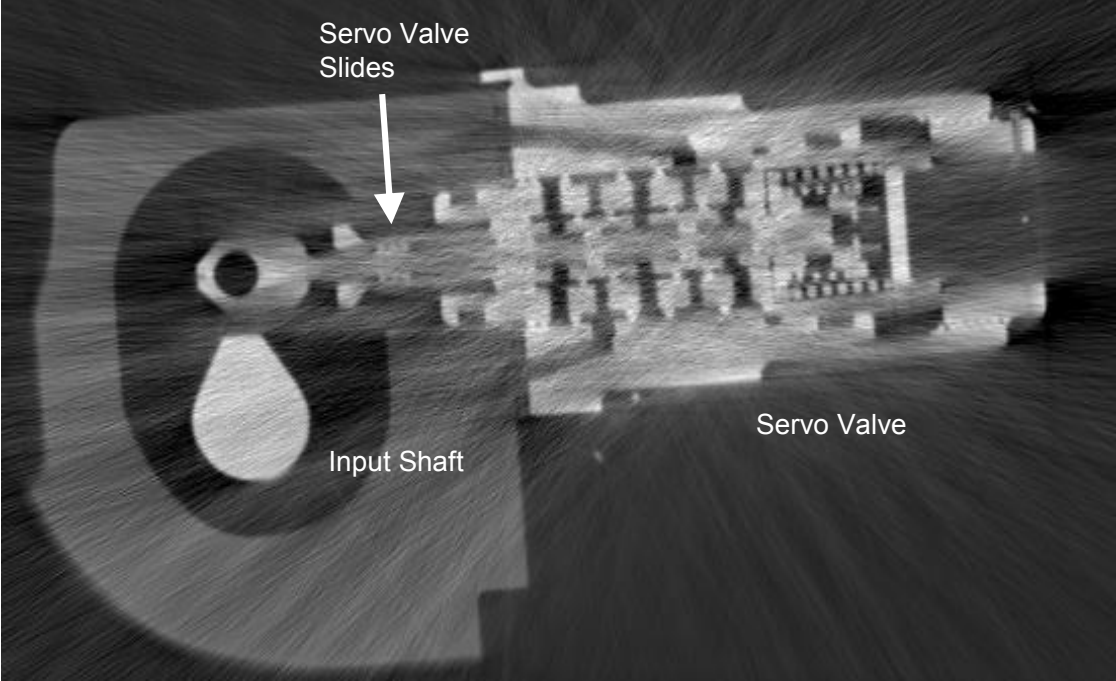
---



# Computed Tomograph Scan



# Computed Tomograph Scan



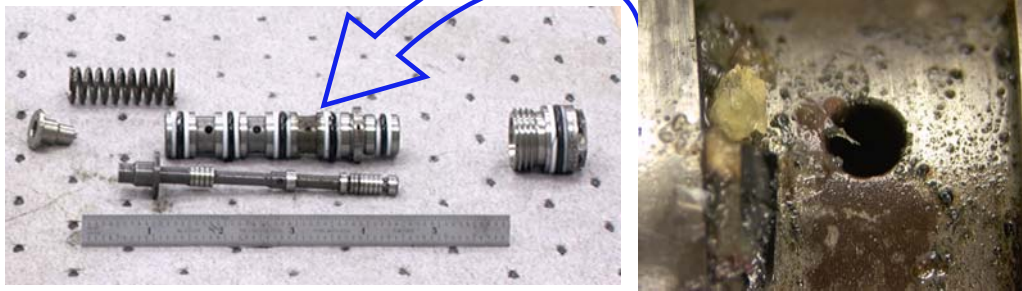
## Filter

- Filter cap and filter element removed
- Fluid sample and filter retained for chemical analysis



## Bypass Valve

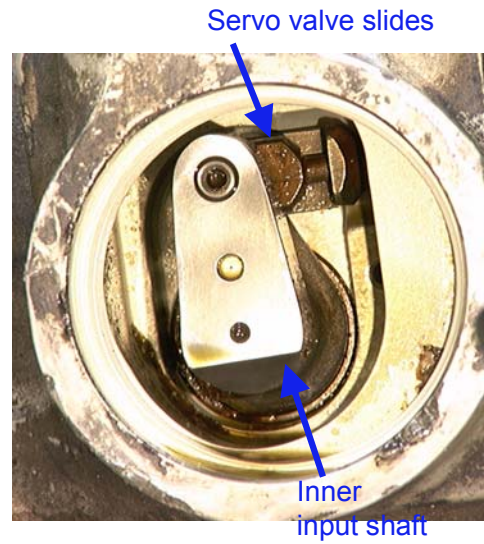
- Some corrosion and contamination on bypass valve sleeve
- Samples retained for chemical analysis



- Metal sliver found on outside of sleeve
- Origin uncertain, retained for chemical analysis

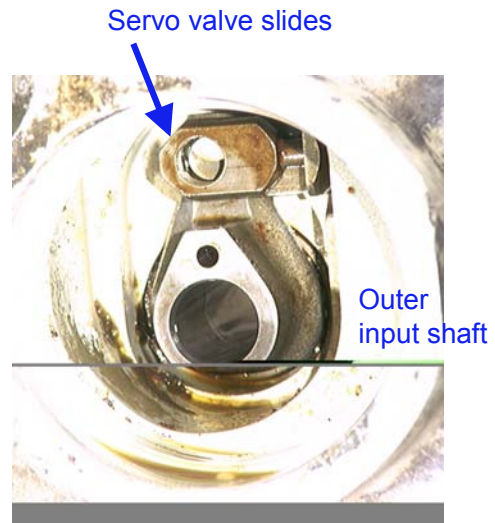
## Input Shafts

- Linkage cavity cover removed
- Some contamination noted in linkage cavity – samples taken for analysis
- View shows end of inner shaft and shaft and mating ends of servo valve slides



## Input Shafts

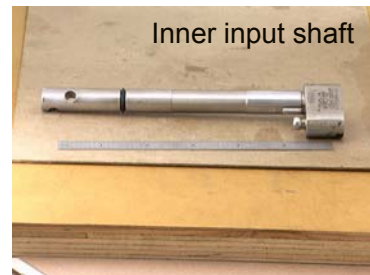
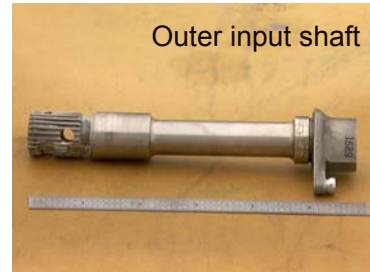
- Inner input shaft pressed out (required removal force much higher than normal)
- View shows outer shaft and mating ends of servo valve slides (inner shaft has been removed)





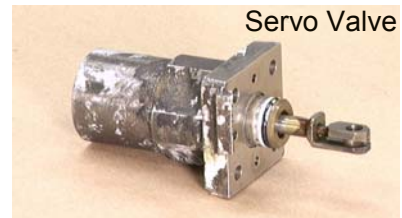
## Input Shafts

- Both shafts found to be bent
- Some corrosion found on shaft bearings, but none on shafts
- Deformed shafts consistent with high removal forces



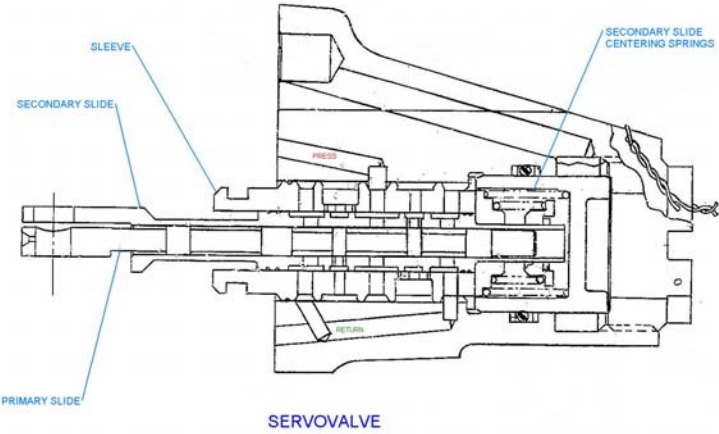
## Servo Valve

- Outer shaft rotated to allow removal of servo valve
- Axial load of 29 lbs applied to primary sleeve – no movement noted
- After removal, slides remain jammed

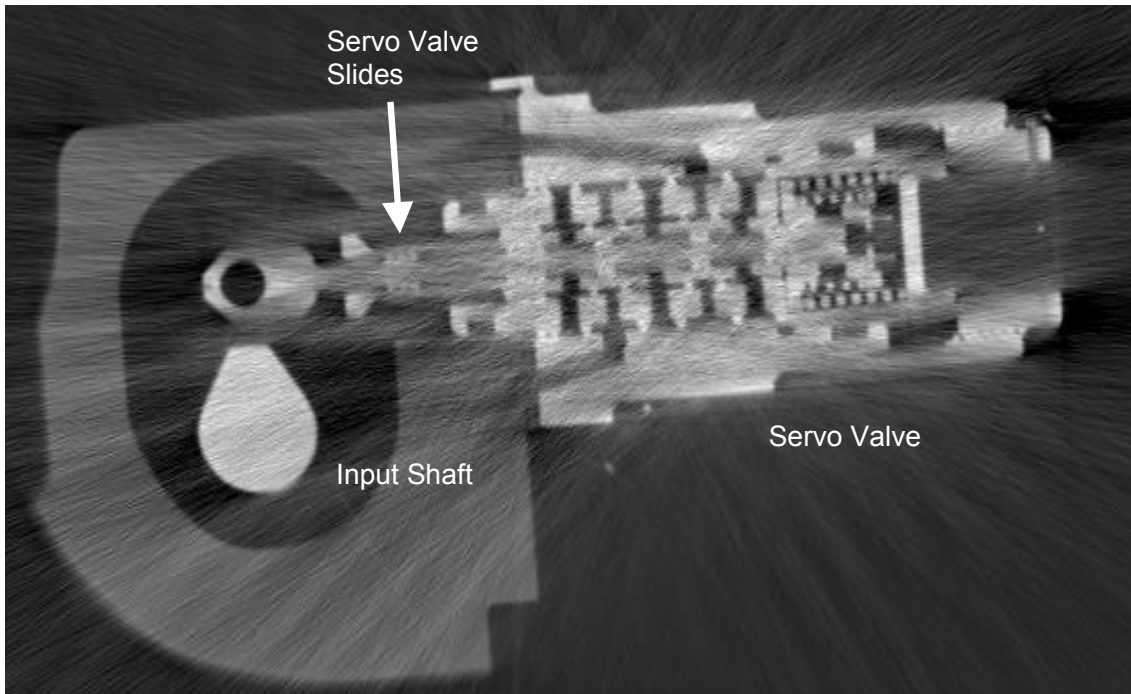


# Servo Valve Cross Section

---



## Computed Tomograph Scan



## Servo Valve

- Decision made to discontinue disassembly of servo valve (driving out slides could cause damage to surfaces)
- If deemed necessary, servo valve can be sectioned by electro-machining discharge (EDM).



▪

**Aileron PCU EQA Report (Aileron PCU EQA Report.pdf)**

**Boeing Proprietary information and will not be available for public use**

1.16.1.8. Master Caution:  
CairoMarch04Slides (March Progress Meeting - Cairo).pdf

## Master Caution Discrete at Time 92465

<u>Flight Controls</u>		<u>Electrical</u>		<u>Engine</u>	
Low Quantity	2	Low Oil Pressure	2	Reverser	3
Low Pressure	2	High Oil Temp	2	PMC-Inop	1
Feel Diff Press	2	Standby Power Off	2	Low Idle	1
Speed Trim Fail	1	Transfer Bus Off	3		
Mach Trim Fail	1	Bus Off	3	<u>Overhead</u>	
Yaw Damper	3			Equipment Cooling - Off	2
Autoslat Fail	2	<u>Overheat Detection</u>		Emer Exit Lts-Not Armed	2
		Engine1 overheat	2	Flight Recorder - Off	3
<u>Hydraulics</u>		Engine 2 overheat	2	Pass Oxy - On	3
Low Press – Elec Pump	3	APU Detection Inop	1		
Overheat – Elec Pump	2			<u>Air Cond</u>	
Low Press – Eng Pump	3	<u>Anti-Ice</u>		Flt Deck Duct Ovht	2
		Window overheat	2	Pax Duct Ovht	2
<u>IRS</u>		Pitot heat	2	Dual Bleed	2
Fault	2	Cowl Anti-Ice	3	Wing-Body Overheat	2
On DC	2	<u>Doors</u>		Bleed Trip Off	2
DC Fail	2	Fwd/Aft Entry	1	Auto Fail	2
		Equipment	1	Off Sched Descent	1
<u>Fuel</u>		Fwd/Aft Cargo	1	Pack Trip Off	2
Low Pressure	1	Fwd/Aft Service	1		
Filter Bypass.	3	Airstairs (not installed on PQ294)			
<u>APU</u>					
Low Oil Pressure	2				
Fault	2				
Overspeed	1				

**Legend**

1 = unknown  
2 = unlikely  
3 = ruled out

1.16.1.9. Auto Flight Systems

CairoMarch04Slides (March Progress Meeting - Cairo).pdf, 040301 Flash 737 Cairo  
Mtg (public release version).pdf  
*Relevant Figures*

**Boeing Proprietary information and will not be available for public use**



*737-300 (PQ294) Flight Director Control Law: (see also FDControlLaw.pdf file)*

**Boeing Proprietary information and will not be available for public use**

# HSI Display

:

## HSI Display Options



M-Cab HSI Control Panel

Full Rose



VOR Mode

Map Mode

Plan Mode

Mode selected by Capt & FO

Expanded



## Display Settings from FDR

Signal Name	Bit True	Bit False	Capt	FO)
FULL COMPASS ROSE	SELECT	NOT SEL	0	0
AIRPORTS	SELECT	NOT SEL	0	0
RTE DATA	SELECTED	NOT SEL	0	0
WPT	SELECT	NOT SEL	0	0
NAV AIDS	SELECT	NOT SEL	0	0
SPARE	SELECTED	NOT SEL		
NAV MODE SELECTED	SELECT	NOT SEL	0	0
ILS (STD) MODE SEL	ILS (STD)	NOT SEL	0	0
VOR (STD) MODE SEL	VOR (STD)	NOT SEL	0	0
PLAN MODE SEL	PLAN MODE	NOT SEL	0	0
ILS (MOD) MODE SEL	ILS (MOD)	NOT SEL	0	0
VOR (MOD) MODE SEL	VOR (MOD)	NOT SEL	1	1
MAP MODE SELECT	MAP MODE	NOT SEL	0	0
160 MI RANGE SEL	SET	NOT SET	0	0
80 MI RANGE SEL	SET	NOT SET	0	0
40 MI RANGE SEL	SET	NOT SET	0	0
20 MI RANGE SEL	SET	NOT SET	1	0
10 MI RANGE SEL	SET	NOT SET	0	1
WXR DATA	WXR SEL	NOT SEL	0	0 to 1 @ 530-534

Boeing Proprietary

# HSI Scale Options



Range selected by Capt

Range selected by FO

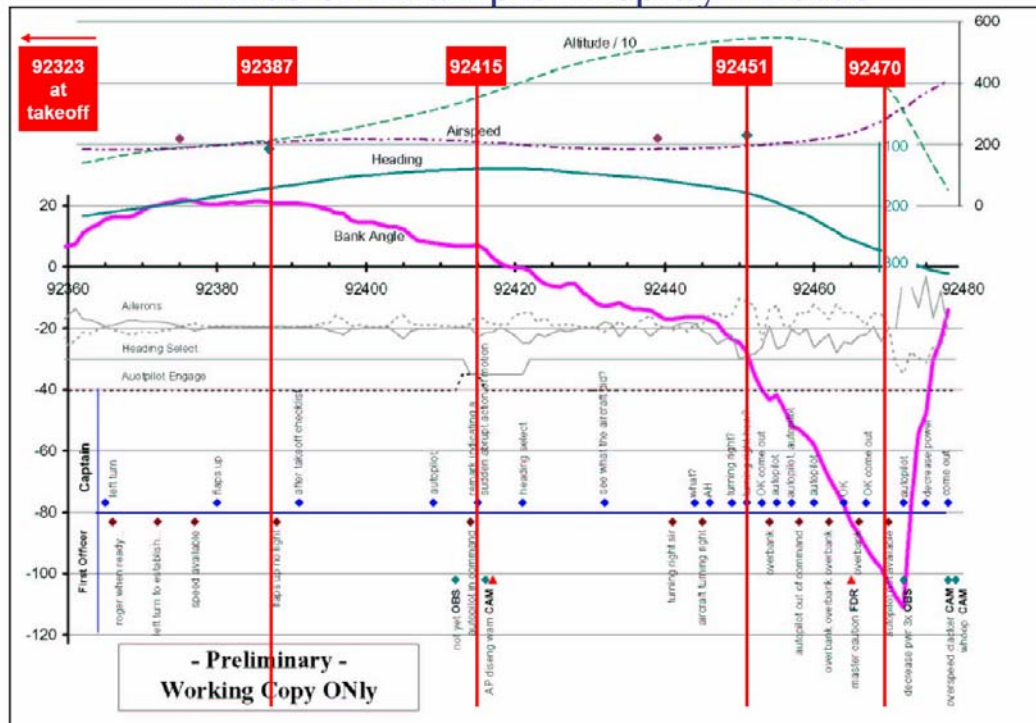


Boeing Proprietary

Note:  
Remaining information is Boeing Proprietary information and will not be available for public use

Times of Example Display Photos:

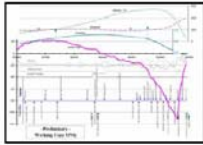
## Times of Example Display Photos



Boeing Proprietary

SU-ZCF  
@ Time  
92323

Takeoff

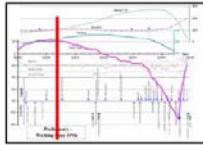


Boeing Proprietary



SU-ZCF  
@ Time  
92387

1st Hdg Sel point



Boeing Proprietary

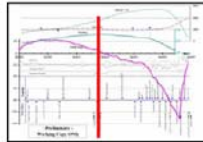


SU-ZCF  
 @ Time  
 923415

AP Engage point

28 seconds after  
 previous photo

assumed  
 value for Hdg Sel



Boeing Proprietary

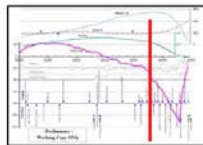




SU-ZCF  
@ Time  
92451

2<sup>nd</sup> Hdg Sel point

36 seconds after  
previous photo



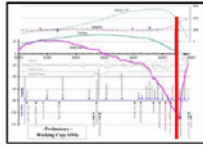
Boeing Proprietary



SU-ZCF  
@ Time  
92470

near max  
bank angle point

19 seconds after  
previous photo

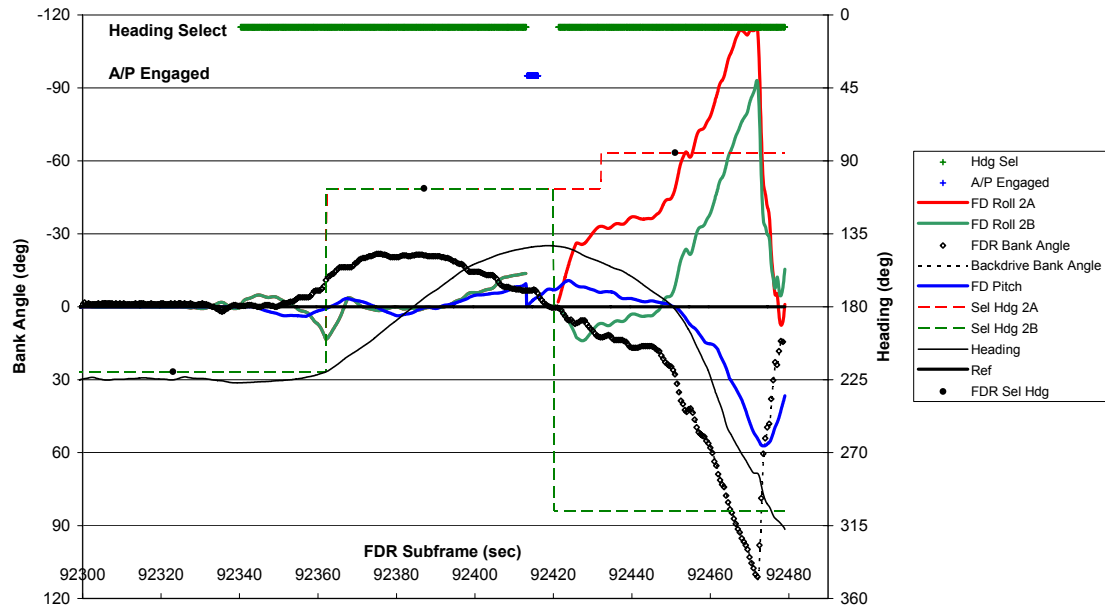


Boeing Proprietary



M-Cab Flight Director Commands (Flight Director Results Boeing.xls)

Flash Airlines SU-ZCF  
M-Cab Flight Director Commands



Display Architecture (Display Architecture.ppt)

**Boeing Proprietary information and will not be available for public use**

Autopilot Engagement Observations

## Autopilot Engagement *Observations*

- **Engage Hold Interlocks**
  - *essentially the same as pre-engage interlocks, see table*
  - *would need to have failed within the 3 seconds since engagement*
- **Engage Synchronization**
  - *syncs AP servo to aft quadrant*
  - *FCC allows 4.0 seconds to complete*
- **Manually Disconnected**

Autopilot Engage Logic

## Autopilot Engage & Engage Hold Interlocks

Condition	Pre-Engage	Engage Hold
	Prevent Engage	Cause Disengage
Pitch CWS force greater than 5 lbs	X	
Roll CWS force greater than 2.25 lbs	X	
Elevator Detent Pressure Switch Indicates Pressurized	X	
Aileron Detent Pressure Switch Indicates Pressurized	X	
Auto Stab Trim Cutout Switch in Cutout	X	X
Both Flap Switches and Stab Trim Motor don't agree as Flaps Up or as Flaps Down	X	X
Main Electric Trim Switch Activated	X	X
Aileron Force Limiter position does not agree with Flaps UP or Flaps Down	X	X
CAS Invalid	X	X
Uncorrected Altitude Invalid	X	X
28 VAC 400 Hz Invalid	X	X
MCP to FCC Bus Invalid	X	X
Pitch Angle Invalid	X	X
Pitch Rate Invalid	X	X
Roll Angle Invalid	X	X
Roll Rate Invalid	X	X
Baro Altitude Invalid (Prevents CMD only)	X	X
Elevator Detent Pressure Switch Indicates Non-Pressurized		X
Aileron Detent Pressure Switch Indicates Non-Pressurized		X
(Magnetic Heading OR TAS Invalid) AND (Roll CWS) AND (Bank Angle <8 degrees)	X	X

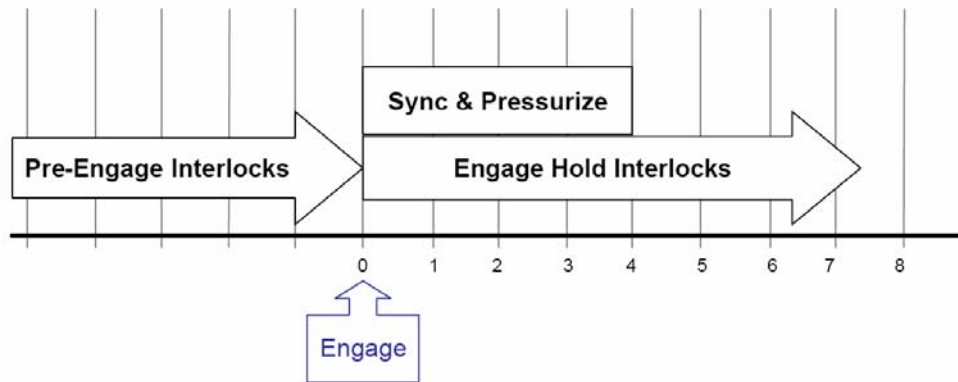
Boeing Proprietary

# Autopilot Engage Logic

## Failure to Sync or Pressurize Scenarios

CMD light ON

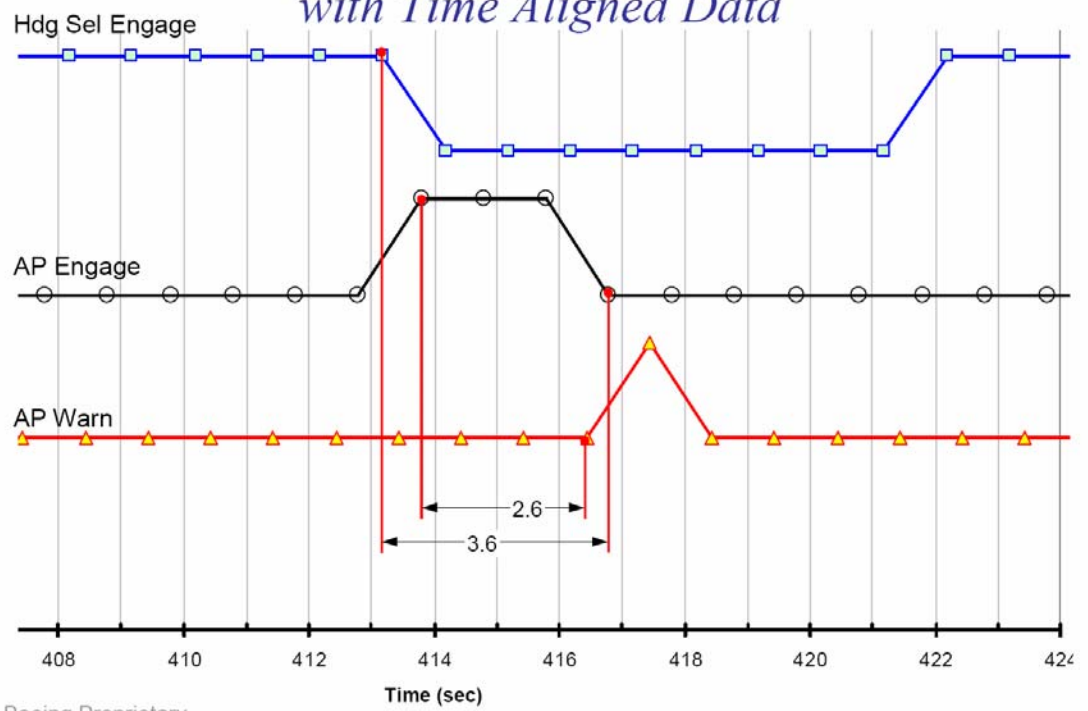
1-Failure to synchronize	4.0 sec
2-sync in 0+ sec but fails to pressurize	3.5 sec
3-sync in 4- sec but fails to pressurize	7.5 sec



Boeing Proprietary

Autopilot Engage Attempt- with Time Aligned Data

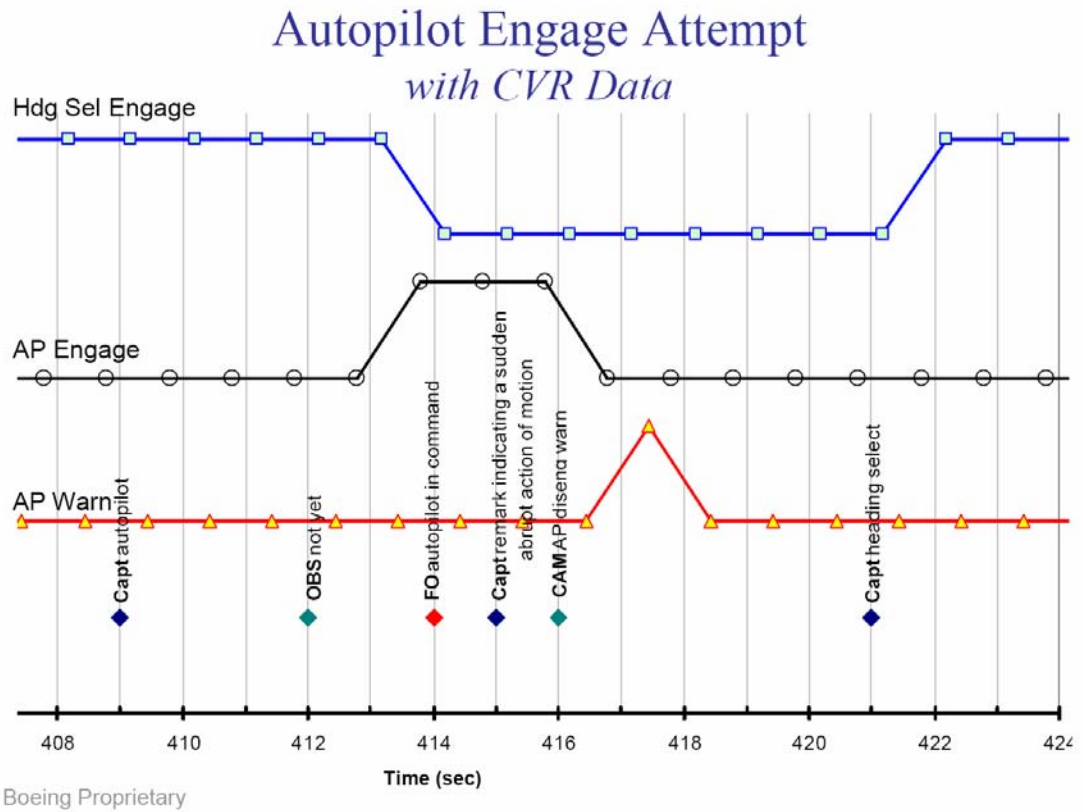
# Autopilot Engage Attempt *with Time Aligned Data*



Boeing Proprietary



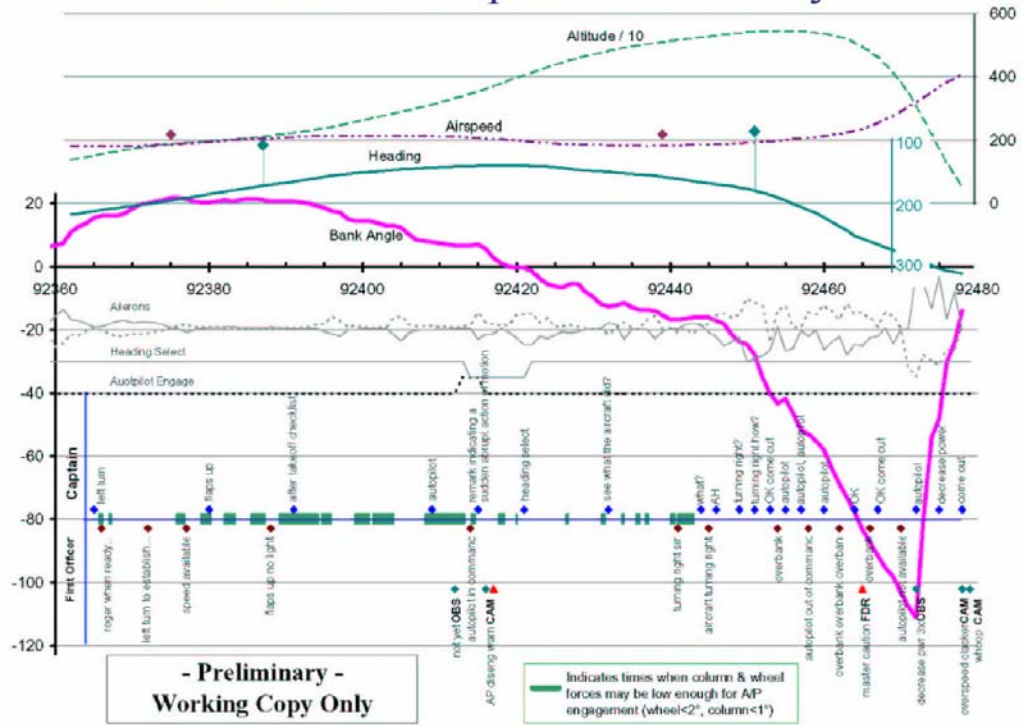
# Autopilot Engage Attempt- with CVR Data



Note:  
The recording "not yet" at 412 seconds is attributed to the captain and not to the observer.

Estimated Autopilot Availability

# Estimated Autopilot Availability



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AP Actuator description and Scenario 12 info b.pdf, AP Actuator description and Scenario 12 info 2.ppt

**Boeing Proprietary information and will not be available for public use**

Scenario 12 ver 2.ppt (Rev - 3 Feb 05)

**Boeing Proprietary information and will not be available for public use**

Honeywell SP-300 DFCS B737-300.ppt

**Honeywell Proprietary information and will not be available for public use**

Flash Airlines Presentation SP-300 DFCS Health Monitoring Honeywell.ppt

**Honeywell Proprietary information and will not be available for public use**

1.16.1.10. Flash Airlines AI236 RAM Simulator Configuration (Flash Airlines AI236 RAM Simulator Configuration.htm, Program\_Pins.pdf)

## **RAM FULL FLIGHT SIMULATOR**

**Subject: Request Configuration of RAM 737-500/400 Training simulator**

**Reference: (a) Email from Capt. Shaker Kelada, Egyptian Ministry of Civil Aviation, to xxxxxxxxx dated 26 May 2005.**

The simulator was agreed by Egyptian authority (CAA Egyptian ) on the 9 Mai 2003 for Flash airlines use. The simulator was used by flash airlines on dry lease, the instructor was flash airlines instructor.

Simulator configuration:

INITIAL CERTIFICATION: FAA AC 120-40 LEVEL D  
ACTUAL CERTIFICATION « JAR STD 1A LEVEL D » BY FRENCH AUTHORITY (DGAC) AND MOROCAIN AUTHORITY (DAC). Also agreed by all users authority like Tunisian, Jordanian, Senegalian, JAT Airlines

→ Simulateur Manufacturer: CAE Electronics LTD  
→ In service Date : 1993  
→ Master Aircraft : B.737-500 Convertible to B.737-400  
→ APU : GTCP36-28 (B) Garette  
→ Basic Engine Data : CFM 56B2 - CFM 56C1  
→ AFCS : Honeywell MCP 4051601-937  
→ EFIS : Collins P/N 622-9436-1014  
→ Flight Management System : Smith industries P/N: 168925-06-01  
→ Host computer : IBM Risc 6000  
→ Motion & Control loading : Hydrostatic actuators with digital control electronics and 6 axis

TCAS – CFIT - Windshear warning system – Low visibility (CAT I- II –III) – ATIS – GPWS

### **VISUAL VITAL VII**

→ Visual System Manufacturer : Flight Safety (V S S).  
→ Computer : Motorola SMM 1467.  
→ Type of Image Generator : Vital VII.  
→ Type of Display : Wide (FOV) 150x40 degre.  
→ Illumination Level : Day / Bright Day / Dusk / Night.

### **INSTRUCTOR STATION**

→ Computer : 2 Computers Iris 4D25.  
→ Display : 2 CRT / Touch Screen  
→ Printer : Color hardcopy unit.  
→ Training Aids : Wind, Wind shear (16 Profils), Rec & Instant replay, FMC copy, Camera, video tape recorder, lesson plan

### **EFIS CUSTOMER OPTIONS:**

EADI FORMAT : EUROPEAN - BASIC  
FAST SLOW/SPEED TAPE : SPEED TAPE – FAST SLOW  
F/S – G/S : REVERSAL – NORMAL  
SPEED TAPE : REVERSAL – NORMAL

SPEED TREND VECTOR	: DISABLE – ENABLE
SPEED TAPE CAS	: CURSOR – ROLLING
MIN OP SPEED	: ENABLE – DISABLE
G/S AND TAS DISPLAY	: DLH – BASIC
EADI TAS DISPLAY	: ENABLE – DISABLE
FD DISP SEL	: FILLED INTEGRA C – INTEGRA C – SPLIT AXIS
FILLED AIRPLANE SYMBOL:	
RA DISP SEL	: ANA - ANA/RR DIG - DIG /RR
PITCH LIMIT IND	: DISABLE – ENABLE
H ALERT SEL	: NO ALERT – 1000 FT 1500FT - 2500 FT
ILS DEVIATION: DISABLE – ENABLE	
WARNING	
DUAL CHANNEL ANN	: DISABLE – ENABLE
COMPARATOR	: ON – OFF
BLINKING COMPARATOR	: DISABLE – ENABLE
EHSI SYMBOLOGY	: SPERRY – BASIC
CENTER MAP	: FULL ROSE – EXP ROSE
MAP ORIENTATION	: HEADING UP – TRACK UP
VOL/ILS ORIENTATION	: HEADING UP – TRACK UP
NAV/IRU POS DIFF	: F/TIME DISP – FMC DISP DISABLE
WIND BEARING	: DISABLE – ENABLE
RANGE ARCS	: DISABLE – ENABLE
WXR TURB COLOR	: MAGENTA – RED
ADF POINTERS MAP	: DISABLE – ENABLE
ADF INSTL	: SINGLE LEFT – SINGLE RGT DUALE - NONE

ENGINE: **20.000 LB**  
**22.000 LB**  
**23.500 LB**  
**18.500 LB**

GPWS CUSTOMER OPTIONS:

WINDSHEAR ALGORITHM: ENABLE – DISABLE

ALTITUDE CALL OUTS: ENABLE – DISABLE  
INCLUDE 'BANK ANGLE – BANK ANGLE' when bank angle exceeds 35, 40 and 45 degrees.

ALTITUDE CALL OUTS SEL ID: .....

VOICE MENU SEL:

ENVELOPE MODULATION: ENABLE – DISABLE  
FMC INPUT SELECTION: ENABLE – DISABLE  
AUDIO LEVEL REDUCTION: ENABLE – DISABLE

**FCC**

The Flight Control Computer System of the B737 Classic is identified as Computer software Component (CSC)

This CSC will simulate the flight Control computer and will consist of the xxxxSL, xxxxSP, xxxxSR, xxxxSC, xxxxST modules called up by the synchronuous dispatcher as entry points SLOGIC, SPITCH, SROLL, SCOMP, SINT and STRIM.



1.16.1.10. Boeing response to raised questions.doc

References

17833 (B-H200-17833-ASI 12 Feb 2004).pdf

CairoMarch04Slides (March Progress Meeting - Cairo).pdf

17848 (B-H200-17848-ASI 04 March 2004).pdf

Cairo March 04 Autopilot Flash 737 March Progress Meeting Flash 737  
March Progress

Flash Airlines Autopilot Answer to Questions - 31 Jan 2005.ppt

Answers to question\_cairo meeting05.ppt

Action Item Response.ppt (Cairo meeting, 1-30-05 to 2-2-05)

Responses to Airplane System Queries  
Flash Airlines 737 SU-ZCF Accident at Sharm el Skeikh – 3 Jan 04

**Questions from the MCA on 25 Jan 04**

A1) Why did the autopilot disengage?

*Answer: There are three possible reasons why the autopilot disengaged: the engage synchronization (actuator to surface) failed to complete; the engage hold interlocks were not satisfied; or it was manually disconnected. Based on the data recorded on the FDR, we are not able to pinpoint which of these caused the autopilot to disengage. Additional information on the details of the interlocks and their operation are provided in the Airplane Maintenance Manual section 22-11-01. ①*

A2) What is the effect of hydraulic systems failures on the flight controls?

*Answer: The hydraulic system arrangement for the 737-300 flight controls is provided in the attached figure. This figure shows which functions would be lost in the event of either an A or B hydraulic system failure.*

A3) What does the FD command? Roll rate? Bank angle?

*Answer: The Flight Director (FD) provides a bank angle command that is primarily a function of selected heading, airplane heading, airplane roll angle, and airplane roll rate. ①*

A4) Please provide the FMEA for the 737-300 autopilot and flight controls related to the roll axis.

*Answer: The following documents were mailed to the NTSB, MCA and BEA:  
D6-14070 737-300 Lateral Failure Analysis (7MB)  
D6-37432 737-300 Autopilot Failure Analysis (20MB)*

A5) What does the flight director do when the airplane bank angle exceeds the selected bank angle limit?

*Answer: It will produce a command to fly back to the desired bank angle. ①*

A6) What does the flight director do when the airplane roll rate exceeds the intended roll rate?

*Answer: It will produce a command to fly back to the desired bank angle. ①*

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① We are preparing additional and more detailed technical information about the operation of the autopilot, flight director, and lateral control systems which will be available for discussion during the next progress meeting in Cairo.

Responses to Airplane System Queries  
Flash Airlines 737 SU-ZCF Accident at Sharm el Skeikh – 3 Jan 04

- A7) What are the aileron travel rates with various hydraulic system availability?  
*Answer: The aileron PCUs are significantly oversized. Because of this, aileron travel rates are not a function of hydraulic system availability - i.e. aileron travel rates are not significantly different whether either or both hydraulic system is pressurized. For reference, the no load rate is approximately 54 degrees per second of aileron.*
- A8) How is Selected Heading recorded on the FDR if it is being turned while the knob is being moved)  
*Answer: The FCC transmits the selected heading value to the DFDAU at a rate of 20 times per second. The DFDAU then takes the latest value once each 64 seconds and sends it to the DFDR for recording. Thus, if selected heading is dynamically changing when the once-per-64-seconds sample is taken, it will record the selected heading value at the time the sample was taken.*
- A9) Is the hydraulic pump capable of outputting 5000 psi of pressure?  
*Answer: The following two failures are required in order to reach 5000 psi: /1/ the pump compensator is failed open (full flow), and /2/ the system relief valve failed closed. For the hydraulic system pressure display, in-range is considered to be from -100 to 4,100 psi, so 5000 psi would be out of range. If the system were to actually go to 5000 psi, the affected hydraulic pressure display (on the EIS) would slew to its lower stop; hold for 2 seconds then the pointer would disappear and dashes would appear in the display.*
- A10) What caused the Master Caution discrete late in the flight?  
*Status: The Master Caution discrete occurs at time 92465 in the FDR data file received by Boeing. There are over 40 inputs that could have caused this discrete to be set. We are still evaluating the possible causes of the setting of this discrete, and expect to have an update for the next progress meeting in Cairo. We did notice that the Master Caution discrete was set several times on previous flights. Airplane records, such as technical log entries, may record the reason for previous Master Caution events. These records may help isolate why the Master Caution was set at time 92465 in the accident flight.*

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① We are preparing additional and more detailed technical information about the operation of the autopilot, flight director, and lateral control systems which will be available for discussion during the next progress meeting in Cairo.

Responses to Airplane System Queries  
Flash Airlines 737 SU-ZCF Accident at Sharm el Skeikh – 3 Jan 04

- B1) Correlation between control inputs and flight control surface deflections, with special emphasis on the inconsistency of control wheel and aileron surface deflection as indicated by the FDR.  
*Answer: A kinematic consistency check and a simulator proof-of-match is being accomplished on the accident data at Boeing. This work is still in progress; however, we have been able to make a few observations on the bias in control wheel position. There is a bias in control wheel position that shifts over time, and possibly a scaling issue. Both issues are being further analyzed for possible explanations. ①*
- B2) Investigate the changes in aileron deflection bias.  
*Answer: The changes in aileron position bias are caused by the airload on the aileron reacting against the cable run in the wing between the aileron and aileron PCU. The bias in aileron position is due to aileron hinge moment which varies as a function of airspeed. ①*
- B3) Investigate the cause(s) for the autopilot disconnect.  
*Answer: See response to question A1.*
- B4) Investigate the cause for HDG SEL disengage when the autopilot was engaged.  
*Answer: If the FD command is greater than 7 degrees at the time autopilot engagement is attempted, the roll mode will change from HDG SEL to CWS. According to the FDR data, this seems consistent with the probable flight director command which existed when A/P engagement was initiated. ①*
- B5) Investigate the possible failure modes of the Flight Director indicator.  
*Status: This is being researched. We will have some preliminary data available to discuss during the next progress meeting in Cairo. ①*

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① We are preparing additional and more detailed technical information about the operation of the autopilot, flight director, and lateral control systems which will be available for discussion during the next progress meeting in Cairo.

Responses to Airplane System Queries  
Flash Airlines 737 SU-ZCF Accident at Sharm el Skeikh – 3 Jan 04

- B6) Investigate availability of autopilot during the captain's requests for "autopilot, autopilot".

*Answer: The autopilot will not initiate the engage sequence if the A/P engage interlocks are not satisfied (ref AMM 22-11-01 page 54). If the engage interlocks are not satisfied, the attempt to engage (A/P button push) will not be recorded on the FDR. In the case of the accident flight it's possible that forces on the column or wheel prevented the engage logic from being satisfied. Additional information on the details of the interlocks and their operation are provided in the Airplane Maintenance Manual section 22-11-01. ①*

- B7) Investigate the effect of flight control surface failures for surfaces like spoiler deflections that are not recorded on the FDR.

*Answer: The effects of various spoiler failures are being examined using kinematic simulations. These results are expected to be available for the next progress meeting in Cairo.*

**Observations on EGT and Engine Oil Pressure Parameters**

During the work in Cairo, it was noted that the EGT and engine oil pressure parameters did not appear to be working properly for either the left or right engines. All four of these parameters are defined in D6-55333 Appendix B and are found in word 61 of the 737-2 data frame, along with a number of other parameters which occupy the same locations. There are several variants of the 737-2 data frame depending upon whether the airplane is equipped with an electronic engine instrument system (EIS) or an electronic flight instrument system (EFIS). The subject airplane, SU-ZCF, was equipped with both and the resulting data frame variant is informally referred to as the 737-2EE data frame. Appendix B lists all variants of the data frame, including the multiple parameters that can be stored in word 61. The order of data selection, e.g. which parameters are actually to be recorded in word 61, is provided in the general notes of appendix B. In this case, the EFIS parameters have priority over the EIS parameters and EGT and engine oil pressure are not recorded. Thus, the attempted conversion of word 61 into EGT and engine oil pressure is not appropriate in the 737-2EE data frame. In the 737-2EE data frame, word 61 is used for a number EFIS mode selection discretes, which appear to be recorded properly on the FDR.

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① We are preparing additional and more detailed technical information about the operation of the autopilot, flight director, and lateral control systems which will be available for discussion during the next progress meeting in Cairo.

## Lateral System-answers to questions

- A2) What is the effect of hydraulic systems failures on the flight controls?  
*Answer: The hydraulic system arrangement for the 737-300 flight controls is provided in the attached figure. This figure shows which functions would be lost in the event of either an A or B hydraulic system failure.*
- A7) What are the aileron travel rates with various hydraulic system availability?  
*Answer: The aileron PCUs are significantly oversized. Because of this, aileron travel rates are not a function of hydraulic system availability. i.e. aileron travel rates are not significantly different whether either or both hydraulic system is pressurized. For reference, the no load rate is approximately 54 degrees per second of aileron.*
- B1) Correlation between control inputs and flight control surface deflections, with special emphasis on the inconsistency of control wheel and aileron surface deflection as indicated by the FDR.  
*Answer: A kinematic consistency check and a simulator proof of match is being accomplished on the accident data at Boeing. This work is still in progress, however, we have been able to make a few observations on the bias in control wheel position. There is a bias in control wheel position that shifts over time, and possibly a scaling issue. Both issues are being further analyzed for possible explanations.*  
①
- B2) Investigate the changes in aileron deflection bias.  
*Answer: The changes in aileron position bias are caused by the airload on the aileron reacting against the wing cable run between the aileron and aileron PCU. Therefore, the bias in aileron position is due to aileron hinge moment which varies as a function of airspeed.* ①
- B7) Investigate the effect of flight control surface failures for surfaces like spoiler deflections that are not recorded on the FDR.  
*Answer: The effects of various spoiler failures are being examined using the Boeing simulation. These results are expected to be available for the next progress meeting in Cairo.*

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# Autopilot - Answers To Questions

A1) Why did the autopilot disengage?

*Answer: There are three possible reasons why the autopilot disengaged: the engage synchronization (actuator to surface) failed to complete; the engage hold interlocks were not satisfied; or it was manually disconnected. Based on the data recorded on the FDR, we are not able to pinpoint which of these caused the autopilot to disengage. Additional information on the details of the interlocks and their operation are provided in the Airplane Maintenance Manual section 22-11-01 . ①*

B3) Investigate the cause(s) for the autopilot disconnect.

*Answer: See response to question A1.*

B6) Investigate availability of autopilot during the captain's requests for "autopilot, autopilot".

*Answer: The autopilot will not initiate the engage sequence if the A/P engage interlocks are not satisfied (ref AMM 22-11-01 page 54). If the engage interlocks are not satisfied, the attempt to engage (A/P button push) will not be recorded on the FDR. In the case of the accident flight it's possible that forces on the column or wheel prevented the engage logic from being satisfied. Additional information on the details of the interlocks and their operation are provided in the Airplane Maintenance Manual section 22-11-01. ①*

# FD-answers to questions

- A3) What does the FD command? Roll rate? Bank angle?  
*Answer: The Flight Director (FD) produces a roll and roll rate command to zero the error between the selected heading and the magnetic heading. ①*
- A5) What does the flight director do when the airplane bank angle exceeds the selected bank angle limit?  
*Answer: It will produce a command to fly back to the desired bank angle. ①*
- A6) What does the flight director do when the airplane roll rate exceeds the intended roll rate?  
*Answer: It will produce a command to fly back to the desired bank angle. ①*
- A8) How is Selected Heading recorded on the FDR if it is being turned while the knob is being moved)  
*Answer: The FCC transmits the Hdg Sel value to the DFDAU at a rate of 20 times per second. The DFDAU then takes the latest value once each 64 seconds and sends it to the DFDR for recording. Thus, if Hdg Sel is dynamically changing when the once-per-64-seconds sample is taken, it will record the Hdg Sel value at the time the sample was taken.*
- B4) Investigate the cause for Hdg Sel disengage when the autopilot was engaged.  
*Answer: If the FD command is greater than 7 degrees at the time autopilot engagement is attempted, the Heading Select mode will be reset and the roll mode will default to CWS. According to the FDR data, this seems consistent with the probable flight director command which existed when A/P engagement was initiated. ①*

Boeing Proprietary



## Other-answers to questions

A4) Please provide the FMEA for the 737-300 autopilot and flight controls related to the roll axis.

*Answer: The following documents were mailed to the NTSB, MCA and BEA:*

*D6-14070 737-300 Lateral Failure Analysis (7MB)*

*D6-37432 737-300 Autopilot Failure Analysis (20MB)*

A9) Is the hydraulic pump capable of outputting 5000 psi of pressure?

*Answer: The following two failures are required in order to reach 5000 psi: /1/ pump compensator failed open (full flow), and /2/ system relief valve failed closed. For the hydraulic system pressure display, in-range is considered to be from -100 to 4,100 psi, so 5000 psi would be out of range. If the system were to actually go to 5000 psi, the affected hydraulic pressure display (on the EIS) would slew to its lower stop; hold for 2 seconds then the pointer would disappear and dashes would appear in the display.*

A10) What caused the Master Caution discrete late in the flight?

*Status: The Master Caution discrete occurs at time 92465 in the FDR data file received by Boeing. There are over 40 inputs that could have caused this discrete to be set. We are still evaluating the possible causes of the setting of this discrete, and expect to have an update for the next progress meeting in Cairo. We did notice that the Master Caution discrete was set several times on previous flights. Airplane records, such as technical log entries, may record the reason for previous Master Caution events. These records may help isolate why the Master Caution was set at time 92465 in the accident flight.*

# Displays-answers to questions

*B5) Investigate the possible failure modes of the Flight Director indicator.*

*Status: This is being researched. We will have some preliminary data available to discuss during the next progress meeting in Cairo.*

Boeing Proprietary

**Questions from 1 March 04**

- 1) How is drift angle matched in KINCON with corrected accelerations?  
*Response: Wheel-well based accelerometer data recorded on the FDR are integrated and converted into a ground speed vectors and altitude. Using IRU information, the ground speed vectors are converted into a drift angle and ground speed. The calculated altitude, drift angle and ground speed are then compared to the recorded altitude and the FMC's recorded drift angle and ground speed. Differences between the two sets of data are minimized by calculating a unique but constant acceleration bias for each axis. The biases are then applied to the recorded accelerometer data. The biases were calculated based on minimizing the error over the entire accident flight.*
  
- 2) With the simulator match data vs FDR data, at the end of the flight when rolling back towards wings level, time 92470 thru the end of data, why does the FDR data show the oscillatory motion, but the simulator match does not?  
*Response: The simulator match is an iterative process in which the difference between the simulator behavior and the recorded FDR data is used as a feedback (with a specific gain) to revise the simulator control inputs. In general, a lower gain produces smoother control inputs (lower frequency content) while a higher gain is required to match highly dynamic maneuvers, but can produce significant noise. The gain used in this iteration was chosen to best match the behavior in the time period from 92337 to 92470. Increasing the gain to match the highly dynamic portion of the flight after time 92470 would have introduced significant noise into the earlier portion of the simulation.*
  
- 3) From FDR time 92470 thru the end of data, are the aileron rates seen on the FDR within the capability of the system (i.e. is it real)?  
*Response: Yes, the aileron rates seen at the end of the FDR data are within the capability of the system.*
  
- 4) With respect to the FDR recorded wheel position data, the wheel bias in the air, just after takeoff, is different on the accident flight than the previous flight, Why?  
*Response: The bias in the recorded control wheel signal appears to change on numerous occasions. As noted in the earlier presentation material, the bias changes during the control wheel sweep prior to every takeoff. In addition, the bias appears to change during every climb out, typically between takeoff and flaps up. Furthermore, the bias also appears to change just prior to landing, either during descent or approach. See attached slides that show the changing wheel bias for the accident flight and the previous flight. Similar behavior is noted in all flights, including the first recorded landing, control sweep and takeoff from Abu Simbel. The behavior of the recorded FDR wheel signal appears consistent with a slipping synchro body.*

Responses to Queries  
Flash Airlines 737 SU-ZCF Accident at Sharm el Skeikh – 3 Jan 04

- 5) What equation is Boeing using to convert raw data into EU for Wheel Position?

*Response: The conversion steps are as follows:*

- 1. The raw data is first converted to a signed quantity using two's complement.*
- 2. The signed counts (C) are converted to synchro degrees (S) using the formula:  $S = C * 360 / 1024$*
- 3. The synchro degrees (S) are converted to degrees of wheel (W) using the formula:  $W = S * 150.7663958 / 180$*

*Additional Information: The control wheel sensor on this airplane is a synchro. The synchro signal is interpreted by the FDAU and passed to the flight recorder as counts. Different FDAUs interpret the synchro signal differently. SU-ZCF was equipped with a Sundstrand FDAU which interprets the synchro linearly. Other FDAU's (e.g. Teledyne) use a non-linear interpretation of synchro data. For Sundstrand FDAU's (and any other that interprets synchros linearly), the correct conversion for wheel data is a linear one such as the one shown above in step 2. For a Teledyne FDAU, a non-linear conversion is required. This conversion is built into the RAPS program and is called "dc\_TELEDYNE\_SYNCHRO". It would not be appropriate to use this function for converting data from a Sundstrand FDAU, such as the SU-ZCF data. In examining the FFD file provided, it appears that this function is being used to convert control wheel data. This conversion will introduce some errors as shown in the attached plots.*

*The MCA also provide a sheet of paper titled "Analog Signal Description" dated 24 May 1991, with the notation "Project BS7372". The data in this sheet appears to match the D6-55333 data for the 737-2 data frame with 2 exceptions:*

*D6-55333 defines control wheel as a 10 bit signal. BS7372 lists the signal as a 12 bit signal. The lower two bits of the actual dataframe are used to discrete bits. If both these bits are set, than a wheel position error of ~0.22 degrees will result.*

*The scaling of the BS7372 differs by a small amount from that of D6-55333. Note: The BS7372 sheet lists separate "Breakpoints" in the data. These "break points" exist to account for the signed nature of the signal (it wraps around from maximum counts back to zero). The function of the "break point" in the BS7372 data is accomplished by the two's complement function listed above and that also exists in the RAPS conversion listed in the FFD file provided.*

Responses to Queries  
Flash Airlines 737 SU-ZCF Accident at Sharm el Skeikh – 3 Jan 04

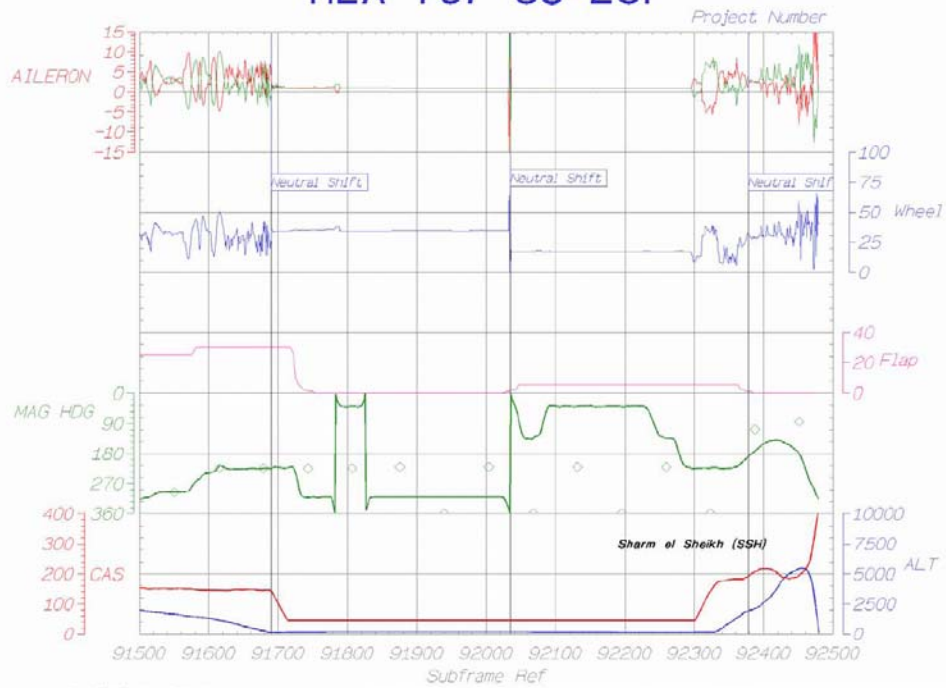
- 6) Please provide a schematic showing the dual concentric control valve in the aileron PCU, and how it attaches to the PCU input rod.  
*Response: Schematics provided.*
- 7) What bias springs are present on the PCU valve, and which direction are they biased?  
*Response: Schematics provided.*
- 8) Is there any delay between the time the autopilot is disconnected and when the disconnect warning is issued.  
*Response: The MCP monitors the CMD and CWS discretes from the FCC and immediately sets the warning (light and aural) when an autopilot disconnect is detected.*
- 9) What method does Boeing use to perform differentiation on flight data? Is there software available for purchase, or what is our algorithm?  
*Response: Without knowing the specifics of the differentiation in question, we can provide a very general answer. Because of the inherent noise associated with differentiation, Boeing tends to avoid differentiation of recorded signals where possible. In some cases, when differentiation is required, we have first modeled the recorded data with a curve fit known to have continuous derivatives and then performed the differentiation on the fitted curve. In other cases, it is possible to take advantage of the known behavior of specific physical quantities and required relationships between different recorded signals when differentiation is required.*

**Questions from 2 March 04**

- 1) Relative to the photo at time 92415, does the "CMD" and "CWS R" text appear on the EADI immediately when the cmd button is pushed or does it wait until the FCC has completed sync & pressurize (i.e. connected to system)?  
*Response: Immediately when CMD is received from the MCP (button push or paddle lift) the FCC retransmits it to the EFIS processor for display on the EADI.*
- 2) Would the roll FD bar really disappear when Hdg Sel was re-set during AP engage. The photo shows the bar gone because Hdg Sel had reset.  
*Response: Yes, the FD bar will be biased out of view in this situation.*
- 3) How does CWS R mode work?  
*Response: In CWS R, the autopilot will enter Heading Hold if the bank angle is less than or equal to 8 degrees or Bank Angle Hold if bank angle is greater than 8 degrees (if bank angle is greater than 30 it will return the airplane to 30).*

Responses to Queries  
Flash Airlines 737 SU-ZCF Accident at Sharm el Skeikh – 3 Jan 04

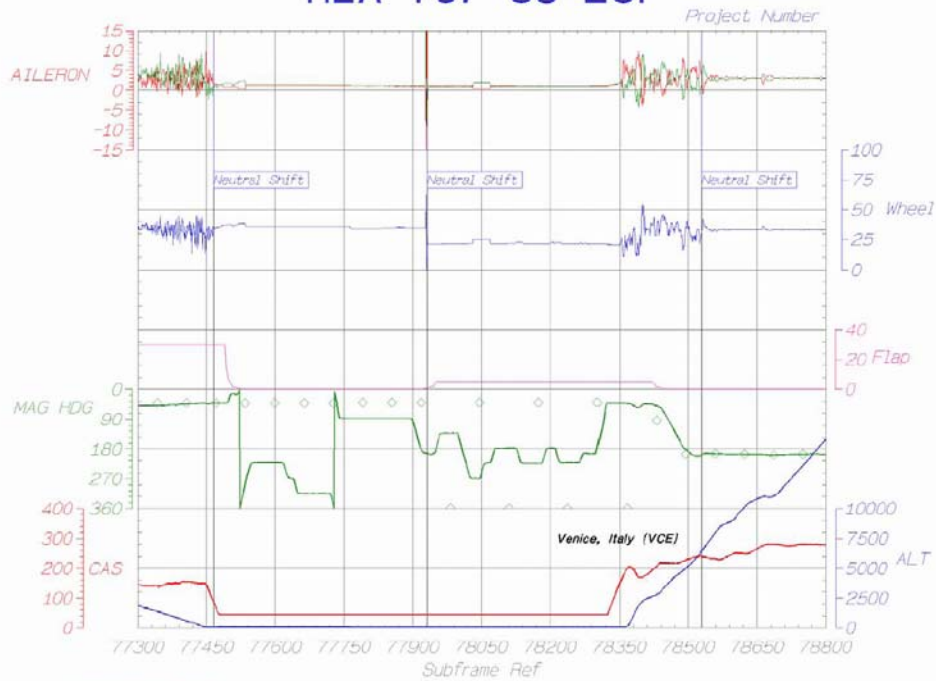
- 4) Relative to the photo at time 92470, does the EADI have the feature that forces the blue/brown line to always be present, even in unusual attitudes?  
*Response: Yes, the forced blue/brown interface is present unless pitch attitude exceeds 85 degrees (up or down), at which point it is removed.*



Preliminary Data  
Created: March 01, 2004

Wheel bias shifts during landing at SSH, control sweep on ground at SSH, and takeoff on accident flight from SSH.

Boeing Proprietary

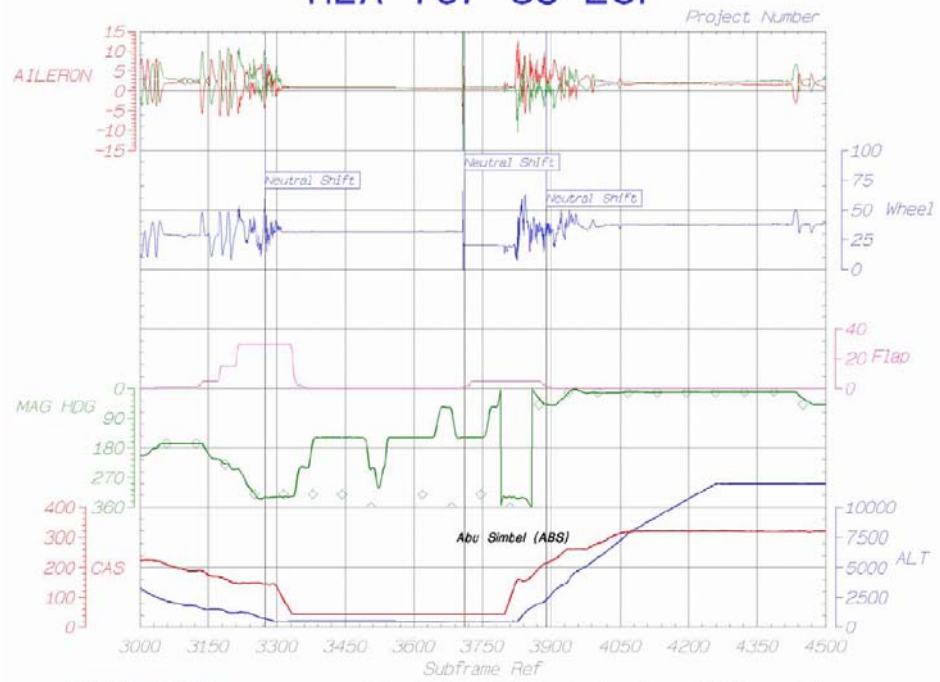


Preliminary Data  
Created: March 01, 2004

Wheel bias shifts during landing at VCE, control sweep on ground at SSH, and takeoff on previous flight from VCE.

Boeing Proprietary



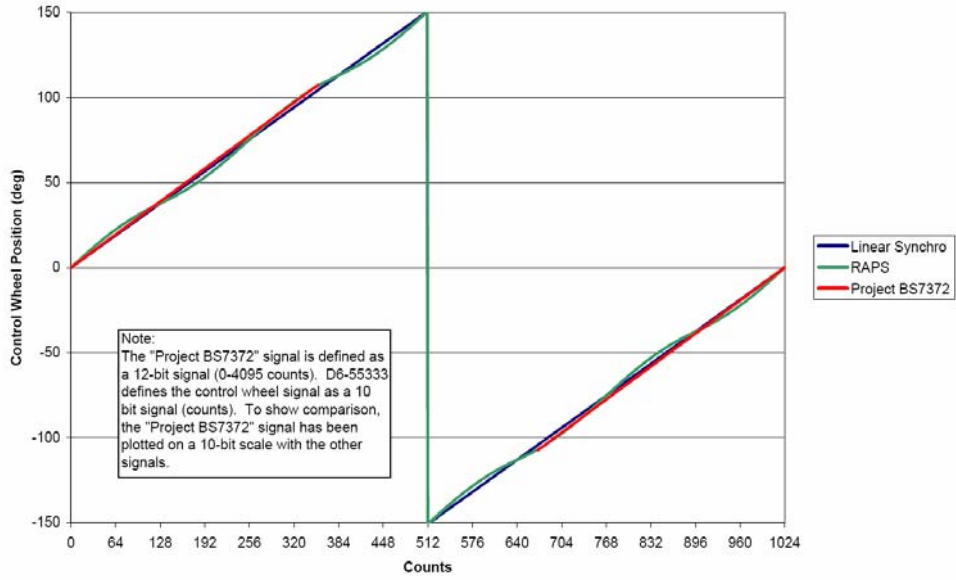


Preliminary Data  
Created: March 01, 2004

Wheel bias shifts during landing at ABS, control sweep on ground at ABS, and first recorded takeoff from ABS.

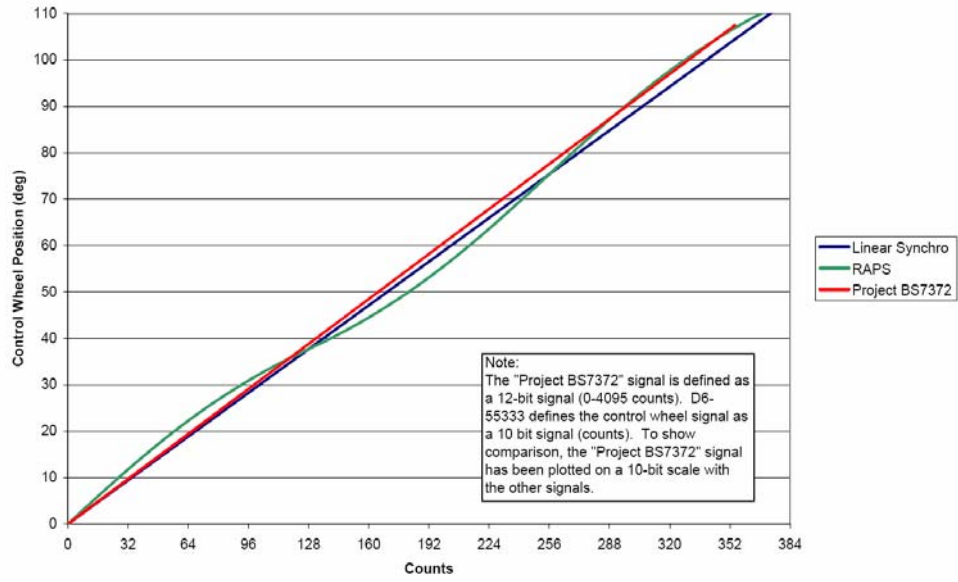
Boeing Proprietary

Control Wheel Conversions



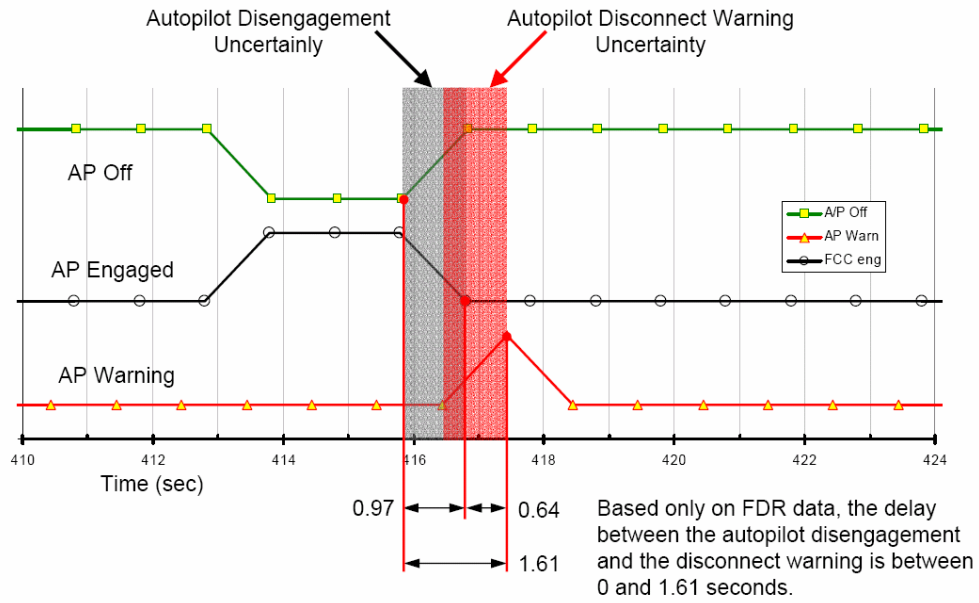
Boeing Proprietary

Control Wheel Conversions



Boeing Proprietary

# Autopilot Disengagement *with Time Aligned Data*



## Autopilot - Answers To Questions

A1) Why did the autopilot disengage?

*Answer: There are three possible reasons why the autopilot disengaged: the engage synchronization (actuator to surface) failed to complete; the engage hold interlocks were not satisfied; or it was manually disconnected. Based on the data recorded on the FDR, we are not able to pinpoint which of these caused the autopilot to disengage. Additional information on the details of the interlocks and their operation are provided in the Airplane Maintenance Manual section 22-11-01 . ①*

B3) Investigate the cause(s) for the autopilot disconnect.

*Answer: See response to question A1.*

B6) Investigate availability of autopilot during the captain's requests for "autopilot, autopilot".

*Answer: The autopilot will not initiate the engage sequence if the A/P engage interlocks are not satisfied (ref AMM 22-11-01 page 54). If the engage interlocks are not satisfied, the attempt to engage (A/P button push) will not be recorded on the FDR. In the case of the accident flight it's possible that forces on the column or wheel prevented the engage logic from being satisfied. Additional information on the details of the interlocks and their operation are provided in the Airplane Maintenance Manual section 22-11-01. ①*

## ***Answers to Questions from 31 Jan 2005 Meeting***

- **Q 1 – What can occur during the A/P engage sequence or after that would cause an aileron command change of 2.91 degrees during R CWS?**
  1. **Input from wheel/force sensors**
    - Pilot command
    - Force sensor failure (CWS command rate to be evaluated against change)
  2. **Heading Hold submode entered**
    - Requires Roll Angle < 6 deg
    - FDR data = -6.7 deg at autopilot engage in left IRU, right IRU used and data not known
    - FDR aileron rates are above the A/P CWS command rates for Heading Hold
  3. **Misrigging or Failure of Quadrant Position Sensor or Actuator LVDT**
    - Actuator LVDT position information continuously monitored for failures
    - Results in successful A/P synchronization when sensors match but surface and actuator do not match mechanically
    - A/P operation did not reflect this in previous flights

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## ***Answers to Questions from 31 Jan 2005 Meeting***

- Q2 – Provide better description of engage “jolt” for scenario 13, Hypothetical Scenarios # 2
  - If this fault exists when the autopilot is trying to engage, the engagement may occur with **minor wheel movement** as the A/P piston would be coupled to the ailerons before the position synchronization is complete

**Note: In-flight engage operation may differ from on ground engage due to aerodynamic loading on control surfaces versus only gravitation forces on surfaces on ground**

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## ***Answers to Questions from 31 Jan 2005 Meeting***

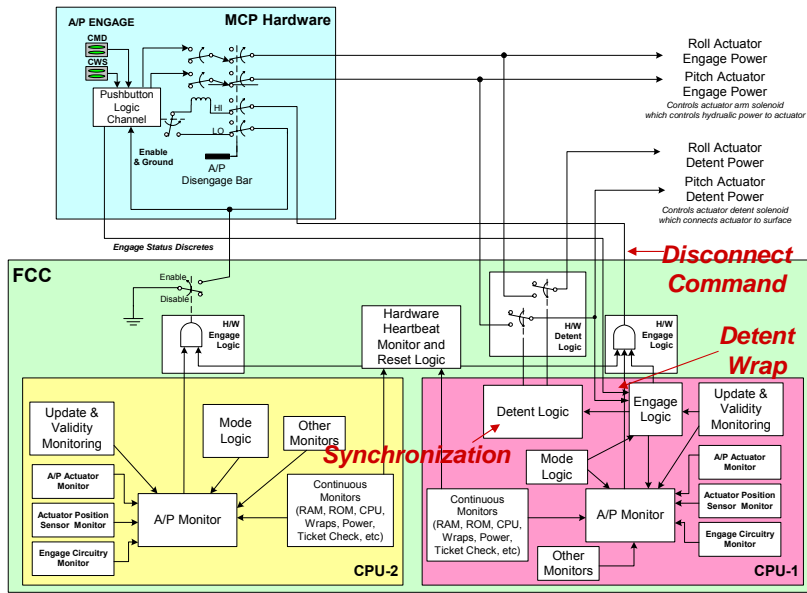
- **Q3 – Provide minimum time for disconnect given immediate A/P synchronization with no detent pressure**
  - FCC receives Local Command from MCP Engage Logic when the A/P CMD button is pressed
    - > Running Time: Start
  - Detent Command logic detects synchronization and sets Aileron Detent Command output (100 ms delay)
    - > Running Time : +100 ms
  - Engage Logic receives Aileron Detent Pressure Command Wrap (50 ms delay)
    - > Running Time : +150 ms
  - Engage Logic does not receive valid Aileron Detent Pressure Switch data and removes power from MCP engage hardware, (3.5 ms delay)
    - > Running Time : +3.65 seconds
  - MCP Engage Logic disconnect (minimum 45 ms, maximum 80 ms)
    - > Running Time : +3.695 seconds

**Minimum Time to A/P Disconnect with No Detent Pressure: 3.695 seconds**

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# Answers to Questions from 31 Jan 2005 Meeting



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## Answers to Questions from 31 Jan 2005 Meeting

- Q4 – Provide relative probability for A/P disconnect given signal invalid in scenario 10 b.

Item	Interlock or Condition	Prevents Engage	Causes Disengage	Probability	Comment
1	A/P Stab Trim Cutout Switch Normal	Yes	Yes	Unlikely	Pilot action or switch failure while A/P in CMD
2	Main Electric Trim Switches (not pressed)	Yes	Yes	Unlikely	Pilot must attempt manual trimming while A/P in CMD
3	A/P Stab Trim Motor Speed Interlock (10 sec)	Yes	Yes	Unlikely	Failure must occur during 2 seconds while A/P in CMD
4	Aileron Force Limiter Authority Limit Interlock (10 sec)	Yes	Yes	Unlikely	Failure must occur during 2 seconds while A/P in CMD
5	<i>Aileron Force Limiter Clutch - disengage</i>	Yes	No	FDR Data rules out	<i>This interlock is only used prior to A/P engage</i>
6	<i>Aileron Force Limiter Clutch - engage in 0.5 sec</i>	Yes	No	FDR Data rules out	<i>FDR recorded disconnect timing too long for this disconnect case</i>
7	A/P Disengage Switch	Yes	Yes	Possible	Pilot could have initiated disconnect
8	<i>A/P Aileron Hydraulic Pressure Switch - stuck in pressurized state</i>	Yes	No	FDR Data rules out	<i>This would have prevented initial engagement and, after engage, not be detectable until after disengage</i>
9	A/P Aileron Hydraulic Pressure Switch - pressure within 3.695 seconds after actuator detent solenoid engaged	No	Yes	FDR Data rules out	Minimum timing greater than FDR data by ~ 0.1 seconds
10	A/P Elevator Hydraulic Pressure Switch - stuck in pressurized state	No	Yes	FDR Data rules out	This would have prevented initial engagement and, after engage, not be detectable until after disengage

Possible cause

Unlikely cause

FDR Mismatch

*Italic Text*

Flight Condition Mismatch

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## Answers to Questions from 31 Jan 2005 Meeting

Item	Interlock or Condition	Prevents Engage	Causes Disengage	Probability	Comment
11	A/P Elevator Hydraulic Pressure Switch - pressure within 3.5 seconds after elevator actuator detent solenoid engaged	No	Yes	FDR Data rules out	Minimum timing greater than FDR data by ~ 0.1 seconds
12	115 VAC	Yes	Yes	Unlikely	Failure must occur during 2 seconds while A/P in CMD
13	28 VDC Engage Interlock Power	Yes	Yes	Unlikely	Failure must occur during 2 seconds while A/P in CMD
14	<i>Not (Foreign FCC In CMD And APP PB And Radio Altitude &lt; 800 ft)</i>	No	Yes	FDR Data rules out	<i>This prevents engage only in approach mode</i>
15	FCC DC And FCC Power Supply	Yes	Yes	FDR Data rules out	FCC continued to provide data to FDR throughout the flight
16	1800 Hz Power Supply	Yes	Yes	Unlikely	Failure must occur during 2 seconds while A/P in CMD
17	<i>Power Up Test Fails</i>	Yes	No	FDR Data rules out	<i>FCC continued to provide data to FDR throughout the flight</i>
18	Continuous Monitor(s) Fail	Yes	Yes	Unlikely	Failure must occur during 2 seconds while A/P in CMD
19	A/P Only Continuous Monitor Valid	Yes	Yes	Unlikely	Failure must occur during 2 seconds while A/P in CMD
20	<i>Less Than 3 lb Force On Control Wheel</i>	Yes	No	FDR Data rules out	<i>This only prevents engagement, will cause mode reversion to CWS with sufficient wheel force after A/P engage.</i>

Possible cause

Unlikely cause

FDR Mismatch

*Italic Text* Flight Condition Mismatch

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## Answers to Questions from 31 Jan 2005 Meeting

Item	Interlock or Condition	Prevents Engage	Causes Disengage	Probability	Comment
21	<i>Less Than 5 lb Force On Control Column</i>	Yes	No	FDR Data rules out	<i>This only prevents engagement, will cause mode reversion to CWS with sufficient column force after A/P engage</i>
22	Selected IRU Roll Angle Valid (norm - off side)	Yes	Yes	Unlikely	Failure must occur during 2 seconds while A/P in CMD
23	Selected IRU Roll Rate Valid (norm - off side)	Yes	Yes	Unlikely	Failure must occur during 2 seconds while A/P in CMD
24	Selected IRU Pitch Angle Valid (norm - on side)	Yes	Yes	Unlikely	Failure must occur during 2 seconds while A/P in CMD and FDR recorded valid Left IRU data
25	Selected IRU Pitch Rate Valid (norm - on side)	Yes	Yes	Unlikely	Failure must occur during 2 seconds while A/P in CMD and FDR recorded valid Left IRU data
26	<i>A/P to CMD and R/A &lt;400 Ft with LOC and GS engaged</i>	No	Yes	FDR Data rules out	<i>Only causes disconnect in approach mode</i>
27	<i>F/D in TO or GA, R/A Alt &lt;400 feet and A/P to CMD</i>	No	Yes	FDR Data rules out	<i>Only causes disconnect when TOGA mode selected</i>
28	ADC CAS Not Valid (except in dual channel operation)	Yes	Yes	Unlikely	Failure must occur during 2 seconds while A/P in CMD and FDR recorded valid Left DADC data
29	IRU Transfer	No	Yes	Unlikely	IRS transfer must occur in 2 seconds while A/P in CMD
30	A/P Engage Switch Swap	No	Yes	FDR Data rules out	FDR data indicates FCC B was not in CMD or CWS during the flight

Possible cause  
Unlikely cause  
FDR Mismatch
*Italic Text* Flight Condition Mismatch

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## Answers to Questions from 31 Jan 2005 Meeting

<i>Item</i>	<i>Interlock or Condition</i>	<i>Prevents Engage</i>	<i>Causes Disengage</i>	<i>Probability</i>	<i>Comment</i>
31	ADC Corrected Baro Altitude Valid	Yes	Yes	Unlikely	Failure must occur during 2 seconds while A/P in CMD and FDR recorded valid Left DADC data
32	ADC Uncorrected Baro Altitude Valid	Yes	Yes	Unlikely	Failure must occur during 2 seconds while A/P in CMD and FDR recorded valid Left DADC data
33	Local Power Bus Transfer	No	Yes	FDR Data rules out	No bus transfers in FDR data
34	Failure Of Aileron Axis To Synchronize	No	Yes	Unlikely	Disengage after 4 seconds of CMD
35	Failure Of Elevator Axis To Synchronize	No	Yes	Unlikely	Disengage after 4 seconds of CMD
36	(RCWS) and (Heading Hold (bank angle < 6 deg)) and (TAS Or Heading Invalid)	No	Yes	Unlikely	Only applicable to Heading Hold mode, Left IRS data showed 6.7 degrees Roll Angle from engage through disconnect

Possible cause  
 Unlikely cause  
 FDR Mismatch

*Italic Text* Flight Condition Mismatch

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## ***Answers to Questions from 31 Jan 2005 Meeting***

- **Q5 – What are the causes for reversion of Heading Select mode to Roll CWS (Control Wheel Steering) when the A/P is engaged?**
  1. Pressing the Heading Select pushbutton on MCP (when Heading Select mode active)
  2. Applying greater than 10 lbs of wheel force after A/P is Engaged
    - A/P needs to be engaged in this case or the wheel force will prevent engagement
  3. Losing True Airspeed (TAS) or Magnetic Heading validity
    - Validity can be lost prior to A/P engage attempt without affecting mode
    - Causes Roll F/D bias out-of-view (BOV) when A/P is not engaged
    - True Airspeed invalid will also cause Level Change to change to CWS P when A/P is engaged and Pitch F/D BOV when F/D On
  4. F/D Bar Command greater than 7 degrees of bank error (Performance Assessment Monitor (PAM) invalid)
    - Based on FDR data, F/D bank error > 7 degrees was present for more than 9 seconds prior to A/P engagement

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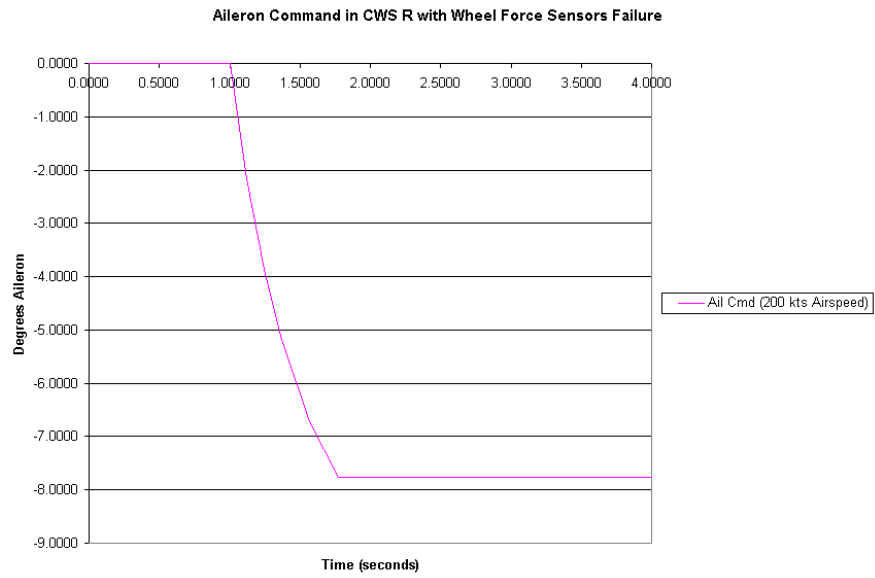
## ***Answers to Questions from 2 Feb 2005 Meeting***

- What is the maximum rate of the Roll CWS command given a failure of the Wheel Force Transducer? (Scenario 9)
  - 15.5 lbs maximum input into the Roll CWS control law based on hardware input scaling limit
  - Command is multiplied by scaling factors and lagged prior to output for a maximum steady state output of 7.64 degrees of aileron (limited by wheel limit) about 0.77 seconds after fault occurs
  - More than 3 lbs force sensor input prevents engagement, so failure in time sequence dependent with the 2.6 to 3.6 second CWS R engage period.

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# Answers to Questions from 2 Feb 2005 Meeting



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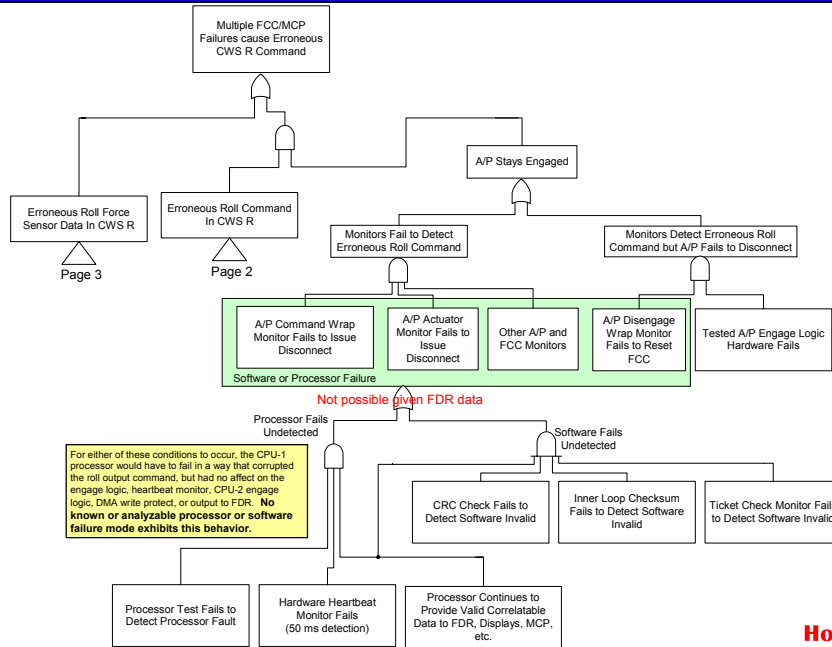
## *Answers to Questions from 2 Feb 2005 Meeting*

- What failures of the Flight Control Computer would cause the A/P to command a 3.64 degree aileron change in Roll CWS? (Scenario 9)

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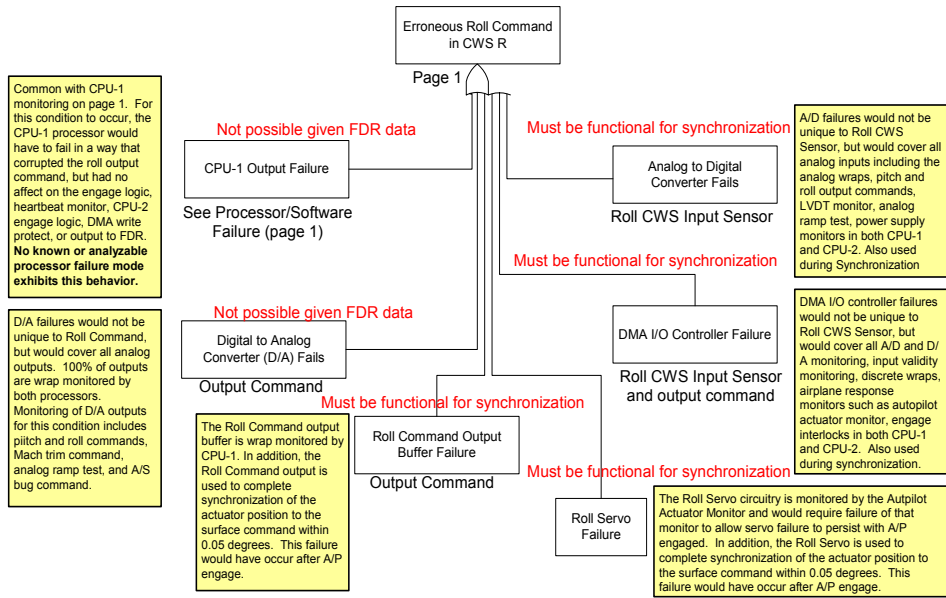
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# Answers to Questions from 2 Feb 2005 Meeting



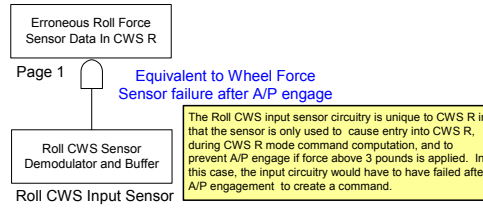
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# Answers to Questions from 2 Feb 2005 Meeting



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## Answers to Questions from 2 Feb 2005 Meeting



Note :This failure does not inhibit manual disconnect nor does it result in failure to disconnect with erroneous FDR disengaged indication. When this failure occurs, pilot can override erroneous command with normal autopilot override forces.

## Answers to Questions from 2 Feb 2005 Meeting

- Provide minimum time for disconnect given immediate A/P synchronization with no detent pressure (Scenario 10)
  - FCC receives Local Command from MCP Engage Logic when the A/P CMD button is pressed
    - > Running Time: Start
  - Detent Command logic detects synchronization and sets Aileron Detent Command output (100 ms delay, based on 0.005% real time clock oscillator/timer)
    - > Running Time : +100 ms  $\pm$  0.005 ms
  - Engage Logic receives Aileron Detent Pressure Command Wrap (50 ms delay, based on 0.005% real time clock oscillator/timer)
    - > Running Time : +150 ms  $\pm$  0.0075 ms
  - Engage Logic does not receive valid Aileron Detent Pressure Switch data and removes power from MCP engage hardware, 3.5 ms delay, based on 50 ms task driven by 0.005% real time clock oscillator timer)
    - > Running Time : +3.65 seconds  $\pm$  0.0001825 seconds
  - MCP Engage Logic disconnect (minimum 45 ms, maximum 80 ms, no additional tolerance)
    - > Running Time : +3.695 seconds  $\pm$  0.0001825 seconds

*Note: No input time penalty assumed through DMA I/O controller. Assumes all I/O exactly aligned in time with input/output timing.*

**Minimum Time to A/P Disconnect with No Detent Pressure: 3.6948175 seconds**

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## ***Answers to Autopilot Questions from 2 Feb 2005***

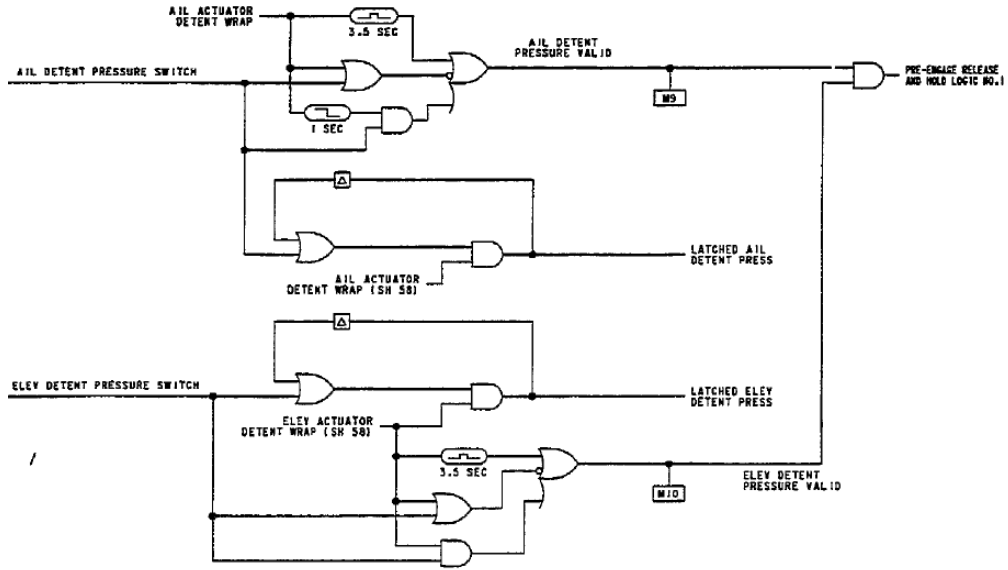
- What is the maximum time for the autopilot to disconnect given the detent solenoid is stuck open prior to A/P engagement? (Scenario 10)
  - The time to the hydraulic pressurization and subsequent detent pressure switch reaction is a maximum of 50 ms
  - DMA I/O cycle maximum time delay of 536  $\mu$ sec for detent pressure input
  - The detent logic of the engage interlocks is executed at 20 Hz (50 ms) so a maximum of one frame delay due to just missing the input data.
  - There is no software delay in reaction to detent pressure input by engage interlocks prior to detent command output.
  - DMA I/O cycle maximum time delay of 536  $\mu$ sec for disconnect command
  - The MCP engage circuitry react in 45 to 80 ms of the processor issuing a disconnect.

Note: This logic is depicted in the SP-300 DFCS Training Manual Volume 4 Sheet 54

**Worst case time the FCC to disconnect for this case is 181.072 ms.**

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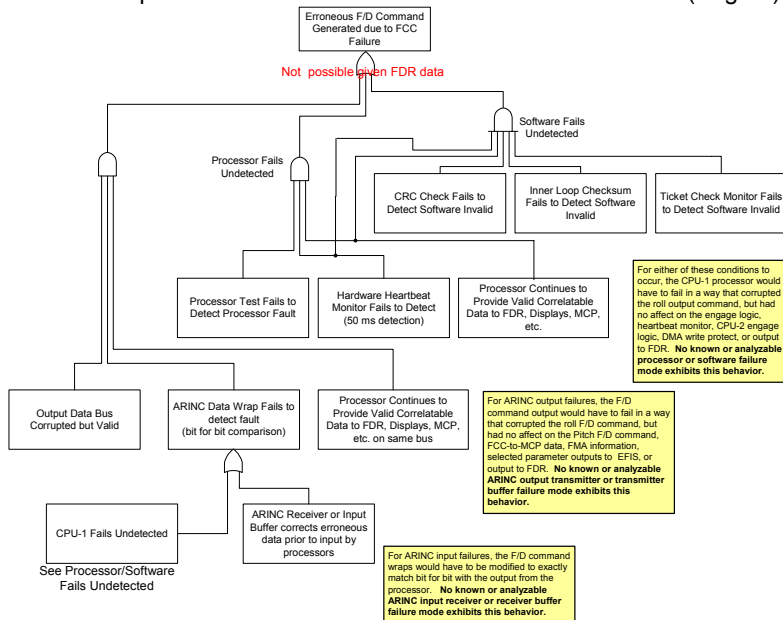
# Answers to Autopilot Questions from 2 Feb 2005



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# Answers to Autopilot Questions from 2 Feb 2005

## Scenario 5 & 11 – Multiple FCC Failures Cause Erroneous F/D Command (Page 5)



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## ***Answers to Autopilot Questions from 2 Feb 2005***

- Scenario 13 – Multiple FCC failures cause erroneous A/P engage and erroneous command output
  - FCCs *CANNOT* engage A/P on their own regardless of failure mode.
    - MCP engage hardware is in control of autopilot engage. FCCs can only disable or enable MCP engage hardware.
    - From page 9 of Scenario 13, since the FCC self-engages, the multiple FCC fault case, the IRU fault case, and the bank limit fault case (page 10) cannot be a function of FCC failure.

Answers to question\_cairo meeting05.ppt

**Boeing/ Honeywell proprietary information and will not be available for public use**

**Action Item Response.ppt (Cairo meeting, 1-30-05 to 2-2-05), Boeing Action Items of 30 January (public release).ppt**

Question 1

Does the aileron PCU bypass valve interconnect the extend and retract side of the main ram when no hydraulic pressure is available?

What is the correct hydraulic schematic for the PCU?

Question 2

Q) Reference Scenario 9 - What will happen to lateral trim capability after the 12 degrees of lost motion is taken up?

A) Lateral trim capability will be limited to +/- 12 degrees of wheel. The force required to break out the transfer mechanism (50 Lb) is in excess of the feel and centering force (~20 Lb peak).

Question 3

What is the airplane level effect of lateral control scenario #9 (spoiler control drum jammed at neutral)?

Boeing to run desktop simulation

Question 4

Provide proposed corrections to scenario #10 write up

See rewrite.

Question 5

Q) Reference Scenario 9-10 – What is breakout force of the aileron spring cartridge?

A) Breakout force of the aileron spring cartridge (reflected at the control wheel) is approximately 16 Lb.

Question 6

Q) Reference Scenario 16 – What is the effect of a failure in the PCA input rod (A or B)?

A) There is no functional effect of a single failure in the PCA input rod. The entire input rod and fasteners are dual load path. The effect of a multiple failure depends on the position of the primary slide at the time of the failure. Worst case effect is a rate jam of the affected PCU, causing a force fight with the other PCU and stalling of both PCUs. Control of spoilers is available from the FO side if the transfer mechanism is broken out. Lateral trim will not be available. Depressurizing the affected PCU will restore normal control.

Question 7

Q) Reference Scenario 17 – What is the effect of a jam between the primary and secondary slide in the aileron PCA?

1. If the primary slide and secondary slide jam together near neutral, the effect is a minor reduction in rate capability.

2. If the jam occurs away from neutral, the feedback motion of the PCU will cause the primary and secondary slides to counter each other (crossflow condition). At a full crossflow condition, the PCU will lose rate capability and be backdriven by the unaffected PCU.

Question 8

Q) Reference Scenario 18 – What is the effect of a jam between the secondary slide and the sleeve in the aileron PCA?

1. If the secondary slide jams near neutral, the effect is a minor reduction in rate capability.

2. If the jam occurs away from neutral, the feedback motion of the PCU will cause the primary and secondary slides to counter each other (crossflow condition). At a full crossflow condition, the PCU will lose rate capability and be backdriven by the unaffected PCU.

Question 9, 10

Q) Reference Scenarios 20, 21 – What is the effect of a piston to cylinder jam in the aileron PCA?

The effect is same as a jam elsewhere in the captain's side aileron control path. The FO must break out the transfer mechanism and aileron spring rod to move the spoilers. Aileron control is limited to deflections within the valve stops.

Question 11

Provide proposed corrections to scenario #34 write up  
See rewrite.

Question 12

Provide proposed corrections to scenario #36 write up  
See rewrite.

Question 13

Provide proposed corrections to scenario #47 write up  
See rewrite

1.16.2. Tests and researches conducted by NTSB:

C.wheel NTSB.ppt



# Introduction

- Define Sensor Malfunction
- Evaluate Data Quality
- Validate Control Wheel Adjustments

6/28/2005

2

# Discussion Points

- Fact - Control Wheel Sensor Maximum Minimum Values Recorded on 25-Hours of FDR data (-2.237deg to 81.5 deg)
- Theory - Control Wheel Sensor Moved Freely Within Active Range (-2.237 and 81.5 degrees.), But due to Internal Binding of Rotating Components will not Exceed this Range.
- Theory - Control Wheel Inputs Outside of Active Range Cause Sensor to Rotate in Mounting Bracket and Reposition Control Wheel Sensor/Cockpit Control Wheel Offset.
- Theory - Rapid Control Wheel Inputs Will Also Cause Sensor to Shift in Mounting Bracket.
- Theory - Control Wheel Sensor Values Can Be Used to Evaluate Crew Inputs When Sensor Offset can be Derived From Known Control Wheel Position (i.e. Before and After Preflight Control Checks, 0 - Aileron Deflection.)

6/28/2005

3

## Discussion Points (cont.)

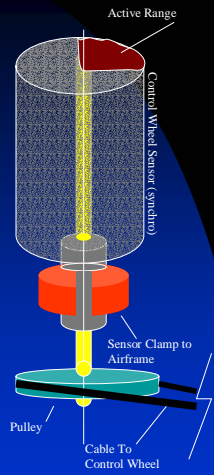
- Control Wheel Position Sensor is a synchro with a range of 0 to 360 degrees or +/- 180 degrees.
- Full Range of Control Wheel as expressed in sensor units (synchro angles) is +/- 128 degrees.
- Full Range of Control Wheel Travel as measured in cockpit is +/- 107 degrees.
- The following discussion will reference sensor units only (ie, synchro angles +/- 128 degrees)
- Theory – Control Wheel Position (Cockpit) values recorded during accident flight can be corrected to actual by applying the following offsets:
  - From Frame 92250 to 92361.92 subtract 17.5444 deg.
  - From Frame 92362.42 to 92445 subtract 28.9 deg.
  - From Frame 92446 to end of data 28.9 deg sensor offset may not apply due to rapid control wheel inputs.

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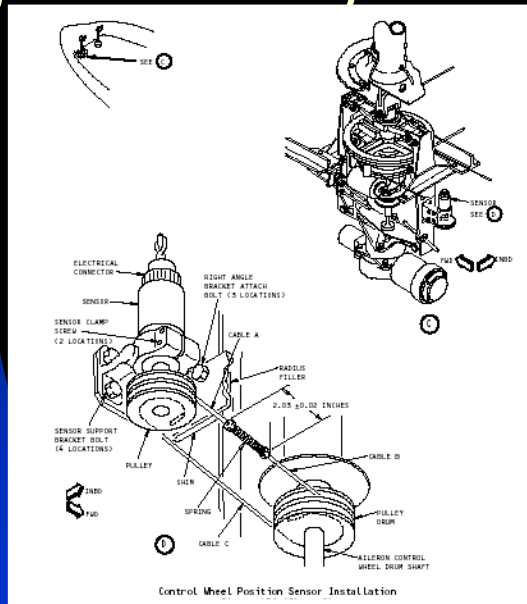
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# Binding Sensor Theory



6/28/2005



Control Wheel Position Sensor Installation

5

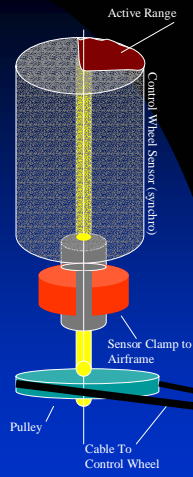
# Control Wheel Position Sensor Values for Neutral Aileron Before & After Preflight Control Checks

	Time in Seconds (FDR Sub Frame)	Control Wheel Position		Control Check Direction
		Before Check	After Check	
1	3713	29.4466	16.7846	Rt. To LT.
2	5568	31.2134	0	Lt. To Rt.
3	7801	58.8932	2.35573	Lt. To Rt.
4	9789	33.8636	3.23913	Lt. To Rt.
5	12124	31.8023	0.294466	Lt. To Rt.
6	14134	28.5632	16.4901	Rt. To LT.
7	17431	29.1521	14.7233	Rt. To LT.
8	22682	30.6245	16.7846	Rt. To LT.
9	30419	37.6915	15.012	Rt. To LT.
10	46964	30.6245	14.1344	Rt. To LT.
11	62156	35.6304	15.6067	Rt. To LT.
12	77924	32.9802	17.668	Rt. To LT.
13	92030	33.5691	14.4288	Rt. To LT.

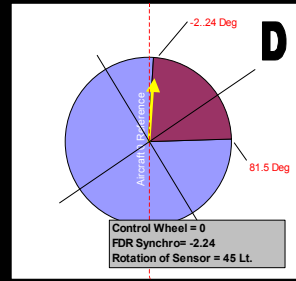
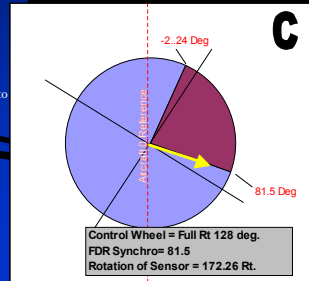
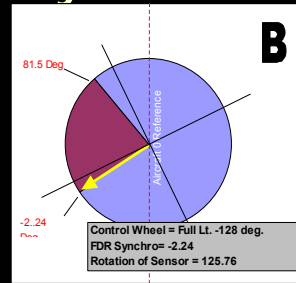
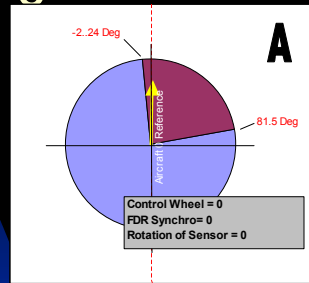
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# Binding Sensor Theory

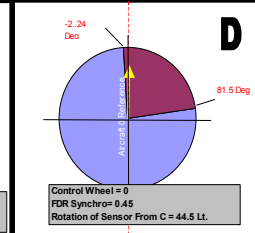
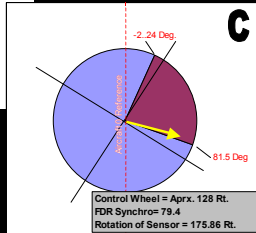
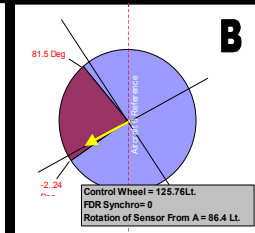
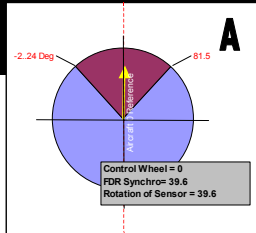
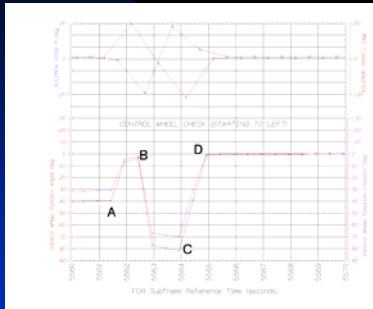


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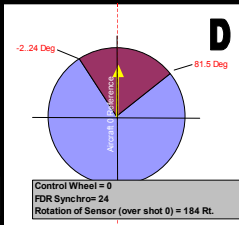
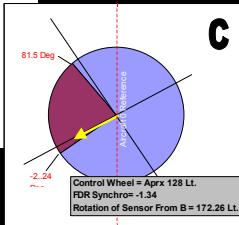
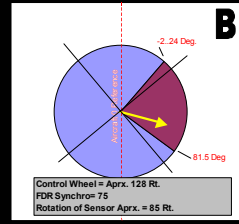
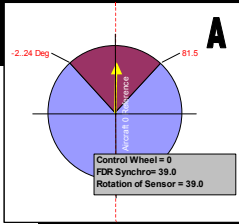
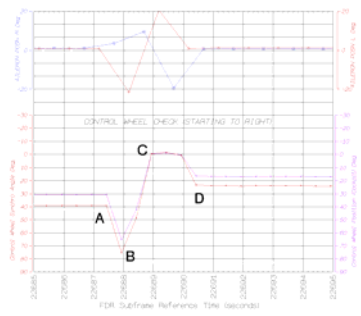
# Evaluation of Preflight Control Wheel Check (Starting to the Lt.)



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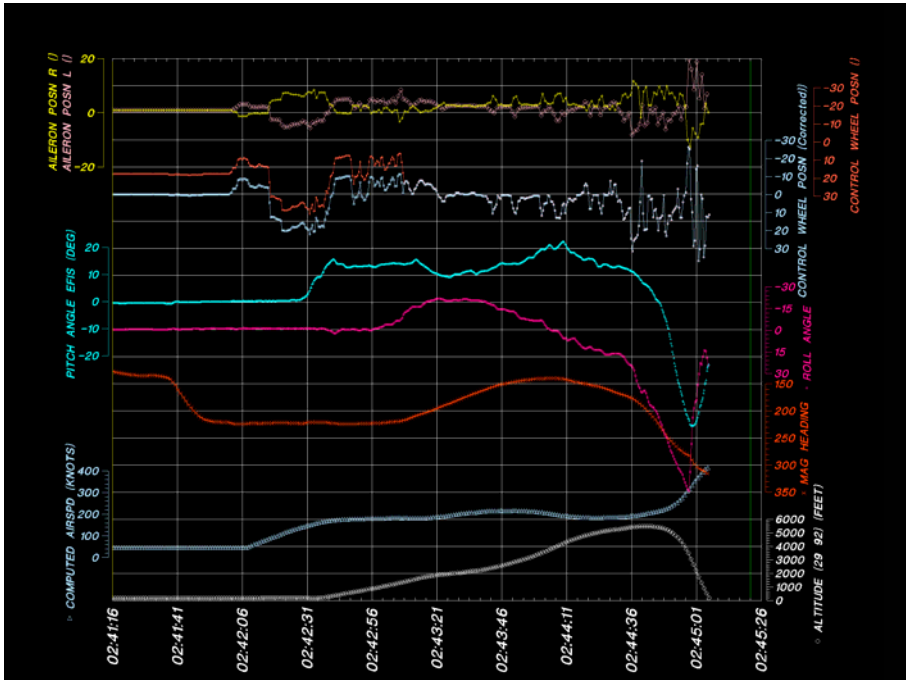
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# Evaluation of Preflight Control Wheel Check (Starting to the Rt.)



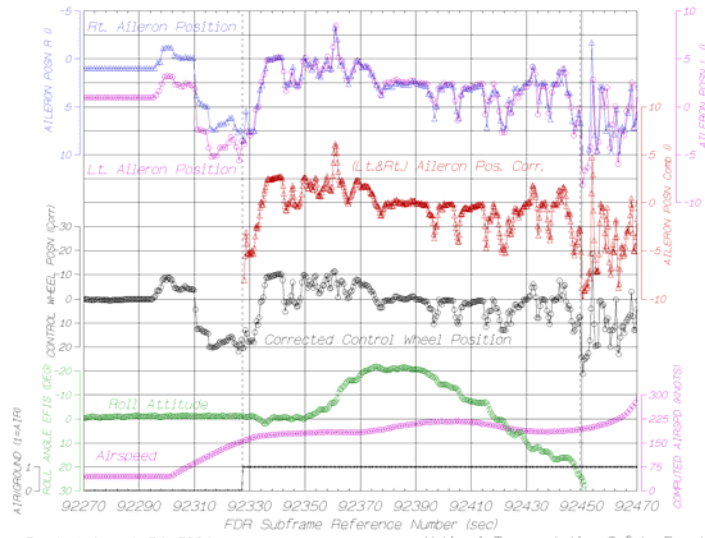
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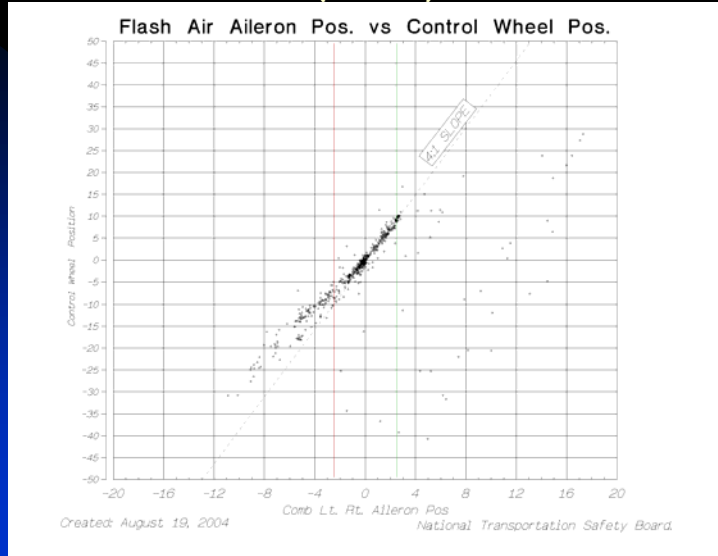
# Correlation of Aileron and Control Wheel Position Data.

Flash Air B737, Correlation Aileron & Contl. Wheel



Created: August 24, 2004 National Transportation Safety Board.

# Cross Plot – Aileron Pos. (Comb.) vs Control Wheel Pos.

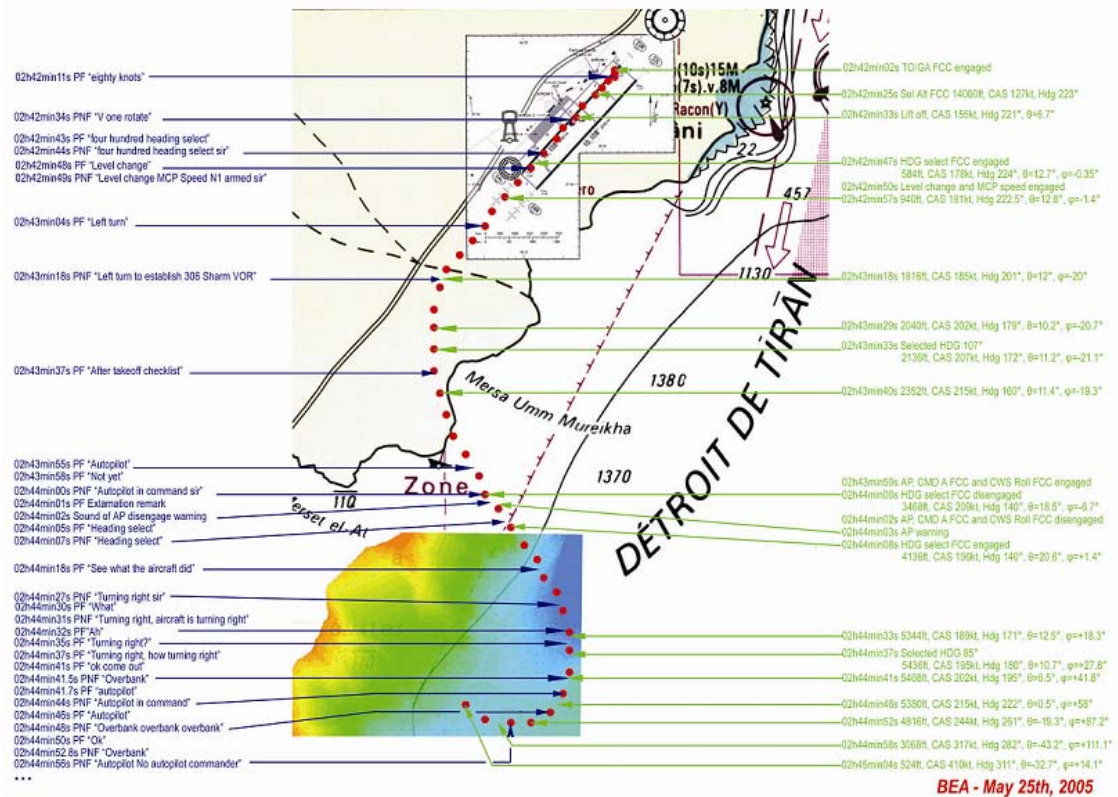


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1.16.3. Tests and researches conducted by BEA  
(Trajecto\_may05.jpg)



1.16.4. Tests and researches introduced by MCA:

Spatial Disorientation<sup>20</sup>

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<sup>20</sup> Reference: "U.S. Army Field Manual, FM 3-04.301, Aeromedical Training for Flight Personnel"

## Spatial Disorientation

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## Spatial Disorientation

Spatial disorientation contributes more to causing aircraft accidents than any other physiological problem in flight. Regardless of their flight-time experience, all aircrew members are subject to disorientation. The human body is structured to perceive changes in movement on land in relation to the surface of the earth. In an aircraft, the human sensory systems—the visual, vestibular, and proprioceptive systems—may give the brain erroneous orientation information. This information can cause sensory illusions, which may lead to spatial disorientation.

### COMMON TERMS OF SPATIAL DISORIENTATION

#### SPATIAL DISORIENTATION

9-1. Spatial disorientation is an individual's inability to determine his or her position, attitude, and motion relative to the surface of the earth or significant objects; for example, trees, poles, or buildings during hover. When it occurs, pilots are unable to see, believe, interpret, or prove the information derived from their flight instruments. Instead, they rely on the false information that their senses provide.

#### SENSORY ILLUSION

9-2. A sensory illusion is a false perception of reality caused by the conflict of orientation information from one or more mechanisms of equilibrium. Sensory illusions are a major cause of spatial disorientation.

#### VERTIGO

9-3. Vertigo is a spinning sensation usually caused by a peripheral vestibular abnormality in the middle ear. Aircrew members often misuse the term vertigo, applying it generically to all forms of spatial disorientation or dizziness.

### TYPES OF SPATIAL DISORIENTATION

#### TYPE I (UNRECOGNIZED)

9-4. A disoriented aviator does not perceive any indication of spatial disorientation. In other words, he does not think anything is wrong. What he sees—or thinks he sees—is corroborated by his other senses. Type I disorientation is the most dangerous type of disorientation. The pilot—unaware of a problem—fails to recognize or correct the disorientation, usually resulting in a fatal aircraft mishap:

- The pilot may see the instruments functioning properly. There is no suspicion of an instrument malfunction.
- There may be no indication of aircraft-control malfunction. The aircraft is performing normally.
- An example of this type of SD would be the height-/depth-perception illusion when the pilot descends into the ground or some obstacle above the ground because of a lack of situational awareness.

#### TYPE II (RECOGNIZED)

9-5. In Type II spatial disorientation, the pilot perceives a problem (resulting from spatial disorientation). The pilot, however, may fail to recognize it as spatial disorientation:

- The pilot may feel that a control is malfunctioning.

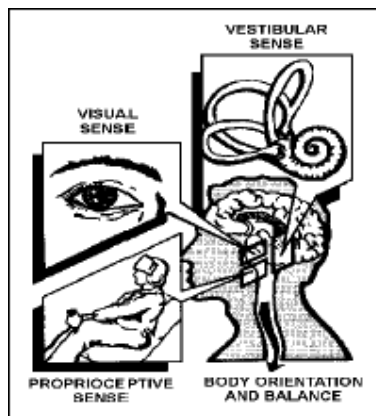
- The pilot may perceive an instrument failure as in the graveyard spiral, a classic example of Type II disorientation. The pilot does not correct the aircraft roll, as indicated by the attitude indicator, because his vestibular indications of straight-and-level flight are so strong.

### **TYPE III (INCAPACITATING)**

9-6. In Type III spatial disorientation, the pilot experiences such an overwhelming sensation of movement that he or she cannot orient himself or herself by using visual cues or the aircraft instruments. Type III spatial disorientation is not fatal if the copilot can gain control of the aircraft.

### **EQUILIBRIUM MAINTENANCE**

9-7. Three sensory systems—the visual, vestibular, and proprioceptive systems—are especially important in maintaining equilibrium and balance. Figure 9-1 shows these systems. Normally, the combined functioning of these senses maintains equilibrium and prevents spatial disorientation. During flight, the visual system is the most reliable. In the absence of the visual system, the vestibular and proprioceptive systems are unreliable in flight.



**Figure 9-1. The Three Equilibrium Systems**

### **VISUAL SYSTEM**

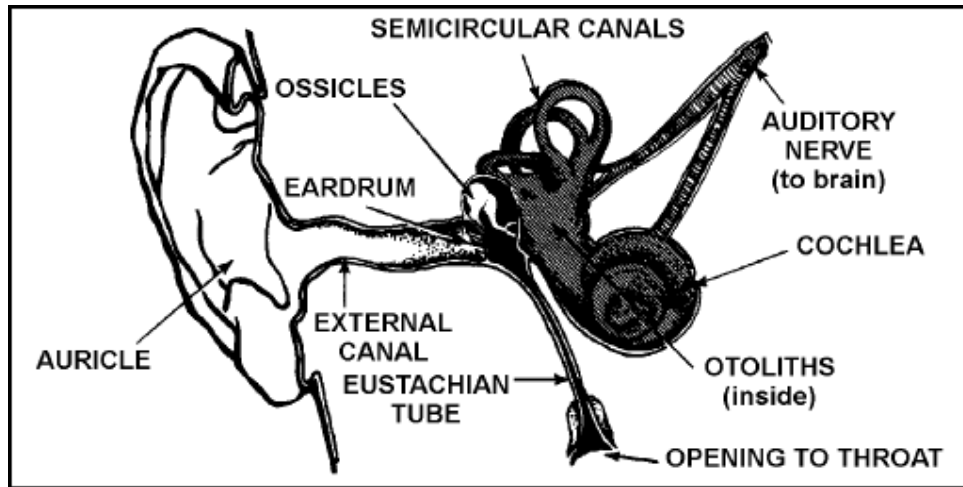
9-8. Of the three sensory systems, the visual system is the most important in maintaining equilibrium and orientation. To some extent, the eyes can help determine the speed and direction of flight by comparing the position of the aircraft relative to some fixed point of reference. Eighty percent of our orientation information comes from the visual system. (Chapter 8 contains information about the eye).

9-9. On flights under IMC, crew members lose fixed points of reference outside of the aircraft. Under IMC, the pilot must rely on visual sensory input from the instruments for spatial orientation. The decision to rely on the visual sense—and to believe the instruments rather than the input of the other senses—demands disciplined training.

9-10. The eyes allow the pilot to scan sensitive flight instruments that give accurate spatial-orientation information. These instruments indicate unusual aircraft attitudes resulting from turbulence, distraction, inattention, mechanical failure, or spatial disorientation.

### **VESTIBULAR SYSTEM**

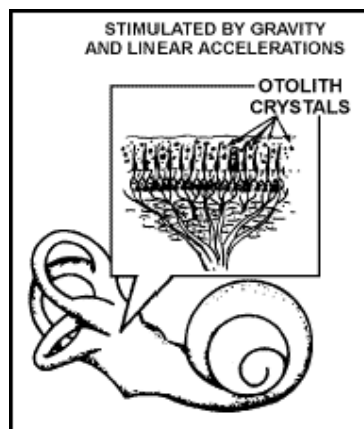
9-11. The inner ear contains the vestibular system, which contains the motion- and gravity-detecting sense organs. This system is located in the temporal bone on each side of the head. Each vestibular apparatus consists of two distinct structures: the semicircular canals and the vestibule proper, which contain the otolith organs. Figure 9-2 depicts the vestibular system. Both the semicircular canals and the otolith organs sense changes in aircraft attitude. The semicircular canals of the inner ear sense changes in angular acceleration and deceleration.



**Figure 9-2. The Vestibular System**

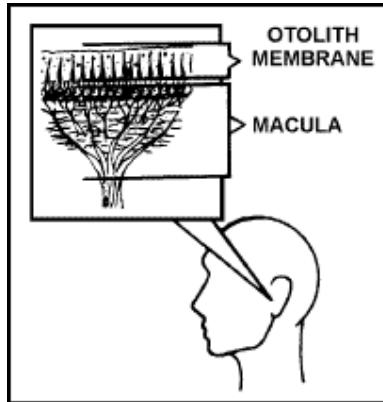
### Otolith Organs

9-12. The otolith organs are small sacs located in the vestibule. Sensory hairs project from each macula into the otolithic membrane, an overlaying gelatinous membrane that contains chalklike crystals, called otoliths. The otolith organs, shown in Figure 9-3, respond to gravity and linear accelerations/decelerations. Changes in the position of the head, relative to the gravitational force, cause the otolithic membrane to shift position on the macula. The sensory hairs bend, signaling a change in the head position.



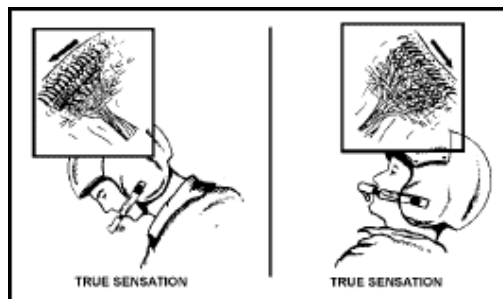
**Figure 9-3. The Otolith Organs**

9-13. When the head is upright, a "resting" frequency of nerve impulses is generated by the hair cells. Figure 9-4 shows the position of the hair cells when the head is upright.



**Figure 9-4. Position of the Hair Cells When the Head Is Upright**

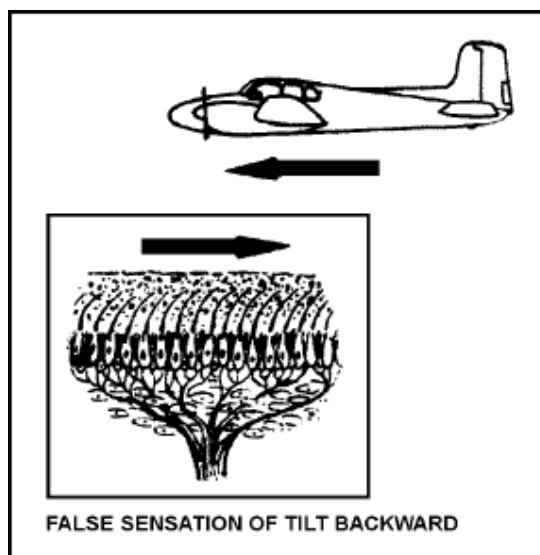
9-14. When the head is tilted, the "resting" frequency is altered. The brain is informed of the new position. The positions of the hair cells when the head is tilted forward and backward are shown in Figure 9-5.



**Figure 9-5. Position of the Hair Cells When the Head Is Tilted Forward and Backward**

9-15. Linear accelerations/decelerations also stimulate the otolith organs. The body cannot physically distinguish between the inertial forces resulting from linear accelerations and the force of gravity. A forward acceleration results in backward displacement of the otolithic membranes. When an adequate visual reference is not available, aircrew members may experience an illusion of backward tilt. Figure 9-6 shows this false sensation of backward tilt.

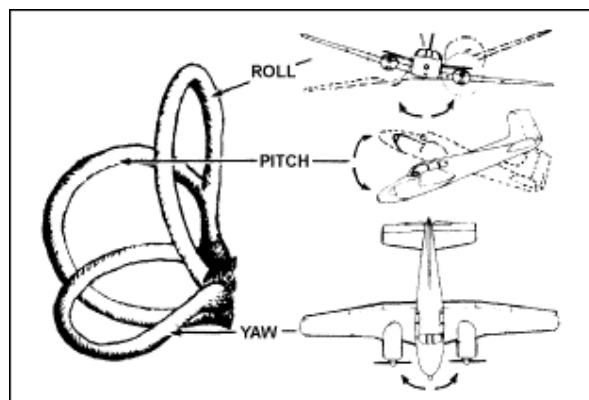




**Figure 9-6. False Sensation During Backward Tilt**

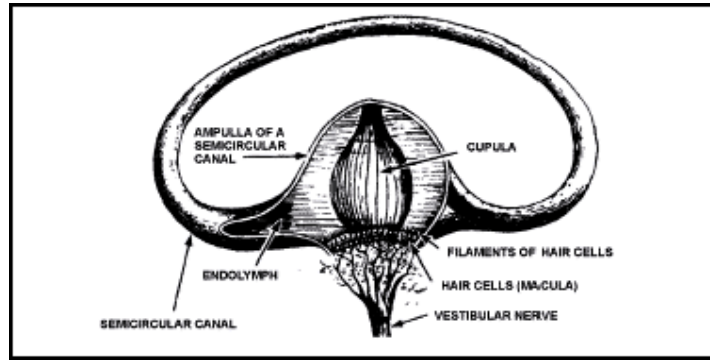
### **SEMICIRCULAR CANALS**

9-16. The semicircular canals of the inner ear sense changes in angular acceleration. The canals will react to any changes in roll, pitch, or yaw attitude. [Figure 9-7](#) shows where these changes are registered in the semicircular canals.



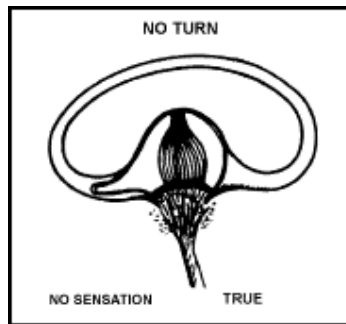
**Figure 9-7. Reaction of the Semicircular Canals to Changes in Angular Acceleration**

9-17. The semicircular canals are situated in three planes, perpendicular to each other. They are filled with a fluid called endolymph. The inertial torque resulting from angular acceleration in the plane of the canal puts this fluid into motion. The motion of the fluid bends the cupula, a gelatinous structure located in the ampulla of the canal. This, in turn, moves the hairs of the hair cells situated beneath the cupula. This movement stimulates the vestibular nerve. These nerve impulses are then transmitted to the brain, where they are interpreted as rotation of the head. [Figure 9-8](#) shows a cutaway section of the semicircular canal.



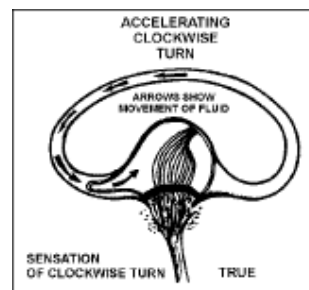
**Figure 9-8. Cutaway View of the Semicircular Canals**

9-18. When no acceleration takes place, the hair cells are upright. The body senses that no turn has occurred. The position of the hair cells and the actual sensation correspond, as shown in Figure 9-9.



**Figure 9-9. Position of Hair Cells During No Acceleration**

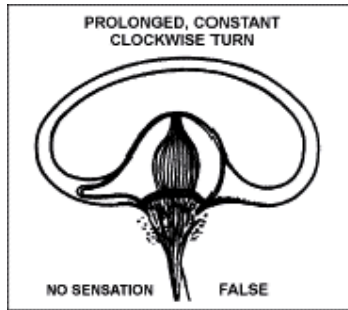
9-19. When a semicircular canal is put into motion during clockwise acceleration, the fluid within the semicircular canal lags behind the accelerated canal walls. This lag creates a relative counterclockwise movement of the fluid within the canal. The canal wall and the cupula move in the opposite direction from the motion of the fluid. The brain interprets the movement of the hairs to be a turn in the same direction as the canal wall. The body correctly senses that a clockwise turn is being made. Figure 9-10 shows the position of the hair cells and the resulting true sensation during a clockwise turn.



**Figure 9-10. Sensation During a Clockwise Turn**

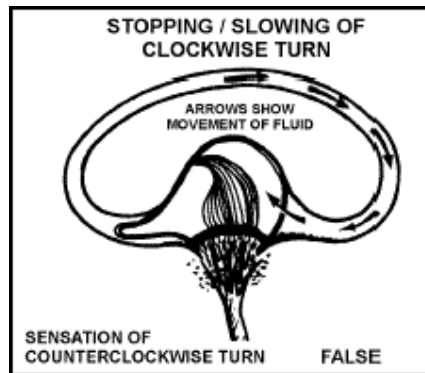
9-20. If the clockwise turn then continues at a constant rate for several seconds or longer, the motion of the fluid in the canals catches up with the canal walls. The hairs are no longer bent, and the brain receives the false impression that turning has stopped. The position of the hair cells and the resulting false sensation during a

prolonged, constant clockwise turn is shown in Figure 9-11. A prolonged constant turn in either direction will result in the false sensation of no turn.



**Figure 9-11. Sensation During a Prolonged Clockwise Turn**

9-21. When the clockwise rotation of the aircraft slows or stops, the fluid in the canal moves briefly in a clockwise direction. This sends a signal to the brain that is falsely interpreted as body movement in the opposite direction. In an attempt to correct the falsely perceived counterclockwise turn, the pilot may turn the aircraft in the original clockwise direction. Figure 9-12 shows the position of the hair cells—and the resulting false sensation when a clockwise turn is suddenly slowed or stopped.



**Figure 9-12. Sensation During Slowing or Stopping of a Clockwise Turn**

### **PROPRIOCEPTIVE SYSTEM**

9-22. This system reacts to the sensation resulting from pressures on joints, muscles, and skin and from slight changes in the position of internal organs. It is closely associated with the vestibular system and, to a lesser degree, the visual system. Forces act upon the seated pilot in flight. With training and experience, the pilot can easily distinguish the most distinct movements of the aircraft by the pressures of the aircraft seat against the body. The recognition of these movements has led to the term "seat-of-the-pants" flying.

### **VISUAL ILLUSIONS**

9-23. Illusions give false impressions or misconceptions of actual conditions; therefore, aircrew members must understand the type of illusions that can occur and the resulting disorientation. Although the visual system is the most reliable of the senses, some illusions can result from misinterpreting what is seen; what is perceived is not always accurate. Even with the references outside the cockpit and the display of instruments inside, aircrew members must be on guard to interpret information correctly.

## RELATIVE-MOTION ILLUSION

9-24. Relative motion is the falsely perceived self-motion in relation to the motion of another object. The most common example is when an individual in a car is stopped at a traffic light and another car pulls alongside. The individual that was stopped at the light perceives the forward motion of the second car as his own motion rearward. This results in the individual applying more pressure to the brakes unnecessarily. This illusion can be encountered during flight in situations such as formation flight, hover taxi, or hovering over water or tall grass.

## CONFUSION WITH GROUND LIGHTS

9-25. Confusion with ground lights occurs when an aviator mistakes ground lights for stars. This illusion prompts the aviator to place the aircraft in an unusual attitude to keep the misperceived ground lights above them. Isolated ground lights can appear as stars and this could lead to the illusion that the aircraft is in a nose high or one wing low attitude (Part A of Figure 9-13). When no stars are visible because of overcast conditions, unlighted areas of terrain can blend with the dark overcast to create the illusion that the unlighted terrain is part of the sky (Part B of Figure 9-13). This illusion can be avoided by referencing the flight instruments and establishing a true horizon and attitude.

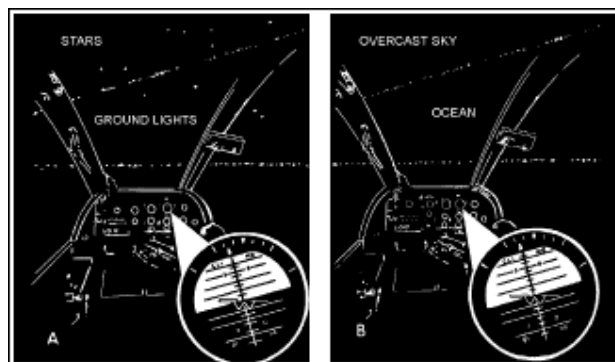
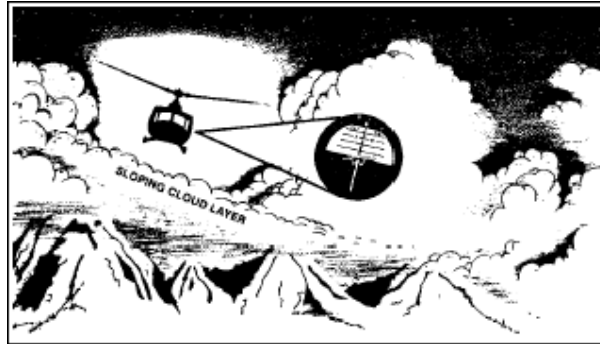


Figure 9-13. Confusion of Ground Lights and Stars at Night

## FALSE HORIZON ILLUSION

The false horizon illusion (Figure 9-14) occurs when the aviator confuses cloud formations with the horizon or the ground. This illusion occurs when an aviator subconsciously chooses the only reference point available for orientation. A sloping cloud deck may be difficult to perceive as anything but horizontal if it extends for any great distance in the pilot's peripheral vision. An aviator may perceive the cloudbank below to be horizontal although it may not be horizontal to the ground; thus, the pilot may fly the aircraft in a banked attitude. This condition is often insidious and goes undetected until the aviator recognizes it and makes the transition to the instruments and corrects it. This illusion can also occur if an aviator looks outside after having given prolonged attention to a task inside the cockpit. The confusion may result in the aviator placing the aircraft parallel to the cloudbank.



**Figure 9-14. False Horizon Illusion**

### **HEIGHT-DEPTH PERCEPTION ILLUSION**

2-27. The height-depth perception illusion is due to a lack of sufficient visual cues and causes an aircrew member to lose depth perception. Flying over an area devoid of visual references—such as desert, snow, or water—will deprive the aircrew member of his perception of height. The aviator, misjudging the aircraft's true altitude, may fly the aircraft dangerously low in reference to the ground or other obstacles above the ground. Flight in an area where visibility is restricted by fog, smoke, or haze can produce the same illusion.

### **CRATER ILLUSION**

9-28. The crater illusion occurs when aircrew members land at night, under NVG conditions, and the IR searchlight is directed too far under the nose of the aircraft. This will cause the illusion of landing with up-sloping terrain in all directions. This misperceived up-sloping terrain will give the aviator the perception of landing into a crater. This illusionary depression lulls the pilot into continuing to lower the collective. This can result in the aircraft prematurely impacting the ground, causing damage to both aircraft and crew. If observing another aircraft during hover taxi, the aviator may perceive that the crater actually appears to move with the aircraft being observed.

### **STRUCTURAL ILLUSIONS**

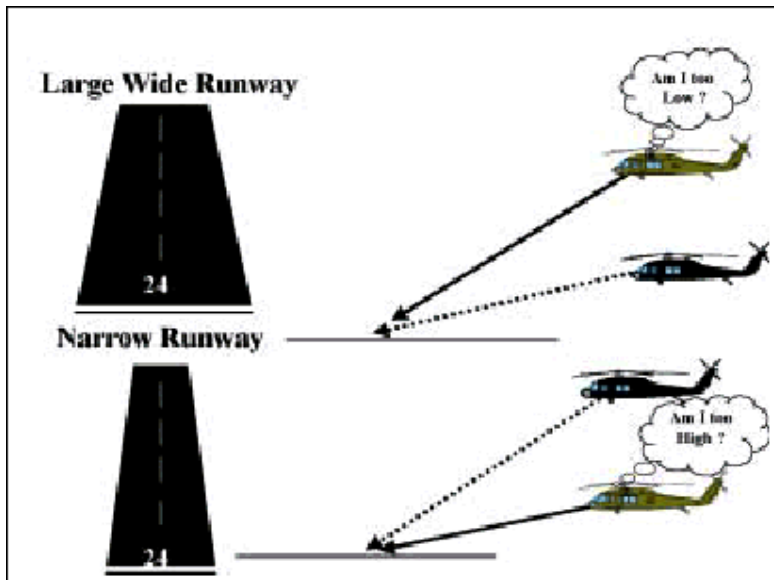
9-29. Structural illusions are caused by the effects of heat waves, rain, snow, sleet, or other visual obscurants. A straight line may appear curved when it is viewed through the heat waves of the desert. A single wing-tip light may appear as a double light or in a different location when it is viewed during a rain shower. The curvature of the aircraft windscreen can also cause structural illusions, as illustrated in Figure 9-15. This illusion is due to the refraction of light rays as they pass through the windscreen. When encountering environments that contain these visual obscurants, the aviator must remain aware that these obscurants may present a false perception.



**Figure 9-15. Structural Illusion**

#### **SIZE-DISTANCE ILLUSION**

9-30. The size-distance illusion (Figure 9-16) is the false perception of distance from an object or the ground, created when a crew member misinterprets an unfamiliar object's size to be the same as an object that he is accustomed to viewing. This illusion can occur if the visual cues, such as a runway or trees, are of a different size than expected. An aviator making an approach to a larger, wider runway may perceive that the aircraft is too low. Conversely, an aviator—making an approach to a smaller, narrower runway—may perceive that the aircraft is too high. A pilot making an approach 25 feet above the trees in the State of Washington, where the average tree is 100 feet tall, may fly the aircraft dangerously low if trying to make the same approach at Fort Rucker, Alabama, where the average tree height is 30 feet. This illusion may also occur when an individual is viewing the position lights of another aircraft at night. If the aircraft being observed suddenly flies into smoke or haze, the aircraft will appear to be farther away than before.



**Figure 9-16. Size-Distance Illusion**

### **FASCINATION (FIXATION) IN FLYING**

9-31. Fascination, or fixation, flying can be separated into two categories: task saturation and target fixation. Task saturation may occur during the accomplishment of simple tasks within the cockpit. Crew members may become so engrossed with a problem or task within the cockpit that they fail to properly scan outside the aircraft. Target fixation, commonly referred to as target hypnosis, occurs when an aircrew member ignores orientation cues and focuses his attention on his object or goal; for example, an attack pilot on a gunnery range becomes so intent on hitting the target that he forgets to fly the aircraft, resulting in the aircraft striking the ground, the target, or the shrapnel created by hitting the target.

### **REVERSIBLE PERSPECTIVE ILLUSION**

9-32. At night, an aircraft may appear to be moving away when it is actually approaching. If the pilot of each aircraft has the same assumption, and the rate of closure is significant, by the time each pilot realizes the misassumption, it may be too late to avoid a mishap. This illusion is termed reversible perspective and is often experienced when an aircrew member observes an aircraft flying a parallel course. In this situation, aircrew coordination is paramount. To determine the direction of flight, the aircrew member should observe the other aircraft's position lights. Remember the following: red on right returning; that is, if you see an aircraft with the red position light on the right and the green position light on the left, the observed aircraft is traveling in the opposite direction of your flight path.

### **ALTERED PLANES OF REFERENCE**

9-33. In altered planes of reference (Figure 9-17), the pilot has an inaccurate sense of altitude, attitude, or flight-path position in relation to an object so great in size that the object becomes the new plane of reference rather than the correct plane of reference, the horizon. A pilot approaching a line of mountains may feel the need to climb although the altitude of the aircraft is adequate. This is because the horizon, which helps the pilot maintain orientation, is subconsciously moved to the top of the ridgeline. Without an adequate horizon, the brain attempts to fix a new horizon. Conversely, an aircraft entering a valley that contains a slowly increasing up-slope

condition may become trapped because the slope may quickly increase and exceed the ability of the aircraft to climb above the hill, causing the aircraft to crash into the surrounding hills.



**Figure 9-17. Altered Planes of Reference**

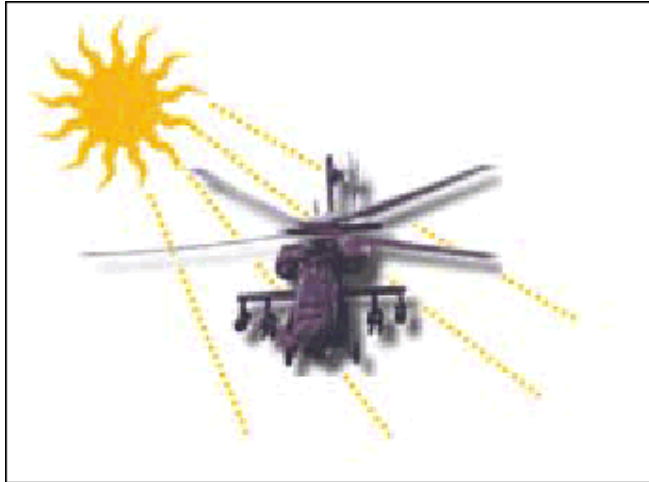
### **AUTOKINESIS**

9-34. Autokinesis primarily occurs at night when ambient visual cues are minimal and a small, dim light is seen against a dark background. After about 6 to 12 seconds of visually fixating on the light, one perceives movement at up to 20 degrees in any particular direction or in several directions in succession, although there is no actual displacement of the object. This illusion may allow an aviator to mistake the object fixated as another aircraft. In addition, a pilot flying at night may perceive a relatively stable lead aircraft to be moving erratically, when in fact, it is not. The unnecessary and undesirable control inputs that the pilot makes to compensate for the illusory movement of the aircraft represent increased work and wasted motion, at best, and an operational hazard at worst.

### **FLICKER VERTIGO**

9-35. Flicker vertigo (Figure 9-18) is technically not an illusion; however, as most people are aware from personal experience, viewing a flickering light can be both distracting and annoying. Flicker vertigo may be created by helicopter rotor blades or airplane propellers interrupting direct sunlight at a rate of 4 to 20 cycles per second. Flashing anticollision strobe lights, especially while the aircraft is in the clouds, can also produce this effect. One should also be aware that photic stimuli at certain frequencies could produce seizures in those rare individuals who are susceptible to flicker-induced epilepsy.





**Figure 9-18. Flicker Vertigo**

### **VESTIBULAR ILLUSIONS**

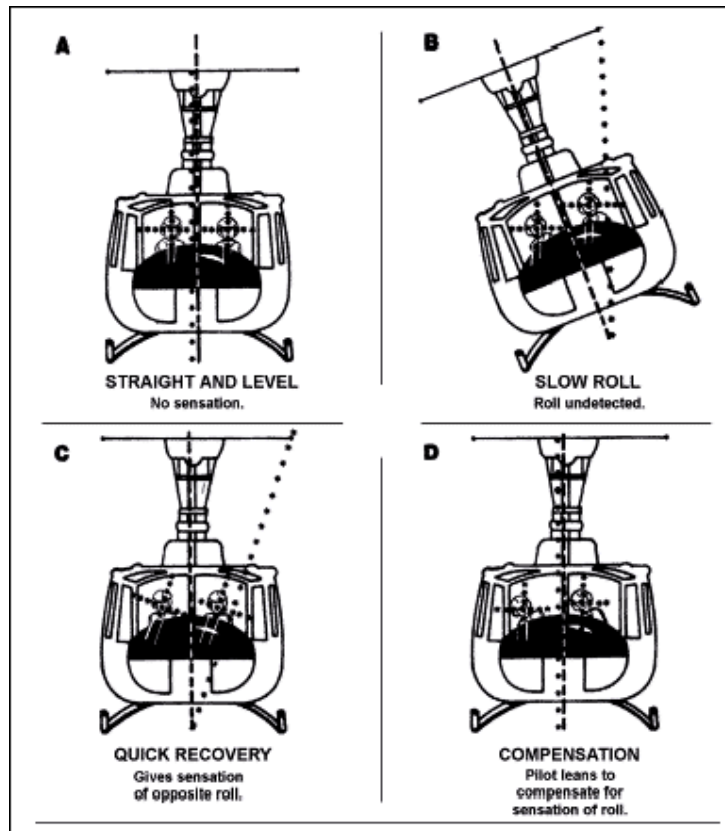
9-36. The vestibular system provides accurate information as long as an individual is on the ground. Once the individual is airborne, however, the system may function incorrectly and cause illusions. These illusions pose the greatest problem with spatial disorientation. Aircrew members must understand vestibular illusions and the conditions under which they occur. They must be able to distinguish between the inputs of the vestibular system that are accurate and those that cause illusion.

### **SOMATOGRAL ILLUSIONS**

9-37. Somatogyral illusions are caused when angular accelerations and decelerations stimulate the semicircular canals. Those that may be encountered in flight are the leans, graveyard spin, and Coriolis illusions.

#### **Leans**

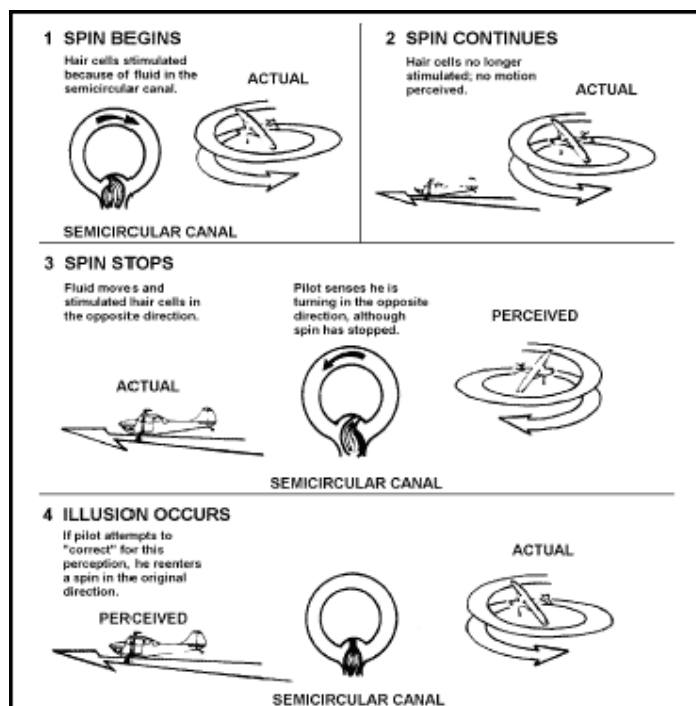
9-38. The most common form of spatial disorientation is the leans. This illusion occurs when the pilot fails to perceive angular motion. During continuous straight-and-level flight, the pilot will correctly perceive that he is straight and level (part A, Figure 9-19). However, a pilot rolling into or out of a bank may experience perceptions that disagree with the reading on the attitude indicator. In a slow roll, for instance, the pilot may fail to perceive that the aircraft is no longer vertical. He may feel that his aircraft is still flying straight and level although the attitude indicator shows that the aircraft is in a bank (part B, Figure 9-19). Once the pilot detects the slow roll, he makes a quick recovery. He rolls out of the bank and resumes straight-and-level flight. The pilot may now perceive that the aircraft is banking in the opposite direction. However, the attitude indicator shows the aircraft flying straight and level (part C, Figure 9-19). The pilot may then feel the need to turn the aircraft so that it aligns with the falsely perceived vertical position. Instead, the pilot should maintain straight-and-level flight as shown by the attitude indicator. To counter the falsely perceived vertical position, the pilot will lean his body in the original direction of the subthreshold roll until the false sensation leaves (part D, Figure 9-19).



**Figure 9-19. Leans**

### **Graveyard Spin**

9-39. This illusion, shown in Figure 9-20, usually occurs in fixed-wing aircraft. For example, a pilot enters a spin and remains in it for several seconds. The pilot's semicircular canals reach equilibrium; no motion is perceived. Upon recovering from the spin, the pilot undergoes deceleration, which is sensed by the semicircular canals. The pilot has a strong sensation of being in a spin in the opposite direction even if the flight instruments contradict that perception. If deprived of external visual references, the pilot may disregard the instrumentation and make control corrections against the falsely perceived spin. The aircraft will then reenter a spin in the original direction.



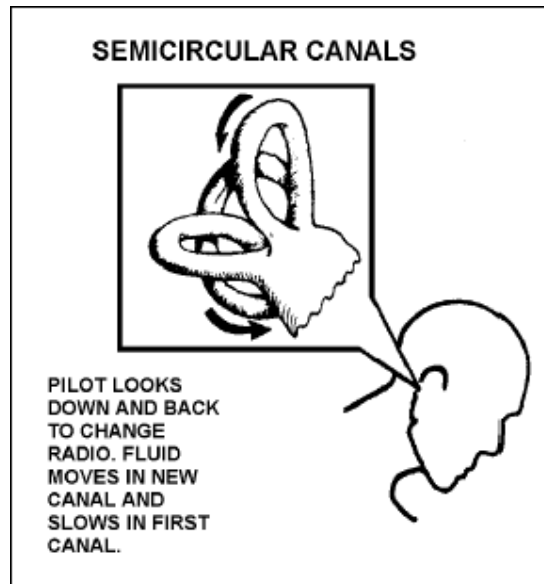
**Figure 9-20. Graveyard Spin**

9-40. To compound the action of the semicircular canals under these conditions, a pilot, noting a loss of altitude as the spin develops, may apply back pressure on the controls and add power in an attempt to gain altitude. This maneuver tightens the spin and may cause the pilot to lose control of the aircraft.

**Coriolis Illusion**

9-41. Regardless of the type of aircraft flown, the Coriolis illusion is the most dangerous of all vestibular illusions. It causes overwhelming disorientation.

9-42. This illusion occurs whenever a prolonged turn is initiated and the pilot makes a head motion in a different geometrical plane. When a pilot enters a turn and then remains in the turn, the semicircular canal corresponding to the yaw axis is equalized. The endolymph fluid no longer deviates, or bends, the cupula. Figure 9-21 shows the movement of the fluid in a semicircular canal when a pilot enters a turn.



**Figure 9-21. Movement of Fluid in the Semicircular Canals During a Turn**

9-43. If the pilot initiates a head movement in a geometrical plane other than that of the turn, the yaw axis semicircular canal is moved from the plane of rotation to a new plane of nonrotation. The fluid then slows in that canal, resulting in a sensation of a turn in the direction opposite that of the original turn.

9-44. Simultaneously, the two other canals are brought within a plane of rotation. The fluid stimulates the two other cupulas. The combined effect of the coupler deflection in all three canals creates the new perception of motion in three different planes of rotation: yaw, pitch, and roll. The pilot experiences an overwhelming head-over-heels tumbling sensation.

### **SOMATOGRAVIC ILLUSIONS**

9-45. Somatogavic illusions are caused by changes in linear accelerations and decelerations or gravity that stimulate the otolith organs. The three types of somatogavic illusions that can be encountered in flight are oculogavic, elevator, and oculoagavic.

#### **Oculogavic Illusion**

9-46. This type of illusion occurs when an aircraft accelerates and decelerates. Inertia from linear accelerations and decelerations cause the otolith organ to sense a nose-high or nose-low attitude. In a linear acceleration, the gelatinous layer, which contains the otolith organ, is shifted aft. The aviator falsely perceives that the aircraft is in a nose-high attitude. A pilot correcting for this illusion without cross-checking the instruments would most likely dive the aircraft. This illusion does not occur if adequate outside references are available. If making an instrument approach in inclement weather or in darkness, the pilot would be considerably more susceptible to the oculogavic illusion. An intuitive reaction to the sensed nose-high attitude could have catastrophic results

#### **Elevator Illusion**

9-47. This illusion occurs during upward acceleration. Because of the inertia encountered, the pilot's eyes will track downward as his body tries, through inputs

supplied by the inner ear, to maintain visual fixation on the environment or instrument panel. With the eyes downward, the pilot will sense that the nose of the aircraft is rising. This illusion is common for aviators flying aircraft that encounter updrafts.

### Oculoagravic Illusion

9-48. This illusion is the opposite of the elevator illusion and results from the downward movement of the aircraft. Because of the inertia encountered, the pilot's eyes will track upward. The pilot's senses then usually indicate that the aircraft is in a nose-low attitude. This illusion is commonly encountered as a helicopter enters autorotation. The pilot's usual intuitive response is to add aft cyclic, which decreases airspeed below the desired level.

### PROPRIOCEPTIVE ILLUSIONS

9-49. Proprioceptive illusions rarely occur alone. They are closely associated with the vestibular system and, to a lesser degree, with the visual system. The proprioceptive information input to the brain may also lead to a false perception of true vertical. During turns, banks, climbs, and descending maneuvers, proprioceptive information is fed into the central nervous system. A properly executed turn vectors gravity and centrifugal force through the vertical axis of the aircraft. Without visual reference, the body only senses being pressed firmly into the seat. Because this sensation is normally associated with climbs, the pilot may falsely interpret it as such. Recovering from turns lightens pressure on the seat and creates an illusion of descending. This false perception of descent may cause the pilot to pull back on the stick, which would reduce airspeed. [Figure 9-22](#) shows proprioceptive illusions.

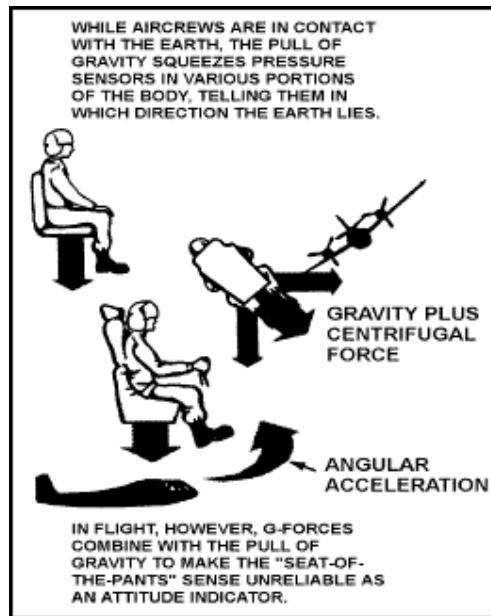


Figure 9-22. Proprioceptive Illusions

### PREVENTION OF SPATIAL DISORIENTATION

9-50. Spatial disorientation cannot be totally eliminated. However, aircrew members need to remember that misleading sensations from sensory systems are predictable. These sensations can happen to anyone because they are due to the normal functions and limitations of the senses. Training, instrument proficiency, good health, and aircraft design minimize spatial disorientation. Spatial disorientation becomes

dangerous when pilots become incapable of making their instruments read right. All pilots, regardless of experience level, can experience spatial disorientation. For that reason, they should be aware of the potential hazards, understand their significance, and learn to overcome them. To prevent disorientation, aviators should—

- Never fly without visual reference points (either the actual horizon or the artificial horizon provided by the instruments).
- Trust the instruments.
- Avoid fatigue, smoking, hypoglycemia, hypoxia, and anxiety, which all heighten illusions.
- Never try to fly VMC and IMC at the same time.

### **TREATMENT OF SPATIAL DISORIENTATION**

9-51. Spatial disorientation can easily occur in the aviation environment. If disorientation occurs, aviators should—

- Refer to the instruments and develop a good cross-check.
- Delay intuitive actions long enough to check both visual references and instruments.
- Transfer control to the other pilot if two pilots are in the aircraft. Rarely will both experience disorientation at the same time.

Note:

The following references were available for the specialized investigation group to assist in the studies.

- Surviving Spatial Disorientation
- Spatial Disorientation, From Wikipedia, the free encyclopedia.
- Spatial Disorientation -Why you shouldn't fly by the seat of your pants
- Spatial Disorientation Deaths of Visual Flight Rules Pilots: J. F. Kennedy, Jr., et. al.
- Spatial Disorientation Stories, From AVWEB Question Of The Week

1.16.5 Systems examination:

- 1.16.5.1 *Cause(s) for the autopilot disconnect*  
(Refer to 1.16.1. (Tests and Researches), Cairo March 04 Autopilot Flash 737 March Progress Meeting Flash 737 March Progress, Autopilot Engagement)
- 1.16.5.2 *Cause(s) for "Heading Select" disengage when the autopilot is engaged* (applied also to the accident aircraft)  
(Refer to 1.16.1. (Tests and Researches), Boeing response to the raised questions, enclosure to B-H200-17833-ASI Question B4)
- 1.16.5.3 *Availability of autopilot during the captain's requests "autopilot, autopilot" (accident aircraft)*  
(Refer to 1.16.1. (Tests and Researches), Cairo March 04 Autopilot Flash 737 March Progress Meeting Flash 737 March Progress, Estimated Autopilot Availability, Boeing response to the raised questions, enclosure to B-H200-17833-ASI Question B6)
- 1.16.5.4 MMEL issues associated with operating the airplane with FD TO/GA mode inoperative (won't stay engaged)  
Relevant information to be added upon Human Factors Group discretion
- 1.16.5.5 *Interlock logic for A/P with the definition of the likelihood (ruled out, not likely, unknown) to the various interlocks regarding the role they may have played in the autopilot disengagement*  
(Refer to 1.16.1. (Tests and Researches), Honeywell SP-300 DFCS B737-300.ppt file, and Flash Airlines Presentation SP-300 DFCS Health Monitoring Honeywell.ppt file)
- 1.16.5.6 *The effects of the TOGA bit dropping out and way it affects the command bars.*  
(Refer to 1.16.1. (Tests and Researches), Boeing AMM 22-03-00, 22-04-00)
- 1.16.5.7 *Examination of the selected course compared to the selected heading (probability for having "dropouts").*
- 1.16.6 *CVR examination:*
  - 1.16.6.1 *Examination of the CVR recording for indications of A/P and heading select switch noises*  
(Could not be identified)
  - 1.16.6.2 *Examination of CVR at 2.58.15 (when the MSR crew says that they heard a message from Flash on 121.5).*  
121.5 recording has been checked, no such message was recorded
- 1.16.7 *FDR examination:*
  - 1.16.7.1 *Spatial disorientation study of the accident flight based on the recorded FDR data*



TBC (CBS group)

- 1.16.8 *PCU inspection and teardown (EQA report):*  
(Refer to 1.16.1.7. Aileron system)

## **1.17. Organizational and Management Information**

### **1.17.1. Flash Airlines**

#### **1.17.1.1. Flash Airlines Air Operator Certificate (AOC)**

Flash Airlines was approved as air operator (charter air carrier) under ECAR 121 by the ECAA, and operating under approval no 018.

Flash Airlines has its main office in Cairo, Egypt at 166b El Hegaz St. Heliopolis. Beginning in 2000, Flash Airlines leased the first B737-300 from the International Lease Financial Corporation (ILFC). In June 2001 another B737-300 from ILFC was added to Flash Airlines fleet, which made the company fleet two aircraft the same type. The Operations Specifications was issued to the company in Feb 2000 and the last revision was on October 29<sup>th</sup> 2003.



ARAB REPUBLIC OF EGYPT  
MINISTRY OF CIVIL AVIATION

## AIR OPERATOR CERTIFICATE

*This certifies that*

***FLASH AIRLINES***

*Has met the requirements of the MINISTRY OF CIVIL AVIATION and related operating regulations and rules prescribed thereunder for the issuance of this certificate and is hereby authorized to conduct Air-Carrier operation in accordance with said operating regulations and rules prescribed thereunder and the terms, conditions and limitations contained in the attached Operation Specifications.*

*This certificate is not transferable and, unless sooner surrendered, suspended or revoked, shall continue in effect until February 23, 2004 or terminated.*

Pilot / Saleh Moussa

SALEH.A.MOUSSA

Head of Operations & Air Transport Sector

20-2-2003



CERTIFICATE N O. : 18

CERTIFICATE ISSUE DATE : February 24, 2000

#### 1.17.1.2. History

Flash Airlines is also approved under ECAR 145 as a repair station. The approval number is CAI/FLASH?AS/1/2001. The certificate is valid until July 30<sup>th</sup>, 2004 and was issued on July 31, 2001. The certificate is limited to line maintenance up to the 8A check for the B737-300. Flash Airlines maintenance base is Cairo international Airport.

Flash Airline Organization Chart:

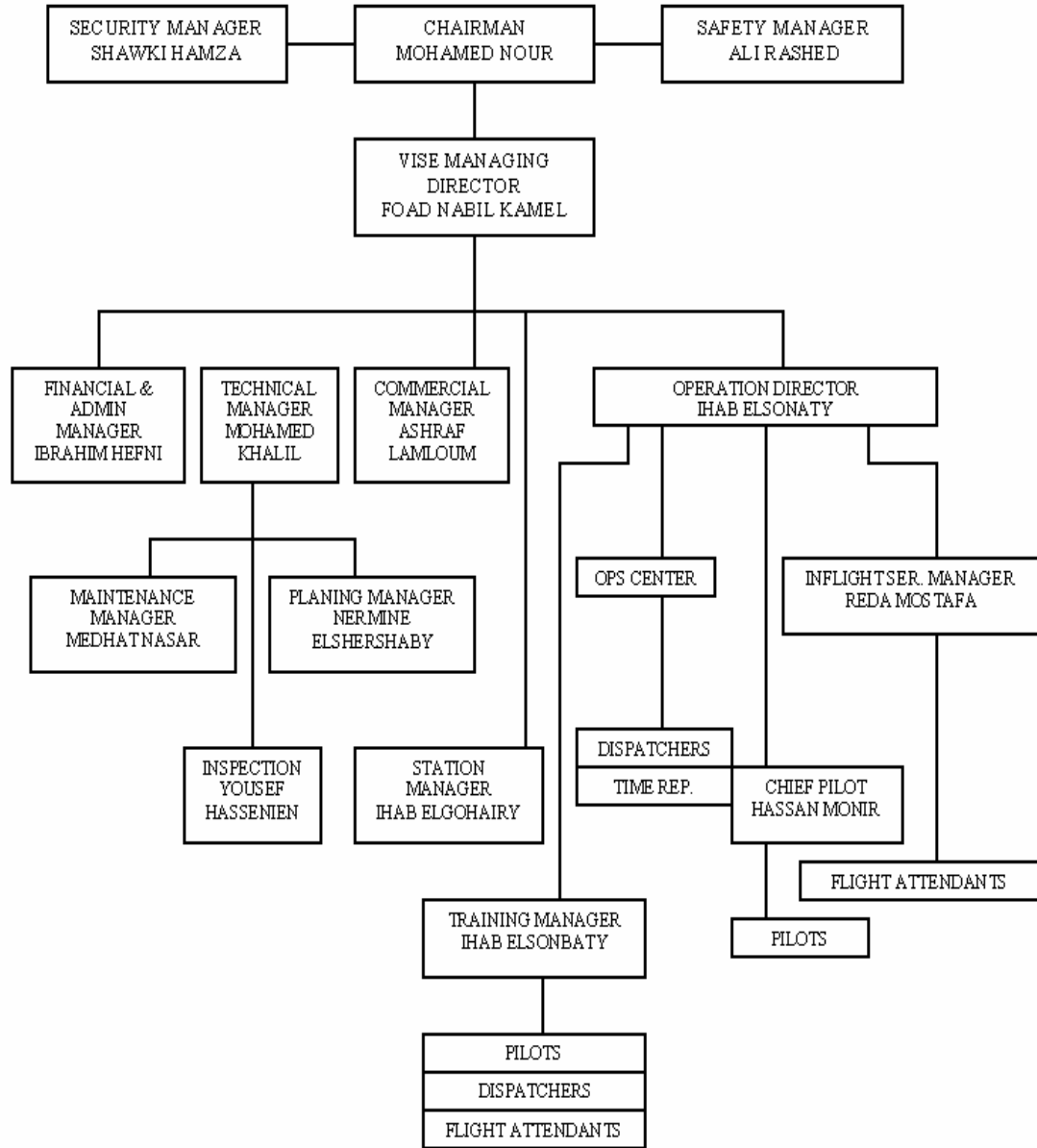


Figure 1.17.1-1 Flash Airlines Organization Chart

Flash Airlines coordinates the maintenance program through its ECAR Part 145 certificate. The Company General Maintenance Manual (GMM) provides guidance related to the Aircraft Maintenance program as the Maintenance Procedures, Maintenance staff Training... etc.

Personnel working on Flash Airlines Fleet at the various maintenance facilities must be familiar with the policies and procedures spelled out in the company GMM. The Quality Control Manager puts the newly hired employees through a twelve-hour Indoctrination Course. The Indoctrination course includes Flash Airlines policy/ procedures, and training practices. It is accomplished before maintenance engineer begins to work at the Flash Airlines facility. The training is documented on a maintenance training attendance record, recorded on the employee's training file.

1.17.1.3. Personnels Training and Authorization  
1.17.1.3.1. Maintenance Engineers

According to ECAR 65 the requirements for granting authorization for ground engineer are as follow:

- 1- Graduation from Faculty of Engineering or an approved training institute.
- 2- Passing the approved Basic training Course at approved Training Center or institute.
- 3- On Job Training for 18 months.
- 4- Passing written, practical and oral exams by the authority for License without Type Rating (LWTR).
- 5- Passing an approved training course for a specific type airframe and engine.
- 6- On Job Training (OJT) on the type airframe and engine for 9 months.
- 7- Attendance of training course for the company exposition procedure manual.
- 8- Passing oral and practical examination in front of the Company Examination Board (approved by the authority)
- 9- Getting the company approval.

Flash Airlines maintains its training program in compliance with Egyptian Civil Aviation Regulation requirements. The Maintenance Director and the Quality Control Manager have joint responsibility for assuring all required training is performed and recorded.

Indoctrination training proceeds an employee's start date. The employee is given a 4-hour introduction course that trains one on Flash Airlines maintenance policies and procedures. The training will be documented on a maintenance training attendance record and maintained in the employee's training file.

The aircraft systems training for the A & C Engineers is accomplished through formal systems training and On-the-Job Training (OJT) Worksheets.

Engineer Mustafa Erfan carried out the last pre- flight release.

1.17.1.3.2. Cockpit Crews

Refer to Exhibit F Operation Group Factual Report, Attachment 1

### 1.17.2. Review of oversight by ECAA on 2003

1.17.2.1 Safety oversight carried out on Flash Airline during the period from 2 Jan, 2003 to 16 Jan 2003 before AOC renewal

The oversight findings and the relevant actions taken by the airline are shown in the table below

#### A- Operation Findings

	<b>Findings</b>	<b>Actions Taken</b>
1	There is no Training Program	Training Program is submitted and approved
2	There is no Internal Evaluation Program (IEP)	IEP is submitted and approved
3	There is no Line check Training for Captains	Line Check Training is performed
4	No ECAR Training Course was performed recently	Training course has started and it will take some time to cover all the operation personnel
5	There is no approved Training Class	Training Class is Approved.
6	There are no DRM & CRM Training course performed for cockpit crews ,dispatchers and cabin crews	The Airline has introduced a training plan starting on Sep 2003 to be done in PAS Airline
7	No of cockpit crews are not fulfilling the minimum requirement of ECAA	The cockpit crews are sufficient for required operation and the airline will recruit more cockpit crews to fulfill the future operation requirements
8	By reviewing the A/C log book sheets found that ,some sheets not filled out and other some have missed data	The airline issued circular for all cockpit crews and maintenance staff to strictly comply with log book sheets filling out instructions
9	By reviewing the airline TM,GOM and Dispatch Manual some findings were discovered	All findings are covered
10	The submitted station manual not fulfilling ECAA requirements	The Station Manual was updated to fulfill the ECAA requirements
11	The Safety Manual which was submitted by the airline does not meet ECAA requirements	New manual revision is in progress
12	Cabin Crew does not use safety and emergency check lists	A circular was issued for the cabin crew to strictly comply with the written instruction for using the check lists
13	There is no security program for Aircraft	The program is submitted and approved
14	Load sheet calculations for some flights not accurate	Load sheet calculations training course is planned to be done for all flight dispatchers

B-Airworthiness Findings

	<b>Findings</b>	<b>Actions Taken</b>
1	There is shortage of some maintenance equipment and tools	The unavailable equipment and tools will be loaned from EgyptAir when required
2	Personnel files are not updated	Files are updated
3	GMM is not Updated	GMM is updated
4	There is no AMM in the library	AMM is Available now in the library
5	MPD, AFM, CMEL, and FOM are not Updated	All manuals are updated
6	There is no Training Program for Recurrent Course	The recurrent training program was submitted and approved
7	Authorization Board does not include electric engineer	The electric engineer authorization will be issued by ECAA
8	The airline has not submitted SOC 121	SOC 121 was submitted and Accepted
9	Some parts are not calibrated	The parts required to be calibrated were sent to EgyptAir for calibration
10	Safety wire of fire bottles do not meet the standards	Safety wire corrected to meet the standards
11	Spare parts in the store are not sufficient	The required spare parts will be loaned from EgyptAir when required
12	A/C tires storage is not according to the storage requirement	Storage requirement familiarization course is performed for the storage keepers
13	The storage keepers are not familiar With GMM	GMM training course is planned to be performed
14	There is no safety requirement program	The program is submitted and approved
15	By reviewing the TLB Sheets ,found that , some sheets not including PDC Maintenance Release and ECM data	An inspection Circular is issued for the maintenance personnel sign PDC Release after PDC performing



1.17.2.2 Safety oversight carried out on Flash Airline on 16 Jul 2003 before AMO  
Certificate renewal

The oversight findings and the relevant actions taken by the airline are shown in the table below

	<b>Findings</b>	<b>Action Taken</b>
1	There is no W&B Program	The program is submitted and approved
2	Human factors training program for the engineers not yet submitted to ECAA for approval	Human factors training program for engineers is submitted to ECAA and approved

1.17.3. Relevant Flash Airlines procedures:

1.17.3.1 Flash Airlines procedures regarding use of autopilot when recovering from unusual attitudes  
Refer to Flash Airline FOM (Ops Group)

1.17.3.2 Flash Airlines procedures regarding Upset Recovery training  
  
MCA requirements regarding Upset Recovery are not mandatory.  
Refer to Flash Airline FOM (Ops Group)

1.17.3.3 Flash Airlines procedures regarding "training about PNF assuming control when the PF is not responding to situations, callouts"

## CREW HEALTH PRECAUTIONS

#### 4. CREW HEALTH PRECAUTIONS

A crew member's sickness/illness, his feeling unwell/indisposed or the impairment of his senses and reflexes by narcotics, drugs or pharmaceutical preparations/medicaments have quite often contributed to incidents and accidents. Therefore, crew health is of the highest importance and has a direct impact upon flight safety. This is reflected in very stringent requirements for regular medical examinations and medical certificates. It hardly needs to be mentioned that living health - consciously is in the self-interest of every crew member.

**Note:** For incapacitation of crew members crew member shall not perform duties on an aeroplane if he is in any doubt of being able to accomplish his assigned duties, or if he knows or suspects that he is suffering from fatigue, or feels unfit to the extent that the flight may be endangered.

##### 4.1 Incapacitation of Crew Members

###### 4.1.1 Definition

Incapacitation of a crew member is defined as any condition which affects the health of a crew member during the performance of duties - associated with the duty/position assigned to him - which renders him incapable of performing the assigned duties.

The definition includes either total or partial incapacitation which does not allow the fulfilment of duties in the "normal" way.

###### 4.1.2 General

In-flight pilot incapacitation is a valid safety hazard and has already caused many accidents. Incapacities have occurred more frequently than other emergencies which are the subject of extensive training (such as engine failure, cabin fire etc). Aviation history and statistics indicate that incapacities may occur in all age groups and during all phases of the flight. There are many forms of incapacitation ranging from obvious sudden death to a lingering and difficult to detect partial loss of functions.

###### 4.1.3 Types of Incapacitation

Obvious incapacitation: means total functional failure and loss of capabilities. This generally will be easily detectable and will be a prolonged condition. Among the possible causes are heart disorders, severe brain disorders, severe internal bleeding, etc.

Subtle incapacitation: this may be considered a more significant operational hazard, because it is difficult to detect and the effects can range from partial loss of functions to a complete unconsciousness. Possible causes might be minor brain seizures, hypoglycemia (low blood sugar), other various medical

disorders or preoccupation with personal problems. Since the crew member concerned may not be aware of, or capable of rationally evaluating his situation, this type of incapacitation is more dangerous!

#### 4.1.4 Causes and Effects

As explained before, incapacitation may range from minor cases of physiological upsets associated with intercurrent mild disease or mental stress which may result in reduced levels of judgement or physical coordination up to a complete collapse.

Among the causes for a mild incapacitation one may list: Body pains such as toothache, headache, gastroenteritis, the delayed effects of alcohol, drugs or medication, common disorders such as a cold, etc. Heart troubles, an acute infection thrombosis, epilepsy, hypoglycemia (extremely low level sugar) and others belong to the more serious causes of a sudden collapse. At least one incident is known, where a crew member had a heart attack right after his aviation medical examination, so a passed medical exam is not a guarantee!

It is obvious that living more health consciously may reduce the number of occurrences of

also the avoidance of stress in your business and private life. Chapter 4.1 covers the subject of health precautions.

#### 4.1.5 Recognition of an Incapacity

An early recognition of a incapacity is of outmost importance. A silent collapse will hardly be detected during normal activities (for instance during the cruise phase of a flight), as communications may sometimes be reduced to a minimum. This requires that all crew members monitor each other very closely.

"Closely" means, observing the other crew members for any "abnormal" reaction/action or behavior. One good method is to use the so called "TWO COMMUNICATION RULE". This simply means, that one crew member's comment must be answered by the other crew member(s).

If - for instance - the PNF reports the aeroplane being left of course, it is essential, that the PF not only corrects this problem but also confirms this verbally. If a crew member doesn't answer any question or checklist item in the normal way, there is reason to believe that there might be the beginning of a subtle incapacitation.

crew member incapacitation.  
This includes avoidance of  
drugs, moderate consumption of  
alcohol, adequate rest time -and  
its proper use for recreation -  
adequate sleep and nutrition but

here is an illustration of the use  
of the Two Communication Rule:

1. the PNF, for example,  
notices the airplane is left of  
course,

## CREW HEALTH PRECAUTIONS

2. the PNF notifies the PF of the abnormal condition (the first communication), but
3. the PF does not respond in any manner (verbally or by correcting the flight path),
4. the PNF repeats the abnormal condition to the PF (the second communication),
5. the PF again fails to respond,
6. after the PF fails to respond to the second communication, the PNF should assume the PF is incapacitated and should take action as described in Section 4.1.6

At the worst he may simply have fallen asleep.

Other symptoms of the beginning of an incapacitation are:

- incoherent speech;
- strange behaviour;
- irregular breathing;
- pale fixed facial expression;
- jerky motions that are either delayed or too rapid.

If any of these are present, incapacitation must be suspected and action taken to check the state of the crew member.

#### 4.1.6 Actions to be taken when an incapacity is recognised.

##### First Step

- take over control of the aeroplane by announcing "I have control",
- engage autopilot,

- declare an urgency or emergency -whichever is applicable -,
- have an incapacitated cockpit crew member removed from his seat. In any case his seat should be moved fully back to prevent obstruction of flight controls, switches, levers, etc. The help of other crew members or passengers might be required,
- if necessary, reset COM and NAV to your side

##### Second Step

- take care of the incapacitated crew member by trying to provide first aid (ask if doctors or other medical persons are aboard),
- arrange a landing as soon as practicable after considering all pertinent factors,
- arrange medical assistance after landing
- giving as many details about the condition of the affected crew member as possible.

##### Third Step

- prepare for landing (cockpit and cabin), but do not press for a hasty approach
- perform approach checklist earlier than normal (request assistance from other crew members or "capable" persons),
- request radar vectoring and make an extended approach where possible - to reduce workload,
- for landing do not change seats - fly the aeroplane from

that position you initially were assigned to.

- organise work after landing; this shall include
  - depending on the situation, a change of seats for taxiing in, but only after the aeroplane has come to a complete stop;
  - having the incapacitated crew member offloaded and to the ambulance as quickly as possible;
  - arrangements for the parking of the aeroplane.

**NOTE:**

1. The company operations department must be kept informed at all times regarding the above circumstances for immediate relay to the Manager Flight Operations.
2. In case of incapacitation of the system panel operator, pilots shall refer to procedures as published in the AOM.

**4.1.7 Summary**

The problems involved with incapacitation of crew members may be summarised as follows:

- 1) If you do not feel well, say "NO" before the flight.
- 2) Remember, that the best medical examination as well as a health conscious life still do not guarantee that an incapacitation during flight will not happen to you or to your other crew members.

- 3) The "TWO COMMUNICATION RULE" must be used in order to have a chance of detecting any incapacitation in time. Take notice of any abnormal or unusual action of another crew member, as this might also be an indication of onset of incapacitation.

4. Once an incapacitation is identified, remember the three basic steps:

Step 1) Take over the aeroplane and bring it under YOUR control.

Step 2) Take care of the incapacitated pilot (either have him removed from his seat or fixed so that he will not interfere the controls).

Step 3) Prepare for landing.

Finally, it is emphasised that incapacitation requires special actions using the good judgement of the crew member left in command of the aeroplane.

**4.2 ALCOHOLIC BEVERAGES**

The use of intoxicating beverages by FLASH AIR flight crew members must of necessity be strictly regulated.

The following rules must be strictly observed by all flight crew members at all times:

1. No alcoholic beverage shall be consumed on the same calendar day that a crew

- 1.17.3.4 *Flash Airlines training/operational information regarding intervention by the non-flying pilot when the flying pilot fails to respond to calls for correcting an unsafe situation.*  
Refer to previous item
- 1.17.3.5 Regularity (or irregularity) rules regarding sleeping schedules on and off-duty. Strategies for obtaining adequate rest and managing crew on-duty alertness  
Refer to Flash Airline FOM (Ops Group)
- 1.17.3.6 General description about Flash Airline.  
(Date of foundation or transition, location of offices and bases, number of aircrafts operated, number of pilots and other personnel, annual flights, passengers carried, revenues, routes flown, and financial health)  
(All relevant information are already included in the Factual Report)
- 1.17.3.7 Labor management issues, growth trends, and main competitors.  
Closed
- 1.17.3.8 Egyptian requirements for the training of pilots at an airline such as Flash Airlines.

**GENERAL.** The following outline is intended to clarify the six categories of training used by operators and defined in Part 121, Subpart N. This clarification is intended to both define the type of training and describe for the Operator when each category of training is applicable.

**APPLICABILITY OF TRAINING CATEGORIES.** Usually, operators will need to conduct training in all six categories of training. Recurrent training applies to all operators. Initial equipment training, transition training, upgrade training, and requalification training apply in most situations. However, transition training is not applicable for an operator who operates only one aircraft type. Initial new hire training applies to operators who train and qualify newly hired personnel or personnel who have not been previously qualified as a crewmember by that operator.

**CATEGORIES OF TRAINING.** There are six basic categories of training applicable to Part 121 operators. The primary factors which determine the appropriate category of training are the student's previous experience with the operator and previous duty position. Each category of training consists of one or more curriculums, each one of which is specific to an aircraft type and a duty position (for example: A-320 SIC, and A-320 PIC). Training should be identified with and organized according to specific categories of training. When discussing training requirements, MoCA inspectors should be specific regarding the category of training being discussed and use the same references as are stated in Part 121 Subpart N. Inspectors should encourage operators to use this nomenclature when developing new training curriculums or revising existing training curriculums. Use of this common nomenclature improves standardization and mutual understanding. The six categories of training are briefly discussed in the following subparagraphs:

**A. Initial New Hire Training.** This training category is for personnel who have not had previous experience with the operator (newly hired personnel). It also applies, however, to personnel employed by the operator who have not previously held a cockpit crewmember duty position with that operator. Initial new hire training includes basic indoctrination training and training for a specific duty position and aircraft type. Except for a basic indoctrination curriculum segment, the regulatory requirements for "initial new hire" and "initial equipment" training are the same. Since initial new hire training is usually the employee's first exposure to specific company methods, systems, and procedures, it must be the most comprehensive of the six categories of training. For this reason, initial new hire training is a distinct separate category of training and should not be confused with initial equipment training. Initial equipment training is a separate category of training.

**B. Initial Equipment Training (PIC and SIC).** This category of training is for personnel who have been previously trained and qualified for a duty position by the operator (not new hires) and who are being reassigned for any of the following reasons:

(a) Reassignment is to any duty position on an airplane of a different group (Group IIIP is reciprocating and turbopropeller powered and Group IIJJ is turbojet powered).

(b) Reassignment is to a different duty position on a different airplane type when the cockpit crewmember has not been previously trained and qualified by the operator for that duty position and airplane type.

**C. Transition Training.** This category of training is for an employee who has been previously trained and qualified for a specific duty position by the operator and who is being assigned to the same duty position on a different aircraft type and the different type aircraft must be in the same group. If it is not in the same group, initial equipment training is the applicable category of training.

**D. Upgrade Training.** This category of training is for an employee who has been previously trained and qualified as SIC or PIC (not eligible for requalification training) by the operator and is being assigned as PIC to the same aircraft type for which the employee was previously trained and qualified as SIC or PIC on the same type.

**E. Recurrent Training.** This category of training is for an employee who has been trained and qualified by the operator, who will continue to serve in the same duty position and aircraft type, and who must receive recurring training and/or checking within an appropriate eligibility period to maintain currency.

**F. Requalification Training.** This category of training is for an employee who has been trained and qualified by the operator, but has become unqualified to serve in a particular duty position and/or aircraft due to not having received recurrent training and/or a required flight or competency check within the appropriate eligibility period. Requalification training is also applicable in the following situations:

\* PICs who are being reassigned as SICs on the same aircraft type when seat dependent training is required

\* PICs and SICs who are being reassigned as FEs on the same aircraft type, provided they were previously qualified as FEs on that aircraft type

**G. Summary of Categories of Training.** The categories of training are summarized in general terms as follows:

(a) All personnel not previously employed by the operator must complete initial new hire training.





(b) All personnel must complete recurrent training for the duty position and aircraft type for which they are currently assigned within the appropriate eligibility period.

(c) All personnel who have become unqualified for a duty position on an aircraft type with the operator must complete requalification training to reestablish qualification for that duty position and aircraft type.

(d) All personnel who are being assigned by the operator to a different duty position and/or aircraft type must complete either initial equipment, transition, upgrade, or requalification training depending on the aircraft type and duty position for which they were previously qualified.

**Experience Hours Pre-Requisites for Different Training**

ECAR Part 121.400 Groups of aircraft	Requirements For	Upgrade	Initial New Equipment		Initial New Hire	
			SIC	PIC	SIC	PIC
(A) 121 - Air Taxi. Not exceed 5700 kg's						
<b>Group (I):</b> Single Engine Airplane	1.Total Flight Experience. 2.Flight Experience on Aeroplane Group. 3.Flight Experience on Aeroplane Type.	2150 300 100	500 300	2150 300	200	2150 300
<b>Group (II):</b> Multi -Engines Airplane	1.Total Flight Experience. 2.Flight Experience on Aeroplane Group. 3.Flight Experience on Aeroplane Type.	2500 500 150	500 300	2500 500	200	2500 500
(B) 121 - Air Carriers & Air Taxi						
<b>Group (IIP) &gt;5700 kg</b>						
Reciprocating power	1.Total Flight Experience. 2. Flight Experience on Aeroplane Group. 3.Flight Experience on Aeroplane Type.	3000 750 300	500 300	3000 750	200	3000 750
Turbopropeller powered	1.Total Flight Experience. 2.Flight Experience on Aeroplane Group. 3.Flight Experience on Aeroplane Type.	3500 1500 500	700 500	3500 1500	200	3000 1500
<b>Group (IIIJ) &gt;5700 kg</b>						
Turbo- Jet Powered	1.Total Flight Experience. 2.Flight Experience on Aeroplane Group. 3.Flight Experience on Aeroplane Type.	4000 2500 300	1200 1000	4000 2500	300	4000 2500
(C) 121 - Air Carriers & Air Taxi Helicopter	1.Total Flight Experience. 2. Flight Experience on Aircraft Category. 3.Flight Experience on Aircraft Type.	1000 300 120	450 300	1000 300	150	1000 300

**Two Pilots Flight Training Minimum Hours Required**

<i>ECAR Part 121.400 Groups of aircraft</i>	Upgrade SIC to PIC	Transition		<i>Initial New Equipment</i>		<i>Initial New Hire</i>	
		<i>SIC</i>	<i>PIC</i>	<i>SIC</i>	<i>PIC</i>	<i>SIC</i>	<i>PIC</i>
<i>(A) 121 - Air Taxi. Not exceed 5700 kg's</i>							
<b>Group (I): Single Engine</b>	2	4	4	4	4	8	8
<b>Group (I) &amp; (II): VFR only</b>	4	4	4	4	4	4	4
<b>Group (II) &amp; (II): IFR/VFR</b>	4	8	8	12	12	16	16
<i>(B) 121 - Air Carriers &amp; Air Taxi</i>							
<b>Group (III): Exceeds 5700 kg</b>							
• <b>Reciprocating power</b>	12	20	20	20	20	24	24
• <b>Turbopropeller powered</b>	12	20	20	20	20	24	24
<b>Group (III): Turbo- Jet Powered</b>	12	24	24	24	24	28	28
<i>(C) 121 - Air Carriers &amp; Air Taxi Helicopter</i>							
• <b>VFR only</b>	4	4	4	4	4	4	4
• <b>IFR/VFR</b>	4	8	8	12	12	16	16

**One Pilot Flight Training Minimum Hours Required**

<i>ECAR Part 121.400 Groups of aircraft</i>	Upgrade SIC to PIC	Transition		<i>Initial New Equipment</i>		<i>Initial New Hire</i>	
		<i>SIC</i>	<i>PIC</i>	<i>SIC</i>	<i>PIC</i>	<i>SIC</i>	<i>PIC</i>
<i>(A) 121 - Air Taxi. Not exceed 5700 kg's</i>							
<b>Group (I): Single Engine</b>	4	4	4	4	4	6	6
<b>Group (I) &amp; (II): VFR only</b>	2	3	3	3	3	4	4
<b>Group (II) &amp; (II): IFR/VFR</b>	4	6	6	6	6	8	8
<i>(B) 121 - Air Carriers &amp; Air Taxi</i>							
<b>Group (III) : Exceeds 5700 kg</b>							
• Reciprocating power	6	12	12	14	14	14	14
• Turbopropeller powered	6	12	12	15	15	15	15
<b>Group (IIIJ) : Turbo- Jet Powered</b>	6	12	12	16	20	16	20
<i>(C) 121 - Air Carriers &amp; Air Taxi Helicopter</i>							
• VFR only	2	3	3	3	3	4	4
• IFR/VFR	4	6	6	8	8	10	10

See also Pilots training documents included in items 1.5.1 and 1.5.2

- 1.17.3.9 The training that was actually provided to all Flash Airlines pilots  
Pilots training documents are included (refer to 1.5.1 and 1.5.2)
- 1.17.3.10 Flash Airlines procedures regarding pilots training and checking on operation of the auto flight system. .  
No specific form is available (refer to 1.5.1 and 1.5.2)
- 1.17.3.11 Flash Airlines program for training and checking pilots in the field of CRM and human factors (as contained in the company training manual)  
No mandatory training was required by ECAR at the time of the accident. However, CRM course is outlined in Flash Airline Training Manual 4.10
- 1.17.3.12 Flash Airlines pilots procedures for training and checking pilots on spatial disorientation countermeasures and upset recovery  
Spatial Disorientation training is not a requirement by Civil Aviation Authorities. However, some literature about this subject is included in Flash Airline Training Manual.
- 1.17.3.13 Flash Airlines policies regarding use of CRM.  
Refer to 1.17.3.11.
- 1.17.3.14 Flash Airlines policies relating to assertiveness and company guidelines as to when a first officer should take control of an aircraft from a captain.  
Refer to 1.17.3.3.
- 1.17.3.15 *Flash Air general company policies related to crew communication, assertiveness, and other CRM-related behaviors*  
Refer to 1.17.3.3.
- 1.17.3.16 Flash Airlines policies regarding use of the auto flight system  
(To be referred to the OPS group)
- 1.17.3.17 Regulations governing operators (like Flash Airlines) regarding Oversight audits by ECAA.  
ECAA regulations require every operator to undergo an oversight audit once every 12 month
- 1.17.3.18 Details about the ECAA oversight audit on Flash Airlines  
Is already included in the Factual Report
- 1.17.3.19 Outcomes of Oversight audits (previous violations, fines, or bans levied by ECAA)  
Is already included in the Factual Report

1.17.3.20 Previous violations, fines, or bans levied foreign aviation regulatory agencies.  
None identified

Reviewing this report indicated that the ban was due to a conflict on financial issues and no relevant safety issues were mentioned.

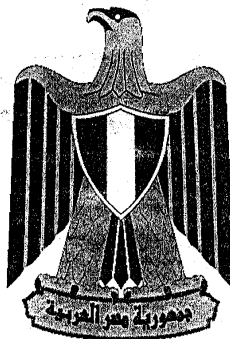
1.17.3.21 Selected additional information regarding Flash Airlines Organization including:

- Organization and responsibilities Chapter 1 FSH 1.5.1/ 1.5.2
- Organization and responsibilities Chapter 1 FSH 1.8.7
- Qualification requirements Chapter 3 FSH 3.3.1/ 3.3.2
- Crew Health Precautions Chapter 4 FSH-4.1.1- 4.1.4
- Operating Procedures Chapter 6 FSH 6.3.44/ 6.3.45/ 6.3.46
- Training details Flash Training Manual Chapt 05 Page 7

All pertinent information are included in the Factual Report

1.17.3.22 Airline Simulator program contract with RAM, ECAA letter of approval

**ARAB REPUBLIC OF EGYPT  
MINISTRY OF CIVIL AVIATION  
Egyptian Civil Aviation Safety & Security Authority**



***Full Flight Simulator Approval Certificate***

***Aircraft Type B737 - 500***

***Issued to: EgyptAir***

***AIR OPERATOR CERTIFICATE***

***AOC Number: MSR-AC 010 (B737/500SIM-2DG)***

***Simulator Operator: Air Maroc – Casablanca***

**CAIRO  
September, 2003**



**Our Ref. MSR - AC010 - B737-500 FLT SIM-2/D**  
**Date: 24, September 2003**

The General Manager Flight Training (GMFT)  
Flight Operations, EgyptAir,  
Cairo International Airport,  
Cairo, Egypt.

**To: GMFT, EgyptAir**

***APPROVAL TO USE THE FLIGHT SIMULATOR SPECIFIED IN  
THE ENCLOSED DOCUMENTATION***

Please find enclosed the required Approval Certificate and Licensing  
Considerations for the use of the requested Flight Simulator.

*Yours sincerely,*

**Issued at: Cairo, Egypt**  
**Date: 24, September 2003**

**Signature: SALEH.A. MOUSSA**  
**Head of,**  
**Egyptian Civil Aviation Safety &  
Security Authority**

**Enclosure.**

- 1. B737-500.FLT. SIM Approval to EgyptAir.**
- 2. Approval Certificate to Air Maroc, Casablanca**
- 3. Licensing Considerations**
- 4. Terms of Approval**





**CERTIFICATE OF APPROVAL  
FLIGHT SIMULATOR**

Number: MSR-AC010-B737-500 FLT. SIM-2D

This Certificate is issued to:

**EgyptAir**

Whose Business Address is:

Cairo International Airport  
Cairo, Egypt.

On behalf of the Egyptian Civil Aviation,  
It is hereby certified that the Flight Simulator for

**B737-500**

Located at

Royal Air Maroc,  
Casablanca Airport  
Anfa

**Has Satisfied the Qualification Requirements Prescribed In**

Egyptian Civil Aviation Regulations (ECARs) Part 121 Section 121- 407 Approval of Aircraft Simulators, and Appendices "E" and "F" Flight Training, Proficiency Check Requirements Respectively, and Appendix H to Part 121- Advanced Simulation. The Simulator must Maintain French DGCA, Approval and Qualification Level with JAR STD 1A as Reference

*Subject to the conditions of the attached Specifications.*

*This Certificate is not transferable, and unless cancelled, revoked, suspended or varied shall continue in effect from September 24<sup>th</sup> 2003 until the end of September, 2004*

Issued at: Cairo, Egypt  
Date: 24, September 2003

Signature: SALEH A. MOUSSA  
Head of,  
Egyptian Civil Aviation Safety &  
Security Authority

Arab Republic of Egypt  
Ministry of Civil Aviation  
Egyptian Civil Aviation Safety  
& Security Authority



جمهورية مصر العربية  
وزارة الطيران المدني  
سلطة الطيران المدني المصري

**APPROVAL CERTIFICATE**  
FLIGHT SIMULATOR

**This Certificate is issued to:**

**Air Maroc**

**Whose Business Address is:**

**Air Maroc,  
Casablanca  
RAM**

Upon finding that its organization complies in all respects with the requirements of the Egyptian Civil Aviation Regulations relating to the establishment of a Flight Simulator as described below, for the approved Training and Testing for **EgyptAir-Cairo**. This certificate, unless cancelled, suspended or revoked, shall continue in effect until **end of September 2004**

**Simulator Specifications:**

Aeroplane/Type/Class Simulated	B737/400-500
Category	: Full Flight Simulator
Data Package	: Boeing STD
Manufactured by	: CAE Electronics LTD - 1993.
Approval and Level	: JAR - STD 1A Level "GD"
Engines Type	: CFM - 56 - C1
Engine Instrumentation	: Boeing Standard
AFCS / EFIS	: Honeywell / Collins
Flight Management System	: Smith Industries
Visual System Manufacturer; and Type	: Vital VII , Day / Bright Day / Dusk / Night . : 180 *40
Motion System/ and control loading Manufacturer	: CAE/Hydraulic actuator with digital control electronics : 6 Degrees of Freedom CAE series 500 6 DOF
Other Equipment	: TCAS-ATIS & RT Chatter-SATCOM-EGPWS-GPS
Simulated Computer Manufacturer; and Type ( Host Computer )	: IBM Risc 6000
Instructor's Station	: Dual Indigo Touchscreen

**Note: (1) A satisfactory assessment of one simulator session is required before use.**

**Note: (2) A satisfactory assessment of flight Simulator Operators is required by ECASSA Flight Inspector.**

No. and Date of Issue:

**MSR-B737/500 2D 24, September 2003**

Signature: **SALEH.A. MOUSSA**  
Head of,  
Egyptian Civil Aviation Safety &  
Security Authority



**TERMS OF APPROVAL**

*Issued To: Royal Air Maroc - Casablanca  
Number: MSR-AOC-AC 010 -B737/500 FLT SIM - Issue 1  
Date of Issue: 24/09/2003*

*The following terms of approval have been granted to Royal Air Maroc - Casablanca  
in respect of their organization at:*

*Royal Air Maroc,  
Casablanca Airport  
Anfa*

- 1. B737/500 Simulator to maintain French DGCA Approval.*
- 2. The Simulator maintains Qualification Level "D" with JAR -STD 1A as reference Document until the end of September 2004, unless sooner refused, revoked, suspend or varied.*

*Issued at: Cairo, Egypt  
Date: 24<sup>th</sup> of September 2003*

*Signature: SALEH.A. MOUSSA  
Head of,  
Egyptian Civil Aviation Safety &  
Security Authority*



ministère de  
l'Équipement  
des Transports  
du Logement  
du Tourisme  
et de la Mer



direction  
générale  
de l'Aviation  
civile

service  
de la Formation aéronautique  
et du Contrôle  
technique

## Certificat de Qualification STD (STD QUALIFICATION CERTIFICATE)

Nr F-173Z

Pour le compte de la Direction Générale de l'Aviation Civile,  
membre des Autorités Conjointes de l'Aviation (JAA), il est  
déclaré par ce document que le simulateur de vol  
*(on behalf of the French DGAC, a member of the Joint Aviation Authorities it is  
hereby certified that the under mentioned flight simulator)*

**B 737-500**

Situé à  
*(located at)*

~~CASABLANCA MARRAKECH~~

A satisfait les exigences de qualification du JAR-STD 1A et est  
qualifié pour le niveau **DG**  
*(has satisfied the qualification requirements prescribed in JAR-STD 1A and is  
qualified for level DG)*

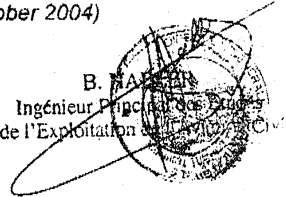
Ce certificat n'est pas transmissible et, à moins qu'il ne soit  
suspendu, retiré ou modifié, reste valable jusqu'au :  
*(this certificate is not transferable and unless sooner suspended revoked or varied,  
shall continue in effect until)*

~~31 Octobre 2005~~ (31<sup>st</sup> October 2005)

L'adjoint au Chef du Bureau des Equipages  
et des Procédures

Paris, le 27 Octobre 2004 (Paris, on 27<sup>th</sup> October 2004)

B. J. A. J. E. N.  
Ingénieur Principal des Equipages  
et de l'Exploitation de l'Aviation Civile



1.17.3.23 Simulator used by Flash Airlines at RAM).

Including

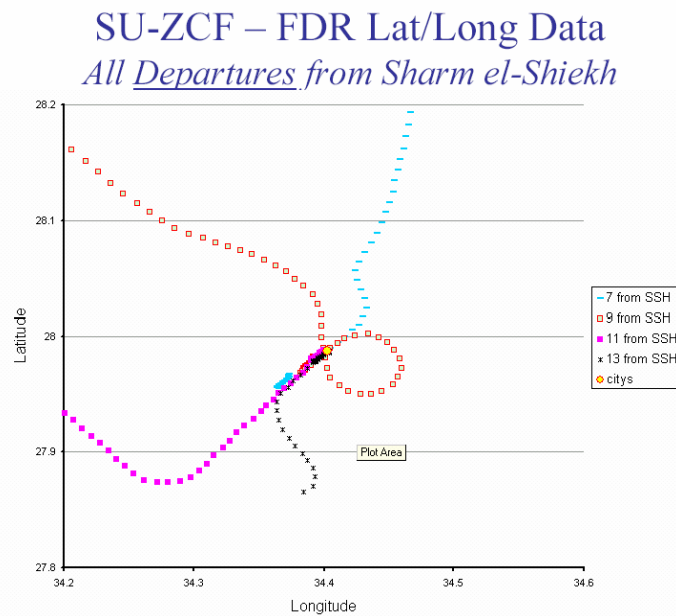
- FCC options
- Ground proximity
- Bank angle options
- Display type installed
- FD type (split or integrated cue)

See section 1.16.1.10.

Boeing answer to MCA request

1.17.3.24 *Flash Airlines procedures regarding which pilot (PF or PNF) engages the autopilot, Boeing recommended practice*  
 No written procedure was found in Flash Airline FOM regarding this issue. Boeing procedures and common practices are for PF to connect the autopilot.

1.17.3.25 Additional information regarding dispatch from SSH  
 A. All departures from SSH (accident aircraft)



- 7 Departure from SSH
- 9 Departure from SSH
- 11 Departure from SSH
- 13 Departure from SSH

Same crew did flight no13 "Accident flight" and flight no 9 "SSH /TRN", following a comparison between the two flights.

FDR SSH Departure no.	Flight 13 Accident Flight	Flight no.9
Date	3 <sup>rd</sup> Jan, 04	2 <sup>nd</sup> Jan, 04
Take off Time	2.42 GMT	4.37 GMT
Runway	22R	04L
Captain	Khedr Aabdalla Saad	Khedr Aabdalla Saad
First officer	Amr Mahmoud Shafe	Amr Mahmoud Shafe

Autopilot in Command	A	A
Autopilot engaged at	3392 ft	2836 ft
Autopilot Mode	CMD /Heading Select	CMD /Heading Select

B- Extension of the outbound legs before beginning the turn

Interviewing Flash Airlines chief pilot:  
Flash Airlines chief pilot stated that during the departure from SSH, Flash Airline pilots might extend the circuit as the situations need whether day or night departures (departure over water is mandatory)

Actual pattern flown depends on airplane performance (weight, OAT, etc). Most airplanes widen the pattern to gain additional altitude as a pilot technique. VOR crossing altitude restriction is shown on charts. This information should be added to Operations Group Notes.

**1.18. Additional Information**

**Flash Airlines Flight 604 Investigation  
Crew Behavior Subcommittee**

**Minutes of a Meeting Held at the Offices of the Ministry of Civil Aviation**

**Cairo, Egypt  
August 23-26, 2004**

**Materials Provided by MCA**

1. Paragraph interview summaries
2. One page summary of medical records provided to MCA by Egyptian Air Force after the retirement of the accident captain
3. Ops group chairman's factual report
4. Capt's flight time summary & schedule for previous 30 days
5. FO's flight time summary & schedule for previous 30 days
6. Capt's MCA pilot certification file
7. Capt's CV (1-page summary of qualifications and type certificates)
8. Captain's meteorology training course certificate from Egyptian Air Force (taken by Capt in 1984 and provided to MCA when he became civil pilot)
9. Capt's Proficiency Check Form from May 12, 2003 and transition training form from May 13, 2003
10. Capt's recurrent training form from Dec 16, 2003
11. Capt's Line Check form from July 23, 2003
12. Capt's Oral Exam form from May 12, 2003
13. Capt's ICE training form from May 28, 2003
14. Capt's Fixed Base Sim training record from April 28, 2003
15. Capt's Full Flight Sim training record from May 3-12, 2003
16. Capt's flight time records from the Air Force, Dec 14, 1999
17. FO's MCA pilot certification file
18. FO's transition training record from June, 2002
19. Flash Air Ground syllabus for 737 -300 course
20. FO's Proficiency Check Form from June 30, 2002
21. page #2 of previous
22. FO's Proficiency Check Form from July 11, 2002 (difficult to read)
23. FO's ICE training form from Aug 12, 2002
24. page #2 and #3 of previous
25. FO's Competency Check (ground school on emergency operations- training conducted at Egypt Air) from May 22, 2002
26. FO's Proficiency Check form from May 15-16, 2003
27. FO's Recurrent Training form from Dec 11, 2003
28. FO's Flash Air special course on emergency procedures, HAZMA T, first aid (practical test tied to handling dangerous goods)
29. FO's MCA test performance and systems certification oral exam
30. FO's basic indoctrination course form (from MCA at Egypt Air facility)
31. FO's ICE form
- 32-39 -FO's full flight simulator training form from June 22-July 7, 2002
40. MCA CVR-FDR overlay plots (3 pages)



Materials made available for review during the meeting:

- MCA medical certification records of the captain
- Flash Air general operations manual
- Flash Air training manual

Definition of spatial disorientation

Spatial disorientation is an incorrect perception of attitude, altitude or motion of one's own aircraft relative to the position of the Earth.

Type I spatial disorientation:

Unrecognized spatial disorientation. No conscious perception of SD. Distractions are often antecedents to the accident. Crash with no distress or concern expressed. No mayday or other than routine communications. Unusual or inappropriate aircraft attitude, but pilot does not make any appropriate corrective action. Pilot is apparently oblivious to the situation.

Type II recognized:

Conscious manifestation of a problem. Pilots often incorrectly refer to this experience as vertigo. Pilot recognizes conflict between perceived and intended or expected attitude. Can assume that the instruments are operating incorrectly. Might not properly react because of difficulty accepting indicated correct control input or might just be puzzled about the situation. Confusion might persist after recovery and lead to compounding of SD problem.

{Veronneau, S.J.H. & Evans, R.. (2004). Spatial disorientation mishap classification, data and investigation. Previc, F.H. & Ercoline, W.R. (Eds) Spatial disorientation in aviation. American institute of Aeronautics and Astronautics.}

Conditions for establishing spatial disorientation

1. Presence of inaccurate or misleading vestibular cues.
2. Absence of visual cues or presence of misleading visual cues.
3. Presence of a distraction capable of drawing attention away from attitude displays.

Closing Comments

This is a preliminary report. More work is needed to comprehensively address all human factors issues relevant to this accident, as needed. Complete minutes of CBS meeting will be made available to the sub committee for further work and analysis

## Interviews regarding Captain Kheider Abdullah

- **A.V.M. Ibrahim Omran,**  
Worked together in the Egyptian Air Force and later in Civil Aviation.  
A religious man, accurate in his work, does not recall medical complaints or use of any significant medication, was aware of maintaining his health, had self respect in all dealing with others.
- **MRS. Olfat – wife of Captain Kheider**  
Spoke very highly of him; he never created any problem for her all through their married life – chose to cure any minor health problem by using natural components such as herbs – played soccer until five years ago – never complained of headaches, dizziness or unbalance, did not mention any work related problems to her or his children.
- **Meeting with Captain Khedr's wife 24/10/2004**  
All his life Captain Khedr motivation for flight was very high he used to care of his health and eat organic foods and much salad. When he is expecting a journey he used to close his room to have a good sleep while taking off the telephone. He was married since 30 years; he has 3 children and one grand child. Two children are living with him.  
No accidents either aeroplane or crush car was reported. He was much praised at work. In the year 1997 he was awarded a prize when he landed in a difficult weather in Sarajevo.
- **First Officer Yasser Elseesy**  
**Important note:** F/O Elseesy flew with Captain Kheider 48 hours prior to the crash.  
Had good relations with everybody regardless of position or rank. The last flight was the F/O birthday and the Captain celebrated the event on the A/C by sharing a cake with all the crew, this gesture left a very positive impression on everybody.
- **First Officer Hany El Meligy**  
Says Captain Kheider was calm and balanced person and in spite of his long experience he always took time to read and prepare well before any flight, he was well disciplined and did not smoke.
- **First Officer Sherif Darwish**  
Flew frequently with Captain Kheider, learnt a lot from him and his long experience, was of good character, calm during flights and he did not observe anything about his behavior that was not normal.
- **First Officer Heba Darwish**  
Flew frequently with Captain Kheider, she says that he was intelligent, observant and highly concentrated on his work during flights, balanced, calm and disciplined.
- **Meeting with traffic officer Mr. Amr Shawky**  
(Sharm El Sheikh Station Manager)

Mr. Amr met the 3 crew members and he know them well during the months proceeding the accident. Crew members:

- 1) Captain Khedr.
- 2) F/O Amr El Shafy.
- 3) Engineer Mostafa Askar.

He used to see them in the office during work and a lot during rest periods in Sharm El Sheikh City. Either staying in a hotel or taking supper together in a restaurant in the City.

He noticed they were pleasant and within normal behavior. No special incidents or accidents or quarries occurred during that period.

Captain Khedr was specially accurate and meticulous in his work and famous for his punctuality. He likes his work very much and talks about it with pride and satisfaction. He used to smile and talk nicely to all crew members specially before flights. Between journeys he used to stay at hotel taking complete rest. I used to see Captain Khedr daily in between trips.

On the 3<sup>rd</sup> day before accident nothing specially was observed with normal relationship with a crew.

- **On the day of the accident**

Due to pressures of reservation in hotels, Captain Khedr and F/O were in Fantasia hotel and the rest of the crew was in Coral Beach Hotel. The bus brought the crew first then the Captain and first officer from the 2<sup>nd</sup> hotel with a difference of 15 min. the aeroplane arrived and I gave them the documents and Captain Khedr requested the usual questions (like the № of passengers).

Captain Khedr was joking with me and told me I can take you with me now to Cairo (on aeroplane) this happened while the first officer is busy checking, the different systems of aeroplane and entering the computerized route plan he is usual a calm person with little but pleasant talking.

## **1.19. New Investigation Techniques**

### 1.19.1 Spatial disorientation :

- Definition
- The way the SD works
- Crew fatigue
- Human related factors

Refer to (tests and researches), 1.16.4. Tests and researches conducted by MCA, Spatial Disorientation Studies  
Additional work can be done through adding the report of the CBS group meeting)

# **Exhibits**

# **Exhibit A**

## **AIRCRAFT MAINTENANCE RECORDS GROUP FACTUAL REPORT**

Ministry of Civil Aviation  
Accident Investigation Central Administration  
Accident Investigation Team  
Cairo, January 26,2004

**AIRCRAFT MAINTENANCE RECORDS GROUP**  
**FACTUAL REPORT**

A. ACCIDENT

Location: Sharm El Sheikh Airport, South Sinai  
Date: January 3, 2004  
Time: 0246 UTC, 0446 Local Time  
Aircraft: Flash Airlines, Flight FSH 604,B737-3Q8, SU-ZCF.

B. AIRCRAFT MAINTENANCE RECORDS GROUP

C. SUMMARY

On January 3, 2004, about 0246 UTC, Flash Airlines flight FSH604, a B737-3Q8, SU-ZCF plunged into the Red Sea shortly after takeoff from Sharm El Sheikh International Airport (SSH) in South Sinai, Egypt. The flight was a passenger charter flight to Charles de Gaulle Airport (CDG), France with a stopover in Cairo international Airport (CAI) for refueling. Two cockpit crewmembers (Pilot and Co-pilot), three cabin attendants and 143 passengers (135 French and 8 Egyptian) onboard were killed. The airplane was destroyed due to impact forces with the red sea.

On January 11, 2004, the Aircraft Maintenance and Records Group convened at Flash Airlines Headquarter in 166b El Hegaz St, Heliopolis, Cairo Egypt in order to meet and interview Flash Airlines Technical Director and his staff. They collected all documents and records available for the subject aircraft. The rest of the aircraft records were delivered to the Accident Investigation Team on January 14, 2004. The Aircraft Maintenance and Records Group examined Flash Airlines maintenance program and the airplane records of SU-ZCF. The Aircraft Maintenance and Records Group completed the field review and examination on January 26, 2004.

The Aircraft Maintenance and Records Group performed a review of airworthiness directives, maintenance program , weight and balance report, supplemental type certificates, maintenance discrepancies, and contracts. Results of these reviews are summarized in this report.

All Interviews are attached to Appendix A of this report.

## **D. DETAILS OF THE INVESTIGATION**

### **1.0 Flash Airlines Air Operator Certificate (AOC)**

Flash Airlines is approved as air operator (charter air carrier) under ECAR 121 by the ECAA, and operating under approval no 018.

Flash Airlines has its main office in Cairo, Egypt at 166b El Hegaz St. Heliopolis . Beginning in 2000, Flash Airlines leased the first B737-300 from the International Lease Financial Cooperation ILFC. In June 2001 another B737-300 from ILFC was added to Flash Airlines fleet which made the company fleet two aircraft the same type. The Operations Specifications was issued to the company in Feb 2000 and last revision was on October 29<sup>th</sup> 2003.

### **2.0 Aircraft History**

Per Egyptian Civil Aviation Safety and Security Authority (ECASSA), civil aviation aircraft registration records , the International Lease Financial Cooperation (ILFC) leased the accident aircraft, serial number 26283, to Flash Airlines on May 14, 2001. It was registered in Egypt on June 17, 2001 under tail number SU-ZCF to be operated by Flash Airlines. The subject aircraft basic information as following:

Aircraft Type	: B737-3Q8
Minimum Crew	: 2 (Pilot and Copilot)
Registration Mark	: SU-ZCF
Serial Number	: 26283
Manufacture Date	: October 1992
Line Number	: 2383
Variable No	: PQ294



Interior Configuration : Total 148 Economy Class

ECAA Minimum Number of Flight Attendant : 3

### 3.0 Aircraft Maintenance

#### 3.1 Maintenance Program Summary- Flash Airlines B737-300

Flash Airlines has developed their customized Maintenance Program . The Maintenance Program last revision was issued on January 20, 2003 and approved by the Egyptian Civil Aviation Safety and Security Authority (ECASSA), Airworthiness Central Administration under approval No MOCA/FLASH/737-300/MP/R2/03. This Maintenance Program was incorporated guidance from Boeing Maintenance Planning Document (MPD) Revision July 2002.

The Periodic Service Check is accomplished on layover. The check is performed as a walk-around, visual inspection and servicing when necessary.

The Routine Inspection is performed every 250 flight-hours (A Checks). A Routine Inspection Procedures Index is used to assure the check is completed. The Inspection consists of a visual inspection of the aircraft's major components, servicing, operational and functional checks.

The Maintenance Program contains subparts related to:

- 1- Line Maintenance Checks: Transient, Daily and Weekly Checks.
- 2- "A" Checks which should be carried out at 250 Flight Hours Interval and its multiples. The following chart will show how are the "A" checks cycled:

"A" Check Cycle (250 Flight Hours Intervals per Cycle – 16 "C" Check)																
Check	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16
A	x	x	x	x	x	x	x	x	x	x	x	x	x	x	x	x
2A		x		x		x		x		x		x		x		x
4A				x				x				x				x
8A								x								x

- 3- "C" Check which should be carried out every 4000 flight hours and its multiples. The following chart will show how are the "C" checks cycled.

"C" Check Cycle (4000 Flight Hours Intervals per Cycle)								
Checks	1	2	3	4	5	6	7	8
1C	x	x	x	x	x	x	x	x
2C		x		x		x		x
4C				x				x
6C						x		
8C								x

- 4- Components: This section contains general information on selected airframe and engine components. They are Condition Monitoring, On Condition or Hard Time.
- 5- Structure Inspection which should be carried out every 24000 Flight Hours. Structural inspections are performed in accordance with guidelines set down by the manufacturer Boeing MPD.
- 6- Corrosion Prevention Control Program (CPCP)
- 7- Pylon Inspections (ATA 54) the 15 Months and 45 Months Checks

The checks and inspection times can not be exceeded except by using the short term escalation as authorized per the Operations Specifications D95 issued by ECASSA to Flash Airlines and considered as a part of the air operator certificate AOC No 18.

The last "A" check accomplished by Flash Airlines and the last "C" check and Structural inspection carried by Braathens Engineering and Maintenance for the SU-ZCF were as follows:

- "8A" Check : December 12, 2003 at 25423:50 Flight Hours
- "7C" Check : From Nov 3 - Dec 21, 2002 at 23531 Flight Hours
- Last SI Check : From Nov 3 - Dec 21, 2002 at 23531 Flight Hours
- Last 15 M Chk : From Nov 3 - Dec 21, 2002
- Last 45 M Chk : From Nov 3 - Dec 21, 2002

Copy of the checks done on the aircraft is attached (attachment 01)

### 3.2 Maintenance Time Limitations

Scheduled maintenance checks are approved by ECASSA (Flash Airlines Operations Specifications D88), and are in accordance with the Boeing 737-300 Maintenance Planning Documents MPD<sup>1</sup>.

<sup>1</sup> The Boeing 737-300 Maintenance Planning Data (MPD) document provides maintenance planning information necessary for each 737 operator to develop a customized scheduled maintenance program

Transient Check:	Before each flight
Daily Check:	Every 24 hours that the airplane is in service.
7 days check:	Every 7 Calendar days.
Check “A” Systems and multiples:	Every 250 Flying hours and multiples.
Check “C” Systems and multiples:	Every 4000 Flying hours.
Structural Inspections:	Every 24000 Flying hours

### 3.3 Aircraft Summary

Total Hours at Time of Accident:	25603 Flight Hours
Total Cycles at Time of Accident:	17976 Flight Cycles

### 3.4 Weights and Balance Summary

According to the Egyptian Civil Aviation Regulations, ECAR 91 Appendix H attachment 1 the aircraft has to be reweighed every three years . Furthermore, aircraft must be reweighed if the effect of modifications on the mass and balance is not accurately known. Flash Airlines aircraft was weighed last time on December 19, 2002 in Braathens SAFE, Stavangar, Norway. and recalculated by Flash Airlines after the reenforced cockpit door modification installation on November 1<sup>st</sup>, 2003, and the results were as follows.

Empty Weight	:	70794 lbs
Moment	:	45921358.6 lb.in
% AMC	:	17.42%

### 3.5 Engines: CFM56-3C-1

Engines are maintained in accordance with Flash Airlines Maintenance program and are based on the life cycle limits of the rotating components. CFMI Engine maintenance manual together with the applicable Service Bulletins and engine teardown data determine these limits. Overhauls are performed at the SNECMA MOROCCO Workshop or other authorized Certified Repair Station.

	<u>Engine Position 1</u> (Left Side)	<u>Engine Position 2</u> (Right Side)
Serial Number (ESN)	857352	856481
Time Since New (TSN)	25314 hours	26045 hours

Cycles Since New (CSN)	17815 Cycles	17523 Cycles
Date of Installation on SU-ZCF	August 1998	Jan 3, 2003
Time Since Last O/H	8741 Hours	1828 Hours
Cycles Since Last O/H	6188 Cycles	909 Cycles

Engine Disks and First Limiters Status as per attached (attachment 02)

### 3.6 Engine Monitoring System

Flash Airlines engines are monitored as per the manufacturer (CFMI) engine condition monitoring program (Sage Trend Analysis program). Sage is a set of programs which collectively provide the functionality to perform standard condition monitoring of CFMI engines. Sage is designed to work in an interactive environment with the major analytical calculations performed at scheduled times throughout the day.

By reviewing the engine condition monitoring trend reports for both engines, they showed no deviation or important shift, the EGT margin is considerable ok. Engine Condition Monitoring cruise trend sheet is attached (attachment 14)

### 3.7 Flight Data Recorder/ Cockpit Voice Recorder.

Description	P/N	S/N	Test Date	Workshop
Sundstrand FDR CVR	980-4120-DXUN 93A100-80	10069 57994	O/H 18/11/02 Tested 12/11/02	Air Transport Avionic Braathens

### 3.8 Aircraft Status

#### 3.8.1 Minimum Equipment List (MEL)

Flash Airlines Customized Minimum Equipment List CMEL was approved by the ECASSA on Feb 23<sup>rd</sup>, 2002 under approval number ECASSA/FLASH/MEL/737-300/02/02 according to MMEL<sup>2</sup> R40, meanwhile another revision according to the last Master Minimum Equipment List (MMEL) revision 45 is currently under approval by the ECAA.

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<sup>2</sup> The Master Minimum Equipment List (MMEL) is a FAA approved document, with participation by the aviation industry, intended to assist airline operations and maintenance organizations in developing the procedures required to operate the aircraft in various nonstandard configurations. It is also intended to permit operation with inoperative items of equipment for a period until repair can be accomplished. In order to maintain an acceptable level of safety and reliability, the MMEL establishes limitations on the duration of and conditions for operation with inoperative equipment. It is the basis for development of individual operator MEL that take into consideration the operator's equipment configuration and operational conditions.

### **3.8.2 Aircraft Condition Report (A/C deferred defects)**

No deferred items were recorded in the aircraft deferred snags log Book

### **3.8.3 Type Certificate Data Sheet**

FAA "Type Certificate Data Sheet" number A16WE (revision 28, dated October 29, 1999) for B737-300 series airplanes was reviewed for compliance conditions and limitations. No discrepancies were noted. Type certificate Data Sheet attached (attachment 15)

### **3.8.4 Supplemental Type Certificates**

Supplemental Type Certificates supplied by Flash Airlines were reviewed. One Supplemental Type Certificate was issued to install a Matsushita Audio Entertainment System in accordance with General Aerospace Engineering Order No GA-23-1042. STC attached (attachment 16)

### **3.8.5 Airworthiness Directives (AD) Summary and Service Bulletins (SB) Summary**

The Airworthiness Directives compliance status list dated January 12<sup>th</sup>, 2004 (attachment 03) submitted by Flash Airlines was reviewed with special concentration on AD's carried out after the aircraft was leased by Flash Airlines.

The previous AD's Status which was forward to Flash Airlines during the aircraft delivery was reviewed with special attention to those AD's which had an open or repetitive status.

All listed Airworthiness Directives and Service Bulletins have been complied with no discrepancies noted.

Service Bulletins compliance status attached (attachment 17).

### **3.8.6 Time Controlled Components**

Time Controlled items listed on the Boeing 737-300 Maintenance Program, including task card number, part/serial numbers, and the time interval, were reviewed. The listing by task card noted categories (inspections, functional check, restoration, or scrap). Flash Airlines has no exceedance for the MPD recommendations. No discrepancies were noted. Components list replaced by Flash Airlines attached (attachment 04)

### **3.8.7 Prior Discrepancies/Accidents Involving SU-ZCF**

Per Flash Airlines records, no previous accidents were reported for the accident aircraft.

### **3.8.8 Logbook Forms**

The original aircraft Technical Log Book sheets were reviewed for the last three months from September 27, 2003 through December 2003 for discrepancies, no trends or discrepancies noted. The list of the reviewed Technical Log Book sheets is attached:

Few number of pilot reports are recorded. Some corrective actions recorded by the maintenance staff without pilot reports. Copy of the Tech Log Book entry listing is attached (attachment 05)

Copies of the Technical Log Book sheets following the original copies (from Dec 27, to Dec 31, 2003) were reviewed also. The following are the review results:

- The Line Maintenance checks (transient, PDC and Daily) are properly carried out and recorded by the certified staff.
- All Pilots acceptance are recorded.
- Pilots reports are very limited, however many corrective actions are recorded by the maintenance staff.
- Some Technical Log Book sheets are missed From serial no 1998 up to the accident flight. (Shown as per attached schedule)

### **4.0 Maintenance Participants**

Prior to the accident, the most recent scheduled maintenance performed on the accident aircraft was (8A check) done by Flash Airlines, Cairo base on December 11, 2003. Also, the PDC check was carried out by Flash Airlines Engineer at SSH station just before the accident. Due to the unavailability of the missed technical log book sheets, an interview, and document review were conducted to obtain information about the maintenance performed at this station before the accident flight.

The on board ground engineer said that there weren't any abnormal problem with the aircraft during the flight to SSH from VCE. And nothing was reported from the pilot. Interview attached (attachment 06)

### **4.1 Flash Airlines Approved Maintenance Organization (AMO)**

Flash Airlines is also approved under ECAR 145 as a repair station . The approval number is CAI/FLASH?AS/1/2001. The certificate is valid until July 30<sup>th</sup>, 2004 and was issued on July 31, 2001. The certificate is limited to line maintenance up to the 8A check for the B737-300. Flash Airlines maintenance base is Cairo international Airport.

Flash Airlines coordinates the maintenance program through its ECAR Part 145 certificate. The Company General Maintenance Manual (GMM) provide guidance related to the Aircraft Maintenance program as the Maintenance Procedures, Maintenance staff Training... etc.

Personnel working on Flash Airlines Fleet at the various maintenance facilities must be familiar with the policies and procedures spelled out in the company GMM. The Quality Control Manager puts the newly hired employees through a twelve-hour Indoctrination Course. The Indoctrination course Flash Airlines policy and procedures, and training practices. It is accomplished before maintenance engineer begins to work at the Flash Airlines facility. The training is documented on a maintenance training attendance record, recorded on the employee's training file.

#### **4.2 Contracted Repair Station Listing**

- EgyptAir Maintenance and Engineering
- Braathens Maintenance and Engineering
- Snecma Morocco Engine Services.

#### **5.0 Personnel Training and Authorization**

According to ECAR 65 the requirements for granting authorization for ground engineer are as follow:

- 1- Graduation from Faculty of Engineering or an approved training institute.
- 2- Passing the approved Basic training Course at approved Training Center or institute.
- 3- On Job Training for 18 months.
- 4- Passing written, practical and oral exams by the authority for License without Type Rating (LWTR).
- 5- Passing an approved training course for a specific type airframe and engine.
- 6- On Job Training (OJT) on the type airframe and engine for 9 months.
- 7- Attendance of training course for the company exposition procedure manual.
- 8- Passing oral and practical examination in front of the Company Examination Board (approved by the authority)
- 9- Getting the company approval.

Flash Airlines maintains its training program in compliance with Egyptian Civil Aviation Regulation requirements. The Maintenance Director and the Quality Control Manager have joint responsibility for assuring all required training is performed and recorded. Indoctrination training proceeds an employee's start date. The employee is given a 4-hour introduction course that trains one on Flash Airlines maintenance policies and procedures. The training will be documented on a maintenance training attendance record and maintained in the employee's training file.

The aircraft systems training for the A & C Engineers is accomplished through formal systems training and On-the-Job Training (OJT) Worksheets.

Engineer Mostafa Erfan Askr does the last flight release.

Engineer Mostafa was graduated from the National Civil Aviation Training Organization on September 6<sup>th</sup> 1972. He worked as a mechanic for the Kuwait Airways for twenty years during which he received the following training courses:

- 1- B 747-269B Mechanics Familiarization during the period between Feb 17<sup>th</sup> 1979 to March 3<sup>rd</sup> 1979. (Kuwait Airways).
- 2- Airbus Mechanics Familiarization Course during the period between October 6<sup>th</sup> to October 18<sup>th</sup> 1984 (Kuwait Airways).
- 3- B767 Mechanics Familiarization A&C Course during the period between February 7<sup>th</sup> to February 19<sup>th</sup>, 1987 (Kuwait Airways).

In 1991 he took the Cessna 188 course at DEVCO training center, then he got his Egyptian license without type rating (LWTR) No 1525 on August 1<sup>st</sup> 1992 which is valid until July 27<sup>th</sup>, 2004.

He joined Flash Airlines two years ago, during this two years he had the following training and exams:

- 1- B737-300 type course at EgyptAir approved training center during the period between December 22<sup>nd</sup>, 2002 to February 27<sup>th</sup>, 2003.
- 2- Basic Indoctrination Course during the period between 13-14 June 2003.
- 3- An on Job Training for 9 months on Flash Airlines B737-300 fleet.
- 4- An approval authorization exam for the engine on November 2<sup>nd</sup>, 2003 and for the airframe November 3<sup>rd</sup>, 2003.

His approval No: 014 Valid until: July 26<sup>th</sup>, 2004 Issued on: Nov 28<sup>th</sup>, 2003  
LWTR No: 1525 Valid until: July 27<sup>th</sup>, 2004 issued on: August 1<sup>st</sup>, 1992

## **6.0 Contracts**

### **6.1 Flash Airlines and EgyptAir Approved Maintenance Organization Contract**

The contract between Flash Airlines and EgyptAir Maintenance and Engineering Approved Maintenance Organization (attachment 07) was signed January , 2000. There are 15 agreement statements throughout the contract identifying conditions in which the two companies will work together.

Per the contract, EgyptAir will perform maintenance routine checks (A check and its multiples and C Checks and its multiples) and any requested AD's accomplishment on the B737-300 operated by Flash Airlines.

Flash Airlines provides the work package for the required routine check including the routine task cards, engineering orders weather for Airworthiness Directives, Service Bulletins, or modifications as well as other non-routine task cards that may be required to be accomplished concurrently with the routine check, in addition to any rectified defects by EgyptAir during the check.



EgyptAir is an approved maintenance organization as per ECAR 145 under approval No CAI/EGYPTAIR/AS/01/98 issued by ECASSA

## **6.2 Flash Airlines and Braathens Maintenance and Engineering Contract.**

The contract between Flash airlines and Braathens Maintenance and Engineering in Stavanger, Norway (attachment 08). It became effective on November 3rd, 2002. There are thirty statements of understanding and two Appendices that explain the conditions of the Agreement.

Flash Airlines provides the required work scope as per their approved maintenance program. Braathens Maintenance and Engineering supplies the necessary consumables, routable parts, and equipment.

Braathens Maintenance and Engineering is approved as Per ECAR 145 approved maintenance organization under approval CAI/BRAATHENS/AS/1/2002.

## **6.3 Flash Airlines and SNECMA MOROCCO ENGINE SERVICES.**

The contract between Flash Airlines and SNECMA MOROCCO ENGINE SERVICES (attachment 09) was signed on November 7<sup>th</sup>, 2002. There are 22 agreement statements throughout the contract identifying conditions in which the two companies will work together.

Per the contract, Flash Airlines and Snecma MOROCCO ENGINE SERVICES have entered into this agreement to stipulate and regulate terms and conditions for repair/overhaul of Flash Airlines CFM56-3C-1 Engines rated 22 klbs. According to the agreed workscope, it includes repair, engine performance restoration, and application of any applicable AD's.

SNECMA MOROCCO ENGINE SERVICES is approved as Per ECAR 145 approved maintenance organization under approval CAI/SNECMA MOROCCO/AS/1/2002

## **7.0 Maintenance Performed on the A/C before the accident flight.**

### **7.1 Maintenance done by Flash Airlines Tech Staff at Cairo Base**

The Last Check carried out on the accident aircraft was an 8A check. The check was performed by Flash Airlines Technical staff at Cairo base station. The check workpackage included visual inspection, servicing, and operational checks. A routine borescope inspection for the HPT nozzles guide vanes and the combustion chamber was performed on both engines by EgyptAir with no findings. The workpackage was reviewed with no discrepancies.

## **7.2 Transient Check carried out for the Flight VCE/SSH**

A transient check was carried out in VCE by engineer Motaz Awad on January 2<sup>nd</sup>, 2004 a copy of the interview with him is attached (attachment 06)

## **7.3 Last PDC Carried out for the Accident Flight**

On January 3<sup>rd</sup>, 2003, aircraft SU-ZCF, a daily check was performed in accordance with the approved checklist as per the company maintenance schedule at SSH station just before the flight. The check was carried out by the accident flight, on board engineer (Eng Mostafa Askar).

## **7.4 Aircraft Refueling before the Accident Flight and investigations done after the accident.**

The Refueling was done for the accident aircraft on January 3<sup>rd</sup>, 2004 between 03:50 and 04:00 local time (UTC +2) for the quantity of 3500Liters by truck no 4432 belonging to Misr Petroleum Company (service invoice is attached) attachment 10.

The same truck had refueled the following airplanes on the same date:

- EgyptAir aircraft A320 SU-GBF at 02:05 LT before the accident aircraft.
- Taroum aircraft YR-GGX at 04:20 LT after the accident aircraft.
- EgyptAir aircraft SU-GCD at 05:10 LT after the accident aircraft.

After the aircraft accident, Three fuel samples had been drawn from the Misr Petroleum fuel truck on January 3<sup>rd</sup>, 2004 at 12:45 local time. One of them was used for a dehydrated Copper Sulfate capsule field inspection for fuel water content, which was satisfactory (attachment 11). The two others samples were sent to the following laboratories for analysis:

- The Egyptian Petroleum Research Institute Nasr City, Cairo (attachment 12).
- Misr Petroleum Company, Ghamra Research Center Laboratory (attachment 13).

The Egyptian Petroleum Research Institute (EPRI) performed the Jet (A-1) fuel analysis, ASTM distillation and ASTM D-86. The results of these analyses show that all the values are within limits except for the water content, ppm, which is 48, and the max is 30.

The Misr Petroleum Co, Ghamra Research Center Laboratory performed the same analyses done by (EPRI), all the results comply with the requirements of DES-STAN 91-91 issue 4 (DERD 2494) and the joint fueling systems "Checklist" specifications for JET A-1 issue 19 Sept, 2002.

*Appendix A*

*Attachment Listing*

Attachment 01: List of Checks done on the accident aircraft.

Attachment 02: Engine Disks and first limiters status

Attachment 03: Airworthiness compliance status.

Attachment 04: Components list replaced by Flash Airlines.

Attachment 05: Copy of the Tech Log Book Entry Listing.

Attachment 06: Eng [REDACTED] Interview.

Attachment 07: EgyptAir Contract

Attachment 08: Braathens Engineering and Maintenance Contract.

Attachment 09: Snecma Morocco Contract

Attachment 10: Fuel Service Invoice.

Attachment 11: On spot fuel field inspection.

Attachment 12: Egyptian Petroleum Research Institute Analyses Report.

Attachment 13: Misr Petroleum Co, Ghamra Laboratory analyses report.

Attachment 14: Engine Condition Monitoring Cruise Trend Sheets.

Attachment 15: Type Certificate Data Sheet.

Attachment 16: Supplemental Type Certificate, STC.

Attachment 17: Service Bulletins compliance list

## Service Bulletins compliance list

### S Dates

1551-1575	From 27-9-03 to 4-10-03
1576-1600	From 3-10-03 to 9-10-03
1601-1625	From 10-10-03 to 18-10-03
1626-1650	From 18-10-03 to 22-10-03
1651-1675	From 23-10-03 to 27-10-03
1676-1700	From 27-10-03 to 1-11-03
1701-1725	From 1-11-03 to 7-11-03
1726-1750	From 7-11-03 to 12-11-03
1751-1775	From 12-11-03 to 17-11-03
1776- 1800	From 17-11-03 to 23-11-03
1801-1825	From 23-11-03 to 30-11-03
1826- 1850	From 30-11-03 to 11-12-03
1851- 1875	From 12-12-03 to 22-12-03
1876- 1900	From 22-12-03 to 27-12-03

# **Exhibit B**

## **Flight Data Recorder (FDR) Group Factual Report**

**Ministry of civil aviation**  
Accidents Department  
Egypt, Cairo

October14, 2004

**Group Chairman's Factual Report - Flight Data Recorder**

**ACCIDENT**

<b>Location:</b>	Red Sea off Sharm el-Sheikh
<b>Date:</b>	January3, 2004
<b>Time:</b>	2:45:06 GMT
<b>Operator:</b>	Flash Airlines – Flight 604

The group convened at MCA headquarters in Cairo from January16, 2004 for readout of the FDR. The readout included transcription of the accident flight data. In addition, a transcription of the entire 25-hour contents of the FDR was accomplished.

**SUMMARY**

On January 3, 2004, about 02:45:06 UTC, 04:45:06 Local time, Flash Airlines flight FSH604, a Boeing 737-300, Egyptian registration SU-ZCF, operated by Flash Airlines, crashed into the Red Sea shortly after takeoff from Sharm el-Sheikh International Airport (SSH) in South Sinai, Egypt. The flight was a passenger charter flight to Charles de Gaulle Airport (CDG), France with a stopover in Cairo international Airport (CAI) for refueling. Flight 604 departed from Sharm el-Sheikh airport with 2 pilots (Captain and First Officer), 1 observer, 4 cabin crew, 6 off-duty crew members and 135 passengers on board. The airplane was destroyed due to impact forces with the red sea with no survivals.

**Details of Investigation**

- The accident airplane's flight data recorder (SSFDR), part number 980-4120-DXUN S/N 10069, was retrieved from the Red Sea on January16, 2004 by the French Navy. The FDR was immersed in water and sealed in an ice chest and transported to MOCA, accident investigation laboratory at Cairo.

- Readout of the FDR was accomplished using the laboratory's playback hardware, Hand held Down Load unit manufactured by ALLIED SIGNAL Part No. 964-0446-001 and recovery/ analysis/ presentation system (RAPS) software.
- Inspite of the damage that had occurred to the external case of SSFDR, the internal solid state memory was in good condition and all the available data was retrieved. RAPS considered the recorded signal and data quality to be very good.
- Data plots and tabular listings of each data parameter for the entire accident flight are included in this report. The entire 25-hour contents of the FDR were also transcribed, and the data provided to the parties to the investigation.

After the cockpit voice recorder (CVR) timing had been compared to the SSFDR vhf microphone keying and Autopilot disengages warning, a time correlation was developed.

### **Unreliable parameters**

- **Control Wheel Position**

The position of the control wheel is sensed by a position transmitter mounted under the flight deck floor. The transmitter measures the rotation of a shaft that is connected to the lateral control system with a cable and pulley arrangement. The body of the transmitter is cylindrical and is held in place by a clamp. The output may be adjusted by rotating the body of transmitter within clamp which is then tightened. The recorded position of the control wheel tended to follow the recorded position of the ailerons, and therefore appears to have the correct profile. However there was an offset or bias between the recorded position and the expected position. The value of the bias changed at irregular intervals, often when large control wheel inputs were made, and also every time that a control wheel freedom-of-motion check was conducted prior to takeoff. The shifting bias was evident in all 25 hours of FDR data.

- **Left Engine N1**

The fan speed of the left engine appears to behave normally during the first 17 hours of recorded data. During the last 8 hours (including the accident flight), the parameter recording fan speed alternates between two fixed values. All other engine parameters

for both the left and right engine are operating normally. The aerodynamic performance and simulation match discussed in section 1.16 indicates that the left engine was operating normally.

- **Slat #1 Mid Extend Discrete**

Slats position is recorded by three discrete parameters as follows:

- “Slats full extended”
- “Slats in transit”
- “Slats mid extended”

. Normally, during cruise, the slats are up, during takeoff, the slats are in the mid-extend position to provide increased low-speed lift capability. During landing, the slats are normally in the fully extended position to further increase low-speed lift capability. The position of each slat is indicated by discrete parameters on the FDR. With the exception of the "LE Slat 1 Mid Extend" parameter, all of the slat indications recorded on the FDR change in a consistent manner



## Comments

- 1) The transition of the Air/Ground discrete parameter from “Ground” to “Air” had occurred at 2:42:33 GMT, the last recovered data was recorded at 2:45:5 GMT.
- 2) TOGA mode had been engaged at 2:42:02 GMT for two seconds, and then disengaged. While checking the TOGA mode operation all over the FDR 25 Hr. Data, We notice that every time the mode engaged, one second or two seconds later disengage.
- 3) During takeoff with the aircraft magnetic heading constant, the right aileron indication was up and the left aileron indication was down.
- 4) Heading Select and Level Change modes had been selected as Flight director modes.
- 5) The FDR data indicates that the airplane was turning to the left after takeoff, and rolling back towards wings level before the autopilot engagement.
- 6) The autopilot had been engaged at 2:43:59 GMT and disengaged at 2:44:02GMT. At 2:44:03 GMT, the autopilot disengage warning was recorded.
- 7) At autopilot engagement, the Heading Select Mode was disengaged and reverted to CWS R Mode.
- 8) Between the time of the autopilot engagement and disengagement, the FDR records momentary aileron surfaces movements. The right aileron deflected to 7.2 degree TEU for one second.

- 9) After autopilot disengagement, the aircraft had turned to the right and on the other hand the ailerons repetitively moved between the neutral and the roll right direction.
- 10) At 2:44:58GMT, the aircraft roll angel reached 111.094° to the right, next second both ailerons reversed their directions and initiated aircraft recovery.
- 11) Hydraulic pressure, Engine Oil Quantity, Speed Brake Handle Position, Selected Heading and Selected Course no.1 Parameters were retrieved according to Boeing Document "Enclosure B-H200-17884-ASI"

**Attachments:**

- A- Attachment 1, Tabular data of the accident flight.
- B- Attachment 2, FDR Plots
- C- Attachment 3, Five plots represent FDR and CVR correlation.

Note: Soft Copy for all 25 hours FDR data is available at MCA upon request

**Attachment 1, Tabular data of the accident flight.**

# Flash Air B737-300 Accident  
 # Preliminary Data Created: January 20 2004  
 # MCA

Time (seconds)	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUDE ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
91864	2	34	50	216	45	309.375	0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.988558	-0.00301	-0.04574	1.23047	0.175781	0
							0.988558	-0.00504	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.988558					
							0.990848					
							0.988558					
91865				216	45	309.375	0.988558	-0.00097	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.990848					
91866				216	45	309.375	0.988558	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00504	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
91867				216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00504	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.990848					
91868	2	34	54	216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.988558					
							0.990848					
							0.990848					
91869				216	45	309.375	0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00504	-0.04574		0.175781	
							0.990848					
							0.988558					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.988558					
							0.990848					
91870				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.988558					
91871				216	45	309.375	0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.988558	-0.00504	-0.04574	1.05469	0.175781	0
							0.990848	-0.00097	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.990848					
91872	2	34	58	216	45	309.375	0.990848	-0.00504	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00097	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
91873				216	45	309.375	0.988558	-0.00301	-0.04574	1.23047	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.0437		0.175781	
							0.988558					
							0.990848					
							0.988558					
							0.990848					
91874				216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.988558					
							0.990848					
							0.988558					
91875				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.988558					
91876	2	35	2	216	45	309.375	0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.988558					
							0.990848					
							0.990848					
91877				216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
91878				216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.988558					
							0.990848					
91879				216	45	309.375	0.988558	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00097	-0.0437		0.175781	
							0.990848					
							0.988558					
							0.988558					
							0.990848					
91880	2	35	6	216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00504	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.0437		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.988558					
91881				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.993137	-0.00504	-0.0437	1.05469	0.175781	0
							0.990848	-0.00504	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.988558					
91882				216	45	309.375	0.988558	-0.00097	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00504	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.988558					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.990848					
							0.990848					
91883				216	45	309.375	0.988558	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00504	-0.0437		0.175781	
							0.990848					
							0.990848					
							0.990848					
91884	2	35	10	216	45	309.375	0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00504	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
91885				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.0437		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
91886				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.990848					
91887				216	45	309.375	0.990848	-0.00504	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
91888	2	35	14	216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00097	-0.04574		0.175781	
							0.988558					
							0.988558					
							0.990848					
							0.988558					
91889				216	45	309.375	0.990848	-0.00504	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.04574	1.05469	0.175781	0

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00097	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.988558					
91890				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558					
							0.988558					
							0.990848					
							0.988558					
91891				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.990848					
91892	2	35	18	216	45	309.375	0.990848	-0.00097	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.0437		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.990848					
91893				216	45	309.375	0.990848	-0.00097	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00504	-0.04574		0.175781	
							0.990848	-0.00504	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.988558					
91894				216	45	309.375	0.990848	-0.00097	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00504	-0.04777		0.175781	
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848					
							0.990848					
							0.990848					
91895				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.0437		0.175781	
							0.988558					
							0.990848					



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.988558					
							0.988558					
91896	2	35	22	216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.23047	0.175781	0
							0.988558	-0.00301	-0.0437		0.175781	
							0.990848	-0.00504	-0.04777		0.175781	
							0.990848					
							0.988558					
							0.988558					
							0.988558					
91897				216	45	309.375	0.988558	-0.00504	-0.04574	1.05469	0.175781	0
							0.990848	-0.00097	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04777		0.175781	
							0.988558	-0.00504	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.988558					
91898				216	45	309.375	0.988558	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04777	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.0437		0.175781	
							0.988558					
							0.993137					
							0.990848					
							0.988558					
91899				216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558					
							0.988558					
							0.990848					
							0.990848					
91900	2	35	26	216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00504	-0.04777		0.175781	
							0.988558					
							0.988558					
							0.990848					
							0.993137					
91901				216	45	309.375	0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04777		0.175781	
							0.990848	-0.00504	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.988558					
91902				216	45	309.375	0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00504	-0.0437		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.990848					
91903				216	45	309.375	0.988558	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.0437		0.175781	
							0.990848					
							0.988558					
							0.988558					
							0.990848					
91904	2	35	30	216	45	309.375	0.993137	-0.00301	-0.04777	1.23047	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.0437		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.988558					
91905				216	45	309.375	0.993137	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.988558					
91906				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00097	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.988558					
91907				216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.993137	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.988558					
							0.990848					
							0.990848					
91908	2	35	34	216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.988558					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.990848					
							0.988558					
91909				216	45	309.375	0.988558	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848					
							0.990848					
							0.988558					
91910				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.0437		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.988558					
91911				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00504	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.988558					
91912	2	35	38	216	45	309.375	0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.988558					
91913				216	45	309.375	0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.988558					
91914				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.0437		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.988558					
91915				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437	1.05469	0.175781	0

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.988558					
91916	2	35	42	216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.990848					
91917				216	45	309.375	0.988558	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.988558					
91918				216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.988558					
							0.988558					
							0.990848					
91919				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.988558					
							0.988558					
							0.990848					
91920	2	35	46	216	45	309.375	0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00504	-0.04574		0.175781	
							0.990848					
							0.988558					
							0.988558					
							0.988558					
91921				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.988558					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.990848					
							0.990848					
91922				216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.988558					
91923				216	45	309.375	0.988558	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.988558					
91924	2	35	50	216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.988558					
							0.990848					
							0.990848					
91925				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00504	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.990848					
91926				216	45	309.375	0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.988558	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00504	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.988558					
91927				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.0437		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.988558					
91928	2	35	54	216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.990848	-0.00301	-0.0437		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.990848					
91929				216	45	309.375	0.988558	-0.00301	-0.04777	1.05469	0.175781	0
							0.986269	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848	-0.00301	-0.04777		0.175781	
							0.988558					
							0.988558					
							0.990848					
							0.990848					
91930				216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.988558					
							0.990848					
91931				216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.988558					
91932	2	35	58	216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.0437		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.988558					
91933				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.990848					
91934				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.990848					
							0.988558					
91935				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.988558					
							0.990848					
							0.990848					
91936	2	36	2	216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.988558					
91937				216	45	309.375	0.988558	-0.00504	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.988558					
							0.990848					
							0.988558					
91938				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.988558					
91939				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.988558					
91940	2	36	6	216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.990848					
91941				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.0437		0.175781	
							0.990848					
							0.988558					
							0.988558					
							0.988558					
91942				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.983979	-0.00301	-0.0437		0.175781	
							0.993137					
							0.995426					
							0.993137					
							0.990848					
91943				216	45	309.375	0.993137	-0.00301	-0.04574	1.05469	0.175781	0
							0.995426	-0.00301	-0.0437	1.05469	0.175781	0
							0.993137	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.993137					
							0.993137					
91944	2	36	10	216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.988558					
							0.990848					
							0.990848					
91945				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.990848					
91946				216	45	309.375	0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.993137					
							0.990848					
							0.988558					
91947				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.990848					
							0.990848					
91948	2	36	14	216	45	309.375	0.990848	-0.00504	-0.0437	1.23047	0.175781	0
							0.988558	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.0437		0.175781	
							0.990848					
							0.990848					
							0.990848					
91949				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848					
							0.990848					
							0.990848					
91950				216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848					
							0.988558					
							0.990848					
91951				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848					
							0.988558					
							0.988558					
							0.990848					
91952	2	36	18	216	45	309.375	0.993137	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
91953				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
91954				216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.23047	0.175781	0

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.993137	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.988558					
							0.990848					
							0.990848					
91955				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437		0.175781	
							0.993137	-0.00301	-0.04574		0.175781	
							0.990848					
							0.988558					
							0.988558					
							0.990848					
91956	2	36	22	216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.988558					
91957				216	45	309.375	0.990848	-0.00301	-0.0437	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.990848					
91958				216	45	309.375	0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.988558					
91959				216	45	309.375	0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.0437		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.990848					
91960	2	36	26	216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.990848					
							0.990848					
91961				216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.988558					
							0.990848					
							0.990848					
91962				216	45	309.375	0.990848	-0.00301	-0.0437	1.23047	0.175781	0
							0.988558	-0.00301	-0.04574	1.23047	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.988558					
							0.988558					
							0.990848					
91963				216	45	309.375	0.990848	-0.00097	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.0437		0.175781	
							0.993137					
							0.988558					
							0.988558					
							0.990848					
91964	2	36	30	216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848					
							0.988558					
							0.990848					
							0.988558					
91965				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.0437		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.990848					
91966				216	45	309.375	0.988558	-0.00504	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.988558					
							0.990848					
							0.990848					
91967				216	45	309.375	0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00504	-0.04574		0.175781	
							0.990848					
							0.988558					
							0.988558					
							0.990848					
91968	2	36	34	216	45	309.375	0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.988558					
91969				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.988558					
91970				216	45	309.375	0.990848	-0.00504	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.990848					
							0.988558					
91971				216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848					
							0.988558					
							0.990848					
							0.990848					
91972	2	36	38	216	45	309.375	0.988558	-0.00301	-0.0437	1.05469	0.175781	0
							0.988558	-0.00504	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848					
							0.988558					
							0.988558					
							0.990848					
91973				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.988558					
							0.988558					
91974				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848					
							0.990848					
							0.990848					
91975				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.05794		0.175781	
							0.990848	-0.00301	-0.05387		0.175781	
							0.990848					
							0.977111					
							0.98169					
							0.98169					
91976	2	36	42	216	45	309.375	0.98169	-0.00301	-0.05387	1.05469	0.175781	0
							0.979401	-0.00301	-0.05387	1.05469	0.175781	0
							0.979401	-0.00301	-0.05387		0.175781	
							0.98169	-0.00301	-0.05387		0.175781	
							0.98169					
							0.98169					
							0.98169					
91977				216	45	309.375	0.98169	-0.00504	-0.05387	1.05469	0.175781	0
							0.98169	-0.00301	-0.05387	1.05469	0.175781	0
							0.98169	-0.00301	-0.05387		0.175781	
							0.98169	-0.00301	-0.05387		0.175781	
							0.98169					
							0.98169					
							0.98169					
91978				216	45	309.375	0.993137	-0.00504	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.990848					
91979				216	45	309.375	0.990848	-0.00504	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
91980	2	36	46	216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00504	-0.04574	1.05469	0.175781	0

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848					
							0.988558					
							0.990848					
							0.990848					
91981				216	45	309.375	0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.988558					
							0.993137					
							0.990848					
91982				216	45	309.375	0.988558	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04777	1.23047	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.993137	-0.00301	-0.0437		0.175781	
							0.990848					
							0.988558					
							0.990848					
							0.990848					
91983				216	45	309.375	0.988558	-0.00097	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00504	-0.04574		0.175781	
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.990848					
91984	2	36	50	216	45	309.375	0.990848	-0.00097	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.988558					
91985				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00097	-0.0437	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.990848					
91986				216	45	309.375	0.990848	-0.00504	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.988558					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.988558					
							0.988558					
91987				216	45	309.375	0.990848	-0.00504	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.988558					
91988	2	36	54	216	45	309.375	0.990848	-0.00504	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.990848					
91989				216	45	309.375	0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.05794		0.175781	
							0.990848	-0.00301	-0.05387		0.175781	
							0.988558					
							0.979401					
							0.98169					
							0.98169					
91990				216	45	309.375	0.98169	-0.00301	-0.05387	1.23047	0.175781	0
							0.98169	-0.00504	-0.05387	1.05469	0.175781	0
							0.98169	-0.00301	-0.05387		0.175781	
							0.98169	-0.00301	-0.05387		0.175781	
							0.98169					
							0.979401					
							0.98169					
							0.98169					
91991				216	45	309.375	0.98169	-0.00301	-0.05387	1.23047	0.175781	0
							0.98169	-0.00301	-0.05387	1.05469	0.175781	0
							0.979401	-0.00301	-0.05387		0.175781	
							0.98169	-0.00504	-0.05591		0.175781	
							0.98169					
							0.98169					
							0.98169					
91992	2	36	58	216	45	309.375	0.98169	-0.00301	-0.05387	1.23047	0.175781	0
							0.98169	-0.00301	-0.05387	1.05469	0.175781	0
							0.979401	-0.00301	-0.05387		0.175781	
							0.98169	-0.00504	-0.05387		0.175781	
							0.98169					
							0.98169					
							0.98169					
							0.979401					
91993				216	45	309.375	0.983979	-0.00301	-0.05387	1.23047	0.175781	0
							0.98169	-0.00301	-0.05387	1.05469	0.175781	0

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.98169	-0.00301	-0.05387		0.175781	
							0.979401	-0.00504	-0.05387		0.175781	
							0.98169					
							0.98169					
							0.98169					
							0.98169					
91994				216	45	309.375	0.98169	-0.00301	-0.05387	1.23047	0.175781	0
							0.98169	-0.00301	-0.05387	1.23047	0.175781	0
							0.98169	-0.00301	-0.05387		0.175781	
							0.979401	-0.00301	-0.0437		0.175781	
							0.98169					
							0.98169					
							0.98169					
							0.988558					
91995				216	45	309.375	0.990848	-0.00504	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.990848					
91996	2	37	2	216	45	309.375	0.990848	-0.00504	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.990848					
91997				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.988558	-0.00504	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.990848					
91998				216	45	309.375	0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.988558	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
91999				216	45	309.375	0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.988558					
							0.988558					
92000	2	37	6	216	45	309.375	0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
92001				216	45	309.375	0.988558	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
92002				216	45	309.375	0.988558	-0.00301	-0.04574	1.23047	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.988558					
							0.990848					
							0.990848					
92003				216	45	309.375	0.990848	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.988558	-0.00301	-0.0437		0.175781	
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.990848					
92004	2	37	10	216	45	309.375	0.993137	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04777	1.23047	0.175781	0
							0.988558	-0.00301	-0.04777		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
92005				216	45	309.375	0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.988558	-0.00301	-0.04574		0.175781	
							0.988558					
							0.990848					
							0.990848					
							0.990848					
92006				216	45	309.375	0.990848	-0.00504	-0.04574	1.23047	0.175781	0
							0.988558	-0.00301	-0.04574	1.05469	0.175781	0

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.988558					
							0.990848					
							0.990848					
92007				216	45	309.375	0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.988558					
							0.990848					
92008	2	37	14	216	45	309.375	0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.988558					
							0.990848					
							0.990848					
92009				216	45	309.375	0.988558	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.990848					
92010				216	45	309.375	0.990848	-0.00301	-0.0437	1.23047	0.175781	0
							0.988558	-0.00301	-0.0437	1.05469	0.175781	0
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848	-0.00301	-0.04574		0.175781	
							0.990848					
							0.990848					
							0.990848					
							0.990848					
92011				216	45	309.375	0.990848	-0.00301	-0.04574	1.23047	0.175781	0
							0.990848	-0.00301	-0.04574	1.05469	0.175781	0
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848					
							0.990848					
							0.990848					
92012	2	37	18	216	45	309.375	0.990848	-0.00301	-0.0437	1.23047	0.175781	0
							0.990848	-0.00301	-0.0437	1.23047	0.175781	0
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848	-0.00301	-0.0437		0.175781	
							0.990848					
							0.990848					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.990848					
							0.988558					
92013				216	45	309.375	0.995426	-0.00301	-0.0437	1.05469	0.175781	0
							0.993137	-0.00301	-0.0437	1.23047	0.175781	0
							0.993137	-0.00301	-0.04167		0.175781	
							0.993137	-0.00301	-0.04167		0.175781	
							0.995426					
							0.993137					
							0.993137					
							0.993137					
92014				216	45	309.375	0.995426	-0.00301	-0.04167	1.05469	0.175781	0
							0.995426	-0.00301	-0.04167	1.05469	0.175781	0
							0.993137	-0.00301	-0.04167		0.175781	
							0.993137	-0.00301	-0.03963		0.175781	
							0.993137					
							0.995426					
							0.993137					
							0.993137					
92015				216	45	309.375	0.993137	-0.00301	-0.03963	1.23047	0.175781	0
							0.993137	-0.00301	-0.0376	1.05469	0.175781	0
							0.995426	-0.00097	-0.03556		0.175781	
							0.993137	-0.00301	-0.03353		0.175781	
							0.993137					
							0.993137					
							0.995426					
							0.993137					
92016	2	37	22	216	45	309.375	0.995426	-0.00301	-0.02946	1.23047	0.175781	0
							0.995426	-0.00504	-0.02743	1.05469	0.175781	0
							0.993137	-0.00504	-0.02743		0.175781	
							0.997715	-0.00301	-0.02539		0.175781	
							0.995426					
							0.995426					
							0.997715					
							0.995426					
92017				216	45	309.375	0.993137	-0.00301	-0.02539	1.05469	0.175781	0
							0.993137	-0.00301	-0.02336	1.05469	0.175781	0
							0.995426	-0.00097	-0.01929		0.175781	
							0.997715	-0.00097	-0.01725		0.175781	
							0.997715					
							0.993137					
							0.997715					
							0.995426					
92018				216	45	309.375	0.995426	0.001057	-0.01318	1.05469	0.175781	0
							0.995426	-0.00301	-0.00911	1.05469	0.175781	0
							0.995426	-0.00504	-0.00505		0.175781	
							0.995426	-0.00301	0.001058		0.175781	
							0.995426					
							0.997715					
							0.997715					
							0.993137					
92019				216	45	309.375	0.995426	-0.00504	0.003092	1.05469	0.175781	0
							0.997715	0.003092	0.003092	1.23047	0.175781	0

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.997715	-0.00097	0.007161		0.175781	
							0.997715	-0.00504	0.007161		0.175781	
							0.997715					
							0.997715					
							0.995426					
							1.00001					
92020	2	37	26	216	45	309.375	1.00001	-0.00301	-0.00098	1.05469	0.175781	0
							1.00001	0.001057	-0.01929	1.23047	0.175781	0
							1.01603	-0.00097	-0.00098		0.175781	
							1.00001	0.003092	-0.01115		0.175781	
							0.963375					
							0.967954					
							1.00001					
							1.02061					
92021				216	45	309.727	1.00001	-0.00504	-0.00911	1.05469	0.175781	0
							0.977111	-0.01114	-0.01725	1.05469	0.175781	0
							0.995426	-0.00097	-0.01318		0.175781	
							1.01374	0.001057	-0.01725		0.175781	
							1.00458					
							0.979401					
							0.977111					
							1.00229					
92022				216	45	309.727	1.02977	0.001057	-0.02743	1.05469	0.175781	0
							1.00458	-0.00504	-0.02946	1.05469	0.175781	0
							0.967954	-0.00301	-0.02743		0.175781	
							0.986269	0.003092	-0.02336		0.175781	
							1.01603					
							1.00001					
							0.967954					
							0.983979					
92023				216	45	310.078	1.01832	0.001057	-0.01929	1.05469	0.175781	0
							1.00916	-0.01114	-0.03353	1.23047	0.175781	0
							0.979401	-0.00301	-0.02743		0.175781	
							0.997715	0.005126	-0.02946		0.175781	
							1.02519					
							0.997715					
							0.961086					
							0.988558					
92024	2	37	30	216	45	311.133	1.02519	-0.00301	-0.03149	1.23047	0.175781	0
							1.00916	-0.00097	-0.02132	1.23047	0.175781	0
							0.963375	0.003092	-0.03963		0.175781	
							0.977111	0.015299	-0.02743		0.175781	
							1.00458					
							1.02977					
							1.00458					
							0.970243					
92025				216	45	312.188	0.98169	0.007161	-0.03353	1.23047	0.175781	0
							1.01603	-0.00301	-0.02946	1.23047	0.175781	0
							1.01145	0.007161	-0.0376		0	
							0.979401	0.013264	-0.02336		0	
							0.977111					
							1.00916					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.01832					
							0.979401					
92026				216	45	314.648	0.956507	0.005126	-0.03149	1.23047	0	0
							1.00229	0.009195	-0.03556	1.05469	0	0
							1.03663	0.017333	-0.03556		0	
							1.00229	0.009195	-0.0376		0	
							0.965664					
							0.983979					
							1.0229					
							1.00229					
92027				216	45	317.109	0.965664	0.013264	-0.02743	1.05469	0	0
							0.970243	0.01123	-0.0376	1.23047	0	0
							1.00001	0.019368	-0.0376		0	
							1.02061	0.003092	-0.03963		0	
							1.01145					
							0.98169					
							0.986269					
							1.00001					
92028	2	37	34	216	45	321.328	1.00229	0.01123	-0.0498	1.05469	0	0
							1.01374	0.019368	-0.03963	1.05469	0.175781	0
							0.993137	0.021403	-0.05387		0	
							0.970243	0.02954	-0.0437		0	
							0.967954					
							1.00916					
							1.02977					
							0.970243					
92029				216	45	325.195	0.972533	0.037679	-0.05184	1.23047	-0.17578	-0.35156
							1.00916	0.02954	-0.04167	1.05469	-0.17578	-0.35156
							1.01145	0.021403	-0.05387		-0.17578	
							0.988558	0.045817	-0.04574		-0.17578	
							0.979401					
							1.00229					
							1.01374					
							0.979401					
92030				216	45	331.523	0.983979	0.039713	-0.05184	1.23047	-0.17578	-0.35156
							1.01374	0.027506	-0.04777	1.23047	-0.17578	-0.35156
							0.995426	0.041747	-0.0498		-0.35156	
							0.977111	0.035644	-0.04777		-0.35156	
							0.997715					
							1.00229					
							0.986269					
							0.986269					
92031				216	45	337.5	1.00001	0.023437	-0.05387	1.23047	-0.35156	0
							1.02519	0.027506	-0.0498	1.05469	-0.35156	0
							1.00001	0.049886	-0.05998		-0.35156	
							0.970243	0.043782	-0.04167		-0.35156	
							0.974822					
							1.00687					
							1.01374					
							0.965664					
92032	2	37	38	216	45	345.234	0.967954	0.031575	-0.04777	1.05469	-0.35156	0
							1.01374	0.031575	-0.04167	1.05469	-0.35156	-0.35156

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.01832	0.035644	-0.0437		-0.35156	
							0.983979	0.049886	-0.04574		-0.17578	
							0.974822					
							1.00458					
							1.01145					
							0.995426					
92033				216	45	351.211	0.977111	0.045817	-0.04574	1.05469	-0.17578	-0.70312
							0.974822	0.053955	-0.04167	1.05469	-0.17578	-0.70312
							1.00001	0.058024	-0.0376		-0.17578	
							0.997715	0.023437	-0.04777		-0.17578	
							0.979401					
							1.00001					
							1.02519					
							1.02061					
92034				216	45	358.945	0.967954	0.027506	-0.0376	1.05469	-0.17578	-0.35156
							0.958796	0.049886	-0.0376	1.05469	-0.17578	-0.35156
							1.00229	0.041747	-0.0437		0	
							1.02061	0.037679	-0.0498		0	
							1.00916					
							0.990848					
							0.990848					
							1.01374					
92035				216	45	4.92188	1.00001	0.035644	-0.04777	1.05469	0	0
							0.986269	0.025471	-0.03963	1.05469	0	-0.35156
							0.983979	0.045817	-0.03963		0	
							0.986269	0.053955	-0.0376		0	
							0.988558					
							0.993137					
							1.00229					
							1.00001					
92036	2	37	42	216	45	12.3047	1.00001	0.039713	-0.0437	1.05469	0	0
							0.993137	0.037679	-0.03963	1.05469	0	0
							0.979401	0.02954	-0.0437		0	
							0.986269	0.02954	-0.03353		0	
							1.00229					
							1.01145					
							0.986269					
							0.972533					
92037				216	45	17.9297	1.00687	0.031575	-0.0437	1.05469	0	0
							1.02748	0.02954	-0.03556	1.05469	0	0.351562
							0.995426	0.025471	-0.0437		0	
							0.965664	0.017333	-0.04167		0	
							0.98169					
							1.02519					
							1.01603					
							0.977111					
92038				216	45	23.5547	0.98169	0.02954	-0.0437	1.05469	0	0.351562
							1.00687	0.023437	-0.04574	1.05469	0	0.703124
							1.00687	0.017333	-0.04167		0	
							0.988558	0.019368	-0.0437		0	
							0.979401					
							0.988558					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.00687					
							1.00687					
92039				216	45	28.4766	0.988558	0.007161	-0.0437	1.05469	0	0.703124
							0.983979	0.015299	-0.0437	1.05469	0	1.05469
							0.997715	0.013264	-0.03963		0	
							1.00229	0.001057	-0.03963		-0.17578	
							0.995426					
							0.988558					
							0.997715					
							1.00687					
92040	2	37	46	216	45	34.1016	0.993137	0.009195	-0.0376	1.05469	-0.17578	1.05469
							0.98169	0.01123	-0.04167	1.05469	-0.17578	1.05469
							1.00229	0.013264	-0.04574		-0.17578	
							1.00001	0.009195	-0.0437		-0.17578	
							0.98169					
							0.997715					
							1.00687					
							0.990848					
92041				216	45	38.3203	0.990848	0.01123	-0.04777	1.05469	-0.17578	1.05469
							0.997715	0.015299	-0.05591	1.05469	-0.35156	1.05469
							0.983979	0.017333	-0.0498		-0.35156	
							0.997715	0.015299	-0.05591		-0.35156	
							1.00458					
							0.986269					
							0.990848					
							1.00916					
92042				216	45	43.5938	0.995426	0.013264	-0.05184	1.05469	-0.35156	1.05469
							0.98169	0.021403	-0.05387	1.05469	-0.35156	0.703124
							0.997715	0.025471	-0.05387		-0.52734	
							1.00458	0.027506	-0.05387		-0.52734	
							0.986269					
							0.986269					
							0.993137					
							1.00458					
92043				216	45	50.625	1.00001	0.031575	-0.05184	1.05469	-0.52734	0.703124
							0.988558	0.02954	-0.05184	1.05469	-0.52734	0.703124
							0.995426	0.035644	-0.05184		-0.52734	
							0.993137	0.037679	-0.05591		-0.52734	
							0.995426					
							0.997715					
							0.983979					
							1.00229					
92044	2	37	50	216	45	56.9531	1.01145	0.039713	-0.04574	1.23047	-0.52734	0.703124
							0.979401	0.041747	-0.04777	1.05469	-0.52734	0.351562
							0.986269	0.043782	-0.05184		-0.52734	
							1.00229	0.049886	-0.0498		-0.52734	
							0.990848					
							0.995426					
							1.00001					
							0.995426					
92045				216	45	65.7422	0.990848	0.047851	-0.04777	1.23047	-0.52734	0.351562
							0.993137	0.047851	-0.0437	1.23047	-0.52734	0

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.997715	0.055989	-0.0437		-0.35156	
							0.993137	0.049886	-0.04167		-0.35156	
							0.988558					
							1.00001					
							1.00001					
							0.993137					
92046				216	45	73.125	0.997715	0.055989	-0.04777	1.23047	-0.35156	0
							0.995426	0.070231	-0.05184	1.23047	-0.35156	-0.35156
							1.00001	0.066162	-0.0498		-0.35156	
							1.00229	0.058024	-0.0498		-0.35156	
							1.00229					
							0.995426					
							0.997715					
							0.993137					
92047				216	45	82.9688	0.98169	0.064127	-0.04574	1.23047	-0.52734	-0.35156
							0.983979	0.0743	-0.04574	1.23047	-0.52734	-0.35156
							0.977111	0.058024	-0.05184		-0.52734	
							1.00001	0.055989	-0.05184		-0.52734	
							1.01603					
							1.01145					
							0.98169					
							0.967954					
92048	2	37	54	216	45	90	1.01603	0.064127	-0.05998	1.23047	-0.52734	-0.35156
							1.02977	0.066162	-0.0498	1.23047	-0.52734	-0.35156
							0.98169	0.060058	-0.05184		-0.52734	
							0.958796	0.062093	-0.05184		-0.52734	
							0.965664					
							1.01374					
							1.05266					
							1.00916					
92049				216	45	99.4922	0.935903	0.058024	-0.04167	1.23047	-0.52734	-0.35156
							0.940481	0.068196	-0.04777	1.23047	-0.52734	-0.70312
							1.01374	0.084472	-0.0437		-0.52734	
							1.02977	0.066162	-0.04167		-0.52734	
							0.970243					
							0.967954					
							1.00458					
							1.01832					
92050				216	45	106.523	1.01374	0.055989	-0.04167	1.23047	-0.52734	-0.35156
							0.983979	0.060058	-0.0437	1.23047	-0.52734	-0.35156
							0.988558	0.05192	-0.04574		-0.52734	
							1.00916	0.060058	-0.03963		-0.52734	
							0.993137					
							0.986269					
							0.997715					
							0.995426					
92051				216	45	115.312	1.00001	0.062093	-0.0437	1.23047	-0.52734	-0.35156
							1.00229	0.045817	-0.04167	1.23047	-0.52734	-0.35156
							0.997715	0.058024	-0.04574		-0.52734	
							0.979401	0.058024	-0.0437		-0.52734	
							0.979401					
							1.01374					



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.02061					
							0.988558					
92052	2	37	58	216	45	121.641	0.967954	0.047851	-0.04167	1.23047	-0.52734	-0.35156
							0.986269	0.060058	-0.04167	1.23047	-0.52734	-0.35156
							1.00916	0.041747	-0.03963		-0.52734	
							1.00916	0.035644	-0.03556		-0.52734	
							1.00458					
							0.983979					
							0.977111					
							0.995426					
92053				216	45	127.969	1.01374	0.039713	-0.03963	1.23047	-0.52734	-0.35156
							1.01145	0.031575	-0.0376	1.23047	-0.52734	-0.35156
							0.986269	0.037679	-0.04574		-0.52734	
							0.972533	0.03361	-0.03963		-0.52734	
							0.995426					
							1.02061					
							1.00916					
							0.963375					
92054				216	45	131.133	0.965664	0.019368	-0.0437	1.23047	-0.52734	-0.35156
							1.00916	0.017333	-0.04777	1.23047	-0.70312	0
							1.03892	0.009195	-0.04574		-0.70312	
							1.00916	0.013264	-0.05387		-0.70312	
							0.94735					
							0.970243					
							1.00916					
							1.03206					
92055				216	45	133.594	1.01374	0.017333	-0.03963	1.23047	-0.52734	0
							0.958796	0.01123	-0.0437	1.23047	-0.52734	0
							0.970243	0.009195	-0.04167		-0.52734	
							1.00458	0.001057	-0.04167		-0.35156	
							1.03892					
							0.997715					
							0.961086					
							0.990848					
92056	2	38	2	216	45	134.648	1.0229	-0.00301	-0.04574	1.23047	-0.35156	0
							1.02061	0.005126	-0.03353	1.23047	0	0
							0.990848	0.009195	-0.04574		0	
							0.954217	0.005126	-0.03353		0.175781	
							0.977111					
							1.0435					
							1.04121					
							0.965664					
92057				216	45	135.703	0.940481	0.001057	-0.03963	1.23047	0.175781	-0.35156
							1.00229	0.01123	-0.0376	1.23047	0.175781	-0.35156
							1.05724	0.007161	-0.03556		0	
							1.00916	0.007161	-0.06201		0	
							0.940481					
							0.956507					
							1.00229					
							1.08471					
92058				216	45	135.703	1.05953	0.005126	-0.05184	1.23047	0	-0.35156
							0.956507	-0.00097	-0.05794	1.23047	0	-0.35156

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.956507	0.001057	-0.05591		-0.17578	
							1.00687	-0.00301	-0.0498		-0.35156	
							1.0435					
							0.995426					
							0.915298					
							0.933613					
92059				216	45	135.352	1.02748	0.003092	-0.04777	1.23047	-0.35156	-0.70312
							1.03663	-0.00097	-0.03963	1.23047	-0.35156	-0.35156
							0.954217	-0.00301	-0.06201		-0.52734	
							0.935903	0.017333	-0.05591		-0.52734	
							1.00916					
							1.07327					
							1.00916					
							0.935903					
92060	2	38	6	216	45	135.352	0.933613	0.007161	-0.06405	1.23047	-0.52734	-0.35156
							1.00458	0.005126	-0.05591	1.23047	-0.52734	-0.35156
							1.05495	0.001057	-0.04167		-0.52734	
							0.983979	0.005126	-0.05184		-0.52734	
							0.929034					
							0.949639					
							1.03892					
							1.05724					
92061				216	45	135.703	0.98169	0.005126	-0.03963	1.23047	-0.52734	-0.35156
							0.933613	-0.00301	-0.05184	1.23047	-0.52734	-0.35156
							0.993137	0.005126	-0.0437		-0.52734	
							1.04808	0.015299	-0.05184		-0.35156	
							1.01145					
							0.958796					
							0.94506					
							1.02061					
92062				212	45	136.055	1.05953	0.01123	-0.04574	1.23047	-0.52734	-0.35156
							0.98169	0.007161	-0.04777	1.23047	-0.52734	0
							0.94277	0.003092	-0.04777		-0.52734	
							0.995426	-0.00301	-0.0498		-0.52734	
							1.02977					
							1.01145					
							0.979401					
							0.979401					
92063				216	45	136.406	1.00458	0.01123	-0.0437	1.23047	-0.52734	-0.35156
							1.00229	0.001057	-0.04574	1.23047	-0.52734	-0.35156
							0.983979	0.001057	-0.0498		-0.52734	
							0.988558	0.007161	-0.03556		-0.52734	
							1.0229					
							1.02061					
							0.94735					
							0.935903					
92064	2	38	10	212	45	137.109	1.02519	-0.00708	-0.05387	1.23047	-0.52734	0
							1.0664	0.001057	-0.04167	1.23047	-0.52734	0
							1.00687	-0.00301	-0.05591		-0.52734	
							0.94735	-0.00301	-0.04574		-0.70312	
							0.956507					
							1.03206					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.04579					
							0.963375					
92065				212	45	137.109	0.922166	-0.00097	-0.05184	1.23047	-0.8789	0
							0.990848	-0.01318	-0.05794	1.23047	-0.8789	0
							1.05953	-0.00097	-0.0437		-0.8789	
							1.02977	-0.00708	-0.05998		-0.8789	
							0.949639					
							0.935903					
							1.00458					
							1.05953					
92066				212	45	136.406	1.00916	-0.00911	-0.04777	1.23047	-0.8789	0
							0.956507	-0.01114	-0.04574	1.23047	-0.8789	0
							0.94506	-0.01521	-0.05184		-0.8789	
							0.988558	-0.00708	-0.04167		-1.05469	
							1.04808					
							1.0229					
							0.94735					
							0.940481					
92067				212	45	134.297	0.990848	-0.00708	-0.05794	1.23047	-1.05469	-0.35156
							1.04579	-0.00504	-0.03963	1.23047	-0.8789	0
							1.03435	-0.01521	-0.05591		-0.8789	
							0.954217	-0.02335	-0.04777		-1.05469	
							0.956507					
							1.0435					
							1.0664					
							0.979401					
92068	2	38	14	212	45	132.891	0.892404	-0.00708	-0.03963	1.23047	-1.05469	-0.35156
							0.931324	-0.00504	-0.05387	1.23047	-0.8789	-0.35156
							1.0435	-0.02132	-0.04574		-0.8789	
							1.09158	-0.01521	-0.05794		-0.8789	
							1.02061					
							0.94277					
							0.935903					
							1.01603					
92069				212	45	131.133	1.05495	-0.00301	-0.04574	1.23047	-0.8789	-0.35156
							0.995426	-0.02335	-0.0498	1.23047	-1.05469	-0.35156
							0.94277	-0.01521	-0.05591		-1.05469	
							0.958796	0.007161	-0.05184		-1.05469	
							1.02519					
							1.03435					
							1.00229					
							0.993137					
92070				212	45	129.727	0.995426	-0.00911	-0.05794	1.23047	-1.05469	-0.35156
							0.997715	-0.01318	-0.04777	1.23047	-1.05469	-0.35156
							0.974822	0.007161	-0.04777		-1.05469	
							0.961086	-0.00911	-0.06201		-1.05469	
							0.990848					
							0.970243					
							0.990848					
							1.02519					
92071				212	45	129.375	1.00229	-0.00911	-0.06201	1.23047	-1.05469	0
							0.963375	-0.01521	-0.07829	1.23047	-1.05469	-0.35156

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.935903	-0.00301	-0.08439		-1.05469	
							1.01145	0.001057	-0.07422		-1.05469	
							1.08929					
							1.00229					
							0.899272					
							0.90843					
92072	2	38	18	212	45	129.023	1.01832	-0.01114	-0.0966	1.23047	-1.23047	-0.35156
							1.04808	-0.00911	-0.08846	1.23047	-1.23047	-0.35156
							0.990848	-0.00504	-0.09863		-1.05469	
							0.94735	-0.00301	-0.08439		-1.23047	
							0.972533					
							1.03435					
							1.01374					
							0.938192					
92073				212	45	128.32	0.90843	0.003092	-0.09456	1.23047	-1.23047	-0.35156
							0.988558	-0.00097	-0.08846	1.23047	-1.05469	-0.35156
							1.05724	-0.00708	-0.09863		-1.05469	
							0.995426	-0.00911	-0.0966		-1.05469	
							0.926745					
							0.988558					
							1.04121					
							0.995426					
92074				212	45	127.266	0.913009	-0.01114	-0.08643	1.23047	-1.05469	-0.35156
							0.94735	-0.00504	-0.09456	1.23047	-1.05469	-0.35156
							1.05953	-0.01114	-0.08236		-1.05469	
							1.07555	-0.00708	-0.0966		-1.05469	
							0.94735					
							0.899272					
							0.974822					
							1.05495					
92075				212	45	126.211	1.01603	-0.01114	-0.06812	1.23047	-1.05469	-0.35156
							0.94735	-0.02945	-0.07625	1.23047	-1.05469	0
							0.922166	-0.03759	-0.07829		-1.23047	
							1.00458	-0.01114	-0.05794		-1.05469	
							1.07327					
							1.00229					
							0.949639					
							0.94735					
92076	2	38	22	212	45	124.102	1.01145	-0.01725	-0.07218	1.23047	-1.05469	-0.35156
							1.08013	-0.03556	-0.07015	1.23047	-1.05469	-0.35156
							1.02519	-0.04166	-0.05998		-1.05469	
							0.958796	-0.02335	-0.04777		-1.05469	
							0.974822					
							0.958796					
							0.954217					
							0.990848					
92077				208	45	121.992	1.03663	-0.03963	-0.05184	1.23047	-1.05469	-0.35156
							0.997715	-0.04573	-0.05184	1.23047	-1.05469	0
							1.00687	-0.0559	-0.04777		-0.8789	
							1.01374	-0.0498	-0.04777		-0.8789	
							0.990848					
							0.983979					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.990848					
							0.997715					
92078				208	45	117.422	1.00229	-0.05183	-0.04777	1.23047	-1.05469	0
							0.995426	-0.06201	-0.05184	1.23047	-1.05469	0
							0.983979	-0.06811	-0.0498		-0.8789	
							1.00229	-0.07014	-0.05184		-1.05469	
							1.01374					
							0.988558					
							0.972533					
							1.00687					
92079				208	45	111.797	1.00229	-0.08032	-0.0498	1.23047	-1.05469	0
							0.979401	-0.08439	-0.05387	1.23047	-1.05469	0.351562
							0.977111	-0.08439	-0.04574		-1.05469	
							1.00458	-0.09456	-0.05591		-1.05469	
							1.01603					
							0.967954					
							0.974822					
							1.01832					
92080	2	38	26	208	45	104.062	1.02519	-0.09863	-0.05184	1.23047	-1.05469	0.351562
							0.990848	-0.1027	-0.05184	1.23047	-1.23047	0.703124
							0.965664	-0.10473	-0.05387		-1.23047	
							0.98169	-0.11084	-0.0498		-1.05469	
							0.995426					
							1.00916					
							0.997715					
							0.98169					
92081				208	45	97.0312	0.951928	-0.12101	-0.05387	1.23047	-1.05469	0.703124
							0.995426	-0.12101	-0.04167	1.23047	-1.05469	0.703124
							1.05495	-0.12915	-0.0437		-1.05469	
							0.990848	-0.1149	-0.0498		-0.8789	
							0.94506					
							0.98169					
							1.02519					
							1.02748					
92082				208	45	87.1875	0.983979	-0.12101	-0.04777	1.23047	-0.8789	0.703124
							0.977111	-0.12915	-0.05591	1.23047	-0.8789	0.351562
							1.01145	-0.12101	-0.03963		-0.8789	
							1.03663	-0.1149	-0.05387		-0.8789	
							0.988558					
							0.917587					
							0.974822					
							1.05266					
92083				208	45	79.4531	1.04808	-0.11694	-0.05184	1.23047	-0.8789	0.703124
							0.986269	-0.11897	-0.0437	1.23047	-0.8789	0.703124
							0.940481	-0.11897	-0.0498		-0.8789	
							0.963375	-0.12508	-0.04574		-0.70312	
							1.00458					
							1.02748					
							1.00687					
							0.974822					
92084	2	38	30	208	45	69.9609	0.993137	-0.12915	-0.04777	1.23047	-0.70312	1.05469
							1.00916	-0.1149	-0.04574	1.23047	-0.70312	1.05469

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.98169	-0.1149	-0.0498		-0.70312	
							0.979401	-0.12915	-0.0498		-0.70312	
							1.00229					
							1.01832					
							1.00916					
							0.98169					
92085				208	45	62.9297	0.98169	-0.11694	-0.04574	1.23047	-0.52734	0.703124
							0.990848	-0.10677	-0.04574	1.23047	-0.70312	0.703124
							0.997715	-0.11084	-0.05184		-0.70312	
							0.993137	-0.10677	-0.04777		-0.52734	
							0.972533					
							1.00916					
							1.0229					
							0.990848					
92086				208	45	54.4922	0.977111	-0.11084	-0.0498	1.23047	-0.52734	0.703124
							0.986269	-0.10473	-0.04777	1.23047	-0.52734	0.703124
							1.00229	-0.09863	-0.0498		-0.52734	
							0.997715	-0.09659	-0.0498		-0.52734	
							0.995426					
							0.990848					
							0.990848					
							1.00001					
92087				208	45	48.8672	0.995426	-0.09252	-0.04777	1.23047	-0.52734	0.703124
							0.974822	-0.08846	-0.0498	1.23047	-0.52734	1.05469
							0.988558	-0.09049	-0.0498		-0.35156	
							1.01603	-0.08439	-0.0498		-0.35156	
							1.00458					
							0.979401					
							0.977111					
							1.00229					
92088	2	38	34	208	45	43.2422	1.01832	-0.07625	-0.0498	1.23047	-0.35156	1.05469
							0.995426	-0.07421	-0.05184	1.23047	-0.35156	1.05469
							0.974822	-0.06201	-0.05184		-0.35156	
							0.993137	-0.05794	-0.05184		-0.35156	
							1.00458					
							0.988558					
							0.983979					
							1.00001					
92089				208	45	40.0781	1.01145	-0.05387	-0.04777	1.23047	-0.35156	1.05469
							0.98169	-0.05183	-0.05591	1.23047	-0.35156	1.05469
							0.983979	-0.05183	-0.05184		-0.35156	
							1.01832	-0.04166	-0.05184		-0.35156	
							1.00229					
							0.983979					
							0.995426					
							0.997715					
92090				208	45	38.3203	0.993137	-0.03352	-0.05387	1.23047	-0.35156	0.703124
							0.990848	-0.02539	-0.05794	1.23047	-0.35156	0.703124
							0.995426	-0.01928	-0.0498		-0.35156	
							0.990848	-0.02335	-0.0437		-0.35156	
							0.993137					
							0.990848					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.970243					
							0.979401					
92091				208	45	37.2656	1.01374	-0.02539	-0.04777	1.23047	-0.35156	0.703124
							1.00916	-0.01725	-0.04574	1.23047	-0.35156	0.703124
							0.979401	-0.00911	-0.0498		-0.35156	
							0.993137	-0.02132	-0.0498		-0.35156	
							1.01145					
							1.00458					
							1.00687					
							0.995426					
92092	2	38	38	208	45	37.2656	0.986269	-0.02132	-0.0498	1.23047	-0.35156	1.05469
							0.993137	-0.01318	-0.04574	1.23047	-0.35156	1.05469
							1.00001	-0.01521	-0.04777		-0.35156	
							0.979401	-0.02335	-0.04574		-0.35156	
							0.995426					
							1.00916					
							0.979401					
							0.986269					
92093				208	45	37.6172	1.00229	-0.01521	-0.04777	1.23047	-0.35156	1.05469
							1.00458	-0.01521	-0.05184	1.23047	-0.35156	1.05469
							1.00229	-0.02742	-0.0498		-0.35156	
							1.00229	-0.02335	-0.0498		-0.35156	
							0.993137					
							0.983979					
							0.990848					
							0.997715					
92094				208	45	37.6172	0.986269	-0.02335	-0.04777	1.23047	-0.35156	1.05469
							0.997715	-0.02539	-0.04574	1.23047	-0.35156	1.05469
							1.00229	-0.02132	-0.04777		-0.35156	
							0.983979	-0.02132	-0.0498		-0.35156	
							0.988558					
							1.00458					
							0.995426					
							0.995426					
92095				208	45	37.2656	1.00001	-0.02132	-0.04777	1.23047	-0.35156	0.703124
							0.993137	-0.01928	-0.04777	1.23047	-0.35156	0.703124
							0.988558	-0.01928	-0.0498		-0.35156	
							0.993137	-0.02539	-0.04777		-0.35156	
							1.00229					
							0.997715					
							0.997715					
							0.988558					
92096	2	38	42	208	45	37.2656	0.993137	-0.02335	-0.0498	1.23047	-0.35156	0.703124
							1.00229	-0.01521	-0.0498	1.23047	-0.35156	0.703124
							0.995426	-0.01725	-0.0498		-0.35156	
							0.986269	-0.01725	-0.05184		-0.35156	
							0.995426					
							1.00229					
							0.990848					
							0.993137					
92097				208	45	36.9141	1.00229	-0.01318	-0.05184	1.23047	-0.35156	0.703124
							0.995426	-0.00708	-0.05387	1.23047	-0.35156	0.351562

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.988558	-0.00708	-0.0498		-0.35156	
							0.993137	-0.00911	-0.0498		-0.35156	
							0.990848					
							0.993137					
							0.997715					
							0.990848					
92098				208	45	37.2656	0.997715	-0.00301	-0.0498	1.23047	-0.35156	0.703124
							0.997715	-0.00097	-0.0498	1.23047	-0.35156	0.703124
							0.979401	-0.00708	-0.04574		-0.35156	
							0.988558	-0.00911	-0.04574		-0.35156	
							1.00916					
							1.00229					
							0.98169					
							0.993137					
92099				208	45	38.3203	1.00229	-0.00708	-0.0498	1.23047	-0.35156	0.703124
							0.997715	-0.01114	-0.04777	1.23047	-0.35156	0.703124
							0.990848	-0.01318	-0.05184		-0.35156	
							0.993137	-0.00708	-0.0498		-0.35156	
							0.997715					
							1.00001					
							0.995426					
							0.993137					
92100	2	38	46	208	45	38.3203	0.988558	-0.00911	-0.0498	1.23047	-0.35156	0.703124
							0.993137	-0.00708	-0.0498	1.23047	-0.35156	0.351562
							0.993137	-0.00911	-0.0498		-0.35156	
							0.997715	-0.00911	-0.0498		-0.35156	
							1.00001					
							0.995426					
							0.997715					
							0.986269					
92101				208	45	38.6719	0.974822	-0.00708	-0.0498	1.23047	-0.35156	0.351562
							0.997715	-0.00708	-0.04777	1.23047	-0.35156	0.703124
							1.01374	-0.00301	-0.0498		-0.35156	
							0.990848	-0.01114	-0.0498		-0.35156	
							0.983979					
							1.00229					
							1.00229					
							0.993137					
92102				208	45	39.0234	0.995426	-0.01114	-0.0498	1.23047	-0.35156	0.703124
							0.993137	-0.00301	-0.0498	1.23047	-0.35156	0.703124
							0.997715	-0.00708	-0.04777		-0.35156	
							0.997715	-0.01114	-0.04777		-0.35156	
							0.979401					
							0.995426					
							1.00458					
							0.983979					
92103				204	45	39.375	0.997715	-0.00504	-0.04777	1.23047	-0.35156	0.703124
							1.00916	-0.00911	-0.0498	1.23047	-0.35156	0.703124
							0.983979	-0.00911	-0.04777		-0.35156	
							0.993137	-0.00301	-0.04574		-0.35156	
							0.993137					
							0.98169					



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.00001					
							1.00458					
92104	2	38	50	204	45	39.7266	0.983979	-0.00911	-0.04777	1.23047	-0.35156	0.703124
							1.00458	-0.01725	-0.04777	1.23047	-0.35156	0.703124
							1.01374	-0.01521	-0.04574		-0.35156	
							0.993137	-0.01725	-0.0498		-0.35156	
							0.993137					
							0.995426					
							0.990848					
							1.00229					
92105				204	45	40.0781	0.995426	-0.01725	-0.04574	1.23047	-0.35156	0.703124
							0.979401	-0.01521	-0.04777	1.23047	-0.35156	0.703124
							0.995426	-0.01725	-0.04777		-0.35156	
							1.00687	-0.01725	-0.05184		-0.35156	
							0.988558					
							0.983979					
							1.00229					
							1.00916					
92106				204	45	39.7266	0.997715	-0.01928	-0.0498	1.23047	-0.35156	0.703124
							0.993137	-0.02335	-0.04777	1.23047	-0.35156	0.703124
							0.986269	-0.01114	-0.0498		-0.35156	
							0.977111	-0.01521	-0.04777		-0.35156	
							0.986269					
							1.00458					
							1.00229					
							0.988558					
92107				204	45	39.7266	1.00001	-0.02132	-0.0498	1.23047	-0.35156	0.703124
							0.995426	-0.01114	-0.0498	1.23047	-0.35156	0.703124
							0.986269	-0.01725	-0.04777		-0.35156	
							0.997715	-0.02132	-0.04777		-0.35156	
							0.995426					
							0.986269					
							1.00458					
							0.997715					
92108	2	38	54	204	45	39.375	0.983979	-0.01928	-0.0498	1.23047	-0.35156	1.05469
							1.00687	-0.02742	-0.04777	1.23047	-0.35156	1.05469
							1.00687	-0.02132	-0.0437		-0.35156	
							0.98169	-0.02539	-0.05184		-0.52734	
							0.997715					
							0.993137					
							0.977111					
							1.01145					
92109				204	45	39.0234	1.01374	-0.02945	-0.0498	1.23047	-0.52734	1.05469
							0.977111	-0.02945	-0.0498	1.23047	-0.52734	0.703124
							0.974822	-0.01725	-0.04574		-0.52734	
							1.00458	-0.01521	-0.04777		-0.52734	
							1.00687					
							0.990848					
							0.986269					
							1.00687					
92110				208	45	39.0234	1.00687	-0.02335	-0.04574	1.23047	-0.52734	0.703124
							0.977111	-0.01928	-0.0498	1.23047	-0.35156	0.703124

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.993137	-0.01318	-0.04777		-0.35156	
							1.00458	-0.02335	-0.0498		-0.52734	
							0.995426					
							0.995426					
							0.993137					
							0.997715					
92111				204	45	39.0234	1.00229	-0.02132	-0.04574	1.23047	-0.35156	0.703124
							0.990848	-0.02132	-0.04777	1.23047	-0.35156	0.703124
							0.995426	-0.01725	-0.04777		-0.52734	
							0.997715	-0.02132	-0.0498		-0.52734	
							0.988558					
							0.997715					
							1.00001					
							0.988558					
92112	2	38	58	204	45	38.6719	1.00001	-0.01928	-0.0498	1.23047	-0.52734	0.703124
							1.00001	-0.01928	-0.04777	1.23047	-0.52734	0.703124
							0.983979	-0.02335	-0.0498		-0.52734	
							0.98169	-0.02335	-0.04574		-0.35156	
							0.997715					
							1.01145					
							1.00001					
							0.988558					
92113				204	45	38.6719	0.995426	-0.02539	-0.04574	1.23047	-0.35156	0.703124
							0.995426	-0.01928	-0.04777	1.23047	-0.35156	0.703124
							0.990848	-0.02132	-0.04777		-0.35156	
							1.00001	-0.02335	-0.0498		-0.35156	
							1.00001					
							0.986269					
							0.997715					
							1.00458					
92114				204	45	38.3203	1.00001	-0.02132	-0.0498	1.23047	-0.35156	0.703124
							0.983979	-0.01928	-0.05184	1.23047	-0.52734	0.703124
							0.986269	-0.01928	-0.0498		-0.52734	
							1.00458	-0.02132	-0.0498		-0.52734	
							0.997715					
							0.983979					
							0.988558					
							0.995426					
92115				204	45	37.9688	0.995426	-0.01725	-0.0498	1.23047	-0.52734	0.703124
							0.995426	-0.02132	-0.0498	1.23047	-0.52734	1.05469
							0.993137	-0.02539	-0.0498		-0.52734	
							1.00687	-0.02132	-0.0498		-0.52734	
							0.995426					
							0.986269					
							0.983979					
							0.995426					
92116	2	39	2	204	45	37.9688	1.00687	-0.01521	-0.0498	1.23047	-0.52734	0.703124
							1.00229	-0.01318	-0.0498	1.23047	-0.52734	0.703124
							0.986269	-0.01318	-0.04777		-0.52734	
							0.990848	-0.01114	-0.0498		-0.52734	
							0.995426					
							0.988558					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.997715					
							1.00001					
92117				204	45	38.3203	0.993137	-0.01521	-0.04777	1.23047	-0.52734	1.05469
							0.990848	-0.01318	-0.04777	1.23047	-0.52734	1.05469
							0.990848	-0.01521	-0.04777		-0.52734	
							1.00001	-0.01928	-0.04574		-0.52734	
							1.00229					
							0.997715					
							0.986269					
							0.983979					
92118				204	45	38.6719	0.993137	-0.01521	-0.04574	1.23047	-0.52734	1.05469
							0.997715	-0.01928	-0.04777	1.23047	-0.52734	0.703124
							1.00001	-0.01521	-0.0498		-0.52734	
							1.00001	-0.01725	-0.04777		-0.52734	
							0.997715					
							0.990848					
							0.990848					
							0.997715					
92119				204	45	38.6719	0.997715	-0.02335	-0.0498	1.23047	-0.52734	0.703124
							1.00001	-0.01725	-0.0498	1.23047	-0.52734	0.703124
							0.995426	-0.01318	-0.05184		-0.52734	
							0.990848	-0.01725	-0.04777		-0.52734	
							0.988558					
							1.00001					
							0.997715					
							0.979401					
92120	2	39	6	204	45	38.6719	0.98169	-0.01521	-0.04777	1.23047	-0.52734	0.703124
							0.997715	-0.00911	-0.04777	1.23047	-0.52734	0.703124
							1.00458	-0.01114	-0.05387		-0.52734	
							0.990848	-0.01318	-0.04777		-0.52734	
							0.988558					
							1.01374					
							0.997715					
							0.974822					
92121				204	45	39.0234	0.983979	-0.00708	-0.0498	1.23047	-0.52734	0.703124
							1.00916	-0.01114	-0.0498	1.23047	-0.52734	0.703124
							1.00458	-0.00708	-0.04777		-0.52734	
							0.988558	-0.01114	-0.04777		-0.52734	
							0.988558					
							0.988558					
							0.993137					
							1.00687					
92122				204	45	39.375	1.00458	-0.01928	-0.0498	1.23047	-0.52734	1.05469
							1.00001	-0.01928	-0.04777	1.23047	-0.52734	0.703124
							0.979401	-0.01114	-0.0498		-0.52734	
							0.977111	-0.01318	-0.04777		-0.52734	
							1.01374					
							1.01832					
							0.98169					
							0.967954					
92123				204	45	39.375	0.997715	-0.01318	-0.05184	1.23047	-0.52734	0.703124
							1.01603	-0.01318	-0.04574	1.23047	-0.52734	0.703124

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.990848	-0.01521	-0.0498		-0.52734	
							0.974822	-0.01725	-0.04574		-0.52734	
							0.993137					
							1.01374					
							1.00229					
							0.979401					
92124	2	39	10	204	45	39.375	0.990848	-0.01318	-0.05184	1.23047	-0.52734	0.703124
							1.01374	-0.01725	-0.04574	1.23047	-0.52734	0.703124
							1.00687	-0.02132	-0.0498		-0.52734	
							0.98169	-0.01725	-0.04777		-0.52734	
							0.979401					
							1.00229					
							1.00001					
							0.988558					
92125				204	45	39.7266	0.993137	-0.01928	-0.0498	1.23047	-0.52734	1.05469
							1.00229	-0.01928	-0.0498	1.23047	-0.52734	1.05469
							1.00687	-0.01725	-0.0498		-0.52734	
							0.983979	-0.01725	-0.05184		-0.52734	
							0.983979					
							1.00458					
							1.00229					
							0.995426					
92126				204	45	39.7266	0.979401	-0.01725	-0.0498	1.23047	-0.52734	0.703124
							0.986269	-0.01318	-0.0498	1.23047	-0.52734	0.703124
							1.00001	-0.01318	-0.04777		-0.52734	
							0.995426	-0.01318	-0.04574		-0.52734	
							0.995426					
							0.995426					
							0.995426					
							0.983979					
92127				204	45	39.7266	0.993137	-0.01521	-0.0498	1.23047	-0.52734	0.703124
							1.00687	-0.01725	-0.04574	1.23047	-0.52734	0.703124
							0.990848	-0.02132	-0.0498		-0.52734	
							0.98169	-0.02539	-0.04574		-0.52734	
							1.00229					
							1.00916					
							0.988558					
							0.974822					
92128	2	39	14	204	45	39.7266	1.00229	-0.01725	-0.05387	1.23047	-0.52734	1.05469
							1.02748	-0.02132	-0.04777	1.23047	-0.52734	1.05469
							1.01145	-0.02945	-0.05387		-0.70312	
							0.970243	-0.01928	-0.0498		-0.70312	
							0.956507					
							0.997715					
							1.01832					
							0.997715					
92129				204	45	39.375	0.970243	-0.02335	-0.04777	1.23047	-0.70312	1.05469
							0.98169	-0.02742	-0.04777	1.23047	-0.70312	1.05469
							1.00916	-0.02539	-0.04777		-0.52734	
							1.00229	-0.01928	-0.0498		-0.52734	
							0.979401					
							0.983979					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.01374					
							1.01145					
92130				200	45	39.375	1.00001	-0.02335	-0.04777	1.23047	-0.52734	1.05469
							0.988558	-0.03149	-0.0498	1.23047	-0.52734	0.703124
							0.983979	-0.02539	-0.04777		-0.52734	
							0.997715	-0.02335	-0.0498		-0.52734	
							0.988558					
							0.988558					
							0.997715					
							1.00458					
92131				204	45	39.0234	0.997715	-0.02335	-0.04777	1.23047	-0.52734	1.05469
							0.979401	-0.02132	-0.05184	1.23047	-0.52734	1.05469
							0.995426	-0.02539	-0.04574		-0.52734	
							1.01145	-0.02539	-0.05387		-0.52734	
							0.995426					
							0.972533					
							0.990848					
							1.02061					
92132	2	39	18	200	45	38.6719	1.00229	-0.02742	-0.04777	1.23047	-0.52734	1.05469
							0.974822	-0.02742	-0.05184	1.23047	-0.52734	0.703124
							0.977111	-0.02335	-0.0498		-0.70312	
							1.00916	-0.02132	-0.05184		-0.70312	
							1.00458					
							0.98169					
							0.986269					
							1.00458					
92133				200	45	38.3203	1.00229	-0.02335	-0.04777	1.23047	-0.70312	0.703124
							0.979401	-0.01928	-0.05184	1.23047	-0.70312	0.703124
							0.98169	-0.01928	-0.04777		-0.70312	
							1.00687	-0.02132	-0.04777		-0.70312	
							1.01145					
							0.979401					
							0.970243					
							1.00229					
92134				200	45	38.3203	1.01374	-0.02132	-0.0498	1.23047	-0.70312	1.05469
							1.00458	-0.02132	-0.0498	1.23047	-0.70312	1.05469
							0.983979	-0.01928	-0.05387		-0.70312	
							0.990848	-0.02132	-0.0498		-0.70312	
							1.00916					
							1.00916					
							0.986269					
							0.979401					
92135				200	45	38.3203	0.986269	-0.01928	-0.05184	1.23047	-0.70312	0.703124
							0.997715	-0.01725	-0.0498	1.23047	-0.70312	0.703124
							1.00916	-0.01928	-0.05184		-0.70312	
							0.993137	-0.02132	-0.05184		-0.70312	
							0.983979					
							1.00001					
							1.00229					
							0.988558					
92136	2	39	22	200	45	38.3203	0.988558	-0.01725	-0.05387	1.23047	-0.70312	0.703124
							0.997715	-0.01928	-0.05184	1.23047	-0.70312	0.703124

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.00229	-0.02132	-0.05387		-0.70312	
							0.983979	-0.01928	-0.05184		-0.70312	
							0.98169					
							1.00001					
							1.01374					
							0.995426					
92137				200	45	37.9688	0.986269	-0.02132	-0.05184	1.23047	-0.70312	0.703124
							0.986269	-0.01725	-0.05387	1.23047	-0.70312	1.05469
							0.997715	-0.02132	-0.05184		-0.70312	
							1.00687	-0.02132	-0.05387		-0.70312	
							0.995426					
							0.986269					
							0.986269					
							0.990848					
92138				200	45	37.9688	1.00458	-0.01114	-0.05387	1.23047	-0.70312	0.703124
							1.00001	-0.00911	-0.05387	1.23047	-0.70312	0.703124
							0.979401	-0.01521	-0.05387		-0.70312	
							0.986269	-0.00911	-0.05387		-0.70312	
							0.995426					
							0.993137					
							0.995426					
							0.993137					
92139				200	45	38.3203	0.993137	-0.01318	-0.05387	1.23047	-0.70312	1.05469
							0.993137	-0.01318	-0.05184	1.23047	-0.70312	1.05469
							0.990848	-0.01318	-0.05184		-0.70312	
							0.993137	-0.02335	-0.05184		-0.70312	
							1.00229					
							1.00229					
							1.00458					
							0.993137					
92140	2	39	26	200	45	38.6719	0.979401	-0.02335	-0.0498	1.23047	-0.70312	1.05469
							0.983979	-0.01725	-0.05387	1.23047	-0.70312	1.40625
							1.00916	-0.02539	-0.0498		-0.70312	
							1.01832	-0.02742	-0.05184		-0.70312	
							0.993137					
							0.974822					
							0.98169					
							1.00458					
92141				200	45	38.6719	1.00687	-0.01725	-0.0498	1.23047	-0.70312	1.05469
							0.995426	-0.02945	-0.05184	1.23047	-0.70312	1.05469
							0.986269	-0.02945	-0.05387		-0.70312	
							0.990848	-0.02539	-0.05387		-0.70312	
							1.00458					
							1.00229					
							0.993137					
							1.00001					
92142				200	45	37.9688	1.00458	-0.02539	-0.05591	1.23047	-0.70312	1.05469
							0.993137	-0.02539	-0.05184	1.23047	-0.70312	0.703124
							0.972533	-0.01725	-0.05387		-0.70312	
							0.965664	-0.02335	-0.05184		-0.70312	
							0.993137					
							1.01145					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.00458					
							0.983979					
92143				196	45	37.9688	0.990848	-0.03149	-0.05184	1.23047	-0.70312	1.05469
							1.00001	-0.02335	-0.0498	1.23047	-0.70312	1.05469
							1.00001	-0.02335	-0.05184		-0.70312	
							0.993137	-0.02539	-0.05387		-0.70312	
							0.986269					
							0.995426					
							1.00687					
							1.00001					
92144	2	39	30	196	45	37.9688	0.993137	-0.01928	-0.05387	1.23047	-0.70312	0.703124
							0.993137	-0.01725	-0.05184	1.23047	-0.70312	0.703124
							1.00001	-0.01521	-0.05184		-0.70312	
							0.997715	-0.01521	-0.05387		-0.70312	
							0.983979					
							0.977111					
							0.993137					
							1.00687					
92145				196	45	37.9688	0.997715	-0.00911	-0.05184	1.23047	-0.70312	0.703124
							0.988558	-0.01318	-0.05387	1.23047	-0.70312	0.703124
							0.988558	-0.01725	-0.05184		-0.70312	
							1.00687	-0.02132	-0.05387		-0.70312	
							1.00229					
							0.990848					
							0.993137					
							0.990848					
92146				196	45	37.9688	0.990848	-0.01928	-0.05184	1.23047	-0.70312	0.703124
							0.993137	-0.01521	-0.05184	1.23047	-0.70312	0.703124
							0.986269	-0.01725	-0.05184		-0.8789	
							0.990848	-0.02335	-0.05184		-0.70312	
							1.00458					
							0.993137					
							0.974822					
							0.986269					
92147				196	45	37.9688	1.01603	-0.02132	-0.05184	1.23047	-0.8789	0.703124
							1.01603	-0.02742	-0.0498	1.23047	-0.70312	0.703124
							0.98169	-0.02945	-0.05387		-0.70312	
							0.98169	-0.02335	-0.0498		-0.70312	
							1.00916					
							1.00687					
							0.977111					
							0.979401					
92148	2	39	34	196	45	37.2656	1.00458	-0.02335	-0.05387	1.23047	-0.70312	0.703124
							1.01374	-0.02945	-0.05184	1.23047	-0.70312	0.703124
							1.00001	-0.02539	-0.05184		-0.70312	
							0.979401	-0.02335	-0.05184		-0.70312	
							0.983979					
							1.00001					
							1.00687					
							0.993137					
92149				196	45	37.2656	0.974822	-0.02132	-0.04777	1.23047	-0.70312	1.05469
							0.983979	-0.02335	-0.05387	1.23047	-0.70312	1.05469

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.02061	-0.02335	-0.05184		-0.70312	
							1.01832	-0.01318	-0.05387		-0.52734	
							0.990848					
							0.974822					
							0.995426					
							1.01603					
92150				196	45	37.2656	1.00001	-0.01521	-0.05184	1.23047	-0.52734	1.05469
							0.983979	-0.01928	-0.05591	1.23047	-0.52734	1.05469
							1.00001	-0.01318	-0.05387		-0.52734	
							1.00916	-0.00911	-0.05387		-0.70312	
							0.993137					
							0.983979					
							0.993137					
							0.990848					
92151				196	45	37.6172	0.98169	-0.00097	-0.05184	1.23047	-0.70312	0.703124
							0.979401	0.001057	-0.0498	1.23047	-0.70312	0.703124
							0.993137	0.001057	-0.05387		-0.70312	
							0.995426	-0.00504	-0.05184		-0.52734	
							1.00229					
							1.01145					
							1.00001					
							0.983979					
92152	2	39	38	196	45	38.6719	0.997715	-0.00301	-0.05387	1.23047	-0.52734	0.703124
							1.00458	-0.00097	-0.0498	1.23047	-0.52734	0.703124
							0.986269	-0.00097	-0.05184		-0.52734	
							0.979401	-0.00708	-0.0498		-0.52734	
							0.995426					
							1.00001					
							1.00229					
							0.997715					
92153				196	45	39.375	0.997715	-0.00911	-0.04777	1.23047	-0.52734	0.703124
							0.986269	-0.00911	-0.05184	1.23047	-0.52734	0.703124
							0.979401	-0.01114	-0.05184		-0.52734	
							1.00687	-0.01521	-0.05387		-0.52734	
							1.01832					
							0.990848					
							0.98169					
							1.00001					
92154				196	45	39.375	1.00916	-0.01725	-0.0498	1.23047	-0.52734	0.703124
							0.986269	-0.01521	-0.0498	1.23047	-0.52734	0.703124
							0.974822	-0.02335	-0.05387		-0.35156	
							0.995426	-0.01725	-0.05184		-0.35156	
							1.01374					
							1.00458					
							0.983979					
							0.983979					
92155				192	45	39.375	1.00458	-0.01114	-0.05184	1.23047	-0.35156	0.703124
							1.00687	-0.01725	-0.05184	1.23047	-0.35156	0.703124
							0.995426	-0.01114	-0.05184		-0.35156	
							0.977111	-0.01114	-0.0498		-0.52734	
							0.988558					
							1.00001					



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.993137					
							0.983979					
92156	2	39	42	192	45	39.7266	0.993137	-0.01928	-0.05184	1.23047	-0.52734	0.703124
							0.997715	-0.01521	-0.05184	1.23047	-0.52734	0.703124
							0.997715	-0.01318	-0.05387		-0.35156	
							0.990848	-0.01725	-0.0498		-0.35156	
							0.997715					
							1.00458					
							0.997715					
							0.974822					
92157				196	45	39.7266	0.983979	-0.01318	-0.05184	1.23047	-0.52734	0.703124
							1.00229	-0.01521	-0.05184	1.23047	-0.52734	0.703124
							1.01145	-0.01521	-0.0498		-0.35156	
							1.00458	-0.01725	-0.05184		-0.35156	
							0.979401					
							0.988558					
							1.02061					
							1.00687					
92158				192	45	39.7266	0.970243	-0.01725	-0.05184	1.23047	-0.35156	0.703124
							0.979401	-0.01318	-0.05387	1.23047	-0.35156	0.703124
							1.00916	-0.01725	-0.05184		-0.35156	
							1.01603	-0.01521	-0.05387		-0.35156	
							0.98169					
							0.979401					
							1.00458					
							1.00458					
92159				192	45	39.7266	0.988558	-0.01521	-0.05387	1.23047	-0.35156	0.703124
							0.990848	-0.01928	-0.05387	1.05469	-0.35156	0.703124
							1.00001	-0.01521	-0.05387		-0.35156	
							0.995426	-0.02132	-0.05387		-0.52734	
							0.977111					
							0.995426					
							1.00458					
							0.988558					
92160	2	39	46	192	45	39.7266	0.977111	-0.02132	-0.05591	1.05469	-0.35156	0.703124
							0.990848	-0.01725	-0.05591	1.23047	-0.35156	1.05469
							1.01374	-0.02742	-0.05387		-0.35156	
							1.01832	-0.02945	-0.05794		-0.35156	
							0.993137					
							0.979401					
							0.986269					
							1.00458					
92161				192	45	39.375	1.00001	-0.02335	-0.05591	1.05469	-0.35156	1.05469
							0.98169	-0.03149	-0.05998	1.23047	-0.52734	1.05469
							0.995426	-0.02539	-0.06201		-0.52734	
							0.997715	-0.02335	-0.06201		-0.35156	
							0.993137					
							0.995426					
							0.990848					
							0.988558					
92162				192	45	39.375	0.995426	-0.01928	-0.06405	1.23047	-0.35156	1.05469
							0.990848	-0.01725	-0.06608	1.23047	-0.35156	0.703124

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.988558	-0.01318	-0.06608		-0.35156	
							0.990848	-0.01114	-0.06405		-0.35156	
							0.995426					
							1.00001					
							1.00001					
							0.995426					
92163				192	45	39.7266	0.983979	-0.01521	-0.06405	1.23047	-0.52734	1.05469
							0.986269	-0.01521	-0.06405	1.05469	-0.35156	1.05469
							0.997715	-0.01521	-0.05794		-0.35156	
							1.00458	-0.01928	-0.05794		-0.35156	
							0.990848					
							0.98169					
							1.00687					
							1.00687					
92164	2	39	50	192	45	39.7266	0.988558	-0.02132	-0.05998	1.23047	-0.35156	1.05469
							0.98169	-0.01928	-0.05998	1.05469	-0.35156	1.05469
							0.995426	-0.02132	-0.05591		-0.35156	
							1.00001	-0.02132	-0.05998		-0.52734	
							0.988558					
							0.979401					
							1.00458					
							1.01603					
92165				192	45	39.375	0.995426	-0.02539	-0.05591	1.05469	-0.52734	1.05469
							0.965664	-0.02539	-0.05998	1.23047	-0.52734	1.05469
							0.972533	-0.01725	-0.06201		-0.52734	
							0.997715	-0.02132	-0.06405		-0.52734	
							1.01374					
							1.00687					
							0.98169					
							0.986269					
92166				192	45	39.375	1.00458	-0.02742	-0.06812	1.05469	-0.35156	1.05469
							1.00458	-0.02132	-0.07015	1.23047	-0.52734	1.05469
							1.00001	-0.01725	-0.06608		-0.52734	
							0.997715	-0.01928	-0.06608		-0.52734	
							0.986269					
							0.979401					
							0.995426					
							1.00229					
92167				192	45	39.375	0.986269	-0.01928	-0.06608	1.23047	-0.52734	1.05469
							0.98169	-0.01928	-0.06608	1.05469	-0.52734	0.703124
							0.997715	-0.02335	-0.06608		-0.52734	
							1.00001	-0.01928	-0.06608		-0.35156	
							1.00916					
							0.993137					
							0.977111					
							0.98169					
92168	2	39	54	192	45	39.375	1.01145	-0.00504	-0.07218	1.05469	-0.35156	0.703124
							1.02061	-0.01114	-0.06812	1.23047	-0.35156	0.703124
							1.00001	-0.01114	-0.07218		-0.35156	
							0.967954	-0.01521	-0.07015		-0.35156	
							0.972533					
							1.01145					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.02977					
							0.995426					
92169				192	45	39.375	0.961086	-0.02539	-0.07015	1.23047	-0.35156	1.05469
							0.977111	-0.01521	-0.06812	1.05469	-0.35156	0.703124
							1.00916	-0.01725	-0.06812		-0.35156	
							1.00687	-0.01928	-0.06405		-0.35156	
							0.98169					
							0.986269					
							1.01603					
							0.990848					
92170				192	45	39.7266	0.967954	-0.01725	-0.05998	1.05469	-0.35156	0.703124
							1.00001	-0.01725	-0.05387	1.05469	-0.35156	0.703124
							1.02061	-0.02335	-0.05591		-0.35156	
							0.997715	-0.02742	-0.05794		-0.35156	
							0.983979					
							0.986269					
							0.997715					
							1.00229					
92171				192	45	39.375	1.00458	-0.02539	-0.05998	1.23047	-0.35156	0.703124
							0.993137	-0.02132	-0.05794	1.05469	-0.35156	1.05469
							0.990848	-0.02539	-0.05387		-0.35156	
							0.986269	-0.03556	-0.05998		-0.35156	
							0.977111					
							0.972533					
							1.00229					
							1.01832					
92172	2	39	58	192	45	39.0234	1.00001	-0.03759	-0.05998	1.23047	-0.17578	0.703124
							0.979401	-0.03149	-0.07015	1.05469	-0.17578	0.703124
							0.990848	-0.02335	-0.07625		-0.17578	
							1.02519	-0.02335	-0.07218		-0.35156	
							1.0229					
							0.997715					
							0.98169					
							0.970243					
92173				192	45	38.3203	0.983979	-0.02945	-0.06608	1.23047	-0.35156	1.05469
							0.979401	-0.02539	-0.06608	1.23047	-0.35156	0.703124
							0.979401	-0.01318	-0.06608		-0.35156	
							1.00001	-0.01521	-0.06405		-0.17578	
							1.00916					
							1.00229					
							0.990848					
							0.988558					
92174				192	45	38.3203	0.997715	-0.02132	-0.06812	1.23047	-0.35156	1.05469
							1.00458	-0.02132	-0.06812	1.05469	-0.35156	1.05469
							0.983979	-0.02539	-0.07218		-0.35156	
							0.977111	-0.02945	-0.06608		-0.52734	
							1.00001					
							1.00687					
							0.993137					
							0.965664					
92175				192	45	38.3203	0.979401	-0.02742	-0.07015	1.23047	-0.52734	1.05469
							1.00916	-0.02132	-0.07015	1.05469	-0.52734	0.703124

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.00458	-0.01318	-0.07218		-0.52734	
							0.988558	-0.00911	-0.07422		-0.52734	
							0.979401					
							0.993137					
							1.00458					
							0.986269					
92176	2	40	2	192	45	38.3203	0.979401	-0.00911	-0.07422	1.05469	-0.52734	0.703124
							0.990848	-0.00911	-0.07422	1.23047	-0.52734	0.703124
							1.01832	-0.00911	-0.07422		-0.52734	
							1.00001	-0.00504	-0.07218		-0.52734	
							0.977111					
							0.98169					
							1.00001					
							1.00458					
92177				192	45	39.0234	0.98169	-0.01725	-0.07218	1.23047	-0.52734	0.703124
							0.990848	-0.01928	-0.07422	1.05469	-0.35156	0.703124
							1.00916	-0.01725	-0.07218		-0.35156	
							1.00916	-0.01521	-0.07422		-0.35156	
							0.986269					
							0.977111					
							0.993137					
							1.00687					
92178				192	45	39.0234	1.00458	-0.01521	-0.07218	1.05469	-0.35156	0.703124
							0.990848	-0.02539	-0.07422	1.23047	-0.35156	0.703124
							0.995426	-0.02132	-0.07422		-0.35156	
							0.995426	-0.01725	-0.07625		-0.35156	
							0.990848					
							0.986269					
							1.00229					
							1.01374					
92179				192	45	38.6719	0.986269	-0.02742	-0.06812	1.05469	-0.35156	0.703124
							0.963375	-0.02335	-0.07015	1.23047	-0.52734	0.351562
							0.974822	-0.01114	-0.07422		-0.52734	
							0.995426	-0.01318	-0.07015		-0.35156	
							0.993137					
							0.997715					
							1.00001					
							0.98169					
92180	2	40	6	192	45	38.3203	0.979401	-0.01318	-0.07015	1.05469	-0.52734	0.351562
							0.997715	-0.01928	-0.07625	1.23047	-0.52734	0.703124
							1.01603	-0.02335	-0.07625		-0.52734	
							1.00458	-0.00708	-0.07422		-0.35156	
							0.977111					
							0.974822					
							1.01603					
							1.00458					
92181				188	45	38.3203	0.972533	-0.02132	-0.07422	1.23047	-0.52734	1.05469
							0.983979	-0.02945	-0.07625	1.05469	-0.35156	0.703124
							1.00458	-0.01725	-0.07625		-0.35156	
							1.01145	-0.02132	-0.07218		-0.35156	
							0.983979					
							0.988558					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.01145					
							1.01374					
92182				192	45	38.3203	0.995426	-0.02335	-0.06405	1.05469	-0.35156	0.703124
							0.972533	-0.00911	-0.06405	1.23047	-0.35156	0.351562
							0.988558	-0.00301	-0.05794		-0.52734	
							1.01145	-0.00504	-0.05794		-0.52734	
							1.00001					
							0.974822					
							0.972533					
							0.997715					
92183				192	45	38.6719	0.993137	-0.00504	-0.05591	1.23047	-0.52734	0.351562
							0.990848	-0.01318	-0.05387	1.05469	-0.52734	0.703124
							0.997715	-0.01521	-0.05591		-0.52734	
							0.995426	-0.00911	-0.05387		-0.52734	
							0.995426					
							0.995426					
							0.993137					
							0.986269					
92184	2	40	10	192	45	38.6719	0.995426	-0.00504	-0.05794	1.23047	-0.52734	0.351562
							0.997715	-0.01114	-0.05794	1.05469	-0.52734	0.703124
							1.00001	-0.02132	-0.05387		-0.52734	
							1.00458	-0.02335	-0.05184		-0.52734	
							0.995426					
							0.988558					
							0.979401					
							0.995426					
92185				188	45	38.6719	1.00458	-0.01725	-0.05387	1.05469	-0.52734	0.703124
							0.995426	-0.01725	-0.0498	1.23047	-0.52734	0.703124
							0.997715	-0.01521	-0.05591		-0.35156	
							0.993137	-0.01521	-0.0498		-0.35156	
							0.990848					
							1.00001					
							1.00916					
							1.00229					
92186				192	45	38.3203	0.98169	-0.01928	-0.05387	1.05469	-0.35156	0.703124
							0.997715	-0.01318	-0.05184	1.23047	-0.35156	0.703124
							1.00458	-0.00911	-0.05184		-0.35156	
							0.990848	-0.01114	-0.0498		-0.35156	
							0.983979					
							0.995426					
							1.00458					
							0.988558					
92187				192	45	38.6719	0.986269	-0.00911	-0.0498	1.23047	-0.35156	0.703124
							0.993137	-0.00708	-0.05184	1.23047	-0.35156	0.703124
							0.997715	-0.01725	-0.05184		-0.35156	
							1.00458	-0.01928	-0.05184		-0.35156	
							1.00001					
							0.997715					
							0.990848					
							0.993137					
92188	2	40	14	188	45	38.6719	0.995426	-0.01521	-0.05387	1.23047	-0.35156	0.703124
							0.997715	-0.01521	-0.0498	1.23047	-0.35156	0.703124

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.988558	-0.01114	-0.05184		-0.35156	
							0.979401	-0.01114	-0.0498		-0.35156	
							0.993137					
							1.00001					
							1.00001					
							0.990848					
92189				192	45	39.0234	0.993137	-0.01928	-0.0498	1.05469	-0.35156	1.05469
							1.00687	-0.01928	-0.05387	1.23047	-0.35156	0.703124
							1.00001	-0.01521	-0.05184		-0.35156	
							0.997715	-0.01114	-0.05184		-0.35156	
							0.995426					
							0.993137					
							0.990848					
							0.986269					
92190				188	45	39.0234	0.988558	-0.00911	-0.05184	1.23047	-0.52734	0.703124
							0.995426	-0.00911	-0.05387	1.05469	-0.52734	0.703124
							1.00229	-0.01318	-0.0498		-0.52734	
							0.995426	-0.01521	-0.05184		-0.52734	
							0.98169					
							0.983979					
							0.995426					
							0.997715					
92191				188	45	39.375	0.986269	-0.02132	-0.0498	1.05469	-0.52734	0.703124
							0.990848	-0.01928	-0.05184	1.23047	-0.52734	0.703124
							1.00229	-0.01521	-0.05387		-0.52734	
							1.00229	-0.02132	-0.0498		-0.52734	
							0.988558					
							0.997715					
							1.01374					
							0.995426					
92192	2	40	18	188	45	39.0234	0.979401	-0.02132	-0.0498	1.05469	-0.52734	0.703124
							0.990848	-0.01521	-0.0498	1.23047	-0.35156	0.703124
							1.00687	-0.01521	-0.04777		-0.35156	
							0.997715	-0.01318	-0.05184		-0.35156	
							0.98169					
							0.986269					
							1.00229					
							1.01832					
92193				188	45	39.375	1.01145	-0.01318	-0.05184	1.23047	-0.35156	0.703124
							0.995426	-0.01928	-0.05184	1.23047	-0.17578	0.703124
							0.983979	-0.01521	-0.05184		-0.17578	
							0.986269	-0.00708	-0.05387		-0.35156	
							1.00001					
							1.00001					
							0.993137					
							0.990848					
92194				188	45	39.375	0.997715	-0.01521	-0.05184	1.05469	-0.35156	0.703124
							0.993137	-0.01521	-0.05184	1.23047	-0.35156	0.703124
							0.983979	-0.01114	-0.05184		-0.52734	
							0.988558	-0.01114	-0.05184		-0.52734	
							0.995426					
							0.993137					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.979401					
							0.986269					
92195				188	45	39.7266	1.00687	-0.00911	-0.05387	1.23047	-0.35156	0.703124
							1.00458	-0.00911	-0.05794	1.23047	-0.35156	0.703124
							0.98169	-0.00911	-0.06405		-0.35156	
							0.98169	-0.00911	-0.07015		-0.52734	
							1.00229					
							1.01603					
							1.00229					
							0.990848					
92196	2	40	22	188	45	40.0781	1.00687	-0.01521	-0.07015	1.23047	-0.52734	0.703124
							1.00229	-0.01725	-0.06405	1.23047	-0.52734	0.703124
							0.974822	-0.01521	-0.06812		-0.52734	
							0.963375	-0.01521	-0.06812		-0.52734	
							0.977111					
							1.00687					
							1.01145					
							1.00001					
92197				188	45	40.0781	0.990848	-0.01725	-0.07218	1.23047	-0.52734	0.703124
							0.995426	-0.01318	-0.07625	1.23047	-0.52734	0.703124
							0.990848	-0.00911	-0.07625		-0.52734	
							0.990848	-0.01114	-0.07218		-0.52734	
							0.997715					
							0.986269					
							0.983979					
							0.993137					
92198				188	45	40.0781	0.995426	-0.01114	-0.07015	1.23047	-0.52734	0.703124
							1.00001	-0.01928	-0.06608	1.23047	-0.52734	0.703124
							1.00687	-0.02539	-0.07015		-0.52734	
							1.00001	-0.02132	-0.07625		-0.52734	
							0.972533					
							0.986269					
							1.01374					
							1.00458					
92199				188	45	40.0781	0.967954	-0.01318	-0.07625	1.23047	-0.52734	0.351562
							0.977111	-0.01725	-0.07829	1.23047	-0.52734	0.351562
							1.01145	-0.01114	-0.07625		-0.52734	
							1.00229	-0.01521	-0.08032		-0.52734	
							0.983979					
							0.986269					
							1.00001					
							1.00458					
92200	2	40	26	188	45	39.7266	0.983979	-0.00911	-0.08439	1.23047	-0.52734	0.351562
							0.98169	-0.01114	-0.09456	1.05469	-0.52734	0.351562
							1.00229	-0.01725	-0.0966		-0.52734	
							1.00687	-0.01928	-0.09456		-0.52734	
							0.995426					
							0.986269					
							0.990848					
							0.993137					
92201				188	45	39.375	0.990848	-0.01521	-0.09456	1.23047	-0.35156	0.703124
							1.00001	-0.02132	-0.09253	1.23047	-0.35156	0.703124

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.993137	-0.02335	-0.0966		-0.35156	
							0.990848	-0.02132	-0.09863		-0.52734	
							0.995426					
							0.995426					
							0.997715					
							0.990848					
92202				188	45	39.0234	0.98169	-0.01725	-0.09863	1.23047	-0.52734	0.703124
							0.983979	-0.01318	-0.10474	1.23047	-0.52734	0.703124
							0.988558	-0.01928	-0.10677		-0.52734	
							1.00001	-0.01928	-0.10677		-0.52734	
							1.00001					
							0.995426					
							0.972533					
							0.98169					
92203				188	45	39.0234	1.00458	-0.01114	-0.10474	1.23047	-0.70312	0.703124
							0.995426	-0.01725	-0.10677	1.23047	-0.52734	0.703124
							0.98169	-0.02132	-0.1027		-0.52734	
							0.988558	-0.01725	-0.1027		-0.52734	
							1.00229					
							0.983979					
							0.986269					
							1.00001					
92204	2	40	30	188	45	39.0234	1.00458	-0.01725	-0.10067	1.23047	-0.52734	0.703124
							0.990848	-0.01725	-0.09049	1.23047	-0.52734	0.703124
							0.979401	-0.01521	-0.07422		-0.52734	
							0.995426	-0.02132	-0.07015		-0.52734	
							1.00687					
							0.997715					
							0.990848					
							0.997715					
92205				188	45	39.0234	1.00001	-0.02539	-0.06812	1.23047	-0.52734	1.05469
							0.986269	-0.02335	-0.06812	1.23047	-0.52734	0.703124
							0.977111	-0.01114	-0.07015		-0.52734	
							0.993137	-0.01928	-0.07422		-0.52734	
							1.00458					
							0.993137					
							0.988558					
							0.995426					
92206				188	45	39.0234	0.995426	-0.02132	-0.07829	1.23047	-0.52734	0.703124
							0.983979	-0.01521	-0.07625	1.23047	-0.52734	0.703124
							0.986269	-0.01725	-0.07625		-0.52734	
							0.995426	-0.02132	-0.07625		-0.52734	
							1.00229					
							0.993137					
							0.986269					
							0.993137					
92207				188	45	39.375	1.00916	-0.01725	-0.06405	1.23047	-0.52734	0.703124
							0.995426	-0.01521	-0.05591	1.23047	-0.52734	0.703124
							0.986269	-0.01725	-0.05794		-0.52734	
							0.988558	-0.02132	-0.05794		-0.52734	
							0.995426					
							1.00001					



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.997715					
							0.988558					
92208	2	40	34	188	45	39.0234	0.993137	-0.02132	-0.06201	1.23047	-0.52734	0.703124
							1.00687	-0.02132	-0.06201	1.23047	-0.52734	0.703124
							0.997715	-0.02132	-0.06608		-0.52734	
							0.979401	-0.01521	-0.06608		-0.52734	
							0.98169					
							1.00001					
							1.00229					
							0.990848					
92209				188	45	38.6719	0.988558	-0.02132	-0.07015	1.23047	-0.52734	0.703124
							1.00001	-0.02132	-0.06608	1.23047	-0.52734	0.703124
							0.997715	-0.01725	-0.06812		-0.52734	
							0.986269	-0.01928	-0.06608		-0.35156	
							0.988558					
							1.00001					
							1.00458					
							0.988558					
92210				188	45	38.3203	0.98169	-0.01928	-0.06608	1.23047	-0.35156	0.703124
							0.997715	-0.02335	-0.06405	1.23047	-0.35156	1.05469
							0.997715	-0.02335	-0.06608		-0.35156	
							0.988558	-0.02742	-0.07015		-0.35156	
							0.993137					
							0.997715					
							0.997715					
							0.993137					
92211				188	45	38.3203	0.988558	-0.02132	-0.07422	1.23047	-0.35156	0.703124
							0.993137	-0.01725	-0.07218	1.23047	-0.52734	0.703124
							0.993137	-0.01725	-0.07218		-0.52734	
							0.997715	-0.02132	-0.06812		-0.52734	
							1.00229					
							1.00229					
							0.988558					
							0.979401					
92212	2	40	38	188	45	37.9688	0.990848	-0.01521	-0.06812	1.23047	-0.52734	0.703124
							0.997715	-0.01114	-0.06608	1.23047	-0.52734	0.703124
							0.983979	-0.02335	-0.07218		-0.70312	
							0.986269	-0.02335	-0.06812		-0.70312	
							1.00001					
							1.00001					
							0.986269					
							0.974822					
92213				188	45	37.9688	0.986269	-0.01521	-0.07218	1.23047	-0.70312	0.703124
							1.00687	-0.01928	-0.07218	1.23047	-0.70312	0.703124
							1.00229	-0.01928	-0.06812		-0.52734	
							0.990848	-0.01521	-0.06812		-0.52734	
							0.979401					
							0.993137					
							1.00916					
							0.993137					
92214				188	45	37.9688	0.98169	-0.01928	-0.06608	1.23047	-0.52734	0.703124
							0.988558	-0.01928	-0.06608	1.23047	-0.52734	0.703124

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.997715	-0.01521	-0.07218		-0.35156	
							0.997715	-0.01725	-0.07422		-0.35156	
							0.990848					
							1.00687					
							1.00001					
							0.995426					
92215				184	45	37.9688	0.983979	-0.01725	-0.07829	1.23047	-0.35156	0.703124
							0.993137	-0.02132	-0.08032	1.23047	-0.35156	0.703124
							1.00229	-0.01521	-0.08236		-0.35156	
							1.00001	-0.01725	-0.08032		-0.35156	
							0.983979					
							0.988558					
							1.01145					
							1.00001					
92216	2	40	42	188	45	37.9688	0.993137	-0.02539	-0.07829	1.23047	-0.17578	1.05469
							0.990848	-0.01725	-0.07829	1.23047	-0.35156	1.05469
							0.986269	-0.01725	-0.08236		-0.35156	
							0.986269	-0.01725	-0.07829		-0.35156	
							0.988558					
							1.00687					
							1.00001					
							0.98169					
92217				188	45	37.9688	0.979401	-0.02132	-0.08236	1.23047	-0.35156	1.05469
							1.00687	-0.01725	-0.07829	1.23047	-0.35156	0.703124
							1.00687	-0.01725	-0.08846		-0.35156	
							0.974822	-0.02335	-0.08439		-0.35156	
							0.972533					
							1.02061					
							1.01832					
							0.98169					
92218				188	45	38.3203	0.970243	-0.00911	-0.07829	1.23047	-0.35156	0.703124
							0.983979	-0.00911	-0.07422	1.23047	-0.52734	1.05469
							1.00458	-0.02945	-0.07422		-0.52734	
							0.995426	-0.02132	-0.07218		-0.52734	
							0.990848					
							0.990848					
							0.990848					
							0.983979					
92219				188	45	38.3203	0.988558	-0.01521	-0.07625	1.23047	-0.52734	0.703124
							1.00229	-0.01928	-0.07829	1.23047	-0.52734	0.703124
							1.00229	-0.01725	-0.08643		-0.52734	
							0.98169	-0.01318	-0.09049		-0.52734	
							0.98169					
							0.993137					
							1.00229					
							0.993137					
92220	2	40	46	188	45	38.3203	0.990848	-0.01725	-0.09049	1.23047	-0.52734	0.703124
							0.993137	-0.01725	-0.08846	1.23047	-0.52734	0.703124
							1.00001	-0.01521	-0.08643		-0.52734	
							0.990848	-0.01114	-0.08643		-0.52734	
							0.983979					
							0.995426					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.00229					
							0.995426					
92221				188	45	38.3203	0.979401	-0.01521	-0.08439	1.23047	-0.52734	0.703124
							0.990848	-0.01725	-0.08439	1.23047	-0.52734	0.703124
							1.00687	-0.01928	-0.08236		-0.52734	
							0.997715	-0.01725	-0.08236		-0.52734	
							0.98169					
							0.983979					
							1.00458					
							1.00458					
92222				184	45	38.6719	0.98169	-0.02132	-0.08032	1.23047	-0.52734	0.703124
							0.979401	-0.01928	-0.08236	1.05469	-0.52734	0.703124
							1.00001	-0.01928	-0.08032		-0.52734	
							1.00458	-0.01521	-0.07829		-0.35156	
							0.990848					
							0.988558					
							0.993137					
							1.00458					
92223				188	45	38.6719	0.997715	-0.01928	-0.07625	1.23047	-0.35156	1.05469
							0.986269	-0.02335	-0.07422	1.23047	-0.35156	0.703124
							0.990848	-0.01725	-0.07422		-0.35156	
							0.997715	-0.02335	-0.07218		-0.35156	
							0.997715					
							0.993137					
							0.986269					
							0.993137					
92224	2	40	50	188	45	38.6719	0.993137	-0.01521	-0.07218	1.23047	-0.35156	0.703124
							0.993137	-0.01114	-0.07218	1.23047	-0.35156	0.703124
							0.990848	-0.01521	-0.06812		-0.35156	
							0.993137	-0.01521	-0.06812		-0.35156	
							1.00001					
							0.988558					
							0.997715					
							1.00001					
92225				184	45	38.6719	0.988558	-0.02132	-0.06812	1.23047	-0.35156	0.703124
							0.995426	-0.02335	-0.07015	1.23047	-0.35156	0.703124
							0.993137	-0.01928	-0.07218		-0.52734	
							0.995426	-0.01725	-0.07625		-0.52734	
							0.995426					
							0.988558					
							0.993137					
							0.995426					
92226				184	45	38.6719	0.997715	-0.01521	-0.07422	1.23047	-0.52734	0.703124
							0.990848	-0.01725	-0.07218	1.23047	-0.52734	0.703124
							0.983979	-0.01114	-0.06812		-0.52734	
							0.995426	-0.00504	-0.07218		-0.52734	
							0.995426					
							0.986269					
							0.993137					
							0.997715					
92227				184	45	39.0234	0.990848	-0.00911	-0.07422	1.23047	-0.52734	0.703124
							0.988558	-0.00708	-0.08032	1.23047	-0.52734	0.703124

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.993137	-0.00504	-0.07625		-0.35156	
							1.00229	-0.00708	-0.07422		-0.35156	
							0.995426					
							0.983979					
							0.993137					
							0.995426					
92228	2	40	54	184	45	40.0781	0.997715	-0.00504	-0.07015	1.23047	-0.35156	0.703124
							0.997715	-0.00708	-0.06405	1.23047	-0.35156	0.703124
							0.988558	-0.00301	-0.05998		-0.35156	
							0.986269	-0.00504	-0.06201		-0.35156	
							0.995426					
							0.995426					
							1.00229					
							0.997715					
92229				184	45	41.4844	0.983979	-0.00708	-0.06405	1.23047	-0.35156	0.703124
							0.990848	-0.00097	-0.06608	1.23047	-0.35156	0.703124
							0.995426	-0.00097	-0.06812		-0.35156	
							0.997715	-0.00097	-0.07218		-0.35156	
							1.00001					
							0.990848					
							0.988558					
							0.993137					
92230				184	45	42.8906	0.995426	0.003092	-0.06812	1.23047	-0.52734	0.351562
							0.993137	0.003092	-0.06608	1.23047	-0.52734	0.351562
							0.990848	0.003092	-0.05591		-0.52734	
							0.997715	0.005126	-0.05387		-0.52734	
							0.993137					
							0.986269					
							0.995426					
							1.00001					
92231				184	45	45	0.997715	0.003092	-0.05387	1.23047	-0.52734	0.351562
							0.990848	0.001057	-0.05387	1.23047	-0.52734	0.351562
							0.990848	0.003092	-0.05387		-0.52734	
							0.997715	0.003092	-0.05591		-0.52734	
							0.993137					
							0.995426					
							1.00229					
							0.997715					
92232	2	40	58	184	45	47.1094	0.986269	0.001057	-0.05387	1.23047	-0.52734	0.351562
							0.986269	0.003092	-0.05387	1.05469	-0.52734	0.351562
							0.990848	0.007161	-0.05387		-0.52734	
							0.997715	0.005126	-0.05387		-0.52734	
							1.00458					
							0.995426					
							0.986269					
							0.988558					
92233				184	45	49.2188	0.995426	0.009195	-0.05184	1.23047	-0.52734	0.351562
							0.993137	0.009195	-0.05591	1.23047	-0.52734	0.351562
							0.993137	0.007161	-0.0498		-0.52734	
							1.00229	0.015299	-0.05387		-0.52734	
							0.988558					
							0.98169					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.997715					
							1.00687					
92234				184	45	52.0312	0.993137	0.013264	-0.05184	1.23047	-0.52734	0.351562
							0.979401	0.013264	-0.05387	1.23047	-0.35156	0.351562
							0.990848	0.013264	-0.05387		-0.52734	
							1.00458	0.017333	-0.05591		-0.52734	
							1.00229					
							0.990848					
							0.993137					
							1.00229					
92235				184	45	54.8438	0.995426	0.015299	-0.05387	1.23047	-0.52734	0.351562
							0.983979	0.019368	-0.05591	1.23047	-0.52734	0.351562
							0.993137	0.019368	-0.05387		-0.52734	
							1.00458	0.019368	-0.05387		-0.52734	
							0.997715					
							0.983979					
							0.988558					
							0.995426					
92236	2	41	2	184	45	59.7656	1.00229	0.015299	-0.05387	1.23047	-0.52734	0.351562
							0.995426	0.017333	-0.05184	1.23047	-0.52734	0.351562
							0.986269	0.019368	-0.05387		-0.70312	
							0.98169	0.017333	-0.05184		-0.70312	
							0.997715					
							1.00229					
							0.993137					
							0.990848					
92237				184	45	63.9844	1.00458	0.019368	-0.05591	1.23047	-0.70312	0.351562
							1.00001	0.019368	-0.05184	1.23047	-0.8789	0.351562
							0.986269	0.023437	-0.05794		-0.8789	
							0.98169	0.023437	-0.0498		-0.8789	
							0.995426					
							1.01374					
							0.990848					
							0.970243					
92238				184	45	69.2578	0.988558	0.02954	-0.05387	1.23047	-0.8789	0.351562
							1.01145	0.03361	-0.05184	1.23047	-0.8789	0.703124
							1.00916	0.023437	-0.0498		-0.70312	
							0.986269	0.025471	-0.04777		-0.70312	
							0.990848					
							0.997715					
							0.979401					
							0.990848					
92239				184	45	74.1797	1.01145	0.02954	-0.04777	1.23047	-0.70312	0.703124
							1.00687	0.019368	-0.04574	1.23047	-0.70312	0.351562
							0.988558	0.02954	-0.04777		-0.52734	
							0.983979	0.043782	-0.0437		-0.52734	
							1.01145					
							1.00458					
							0.988558					
							0.986269					
92240	2	41	6	184	45	80.1562	0.995426	0.043782	-0.04574	1.23047	-0.52734	0
							1.00001	0.045817	-0.04574	1.23047	-0.35156	0

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.98169	0.041747	-0.04574		-0.35156	
							1.00229	0.043782	-0.04777		-0.35156	
							1.02519					
							0.993137					
							0.970243					
							0.995426					
92241				184	45	85.0781	1.02519	0.049886	-0.05387	1.05469	-0.35156	0
							1.01374	0.045817	-0.05794	1.23047	-0.35156	-0.35156
							0.986269	0.043782	-0.06608		-0.35156	
							0.986269	0.058024	-0.06812		-0.52734	
							0.990848					
							0.986269					
							1.00458					
							1.00916					
92242				184	45	91.7578	0.990848	0.045817	-0.06201	1.23047	-0.52734	-0.35156
							0.974822	0.045817	-0.05998	1.23047	-0.52734	-0.35156
							0.986269	0.058024	-0.05794		-0.70312	
							1.00229	0.05192	-0.05998		-0.70312	
							0.986269					
							0.979401					
							0.995426					
							1.00458					
92243				184	45	96.6797	1.00001	0.05192	-0.05794	1.23047	-0.70312	-0.35156
							0.986269	0.043782	-0.05794	1.23047	-0.70312	-0.35156
							0.988558	0.039713	-0.05591		-0.70312	
							0.990848	0.047851	-0.05794		-0.70312	
							0.988558					
							0.993137					
							0.990848					
							0.997715					
92244	2	41	10	184	45	102.656	0.983979	0.047851	-0.05591	1.23047	-0.70312	-0.70312
							0.979401	0.053955	-0.05794	1.23047	-0.70312	-0.35156
							1.00001	0.047851	-0.05794		-0.70312	
							1.00687	0.035644	-0.05794		-0.70312	
							1.00001					
							0.993137					
							0.990848					
							0.988558					
92245				184	45	106.875	0.993137	0.045817	-0.05794	1.23047	-0.70312	-0.35156
							0.990848	0.045817	-0.06201	1.23047	-0.70312	-0.35156
							1.00229	0.041747	-0.06201		-0.70312	
							1.00458	0.05192	-0.05998		-0.70312	
							0.983979					
							0.977111					
							0.993137					
							1.01374					
92246				184	45	112.5	1.00687	0.041747	-0.04574	1.23047	-0.70312	-0.35156
							0.967954	0.02954	-0.05591	1.23047	-0.70312	-0.35156
							0.990848	0.045817	-0.04777		-0.70312	
							1.01603	0.043782	-0.05184		-0.70312	
							0.990848					
							0.979401					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.990848					
							1.00916					
92247				184	45	116.719	1.00001	0.035644	-0.0498	1.05469	-0.70312	-0.35156
							0.979401	0.041747	-0.05184	1.23047	-0.70312	-0.35156
							1.00001	0.043782	-0.0498		-0.70312	
							1.00916	0.037679	-0.0498		-0.70312	
							0.988558					
							0.979401					
							0.995426					
							1.00687					
92248	2	41	14	184	45	121.641	0.995426	0.03361	-0.04777	1.23047	-0.70312	-0.35156
							0.98169	0.039713	-0.05184	1.23047	-0.70312	-0.35156
							1.00001	0.035644	-0.0437		-0.70312	
							1.01145	0.02954	-0.04777		-0.70312	
							1.00001					
							0.974822					
							0.977111					
							1.00916					
92249				184	45	124.805	1.01832	0.02954	-0.0437	1.23047	-0.52734	-0.35156
							0.995426	0.015299	-0.04574	1.23047	-0.52734	-0.35156
							0.979401	0.017333	-0.04777		-0.52734	
							0.993137	0.019368	-0.04777		-0.52734	
							1.00458					
							0.995426					
							0.988558					
							0.988558					
92250				184	45	127.266	1.00229	0.013264	-0.0498	1.23047	-0.52734	-0.35156
							1.00001	0.013264	-0.0498	1.23047	-0.52734	-0.35156
							0.986269	0.01123	-0.0498		-0.52734	
							0.993137	0.009195	-0.0498		-0.52734	
							1.00001					
							0.993137					
							0.993137					
92251				184	45	128.672	0.995426	0.007161	-0.0498	1.23047	-0.52734	-0.35156
							1.00001	0.009195	-0.0498	1.23047	-0.52734	-0.35156
							0.988558	0.01123	-0.0498		-0.52734	
							0.995426	0.013264	-0.0498		-0.52734	
							1.00229					
							0.995426					
							0.990848					
							0.990848					
92252	2	41	18	184	45	129.375	1.00687	0.013264	-0.0498	1.23047	-0.52734	-0.35156
							0.995426	0.01123	-0.04777	1.23047	-0.52734	-0.35156
							0.979401	0.003092	-0.05184		-0.70312	
							0.997715	0.009195	-0.0498		-0.70312	
							1.00916					
							1.00001					
							0.98169					
							0.98169					
92253				184	45	130.43	0.997715	0.017333	-0.0498	1.23047	-0.70312	-0.35156
							1.01145	0.009195	-0.0498	1.23047	-0.70312	-0.35156

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.997715	0.01123	-0.05387		-0.70312	
							0.979401	0.01123	-0.0498		-0.70312	
							0.988558					
							1.01145					
							1.00229					
							0.979401					
92254				184	45	131.133	0.986269	0.01123	-0.05184	1.23047	-0.70312	-0.35156
							1.00229	0.01123	-0.04777	1.23047	-0.70312	-0.35156
							1.00458	0.01123	-0.0498		-0.52734	
							0.988558	0.007161	-0.0498		-0.70312	
							0.986269					
							1.00001					
							1.00229					
							0.986269					
92255				184	45	131.836	0.983979	0.013264	-0.04777	1.23047	-0.70312	-0.70312
							0.997715	0.013264	-0.0498	1.23047	-0.52734	-0.35156
							1.00687	0.009195	-0.0498		-0.52734	
							0.997715	0.01123	-0.05184		-0.52734	
							0.983979					
							0.988558					
							1.00458					
							1.00229					
92256	2	41	22	184	45	132.539	0.993137	0.01123	-0.04574	1.23047	-0.52734	-0.70312
							0.979401	0.013264	-0.0498	1.23047	-0.52734	-0.70312
							0.995426	0.013264	-0.04777		-0.52734	
							1.00687	0.01123	-0.05184		-0.52734	
							1.00229					
							0.986269					
							0.990848					
							1.00687					
92257				184	45	133.242	1.00001	0.01123	-0.04777	1.23047	-0.52734	-0.70312
							0.983979	0.015299	-0.0498	1.23047	-0.52734	-0.70312
							0.98169	0.013264	-0.04777		-0.52734	
							1.00001	0.013264	-0.05184		-0.52734	
							1.00687					
							0.990848					
							0.98169					
							1.00001					
92258				184	45	133.594	1.00458	0.01123	-0.05184	1.23047	-0.52734	-0.70312
							0.990848	0.01123	-0.04777	1.23047	-0.52734	-0.70312
							0.988558	0.01123	-0.05184		-0.52734	
							0.988558	0.009195	-0.04777		-0.52734	
							1.00229					
							1.00687					
							0.988558					
							0.983979					
92259				180	45	134.297	0.993137	0.009195	-0.05184	1.23047	-0.52734	-0.70312
							1.00229	0.01123	-0.04777	1.23047	-0.52734	-0.70312
							0.995426	0.013264	-0.05184		-0.52734	
							0.979401	0.01123	-0.0498		-0.52734	
							0.990848					
							1.00687					



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.00001					
							0.983979					
92260	2	41	26	180	45	134.648	0.993137	0.013264	-0.0498	1.05469	-0.52734	-0.70312
							1.00458	0.009195	-0.04574	1.23047	-0.52734	-0.70312
							1.00687	0.013264	-0.05184		-0.52734	
							0.983979	0.01123	-0.04777		-0.52734	
							0.977111					
							1.00687					
							1.01374					
							0.988558					
92261				180	45	135	0.972533	0.01123	-0.0498	1.23047	-0.52734	-0.70312
							1.00001	0.007161	-0.04574	1.23047	-0.52734	-0.70312
							1.01603	0.005126	-0.04777		-0.52734	
							0.988558	0.007161	-0.04574		-0.52734	
							0.979401					
							-0.26602					
							1.00458					
							0.988558					
92262				180	45	135	0.98169	0.013264	-0.04777	1.23047	-0.52734	-0.70312
							1.00001	0.005126	-0.0498	1.05469	-0.52734	-0.70312
							1.00458	0.003092	-0.05184		-0.52734	
							1.00001	0.001057	-0.0498		-0.52734	
							0.993137					
							0.995426					
							0.995426					
							0.990848					
92263				180	45	134.648	0.986269	0.009195	-0.05184	1.23047	-0.52734	-0.70312
							0.997715	0.005126	-0.0498	1.23047	-0.52734	-0.70312
							1.00687	0.003092	-0.05184		-0.52734	
							0.995426	0.003092	-0.05184		-0.52734	
							0.986269					
							0.995426					
							1.00458					
							0.995426					
92264	2	41	30	180	45	134.297	0.98169	0.003092	-0.0498	1.05469	-0.52734	-0.70312
							0.98169	0.005126	-0.05591	1.05469	-0.52734	-0.70312
							1.00229	0.007161	-0.05998		-0.52734	
							1.00229	0.003092	-0.07015		-0.52734	
							0.988558					
							0.990848					
							1.00001					
							0.993137					
92265				180	45	133.945	0.986269	0.01123	-0.07625	1.23047	-0.52734	-0.70312
							0.990848	0.007161	-0.08439	1.23047	-0.52734	-0.35156
							1.00001	0.009195	-0.08846		-0.52734	
							1.00001	0.01123	-0.09049		-0.52734	
							0.98169					
							0.988558					
							0.995426					
							1.00001					
92266				180	45	134.297	0.988558	0.009195	-0.09049	1.23047	-0.52734	-0.35156
							0.983979	0.009195	-0.09253	1.05469	-0.52734	-0.35156

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.995426	0.01123	-0.0966		-0.52734	
							1.00001	0.007161	-0.09253		-0.52734	
							0.98169					
							0.993137					
							1.00229					
							0.993137					
92267				180	45	135	0.98169	0.01123	-0.09253	1.05469	-0.52734	-0.35156
							0.990848	0.009195	-0.08846	1.23047	-0.52734	-0.35156
							1.00458	0.01123	-0.08846		-0.52734	
							0.993137	0.007161	-0.08846		-0.52734	
							0.977111					
							0.993137					
							1.00916					
							0.995426					
92268	2	41	34	180	45	135.352	0.98169	0.01123	-0.08032	1.23047	-0.52734	-0.35156
							0.988558	0.009195	-0.07625	1.23047	-0.52734	-0.35156
							0.997715	0.009195	-0.07015		-0.52734	
							0.995426	0.017333	-0.06608		-0.52734	
							0.993137					
							0.988558					
							0.993137					
							1.00001					
92269				180	45	136.406	0.995426	0.009195	-0.06608	1.23047	-0.52734	-0.35156
							0.988558	0.01123	-0.06608	1.23047	-0.52734	-0.35156
							0.983979	0.01123	-0.06812		-0.52734	
							0.997715	0.017333	-0.06812		-0.52734	
							1.00229					
							0.997715					
							0.986269					
							0.990848					
92270				180	45	137.109	0.997715	0.015299	-0.07422	1.23047	-0.52734	-0.35156
							1.00229	0.021403	-0.07422	1.23047	-0.70312	-0.35156
							0.997715	0.021403	-0.07422		-0.70312	
							0.983979	0.019368	-0.07218		-0.8789	
							0.995426					
							0.993137					
							0.986269					
							0.993137					
92271				180	45	138.867	1.00229	0.027506	-0.07218	1.23047	-0.8789	-0.35156
							0.995426	0.025471	-0.07625	1.23047	-0.8789	-0.70312
							0.983979	0.035644	-0.07015		-0.8789	
							0.986269	0.031575	-0.07218		-0.8789	
							1.00687					
							1.00001					
							0.958796					
							0.983979					
92272	2	41	38	180	45	141.328	1.01145	0.03361	-0.06812	1.23047	-0.8789	-0.70312
							0.993137	0.045817	-0.06405	1.23047	-0.8789	-0.70312
							0.977111	0.041747	-0.06201		-0.70312	
							0.988558	0.060058	-0.05591		-0.70312	
							1.00001					
							0.993137					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.979401					
							0.98169					
92273				180	45	146.602	0.993137	0.066162	-0.05794	1.23047	-0.52734	-1.05469
							1.02061	0.060058	-0.05591	1.23047	-0.35156	-1.05469
							1.00229	0.078369	-0.05794		-0.17578	
							0.977111	0.080403	-0.06201		-0.17578	
							0.988558					
							1.00001					
							1.02748					
							1.01374					
92274				180	45	152.227	0.979401	0.053955	-0.06405	1.23047	0	-0.70312
							0.988558	0.064127	-0.06201	1.23047	0	-0.70312
							1.01145	0.080403	-0.06608		0	
							0.993137	0.0743	-0.06608		0	
							0.977111					
							1.00458					
							1.01374					
							0.986269					
92275				180	45	160.664	0.98169	0.078369	-0.06812	1.23047	0	-0.70312
							1.00001	0.082438	-0.06812	1.23047	0	-0.35156
							1.00916	0.066162	-0.07015		0	
							1.00001	0.055989	-0.05591		-0.17578	
							0.993137					
							1.00687					
							1.00001					
							0.98169					
92276	2	41	42	180	45	167.695	0.983979	0.060058	-0.05184	1.23047	-0.17578	-0.35156
							0.997715	0.064127	-0.04574	1.23047	-0.17578	-0.70312
							0.990848	0.070231	-0.0498		-0.17578	
							0.977111	0.0743	-0.0498		-0.17578	
							0.988558					
							1.00687					
							0.995426					
							0.983979					
92277				180	45	175.078	0.988558	0.068196	-0.0498	1.23047	-0.17578	-0.70312
							0.997715	0.068196	-0.04574	1.23047	-0.17578	-0.70312
							0.995426	0.062093	-0.04777		-0.17578	
							0.990848	0.053955	-0.04777		-0.17578	
							0.993137					
							1.00916					
							1.00229					
							0.983979					
92278				180	45	182.109	0.983979	0.066162	-0.0498	1.23047	-0.17578	-0.70312
							1.00229	0.066162	-0.05184	1.05469	-0.17578	-0.70312
							1.00458	0.053955	-0.0498		-0.17578	
							0.993137	0.058024	-0.04574		-0.17578	
							0.986269					
							0.986269					
							1.00229					
							1.00001					
92279				180	45	188.438	0.993137	0.045817	-0.04777	1.23047	-0.17578	-0.70312
							0.993137	0.049886	-0.04777	1.23047	-0.17578	-0.70312

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.988558	0.064127	-0.0498		-0.17578	
							0.995426	0.05192	-0.05184		-0.17578	
							0.993137					
							1.00001					
							1.00687					
							0.993137					
92280	2	41	46	180	45	193.711	0.972533	0.045817	-0.04777	1.23047	-0.17578	-0.70312
							0.98169	0.049886	-0.05184	1.23047	-0.17578	-0.70312
							1.01374	0.043782	-0.04574		-0.17578	
							1.01832	0.05192	-0.05387		-0.17578	
							0.986269					
							0.965664					
							0.986269					
							1.01603					
92281				180	45	199.336	1.00458	0.058024	-0.04777	1.23047	-0.17578	-1.05469
							0.974822	0.05192	-0.0498	1.23047	-0.17578	-1.05469
							0.979401	0.055989	-0.04574		-0.17578	
							1.00229	0.05192	-0.05184		-0.17578	
							1.01145					
							0.988558					
							0.979401					
							1.00458					
92282				180	45	203.906	1.00458	0.053955	-0.04574	1.23047	-0.17578	-1.05469
							0.98169	0.05192	-0.05184	1.23047	-0.17578	-1.05469
							0.98169	0.047851	-0.0498		-0.17578	
							1.01145	0.049886	-0.05184		-0.17578	
							1.01374					
							0.98169					
							0.972533					
							1.00229					
92283				180	45	208.828	1.01374	0.047851	-0.04777	1.23047	-0.17578	-1.05469
							0.988558	0.043782	-0.04777	1.23047	-0.17578	-0.70312
							0.977111	0.043782	-0.04777		-0.17578	
							0.990848	0.039713	-0.0437		-0.17578	
							1.00916					
							1.00229					
							0.979401					
							0.986269					
92284	2	41	50	180	45	212.344	1.00916	0.02954	-0.03963	1.23047	0	-0.70312
							1.00001	0.031575	-0.03963	1.23047	0	-0.70312
							0.979401	0.031575	-0.04167		0	
							0.997715	0.027506	-0.0437		0	
							1.01374					
							1.00001					
							0.979401					
							0.988558					
92285				180	45	215.156	1.01145	0.031575	-0.0437	1.23047	0	-0.70312
							1.00458	0.023437	-0.0437	1.23047	0	-1.05469
							0.983979	0.025471	-0.04574		0	
							0.983979	0.025471	-0.0437		0	
							1.00458					
							1.00458					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.990848					
							0.990848					
92286				180	45	216.562	1.00001	0.021403	-0.0437	1.23047	0	-1.05469
							1.00001	0.025471	-0.04574	1.23047	0	-1.05469
							0.995426	0.025471	-0.03963		0	
							0.995426	0.021403	-0.0437		0	
							0.990848					
							0.983979					
							0.995426					
							1.00458					
92287				180	45	217.969	1.00001	0.025471	-0.04167	1.23047	0	-1.05469
							0.993137	0.021403	-0.04167	1.23047	0	-1.05469
							0.993137	0.023437	-0.04167		0	
							0.993137	0.02954	-0.03963		0	
							0.988558					
							0.997715					
							1.00001					
							0.997715					
92288	2	41	54	180	45	219.023	0.990848	0.021403	-0.03556	1.23047	0	-1.40625
							0.983979	0.025471	-0.03556	1.23047	0	-1.40625
							0.98169	0.027506	-0.03353		0	
							1.00687	0.021403	-0.03353		0	
							1.01374					
							0.986269					
							0.979401					
							0.997715					
92289				180	45	219.727	1.00916	0.025471	-0.03149	1.23047	0	-1.40625
							0.993137	0.021403	-0.02946	1.05469	0	-1.40625
							0.986269	0.019368	-0.02743		0	
							0.995426	0.023437	-0.02539		0	
							0.997715					
							1.00001					
							0.990848					
							0.993137					
92290				180	45	220.078	1.00229	0.013264	-0.02336	1.05469	0	-1.05469
							1.00229	0.019368	-0.02132	1.23047	0	-1.05469
							0.993137	0.015299	-0.01725		0	
							0.988558	0.017333	-0.01115		0	
							1.00458					
							1.00458					
							0.997715					
							0.993137					
92291				184	45	220.43	0.995426	0.021403	-0.00708	1.23047	0	-1.05469
							1.00229	0.015299	0.005127	1.23047	0	-1.05469
							1.01145	0.015299	0.017334		0	
							0.988558	0.017333	0.027507		0	
							0.98169					
							1.00001					
							1.00687					
							1.00001					
92292	2	41	58	180	45	220.781	0.988558	0.015299	0.029541	1.23047	0	-1.05469
							0.995426	0.017333	0.029541	1.23047	0	-1.05469

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUDE ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.00916	0.015299	0.029541		0	
							1.00229	0.01123	0.029541		0	
							0.997715					
							0.993137					
							1.00458					
							1.00916					
92293				184	45	220.781	0.993137	0.01123	0.035644	1.23047	0	-1.05469
							0.993137	0.003092	0.043783	1.23047	0	-1.05469
							1.00458	0.013264	0.049886		0	
							1.00458	0.017333	0.05599		0	
							0.986269					
							0.986269					
							1.01145					
							1.00458					
92294				184	45	221.133	0.990848	0.021403	0.060059	1.23047	0	-1.40625
							1.00687	0.017333	0.070231	1.23047	0	-1.05469
							1.01145	0.023437	0.0743		0	
							1.00229	0.027506	0.076335		0	
							0.997715					
							1.00229					
							0.995426					
							0.995426					
92295				184	45	221.836	1.01145	0.025471	0.076335	1.23047	0	-1.40625
							1.01145	0.027506	0.076335	1.23047	0.175781	-1.40625
							0.995426	0.031575	0.078369		0.175781	
							0.988558	0.03361	0.080404		0.175781	
							1.00916					
							1.00687					
							0.98169					
							0.983979					
92296	2	42	2	188	45	223.242	1.00229	0.035644	0.080404	1.23047	0.175781	-1.40625
							1.00687	0.027506	0.076335	1.23047	0.351562	-1.05469
							1.02061	0.019368	0.082438		0.351562	
							1.02061	0.017333	0.094645		0.351562	
							1.01145					
							1.00001					
							0.997715					
							1.00687					
92297				188	45	223.594	1.00916	0.005126	0.112956	1.23047	0.175781	-0.70312
							0.997715	0.003092	0.127197	1.23047	0.175781	-0.70312
							0.995426	0.003092	0.151611		0.175781	
							1.00229	-0.00301	0.159749		0.175781	
							1.01145					
							0.986269					
							0.993137					
							1.00916					
92298				188	45	223.594	1.02061	-0.00504	0.171956	1.23047	0.175781	-0.70312
							1.00229	0.01123	0.167887	1.23047	0.175781	-1.05469
							1.00001	0.009195	0.169922		0.175781	
							1.0229	0.001057	0.163818		0.175781	
							1.0229					
							0.986269					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUDE ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.972533					
							1.00687					
92299				188	45	223.945	1.01603	0.01123	0.163818	1.23047	0.175781	-1.05469
							1.01832	0.009195	0.171956	1.05469	0.175781	-1.05469
							1.00229	0.001057	0.180094		0.175781	
							0.997715	0.013264	0.184163		0.175781	
							1.01145					
							1.00458					
							1.00001					
							1.01374					
92300	2	42	6	192	45	223.594	1.0229	0.007161	0.17806	1.23047	0.351562	-1.05469
							1.0435	-0.00504	0.180094	1.23047	0.351562	-1.75781
							1.00687	0.001057	0.182129		0.351562	
							0.983979	0.027506	0.188232		0.175781	
							1.00458					
							1.01374					
							1.00458					
							0.977111					
92301				192	45.5	223.594	0.951928	0.009195	0.182129	1.23047	0.175781	-1.05469
							1.00916	-0.00708	0.180094	1.23047	0.175781	-1.40625
							1.07098	-0.00708	0.184163		0.175781	
							1.02748	-0.00911	0.182129		0.175781	
							0.972533					
							0.970243					
							1.00916					
							1.03206					
92302				192	49.5	222.891	1.00458	-0.01318	0.192301	1.23047	0.175781	-1.05469
							0.997715	-0.01318	0.200439	1.23047	0.175781	-1.05469
							1.05266	-0.01521	0.216715		0.351562	
							1.02977	-0.01521	0.206543		0.175781	
							0.979401					
							0.94277					
							0.986269					
							1.06411					
92303				196	56	222.188	1.05266	-0.01114	0.222819	1.23047	0.175781	-1.05469
							0.986269	-0.00911	0.222819	1.05469	0.175781	-1.05469
							0.974822	-0.00301	0.228922		0.351562	
							1.02519	-0.00097	0.23706		0.351562	
							1.07555					
							1.02748					
							0.951928					
							0.94506					
92304	2	42	10	196	61	222.188	1.00687	0.005126	0.226888	1.05469	0.175781	-1.40625
							1.03892	0.003092	0.228922	1.23047	0.175781	-1.05469
							0.993137	0.001057	0.21875		0	
							0.990848	0.007161	0.228922		0.175781	
							1.05495					
							1.04808					
							1.00001					
							0.98169					
92305				196	65	222.188	0.972533	0.003092	0.224853	1.05469	0.351562	-1.40625
							1.01603	0.007161	0.228922	1.05469	0.351562	-1.40625

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.03206	0.025471	0.226888		0.175781	
							1.01603	0.031575	0.228922		0.351562	
							1.0435					
							1.01603					
							1.04579					
							0.967954					
92306				196	70	222.891	0.883247	0.02954	0.226888	1.23047	0.351562	-1.40625
							0.94277	0.02954	0.220784	1.23047	0.351562	-1.40625
							1.11676	0.02954	0.230957		0.351562	
							1.12134	0.027506	0.228922		0.351562	
							1.01374					
							0.967954					
							0.931324					
							1.00916					
92307				200	75.5	222.891	1.0664	0.009195	0.224853	1.05469	0.351562	-1.40625
							1.03892	0.003092	0.230957	1.05469	0.351562	-1.40625
							0.986269	0.003092	0.210612		0.351562	
							0.913009	0.003092	0.212646		0.175781	
							0.995426					
							1.04808					
							1.04121					
							1.03663					
92308	2	42	14	200	78.5	222.188	1.02061	-0.01114	0.222819	0.878905	0.175781	-1.05469
							0.965664	-0.02945	0.216715	0.878905	0.175781	-1.40625
							0.954217	-0.01928	0.210612		0.351562	
							1.00458	-0.00301	0.222819		0.351562	
							1.05953					
							1.08471					
							1.01603					
							0.94735					
92309				200	83.5	222.188	0.979401	0.001057	0.214681	0.878905	0.351562	-1.05469
							0.974822	0.009195	0.212646	0.878905	0.351562	-1.40625
							0.986269	0.023437	0.214681		0.175781	
							1.01145	0.023437	0.212646		0.175781	
							1.05953					
							1.05266					
							0.94506					
							0.997715					
92310				200	89	222.539	1.09158	0.019368	0.216715	0.703124	0.351562	-1.05469
							1.02748	0.017333	0.212646	1.05469	0.351562	-1.40625
							0.929034	0.02954	0.202474		0.351562	
							0.940481	0.01123	0.206543		0.351562	
							1.00687					
							1.08013					
							1.02061					
							0.993137					
92311				200	93	222.188	0.965664	-0.01928	0.206543	0.878905	0.175781	-1.05469
							1.00458	-0.01928	0.204508	0.703124	0.175781	-1.05469
							1.06411	-0.01318	0.202474		0.351562	
							1.03892	-0.01318	0.200439		0.351562	
							0.972533					
							0.986269					



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.00229					
							1.01603					
92312	2	42	18	200	97.5	221.836	1.02748	-0.02539	0.208577	0.527343	0.351562	-1.05469
							0.949639	-0.01928	0.198405	0.878905	0.351562	-1.05469
							0.94506	-0.00911	0.202474		0.175781	
							1.05266	-0.01318	0.204508		0.175781	
							1.05953					
							1.06182					
							0.979401					
							0.979401					
92313				204	101	221.836	0.979401	-0.00708	0.206543	0.527343	0.351562	-1.05469
							0.983979	0.013264	0.190267	0.878905	0.351562	-1.40625
							1.00001	0.023437	0.200439		0.351562	
							1.087	-0.00911	0.198405		0.351562	
							1.05266					
							1.00229					
							0.938192					
							0.977111					
92314				204	106.5	221.836	1.06869	-0.00504	0.190267	0.703124	0.351562	-1.05469
							1.0664	0.007161	0.194336	0.703124	0.351562	-1.40625
							0.933613	-0.00708	0.200439		0.351562	
							0.972533	-0.01318	0.194336		0.351562	
							0.997715					
							1.02061					
							1.13508					
							1.07555					
92315				204	109.5	221.484	0.885536	-0.00504	0.198405	0.351562	0.351562	-1.40625
							0.890115	-0.00301	0.186198	0.878905	0.351562	-1.40625
							0.949639	-0.01521	0.190267		0.351562	
							1.03663	0.009195	0.19637		0.351562	
							1.10761					
							1.07555					
							0.993137					
							0.967954					
92316	2	42	22	204	115.5	221.836	0.926745	0.009195	0.190267	0.703124	0.351562	-1.05469
							1.00001	0.013264	0.184163	0.878905	0.351562	-1.05469
							1.03206	0.021403	0.186198		0.351562	
							1.04121	0.015299	0.188232		0.351562	
							1.04808					
							0.98169					
							0.917587					
							1.01145					
92317				204	119.5	221.836	1.08013	-0.00301	0.17806	0.703124	0.351562	-1.05469
							1.07098	0.005126	0.184163	1.05469	0.351562	-1.40625
							0.988558	0.021403	0.17806		0.351562	
							0.876379	0.023437	0.184163		0.351562	
							0.98169					
							1.06869					
							1.0664					
							1.05266					
92318				204	123.5	222.188	0.983979	0.037679	0.190267	0.351562	0.351562	-1.05469
							0.94277	0.041747	0.180094	0.878905	0.351562	-1.05469

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.926745	0.035644	0.182129		0.527343	
							1.0435	0.045817	0.190267		0.527343	
							1.1099					
							1.0435					
							1.00916					
							0.954217					
92319				208	127.5	222.539	0.910719	0.02954	0.184163	0.703124	0.527343	-1.05469
							0.993137	0.015299	0.17806	0.878905	0.527343	-1.05469
							1.05495	0.035644	0.186198		0.527343	
							1.03892	0.003092	0.180094		0.527343	
							1.06869					
							0.974822					
							0.901562					
							0.972533					
92320	2	42	26	208	131.5	222.188	0.94735	-0.03149	0.163818	0.703124	0.527343	-1.40625
							1.08242	-0.01928	0.17806	0.527343	0.527343	-1.40625
							1.20376	-0.01521	0.180094		0.527343	
							1.02519	0.003092	0.173991		0.351562	
							0.844327					
							0.890115					
							1.03892					
							1.06411					
92321				208	135.5	222.539	1.08471	0.023437	0.180094	0.878905	0.527343	-1.05469
							0.961086	0.027506	0.157715	0.878905	0.527343	-1.05469
							0.874089	0.045817	0.169922		0.527343	
							1.06182	0.043782	0.188232		0.527343	
							1.16713					
							1.06411					
							0.977111					
							0.858064					
92322				208	139	222.891	0.848906	0.039713	0.169922	1.05469	0.527343	-1.05469
							1.08013	0.041747	0.17806	0.878905	0.703124	-1.40625
							1.18316	0.05192	0.194336		0.878905	
							1.09387	0.068196	0.186198		0.878905	
							0.963375					
							0.890115					
							0.963375					
							1.05266					
92323				204	142.5	222.891	1.08013	0.05192	0.17806	1.40625	1.23047	-1.05469
							1.06869	0.009195	0.188232	1.75781	1.40625	-1.40625
							0.954217	0.005126	0.188232		1.58203	
							0.915298	-0.00708	0.188232		1.58203	
							0.995426					
							1.00687					
							1.01145					
							1.00916					
92324	2	42	30	204	146	222.188	0.983979	-0.03759	0.184163	2.28515	1.75781	-1.05469
							0.995426	-0.05183	0.190267	2.46093	1.93359	-1.40625
							1.02748	-0.03963	0.19637		1.93359	
							1.00001	-0.03352	0.198405		2.10937	
							0.993137					
							0.970243					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUDE ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.997715					
							1.02977					
92325				196	150	221.133	1.07784	-0.02945	0.210612	2.98828	2.63671	-1.40625
							0.988558	-0.02742	0.212646	4.04296	2.8125	-1.05469
							0.970243	-0.0437	0.216715		3.33984	
							1.00229	-0.03963	0.226888		3.86718	
							0.98169					
							0.986269					
							0.979401					
							0.970243					
92326				192	152	220.781	1.01145	-0.02335	0.23706	5.62499	4.21874	-1.40625
							1.00687	-0.01725	0.245198	6.85546	5.09765	-1.05469
							0.993137	-0.02742	0.249267		5.27343	
							1.03892	-0.02539	0.255371		6.32812	
							1.05037					
							0.990848					
							0.993137					
							1.01145					
92327				192	155.5	221.133	1.00458	-0.01318	0.261474	8.43749	6.67968	-1.05469
							0.997715	-0.01318	0.263509	9.84374	7.03124	-1.40625
							1.02748	-0.00504	0.269613		7.73436	
							1.01603	-0.00911	0.273682		7.91014	
							0.977111					
							1.00458					
							1.02061					
							1.03435					
92328	2	42	34	196	159	221.133	1.03892	-0.01318	0.273682	10.7226	8.26171	-1.05469
							1.0435	-0.00708	0.273682	10.8984	8.61327	-0.70312
							1.04579	-0.00301	0.273682		8.78905	
							1.05037	-0.01318	0.269613		8.96483	
							1.06411					
							1.06182					
							1.05495					
							1.05724					
92329				208	162	220.781	1.04579	-0.01521	0.265544	10.7226	8.96483	-0.70312
							1.03663	-0.01725	0.263509	10.1953	8.96483	-1.05469
							1.02977	-0.01928	0.263509		8.96483	
							1.01832	-0.01928	0.261474		8.96483	
							1.01374					
							1.01145					
							1.01603					
							1.0229					
92330				220	165.5	220.781	1.03435	-0.01318	0.261474	10.3711	9.14061	-1.05469
							1.05037	-0.01114	0.265544	10.7226	9.49217	-1.40625
							1.0664	-0.01114	0.269613		9.84374	
							1.07784	-0.01725	0.273682		10.5469	
							1.08929					
							1.1099					
							1.12134					
							1.13508					
92331				240	167.5	220.781	1.14195	-0.02539	0.275716	11.6015	10.8984	-1.05469
							1.15568	-0.01725	0.277751	11.9531	11.0742	-1.05469

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUDE ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.15797	-0.00708	0.277751		11.9531	
							1.16713	-0.00911	0.279785		12.3047	
							1.16484					
							1.16942					
							1.174					
							1.18316					
92332	2	42	38	268	169.5	221.133	1.18773	-0.00708	0.28182	12.3047	12.832	-0.70312
							1.1946	-0.00708	0.277751	12.3047	13.0078	-0.35156
							1.19002	-0.00911	0.277751		13.3594	
							1.1946	-0.00097	0.275716		13.7109	
							1.19231					
							1.19231					
							1.19231					
							1.18544					
92333				300	171.5	221.836	1.174	-0.00301	0.271647	11.9531	13.8867	0
							1.16713	-0.00097	0.271647	11.4258	14.0625	0
							1.16255	0.003092	0.269613		14.414	
							1.16255	0.01123	0.273682		14.5898	
							1.15339					
							1.14653					
							1.1511					
							1.15568					
92334				328	172	222.188	1.15568	0.009195	0.277751	11.4258	14.7656	0.703124
							1.15568	0.009195	0.279785	11.25	15.1172	1.05469
							1.16484	0.007161	0.275716		15.2929	
							1.16942	0.003092	0.271647		15.6445	
							1.16484					
							1.17171					
							1.16026					
							1.1511					
92335				364	173	222.539	1.14424	0.001057	0.267578	10.8984	15.6445	1.75781
							1.12821	-0.00097	0.25944	9.66795	15.6445	2.10937
							1.09845	0.001057	0.257406		15.2929	
							1.07098	-0.00504	0.249267		15.1172	
							1.04121					
							1.02061					
							0.993137					
							0.979401					
92336	2	42	42	400	174	222.891	0.963375	0.001057	0.243164	8.26171	14.5898	1.75781
							0.951928	0.003092	0.24113	7.3828	14.414	1.05469
							0.933613	0.005126	0.24113		14.2383	
							0.931324	0.003092	0.24113		13.8867	
							0.919877					
							0.90843					
							0.917587					
							0.926745					
92337				440	174.5	223.594	0.926745	0.005126	0.24113	7.55858	13.8867	0.703124
							0.935903	0.007161	0.243164	7.91014	13.8867	0
							0.940481	0.005126	0.245198		13.8867	
							0.94735	0.01123	0.245198		13.8867	
							0.958796					
							0.965664					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.979401					
							0.993137					
92338				480	176	223.945	0.997715	0.015299	0.245198	8.08593	13.8867	-0.35156
							0.995426	0.015299	0.243164	7.73436	13.8867	-0.70312
							0.988558	0.015299	0.239095		13.8867	
							0.98169	0.013264	0.23706		13.7109	
							0.974822					
							0.965664					
							0.956507					
							0.940481					
92339				512	176.5	223.945	0.929034	0.013264	0.235026	7.20702	13.7109	-0.70312
							0.924456	0.015299	0.235026	6.85546	13.3594	-0.35156
							0.924456	0.01123	0.230957		13.1836	
							0.919877	0.009195	0.230957		13.0078	
							0.922166					
							0.917587					
							0.915298					
							0.90614					
92340	2	42	46	548	177	223.945	0.901562	0.007161	0.230957	6.5039	12.832	-0.35156
							0.90614	0.01123	0.230957	6.5039	12.6562	-0.35156
							0.901562	0.01123	0.230957		12.6562	
							0.903851	0.01123	0.232991		12.6562	
							0.90843					
							0.90614					
							0.913009					
							0.919877					
92341				584	178	223.945	0.926745	0.013264	0.232991	6.67968	12.6562	-0.35156
							0.935903	0.01123	0.232991	6.85546	12.6562	-0.35156
							0.938192	0.009195	0.235026		12.6562	
							0.94506	0.013264	0.23706		12.832	
							0.94735					
							0.949639					
							0.958796					
							0.961086					
92342				616	178.5	223.945	0.967954	0.01123	0.23706	7.20702	12.832	-0.70312
							0.972533	0.009195	0.239095	7.55858	13.0078	-0.70312
							0.974822	0.009195	0.24113		13.0078	
							0.986269	0.009195	0.243164		13.1836	
							0.993137					
							1.00229					
							1.01145					
							1.01603					
92343				652	179	223.594	1.01832	0.007161	0.24113	7.91014	13.1836	-0.70312
							1.01832	0.005126	0.24113	7.73436	13.1836	-0.35156
							1.00916	0.009195	0.239095		13.1836	
							1.00458	0.007161	0.235026		13.1836	
							1.00001					
							0.995426					
							0.986269					
							0.972533					
92344	2	42	50	688	178.5	223.594	0.958796	0.007161	0.235026	7.20702	13.1836	-0.35156
							0.954217	0.005126	0.235026	7.20702	13.0078	0

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUDE ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.958796	0.003092	0.235026		13.0078	
							0.965664	0.003092	0.235026		13.0078	
							0.967954					
							0.972533					
							0.970243					
							0.963375					
92345				720	179.5	223.242	0.970243	0.003092	0.235026	7.20702	13.0078	0
							0.98169	0.003092	0.235026	7.3828	13.0078	0
							0.986269	0.003092	0.23706		13.0078	
							0.990848	0.003092	0.235026		13.0078	
							0.993137					
							0.997715					
							0.995426					
							0.990848					
92346				756	179.5	223.242	0.979401	0.005126	0.23706	7.3828	13.0078	-0.35156
							0.979401	0.009195	0.239095	7.20702	13.0078	-0.70312
							0.993137	0.01123	0.23706		13.1836	
							0.986269	0.01123	0.239095		13.1836	
							0.98169					
							0.983979					
							0.993137					
							1.00458					
92347				792	180	223.242	0.997715	0.009195	0.235026	7.20702	13.1836	-0.70312
							0.983979	0.001057	0.235026	7.3828	13.1836	-0.70312
							0.972533	0.003092	0.235026		13.1836	
							0.972533	0.009195	0.235026		13.1836	
							0.98169					
							0.986269					
							0.988558					
							0.990848					
92348	2	42	54	832	180	222.891	1.00001	0.01123	0.235026	7.3828	13.1836	-0.35156
							0.988558	0.007161	0.235026	7.20702	13.1836	-0.35156
							0.983979	0.009195	0.235026		13.3594	
							0.986269	0.01123	0.235026		13.3594	
							0.988558					
							0.983979					
							0.983979					
							0.988558					
92349				868	181	222.891	0.983979	0.007161	0.230957	7.20702	13.3594	0
							0.98169	0.003092	0.232991	7.20702	13.1836	0
							0.972533	-0.00301	0.232991		13.1836	
							0.970243	0.003092	0.230957		13.1836	
							0.98169					
							0.983979					
							0.979401					
							0.979401					
92350				904	180.5	222.539	0.967954	-0.00301	0.228922	6.85546	13.0078	-0.35156
							0.965664	0.005126	0.228922	6.5039	13.0078	-0.70312
							0.956507	0.007161	0.228922		12.832	
							0.94277	0.009195	0.230957		12.832	
							0.933613					
							0.938192					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUDE ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.940481					
							0.951928					
92351				940	181.5	222.539	0.951928	0.013264	0.228922	6.85546	12.832	-1.40625
							0.958796	0.009195	0.232991	7.03124	12.832	-1.40625
							0.954217	0.01123	0.232991		13.0078	
							0.974822	0.01123	0.235026		13.1836	
							0.997715					
							0.993137					
							0.988558					
							1.00458					
92352	2	42	58	976	181	222.539	1.00001	0.013264	0.23706	7.20702	13.1836	-1.40625
							0.990848	0.01123	0.232991	7.20702	13.3594	-1.75781
							0.995426	0.007161	0.232991		13.3594	
							0.986269	0.009195	0.23706		13.5351	
							0.988558					
							0.990848					
							0.990848					
							1.00458					
92353				1016	181.5	222.188	1.0229	0.009195	0.239095	7.73436	13.5351	-2.10937
							1.03435	0.007161	0.232991	7.55858	13.7109	-2.10937
							1.02977	0.007161	0.235026		13.7109	
							1.0229	0.01123	0.235026		13.7109	
							1.0229					
							1.02977					
							1.03892					
							1.01145					
92354				1052	181.5	221.836	0.993137	0.01123	0.232991	7.03124	13.7109	-2.46093
							0.997715	0.007161	0.235026	7.20702	13.7109	-3.16406
							0.995426	0.003092	0.232991		13.7109	
							0.993137	0.007161	0.232991		13.7109	
							0.997715					
							0.997715					
							1.01145					
							1.02519					
92355				1096	183	221.484	1.02061	0.009195	0.235026	7.3828	13.8867	-3.86718
							1.01374	0.013264	0.232991	7.20702	13.8867	-3.86718
							1.01374	0.009195	0.230957		13.8867	
							1.00916	0.013264	0.232991		13.8867	
							0.997715					
							1.00687					
							1.00458					
							1.00001					
92356	2	43	2	1136	183	221.133	0.997715	0.015299	0.230957	7.03124	13.8867	-3.86718
							1.00001	0.015299	0.230957	7.03124	13.8867	-3.86718
							0.995426	0.007161	0.230957		14.0625	
							1.00229	0.005126	0.230957		14.0625	
							1.00916					
							1.00001					
							1.00687					
							1.01374					
92357				1180	184	220.43	1.00687	0.007161	0.230957	7.03124	14.0625	-3.86718
							1.01145	0.01123	0.230957	7.03124	14.0625	-3.86718

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.01374	0.007161	0.228922		14.0625	
							1.01145	0.001057	0.228922		14.2383	
							1.01374					
							1.01374					
							1.00916					
							1.00687					
92358				1220	184	220.078	1.00916	0.005126	0.228922	7.03124	14.2383	-4.21874
							1.01374	0.005126	0.226888	7.03124	14.2383	-5.27343
							1.01603	0.005126	0.220784		14.2383	
							1.00916	0.007161	0.21875		14.2383	
							1.00916					
							1.00458					
							1.00001					
							0.990848					
92359				1268	184	219.375	0.972533	0.01123	0.216715	6.67968	14.0625	-6.32812
							0.979401	0.015299	0.214681	6.85546	14.0625	-6.67968
							0.972533	0.025471	0.214681		14.0625	
							0.972533	0.021403	0.214681		14.0625	
							0.986269					
							0.98169					
							0.974822					
							0.977111					
92360	2	43	6	1312	184	219.023	0.977111	0.023437	0.212646	6.67968	14.0625	-6.67968
							0.974822	0.017333	0.210612	6.85546	14.0625	-6.67968
							0.979401	0.017333	0.210612		14.0625	
							0.988558	0.015299	0.210612		14.0625	
							0.988558					
							0.986269					
							0.988558					
							0.995426					
92361				1352	183	218.32	0.970243	0.01123	0.210612	6.67968	14.0625	-7.3828
							0.974822	0.009195	0.210612	6.5039	13.8867	-8.43749
							0.967954	0.009195	0.208577		13.8867	
							0.961086	0.017333	0.208577		13.8867	
							0.949639					
							0.94735					
							0.94506					
							0.94506					
92362				1396	184	216.914	0.935903	0.019368	0.210612	6.32812	13.8867	-10.8984
							0.940481	0.015299	0.210612	6.5039	13.7109	-12.3047
							0.954217	0.019368	0.212646		13.7109	
							0.967954	0.027506	0.216715		13.8867	
							0.967954					
							0.970243					
							0.977111					
							0.997715					
92363				1440	184	215.859	1.01145	0.021403	0.216715	7.20702	13.8867	-12.6562
							1.0229	0.01123	0.216715	7.20702	13.8867	-13.3594
							1.02748	0.015299	0.21875		14.0625	
							1.0435	0.015299	0.214681		14.0625	
							1.03206					
							1.01145					



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUDE ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.993137					
							0.983979					
92364	2	43	10	1484	183.5	213.75	0.990848	0.015299	0.216715	7.03124	14.0625	-13.7109
							1.00001	0.015299	0.21875	7.55858	14.0625	-14.7656
							1.00687	0.007161	0.220784		14.2383	
							1.01374	0.009195	0.220784		14.2383	
							1.01832					
							1.02519					
							1.02519					
							1.02061					
92365				1528	183	212.344	1.01145	0.017333	0.224853	7.55858	14.414	-15.4687
							1.01145	0.013264	0.222819	7.55858	14.414	-16.1719
							1.02061	0.013264	0.224853		14.5898	
							1.01603	0.01123	0.230957		14.7656	
							1.01832					
							1.03206					
							1.06182					
							1.08471					
92366				1576	183.5	210.234	1.09387	0.013264	0.232991	8.43749	14.9414	-16.1719
							1.09387	0.015299	0.235026	8.96483	15.2929	-16.1719
							1.1099	0.007161	0.232991		15.4687	
							1.10761	0.007161	0.222819		15.4687	
							1.11676					
							1.11447					
							1.08929					
							1.0664					
92367				1624	183	208.477	1.04808	0.013264	0.224853	8.26171	15.4687	-16.1719
							1.02519	0.019368	0.222819	7.91014	15.4687	-16.1719
							1.00687	0.019368	0.220784		15.2929	
							0.993137	0.017333	0.216715		14.9414	
							0.988558					
							0.977111					
							0.970243					
							0.956507					
92368	2	43	14	1668	182.5	207.07	0.949639	0.013264	0.214681	7.3828	14.7656	-16.1719
							0.94277	0.009195	0.212646	7.03124	14.414	-16.1719
							0.933613	0.009195	0.210612		14.2383	
							0.929034	0.009195	0.210612		14.0625	
							0.924456					
							0.919877					
							0.917587					
							0.915298					
92369				1708	183	205.312	0.910719	0.007161	0.210612	6.85546	13.8867	-16.875
							0.90614	0.009195	0.210612	6.67968	13.7109	-17.9297
							0.903851	0.01123	0.210612		13.5351	
							0.903851	0.013264	0.210612		13.3594	
							0.903851					
							0.901562					
							0.903851					
							0.903851					
92370				1748	183.5	203.906	0.901562	0.013264	0.210612	6.67968	13.3594	-18.2812
							0.903851	0.013264	0.210612	6.67968	13.0078	-19.3359

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.901562	0.015299	0.210612		13.0078	
							0.90614	0.017333	0.210612		12.832	
							0.903851					
							0.901562					
							0.903851					
							0.901562					
92371				1784	184.5	202.148	0.903851	0.017333	0.210612	6.67968	12.6562	-19.6875
							0.901562	0.017333	0.210612	6.67968	12.4805	-20.039
							0.901562	0.017333	0.210612		12.3047	
							0.899272	0.017333	0.208577		12.1289	
							0.899272					
							0.899272					
							0.896983					
							0.899272					
92372	2	43	18	1816	185.5	200.742	0.892404	0.019368	0.208577	6.5039	11.9531	-20.039
							0.890115	0.019368	0.208577	6.5039	11.7773	-20.3906
							0.890115	0.015299	0.206543		11.4258	
							0.890115	0.017333	0.208577		11.4258	
							0.892404					
							0.890115					
							0.887825					
							0.883247					
92373				1844	186.5	198.984	0.885536	0.015299	0.206543	6.32812	11.0742	-20.7422
							0.887825	0.015299	0.208577	6.5039	10.8984	-20.7422
							0.890115	0.013264	0.210612		10.7226	
							0.896983	0.009195	0.210612		10.7226	
							0.903851					
							0.913009					
							0.926745					
							0.935903					
92374				1868	187.5	196.875	0.940481	0.013264	0.212646	6.85546	10.5469	-21.0937
							0.94506	0.013264	0.212646	7.20702	10.5469	-21.4453
							0.949639	0.013264	0.214681		10.3711	
							0.954217	0.013264	0.216715		10.3711	
							0.958796					
							0.965664					
							0.972533					
							0.977111					
92375				1892	188.5	194.766	0.979401	0.015299	0.216715	7.3828	10.3711	-21.7968
							0.974822	0.015299	0.216715	7.3828	10.1953	-21.7968
							0.977111	0.015299	0.216715		10.1953	
							0.98169	0.015299	0.214681		10.0195	
							0.979401					
							0.979401					
							0.977111					
							0.977111					
92376	2	43	22	1912	190	193.008	0.972533	0.015299	0.216715	7.3828	9.84374	-21.7968
							0.977111	0.015299	0.216715	7.3828	9.84374	-21.4453
							0.979401	0.013264	0.216715		9.66795	
							0.979401	0.013264	0.216715		9.66795	
							0.986269					
							0.988558					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUDE ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.986269					
							0.983979					
92377				1932	191.5	190.898	0.979401	0.015299	0.214681	7.3828	9.49217	-21.4453
							0.98169	0.013264	0.216715	7.3828	9.49217	-21.0937
							0.986269	0.01123	0.216715		9.49217	
							0.983979	0.01123	0.216715		9.49217	
							0.990848					
							0.993137					
							0.995426					
							0.997715					
92378				1948	193	189.141	1.00001	0.01123	0.216715	7.55858	9.31639	-20.7422
							1.00229	0.01123	0.216715	7.3828	9.31639	-20.3906
							1.00001	0.01123	0.216715		9.31639	
							1.00001	0.01123	0.216715		9.14061	
							0.997715					
							0.995426					
							1.00001					
							1.00001					
92379				1964	194.5	187.031	1.00229	0.009195	0.214681	7.3828	9.14061	-20.3906
							1.00001	0.009195	0.214681	7.3828	9.14061	-20.3906
							0.997715	0.009195	0.216715		8.96483	
							0.995426	0.01123	0.216715		8.96483	
							0.993137					
							0.995426					
							1.00001					
							1.00687					
92380	2	43	26	1980	196.5	185.273	1.01374	0.01123	0.220784	7.55858	8.96483	-20.3906
							1.02061	0.01123	0.220784	7.91014	9.14061	-20.3906
							1.02748	0.01123	0.224853		9.14061	
							1.04121	0.009195	0.228922		9.49217	
							1.05266					
							1.06869					
							1.08471					
							1.09387					
92381				2000	198.5	183.164	1.10532	0.01123	0.230957	8.43749	9.66795	-20.7422
							1.11447	0.015299	0.232991	8.96483	9.84374	-21.0937
							1.11905	0.017333	0.235026		10.0195	
							1.12821	0.017333	0.232991		10.1953	
							1.1305					
							1.13279					
							1.12592					
							1.11676					
92382				2020	200.5	181.406	1.11218	0.019368	0.232991	8.78905	10.1953	-21.0937
							1.10303	0.017333	0.230957	8.61327	10.1953	-21.0937
							1.10074	0.017333	0.228922		10.1953	
							1.08929	0.017333	0.228922		10.1953	
							1.08242					
							1.07784					
							1.07098					
							1.06411					
92383				2040	202	179.297	1.06182	0.017333	0.226888	8.43749	10.1953	-20.7422
							1.05724	0.015299	0.226888	8.43749	10.1953	-20.7422

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.05495	0.01123	0.228922		10.1953	
							1.05724	0.007161	0.230957		10.1953	
							1.05724					
							1.05037					
							1.05037					
							1.05495					
92384	2	43	30	2064	203.5	177.539	1.05953	0.007161	0.230957	8.43749	10.1953	-21.0937
							1.06411	0.015299	0.232991	8.61327	10.3711	-21.0937
							1.07327	0.013264	0.235026		10.3711	
							1.08242	0.015299	0.23706		10.7226	
							1.08242					
							1.08471					
							1.09387					
							1.10074					
92385				2084	205	175.43	1.10532	0.013264	0.23706	8.96483	10.7226	-21.4453
							1.1099	0.013264	0.23706	8.96483	10.7226	-21.4453
							1.11218	0.015299	0.23706		10.8984	
							1.11218	0.015299	0.23706		11.0742	
							1.11218					
							1.11218					
							1.11676					
92386				2112	206	173.672	1.11447	0.013264	0.23706	8.96483	11.0742	-21.4453
							1.11676	0.013264	0.23706	8.78905	11.0742	-21.4453
							1.11676	0.015299	0.23706		11.25	
							1.11676	0.017333	0.235026		11.25	
							1.11676					
							1.11447					
							1.11447					
							1.11218					
92387				2136	207.5	171.562	1.11218	0.015299	0.235026	8.61327	11.25	-21.0937
							1.11447	0.015299	0.23706	8.61327	11.4258	-21.0937
							1.11447	0.015299	0.235026		11.4258	
							1.11676	0.015299	0.232991		11.4258	
							1.12134					
							1.12134					
							1.12134					
							1.11676					
92388	2	43	34	2168	208.5	169.805	1.1099	0.015299	0.230957	8.61327	11.6015	-20.7422
							1.09845	0.015299	0.228922	8.26171	11.4258	-20.7422
							1.08929	0.013264	0.224853		11.4258	
							1.08242	0.01123	0.220784		11.0742	
							1.07784					
							1.06869					
							1.05495					
							1.03663					
92389				2196	209	168.047	1.01603	0.01123	0.216715	7.55858	10.8984	-20.7422
							1.00001	0.01123	0.214681	7.03124	10.8984	-20.7422
							0.983979	0.01123	0.214681		10.5469	
							0.977111	0.009195	0.212646		10.3711	
							0.967954					
							0.967954					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.972533					
							0.965664					
92390				2224	210.5	166.992	0.963375	0.01123	0.214681	6.67968	10.1953	-20.7422
							0.974822	0.015299	0.216715	7.03124	10.1953	-20.7422
							0.990848	0.015299	0.21875		10.3711	
							1.01145	0.015299	0.220784		10.5469	
							1.02748					
							1.04121					
							1.05495					
							1.06869					
92391				2252	212	164.883	1.08242	0.015299	0.222819	7.55858	10.7226	-20.7422
							1.08929	0.015299	0.226888	7.91014	10.7226	-20.3906
							1.09616	0.015299	0.226888		11.0742	
							1.10074	0.017333	0.226888		11.0742	
							1.1099					
							1.11447					
							1.11447					
							1.1099					
92392	2	43	38	2284	213.5	163.125	1.1099	0.017333	0.224853	7.91014	11.25	-20.3906
							1.1099	0.017333	0.224853	7.73436	11.25	-20.039
							1.10532	0.017333	0.222819		11.25	
							1.10532	0.015299	0.220784		11.25	
							1.10074					
							1.09845					
							1.08929					
							1.08242					
92393				2320	214.5	161.367	1.08013	0.017333	0.220784	7.55858	11.25	-20.039
							1.08929	0.015299	0.220784	7.20702	11.4258	-19.6875
							1.08929	0.017333	0.21875		11.4258	
							1.08471	0.021403	0.21875		11.4258	
							1.07555					
							1.0664					
							1.07098					
							1.07784					
92394				2352	215.5	159.609	1.08471	0.013264	0.21875	7.3828	11.4258	-19.3359
							1.09158	0.009195	0.21875	7.55858	11.6015	-18.6328
							1.10532	0.01123	0.220784		11.6015	
							1.11218	0.01123	0.21875		11.7773	
							1.11218					
							1.09845					
							1.09158					
							1.08471					
92395				2392	215.5	157.5	1.08471	0.013264	0.21875	7.3828	11.7773	-18.6328
							1.09158	0.015299	0.21875	7.3828	11.9531	-17.9297
							1.09387	0.013264	0.220784		11.9531	
							1.08929	0.007161	0.220784		12.1289	
							1.08471					
							1.09616					
							1.12134					
							1.13737					
92396	2	43	42	2432	216	155.742	1.12821	0.001057	0.220784	7.73436	12.3047	-17.5781
							1.12363	0.003092	0.220784	7.55858	12.3047	-17.5781

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.12363	0.003092	0.220784		12.4805	
							1.11447	0.003092	0.21875		12.4805	
							1.1099					
							1.10761					
							1.10303					
							1.09845					
92397				2472	216.5	154.336	1.09387	0.005126	0.216715	7.3828	12.6562	-17.2265
							1.09387	0.013264	0.216715	7.20702	12.832	-16.1719
							1.08929	0.017333	0.216715		12.832	
							1.08471	0.013264	0.21875		12.832	
							1.07555					
							1.07784					
							1.09158					
							1.10303					
92398				2520	216.5	152.93	1.10761	0.005126	0.21875	7.3828	13.1836	-15.1172
							1.12134	0.005126	0.222819	7.91014	13.3594	-14.414
							1.1305	0.005126	0.222819		13.5351	
							1.14195	0.003092	0.222819		13.7109	
							1.15797					
							1.15797					
							1.16484					
							1.16713					
92399				2572	217	151.523	1.16026	0.003092	0.222819	8.08593	13.8867	-14.414
							1.14195	0.009195	0.21875	7.73436	14.0625	-14.414
							1.1305	0.007161	0.216715		14.0625	
							1.12134	0.007161	0.214681		14.0625	
							1.10303					
							1.10303					
							1.09616					
							1.087					
92400	2	43	46	2624	216.5	150.469	1.10074	0.015299	0.214681	7.55858	14.2383	-14.414
							1.10303	0.009195	0.212646	7.03124	14.2383	-14.414
							1.08929	0.01123	0.210612		14.2383	
							1.06182	0.01123	0.206543		14.0625	
							1.05037					
							1.0435					
							1.03663					
							1.02748					
92401				2676	216.5	149.766	1.01145	0.01123	0.204508	6.5039	14.0625	-14.414
							0.993137	0.017333	0.204508	6.5039	13.8867	-14.0625
							0.993137	0.015299	0.204508		13.8867	
							0.997715	0.017333	0.204508		13.8867	
							1.00458					
							1.00687					
							0.995426					
							0.98169					
92402				2728	216	148.711	0.98169	0.017333	0.202474	6.15233	13.8867	-13.7109
							0.983979	0.015299	0.202474	6.32812	13.8867	-13.3594
							0.983979	0.013264	0.204508		13.8867	
							0.988558	0.01123	0.204508		13.8867	
							0.995426					
							1.00001					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUDE ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.997715					
							0.997715					
92403				2784	216.5	147.656	1.00001	0.009195	0.202474	6.32812	13.8867	-13.0078
							1.00001	0.005126	0.204508	6.32812	13.8867	-13.0078
							0.997715	0.003092	0.204508		13.8867	
							0.997715	0.005126	0.204508		13.8867	
							1.00687					
							1.01145					
							1.01145					
							1.00687					
92404	2	43	50	2840	217	146.602	1.01145	0.007161	0.204508	6.5039	14.0625	-13.0078
							1.01603	0.009195	0.204508	6.67968	14.0625	-13.0078
							1.01832	0.01123	0.204508		14.0625	
							1.01374	0.009195	0.204508		14.0625	
							1.01603					
							1.01832					
							1.01832					
							1.02061					
92405				2892	217	145.547	1.01832	0.007161	0.204508	6.67968	14.0625	-12.3047
							1.01374	0.003092	0.202474	6.32812	14.0625	-11.6015
							1.00916	0.005126	0.202474		14.0625	
							1.00458	0.007161	0.200439		13.8867	
							1.00001					
							0.98169					
							0.977111					
							0.983979					
92406				2948	216.5	144.844	0.98169	0.005126	0.202474	6.15233	13.8867	-10.1953
							0.990848	0.007161	0.202474	6.32812	13.8867	-9.14061
							1.00458	0.005126	0.202474		14.0625	
							1.00916	0.003092	0.202474		14.0625	
							1.00687					
							1.00229					
							0.997715					
							0.997715					
92407				3004	216.5	144.141	1.00916	0.003092	0.204508	6.32812	14.2383	-8.43749
							1.01832	0.007161	0.206543	6.85546	14.414	-8.08593
							1.02519	0.005126	0.208577		14.414	
							1.03435	0.003092	0.208577		14.5898	
							1.05037					
							1.05724					
							1.05266					
							1.05266					
92408	2	43	54	3064	216	143.438	1.04579	0.005126	0.208577	6.85546	14.7656	-8.08593
							1.04121	0.005126	0.208577	6.85546	14.7656	-8.08593
							1.02977	0.007161	0.210612		14.9414	
							1.03206	0.013264	0.212646		15.1172	
							1.03892					
							1.05266					
							1.06869					
							1.07555					
92409				3124	216	142.734	1.07784	0.01123	0.212646	7.20702	15.2929	-7.73436
							1.087	0.009195	0.214681	7.3828	15.6445	-7.3828

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.09387	0.007161	0.216715		15.8203	
							1.09845	0.009195	0.216715		15.9961	
							1.09616					
							1.10303					
							1.10303					
							1.09616					
92410				3188	214.5	142.383	1.10074	0.009195	0.214681	7.55858	16.1719	-7.3828
							1.10761	0.009195	0.214681	7.55858	16.1719	-7.03124
							1.09158	0.009195	0.210612		16.3476	
							1.087	0.01123	0.206543		16.3476	
							1.08471					
							1.0664					
							1.03892					
							1.02519					
92411				3252	214	141.68	1.01603	0.009195	0.206543	6.85546	16.1719	-7.03124
							1.00687	0.009195	0.206543	6.5039	16.1719	-7.03124
							0.997715	0.007161	0.204508		16.1719	
							0.986269	0.005126	0.204508		16.1719	
							0.98169					
							0.98169					
							0.98169					
							0.993137					
92412	2	43	58	3320	213.5	141.328	1.00458	0.007161	0.208577	6.67968	16.3476	-6.67968
							1.01832	0.007161	0.210612	7.20702	16.3476	-6.67968
							1.04121	-0.00301	0.214681		16.875	
							1.05495	0.001057	0.21875		17.2265	
							1.0664					
							1.07555					
							1.09616					
							1.10761					
92413				3392	212	140.625	1.11218	0.005126	0.222819	7.91014	17.5781	-6.67968
							1.11905	0.013264	0.224853	8.26171	17.7539	-6.67968
							1.12363	0.017333	0.224853		17.9297	
							1.12592	0.013264	0.222819		18.2812	
							1.12821					
							1.12821					
							1.1099					
							1.10303					
92414				3468	209.5	140.273	1.1099	0.001057	0.21875	7.91014	18.457	-6.67968
							1.11676	-0.02742	0.220784	8.43749	18.6328	-6.67968
							1.10303	-0.00097	0.224853		18.8086	
							1.09616	0.007161	0.228922		18.9843	
							1.11447					
							1.11676					
							1.09158					
							1.1099					
92415				3544	209.5	139.922	1.12592	0.003092	0.226888	8.78905	19.1601	-7.03124
							1.12134	0.019368	0.226888	8.26171	19.3359	-6.67968
							1.1099	0.009195	0.21875		19.3359	
							1.10074	0.003092	0.220784		19.3359	
							1.07784					
							1.05266					



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.04121					
							1.04121					
92416	2	44	2	3624	207	139.922	1.00687	0.013264	0.216715	7.91014	19.3359	-5.62499
							1.0229	0.005126	0.216715	8.08593	19.1601	-3.86718
							1.04121	0.003092	0.224853		19.3359	
							1.03435	0.009195	0.222819		19.5117	
							1.05037					
							1.06869					
							1.06411					
							1.06182					
92417				3712	206	139.57	1.05724	0.01123	0.222819	7.91014	19.6875	-2.8125
							1.05266	0.013264	0.224853	8.26171	20.039	-1.75781
							1.05495	0.01123	0.228922		20.2148	
							1.05266	0.017333	0.228922		20.5664	
							1.0664					
							1.08013					
							1.08929					
							1.10303					
92418				3796	204.5	139.57	1.10761	0.01123	0.232991	8.96483	20.7422	-1.05469
							1.11447	0.009195	0.230957	8.96483	20.9179	-0.35156
							1.12134	0.001057	0.226888		20.9179	
							1.09616	0.005126	0.222819		20.9179	
							1.09158					
							1.08242					
							1.06182					
							1.03892					
92419				3880	203	139.57	1.00916	-0.00301	0.214681	8.26171	20.5664	0
							0.977111	-0.01114	0.202474	6.67968	20.3906	0.351562
							0.933613	-0.00504	0.202474		20.039	
							0.899272	0.001057	0.204508		19.6875	
							0.864932					
							0.853485					
							0.862642					
							0.860353					
92420	2	44	6	3964	201	139.57	0.874089	0.003092	0.206543	6.5039	19.5117	0.351562
							0.896983	-0.00097	0.208577	7.20702	19.5117	0.351562
							0.90614	-0.00911	0.208577		19.5117	
							0.90614	-0.01725	0.210612		19.6875	
							0.915298					
							0.922166					
							0.90614					
							0.90843					
92421				4056	199	139.57	0.913009	-0.01521	0.210612	7.03124	19.8633	0.351562
							0.922166	-0.00504	0.214681	7.03124	20.039	0.703124
							0.929034	0.007161	0.214681		20.2148	
							0.933613	0.015299	0.222819		20.3906	
							0.924456					
							0.90843					
							0.917587					
							0.94506					
92422				4136	196.5	140.273	0.963375	0.009195	0.224853	7.91014	20.5664	1.40625
							0.974822	-0.00301	0.226888	8.78905	21.0937	2.8125

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUDE ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.974822	-0.00911	0.230957		21.2695	
							0.993137	-0.00301	0.235026		21.6211	
							1.02061					
							1.03435					
							1.03663					
							1.03892					
92423				4220	194.5	140.625	1.05266	0.001057	0.23706	9.49217	21.7968	3.86718
							1.05495	-0.00911	0.235026	9.66795	21.9726	5.27343
							1.05266	-0.01521	0.232991		22.1484	
							1.0435	-0.02132	0.226888		22.1484	
							1.05724					
							1.05495					
							1.02977					
							1.00458					
92424	2	44	10	4308	195	141.328	0.995426	-0.01521	0.220784	9.14061	21.9726	5.62499
							0.977111	-0.00708	0.224853	8.43749	21.4453	5.27343
							0.967954	-0.01725	0.216715		21.0937	
							0.963375	-0.01521	0.208577		20.9179	
							0.94735					
							0.880957					
							0.823723					
							0.79625					
92425				4388	192	142.383	0.814565	-0.01928	0.208577	7.55858	20.2148	6.32812
							0.842038	-0.02742	0.206543	6.85546	19.8633	7.03124
							0.83517	-0.01521	0.200439		19.5117	
							0.837459	-0.00504	0.204508		19.1601	
							0.821434					
							0.79854					
							0.757331					
							0.743595					
92426				4460	190	143.438	0.732148	0.013264	0.208577	6.32812	18.9843	6.67968
							0.734437	-0.01318	0.202474	6.85546	18.457	6.32812
							0.752752	-0.02132	0.204508		18.2812	
							0.791671	-0.01114	0.210612		18.2812	
							0.800829					
							0.803118					
							0.826012					
							0.821434					
92427				4532	190	144.844	0.821434	-0.00708	0.212646	7.3828	18.1054	5.62499
							0.858064	-0.00097	0.212646	7.55858	18.1054	5.62499
							0.858064	0.009195	0.212646		18.1054	
							0.846617	0.017333	0.214681		18.1054	
							0.851195					
							0.869511					
							0.8718					
							0.864932					
92428	2	44	14	4600	188.5	146.25	0.864932	0.015299	0.216715	7.73436	18.1054	5.62499
							0.855774	0.013264	0.214681	7.73436	17.7539	7.03124
							0.842038	0.009195	0.212646		17.4023	
							0.839748	0.009195	0.212646		17.4023	
							0.83517					
							0.837459					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.837459					
							0.839748					
92429				4660	188	146.953	0.848906	0.009195	0.214681	7.73436	17.0508	8.08593
							0.858064	0.009195	0.214681	7.73436	17.0508	9.14061
							0.853485	0.009195	0.212646		16.875	
							0.844327	0.009195	0.214681		16.6992	
							0.848906					
							0.853485					
							0.855774					
							0.855774					
92430				4720	187.5	148.008	0.846617	0.009195	0.214681	7.73436	16.6992	9.84374
							0.846617	0.005126	0.214681	7.91014	16.5234	10.8984
							0.858064	0.005126	0.214681		16.3476	
							0.864932	0.003092	0.214681		16.1719	
							0.864932					
							0.855774					
							0.848906					
							0.844327					
92431				4772	187	148.711	0.837459	-0.00097	0.214681	7.73436	15.8203	11.9531
							0.83288	-0.00097	0.214681	7.73436	15.6445	12.3047
							0.83517	-0.00097	0.214681		15.4687	
							0.83288	-0.00301	0.216715		15.4687	
							0.837459					
							0.848906					
							0.853485					
							0.862642					
92432	2	44	18	4824	186.5	149.414	0.878668	-0.00301	0.21875	8.08593	15.4687	12.6562
							0.890115	-0.00097	0.222819	8.96483	15.6445	12.6562
							0.901562	-0.00097	0.228922		15.6445	
							0.917587	0.001057	0.228922		15.8203	
							0.929034					
							0.94277					
							0.956507					
							0.956507					
92433				4876	186	150.82	0.951928	0.003092	0.228922	9.14061	15.9961	12.3047
							0.94506	0.005126	0.230957	9.49217	15.9961	11.9531
							0.949639	0.001057	0.230957		15.9961	
							0.951928	0.001057	0.230957		15.9961	
							0.94506					
							0.94277					
							0.940481					
							0.935903					
92434				4920	185.5	151.875	0.935903	0.003092	0.228922	9.31639	15.8203	11.6015
							0.933613	0.009195	0.230957	9.31639	15.8203	11.9531
							0.926745	0.003092	0.230957		15.8203	
							0.931324	-0.00097	0.228922		15.8203	
							0.938192					
							0.94277					
							0.94735					
							0.956507					
92435				4968	185.5	152.93	0.956507	-0.00301	0.228922	9.49217	15.8203	13.0078
							0.956507	-0.00097	0.226888	9.31639	15.6445	13.7109

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUDE ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.958796	-0.00301	0.222819		15.4687	
							0.94735	-0.00504	0.21875		15.1172	
							0.933613					
							0.917587					
							0.903851					
							0.876379					
92436	2	44	22	5008	185	153.633	0.858064	-0.00504	0.214681	8.61327	14.9414	13.7109
							0.846617	0.001057	0.214681	7.91014	14.414	13.7109
							0.83517	-0.00301	0.212646		14.0625	
							0.823723	-0.00301	0.210612		13.3594	
							0.816854					
							0.805408					
							0.800829					
							0.79854					
92437				5044	184.5	154.688	0.793961	-0.00504	0.210612	7.55858	13.1836	13.7109
							0.793961	-0.00708	0.212646	7.55858	13.0078	13.7109
							0.79854	-0.00504	0.214681		13.0078	
							0.807697	-0.00301	0.220784		13.0078	
							0.814565					
							0.826012					
							0.844327					
							0.869511					
92438				5076	185.5	155.742	0.890115	-0.00301	0.224853	8.61327	13.0078	14.0625
							0.90843	-0.00708	0.226888	9.31639	13.1836	14.414
							0.924456	-0.00301	0.232991		13.3594	
							0.938192	-0.00097	0.23706		13.5351	
							0.951928					
							0.967954					
							0.983979					
							0.993137					
92439				5112	186	157.5	1.00687	-0.00301	0.23706	10.0195	13.7109	15.4687
							1.00687	-0.00708	0.239095	10.1953	13.7109	16.5234
							1.00687	-0.00504	0.239095		13.7109	
							1.01374	-0.00708	0.239095		13.7109	
							1.01145					
							1.00687					
							1.00916					
							1.01145					
92440	2	44	26	5144	186.5	158.906	1.00687	-0.00708	0.23706	10.3711	13.7109	16.875
							1.00229	-0.00504	0.235026	10.0195	13.7109	16.875
							1.00001	-0.00708	0.235026		13.7109	
							1.00001	-0.00911	0.232991		13.5351	
							0.990848					
							0.988558					
							0.98169					
							0.970243					
92441				5172	186	160.664	0.963375	-0.01114	0.230957	9.84374	13.3594	16.875
							0.967954	-0.00504	0.232991	9.84374	13.3594	16.875
							0.965664	-0.00301	0.232991		13.1836	
							0.972533	-0.00097	0.232991		13.0078	
							0.977111					
							0.974822					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUDE ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							0.970243					
							0.970243					
92442				5204	186.5	162.422	0.972533	-0.00097	0.230957	9.66795	13.0078	16.5234
							0.961086	-0.00504	0.228922	9.31639	12.832	16.1719
							0.954217	-0.00504	0.228922		12.832	
							0.94506	-0.00301	0.230957		12.6562	
							0.940481					
							0.94735					
							0.951928					
							0.961086					
92443				5232	187	164.18	0.963375	0.001057	0.232991	9.66795	12.6562	16.1719
							0.967954	0.007161	0.232991	10.0195	12.6562	16.1719
							0.98169	0.005126	0.23706		12.6562	
							0.986269	0.005126	0.239095		12.832	
							0.986269					
							1.00916					
							1.0229					
							1.02061					
92444	2	44	30	5260	187.5	165.938	1.02748	-0.00097	0.24113	10.3711	13.0078	16.1719
							1.0435	-0.00097	0.239095	10.3711	13.0078	16.1719
							1.05037	-0.00301	0.24113		13.1836	
							1.03892	-0.00911	0.239095		13.1836	
							1.03663					
							1.04121					
							1.03663					
							1.03435					
92445				5288	188.5	167.695	1.02977	-0.00097	0.23706	10.3711	13.1836	16.1719
							1.03206	0.007161	0.239095	10.3711	13.1836	16.5234
							1.03663	0.007161	0.239095		13.0078	
							1.03435	-0.00301	0.232991		13.0078	
							1.03663					
							1.02519					
							1.01603					
							1.01145					
92446				5320	189	169.102	0.990848	-0.00708	0.228922	9.84374	12.832	17.2265
							0.993137	-0.00504	0.230957	9.66795	12.6562	17.9297
							0.988558	-0.00301	0.228922		12.6562	
							0.979401	-0.00504	0.226888		12.4805	
							0.988558					
							0.979401					
							0.970243					
							0.970243					
92447				5344	189.5	170.859	0.972533	-0.00708	0.228922	9.31639	12.4805	18.2812
							0.979401	-0.00708	0.228922	9.49217	12.3047	20.039
							0.988558	-0.00708	0.228922		12.3047	
							0.995426	-0.00708	0.226888		12.1289	
							0.993137					
							0.995426					
							0.993137					
							0.993137					
92448	2	44	34	5372	191	172.266	1.00001	-0.00911	0.228922	9.49217	12.1289	21.4453
							1.00229	-0.00911	0.226888	9.14061	12.1289	22.8515

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.01145	-0.01725	0.224853		11.9531	
							0.993137	-0.01521	0.224853		11.9531	
							0.979401					
							0.988558					
							0.990848					
							0.983979					
92449				5396	192	174.727	0.990848	-0.01521	0.224853	9.31639	11.7773	23.5547
							0.988558	-0.01521	0.224853	9.31639	11.6015	24.2578
							0.993137	-0.01928	0.224853		11.6015	
							0.997715	-0.01725	0.222819		11.4258	
							0.993137					
							0.988558					
							0.993137					
							0.988558					
92450				5420	193.5	176.484	0.974822	-0.01521	0.220784	9.14061	11.25	24.6093
							0.963375	-0.02132	0.220784	8.96483	11.0742	26.0156
							0.963375	-0.01928	0.220784		11.0742	
							0.963375	-0.01928	0.220784		10.8984	
							0.958796					
							0.963375					
							0.972533					
							0.974822					
92451				5436	195	179.648	0.979401	-0.02335	0.220784	8.96483	10.7226	27.7734
							0.98169	-0.02539	0.220784	8.96483	10.3711	31.6406
							0.988558	-0.02539	0.220784		10.1953	
							1.00001	-0.02945	0.220784		10.1953	
							0.997715					
							0.995426					
							1.00229					
							1.00229					
92452	2	44	38	5452	196.5	182.812	1.00001	-0.03149	0.21875	8.96483	9.84374	35.1562
							0.997715	-0.03352	0.21875	8.78905	9.66795	38.6718
							0.997715	-0.03556	0.21875		9.49217	
							1.00001	-0.03556	0.21875		9.14061	
							1.00001					
							1.00001					
							1.00229					
							1.00916					
92453				5460	198.5	186.328	1.00687	-0.03352	0.21875	8.78905	8.78905	40.0781
							1.00916	-0.03352	0.21875	8.96483	8.26171	42.539
							1.01374	-0.02742	0.222819		8.08593	
							1.0229	-0.01928	0.220784		7.91014	
							1.04121					
							1.06411					
							1.07098					
							1.06411					
92454				5464	200.5	190.547	1.06411	-0.02335	0.222819	8.96483	7.3828	43.2421
							1.0664	-0.01928	0.226888	9.14061	7.03124	42.1874
							1.05953	-0.01114	0.228922		6.85546	
							1.06411	-0.00911	0.232991		6.67968	
							1.08013					
							1.08929					

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUDE ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.09845					
							1.12134					
92455				5468	202.5	194.766	1.13508	-0.01318	0.239095	9.66795	6.5039	41.8359
							1.15339	-0.00504	0.243164	10.3711	6.32812	43.5937
							1.17629	-0.00097	0.249267		6.32812	
							1.1946	-0.00301	0.253337		6.15233	
							1.21521					
							1.23352					
							1.25413					
							1.26557					
92456	2	44	42	5460	205.5	200.742	1.2816	-0.00301	0.255371	10.8984	5.97655	46.4062
							1.29305	-0.00911	0.253337	10.8984	5.80077	49.5702
							1.29534	-0.00911	0.253337		5.62499	
							1.3022	-0.01114	0.249267		5.44921	
							1.31594					
							1.31594					
							1.3022					
							1.29534					
92457				5452	207.5	205.312	1.27931	-0.01318	0.245198	10.5469	5.09765	51.6796
							1.27015	-0.01521	0.243164	10.3711	4.57031	52.3827
							1.27473	-0.01114	0.245198		4.39453	
							1.27702	-0.00911	0.24113		4.21874	
							1.27702					
							1.28847					
							1.27931					
							1.26328					
92458				5432	209.5	210.586	1.25184	-0.00911	0.239095	10.1953	3.51562	53.0859
							1.24497	-0.00708	0.23706	9.66795	3.16406	53.4374
							1.24955	-0.00911	0.235026		2.63671	
							1.24497	-0.00097	0.23706		2.28515	
							1.23352					
							1.2381					
							1.25184					
							1.26099					
92459				5408	212	215.156	1.26099	0.001057	0.23706	9.66795	1.75781	55.1952
							1.26328	0.001057	0.239095	9.49217	1.05469	56.2499
							1.25413	-0.00301	0.245198		0.878905	
							1.23352	-0.00097	0.255371		0.527343	
							1.2175					
							1.27015					
							1.34799					
							1.40751					
92460	2	44	46	5380	215	222.188	1.46017	-0.00301	0.269613	10.7226	0.527343	58.0077
							1.48306	-0.01114	0.277751	11.9531	0.351562	60.1171
							1.50367	-0.00301	0.289958		0	
							1.52656	0.001057	0.296061		-0.17578	
							1.54488					
							1.55861					
							1.58609					
							1.60898					
92461				5332	218.5	229.219	1.64332	-0.00097	0.298096	12.4805	-0.52734	63.6327
							1.65706	-0.00708	0.291992	12.4805	-1.05469	65.3905

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUDE ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.65935	-0.00708	0.283854		-1.58203	
							1.63874	-0.00301	0.271647		-2.8125	
							1.60669					
							1.58838					
							1.56548					
							1.5403					
92462				5276	222	235.898	1.53114	-0.00301	0.269613	11.4258	-3.51562	68.9062
							1.52885	0.009195	0.269613	11.0742	-4.04296	71.3671
							1.53343	0.001057	0.265544		-5.09765	
							1.54488	-0.00911	0.269613		-5.62499	
							1.52198					
							1.51283					
							1.52656					
							1.54946					
92463				5204	225.5	242.578	1.57922	-0.00504	0.279785	10.8984	-6.15233	73.1249
							1.6044	-0.01318	0.289958	12.1289	-6.85546	74.1796
							1.6479	-0.01318	0.298096		-7.3828	
							1.70284	-0.01114	0.296061		-8.61327	
							1.74405					
							1.7761					
							1.78984					
							1.77839					
92464	2	44	50	5096	230.5	251.367	1.76237	-0.00911	0.28182	12.3047	-9.49217	77.6952
							1.73261	-0.00911	0.25944	10.7226	-9.84374	80.5077
							1.68224	-0.00504	0.239095		-11.7773	
							1.63187	-0.00301	0.222819		-12.6562	
							1.58151					
							1.52198					
							1.46246					
							1.40751					
92465				4972	236.5	255.586	1.36402	-0.00097	0.208577	8.43749	-13.7109	83.3202
							1.31594	0.003092	0.198405	6.5039	-15.2929	84.7264
							1.27931	0.009195	0.192301		-16.3476	
							1.25871	0.013264	0.190267		-18.457	
							1.24726					
							1.24039					
							1.24039					
							1.24726					
92466				4816	244.5	260.508	1.25184	0.015299	0.188232	6.15233	-19.3359	87.1874
							1.25871	0.017333	0.188232	5.80077	-20.7422	89.2967
							1.26328	0.017333	0.184163		-22.6757	
							1.27473	0.021403	0.180094		-23.7304	
							1.28618					
							1.29534					
							1.29762					
							1.27931					
92467				4628	254	265.078	1.27702	0.025471	0.17806	5.44921	-25.1367	91.4061
							1.27473	0.017333	0.173991	4.57031	-26.0156	92.8124
							1.26099	0.025471	0.165853		-27.0703	
							1.24726	0.02954	0.161784		-28.8281	
							1.22894					
							1.22665					



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUDE ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.20605					
							1.18544					
92468	2	44	54	4388	264.5	270	1.174	0.03361	0.157715	3.86718	-29.707	95.2733
							1.14424	0.027506	0.151611	3.33984	-30.2343	96.6796
							1.14653	0.021403	0.145508		-31.1132	
							1.14881	0.021403	0.141439		-31.8164	
							1.14195					
							1.1511					
							1.15339					
							1.14195					
92469				4124	275.5	273.516	1.12821	0.019368	0.13737	2.98828	-33.0468	98.0858
							1.11676	0.019368	0.133301	2.10937	-33.9257	99.8436
							1.08929	0.025471	0.129232		-34.8046	
							1.06182	0.031575	0.127197		-36.5624	
							1.04121					
							1.01145					
							0.990848					
							0.979401					
92470				3820	289.5	277.031	0.965664	0.039713	0.123128	1.23047	-36.914	103.008
							0.958796	0.041747	0.119059	0.703124	-37.7929	105.469
							0.94735	0.047851	0.112956		-39.5507	
							0.954217	0.062093	0.104818		-40.2538	
							0.967954					
							0.983979					
							0.972533					
							0.915298					
92471				3508	306.5	279.844	0.826012	0.047851	0.104818	0	-41.3085	107.578
							0.718411	0.041747	0.112956	-2.63671	-41.6601	110.039
							0.560444	0.027506	0.108887		-42.0117	
							0.445975	0.047851	0.094645		-43.0663	
							0.365847					
							0.31777					
							0.352111					
							0.372715					
92472	2	44	58	3068	317.5	281.602	0.489473	0.009195	0.060059	-2.28515	-43.2421	111.094
							0.633704	-0.02742	0.053955	0.527343	-43.9452	98.0858
							0.752752	-0.0437	0.060059		-45.1757	
							0.855774	-0.03149	0.086507		-45.5273	
							0.997715					
							1.14195					
							1.2816					
							1.39607					
92473				2640	334	290.391	1.50596	-0.02335	0.100749	2.98828	-45.7031	78.7499
							1.52656	0.025471	0.092611	3.16406	-45.8788	60.4687
							1.54259	0.062093	0.102783		-45.8788	
							1.57006	0.098714	0.104818		-45.8788	
							1.58609					
							1.61585					
							1.65706					
							1.67079					
92474				2216	352	298.477	1.67079	0.121094	0.094645	2.8125	-45.7031	54.1405
							1.64561	0.123128	0.072266	2.28515	-45.3515	49.5702

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTE AIRSPD (KNOTS)	MAGNETIC HEADING EFIS (DEG)	VERT ACCEL (G's)	LATERAL ACCEL (G's)	LONGITUD ACCEL (G's)	AOA (DEG)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)
							1.67766	0.117025	0.045817		-44.9999	
							1.72116	0.100749	0.021403		-44.6484	
							1.68682					
							1.70056					
							1.68911					
							1.68911					
92475				1748	368.5	302.695	1.74176	0.098714	-0.00911	2.28515	-44.121	48.164
							1.79213	0.0743	-0.02132	3.33984	-43.4179	37.9687
							1.82189	0.055989	-0.03149		-42.7148	
							1.89057	0.049886	-0.02336		-41.4843	
							1.93178					
							1.94552					
							1.9501					
							1.97757					
92476	2	45	2	1320	382.5	306.914	2.08059	0.064127	-0.02336	3.51562	-40.6054	30.2343
							2.16759	0.094645	-0.03556	3.33984	-39.0234	22.8515
							2.25687	0.108887	-0.0498		-38.3203	
							2.25916	0.09261	-0.06608		-37.9687	
							2.23398					
							2.21338					
							2.14698					
							2.11722					
92477				904	395	309.023	2.09891	0.070231	-0.08236	2.63671	-36.914	23.9062
							2.07372	0.03361	-0.08032	3.51562	-36.2109	18.2812
							2.09662	0.049886	-0.06201		-35.332	
							2.21338	0.080403	-0.06405		-33.75	
							2.30953					
							2.41942					
							2.45605					
							2.43316					
92478				524	410	311.133	2.54534	0.106852	-0.0498	3.51562	-32.6953	14.0625
							2.60257	0.147542	-0.02336	4.92187	-30.5859	14.414
							2.72849	0.149577	0.017334		-29.8828	
							2.99405	0.131266	0.031575		-29.0039	
							3.30312					
							3.48169					
							3.69232					
							3.81594					
92479				180	416	315.703	3.96246	0.131266	0.023437	6.85546	-25.4882	19.3359
							3.8892	0.082438	-0.01929	5.44921	-24.4336	24.6093
							3.70147	0.076334	-0.05794		-23.7304	
							3.51832	0.117025	-0.07625		-23.2031	
							3.28023					
							3.05358					
							2.93224					
							2.76741					
92480												









Time	GMT	GMT	GMT	ALTITUDE	COMPUTED	ALT FLAPS	LE FLAP 1	LE FLAP 1	LE FLAP 2	LE FLAP 2	LE FLAP 3	LE FLAP 3	LE FLAP 4	LE FLAP 4	LE FLAP 5	LE FLAP 5	TOO LOW FLAP	GEAR DN R	GEAR DN L	NOSE GEAR DN	LE SLAT 1 FULL	LE SLAT 1 FULL	LE SLAT 1 MID	LE SLAT 2 FULL	LE SLAT 2 FULL	LE SLAT 2 MID	LE SLAT 3 FULL	LE SLAT 3 FULL	LE SLAT 3 MID	LE SLAT 4 FULL	LE SLAT 4 FULL	LE SLAT 4 MID	LE SLAT 5 FULL	LE SLAT 5 FULL	LE SLAT 5 MID	LE SLAT 6 FULL	LE SLAT 6 FULL	LE SLAT 6 MID																						
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(99 99)	(AIR SPD)	(I-ARMED)	(0-EXTEND)	(0-INTRNS)	(0-EXTEND)	(0-INTRNS)	(0-EXTEND)	(0-INTRNS)	(0-EXTEND)	(0-INTRNS)	(0-EXTEND)	(0-INTRNS)	(1-TRUE)	(0-DOWN)	(0-DOWN)	(0-DOWN)	(0-FULEXT)	(0-INTRNS)	(0-MIDEXT)	(0-FULEXT)	(0-INTRNS)	(0-MIDEXT)	(0-FULEXT)	(0-INTRNS)	(0-MIDEXT)	(0-FULEXT)	(0-INTRNS)	(0-MIDEXT)	(0-FULEXT)	(0-INTRNS)	(0-MIDEXT)	(0-FULEXT)	(0-INTRNS)	(0-MIDEXT)																						
92451				5276	222												0																																											
92453				5204	225.5												0																																											
92461	2	42		5096	230.5												0																																											
92463				4972	236.5																																																							
92465				4818	244.5													0																																										
92467				4620	254																																																							
92469	2	44		4388	264.5																																																							
92471				4154	272.5													0																																										
92473				3920	289.5																																																							
92475				3508	306.5													0																																										
92477	2	44		3068	317.5																																																							
92479				2640	334																																																							
92481				2218	355																																																							
92483				1748	368.5																																																							
92485	2	45		1320	382.5													0																																										
92487				924	395																																																							
92489				524	410																																																							
92491				180	418																																																							
92493																																																												















Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	OIL PRES L	OIL PRES R	HYD OIL PRES A	HYD OIL PRES B	EVENT MARKER (RESV)	RADIO HEIGHT EFIS	SINK RATE	DONT SINK	PULL UP	TERRAIN PULL UP	TERRAIN	TOO LOW TERRAIN	TOO LOW GEAR	TOO LOW FLAP	G/S DEV EFIS	G/S ENGA FCC	G/S GPWS	MINIMUMS	WINDSHEAR	WINDSHEAR CAUTN	
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(PSI)	(PSI)	(PSI)	(PSI)	(0-EVENT 1-.)	(FEET)	(0- 1-TRUE)	(0- 1-TRUE)	(0- 1-TRUE)	(0-FALSE 1-TRUE)	(0- 1-TRUE)	(0- 1-TRUE)	(0- 1-TRUE)	(0- 1-TRUE)	(DDM)	(0- 1-ENGA)	(0- 1-TRUE)	(0- 1-TRUE)	(0-FALSE 1-TRUE)	(0-FALSE 1-TRUE)	
92428	2	44	14	4600	188.5						2630										-0.24218				FALSE	FALSE
92429				4660	188		2				2630										-0.24218				FALSE	FALSE
92430				4720	187.5						2630										-0.24218				FALSE	FALSE
92431				4772	187	2			3248		2630										-0.24218				FALSE	FALSE
92432	2	44	18	4824	186.5						2630										-0.24218				FALSE	FALSE
92433				4876	186			2			2630										-0.24218				FALSE	FALSE
92434				4920	185.5						2630										-0.24218				FALSE	FALSE
92435				4968	185.5	2					2630										-0.24218				FALSE	FALSE
92436	2	44	22	5008	185						2630										-0.24218				FALSE	FALSE
92437				5044	184.5			2			2630										-0.24218				FALSE	FALSE
92438				5076	185.5						2630										-0.24218				FALSE	FALSE
92439				5112	186	2					2630										-0.24218				FALSE	FALSE
92440	2	44	26	5144	186.5						2630										-0.24218				FALSE	FALSE
92441				5172	186			2			2630										-0.24218				FALSE	FALSE
92442				5204	186.5						2630										-0.24218				FALSE	FALSE
92443				5232	187	2					2630										-0.24218				FALSE	FALSE
92444	2	44	30	5260	187.5						2630										-0.24218				FALSE	FALSE
92445				5288	188.5			2			2630										-0.24218				FALSE	FALSE
92446				5320	189						2630										-0.24218				FALSE	FALSE
92447				5344	189.5	2					2630										-0.24218				FALSE	FALSE
92448	2	44	34	5372	191						2630										-0.24218				FALSE	FALSE
92449				5396	192			2			2630										-0.24218				FALSE	FALSE
92450				5420	193.5						2630										-0.24218				FALSE	FALSE
92451				5436	195		2				2630										-0.24218				FALSE	FALSE
92452	2	44	38	5452	196.5						2630										-0.24218				FALSE	FALSE
92453				5460	198.5			2			2630										-0.24218				FALSE	FALSE
92454				5464	200.5						2630										-0.24218				FALSE	FALSE
92455				5468	202.5	2					2630										-0.24218				FALSE	FALSE
92456	2	44	42	5460	205.5						2630										-0.24218				FALSE	FALSE
92457				5452	207.5			2			2630										-0.24218				FALSE	FALSE
92458				5432	209.5						2630										-0.24218				FALSE	FALSE
92459				5408	212	2					2630										-0.24218				FALSE	FALSE
92460	2	44	46	5380	215						2630										-0.24218				FALSE	FALSE
92461				5332	218.5			2			2630										-0.24218				FALSE	FALSE
92462				5276	222						2630										-0.24218				FALSE	FALSE
92463				5204	225.5	2					2630										-0.24218				FALSE	FALSE
92464	2	44	50	5096	230.5						2630										-0.24218				FALSE	FALSE
92465				4972	236.5			2	3300		2630										-0.24218				FALSE	FALSE
92466				4816	244.5						2630										-0.24218				FALSE	FALSE
92467				4628	254	2					2630										-0.24218				FALSE	FALSE
92468	2	44	54	4388	264.5						2630										-0.24218				FALSE	FALSE
92469				4124	275.5			2			2630										-0.24218				FALSE	FALSE
92470				3820	289.5						2630										-0.24218				FALSE	FALSE
92471				3508	306.5			2			2630										-0.24218				FALSE	FALSE
92472	2	44	58	3068	317.5						2630										-0.24218				FALSE	FALSE
92473				2640	334			2			2630										-0.24218				FALSE	FALSE
92474				2216	352						2630										-0.24218				FALSE	FALSE
92475				1748	368.5	2					2630										-0.24218				FALSE	FALSE
92476	2	45	2	1320	382.5						1530										-0.24218				FALSE	FALSE
92477				904	395			2			1234										-0.24218				FALSE	FALSE
92478				524	410						791										-0.24218				FALSE	FALSE
92479				180	416	2					421										-0.24218				FALSE	FALSE















Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	N1 L	N1 R	N2 L	N2 R	FUEL FLOW L	FUEL FLOW R	ENG 1 T/R L SLV DEPLOYED	ENG 1 T/R L SLV NOT STWD	ENG 1 T/R R SLV DEPLOYED	ENG 1 T/R R SLV NOT STWD	ENG 2 T/R L SLV DEPLOYED	ENG 2 T/R L SLV NOT STWD	ENG 2 T/R R SLV DEPLOYED	ENG 2 T/R R SLV NOT STWD	ENG 1 FIRE	ENG 2 FIRE	APU FIRE	THR LEVER ANGLE L (DEG)	THR LEVER ANGLE R (DEG)	ENG OIL QUANT L (PINTS)	ENG OIL QUANT R (PINTS)	OIL PRES L (PSI)	OIL PRES R (PSI)	
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(%RPM)	(%RPM)	(%RPM)	(%RPM)	(PPH)	(PPH)	(0-DEPLOY 1-)	(0-UNLOCK 1-)	(0-DEPLOY 1-)	(0-UNLOCK 1-)	(0-DEPLOY 1-)	(0-UNLOCK 1-)	(0-DEPLOY 1-)	(0-UNLOCK 1-)	(0- 1-FIRE)	(0- 1-FIRE)	(0- 1-FIRE)							
92465				4972	236.5	15.875	89				6816												43.5937					2	
92466				4816	244.5	0	89	96																42.7148					
92467				4628	254	15.875	89.625			7040													43.7695					2	
92468	2	44	54	4388	264.5	0	89.875			96.75														44.121					
92469				4124	275.5	15.875	90				7200													43.9452					2
92470				3820	289.5	0	89.875	96.125																	44.121				
92471				3508	306.5	15.875	89.875			7280														44.2968					2
92472	2	44	58	3068	317.5	0	89.625			96.75															43.9452				
92473				2640	334	15.875	89.125				7456													44.6484					2
92474				2216	352	0	87.5	95.875																	43.9452				
92475				1748	368.5	15.875	77.125			6160														31.289					2
92476	2	45	2	1320	382.5	0	63.375			90.5															19.3359				
92477				904	395	15.875	55.75				3168													2.28515					2
92478				524	410	0	51.5	86.25																	2.98828				
92479				180	416	15.875	48.375			2128														5.27343					2

# Flash Air B737-300 Accident

# Preliminary Data Created: January 23 2004

# MCA

Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	HF L KEYING	HF R KEYING	VHF C KEYING	VHF L KEYING	VHF R KEYING	A/P WARN
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-WARN 1-.)
91864	2	34	50	216	45	.	.	.	.	.	.
91865				216	45	.	.	.	.	.	.
91866				216	45	.	.	.	.	.	.
91867				216	45	.	.	.	.	.	.
91868	2	34	54	216	45	.	.	.	.	.	.
91869				216	45	.	.	.	.	.	.
91870				216	45	.	.	.	.	.	.
91871				216	45	.	.	.	.	.	.
91872	2	34	58	216	45	.	.	.	.	.	.
91873				216	45	.	.	.	.	.	.
91874				216	45	.	.	.	.	.	.
91875				216	45	.	.	.	.	.	.
91876	2	35	2	216	45	.	.	.	.	.	.
91877				216	45	.	.	.	.	.	.
91878				216	45	.	.	.	.	.	.
91879				216	45	.	.	.	.	.	.
91880	2	35	6	216	45	.	.	.	.	.	.
91881				216	45	.	.	.	.	.	.
91882				216	45	.	.	.	.	.	.
91883				216	45	.	.	.	.	.	.
91884	2	35	10	216	45	.	.	.	.	.	.
91885				216	45	.	.	.	.	.	.
91886				216	45	.	.	.	.	.	.
91887				216	45	.	.	.	.	.	.
91888	2	35	14	216	45	.	.	.	.	.	.
91889				216	45	.	.	.	.	.	.
91890				216	45	.	.	.	.	.	.
91891				216	45	.	.	.	.	.	.
91892	2	35	18	216	45	.	.	.	.	.	.
91893				216	45	.	.	.	.	.	.
91894				216	45	.	.	.	.	.	.
91895				216	45	.	.	.	.	.	.
91896	2	35	22	216	45	.	.	.	.	.	.
91897				216	45	.	.	.	.	.	.
91898				216	45	.	.	.	.	.	.

Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	HF L KEYING	HF R KEYING	VHF C KEYING	VHF L KEYING	VHF R KEYING	A/P WARN
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-WARN 1-.)
91899				216	45	.	.	.	.	.	.
91900	2	35	26	216	45	.	.	.	.	.	.
91901				216	45	.	.	.	.	.	.
91902				216	45	.	.	.	.	.	.
91903				216	45	.	.	.	.	.	.
91904	2	35	30	216	45	.	.	.	.	.	.
91905				216	45	.	.	.	.	.	.
91906				216	45	.	.	.	.	.	.
91907				216	45	.	.	.	.	.	.
91908	2	35	34	216	45	.	.	.	.	.	.
91909				216	45	.	.	.	.	.	.
91910				216	45	.	.	.	.	.	.
91911				216	45	.	.	.	.	.	.
91912	2	35	38	216	45	.	.	.	.	.	.
91913				216	45	.	.	.	.	.	.
91914				216	45	.	.	.	.	.	.
91915				216	45	.	.	.	.	.	.
91916	2	35	42	216	45	.	.	.	.	.	.
91917				216	45	.	.	.	.	.	.
91918				216	45	.	.	.	.	.	.
91919				216	45	.	.	.	.	.	.
91920	2	35	46	216	45	.	.	.	.	.	.
91921				216	45	.	.	.	.	.	.
91922				216	45	.	.	.	.	.	.
91923				216	45	.	.	.	.	.	.
91924	2	35	50	216	45	.	.	.	.	.	.
91925				216	45	.	.	.	.	.	.
91926				216	45	.	.	.	.	.	.
91927				216	45	.	.	.	.	.	.
91928	2	35	54	216	45	.	.	.	.	.	.
91929				216	45	.	.	.	.	.	.
91930				216	45	.	.	.	.	.	.
91931				216	45	.	.	.	.	.	.
91932	2	35	58	216	45	.	.	.	.	.	.
91933				216	45	.	.	.	.	.	.
91934				216	45	.	.	.	.	.	.
91935				216	45	.	.	.	.	.	.
91936	2	36	2	216	45	.	.	.	.	.	.
91937				216	45	.	.	.	.	.	.
91938				216	45	.	.	.	.	.	.
91939				216	45	.	.	.	.	.	.

Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	HF L KEYING	HF R KEYING	VHF C KEYING	VHF L KEYING	VHF R KEYING	A/P WARN
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-WARN 1-.)
91940	2	36	6	216	45	.	.	.	.	.	.
91941				216	45	.	.	.	.	.	.
91942				216	45	.	.	.	.	.	.
91943				216	45	.	.	.	.	.	.
91944	2	36	10	216	45	.	.	.	.	.	.
91945				216	45	.	.	.	.	.	.
91946				216	45	.	.	.	.	.	.
91947				216	45	.	.	.	.	.	.
91948	2	36	14	216	45	.	.	.	.	.	.
91949				216	45	.	.	.	.	.	.
91950				216	45	.	.	.	.	.	.
91951				216	45	.	.	.	.	.	.
91952	2	36	18	216	45	.	.	.	.	.	.
91953				216	45	.	.	.	.	.	.
91954				216	45	.	.	.	.	.	.
91955				216	45	.	.	.	.	.	.
91956	2	36	22	216	45	.	.	.	.	.	.
91957				216	45	.	.	.	.	.	.
91958				216	45	.	.	.	.	.	.
91959				216	45	.	.	.	.	.	.
91960	2	36	26	216	45	.	.	.	.	.	.
91961				216	45	.	.	.	.	.	.
91962				216	45	.	.	.	.	.	.
91963				216	45	.	.	.	.	.	.
91964	2	36	30	216	45	.	.	.	.	.	.
91965				216	45	.	.	.	.	.	.
91966				216	45	.	.	.	.	.	.
91967				216	45	.	.	.	.	.	.
91968	2	36	34	216	45	.	.	.	.	.	.
91969				216	45	.	.	.	.	.	.
91970				216	45	.	.	.	.	.	.
91971				216	45	.	.	.	.	.	.
91972	2	36	38	216	45	.	.	.	.	.	.
91973				216	45	.	.	.	.	.	.
91974				216	45	.	.	.	.	.	.
91975				216	45	.	.	.	.	.	.
91976	2	36	42	216	45	.	.	.	KEYED	.	.
91977				216	45	.	.	.	KEYED	.	.
91978				216	45	.	.	.	.	.	.
91979				216	45	.	.	.	.	.	.
91980	2	36	46	216	45	.	.	.	.	.	.



Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	HF L KEYING	HF R KEYING	VHF C KEYING	VHF L KEYING	VHF R KEYING	A/P WARN
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-WARN 1-.)
91981				216	45	.	.	.	.	.	.
91982				216	45	.	.	.	.	.	.
91983				216	45	.	.	.	.	.	.
91984	2	36	50	216	45	.	.	.	.	.	.
91985				216	45	.	.	.	.	.	.
91986				216	45	.	.	.	.	.	.
91987				216	45	.	.	.	.	.	.
91988	2	36	54	216	45	.	.	.	.	.	.
91989				216	45	.	.	.	.	.	.
91990				216	45	.	.	.	KEYED	.	.
91991				216	45	.	.	.	KEYED	.	.
91992	2	36	58	216	45	.	.	.	KEYED	.	.
91993				216	45	.	.	.	KEYED	.	.
91994				216	45	.	.	.	KEYED	.	.
91995				216	45	.	.	.	.	.	.
91996	2	37	2	216	45	.	.	.	.	.	.
91997				216	45	.	.	.	.	.	.
91998				216	45	.	.	.	.	.	.
91999				216	45	.	.	.	.	.	.
92000	2	37	6	216	45	.	.	.	.	.	.
92001				216	45	.	.	.	.	.	.
92002				216	45	.	.	.	.	.	.
92003				216	45	.	.	.	.	.	.
92004	2	37	10	216	45	.	.	.	.	.	.
92005				216	45	.	.	.	.	.	.
92006				216	45	.	.	.	.	.	.
92007				216	45	.	.	.	.	.	.
92008	2	37	14	216	45	.	.	.	.	.	.
92009				216	45	.	.	.	.	.	.
92010				216	45	.	.	.	.	.	.
92011				216	45	.	.	.	.	.	.
92012	2	37	18	216	45	.	.	.	.	.	.
92013				216	45	.	.	.	.	.	.
92014				216	45	.	.	.	.	.	.
92015				216	45	.	.	.	.	.	.
92016	2	37	22	216	45	.	.	.	.	.	.
92017				216	45	.	.	.	.	.	.
92018				216	45	.	.	.	.	.	.
92019				216	45	.	.	.	.	.	.
92020	2	37	26	216	45	.	.	.	.	.	.
92021				216	45	.	.	.	.	.	.

Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	HF L KEYING	HF R KEYING	VHF C KEYING	VHF L KEYING	VHF R KEYING	A/P WARN
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-WARN 1-.)
92022				216	45	.	.	.	.	.	.
92023				216	45	.	.	.	.	.	.
92024	2	37	30	216	45	.	.	.	.	.	.
92025				216	45	.	.	.	.	.	.
92026				216	45	.	.	.	.	.	.
92027				216	45	.	.	.	.	.	.
92028	2	37	34	216	45	.	.	.	.	.	.
92029				216	45	.	.	.	.	.	.
92030				216	45	.	.	.	.	.	.
92031				216	45	.	.	.	.	.	.
92032	2	37	38	216	45	.	.	.	.	.	.
92033				216	45	.	.	.	.	.	.
92034				216	45	.	.	.	.	.	.
92035				216	45	.	.	.	.	.	.
92036	2	37	42	216	45	.	.	.	.	.	.
92037				216	45	.	.	.	.	.	.
92038				216	45	.	.	.	.	.	.
92039				216	45	.	.	.	.	.	.
92040	2	37	46	216	45	.	.	.	.	.	.
92041				216	45	.	.	.	.	.	.
92042				216	45	.	.	.	.	.	.
92043				216	45	.	.	.	.	.	.
92044	2	37	50	216	45	.	.	.	.	.	.
92045				216	45	.	.	.	.	.	.
92046				216	45	.	.	.	.	.	.
92047				216	45	.	.	.	.	.	.
92048	2	37	54	216	45	.	.	.	.	.	.
92049				216	45	.	.	.	.	.	.
92050				216	45	.	.	.	.	.	.
92051				216	45	.	.	.	.	.	.
92052	2	37	58	216	45	.	.	.	.	.	.
92053				216	45	.	.	.	.	.	.
92054				216	45	.	.	.	.	.	.
92055				216	45	.	.	.	.	.	.
92056	2	38	2	216	45	.	.	.	.	.	.
92057				216	45	.	.	.	.	.	.
92058				216	45	.	.	.	.	.	.
92059				216	45	.	.	.	.	.	.
92060	2	38	6	216	45	.	.	.	KEYED	.	.
92061				216	45	.	.	.	.	.	.
92062				212	45	.	.	.	.	.	.

Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	HF L KEYING	HF R KEYING	VHF C KEYING	VHF L KEYING	VHF R KEYING	A/P WARN
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-WARN 1-.)
92063				216	45	.	.	.	.	.	.
92064	2	38	10	212	45	.	.	.	.	.	.
92065				212	45	.	.	.	.	.	.
92066				212	45	.	.	.	.	.	.
92067				212	45	.	.	.	.	.	.
92068	2	38	14	212	45	.	.	.	.	.	.
92069				212	45	.	.	.	.	.	.
92070				212	45	.	.	.	.	.	.
92071				212	45	.	.	.	KEYED	.	.
92072	2	38	18	212	45	.	.	.	KEYED	.	.
92073				212	45	.	.	.	KEYED	.	.
92074				212	45	.	.	.	KEYED	.	.
92075				212	45	.	.	.	KEYED	.	.
92076	2	38	22	212	45	.	.	.	KEYED	.	.
92077				208	45	.	.	.	KEYED	.	.
92078				208	45	.	.	.	KEYED	.	.
92079				208	45	.	.	.	KEYED	.	.
92080	2	38	26	208	45	.	.	.	.	.	.
92081				208	45	.	.	.	.	.	.
92082				208	45	.	.	.	.	.	.
92083				208	45	.	.	.	.	.	.
92084	2	38	30	208	45	.	.	.	.	.	.
92085				208	45	.	.	.	KEYED	.	.
92086				208	45	.	.	.	.	.	.
92087				208	45	.	.	.	.	.	.
92088	2	38	34	208	45	.	.	.	.	.	.
92089				208	45	.	.	.	.	.	.
92090				208	45	.	.	.	KEYED	.	.
92091				208	45	.	.	.	KEYED	.	.
92092	2	38	38	208	45	.	.	.	.	.	.
92093				208	45	.	.	.	.	.	.
92094				208	45	.	.	.	.	.	.
92095				208	45	.	.	.	.	.	.
92096	2	38	42	208	45	.	.	.	.	.	.
92097				208	45	.	.	.	.	.	.
92098				208	45	.	.	.	.	.	.
92099				208	45	.	.	.	.	.	.
92100	2	38	46	208	45	.	.	.	.	.	.
92101				208	45	.	.	.	.	.	.
92102				208	45	.	.	.	.	.	.
92103				204	45	.	.	.	.	.	.

Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	HF L KEYING	HF R KEYING	VHF C KEYING	VHF L KEYING	VHF R KEYING	A/P WARN
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-WARN 1-.)
92104	2	38	50	204	45	.	.	.	.	.	.
92105				204	45	.	.	.	.	.	.
92106				204	45	.	.	.	.	.	.
92107				204	45	.	.	.	.	.	.
92108	2	38	54	204	45	.	.	.	.	.	.
92109				204	45	.	.	.	.	.	.
92110				208	45	.	.	.	.	.	.
92111				204	45	.	.	.	.	.	.
92112	2	38	58	204	45	.	.	.	.	.	.
92113				204	45	.	.	.	.	.	.
92114				204	45	.	.	.	.	.	.
92115				204	45	.	.	.	.	.	.
92116	2	39	2	204	45	.	.	.	.	.	.
92117				204	45	.	.	.	.	.	.
92118				204	45	.	.	.	.	.	.
92119				204	45	.	.	.	.	.	.
92120	2	39	6	204	45	.	.	.	.	.	.
92121				204	45	.	.	.	.	.	.
92122				204	45	.	.	.	.	.	.
92123				204	45	.	.	.	.	.	.
92124	2	39	10	204	45	.	.	.	.	.	.
92125				204	45	.	.	.	.	.	.
92126				204	45	.	.	.	.	.	.
92127				204	45	.	.	.	.	.	.
92128	2	39	14	204	45	.	.	.	.	.	.
92129				204	45	.	.	.	.	.	.
92130				200	45	.	.	.	.	.	.
92131				204	45	.	.	.	.	.	.
92132	2	39	18	200	45	.	.	.	.	.	.
92133				200	45	.	.	.	.	.	.
92134				200	45	.	.	.	.	.	.
92135				200	45	.	.	.	.	.	.
92136	2	39	22	200	45	.	.	.	.	.	.
92137				200	45	.	.	.	.	.	.
92138				200	45	.	.	.	.	.	.
92139				200	45	.	.	.	.	.	.
92140	2	39	26	200	45	.	.	.	.	.	.
92141				200	45	.	.	.	.	.	.
92142				200	45	.	.	.	.	.	.
92143				196	45	.	.	.	.	.	.
92144	2	39	30	196	45	.	.	.	.	.	.

Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	HF L KEYING	HF R KEYING	VHF C KEYING	VHF L KEYING	VHF R KEYING	A/P WARN
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-WARN 1-.)
92145				196	45	.	.	.	.	.	.
92146				196	45	.	.	.	.	.	.
92147				196	45	.	.	.	.	.	.
92148	2	39	34	196	45	.	.	.	.	.	.
92149				196	45	.	.	.	.	.	.
92150				196	45	.	.	.	.	.	.
92151				196	45	.	.	.	.	.	.
92152	2	39	38	196	45	.	.	.	.	.	.
92153				196	45	.	.	.	.	.	.
92154				196	45	.	.	.	.	.	.
92155				192	45	.	.	.	.	.	.
92156	2	39	42	192	45	.	.	.	.	.	.
92157				196	45	.	.	.	.	.	.
92158				192	45	.	.	.	.	.	.
92159				192	45	.	.	.	.	.	.
92160	2	39	46	192	45	.	.	.	.	.	.
92161				192	45	.	.	.	.	.	.
92162				192	45	.	.	.	.	.	.
92163				192	45	.	.	.	.	.	.
92164	2	39	50	192	45	.	.	.	.	.	.
92165				192	45	.	.	.	.	.	.
92166				192	45	.	.	.	.	.	.
92167				192	45	.	.	.	.	.	.
92168	2	39	54	192	45	.	.	.	.	.	.
92169				192	45	.	.	.	.	.	.
92170				192	45	.	.	.	.	.	.
92171				192	45	.	.	.	.	.	.
92172	2	39	58	192	45	.	.	.	.	.	.
92173				192	45	.	.	.	.	.	.
92174				192	45	.	.	.	.	.	.
92175				192	45	.	.	.	.	.	.
92176	2	40	2	192	45	.	.	.	.	.	.
92177				192	45	.	.	.	.	.	.
92178				192	45	.	.	.	.	.	.
92179				192	45	.	.	.	.	.	.
92180	2	40	6	192	45	.	.	.	.	.	.
92181				188	45	.	.	.	.	.	.
92182				192	45	.	.	.	.	.	.
92183				192	45	.	.	.	.	.	.
92184	2	40	10	192	45	.	.	.	.	.	.
92185				188	45	.	.	.	.	.	.

Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	HF L KEYING	HF R KEYING	VHF C KEYING	VHF L KEYING	VHF R KEYING	A/P WARN
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-WARN 1-.)
92186				192	45	.	.	.	.	.	.
92187				192	45	.	.	.	.	.	.
92188	2	40	14	188	45	.	.	.	.	.	.
92189				192	45	.	.	.	.	.	.
92190				188	45	.	.	.	.	.	.
92191				188	45	.	.	.	.	.	.
92192	2	40	18	188	45	.	.	.	.	.	.
92193				188	45	.	.	.	.	.	.
92194				188	45	.	.	.	.	.	.
92195				188	45	.	.	.	.	.	.
92196	2	40	22	188	45	.	.	.	.	.	.
92197				188	45	.	.	.	.	.	.
92198				188	45	.	.	.	.	.	.
92199				188	45	.	.	.	.	.	.
92200	2	40	26	188	45	.	.	.	.	.	.
92201				188	45	.	.	.	.	.	.
92202				188	45	.	.	.	.	.	.
92203				188	45	.	.	.	.	.	.
92204	2	40	30	188	45	.	.	.	.	.	.
92205				188	45	.	.	.	.	.	.
92206				188	45	.	.	.	.	.	.
92207				188	45	.	.	.	.	.	.
92208	2	40	34	188	45	.	.	.	.	.	.
92209				188	45	.	.	.	.	.	.
92210				188	45	.	.	.	.	.	.
92211				188	45	.	.	.	.	.	.
92212	2	40	38	188	45	.	.	.	.	.	.
92213				188	45	.	.	.	.	.	.
92214				188	45	.	.	.	KEYED	.	.
92215				184	45	.	.	.	KEYED	.	.
92216	2	40	42	188	45	.	.	.	.	.	.
92217				188	45	.	.	.	.	.	.
92218				188	45	.	.	.	.	.	.
92219				188	45	.	.	.	.	.	.
92220	2	40	46	188	45	.	.	.	.	.	.
92221				188	45	.	.	.	.	.	.
92222				184	45	.	.	.	.	.	.
92223				188	45	.	.	.	.	.	.
92224	2	40	50	188	45	.	.	.	.	.	.
92225				184	45	.	.	.	.	.	.
92226				184	45	.	.	.	.	.	.

Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	HF L KEYING	HF R KEYING	VHF C KEYING	VHF L KEYING	VHF R KEYING	A/P WARN
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-WARN 1-.)
92227				184	45	.	.	.	.	.	.
92228	2	40	54	184	45	.	.	.	.	.	.
92229				184	45	.	.	.	.	.	.
92230				184	45	.	.	.	KEYED	.	.
92231				184	45	.	.	.	KEYED	.	.
92232	2	40	58	184	45	.	.	.	KEYED	.	.
92233				184	45	.	.	.	KEYED	.	.
92234				184	45	.	.	.	KEYED	.	.
92235				184	45	.	.	.	KEYED	.	.
92236	2	41	2	184	45	.	.	.	.	.	.
92237				184	45	.	.	.	.	.	.
92238				184	45	.	.	.	.	.	.
92239				184	45	.	.	.	.	.	.
92240	2	41	6	184	45	.	.	.	.	.	.
92241				184	45	.	.	.	.	.	.
92242				184	45	.	.	.	.	.	.
92243				184	45	.	.	.	.	.	.
92244	2	41	10	184	45	.	.	.	.	.	.
92245				184	45	.	.	.	.	.	.
92246				184	45	.	.	.	.	.	.
92247				184	45	.	.	.	.	.	.
92248	2	41	14	184	45	.	.	.	.	.	.
92249				184	45	.	.	.	.	.	.
92250				184	45	.	.	.	.	.	.
92251				184	45	.	.	.	.	.	.
92252	2	41	18	184	45	.	.	.	.	.	.
92253				184	45	.	.	.	.	.	.
92254				184	45	.	.	.	.	.	.
92255				184	45	.	.	.	.	.	.
92256	2	41	22	184	45	.	.	.	.	.	.
92257				184	45	.	.	.	.	.	.
92258				184	45	.	.	.	.	.	.
92259				180	45	.	.	.	.	.	.
92260	2	41	26	180	45	.	.	.	.	.	.
92261				180	45	.	.	.	.	.	.
92262				180	45	.	.	.	.	.	.
92263				180	45	.	.	.	.	.	.
92264	2	41	30	180	45	.	.	.	.	.	.
92265				180	45	.	.	.	.	.	.
92266				180	45	.	.	.	.	.	.
92267				180	45	.	.	.	.	.	.

Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	HF L KEYING	HF R KEYING	VHF C KEYING	VHF L KEYING	VHF R KEYING	A/P WARN
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-WARN 1-.)
92268	2	41	34	180	45	.	.	.	.	.	.
92269				180	45	.	.	.	.	.	.
92270				180	45	.	.	.	.	.	.
92271				180	45	.	.	.	KEYED	.	.
92272	2	41	38	180	45	.	.	.	KEYED	.	.
92273				180	45	.	.	.	KEYED	.	.
92274				180	45	.	.	.	KEYED	.	.
92275				180	45	.	.	.	.	.	.
92276	2	41	42	180	45	.	.	.	.	.	.
92277				180	45	.	.	.	.	.	.
92278				180	45	.	.	.	.	.	.
92279				180	45	.	.	.	KEYED	.	.
92280	2	41	46	180	45	.	.	.	KEYED	.	.
92281				180	45	.	.	.	.	.	.
92282				180	45	.	.	.	.	.	.
92283				180	45	.	.	.	.	.	.
92284	2	41	50	180	45	.	.	.	.	.	.
92285				180	45	.	.	.	.	.	.
92286				180	45	.	.	.	.	.	.
92287				180	45	.	.	.	.	.	.
92288	2	41	54	180	45	.	.	.	.	.	.
92289				180	45	.	.	.	.	.	.
92290				180	45	.	.	.	.	.	.
92291				184	45	.	.	.	.	.	.
92292	2	41	58	180	45	.	.	.	.	.	.
92293				184	45	.	.	.	.	.	.
92294				184	45	.	.	.	.	.	.
92295				184	45	.	.	.	.	.	.
92296	2	42	2	188	45	.	.	.	.	.	.
92297				188	45	.	.	.	.	.	.
92298				188	45	.	.	.	.	.	.
92299				188	45	.	.	.	.	.	.
92300	2	42	6	192	45	.	.	.	.	.	.
92301				192	45.5	.	.	.	.	.	.
92302				192	49.5	.	.	.	.	.	.
92303				196	56	.	.	.	.	.	.
92304	2	42	10	196	61	.	.	.	.	.	.
92305				196	65	.	.	.	.	.	.
92306				196	70	.	.	.	.	.	.
92307				200	75.5	.	.	.	.	.	.
92308	2	42	14	200	78.5	.	.	.	.	.	.



Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	HF L KEYING	HF R KEYING	VHF C KEYING	VHF L KEYING	VHF R KEYING	A/P WARN
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-WARN 1-.)
92309				200	83.5	.	.	.	.	.	.
92310				200	89	.	.	.	.	.	.
92311				200	93	.	.	.	.	.	.
92312	2	42	18	200	97.5	.	.	.	.	.	.
92313				204	101	.	.	.	.	.	.
92314				204	106.5	.	.	.	.	.	.
92315				204	109.5	.	.	.	.	.	.
92316	2	42	22	204	115.5	.	.	.	.	.	.
92317				204	119.5	.	.	.	.	.	.
92318				204	123.5	.	.	.	.	.	.
92319				208	127.5	.	.	.	.	.	.
92320	2	42	26	208	131.5	.	.	.	.	.	.
92321				208	135.5	.	.	.	.	.	.
92322				208	139	.	.	.	.	.	.
92323				204	142.5	.	.	.	.	.	.
92324	2	42	30	204	146	.	.	.	.	.	.
92325				196	150	.	.	.	.	.	.
92326				192	152	.	.	.	.	.	.
92327				192	155.5	.	.	.	.	.	.
92328	2	42	34	196	159	.	.	.	.	.	.
92329				208	162	.	.	.	.	.	.
92330				220	165.5	.	.	.	.	.	.
92331				240	167.5	.	.	.	.	.	.
92332	2	42	38	268	169.5	.	.	.	.	.	.
92333				300	171.5	.	.	.	.	.	.
92334				328	172	.	.	.	.	.	.
92335				364	173	.	.	.	.	.	.
92336	2	42	42	400	174	.	.	.	.	.	.
92337				440	174.5	.	.	.	.	.	.
92338				480	176	.	.	.	.	.	.
92339				512	176.5	.	.	.	.	.	.
92340	2	42	46	548	177	.	.	.	.	.	.
92341				584	178	.	.	.	.	.	.
92342				616	178.5	.	.	.	.	.	.
92343				652	179	.	.	.	.	.	.
92344	2	42	50	688	178.5	.	.	.	.	.	.
92345				720	179.5	.	.	.	.	.	.
92346				756	179.5	.	.	.	.	.	.
92347				792	180	.	.	.	.	.	.
92348	2	42	54	832	180	.	.	.	.	.	.
92349				868	181	.	.	.	.	.	.

Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	HF L KEYING	HF R KEYING	VHF C KEYING	VHF L KEYING	VHF R KEYING	A/P WARN
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-WARN 1-.)
92350				904	180.5	.	.	.	.	.	.
92351				940	181.5	.	.	.	.	.	.
92352	2	42	58	976	181	.	.	.	.	.	.
92353				1016	181.5	.	.	.	.	.	.
92354				1052	181.5	.	.	.	.	.	.
92355				1096	183	.	.	.	.	.	.
92356	2	43	2	1136	183	.	.	.	.	.	.
92357				1180	184	.	.	.	.	.	.
92358				1220	184	.	.	.	.	.	.
92359				1268	184	.	.	.	.	.	.
92360	2	43	6	1312	184	.	.	.	.	.	.
92361				1352	183	.	.	.	.	.	.
92362				1396	184	.	.	.	.	.	.
92363				1440	184	.	.	.	.	.	.
92364	2	43	10	1484	183.5	.	.	.	.	.	.
92365				1528	183	.	.	.	.	.	.
92366				1576	183.5	.	.	.	.	.	.
92367				1624	183	.	.	.	KEYED	.	.
92368	2	43	14	1668	182.5	.	.	.	KEYED	.	.
92369				1708	183	.	.	.	KEYED	.	.
92370				1748	183.5	.	.	.	KEYED	.	.
92371				1784	184.5	.	.	.	KEYED	.	.
92372	2	43	18	1816	185.5	.	.	.	KEYED	.	.
92373				1844	186.5	.	.	.	.	.	.
92374				1868	187.5	.	.	.	.	.	.
92375				1892	188.5	.	.	.	.	.	.
92376	2	43	22	1912	190	.	.	.	.	.	.
92377				1932	191.5	.	.	.	.	.	.
92378				1948	193	.	.	.	.	.	.
92379				1964	194.5	.	.	.	.	.	.
92380	2	43	26	1980	196.5	.	.	.	.	.	.
92381				2000	198.5	.	.	.	.	.	.
92382				2020	200.5	.	.	.	.	.	.
92383				2040	202	.	.	.	.	.	.
92384	2	43	30	2064	203.5	.	.	.	.	.	.
92385				2084	205	.	.	.	.	.	.
92386				2112	206	.	.	.	.	.	.
92387				2136	207.5	.	.	.	.	.	.
92388	2	43	34	2168	208.5	.	.	.	.	.	.
92389				2196	209	.	.	.	.	.	.
92390				2224	210.5	.	.	.	.	.	.

Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	HF L KEYING	HF R KEYING	VHF C KEYING	VHF L KEYING	VHF R KEYING	A/P WARN
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-WARN 1-.)
92391				2252	212	.	.	.	.	.	.
92392	2	43	38	2284	213.5	.	.	.	.	.	.
92393				2320	214.5	.	.	.	.	.	.
92394				2352	215.5	.	.	.	.	.	.
92395				2392	215.5	.	.	.	.	.	.
92396	2	43	42	2432	216	.	.	.	.	.	.
92397				2472	216.5	.	.	.	.	.	.
92398				2520	216.5	.	.	.	.	.	.
92399				2572	217	.	.	.	.	.	.
92400	2	43	46	2624	216.5	.	.	.	.	.	.
92401				2676	216.5	.	.	.	.	.	.
92402				2728	216	.	.	.	.	.	.
92403				2784	216.5	.	.	.	.	.	.
92404	2	43	50	2840	217	.	.	.	.	.	.
92405				2892	217	.	.	.	.	.	.
92406				2948	216.5	.	.	.	.	.	.
92407				3004	216.5	.	.	.	.	.	.
92408	2	43	54	3064	216	.	.	.	.	.	.
92409				3124	216	.	.	.	.	.	.
92410				3188	214.5	.	.	.	.	.	.
92411				3252	214	.	.	.	.	.	.
92412	2	43	58	3320	213.5	.	.	.	.	.	.
92413				3392	212	.	.	.	.	.	.
92414				3468	209.5	.	.	.	.	.	.
92415				3544	209.5	.	.	.	.	.	.
92416	2	44	2	3624	207	.	.	.	.	.	.
92417				3712	206	.	.	.	.	.	WARN
92418				3796	204.5	.	.	.	.	.	.
92419				3880	203	.	.	.	.	.	.
92420	2	44	6	3964	201	.	.	.	.	.	.
92421				4056	199	.	.	.	.	.	.
92422				4136	196.5	.	.	.	.	.	.
92423				4220	194.5	.	.	.	.	.	.
92424	2	44	10	4308	195	.	.	.	.	.	.
92425				4388	192	.	.	.	.	.	.
92426				4460	190	.	.	.	.	.	.
92427				4532	190	.	.	.	.	.	.
92428	2	44	14	4600	188.5	.	.	.	.	.	.
92429				4660	188	.	.	.	.	.	.
92430				4720	187.5	.	.	.	.	.	.
92431				4772	187	.	.	.	.	.	.

Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	HF L KEYING	HF R KEYING	VHF C KEYING	VHF L KEYING	VHF R KEYING	A/P WARN
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-WARN 1-.)
92432	2	44	18	4824	186.5	.	.	.	.	.	.
92433				4876	186	.	.	.	.	.	.
92434				4920	185.5	.	.	.	.	.	.
92435				4968	185.5	.	.	.	.	.	.
92436	2	44	22	5008	185	.	.	.	.	.	.
92437				5044	184.5	.	.	.	.	.	.
92438				5076	185.5	.	.	.	.	.	.
92439				5112	186	.	.	.	.	.	.
92440	2	44	26	5144	186.5	.	.	.	.	.	.
92441				5172	186	.	.	.	.	.	.
92442				5204	186.5	.	.	.	.	.	.
92443				5232	187	.	.	.	.	.	.
92444	2	44	30	5260	187.5	.	.	.	.	.	.
92445				5288	188.5	.	.	.	.	.	.
92446				5320	189	.	.	.	.	.	.
92447				5344	189.5	.	.	.	.	.	.
92448	2	44	34	5372	191	.	.	.	.	.	.
92449				5396	192	.	.	.	.	.	.
92450				5420	193.5	.	.	.	.	.	.
92451				5436	195	.	.	.	.	.	.
92452	2	44	38	5452	196.5	.	.	.	.	.	.
92453				5460	198.5	.	.	.	.	.	.
92454				5464	200.5	.	.	.	.	.	.
92455				5468	202.5	.	.	.	.	.	.
92456	2	44	42	5460	205.5	.	.	.	.	.	.
92457				5452	207.5	.	.	.	.	.	.
92458				5432	209.5	.	.	.	.	.	.
92459				5408	212	.	.	.	.	.	.
92460	2	44	46	5380	215	.	.	.	.	.	.
92461				5332	218.5	.	.	.	.	.	.
92462				5276	222	.	.	.	.	.	.
92463				5204	225.5	.	.	.	.	.	.
92464	2	44	50	5096	230.5	.	.	.	.	.	.
92465				4972	236.5	.	.	.	.	.	.
92466				4816	244.5	.	.	.	.	.	.
92467				4628	254	.	.	.	.	.	.
92468	2	44	54	4388	264.5	.	.	.	.	.	.
92469				4124	275.5	.	.	.	.	.	.
92470				3820	289.5	.	.	.	.	.	.
92471				3508	306.5	.	.	.	.	.	.
92472	2	44	58	3068	317.5	.	.	.	.	.	.

Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	HF L KEYING	HF R KEYING	VHF C KEYING	VHF L KEYING	VHF R KEYING	A/P WARN
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-KEYED 1-.)	(0-WARN 1-.)
92473				2640	334	.	.	.	.	.	.
92474				2216	352	.	.	.	.	.	.
92475				1748	368.5	.	.	.	.	.	.
92476	2	45	2	1320	382.5	.	.	.	.	.	.
92477				904	395	.	.	.	.	.	.
92478				524	410	.	.	.	.	.	.
92479				180	416	.	.	.	.	.	.
92480											



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )
91878				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	21.25	1.7	-0.24244	-0.31481	-3.64084 -3.59122	34.9172 34.9172
91879				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21.375	1.7	-0.24244	-0.31481	-3.64084 -3.64084	34.9172 34.9172
91880	2	35	6	216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	21.375	1.7	-0.24244	-0.31481	-3.64084 -3.59122	34.9172 34.9172
91881				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21.25	1.7	-0.24244	-0.31481	-3.59122 -3.59122	34.9172 34.9172
91882				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	21.25	1.7	-0.24244	-0.31481	-3.59122 -3.59122	34.9172 34.9172
91883				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21.25	1.7	-0.24244	-0.31481	-3.59122 -3.64084	34.9172 34.9172
91884	2	35	10	216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	21.25	1.7	-0.24244	-0.31481	-3.59122 -3.59122	34.9172 34.9172
91885				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21.25	1.7	-0.32326	-0.31481	-3.64084 -3.64084	34.9172 34.9172
91886				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	21.125	1.7	-0.24244	-0.31481	-3.64084 -3.64084	34.9172 34.9172
91887				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.23047	15.875	21	1.7	-0.24244	-0.31481	-3.64084 -3.64084	34.9172 34.9172
91888	2	35	14	216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	21	1.7	-0.24244	-0.31481	-3.64084 -3.64084	34.9172 34.9172
91889				216	45	-3.82096	-4.69666	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21	1.7	-0.24244	-0.27985	-3.64084 -3.64084	34.9172 34.9172
91890				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	21	1.7	-0.24244	-0.27985	-3.64084 -3.64084	34.9172 34.9172
91891				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21.125	1.7	-0.24244	-0.31481	-3.64084 -3.64084	34.9172 34.9172
91892	2	35	18	216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	21.125	1.7	-0.24244	-0.31481	-3.64084 -3.59122	34.9172 34.9172
91893				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21.125	1.7	-0.32326	-0.31481	-3.59122 -3.59122	34.9172 34.9172





Time (seconds)	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )	
91909				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21	1.7	-0.24244	-0.31481	-3.59122	34.9172	
91910				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	21	1.7	-0.24244	-0.31481	-3.59122	34.9172	
91911				216	45	-3.82096	-4.69666	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21	1.7	-0.32326	-0.31481	-3.59122	34.9172	
91912	2	35	38	216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.23047 1.05469	0	21	1.7	-0.32326	-0.31481	-3.59122	34.9172	
91913				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21	1.7	-0.32326	-0.31481	-3.59122	34.9172	
91914				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	20.875	1.7	-0.24244	-0.31481	-3.59122	34.9172	
91915				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	20.875	1.7	-0.24244	-0.31481	-3.59122	34.9172	
91916	2	35	42	216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	20.875	1.7	-0.24244	-0.31481	-3.59122	34.9172	
91917				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21	1.7	-0.24244	-0.31481	-3.59122	34.9172	
91918				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	21	1.7	-0.24244	-0.31481	-3.59122	34.9172	
91919				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21	1.7	-0.24244	-0.31481	-3.59122	34.9172	
91920	2	35	46	216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	21	1.7	-0.24244	-0.31481	-3.59122	34.9172	
91921				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21	1.7	-0.24244	-0.31481	-3.59122	34.9172	
91922				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	21	1.7	-0.24244	-0.31481	-3.59122	34.9172	
91923				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21	1.7	-0.24244	-0.31481	-3.59122	34.9172	
91924	2	35	50	216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	21.125	1.7	-0.24244	-0.31481	-3.59122	34.9172	



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )
91940	2	36	6	216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	21.125	1.7	-0.24244	-0.31481	-3.59122	34.9172
91941				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21.125	1.7	-0.24244	-0.31481	-3.59122	34.9172
91942				216	45	-3.88063	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	21.125	1.7	-0.24244	-0.31481	-3.59122	34.9172
91943				216	45	-3.88063	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21.125	1.7	-0.24244	-0.31481	-3.59122	34.9172
91944	2	36	10	216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	21.125	1.7	-0.24244	-0.31481	-3.59122	34.9172
91945				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	21.125	1.7	-0.24244	-0.27985	-3.59122	34.9172
91946				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.23047	0	20.875	1.7	-0.24244	-0.27985	-3.59122	34.9172
91947				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	20.875	1.7	-0.24244	-0.27985	-3.59122	34.9172
91948	2	36	14	216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.23047	0	20.75	1.7	-0.24244	-0.24489	-3.59122	34.9172
91949				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	20.75	1.7	-0.16164	-0.24489	-3.64084	34.9172
91950				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	20.875	1.7	-0.24244	-0.03499	-3.59122	34.9172
91951				216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	20.875	1.7	-0.24244	-0.27985	-3.59122	34.7073
91952	2	36	18	216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	20.875	1.7	-0.24244	-0.27985	-3.59122	34.7073
91953				216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	20.875	1.7	-0.24244	-0.27985	-3.59122	34.7073
91954				216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	0	20.75	1.7	-0.24244	-0.27985	-3.59122	34.7073
91955				216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769 9.54769	0.17578 0.17578	0	309.375	1.05469	15.875	20.875	1.7	-0.24244	-0.27985	-3.59122	34.7073

Time	GMT	GMT	GMT	ALTITUDE	COMPUTED	ELEVATOR	ELEVATOR	AILERON	AILERON	SPD	PITCH	ROLL	MAGNETI	AOA	N1 L	N1 R	PITCH	RUDDER	RUDDER	CONTROL	CONTROL	
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	( )	( )	( )	( )	( )	(DEG)	(DEG)	(DEG)	(DEG)	(%RPM)	(%RPM)	( )	( )	( )	( )	( )	
											0.17578											
											0.17578											
91956	2	36	22	216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.375	1.05469	0	20.875	1.7	-0.24244	-0.27985	-3.59122	34.7073	
										9.54769	0.17578	0		1.05469							-3.59122	34.7073
											0.17578											
											0.17578											
91957				216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.375	1.23047	15.875	20.875	1.7	-0.24244	-0.27985	-3.59122	34.7073	
										9.54769	0.17578	0		1.05469							-3.59122	34.7073
											0.17578											
											0.17578											
91958				216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.375	1.05469	0	20.875	1.7	-0.24244	-0.27985	-3.59122	34.7073	
										9.54769	0.17578	0		1.05469							-3.59122	34.7073
											0.17578											
											0.17578											
91959				216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.375	1.05469	15.875	20.875	1.7	-0.24244	-0.27985	-3.59122	34.7073	
										9.54769	0.17578	0		1.05469							-3.59122	34.7073
											0.17578											
											0.17578											
91960	2	36	26	216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.375	1.05469	0	21	1.7	-0.24244	-0.27985	-3.59122	34.7073	
										9.54769	0.17578	0		1.05469							-3.59122	34.7073
											0.17578											
											0.17578											
91961				216	45	-3.88063	-4.63334	0.969642	0.969645	9.54769	0.17578	0	309.375	1.05469	15.875	20.875	1.7	-0.24244	-0.27985	-3.59122	34.7073	
										9.54769	0.17578	0		1.05469							-3.59122	34.7073
											0.17578											
											0.17578											
91962				216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.375	1.23047	0	20.875	1.7	-0.24244	-0.27985	-3.59122	34.7073	
										9.54769	0.17578	0		1.23047							-3.59122	34.7073
											0.17578											
											0.17578											
91963				216	45	-3.82096	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.375	1.05469	15.875	20.875	1.7	-0.24244	-0.27985	-3.59122	34.7073	
										9.54769	0.17578	0		1.05469							-3.59122	34.7073
											0.17578											
											0.17578											
91964	2	36	30	216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.375	1.05469	0	21	1.7	-0.24244	-0.27985	-3.59122	34.7073	
										9.54769	0.17578	0		1.05469							-3.59122	34.7073
											0.17578											
											0.17578											
91965				216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.375	1.05469	15.875	20.875	1.7	-0.32326	-0.27985	-3.59122	34.7073	
										9.54769	0.17578	0		1.05469							-3.59122	34.7073
											0.17578											
											0.17578											
91966				216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.375	1.05469	0	20.875	1.7	-0.24244	-0.27985	-3.59122	34.7073	
										9.54769	0.17578	0		1.05469							-3.59122	34.7073
											0.17578											
											0.17578											
91967				216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.375	1.05469	15.875	20.875	1.7	-0.24244	-0.27985	-3.59122	34.7073	
										9.54769	0.17578	0		1.05469							-3.59122	34.7073
											0.17578											
											0.17578											
91968	2	36	34	216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.375	1.05469	0	20.875	1.7	-0.24244	-0.27985	-3.59122	34.7073	
										9.54769	0.17578	0		1.05469							-3.59122	34.7073
											0.17578											
											0.17578											
91969				216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.375	1.05469	15.875	20.875	1.7	-0.24244	-0.27985	-3.59122	34.7073	
										9.54769	0.17578	0		1.05469							-3.59122	34.7073
											0.17578											
											0.17578											
91970				216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.375	1.05469	0	20.875	1.7	-0.24244	-0.27985	-3.59122	34.7073	
										9.54769	0.17578	0		1.05469							-3.59122	34.7073
											0.17578											
											0.17578											







Time (seconds)	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )	
											0.17578											
											0.17578											
92018				216	45	-3.82096	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.375	1.05469	0	40	1.7	-0.24244	-0.27985	-3.64084	34.9172	
										9.54769	0.17578	0		1.05469							-3.64084	34.9172
											0.17578											
											0.17578											
92019				216	45	-3.76128	-4.75997	0.969642	0.969645	9.54769	0.17578	0	309.375	1.05469	15.875	39.875	1.7	-0.40409	-0.24489	-3.64084	34.9172	
										9.54769	0.17578	0		1.23047							-3.64084	34.9172
											0.17578											
											0.17578											
92020	2	37	26	216	45	-3.82096	-4.82328	0.969642	0.969645	9.54769	0.17578	0	309.375	1.05469	0	38.375	1.7	-0.48489	-0.24489	-3.64084	34.9172	
										9.54769	0.17578	0		1.23047							-3.64084	34.9172
											0.17578											
											0.17578											
92021				216	45	-3.76128	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.727	1.05469	15.875	34.875	1.7	-0.24244	-0.24489	-3.64084	34.9172	
										9.54769	0.17578	0		1.05469							-3.64084	34.9172
											0.17578											
											0.17578											
92022				216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0.17578	0	309.727	1.05469	0	31.875	1.7	-0.56571	-0.24489	-3.64084	34.9172	
										9.54769	0.17578	0		1.05469							-3.64084	34.9172
											0.17578											
											0.17578											
92023				216	45	-3.94032	-4.69666	0.969642	0.969645	9.54769	0.17578	0	310.078	1.05469	15.875	29.875	1.7	-1.5349	-0.24489	-3.64084	34.9172	
										9.54769	0.17578	0		1.23047							-3.64084	34.9172
											0.17578											
											0.17578											
92024	2	37	30	216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0.17578	0	311.133	1.23047	0	30.375	1.7	-1.93828	-0.24489	-3.64084	34.9172	
										9.54769	0.17578	0		1.23047							-3.64084	34.9172
											0.17578											
											0.17578											
92025				216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0.17578	0	312.188	1.23047	15.875	30.625	1.7	-2.34132	-0.24489	-3.64084	34.9172	
										9.54769	0.17578	0		1.23047							-3.64084	34.9172
											0											
											0											
92026				216	45	-3.88063	-4.63334	0.969642	0.969645	9.54769	0	0	314.648	1.23047	0	30.5	1.7	-3.3066	1.92954	-3.64084	34.9172	
										9.54769	0	0		1.05469							-3.64084	34.9172
											0											
											0											
92027				216	45	-3.82096	-4.63334	0.969642	0.969645	9.54769	0	0	317.109	1.05469	15.875	28.75	1.7	19.7637	12.9665	-3.64084	34.9172	
										9.54769	0	0		1.23047							-3.64084	34.9172
											0											
											0											
92028	2	37	34	216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	0	0	321.328	1.05469	0	26.75	1.7	25.8946	0.767971	-3.64084	34.9172	
										9.54769	0.17578	0		1.05469							-3.64084	34.9172
											0											
											0											
92029				216	45	-3.88063	-4.75997	0.969642	0.969645	9.54769	-0.17578	-0.35156	325.195	1.23047	15.875	25.25	1.7	-6.17691	-4.01553	-3.64084	34.9172	
										9.54769	-0.17578	-0.35156		1.05469							-3.64084	34.9172
											-0.17578											
											-0.17578											
92030				216	45	-3.94032	-4.63334	0.969642	0.969645	9.54769	-0.17578	-0.35156	331.523	1.23047	0	25	1.7	-26.5765	-12.4389	-3.64084	34.9172	
										9.54769	-0.17578	-0.35156		1.23047							-3.64084	34.9172
											-0.35156											
											-0.35156											
92031				216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	-0.35156	0	337.5	1.23047	15.875	23.625	1.7	-16.3136	-0.24489	-3.64084	34.9172	
										9.54769	-0.35156	0		1.05469							-3.64084	34.9172
											-0.35156											
											-0.35156											
92032	2	37	38	216	45	-3.88063	-4.69666	0.969642	0.969645	9.54769	-0.35156	0	345.234	1.05469	0	22.375	1.7	-2.90476	-0.17494	-3.64084	34.9172	
										9.54769	-0.35156	-0.35156		1.05469							-3.64084	56.5421
											-0.35156											
											-0.17578											



Time (seconds)	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )
92033				216	45	-3.52258	-4.5067	-17.9471	18.4694	3.21902 9.54769	-0.17578 -0.17578	-0.70312 -0.70312	351.211	1.05469	15.875	22.25	1.7	-2.18013	-0.20992	-3.78912 -3.64084	64.3333 1.87282
92034				216	45	-3.88063	-5.70867	10.9874	-21.2343	5.34222 10.5907	-0.17578 -0.17578	-0.35156 -0.35156	358.945	1.05469	0	22.75	1.7	-1.21196	-0.24489	-3.03913 -3.29145	-1.1254 -0.37574
92035				216	45	-4.06334	-0.78898	4.30866	1.19328	10.5907 10.5907	0 0	0 -0.35156	4.92188	1.05469	15.875	22.625	1.8	-0.08082	-0.24489	-3.69037 -10.9023	19.1577 17.2165
92036	2	37	42	216	45	18.5069	10.7375	0.969642	0.969645	10.5907 10.5907	0 0	0 0	12.3047	1.05469	0	22.375	1.8	0.969673	-0.24489	-14.807 1.11645	17.8705 17.5444
92037				216	45	-21.3483	-22.6033	0.969642	0.969645	10.5907 10.5907	0 0	0 0.351562	17.9297	1.05469	15.875	22.5	1.7	1.37345	-0.20992	11.0127 -1.27508	17.5444 17.2165
92038				216	45	-1.91434	-4.94987	0.969642	0.969645	10.5907 10.5907	0 0	0.351562 0.703124	23.5547	1.05469	0	22.5	1.7	2.6634	-0.24489	-3.59122 -3.59122	17.2165 17.5444
92039				216	45	-3.82096	-4.69666	0.969642	0.969645	10.5907 10.5907	0 0	0.703124 1.05469	28.4766	1.05469	15.875	22.5	1.7	2.26073	-0.20992	-3.59122 -3.59122	17.5444 17.5444
92040	2	37	46	216	45	-3.88063	-4.69666	0.969642	0.969645	10.5907 10.5907	-0.17578 -0.17578	1.05469 1.05469	34.1016	1.05469	0	22.625	1.7	3.7078	-0.10497	-3.59122 -3.59122	17.5444 17.5444
92041				216	45	-3.82096	-4.69666	0.969642	0.969645	10.5907 10.5907	-0.17578 -0.35156	1.05469 1.05469	38.3203	1.05469	15.875	22.625	1.7	2.50239	-0.10497	-3.64084 -3.59122	17.5444 17.5444
92042				216	45	-3.88063	-4.69666	0.969642	0.969645	10.5907 10.5907	-0.35156 -0.35156	1.05469 0.703124	43.5938	1.05469	0	22.5	1.7	-1.45419	-0.06998	-3.59122 -3.59122	17.5444 17.5444
92043				216	45	-3.88063	-4.69666	0.969642	0.969645	10.5907 10.5907	-0.52734 -0.52734	0.703124 0.703124	50.625	1.05469	15.875	22.5	1.7	-3.38689	-0.06998	-3.59122 -3.59122	17.5444 17.5444
92044	2	37	50	216	45	-4	-4.69666	0.969642	0.969645	10.5907 10.5907	-0.52734 -0.52734	0.703124 0.351562	56.9531	1.23047	0	22.75	1.7	-3.14593	-0.06998	-3.59122 -3.59122	17.5444 17.5444
92045				216	45	-3.94032	-4.63334	0.969642	0.969645	10.5907 10.5907	-0.52734 -0.52734	0.351562 0	65.7422	1.23047	15.875	22.75	1.7	-3.3066	-0.06998	-3.59122 -3.64084	17.5444 17.5444
92046				216	45	-3.82096	-4.69666	0.969642	0.969645	10.5907 10.5907	-0.35156 -0.35156	0 -0.35156	73.125	1.23047	0	22.75	1.7	-2.42185	-0.06998	-3.64084 -3.59122	17.5444 17.5444
92047				216	45	-3.88063	-4.75997	0.969642	0.969645	10.5907 10.5907	-0.52734 -0.52734	-0.35156 -0.35156	82.9688	1.23047	15.875	22.75	1.7	-0.80809	-0.10497	-3.59122 -3.59122	17.5444 17.5444
92048	2	37	54	216	45	-3.94032	-4.69666	0.969642	0.969645	10.5907 10.5907	-0.52734 -0.52734	-0.35156 -0.35156	90	1.23047	0	22.625	1.7	0	-0.13996	-3.64084 -3.59122	17.5444 17.5444



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L (°)	ELEVATOR POSN R (°)	AILERON POSN L (°)	AILERON POSN R (°)	SPD BRAKE HANDLE (°)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION (°)	RUDDER POSN (°)	RUDDER PEDAL POSN (°)	CONTROL COLUMN POSN (°)	CONTROL WHEEL POSN (°)	
92064	2	38	10	212	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0	137.109	1.23047	0	22.25	1.7	-0.32326	-0.31481	-3.59122	17.5444	
										10.5907	-0.52734	0		1.23047						-3.64084	17.8705	
											-0.52734											
											-0.70312											
92065				212	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.8789	0	137.109	1.23047	15.875	22.25	1.7	1.37345	-0.31481	-3.59122	17.8705	
										10.5907	-0.8789	0		1.23047						-3.59122	17.8705	
											-0.8789											
92066				212	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.8789	0	136.406	1.23047	0	22.25	1.7	2.3413	-0.31481	-3.64084	17.8705	
										10.5907	-0.8789	0		1.23047						-3.59122	17.5444	
											-0.8789											
											-1.05469											
92067				212	45	-3.82096	-4.63334	0.969642	0.969645	10.5907	-1.05469	-0.35156	134.297	1.23047	15.875	22.25	1.8	1.6156	-0.41961	-3.59122	17.5444	
										10.5907	-0.8789	0		1.23047						-3.59122	17.5444	
											-0.8789											
											-1.05469											
92068	2	38	14	212	45	-3.76128	-4.69666	0.969642	0.969645	10.5907	-1.05469	-0.35156	132.891	1.23047	0	22.25	1.7	0.565711	0.66366	-3.59122	17.5444	
										10.5907	-0.8789	-0.35156		1.23047						-3.59122	17.5444	
											-0.8789											
											-0.8789											
92069				212	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.8789	-0.35156	131.133	1.23047	15.875	22.25	1.7	-0.72731	-0.38469	-3.59122	17.5444	
										10.5907	-1.05469	-0.35156		1.23047						-3.64084	17.5444	
											-1.05469											
											-1.05469											
92070				212	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-1.05469	-0.35156	129.727	1.23047	0	22.25	1.7	-2.09953	-0.34976	-3.64084	17.5444	
										10.5907	-1.05469	-0.35156		1.23047						-3.59122	18.195	
											-1.05469											
											-1.05469											
92071				212	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-1.05469	0	129.375	1.23047	15.875	22.25	1.7	-2.5829	-0.34976	-3.59122	18.195	
										10.5907	-1.05469	-0.35156		1.23047						-3.59122	18.195	
											-1.05469											
											-1.05469											
92072	2	38	18	212	45	-3.82096	-4.69666	0.969642	1.04419	10.5907	-1.23047	-0.35156	129.023	1.23047	0	22.25	1.7	-2.34132	-0.31481	-3.59122	18.195	
										10.5907	-1.23047	-0.35156		1.23047						-3.59122	18.195	
											-1.05469											
											-1.23047											
92073				212	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-1.23047	-0.35156	128.32	1.23047	15.875	22.375	1.7	-0.32326	-0.27985	-3.59122	18.195	
										10.5907	-1.05469	-0.35156		1.23047						-3.64084	18.195	
											-1.05469											
											-1.05469											
92074				212	45	-3.82096	-4.69666	0.969642	1.04419	10.5907	-1.05469	-0.35156	127.266	1.23047	0	22.25	1.7	-0.48489	-0.24489	-3.64084	18.195	
										10.5907	-1.05469	-0.35156		1.23047						-3.64084	18.195	
											-1.05469											
											-1.05469											
92075				212	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-1.05469	-0.35156	126.211	1.23047	15.875	22.375	1.7	0.484903	-0.24489	-3.59122	18.195	
										10.5907	-1.05469	0		1.23047						-3.64084	18.195	
											-1.23047											
											-1.05469											
92076	2	38	22	212	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-1.05469	-0.35156	124.102	1.23047	0	22.375	1.7	1.77697	-0.24489	-3.64084	18.195	
										10.5907	-1.05469	-0.35156		1.23047						-3.64084	18.195	
											-1.05469											
											-1.05469											
92077				208	45	-3.94032	-4.57003	0.969642	0.969645	10.5907	-1.05469	-0.35156	121.992	1.23047	15.875	22.375	1.7	3.5474	-0.20992	-3.59122	18.195	
										10.5907	-1.05469	0		1.23047						-3.59122	18.195	
											-0.8789											
											-0.8789											
92078				208	45	-3.7016	-4.69666	0.969642	0.969645	10.5907	-1.05469	0	117.422	1.23047	0	22.375	1.7	3.5474	-0.24489	-3.59122	18.195	
										10.5907	-1.05469	0		1.23047						-3.59122	18.195	
											-0.8789											
											-1.05469											
92079				208	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-1.05469	0	111.797	1.23047	15.875	22.25	1.7	3.5474	-0.24489	-3.59122	18.195	
										10.5907	-1.05469	0.351562		1.23047						-3.59122	17.5444	





Time	GMT	GMT	GMT	ALTITUDE	COMPUTED	ELEVATOR	ELEVATOR	AILERON	AILERON	SPD	PITCH	ROLL	MAGNETI	AOA	N1 L	N1 R	PITCH	RUDDER	RUDDER	CONTROL	CONTROL	
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	( )	( )	( )	( )	( )	( )	( )	( )	( )	(%RPM)	(%RPM)	( )	( )	( )	( )	( )	( )
											-0.35156											
											-0.52734											
92111				204	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	39.0234	1.23047	15.875	22.375	1.7	0.24246	-0.24489	-3.59122	17.5444	
										10.5907	-0.35156	0.703124		1.23047							-3.59122	17.5444
											-0.52734											
											-0.52734											
92112	2	38	58	204	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	38.6719	1.23047	0	22.375	1.7	0	-0.24489	-3.59122	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.59122	17.5444
											-0.52734											
											-0.35156											
92113				204	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	38.6719	1.23047	15.875	22.25	1.7	-0.16164	-0.24489	-3.59122	17.5444	
										10.5907	-0.35156	0.703124		1.23047							-3.59122	17.5444
											-0.35156											
											-0.35156											
92114				204	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	38.3203	1.23047	0	22.125	1.7	-0.24244	-0.24489	-3.59122	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.59122	17.5444
											-0.52734											
											-0.52734											
92115				204	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	37.9688	1.23047	15.875	22.25	1.7	-0.24244	-0.24489	-3.59122	17.5444	
										10.5907	-0.52734	1.05469		1.23047							-3.59122	17.5444
											-0.52734											
											-0.52734											
92116	2	39	2	204	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	37.9688	1.23047	0	22.25	1.7	-0.80809	-0.24489	-3.59122	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.59122	17.5444
											-0.52734											
											-0.52734											
92117				204	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.52734	1.05469	38.3203	1.23047	15.875	22.25	1.7	-1.13121	-0.24489	-3.59122	17.5444	
										10.5907	-0.52734	1.05469		1.23047							-3.59122	17.5444
											-0.52734											
											-0.52734											
92118				204	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.52734	1.05469	38.6719	1.23047	0	22.25	1.7	0	-0.27985	-3.59122	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.59122	17.5444
											-0.52734											
											-0.52734											
92119				204	45	-3.82096	-4.63334	0.969642	0.969645	10.5907	-0.52734	0.703124	38.6719	1.23047	15.875	22.25	1.7	-0.08082	-0.24489	-3.59122	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.59122	17.5444
											-0.52734											
											-0.52734											
92120	2	39	6	204	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	38.6719	1.23047	0	22.375	1.7	-0.32326	-0.24489	-3.59122	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92121				204	45	-3.94032	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	39.0234	1.23047	15.875	22.25	1.8	-0.96967	-0.27985	-3.59122	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.59122	17.5444
											-0.52734											
											-0.52734											
92122				204	45	-3.94032	-4.63334	0.969642	0.969645	10.5907	-0.52734	1.05469	39.375	1.23047	0	22.25	1.7	0	-0.24489	-3.59122	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.59122	17.5444
											-0.52734											
											-0.52734											
92123				204	45	-3.82096	-4.63334	0.969642	0.969645	10.5907	-0.52734	0.703124	39.375	1.23047	15.875	22.25	1.7	0.484903	-0.24489	-3.59122	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.59122	17.5444
											-0.52734											
											-0.52734											
92124	2	39	10	204	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.52734	0.703124	39.375	1.23047	0	22.25	1.7	-0.24244	-0.24489	-3.59122	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.59122	17.5444
											-0.52734											
											-0.52734											
92125				204	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	1.05469	39.7266	1.23047	15.875	22.25	1.7	0.323277	-0.24489	-3.59122	17.5444	
										10.5907	-0.52734	1.05469		1.23047							-3.59122	17.5444
											-0.52734											
											-0.52734											



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L (°)	ELEVATOR POSN R (°)	AILERON POSN L (°)	AILERON POSN R (°)	SPD BRAKE HANDLE (°)	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION (°)	RUDDER POSN (°)	RUDDER PEDAL POSN (°)	CONTROL COLUMN POSN (°)	CONTROL WHEEL POSN (°)					
92142				200	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.70312	1.05469	37.9688	1.23047	0	21	1.7	0	-0.24489	-3.59122	17.5444					
										10.5907	-0.70312	0.703124		1.23047							-3.59122	17.5444				
											-0.70312															
92143				196	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.70312	1.05469	37.9688	1.23047	15.875	21	1.7	-0.24244	-0.24489	-3.59122	17.5444					
										10.5907	-0.70312	1.05469		1.23047								-3.59122	17.5444			
											-0.70312															
92144	2	39	30	196	45	-3.94032	-4.69666	0.969642	0.969645	10.5907	-0.70312	0.703124	37.9688	1.23047	0	21	1.7	-0.32326	-0.24489	-3.59122	17.5444					
										10.5907	-0.70312	0.703124		1.23047									-3.59122	17.5444		
											-0.70312															
92145				196	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.70312	0.703124	37.9688	1.23047	15.875	21	1.7	-0.56571	-0.24489	-3.59122	17.5444					
										10.5907	-0.70312	0.703124		1.23047									-3.59122	17.5444		
											-0.70312															
92146				196	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.70312	0.703124	37.9688	1.23047	0	21	1.7	-0.16164	-0.24489	-3.59122	17.5444					
										10.5907	-0.70312	0.703124		1.23047									-3.59122	17.5444		
											-0.70312															
92147				196	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.70312	0.703124	37.9688	1.23047	15.875	21	1.7	0.323277	-0.24489	-3.59122	17.5444					
										10.5907	-0.70312	0.703124		1.23047										-3.59122	17.5444	
											-0.70312															
92148	2	39	34	196	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.70312	0.703124	37.2656	1.23047	0	21	1.7	0.161641	-0.24489	-3.59122	17.5444					
										10.5907	-0.70312	0.703124		1.23047										-3.59122	17.5444	
											-0.70312															
92149				196	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.70312	1.05469	37.2656	1.23047	15.875	21	1.7	-0.56571	-0.24489	-3.59122	17.5444					
										10.5907	-0.70312	1.05469		1.23047										-3.59122	17.5444	
											-0.70312															
92150				196	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.52734	1.05469	37.2656	1.23047	0	21	1.7	-1.61561	-0.24489	-3.59122	17.5444					
										10.5907	-0.52734	1.05469		1.23047										-3.59122	17.5444	
											-0.52734															
92151				196	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.70312	0.703124	37.6172	1.23047	15.875	21	1.8	-2.01891	-0.24489	-3.59122	17.5444					
										10.5907	-0.70312	0.703124		1.23047										-3.59122	17.5444	
											-0.70312															
92152	2	39	38	196	45	-3.88063	-4.63334	0.969642	1.04419	10.5907	-0.52734	0.703124	38.6719	1.23047	0	21	1.7	-1.13121	-0.24489	-3.59122	17.5444					
										10.5907	-0.52734	0.703124		1.23047											-3.59122	17.5444
											-0.52734															
92153				196	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.52734	0.703124	39.375	1.23047	15.875	21	1.7	0.565711	-0.24489	-3.59122	17.5444					
										10.5907	-0.52734	0.703124		1.23047											-3.59122	17.5444
											-0.52734															
92154				196	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.52734	0.703124	39.375	1.23047	0	21	1.7	0.565711	-0.20992	-3.59122	17.5444					
										10.5907	-0.52734	0.703124		1.23047											-3.59122	17.5444
											-0.35156															
92155				192	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.35156	0.703124	39.375	1.23047	15.875	21	1.7	0	-0.38469	-3.59122	17.5444					
										10.5907	-0.35156	0.703124		1.23047											-3.64084	17.5444
											-0.35156															
92156	2	39	42	192	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.52734	0.703124	39.7266	1.23047	0	21	1.7	0.08082	-0.41961	-3.64084	17.5444					
										10.5907	-0.52734	0.703124		1.23047											-3.64084	17.5444
											-0.35156															
											-0.35156															



Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	ELEVATOR POSN L	ELEVATOR POSN R	AILERON POSN L	AILERON POSN R	SPD BRAKE HANDLE	PITCH ANGLE EFIS	ROLL ANGLE EFIS	MAGNETI HEADING EFIS	AOA	N1 L	N1 R	PITCH TRIM POSITION	RUDDER POSN	RUDDER PEDAL POSN	CONTROL COLUMN POSN	CONTROL WHEEL POSN	
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	()	()	()	()	()	(DEG)	(DEG)	(DEG)	(DEG)	(%RPM)	(%RPM)	()	()	()	()	()	
92157				196	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	39.7266	1.23047	15.875	21	1.7	-0.40409	-0.34976	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047						-3.64084	17.5444	
											-0.35156											
											-0.35156											
92158				192	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	39.7266	1.23047	0	21	1.7	0.323277	-0.34976	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.23047						-3.64084	17.5444	
											-0.35156											
											-0.35156											
92159				192	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.35156	0.703124	39.7266	1.23047	15.875	21	1.7	-0.32326	-0.34976	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.05469						-3.64084	17.5444	
											-0.35156											
											-0.52734											
92160	2	39	46	192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	39.7266	1.05469	0	21	1.7	0	-0.27985	-3.64084	17.5444	
										10.5907	-0.35156	1.05469		1.23047						-3.64084	17.5444	
											-0.35156											
											-0.35156											
92161				192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	1.05469	39.375	1.05469	15.875	21	1.7	0.161641	-0.27985	-3.64084	17.5444	
										10.5907	-0.52734	1.05469		1.23047						-3.64084	17.5444	
											-0.52734											
											-0.35156											
92162				192	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.35156	1.05469	39.375	1.23047	0	21	1.7	-0.96967	-0.27985	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.23047						-3.64084	17.5444	
											-0.35156											
											-0.35156											
92163				192	45	-3.88063	-4.69666	0.969642	1.04419	10.5907	-0.52734	1.05469	39.7266	1.23047	15.875	21	1.7	-0.72731	-0.24489	-3.64084	17.5444	
										10.5907	-0.35156	1.05469		1.05469						-3.64084	17.5444	
											-0.35156											
											-0.35156											
92164	2	39	50	192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	1.05469	39.7266	1.23047	0	21	1.7	0	-0.27985	-3.64084	17.5444	
										10.5907	-0.35156	1.05469		1.05469						-3.64084	17.5444	
											-0.35156											
											-0.52734											
92165				192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	1.05469	39.375	1.05469	15.875	21	1.7	-0.08082	-0.24489	-3.64084	17.5444	
										10.5907	-0.52734	1.05469		1.23047						-3.64084	17.5444	
											-0.52734											
											-0.52734											
92166				192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	1.05469	39.375	1.05469	0	21	1.7	-0.56571	-0.27985	-3.64084	17.5444	
										10.5907	-0.52734	1.05469		1.23047						-3.64084	17.5444	
											-0.52734											
											-0.52734											
92167				192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	1.05469	39.375	1.23047	15.875	21	1.7	-0.24244	-0.24489	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.05469						-3.64084	17.5444	
											-0.52734											
											-0.35156											
92168	2	39	54	192	45	-3.94032	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	39.375	1.05469	0	21	1.7	-0.24244	-0.24489	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.23047						-3.64084	17.5444	
											-0.35156											
											-0.35156											
92169				192	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.35156	1.05469	39.375	1.23047	15.875	21	1.8	-0.32326	-0.24489	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.05469						-3.64084	17.5444	
											-0.35156											
											-0.35156											
92170				192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	39.7266	1.05469	0	21	1.7	-0.08082	-0.24489	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.05469						-3.64084	17.5444	
											-0.35156											
											-0.35156											
92171				192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	39.375	1.23047	15.875	21	1.7	0.565711	-0.24489	-3.59122	17.5444	
										10.5907	-0.35156	1.05469		1.05469						-3.64084	17.5444	
											-0.35156											
											-0.35156											
92172	2	39	58	192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.17578	0.703124	39.0234	1.23047	0	21	1.7	0.404091	-0.20992	-3.64084	17.5444	
										10.5907	-0.17578	0.703124		1.05469						-3.64084	17.5444	

Time (seconds)	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )	
											-0.17578											
											-0.35156											
92173				192	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.35156	1.05469	38.3203	1.23047	15.875	21	1.7	-1.13121	-0.24489	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.23047							-3.64084	17.5444
											-0.35156											
											-0.17578											
92174				192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	1.05469	38.3203	1.23047	0	21	1.7	-0.24244	-0.20992	-3.64084	17.5444	
										10.5907	-0.35156	1.05469		1.05469							-3.64084	17.5444
											-0.35156											
											-0.52734											
92175				192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	1.05469	38.3203	1.23047	15.875	21	1.7	-0.88889	-0.17494	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.05469							-3.64084	17.5444
											-0.52734											
											-0.52734											
92176	2	40	2	192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	38.3203	1.05469	0	21	1.7	-1.5349	-0.20992	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92177				192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	39.0234	1.23047	15.875	21	1.7	-0.16164	-0.20992	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.05469							-3.64084	17.5444
											-0.35156											
											-0.35156											
92178				192	45	-3.94032	-4.63334	0.969642	0.969645	10.5907	-0.35156	0.703124	39.0234	1.05469	0	21	1.7	0.404091	-0.20992	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.23047							-3.64084	17.5444
											-0.35156											
											-0.35156											
92179				192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	38.6719	1.05469	15.875	21	1.7	0.323277	-0.13996	-3.64084	17.5444	
										10.5907	-0.52734	0.351562		1.23047							-3.64084	17.5444
											-0.52734											
											-0.35156											
92180	2	40	6	192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.351562	38.3203	1.05469	0	21	1.7	0.08082	-0.10497	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.35156											
92181				188	45	-3.76128	-4.63334	0.969642	0.969645	10.5907	-0.52734	1.05469	38.3203	1.23047	15.875	21	1.7	0	-0.13996	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.05469							-3.64084	17.5444
											-0.35156											
											-0.35156											
92182				192	45	-3.94032	-4.63334	0.969642	0.969645	10.5907	-0.35156	0.703124	38.3203	1.05469	0	21	1.7	-1.13121	-0.17494	-3.64084	17.5444	
										10.5907	-0.35156	0.351562		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92183				192	45	-3.76128	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.351562	38.6719	1.23047	15.875	21	1.8	-0.48489	-0.17494	-3.59122	17.5444	
										10.5907	-0.52734	0.703124		1.05469							-3.64084	17.5444
											-0.52734											
											-0.52734											
92184	2	40	10	192	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.52734	0.351562	38.6719	1.23047	0	21	1.7	0.404091	-0.20992	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.05469							-3.64084	17.5444
											-0.52734											
											-0.52734											
92185				188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	38.6719	1.05469	15.875	21	1.7	0.323277	-0.24489	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.35156											
											-0.35156											
92186				192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	38.3203	1.05469	0	21	1.8	-0.72731	-0.20992	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.23047							-3.64084	17.5444
											-0.35156											
											-0.35156											
92187				192	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	38.6719	1.23047	15.875	21	1.7	-0.32326	-0.20992	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.23047							-3.59122	17.5444
											-0.35156											
											-0.35156											

Time (seconds)	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )	
92188	2	40	14	188	45	-3.94032	-4.63334	0.969642	0.969645	10.5907	-0.35156	0.703124	38.6719	1.23047	0	21	1.8	-0.16164	-0.20992	-3.59122	17.5444	
										10.5907	-0.35156	0.703124		1.23047							-3.64084	17.5444
											-0.35156											
											-0.35156											
92189				192	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.35156	1.05469	39.0234	1.05469	15.875	21	1.7	-0.24244	-0.20992	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.23047							-3.64084	17.5444
											-0.35156											
											-0.35156											
92190				188	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	39.0234	1.23047	0	21	1.7	-0.48489	-0.20992	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.05469							-3.59122	17.5444
											-0.52734											
											-0.52734											
92191				188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	39.375	1.05469	15.875	21	1.7	0.323277	-0.24489	-3.59122	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92192	2	40	18	188	45	-3.94032	-4.63334	0.969642	0.969645	10.5907	-0.52734	0.703124	39.0234	1.05469	0	21	1.7	-0.16164	-0.24489	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.23047							-3.64084	17.5444
											-0.35156											
											-0.35156											
92193				188	45	-3.94032	-4.63334	0.969642	0.969645	10.5907	-0.35156	0.703124	39.375	1.23047	15.875	21	1.7	-0.24244	-0.24489	-3.64084	17.5444	
										10.5907	-0.17578	0.703124		1.23047							-3.64084	17.5444
											-0.17578											
											-0.35156											
92194				188	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.35156	0.703124	39.375	1.05469	0	21	1.8	-0.48489	-0.20992	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92195				188	45	-3.82096	-4.63334	0.969642	0.969645	10.5907	-0.35156	0.703124	39.7266	1.23047	15.875	21	1.8	-0.72731	-0.20992	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.23047							-3.64084	17.5444
											-0.35156											
											-0.52734											
92196	2	40	22	188	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	40.0781	1.23047	0	21	1.8	-0.16164	-0.20992	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92197				188	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	40.0781	1.23047	15.875	21	1.7	0.565711	-0.20992	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92198				188	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	40.0781	1.23047	0	21	1.7	-0.32326	-0.20992	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92199				188	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.52734	0.351562	40.0781	1.23047	15.875	21	1.7	1.21197	-0.24489	-3.64084	17.5444	
										10.5907	-0.52734	0.351562		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92200	2	40	26	188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.351562	39.7266	1.23047	0	21	1.8	-0.24244	-0.24489	-3.64084	17.5444	
										10.5907	-0.52734	0.351562		1.05469							-3.64084	17.5444
											-0.52734											
											-0.52734											
92201				188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	39.375	1.23047	15.875	21	1.7	-0.16164	-0.24489	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.23047							-3.64084	17.5444
											-0.35156											
											-0.52734											
92202				188	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	39.0234	1.23047	0	21	1.7	-0.40409	-0.24489	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92203				188	45	-3.82096	-4.63334	0.969642	0.969645	10.5907	-0.70312	0.703124	39.0234	1.23047	15.875	21	1.7	-0.72731	-0.24489	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444

Time (seconds)	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )	
											-0.52734											
											-0.52734											
92204	2	40	30	188	45	-3.82096	-4.63334	0.969642	0.969645	10.5907	-0.52734	0.703124	39.0234	1.23047	0	20.875	1.8	-0.24244	-0.24489	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92205				188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	1.05469	39.0234	1.23047	15.875	21	1.7	-0.64651	-0.20992	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92206				188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	39.0234	1.23047	0	21	1.7	-0.32326	-0.20992	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92207				188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	39.375	1.23047	15.875	21	1.8	-0.24244	-0.20992	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92208	2	40	34	188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	39.0234	1.23047	0	21	1.7	0.646514	-0.24489	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92209				188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	38.6719	1.23047	15.875	21	1.7	-0.08082	-0.24489	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.35156											
92210				188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	38.3203	1.23047	0	21	1.7	-0.24244	-0.20992	-3.64084	17.5444	
										10.5907	-0.35156	1.05469		1.23047							-3.64084	17.5444
											-0.35156											
											-0.35156											
92211				188	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.35156	0.703124	38.3203	1.23047	15.875	21	1.7	-0.48489	-0.20992	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92212	2	40	38	188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	37.9688	1.23047	0	21	1.7	-0.40409	-0.20992	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.70312											
											-0.70312											
92213				188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.70312	0.703124	37.9688	1.23047	15.875	21	1.7	-0.08082	-0.17494	-3.64084	17.5444	
										10.5907	-0.70312	0.703124		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											
92214				188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	37.9688	1.23047	0	21	1.7	-0.32326	-0.20992	-3.64084	17.5444	
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444
											-0.35156											
											-0.35156											
92215				184	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	37.9688	1.23047	15.875	21	1.8	-0.48489	-0.20992	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.23047							-3.64084	17.5444
											-0.35156											
											-0.35156											
92216	2	40	42	188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.17578	1.05469	37.9688	1.23047	0	21	1.8	-0.64651	-0.20992	-3.64084	17.5444	
										10.5907	-0.35156	1.05469		1.23047							-3.64084	17.5444
											-0.35156											
											-0.35156											
92217				188	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.35156	1.05469	37.9688	1.23047	15.875	20.875	1.8	-0.48489	-0.20992	-3.64084	17.5444	
										10.5907	-0.35156	0.703124		1.23047							-3.64084	17.5444
											-0.35156											
											-0.35156											
92218				188	45	-3.94032	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	38.3203	1.23047	0	20.875	1.8	-0.32326	-0.20992	-3.64084	17.5444	
										10.5907	-0.52734	1.05469		1.23047							-3.64084	17.5444
											-0.52734											
											-0.52734											

Time (seconds)	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )				
92219				188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	38.3203	1.23047	15.875	20.875	1.8	-0.16164	-0.17494	-3.64084	17.5444				
										10.5907	-0.52734	0.703124		1.23047							-3.64084	17.5444			
											-0.52734														
											-0.52734														
92220	2	40	46	188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	38.3203	1.23047	0	20.875	1.7	-0.24244	-0.17494	-3.64084	17.5444				
										10.5907	-0.52734	0.703124		1.23047								-3.64084	17.5444		
											-0.52734														
											-0.52734														
92221				188	45	-3.82096	-4.63334	0.969642	0.969645	10.5907	-0.52734	0.703124	38.3203	1.23047	15.875	20.875	1.8	-0.24244	-0.17494	-3.64084	17.5444				
										10.5907	-0.52734	0.703124		1.23047									-3.64084	17.5444	
											-0.52734														
											-0.52734														
92222				184	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.703124	38.6719	1.23047	0	20.875	1.7	-0.16164	-0.17494	-3.64084	17.5444				
										10.5907	-0.52734	0.703124		1.05469									-3.64084	17.5444	
											-0.52734														
											-0.35156														
92223				188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	1.05469	38.6719	1.23047	15.875	21	1.8	0	-0.17494	-3.64084	17.5444				
										10.5907	-0.35156	0.703124		1.23047									-3.64084	17.5444	
											-0.35156														
											-0.35156														
92224	2	40	50	188	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	38.6719	1.23047	0	21	1.7	-0.08082	-0.20992	-3.64084	17.5444				
										10.5907	-0.35156	0.703124		1.23047										-3.64084	17.5444
											-0.35156														
											-0.35156														
92225				184	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.35156	0.703124	38.6719	1.23047	15.875	20.875	1.8	0.08082	-0.17494	-3.64084	17.8705				
										10.5907	-0.35156	0.703124		1.23047										-3.64084	17.8705
											-0.52734														
											-0.52734														
92226				184	45	-3.88063	-4.57003	0.969642	0.969645	10.5907	-0.52734	0.703124	38.6719	1.23047	0	20.875	1.8	-0.96967	-0.17494	-3.64084	17.8705				
										10.5907	-0.52734	0.703124		1.23047										-3.64084	17.8705
											-0.52734														
											-0.52734														
92227				184	45	-3.82096	-4.57003	0.969642	0.969645	10.5907	-0.52734	0.703124	39.0234	1.23047	15.875	20.875	1.8	-1.77697	-0.17494	-3.64084	17.8705				
										10.5907	-0.52734	0.703124		1.23047										-3.64084	17.8705
											-0.35156														
											-0.35156														
92228	2	40	54	184	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.35156	0.703124	40.0781	1.23047	0	20.875	1.7	-1.45419	-0.20992	-3.64084	17.8705				
										10.5907	-0.35156	0.703124		1.23047										-3.64084	17.8705
											-0.35156														
											-0.35156														
92229				184	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.35156	0.703124	41.4844	1.23047	15.875	21	1.7	-1.5349	-0.20992	-3.64084	17.8705				
										10.5907	-0.35156	0.703124		1.23047										-3.64084	17.8705
											-0.35156														
											-0.35156														
92230				184	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.52734	0.351562	42.8906	1.23047	0	21	1.7	-2.01891	-0.20992	-3.64084	17.8705				
										10.5907	-0.52734	0.351562		1.23047										-3.64084	17.8705
											-0.52734														
											-0.52734														
92231				184	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.52734	0.351562	45	1.23047	15.875	21	1.8	-1.29272	-0.24489	-3.64084	17.8705				
										10.5907	-0.52734	0.351562		1.23047										-3.64084	17.8705
											-0.52734														
											-0.52734														
92232	2	40	58	184	45	-3.82096	-4.63334	0.969642	0.969645	10.5907	-0.52734	0.351562	47.1094	1.23047	0	20.875	1.8	-0.80809	-0.24489	-3.64084	17.8705				
										10.5907	-0.52734	0.351562		1.05469										-3.64084	17.8705
											-0.52734														
											-0.52734														
92233				184	45	-3.82096	-4.63334	0.969642	0.969645	10.5907	-0.52734	0.351562	49.2188	1.23047	15.875	20.875	1.8	-0.88889	-0.24489	-3.64084	17.8705				
										10.5907	-0.52734	0.351562		1.23047										-3.64084	17.8705
											-0.52734														
											-0.52734														
92234				184	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.52734	0.351562	52.0312	1.23047	0	20.875	1.7	-2.18013	-0.27985	-3.64084	17.8705				
										10.5907	-0.35156	0.351562		1.23047										-3.64084	17.8705

Time (seconds)	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )		
											-0.52734												
											-0.52734												
92235				184	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.351562	54.8438	1.23047	15.875	21	1.8	-3.22628	-0.27985	-3.64084	17.8705		
										10.5907	-0.52734	0.351562		1.23047							-3.64084	17.8705	
											-0.52734												
											-0.52734												
92236	2	41	2	184	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0.351562	59.7656	1.23047	0	21	1.8	-3.3066	-0.27985	-3.64084	17.8705		
										10.5907	-0.52734	0.351562		1.23047								-3.64084	17.8705
											-0.70312												
											-0.70312												
92237				184	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.70312	0.351562	63.9844	1.23047	15.875	21	1.8	-2.74389	-0.27985	-3.64084	17.5444		
										10.5907	-0.8789	0.351562		1.23047								-3.64084	17.8705
											-0.8789												
											-0.8789												
92238				184	45	-3.82096	-4.63334	0.969642	0.969645	10.5907	-0.8789	0.351562	69.2578	1.23047	0	21	1.8	-2.26074	-0.27985	-3.64084	17.5444		
										10.5907	-0.8789	0.703124		1.23047								-3.64084	17.5444
											-0.70312												
											-0.70312												
92239				184	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.70312	0.703124	74.1797	1.23047	15.875	21.125	1.8	-1.61561	-0.31481	-3.64084	17.5444		
										10.5907	-0.70312	0.351562		1.23047								-3.64084	17.5444
											-0.52734												
											-0.52734												
92240	2	41	6	184	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	0	80.1562	1.23047	0	21.125	1.8	-1.61561	-0.27985	-3.64084	17.5444		
										10.5907	-0.35156	0		1.23047								-3.64084	17.5444
											-0.35156												
											-0.35156												
92241				184	45	-3.94032	-4.63334	0.969642	0.969645	10.5907	-0.35156	0	85.0781	1.05469	15.875	21.125	1.7	-1.29272	-0.27985	-3.64084	17.5444		
										10.5907	-0.35156	-0.35156		1.23047								-3.64084	17.5444
											-0.35156												
											-0.52734												
92242				184	45	-3.82096	-4.63334	0.969642	0.969645	10.5907	-0.52734	-0.35156	91.7578	1.23047	0	21	1.8	-0.56571	-0.27985	-3.64084	17.5444		
										10.5907	-0.52734	-0.35156		1.23047								-3.64084	17.5444
											-0.70312												
											-0.70312												
92243				184	45	-3.82096	-4.63334	0.969642	0.969645	10.5907	-0.70312	-0.35156	96.6797	1.23047	15.875	21	1.8	0.888893	-0.27985	-3.64084	17.5444		
										10.5907	-0.70312	-0.35156		1.23047								-3.64084	17.5444
											-0.70312												
											-0.70312												
92244	2	41	10	184	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.70312	-0.70312	102.656	1.23047	0	21	1.8	1.29271	-0.27985	-3.64084	17.5444		
										10.5907	-0.70312	-0.35156		1.23047								-3.64084	17.5444
											-0.70312												
											-0.70312												
92245				184	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.70312	-0.35156	106.875	1.23047	15.875	21	1.8	1.37345	-0.27985	-3.64084	17.5444		
										10.5907	-0.70312	-0.35156		1.23047								-3.64084	17.5444
											-0.70312												
											-0.70312												
92246				184	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.70312	-0.35156	112.5	1.23047	0	21	1.7	1.37345	-0.27985	-3.64084	17.5444		
										10.5907	-0.70312	-0.35156		1.23047								-3.64084	17.5444
											-0.70312												
											-0.70312												
92247				184	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.70312	-0.35156	116.719	1.05469	15.875	21	1.7	1.69629	-0.27985	-3.64084	17.5444		
										10.5907	-0.70312	-0.35156		1.23047								-3.64084	17.5444
											-0.70312												
											-0.70312												
92248	2	41	14	184	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.70312	-0.35156	121.641	1.23047	0	21	1.7	2.01892	-0.27985	-3.64084	17.5444		
										10.5907	-0.70312	-0.35156		1.23047								-3.59122	17.5444
											-0.70312												
											-0.70312												
92249				184	45	-3.94032	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.35156	124.805	1.23047	15.875	20.875	1.8	3.46716	-0.27985	-3.59122	17.5444		
										10.5907	-0.52734	-0.35156		1.23047								-3.59122	17.5444
											-0.52734												
											-0.52734												

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )	
92250				184	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.35156	127.266	1.23047	0	20.875	1.8	3.46716	-0.27985	-3.59122	17.5444	
										10.5907	-0.52734	-0.35156		1.23047						-3.59122	17.5444	
											-0.52734											
											-0.52734											
92251				184	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.35156	128.672	1.23047	15.875	20.875	1.7	3.46716	-0.31481	-3.59122	17.5444	
										10.5907	-0.52734	-0.35156		1.23047						-3.59122	17.5444	
											-0.52734											
											-0.52734											
92252	2	41	18	184	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.35156	129.375	1.23047	0	20.875	1.8	2.98518	-0.31481	-3.59122	17.5444	
										10.5907	-0.52734	-0.35156		1.23047						-3.59122	17.5444	
											-0.70312											
											-0.70312											
92253				184	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.70312	-0.35156	130.43	1.23047	15.875	20.875	1.7	1.93828	-0.27985	-3.59122	17.5444	
										10.5907	-0.70312	-0.35156		1.23047						-3.59122	17.5444	
											-0.70312											
											-0.70312											
92254				184	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.70312	-0.35156	131.133	1.23047	0	20.875	1.8	1.13121	-0.20992	-3.59122	17.5444	
										10.5907	-0.70312	-0.35156		1.23047						-3.59122	17.5444	
											-0.52734											
											-0.70312											
92255				184	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.70312	-0.70312	131.836	1.23047	15.875	20.875	1.8	0.808106	-0.24489	-3.59122	17.5444	
										10.5907	-0.52734	-0.35156		1.23047						-3.59122	17.5444	
											-0.52734											
											-0.52734											
92256	2	41	22	184	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.70312	132.539	1.23047	0	20.875	1.8	0.323277	-0.24489	-3.59122	17.5444	
										10.5907	-0.52734	-0.70312		1.23047						-3.59122	17.5444	
											-0.52734											
											-0.52734											
92257				184	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.70312	133.242	1.23047	15.875	20.875	1.8	-0.24244	-0.34976	-3.59122	17.5444	
										10.5907	-0.52734	-0.70312		1.23047						-3.59122	17.5444	
											-0.52734											
											-0.52734											
92258				184	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.70312	133.594	1.23047	0	20.875	1.8	-0.40409	-0.38469	-3.59122	17.5444	
										10.5907	-0.52734	-0.70312		1.23047						-3.59122	17.5444	
											-0.52734											
											-0.52734											
92259				180	45	-3.82096	-4.63334	0.969642	0.969645	10.5907	-0.52734	-0.70312	134.297	1.23047	15.875	20.875	1.8	-0.56571	-0.38469	-3.59122	17.5444	
										10.5907	-0.52734	-0.70312		1.23047						-3.59122	17.5444	
											-0.52734											
											-0.52734											
92260	2	41	26	180	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.70312	134.648	1.05469	0	20.875	1.8	-0.80809	-0.38469	-3.59122	17.5444	
										10.5907	-0.52734	-0.70312		1.23047						-3.59122	17.5444	
											-0.52734											
											-0.52734											
92261				180	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.70312	135	1.23047	15.875	20.875	1.8	-0.80809	-0.38469	-3.59122	17.5444	
										10.5907	-0.52734	-0.70312		1.23047						-3.59122	17.5444	
											-0.52734											
											-0.52734											
92262				180	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.70312	135	1.23047	0	20.875	1.8	0.727313	-0.34976	-3.59122	17.5444	
										10.5907	-0.52734	-0.70312		1.05469						-3.64084	17.5444	
											-0.52734											
											-0.52734											
92263				180	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.70312	134.648	1.23047	15.875	20.875	1.8	-0.16164	-0.34976	-3.64084	17.5444	
										10.5907	-0.52734	-0.70312		1.23047						-3.64084	17.5444	
											-0.52734											
											-0.52734											
92264	2	41	30	180	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.70312	134.297	1.05469	0	20.875	1.8	-0.80809	-0.34976	-3.64084	17.5444	
										10.5907	-0.52734	-0.70312		1.05469						-3.64084	17.5444	
											-0.52734											
											-0.52734											
92265				180	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.70312	133.945	1.23047	15.875	20.875	1.8	-2.09953	-0.34976	-3.59122	17.5444	
										10.5907	-0.52734	-0.35156		1.23047						-3.64084	17.5444	

Time (seconds)	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )	
											-0.52734											
											-0.52734											
92266				180	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.35156	134.297	1.23047	0	20.875	1.8	-2.01891	-0.31481	-3.64084	17.5444	
										10.5907	-0.52734	-0.35156		1.05469							-3.59122	17.5444
											-0.52734											
											-0.52734											
92267				180	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.35156	135	1.05469	15.875	20.875	1.8	-1.85764	-0.31481	-3.59122	17.5444	
										10.5907	-0.52734	-0.35156		1.23047							-3.59122	17.5444
											-0.52734											
											-0.52734											
92268	2	41	34	180	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.35156	135.352	1.23047	0	20.875	1.8	-1.69628	-0.31481	-3.59122	17.5444	
										10.5907	-0.52734	-0.35156		1.23047							-3.59122	17.5444
											-0.52734											
											-0.52734											
92269				180	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.35156	136.406	1.23047	15.875	20.875	1.8	-1.85764	-0.27985	-3.59122	17.5444	
										10.5907	-0.52734	-0.35156		1.23047							-3.59122	17.5444
											-0.52734											
											-0.52734											
92270				180	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	-0.35156	137.109	1.23047	0	20.75	1.8	-2.09953	-0.27985	-3.64084	17.5444	
										10.5907	-0.70312	-0.35156		1.23047							-3.64084	17.5444
											-0.70312											
											-0.8789											
92271				180	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.8789	-0.35156	138.867	1.23047	15.875	20.75	1.8	-3.86808	-0.27985	-3.64084	17.5444	
										10.5907	-0.8789	-0.70312		1.23047							-3.64084	17.8705
											-0.8789											
											-0.8789											
92272	2	41	38	180	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.8789	-0.70312	141.328	1.23047	0	20.75	1.8	-4.10832	-0.27985	-3.64084	17.8705	
										10.5907	-0.8789	-0.70312		1.23047							-3.64084	17.8705
											-0.70312											
											-0.70312											
92273				180	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.52734	-1.05469	146.602	1.23047	15.875	20.75	1.8	-4.10832	-0.27985	-3.64084	17.8705	
										10.5907	-0.35156	-1.05469		1.23047							-3.64084	17.8705
											-0.17578											
											-0.17578											
92274				180	45	-3.94032	-4.63334	0.969642	0.969645	10.5907	0	-0.70312	152.227	1.23047	0	20.75	1.8	-4.10832	-0.27985	-3.64084	17.5444	
										10.5907	0	-0.70312		1.23047							-3.64084	17.8705
											0											
											0											
92275				180	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	0	-0.70312	160.664	1.23047	15.875	20.75	1.8	-4.10832	-0.27985	-3.64084	17.5444	
										10.5907	0	-0.35156		1.23047							-3.64084	17.8705
											0											
											-0.17578											
92276	2	41	42	180	45	-3.88063	-4.69666	0.969642	0.969645	10.5907	-0.17578	-0.35156	167.695	1.23047	0	20.875	1.8	-4.10832	-0.27985	-3.64084	17.8705	
										10.5907	-0.17578	-0.70312		1.23047							-3.64084	17.8705
											-0.17578											
											-0.17578											
92277				180	45	-3.82096	-4.63334	0.969642	0.969645	10.5907	-0.17578	-0.70312	175.078	1.23047	15.875	20.875	1.8	-1.77697	-0.27985	-3.64084	17.8705	
										10.5907	-0.17578	-0.70312		1.23047							-3.64084	17.8705
											-0.17578											
											-0.17578											
92278				180	45	-3.82096	-4.63334	0.969642	0.969645	10.5907	-0.17578	-0.70312	182.109	1.23047	0	21	1.8	0.404091	-0.27985	-3.64084	17.8705	
										10.5907	-0.17578	-0.70312		1.05469							-3.64084	17.8705
											-0.17578											
											-0.17578											
92279				180	45	-3.82096	-4.69666	0.969642	0.969645	10.5907	-0.17578	-0.70312	188.438	1.23047	15.875	21	1.8	1.6156	-0.27985	-3.64084	18.195	
										10.5907	-0.17578	-0.70312		1.23047							-3.64084	18.195
											-0.17578											
											-0.17578											
92280	2	41	46	180	45	-3.88063	-4.63334	0.969642	0.969645	10.5907	-0.17578	-0.70312	193.711	1.23047	0	20.875	1.8	3.06557	-0.27985	-3.64084	17.8705	
										10.5907	-0.17578	-0.70312		1.23047							-3.64084	17.8705
											-0.17578											
											-0.17578											





Time (seconds)	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETIC HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )	
											0.35156											
											0.35156											
92297				188	45	-3.76128	-4.75997	1.64028	0.298401	10.5907	0.17578	-0.70312	223.594	1.23047	15.875	70.875	1.7	-6.25592	-5.54503	-3.69037	14.1923	
										10.5907	0.17578	-0.70312		1.23047							-3.7398	12.4588
											0.17578											
											0.17578											
92298				188	45	-3.64193	-4.38003	2.38422	-1.11874	10.5907	0.17578	-0.70312	223.594	1.23047	0	78.875	1.8	-14.5619	-7.07086	-3.83835	9.61627	
										10.5907	0.17578	-1.05469		1.23047							-3.83835	9.25566
											0.17578											
											0.17578											
92299				188	45	-3.16465	-4.5067	3.12636	-1.19327	10.5907	0.17578	-1.05469	223.945	1.23047	15.875	79.5	1.7	-18.7308	-5.56834	-3.83835	8.894	
										10.5907	0.17578	-1.05469		1.05469							-3.69037	9.61627
											0.17578											
											0.17578											
92300	2	42	6	192	45	-3.88063	-4.57003	3.20046	-1.19327	10.5907	0.35156	-1.05469	223.594	1.23047	0	82.75	1.7	-13.3384	-6.64858	-3.78912	8.894	
										10.5907	0.35156	-1.75781		1.23047							-3.83835	8.894
											0.35156											
											0.17578											
92301				192	45.5	-3.58224	-4.82328	3.20046	-1.19327	10.5907	0.17578	-1.05469	223.594	1.23047	15.875	83.75	1.7	-16.4512	-6.27028	-3.78912	9.25566	
										10.5907	0.17578	-1.40625		1.23047							-3.7398	9.61627
											0.17578											
											0.17578											
92302				192	49.5	-3.82096	-4.63334	3.12636	-0.373	10.5907	0.17578	-1.05469	222.891	1.23047	0	84.625	1.7	-13.4836	-2.84137	-3.64084	11.4022	
										10.5907	0.17578	-1.05469		1.23047							-3.7398	12.8083
											0.35156											
											0.17578											
92303				196	56	-3.82096	-4.88658	2.3099	-0.2984	10.5907	0.17578	-1.05469	222.188	1.23047	15.875	87.25	1.7	-9.21957	-4.35142	-3.69037	13.1564	
										10.5907	0.17578	-1.05469		1.05469							-3.69037	13.1564
											0.35156											
											0.35156											
92304	2	42	10	196	61	-3.52258	-4.5067	2.3099	-0.1492	10.5907	0.17578	-1.40625	222.188	1.05469	0	89.5	1.7	-11.5695	-0.69845	-3.69037	13.5032	
										10.5907	0.17578	-1.05469		1.23047							-3.69037	13.8484
											0											
											0.17578											
92305				196	65	-3.7016	-4.44337	2.08683	-0.0746	10.5907	0.35156	-1.40625	222.188	1.05469	15.875	89.875	1.7	-7.90374	-4.46056	-3.78912	13.5032	
										10.5907	0.35156	-1.40625		1.05469							-3.83835	13.1564
											0.17578											
											0.35156											
92306				196	70	-3.64193	-4.3167	2.1612	-0.2238	10.5907	0.35156	-1.40625	222.891	1.23047	0	90	1.7	-12.9738	-6.0167	-3.88747	12.1079	
										10.5907	0.35156	-1.40625		1.23047							-3.88747	12.4588
											0.35156											
											0.35156											
92307				200	75.5	-3.88063	-4.57003	2.45853	-0.1492	10.5907	0.35156	-1.40625	222.891	1.05469	15.875	90.5	1.7	-13.3384	-3.51894	-3.83835	13.1564	
										10.5907	0.35156	-1.40625		1.05469							-3.78912	13.1564
											0.35156											
											0.17578											
92308	2	42	14	200	78.5	-3.7016	-4.44337	2.3099	-0.0746	10.5907	0.17578	-1.05469	222.188	0.878905	0	90.625	1.7	-1.85764	-2.48924	-3.78912	13.5032	
										10.5907	0.17578	-1.40625		0.878905							-3.78912	13.5032
											0.35156											
											0.35156											
92309				200	83.5	-3.58224	-4.38003	2.23556	-0.0746	10.5907	0.35156	-1.05469	222.188	0.878905	15.875	90.5	1.7	-4.5879	-3.30745	-3.78912	13.5032	
										10.5907	0.35156	-1.40625		0.878905							-3.88747	13.8484
											0.17578											
											0.17578											
92310				200	89	-3.64193	-4.3167	1.86362	3.7923	10.5907	0.35156	-1.05469	222.539	0.703124	0	90.375	1.7	-12.1643	-4.80545	-3.93649	21.9487	
										10.5907	0.35156	-1.40625		1.05469							-3.88747	30.1527
											0.35156											
											0.35156											
92311				200	93	-3.46291	-4.38003	-2.38422	4.52948	10.5907	0.17578	-1.05469	222.188	0.878905	15.875	90.375	1.7	-3.9482	-1.11405	-3.88747	29.9064	
										10.5907	0.17578	-1.05469		0.703124							-3.83835	29.9064
											0.35156											
											0.35156											



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )	
											7.73436											
											7.91014											
92328	2	42	34	196	159	2.3543	1.81456	-3.86615	5.47019	10.5907	8.26171	-1.05469	221.133	10.7226	0	90.375	1.8	-2.42185	-1.35444	-7.5427	30.3972	
											10.5907	8.61327	-0.70312		10.8984						-7.65785	32.0582
											8.78905											
											8.96483											
92329				208	162	3.42392	3.15781	-2.60708	7.58571	10.5907	8.96483	-0.70312	220.781	10.7226	15.875	90.375	1.8	-1.93828	-1.25164	-7.84708	35.3321	
											10.5907	8.96483	-1.05469		10.1953						-8.17929	35.5372
											8.96483											
											8.96483											
92330				220	165.5	4.54443	2.62298	-2.97811	7.5159	10.5907	9.14061	-1.05469	220.781	10.3711	0	90.375	1.8	-2.42185	-1.2173	-7.84708	35.1254	
											10.5907	9.49217	-1.40625		10.7226						-7.6196	34.7073
											9.84374											
											10.5469											
92331				240	167.5	3.22443	0.726283	-2.68131	7.5159	10.5907	10.8984	-1.05469	220.781	11.6015	15.875	90.375	1.8	-1.29272	-0.55916	-7.34809	35.5372	
											10.5907	11.0742	-1.05469		11.9531						-6.90805	31.357
											11.9531											
											12.3047											
92332	2	42	38	268	169.5	1.47568	-0.67097	-0.2984	4.97021	10.5907	12.832	-0.70312	221.133	12.3047	0	90.375	1.8	-1.5349	-0.55916	-6.28172	26.8063	
											10.5907	13.0078	-0.35156		12.3047						-6.32441	28.39
											13.3594											
											13.7109											
92333				300	171.5	0.246849	-0.90708	0	3.20046	10.5907	13.8867	0	221.836	11.9531	15.875	90.5	1.8	-1.05045	-0.55916	-6.19593	25.1545	
											10.5907	14.0625	0		11.4258						-6.06626	18.8386
											14.414											
											14.5898											
92334				328	172	-0.55307	-4.06334	2.60708	2.75555	10.5907	14.7656	0.703124	222.188	11.4258	0	90.5	1.8	-0.56571	-0.55916	-5.49076	18.195	
											10.5907	15.1172	1.05469		11.25						-3.59122	16.887
											15.2929											
											15.6445											
92335				364	173	-4.44337	-5.07644	3.34857	0.969645	10.5907	15.6445	1.75781	222.539	10.8984	15.875	90.5	1.8	-0.48489	-0.55916	-2.88671	11.7557	
											10.5907	15.6445	2.10937		9.66795						-3.54149	8.894
											15.2929											
											15.1172											
92336	2	42	42	400	174	-3.82096	-4.5067	4.82345	0	10.5907	14.5898	1.75781	222.891	8.26171	0	90.625	1.8	-0.64651	-0.55916	-3.7398	8.5313	
											10.5907	14.414	1.05469		7.3828						-3.93649	8.894
											14.2383											
											13.8867											
92337				440	174.5	-3.64193	-4.69666	4.89685	0.074605	10.5907	13.8867	0.703124	223.594	7.55858	15.875	90.75	1.8	-0.64651	-0.55916	-3.54149	8.16762	
											10.5907	13.8867	0		7.91014						-3.03913	8.5313
											13.8867											
											13.8867											
92338				480	176	-4.63335	-5.96125	4.97022	-0.0746	10.5907	13.8867	-0.35156	223.945	8.08593	0	90.625	1.8	-0.08082	-0.55916	-2.68234	8.16762	
											10.5907	13.8867	-0.70312		7.73436						-2.9376	7.80299
											13.8867											
											13.7109											
92339				512	176.5	-4.75998	-5.77184	5.04242	-0.1492	10.5907	13.7109	-0.70312	223.945	7.20702	15.875	90.75	1.8	0.08082	-0.55916	-2.73355	7.80299	
											10.5907	13.3594	-0.35156		6.85546						-3.08977	7.43745
											13.1836											
											13.0078											
92340	2	42	46	548	177	-4.06334	-5.01316	5.1138	-0.2238	10.5907	12.832	-0.35156	223.945	6.5039	0	90.75	1.8	0	-0.55916	-3.29145	7.43745	
											10.5907	12.6562	-0.35156		6.5039						-3.49167	7.43745
											12.6562											
											12.6562											
92341				584	178	-4	-4.69666	5.1138	-0.1492	10.5907	12.6562	-0.35156	223.945	6.67968	15.875	90.75	1.8	0	-0.5243	-3.54149	7.43745	
											10.5907	12.6562	-0.35156		6.85546						-3.59122	9.61627
											12.6562											
											12.832											
92342				616	178.5	-3.82096	-4.69666	3.7184	2.68131	10.5907	12.832	-0.70312	223.945	7.20702	0	90.75	1.8	-0.08082	-0.5243	-3.59122	18.5176	
											10.5907	13.0078	-0.70312		7.55858						-3.19079	22.8457
											13.0078											
											13.1836											

Time (seconds)	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )
92343				652	179	-4.25337	-5.13971	1.64028	2.68131	10.5907 10.5907	13.1836 13.1836 13.1836	-0.70312 -0.35156	223.594	7.91014 7.73436	15.875	90.75	1.8	-0.40409	-0.48942	-3.14032 -3.19079	18.8386 19.1577
92344	2	42	50	688	178.5	-4.19003	-4.69666	2.45853	2.60708	10.5907 10.5907	13.1836 13.0078	-0.35156 0	223.594	7.20702 7.20702	0	90.875	1.8	-0.48489	-0.48942	-3.59122 -3.49167	18.5176 16.2229
92345				720	179.5	-3.82096	-4.75997	3.64449	0.820516	10.5907 10.5907	13.0078 13.0078	0 0	223.242	7.20702 7.3828	15.875	90.875	1.8	-0.64651	-0.48942	-3.44176 -3.49167	10.6914 18.5176
92346				756	179.5	-3.82096	-4.69666	2.53281	2.68131	10.5907 10.5907	13.0078 13.0078	-0.35156 -0.70312	223.242	7.3828 7.20702	0	90.875	1.8	-0.88889	-0.48942	-3.49167 -3.39175	19.1577 20.7266
92347				792	180	-3.94032	-4.75997	2.08683	2.75555	10.5907 10.5907	13.1836 13.1836	-0.70312 -0.70312	223.242	7.20702 7.3828	15.875	90.875	1.8	-0.48489	-0.48942	-3.49167 -3.49167	20.4165 18.5176
92348	2	42	54	832	180	-4	-4.88658	2.53281	1.86361	10.5907 10.5907	13.1836 13.1836 13.3594	-0.35156 -0.35156	222.891	7.3828 7.20702	0	90.875	1.8	-0.56571	-0.48942	-3.34164 -3.03913	17.2165 10.6914
92349				868	181	-4.44337	-5.32945	4.75001	-0.0746	10.5907 10.5907	13.3594 13.1836	0 0	222.891	7.20702 7.20702	15.875	90.875	1.8	-0.32326	-0.5243	-3.08977 -3.14032	7.80299 7.80299
92350				904	180.5	-4.25337	-4.82328	5.1138	0.522196	10.5907 10.5907	13.0078 13.0078	-0.35156 -0.70312	222.539	6.85546 6.5039	0	91	1.8	-0.88889	-0.48942	-3.29145 -3.69037	8.5313 12.8083
92351				940	181.5	-3.7016	-4.57003	3.93997	0.447591	10.5907 10.5907	12.832 12.832	-1.40625 -1.40625	222.539	6.85546 7.03124	15.875	90.875	1.8	-0.56571	-0.5243	-3.83835 -3.59122	11.7557 10.3342
92352	2	42	58	976	181	-3.76128	-4.57003	4.52949	1.11873	10.5907 10.5907	13.1836 13.3594	-1.40625 -1.75781	222.539	7.20702 7.20702	0	91	1.8	-0.24244	-0.48942	-3.59122 -3.64084	11.0474 12.8083
92353				1016	181.5	-3.76128	-4.63334	4.08754	0	10.5907 10.5907	13.5351 13.7109	-2.10937 -2.10937	222.188	7.73436 7.55858	15.875	91	1.8	-0.48489	-0.5243	-3.59122 -3.69037	8.5313 8.16762
92354				1052	181.5	-3.76128	-4.69666	4.67654	1.71474	10.5907 10.5907	13.7109 13.7109	-2.46093 -3.16406	221.836	7.03124 7.20702	0	91	1.8	-0.32326	-0.48942	-3.69037 -3.69037	12.4588 15.2146
92355				1096	183	-3.7016	-4.63334	3.34857	1.86361	10.5907 10.5907	13.8867 13.8867	-3.86718 -3.86718	221.484	7.3828 7.20702	15.875	91	1.8	-0.64651	-0.48942	-3.59122 -3.59122	14.8754 13.8484
92356	2	43	2	1136	183	-3.88063	-4.69666	3.86615	1.04419	10.5907 10.5907	13.8867 14.0625	-3.86718 -3.86718	221.133	7.03124 7.03124	0	91	1.7	-0.32326	-0.48942	-3.54149 -3.49167	11.7557 11.7557
92357				1180	184	-3.82096	-4.82328	4.23498	0.14921	10.5907 10.5907	14.0625 14.0625	-3.86718 -3.86718	220.43	7.03124 7.03124	15.875	91	1.7	-0.56571	-0.5243	-3.39175 -3.49167	9.61627 8.16762
92358				1220	184	-3.82096	-4.69666	5.25647	0.969645	10.5907 10.5907	14.2383 14.2383	-4.21874 -5.27343	220.078	7.03124 7.03124	0	91	1.7	-0.56571	-0.48942	-3.54149 -3.54149	9.25566 13.1564

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ()	ELEVATOR POSN R ()	AILERON POSN L ()	AILERON POSN R ()	SPD BRAKE HANDLE ()	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETIC HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ()	RUDDER POSN ()	RUDDER PEDAL POSN ()	CONTROL COLUMN POSN ()	CONTROL WHEEL POSN ()	
											14.2383											
											14.2383											
92359				1268	184	-3.88063	-4.63334	3.86615	1.04419	10.5907	14.0625	-6.32812	219.375	6.67968	15.875	90	1.7	-0.72731	-0.48942	-3.59122	12.8083	
											10.5907	14.0625	-6.67968		6.85546						-3.49167	7.07103
											14.0625											
											14.0625											
92360	2	43	6	1312	184	-3.94032	-4.69666	6.25067	-3.27453	10.5907	14.0625	-6.67968	219.023	6.67968	0	89.125	1.7	-0.24244	-0.5243	-3.49167	6.33574	
											10.5907	14.0625	-6.67968		6.85546						-3.44176	6.33574
											14.0625											
											14.0625											
92361				1352	183	-3.88063	-4.57003	8.48829	-2.01244	10.5907	14.0625	-7.3828	218.32	6.67968	15.875	89.125	1.8	-0.08082	-0.5243	-3.59122	11.0474	
											10.5907	13.8867	-8.43749		6.5039						-3.78912	16.887
											13.8867											
											13.8867											
92362				1396	184	-3.52258	-3.94032	5.1138	0.671366	10.5907	13.8867	-10.8984	216.914	6.32812	0	89.125	1.7	-0.56571	-0.5243	-4.18001	22.8457	
											10.5907	13.7109	-12.3047		6.5039						-4.08292	21.646
											13.7109											
											13.8867											
92363				1440	184	-3.22428	-3.94032	4.82345	0.373006	10.5907	13.8867	-12.6562	215.859	7.20702	15.875	89.125	1.8	-0.40409	-0.5243	-4.13152	21.646	
											10.5907	13.8867	-13.3594		7.20702						-4.13152	22.5486
											14.0625											
											14.0625											
92364	2	43	10	1484	183.5	-3.16465	-3.76127	3.86615	1.71474	10.5907	14.0625	-13.7109	213.75	7.03124	0	89.125	1.8	-0.48489	-0.48942	-4.32482	25.7127	
											10.5907	14.0625	-14.7656		7.55858						-4.32482	26.8063
											14.2383											
											14.2383											
92365				1528	183	-2.86654	-3.58224	2.75555	2.60708	10.5907	14.414	-15.4687	212.344	7.55858	15.875	89.25	2.1	-0.56571	-0.45452	-4.51633	28.6474	
											10.5907	14.414	-16.1719		7.55858						-4.27666	28.39
											14.5898											
											14.7656											
92366				1576	183.5	-3.28394	-5.07644	2.97811	1.71474	10.5907	14.9414	-16.1719	210.234	8.43749	0	89.125	2.2	-0.64651	-0.48942	-3.64084	25.989	
											10.5907	15.2929	-16.1719		8.96483						-2.57969	25.4345
											15.4687											
											15.4687											
92367				1624	183	-5.32946	-5.77184	3.7184	1.04419	10.5907	15.4687	-16.1719	208.477	8.26171	15.875	89.25	2.2	-0.80809	-0.45452	-2.78468	23.7257	
											10.5907	15.4687	-16.1719		7.91014						-2.73355	21.9487
											15.2929											
											14.9414											
92368	2	43	14	1668	182.5	-5.07643	-5.89811	4.60303	0.14921	10.5907	14.7656	-16.1719	207.07	7.3828	0	89.125	2.2	-0.48489	-0.45452	-2.68234	21.0349	
											10.5907	14.414	-16.1719		7.03124						-2.78468	21.0349
											14.2383											
											14.0625											
92369				1708	183	-5.07643	-5.89811	4.82345	0.298401	10.5907	13.8867	-16.875	205.312	6.85546	15.875	89.125	2.2	-0.64651	-0.45452	-2.73355	21.3414	
											10.5907	13.7109	-17.9297		6.67968						-2.78468	22.5486
											13.5351											
											13.3594											
92370				1748	183.5	-5.07643	-5.89811	4.16128	0.895081	10.5907	13.3594	-18.2812	203.906	6.67968	0	89.125	2.2	-0.56571	-0.45452	-2.78468	23.4343	
											10.5907	13.0078	-19.3359		6.67968						-2.78468	23.1409
											13.0078											
											12.832											
92371				1784	184.5	-5.07643	-5.83499	4.23498	0.745931	10.5907	12.6562	-19.6875	202.148	6.67968	15.875	89	2.2	-0.48489	-0.45452	-2.73355	22.8457	
											10.5907	12.4805	-20.039		6.67968						-2.73355	23.1409
											12.3047											
											12.1289											
92372	2	43	18	1816	185.5	-5.26621	-5.77184	4.23498	0.820516	10.5907	11.9531	-20.039	200.742	6.5039	0	89	2.2	-0.40409	-0.45452	-2.68234	23.1409	
											10.5907	11.7773	-20.3906		6.5039						-3.24116	23.7257
											11.4258											
											11.4258											
92373				1844	186.5	-4.3167	-4.82328	4.01377	1.5658	10.5907	11.0742	-20.7422	198.984	6.32812	15.875	89	2.2	-0.40409	-0.45452	-3.44176	24.8725	
											10.5907	10.8984	-20.7422		6.5039						-3.54149	25.7127
											10.7226											
											10.7226											



Time (seconds)	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETIC HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )	
											10.5469											
											10.3711											
92390				2224	210.5	-4.5067	-5.39269	2.38422	2.60708	10.5907	10.1953	-20.7422	166.992	6.67968	0	89.125	2.7	-0.64651	-0.45452	-3.14032	29.1565	
											10.5907	10.1953	-20.7422		7.03124						-3.08977	29.1565
											10.3711											
											10.5469											
92391				2252	212	-4.69666	-5.45591	2.45853	2.68131	10.5907	10.7226	-20.7422	164.883	7.55858	15.875	89.125	2.7	-0.48489	-0.45452	-3.08977	29.1565	
											10.5907	10.7226	-20.3906		7.91014						-2.98841	29.1565
											11.0742											
											11.0742											
92392	2	43	38	2284	213.5	-4.94987	-5.51912	2.45853	2.75555	10.5907	11.25	-20.3906	163.125	7.91014	0	89.125	2.7	-0.48489	-0.45452	-3.03913	29.6583	
											10.5907	11.25	-20.039		7.73436						-3.08977	29.9064
											11.25											
											11.25											
92393				2320	214.5	-4.69666	-5.45591	2.38422	2.82977	10.5907	11.25	-20.039	161.367	7.55858	15.875	89.125	2.7	-0.40409	-0.41961	-3.08977	29.9064	
											10.5907	11.4258	-19.6875		7.20702						-3.08977	29.9064
											11.4258											
											11.4258											
92394				2352	215.5	-4.69666	-5.51912	2.38422	3.64449	10.5907	11.4258	-19.3359	159.609	7.3828	0	89.125	2.7	-0.48489	-0.41961	-3.08977	31.1198	
											10.5907	11.6015	-18.6328		7.55858						-3.08977	32.9685
											11.6015											
											11.7773											
92395				2392	215.5	-4.69666	-5.45591	1.34232	3.7923	10.5907	11.7773	-18.6328	157.5	7.3828	15.875	89.25	2.7	-0.56571	-0.41961	-3.14032	32.9685	
											10.5907	11.9531	-17.9297		7.3828						-3.14032	33.1917
											11.9531											
											12.1289											
92396	2	43	42	2432	216	-4.63335	-5.51912	0.745944	6.25067	10.5907	12.3047	-17.5781	155.742	7.73436	0	89.125	2.7	-0.72731	-0.41961	-3.08977	39.4382	
											10.5907	12.3047	-17.5781		7.55858						-3.14032	37.1213
											12.4805											
											12.4805											
92397				2472	216.5	-4.5067	-5.13971	0	4.97021	10.5907	12.6562	-17.2265	154.336	7.3828	15.875	89.125	2.7	-0.88889	-0.41961	-3.29145	35.9425	
											10.5907	12.832	-16.1719		7.20702						-3.39175	34.4958
											12.832											
											12.832											
92398				2520	216.5	-4.25337	-5.51912	1.86362	2.82977	10.5907	13.1836	-15.1172	152.93	7.3828	0	89.125	2.7	-0.56571	-0.41961	-3.14032	29.1565	
											10.5907	13.3594	-14.414		7.91014						-2.98841	30.3972
											13.5351											
											13.7109											
92399				2572	217	-4.75998	-5.83499	1.93804	3.20046	10.5907	13.8867	-14.414	151.523	8.08593	15.875	89.125	2.7	-0.88889	-0.45452	-2.88671	31.1198	
											10.5907	14.0625	-14.414		7.73436						-2.63105	29.4084
											14.0625											
											14.0625											
92400	2	43	46	2624	216.5	-5.51912	-6.59153	2.45853	2.75555	10.5907	14.2383	-14.414	150.469	7.55858	0	89	2.7	-0.64651	-0.45452	-2.52825	29.4084	
											10.5907	14.2383	-14.414		7.03124						-2.42516	29.4084
											14.2383											
											14.0625											
92401				2676	216.5	-5.70868	-6.27659	2.3099	2.68131	10.5907	14.0625	-14.414	149.766	6.5039	15.875	89	2.7	-0.48489	-0.45452	-2.52825	29.4084	
											10.5907	13.8867	-14.0625		6.5039						-2.37351	29.4084
											13.8867											
											13.8867											
92402				2728	216	-6.02434	-6.65446	2.3099	2.68131	10.5907	13.8867	-13.7109	148.711	6.15233	0	89.125	2.7	-0.32326	-0.45452	-2.32178	29.4084	
											10.5907	13.8867	-13.3594		6.32812						-2.37351	29.6583
											13.8867											
											13.8867											
92403				2784	216.5	-5.70868	-6.4656	2.38422	2.90393	10.5907	13.8867	-13.0078	147.656	6.32812	15.875	89.125	2.6	-0.48489	-0.45452	-2.63105	29.6583	
											10.5907	13.8867	-13.0078		6.32812						-2.32178	33.4133
											13.8867											
											13.8867											
92404	2	43	50	2840	217	-5.70868	-6.21355	0.596783	5.47019	10.5907	14.0625	-13.0078	146.602	6.5039	0	89.125	2.6	-0.56571	-0.41961	-2.57969	36.5388	
											10.5907	14.0625	-13.0078		6.67968						-2.27	38.6462
											14.0625											
											14.0625											











Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	ELEVATOR POSN L ( )	ELEVATOR POSN R ( )	AILERON POSN L ( )	AILERON POSN R ( )	SPD BRAKE HANDLE ( )	PITCH ANGLE EFIS (DEG)	ROLL ANGLE EFIS (DEG)	MAGNETI HEADING EFIS (DEG)	AOA (DEG)	N1 L (%RPM)	N1 R (%RPM)	PITCH TRIM POSITION ( )	RUDDER POSN ( )	RUDDER PEDAL POSN ( )	CONTROL COLUMN POSN ( )	CONTROL WHEEL POSN ( )	
92467				4628	254	-3.94032	-5.45591	0	3.05225	10.5907	-25.1367	91.4061	265.078	5.44921	15.875	89.625	2.5	2.09954	0.349758	-3.64084	35.5372	
										10.5907	-26.0156	92.8124		4.57031						-3.19079	25.7127	
											-27.0703											
											-28.8281											
92468	2	44	54	4388	264.5	-4	-6.08744	2.53281	7.16603	10.5907	-29.707	95.2733	270	3.86718	0	89.875	2.5	1.85763	0.419615	-3.14032	36.1428	
										10.5907	-30.2343	96.6796		3.33984						-1.90571	41.7476	
											-31.1132											
											-31.8164											
92469				4124	275.5	-6.1505	-7.02937	-2.23555	6.46262	10.5907	-33.0468	98.0858	273.516	2.98828	15.875	90	2.5	2.26073	0.66366	-1.53866	41.7476	
										10.5907	-33.9257	99.8436		2.10937						-1.38063	33.1917	
											-34.8046											
											-36.5624											
92470				3820	289.5	-6.33962	-9.11425	2.01244	2.68131	10.5907	-36.914	103.008	277.031	1.23047	0	89.875	2.5	1.93828	-0.20992	-0.90449	29.6583	
										10.5907	-37.7929	105.469		0.703124						2.32179	27.6066	
											-39.5507											
											-40.2538											
92471				3508	306.5	-9.8567	-8.30798	2.23556	-9.92503	10.5907	-41.3085	107.578	279.844	0	15.875	89.875	2.5	1.53489	-0.24489	0.957521	11.7557	
										7.45171	-41.6601	110.039		-2.63671						-1.69627	2.621	
											-42.0117											
											-43.0663											
92472	2	44	58	3068	317.5	-5.45591	-6.21355	19.9852	-12.609	8.50137	-43.2421	111.094	281.602	-2.28515	0	89.625	2.5	1.21197	-0.20992	-1.48603	3.74078	
										9.54769	-43.9452	98.0858		0.527343						-1.59124	31.357	
											-45.1757											
											-45.5273											
92473				2640	334	-5.07643	-5.77184	16.2187	-5.61246	10.5907	-45.7031	78.7499	290.391	2.98828	15.875	89.125	2.5	1.77697	1.11405	-2.47674	38.6462	
										10.5907	-45.8788	60.4687		3.16406						-1.95793	57.5102	
											-45.8788											
											-45.8788											
92474				2216	352	-5.07643	-5.32945	8.6952	-7.65547	10.5907	-45.7031	54.1405	298.477	2.8125	0	87.5	2.5	2.3413	1.01051	-2.01009	54.3458	
										9.54769	-45.3515	49.5702		2.28515						-2.98841	14.8754	
											-44.9999											
											-44.6484											
92475				1748	368.5	-4.44337	-5.45591	18.83	-9.1074	9.54769	-44.121	48.164	302.695	2.28515	15.875	77.125	2.6	2.98518	1.01051	-2.27	12.8083	
										10.5907	-43.4179	37.9687		3.33984						-3.59122	65.7645	
											-42.7148											
											-41.4843											
92476	2	45	2	1320	382.5	-3.34359	-4.75997	5.39898	-4.23498	10.5907	-40.6054	30.2343	306.914	3.51562	0	63.375	2.4	1.37345	1.07957	-2.98841	59.1578	
										10.5907	-39.0234	22.8515		3.33984						-3.19079	28.9029	
											-38.3203											
											-37.9687											
92477				904	395	-2.80693	-3.64192	14.1822	-4.01377	10.5907	-36.914	23.9062	309.023	2.63671	15.875	55.75	2.4	2.18014	1.86262	-3.98541	31.1198	
										10.5907	-36.2109	18.2812		3.51562						-6.4094	63.6251	
											-35.332											
											-33.75											
92478				524	410	0.999374	2.15226	1.64028	3.42259	10.5907	-32.6953	14.0625	311.133	3.51562	0	51.5	2.2	3.46716	3.33793	-8.81175	57.1858	
										10.5907	-30.5859	14.414		4.92187						-7.88453	41.0981	
											-29.8828											
											-29.0039											
92479				180	416	-0.67098	-3.28393	6.95553	0.14921	10.5907	-25.4882	19.3359	315.703	6.85546	15.875	48.375	2.4	3.3066	4.48769	-5.89152	41.5294	
										10.5907	-24.4336	24.6093		5.44921						-5.35487	40.0486	
											-23.7304											
											-23.2031											
92480																						



Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	MASTER CAUTION	EFIS SEL SW CAPT	V/S MODE FCC	A/T ENGA	A/T GA	A/T LIMIT	A/T MAN DISC	A/T MCP SPEED	A/T MIN SPEED	A/T N1	A/T RETARD	A/T WARN	N1 LIMIT MODE A/T	THR LEVER ANGLE L (DEG)	THR LEVER ANGLE R (DEG)
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-WARN 1-)	(0-LEFT 1-RIGHT)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-LIMIT)	(0-DISC 1-)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0-WARN 1-)	(0-NOCODE 1-CODED)		
91978				216	45															1.23047
91979				216	45		LEFT													2.46093
91980	2	36	48	216	45													T/O		1.23047
91981				216	45															2.46093





Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	MASTER CAUTION	EFIS SEL SW CAPT	V/S MODE FCC	A/T ENGA	A/T GA	A/T LIMIT	A/T MAN DISC	A/T MCP SPEED	A/T MIN SPEED	A/T N1	A/T RETARD	A/T WARN	N1 LIMIT MODE A/T	THR LEVER ANGLE L (DEG)	THR LEVER ANGLE R (DEG)
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-WARN 1-)	(0-LEFT 1-RIGHT)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-LIMIT)	(0-DISC 1-)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0-WARN 1-)	(0-NOCODE 1-CODED)		
92102				208	45															5.62499
92103				204	45		LEFT													7.03124
92104	2	38	50	204	45													T/O		5.62499
92105				204	45															7.03124



Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	MASTER CAUTION	EFIS SEL SW CAPT	V/S MODE FCC	A/T ENGA	A/T GA	A/T LIMIT	A/T MAN DISC	A/T MCP SPEED	A/T MIN SPEED	A/T N1	A/T RETARD	A/T WARN	N1 LIMIT MODE A/T	THR LEVER ANGLE L (DEG)	THR LEVER ANGLE R (DEG)
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-WARN 1-)	(0-LEFT 1-RIGHT)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-LIMIT)	(0-DISC 1-)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0-WARN 1-)	(0-NOCODE 1-CODED)		
92226				184	45				ENGA											1.23047
92227				184	45		LEFT		ENGA											2.8125
92228	2	40	54	184	45				ENGA										T/O	1.23047
92229				184	45				ENGA											2.8125



Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	MASTER CAUTION	EFIS SEL SW CAPT	V/S MODE FCC	A/T ENGA	A/T GA	A/T LIMIT	A/T MAN DISC	A/T MCP SPEED	A/T MIN SPEED	A/T N1	A/T RETARD	A/T WARN	N1 LIMIT MODE A/T	THR LEVER ANGLE L (DEG)	THR LEVER ANGLE R (DEG)
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-WARN 1-)	(0-LEFT 1-RIGHT)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-LIMIT)	(0-DISC 1-)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0-WARN 1-)	(0-NOCODE 1-CODED)		
92350				904	180.5				ENGA											45.7031
92351				940	181.5		LEFT		ENGA											46.2304
92352	2	42	58	976	181				ENGA										T/O	45.7031
92353				1016	181.5				ENGA											46.2304



Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	MASTER CAUTION	EFIS SEL SW CAPT	V/S MODE FCC	A/T ENGA	A/T GA	A/T LIMIT	A/T MAN DISC	A/T MCP SPEED	A/T MIN SPEED	A/T N1	A/T RETARD	A/T WARN	N1 LIMIT MODE A/T	THR LEVER ANGLE L (DEG)	THR LEVER ANGLE R (DEG)
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-WARN 1-)	(0-LEFT 1-RIGHT)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-LIMIT)	(0-DISC 1-)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0-WARN 1-)	(0-NOCODE 1-CODED)		
92474				2216	352				ENGA						ENGA					43.9452
92475				1748	363.5		LEFT		ENGA						ENGA					31.289
92476	2	45	2	1320	382.5				ENGA						ENGA			CLB		19.3359
92477				904	395				ENGA						ENGA					2.28515

Time	GMT HOURS	GMT MINUTES	GMT SECONDS	ALTITUDE (29 92)	COMPUTED AIRSPD	MASTER CAUTION	EFIS SEL SW CAPT	V/S MODE FCC	A/T ENGA	A/T GA	A/T LIMIT	A/T MAN DISC	A/T MCP SPEED	A/T MIN SPEED	A/T N1	A/T RETARD	A/T WARN	N1 LIMIT MODE A/T	THR LEVER ANGLE L (DEG)	THR LEVER ANGLE R (DEG)
(seconds)	(HOURS)	(MINUTES)	(SECONDS)	(FEET)	(KNOTS)	(0-WARN 1-)	(0-LEFT 1-RIGHT)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-LIMIT)	(0-DISC 1-)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0- 1-ENGA)	(0-WARN 1-)	(0-NOCODE 1-CODED)		
92478				524	410				ENGA						ENGA					2.98828
92479				180	416		LEFT		ENGA						ENGA					5.27343
92480																				

















# Flash /  
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Time	TRIM UP A/P
(seconds)	(1-TRIM)
91864	
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Time	TRIM UP A/P
(seconds)	(1-TRIM)
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Time	TRIM UP A/P
(seconds)	(1-TRIM)
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Time	TRIM UP A/P
(seconds)	(1-TRIM)
92143	
92144	
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Time	TRIM UP A/P
(seconds)	(1-TRIM)
92238	
92239	
92240	
92241	
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92243	
92244	
92245	
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Time	TRIM UP A/P
(seconds)	(1-TRIM)
92333	
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Time	TRIM UP A/P
(seconds)	(1-TRIM)
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# Flash Air B737-300 Accident  
 # Preliminary Data Created: January 20 2004  
 # MCA

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	CN1 TRACKED VIB L (SCALAR)	CN1 TRACKED VIB R (SCALAR)	CN2 TRACKED VIB L (SCALAR)	CN2 TRACKED VIB R (SCALAR)	TN1 TRACKED VIB L (SCALAR)	TN1 TRACKED VIB R (SCALAR)	FAN IMB ANGLE L (DEG)	FAN IMB ANGLE R (DEG)	LPT IMB ANGLE L (DEG)	LPT IMB ANGLE R (DEG)
91864	2	34	50	216	45										
91865				216	45										
91866				216	45										
91867				216	45										
91868	2	34	54	216	45			0							
91869				216	45										
91870				216	45										
91871				216	45										
91872	2	34	58	216	45					0					
91873				216	45										
91874				216	45										
91875				216	45										
91876	2	35	2	216	45										
91877				216	45										
91878				216	45										
91879				216	45										
91880	2	35	6	216	45		0.26								
91881				216	45										
91882				216	45										
91883				216	45										
91884	2	35	10	216	45				0.44						
91885				216	45										
91886				216	45										
91887				216	45										
91888	2	35	14	216	45						0.12				
91889				216	45										
91890				216	45										
91891				216	45										
91892	2	35	18	216	45										
91893				216	45										
91894				216	45										
91895				216	45										
91896	2	35	22	216	45							0			
91897				216	45										
91898				216	45										
91899				216	45										
91900	2	35	26	216	45									0	
91901				216	45										
91902				216	45										
91903				216	45										
91904	2	35	30	216	45								0		

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	CN1 TRACKED VIB L (SCALAR)	CN1 TRACKED VIB R (SCALAR)	CN2 TRACKED VIB L (SCALAR)	CN2 TRACKED VIB R (SCALAR)	TN1 TRACKED VIB L (SCALAR)	TN1 TRACKED VIB R (SCALAR)	FAN IMB ANGLE L (DEG)	FAN IMB ANGLE R (DEG)	LPT IMB ANGLE L (DEG)	LPT IMB ANGLE R (DEG)
91905				216	45										
91906				216	45										
91907				216	45										
91908	2	35	34	216	45										2
91909				216	45										
91910				216	45										
91911				216	45										
91912	2	35	38	216	45										
91913				216	45										
91914				216	45										
91915				216	45										
91916	2	35	42	216	45										
91917				216	45										
91918				216	45										
91919				216	45										
91920	2	35	46	216	45										
91921				216	45										
91922				216	45										
91923				216	45										
91924	2	35	50	216	45										
91925				216	45										
91926				216	45										
91927				216	45										
91928	2	35	54	216	45	0.36									
91929				216	45										
91930				216	45										
91931				216	45										
91932	2	35	58	216	45			3.2							
91933				216	45										
91934				216	45										
91935				216	45										
91936	2	36	2	216	45					0.74					
91937				216	45										
91938				216	45										
91939				216	45										
91940	2	36	6	216	45										
91941				216	45										
91942				216	45										
91943				216	45										
91944	2	36	10	216	45		0.3								
91945				216	45										
91946				216	45										
91947				216	45										
91948	2	36	14	216	45				0.22						
91949				216	45										
91950				216	45										
91951				216	45										



Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	CN1 TRACKED VIB L (SCALAR)	CN1 TRACKED VIB R (SCALAR)	CN2 TRACKED VIB L (SCALAR)	CN2 TRACKED VIB R (SCALAR)	TN1 TRACKED VIB L (SCALAR)	TN1 TRACKED VIB R (SCALAR)	FAN IMB ANGLE L (DEG)	FAN IMB ANGLE R (DEG)	LPT IMB ANGLE L (DEG)	LPT IMB ANGLE R (DEG)
91952	2	36	18	216	45						0.08				
91953				216	45										
91954				216	45										
91955				216	45										
91956	2	36	22	216	45										
91957				216	45										
91958				216	45										
91959				216	45										
91960	2	36	26	216	45							0			
91961				216	45										
91962				216	45										
91963				216	45										
91964	2	36	30	216	45									0	
91965				216	45										
91966				216	45										
91967				216	45										
91968	2	36	34	216	45								0		
91969				216	45										
91970				216	45										
91971				216	45										
91972	2	36	38	216	45										2
91973				216	45										
91974				216	45										
91975				216	45										
91976	2	36	42	216	45										
91977				216	45										
91978				216	45										
91979				216	45										
91980	2	36	46	216	45										
91981				216	45										
91982				216	45										
91983				216	45										
91984	2	36	50	216	45										
91985				216	45										
91986				216	45										
91987				216	45										
91988	2	36	54	216	45										
91989				216	45										
91990				216	45										
91991				216	45										
91992	2	36	58	216	45	0.06									
91993				216	45										
91994				216	45										
91995				216	45										
91996	2	37	2	216	45			0.3							
91997				216	45										
91998				216	45										

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	CN1 TRACKED VIB L (SCALAR)	CN1 TRACKED VIB R (SCALAR)	CN2 TRACKED VIB L (SCALAR)	CN2 TRACKED VIB R (SCALAR)	TN1 TRACKED VIB L (SCALAR)	TN1 TRACKED VIB R (SCALAR)	FAN IMB ANGLE L (DEG)	FAN IMB ANGLE R (DEG)	LPT IMB ANGLE L (DEG)	LPT IMB ANGLE R (DEG)
91999				216	45										
92000	2	37	6	216	45					0.1					
92001				216	45										
92002				216	45										
92003				216	45										
92004	2	37	10	216	45										
92005				216	45										
92006				216	45										
92007				216	45										
92008	2	37	14	216	45		0.32								
92009				216	45										
92010				216	45										
92011				216	45										
92012	2	37	18	216	45				0.38						
92013				216	45										
92014				216	45										
92015				216	45										
92016	2	37	22	216	45						0.1				
92017				216	45										
92018				216	45										
92019				216	45										
92020	2	37	26	216	45										
92021				216	45										
92022				216	45										
92023				216	45										
92024	2	37	30	216	45							100			
92025				216	45										
92026				216	45										
92027				216	45										
92028	2	37	34	216	45									0	
92029				216	45										
92030				216	45										
92031				216	45										
92032	2	37	38	216	45								0		
92033				216	45										
92034				216	45										
92035				216	45										
92036	2	37	42	216	45										2
92037				216	45										
92038				216	45										
92039				216	45										
92040	2	37	46	216	45										
92041				216	45										
92042				216	45										
92043				216	45										
92044	2	37	50	216	45										
92045				216	45										

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	CN1 TRACKED VIB L (SCALAR)	CN1 TRACKED VIB R (SCALAR)	CN2 TRACKED VIB L (SCALAR)	CN2 TRACKED VIB R (SCALAR)	TN1 TRACKED VIB L (SCALAR)	TN1 TRACKED VIB R (SCALAR)	FAN IMB ANGLE L (DEG)	FAN IMB ANGLE R (DEG)	LPT IMB ANGLE L (DEG)	LPT IMB ANGLE R (DEG)
92046				216	45										
92047				216	45										
92048	2	37	54	216	45										
92049				216	45										
92050				216	45										
92051				216	45										
92052	2	37	58	216	45										
92053				216	45										
92054				216	45										
92055				216	45										
92056	2	38	2	216	45	0.04									
92057				216	45										
92058				216	45										
92059				216	45										
92060	2	38	6	216	45			0.12							
92061				216	45										
92062				212	45										
92063				216	45										
92064	2	38	10	212	45					0.04					
92065				212	45										
92066				212	45										
92067				212	45										
92068	2	38	14	212	45										
92069				212	45										
92070				212	45										
92071				212	45										
92072	2	38	18	212	45		0.38								
92073				212	45										
92074				212	45										
92075				212	45										
92076	2	38	22	212	45				0.24						
92077				208	45										
92078				208	45										
92079				208	45										
92080	2	38	26	208	45						0.14				
92081				208	45										
92082				208	45										
92083				208	45										
92084	2	38	30	208	45										
92085				208	45										
92086				208	45										
92087				208	45										
92088	2	38	34	208	45							0			
92089				208	45										
92090				208	45										
92091				208	45										
92092	2	38	38	208	45									0	

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	CN1 TRACKED VIB L (SCALAR)	CN1 TRACKED VIB R (SCALAR)	CN2 TRACKED VIB L (SCALAR)	CN2 TRACKED VIB R (SCALAR)	TN1 TRACKED VIB L (SCALAR)	TN1 TRACKED VIB R (SCALAR)	FAN IMB ANGLE L (DEG)	FAN IMB ANGLE R (DEG)	LPT IMB ANGLE L (DEG)	LPT IMB ANGLE R (DEG)
92093				208	45										
92094				208	45										
92095				208	45										
92096	2	38	42	208	45								0		
92097				208	45										
92098				208	45										
92099				208	45										
92100	2	38	46	208	45										2
92101				208	45										
92102				208	45										
92103				204	45										
92104	2	38	50	204	45										
92105				204	45										
92106				204	45										
92107				204	45										
92108	2	38	54	204	45										
92109				204	45										
92110				208	45										
92111				204	45										
92112	2	38	58	204	45										
92113				204	45										
92114				204	45										
92115				204	45										
92116	2	39	2	204	45										
92117				204	45										
92118				204	45										
92119				204	45										
92120	2	39	6	204	45	0.06									
92121				204	45										
92122				204	45										
92123				204	45										
92124	2	39	10	204	45			0.12							
92125				204	45										
92126				204	45										
92127				204	45										
92128	2	39	14	204	45					0.06					
92129				204	45										
92130				200	45										
92131				204	45										
92132	2	39	18	200	45										
92133				200	45										
92134				200	45										
92135				200	45										
92136	2	39	22	200	45		0.32								
92137				200	45										
92138				200	45										
92139				200	45										

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	CN1 TRACKED VIB L (SCALAR)	CN1 TRACKED VIB R (SCALAR)	CN2 TRACKED VIB L (SCALAR)	CN2 TRACKED VIB R (SCALAR)	TN1 TRACKED VIB L (SCALAR)	TN1 TRACKED VIB R (SCALAR)	FAN IMB ANGLE L (DEG)	FAN IMB ANGLE R (DEG)	LPT IMB ANGLE L (DEG)	LPT IMB ANGLE R (DEG)
92140	2	39	26	200	45				0.16						
92141				200	45										
92142				200	45										
92143				196	45										
92144	2	39	30	196	45						0.1				
92145				196	45										
92146				196	45										
92147				196	45										
92148	2	39	34	196	45										
92149				196	45										
92150				196	45										
92151				196	45										
92152	2	39	38	196	45							0			
92153				196	45										
92154				196	45										
92155				192	45										
92156	2	39	42	192	45									0	
92157				196	45										
92158				192	45										
92159				192	45										
92160	2	39	46	192	45								0		
92161				192	45										
92162				192	45										
92163				192	45										
92164	2	39	50	192	45										2
92165				192	45										
92166				192	45										
92167				192	45										
92168	2	39	54	192	45										
92169				192	45										
92170				192	45										
92171				192	45										
92172	2	39	58	192	45										
92173				192	45										
92174				192	45										
92175				192	45										
92176	2	40	2	192	45										
92177				192	45										
92178				192	45										
92179				192	45										
92180	2	40	6	192	45										
92181				188	45										
92182				192	45										
92183				192	45										
92184	2	40	10	192	45	0.04									
92185				188	45										
92186				192	45										

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	CN1 TRACKED VIB L (SCALAR)	CN1 TRACKED VIB R (SCALAR)	CN2 TRACKED VIB L (SCALAR)	CN2 TRACKED VIB R (SCALAR)	TN1 TRACKED VIB L (SCALAR)	TN1 TRACKED VIB R (SCALAR)	FAN IMB ANGLE L (DEG)	FAN IMB ANGLE R (DEG)	LPT IMB ANGLE L (DEG)	LPT IMB ANGLE R (DEG)
92187				192	45										
92188	2	40	14	188	45			0.14							
92189				192	45										
92190				188	45										
92191				188	45										
92192	2	40	18	188	45					0.1					
92193				188	45										
92194				188	45										
92195				188	45										
92196	2	40	22	188	45										
92197				188	45										
92198				188	45										
92199				188	45										
92200	2	40	26	188	45		0.24								
92201				188	45										
92202				188	45										
92203				188	45										
92204	2	40	30	188	45				0.28						
92205				188	45										
92206				188	45										
92207				188	45										
92208	2	40	34	188	45						0.1				
92209				188	45										
92210				188	45										
92211				188	45										
92212	2	40	38	188	45										
92213				188	45										
92214				188	45										
92215				184	45										
92216	2	40	42	188	45							0			
92217				188	45										
92218				188	45										
92219				188	45										
92220	2	40	46	188	45									0	
92221				188	45										
92222				184	45										
92223				188	45										
92224	2	40	50	188	45								0		
92225				184	45										
92226				184	45										
92227				184	45										
92228	2	40	54	184	45										0
92229				184	45										
92230				184	45										
92231				184	45										
92232	2	40	58	184	45										
92233				184	45										

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	CN1 TRACKED VIB L (SCALAR)	CN1 TRACKED VIB R (SCALAR)	CN2 TRACKED VIB L (SCALAR)	CN2 TRACKED VIB R (SCALAR)	TN1 TRACKED VIB L (SCALAR)	TN1 TRACKED VIB R (SCALAR)	FAN IMB ANGLE L (DEG)	FAN IMB ANGLE R (DEG)	LPT IMB ANGLE L (DEG)	LPT IMB ANGLE R (DEG)
92234				184	45										
92235				184	45										
92236	2	41	2	184	45										
92237				184	45										
92238				184	45										
92239				184	45										
92240	2	41	6	184	45										
92241				184	45										
92242				184	45										
92243				184	45										
92244	2	41	10	184	45										
92245				184	45										
92246				184	45										
92247				184	45										
92248	2	41	14	184	45	0.04									
92249				184	45										
92250				184	45										
92251				184	45										
92252	2	41	18	184	45			0.12							
92253				184	45										
92254				184	45										
92255				184	45										
92256	2	41	22	184	45					0.1					
92257				184	45										
92258				184	45										
92259				180	45										
92260	2	41	26	180	45										
92261				180	45										
92262				180	45										
92263				180	45										
92264	2	41	30	180	45		0.24								
92265				180	45										
92266				180	45										
92267				180	45										
92268	2	41	34	180	45			0.16							
92269				180	45										
92270				180	45										
92271				180	45										
92272	2	41	38	180	45						0.1				
92273				180	45										
92274				180	45										
92275				180	45										
92276	2	41	42	180	45										
92277				180	45										
92278				180	45										
92279				180	45										
92280	2	41	46	180	45							0			

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	CN1 TRACKED VIB L (SCALAR)	CN1 TRACKED VIB R (SCALAR)	CN2 TRACKED VIB L (SCALAR)	CN2 TRACKED VIB R (SCALAR)	TN1 TRACKED VIB L (SCALAR)	TN1 TRACKED VIB R (SCALAR)	FAN IMB ANGLE L (DEG)	FAN IMB ANGLE R (DEG)	LPT IMB ANGLE L (DEG)	LPT IMB ANGLE R (DEG)
92281				180	45										
92282				180	45										
92283				180	45										
92284	2	41	50	180	45									0	
92285				180	45										
92286				180	45										
92287				180	45										
92288	2	41	54	180	45								0		
92289				180	45										
92290				180	45										
92291				184	45										
92292	2	41	58	180	45										0
92293				184	45										
92294				184	45										
92295				184	45										
92296	2	42	2	188	45										
92297				188	45										
92298				188	45										
92299				188	45										
92300	2	42	6	192	45										
92301				192	45.5										
92302				192	49.5										
92303				196	56										
92304	2	42	10	196	61										
92305				196	65										
92306				196	70										
92307				200	75.5										
92308	2	42	14	200	78.5										
92309				200	83.5										
92310				200	89										
92311				200	93										
92312	2	42	18	200	97.5	0.18									
92313				204	101										
92314				204	106.5										
92315				204	109.5										
92316	2	42	22	204	115.5			1.16							
92317				204	119.5										
92318				204	123.5										
92319				208	127.5										
92320	2	42	26	208	131.5					0.42					
92321				208	135.5										
92322				208	139										
92323				204	142.5										
92324	2	42	30	204	146										
92325				196	150										
92326				192	152										
92327				192	155.5										



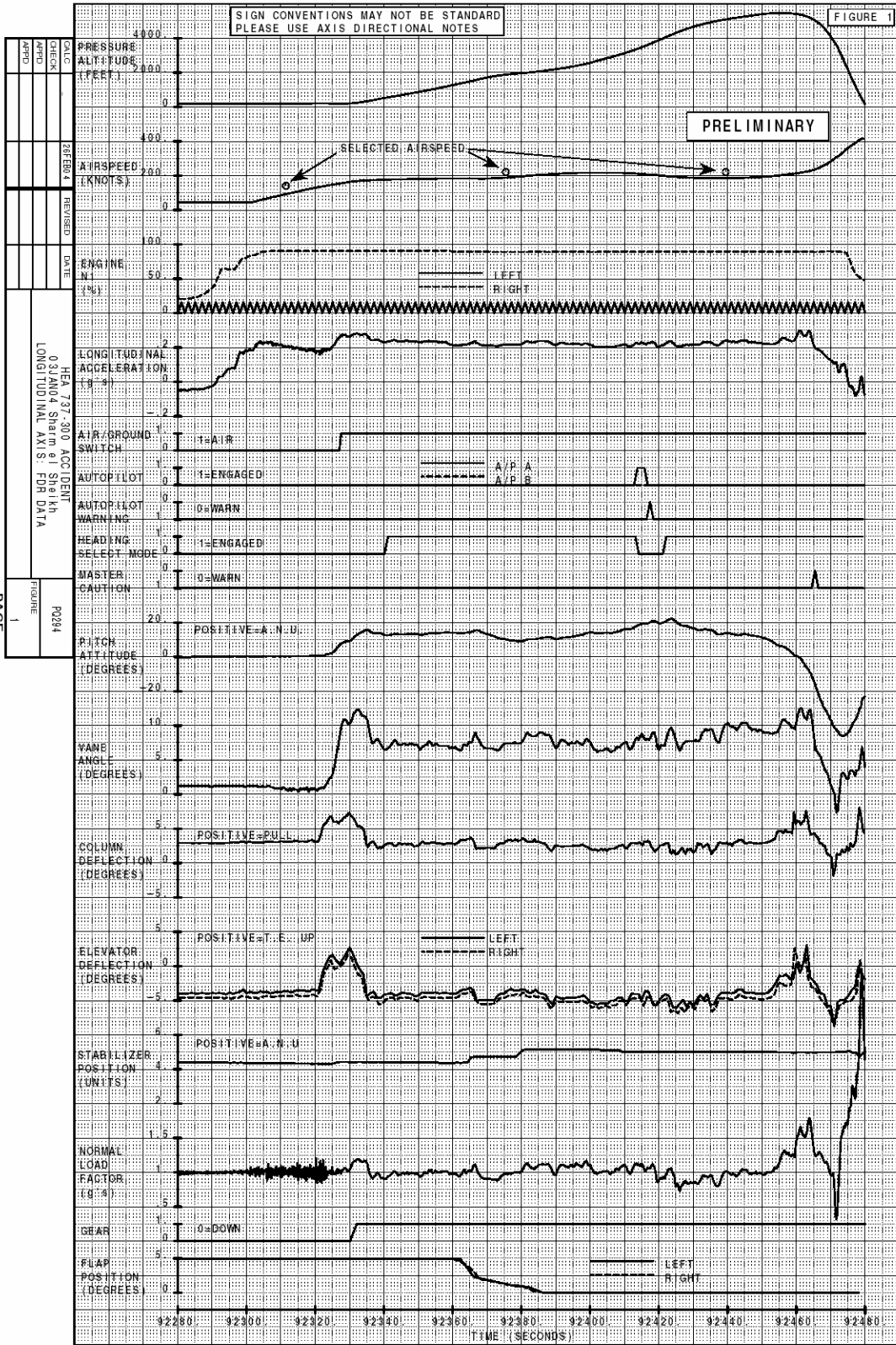
Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	CN1 TRACKED VIB L (SCALAR)	CN1 TRACKED VIB R (SCALAR)	CN2 TRACKED VIB L (SCALAR)	CN2 TRACKED VIB R (SCALAR)	TN1 TRACKED VIB L (SCALAR)	TN1 TRACKED VIB R (SCALAR)	FAN IMB ANGLE L (DEG)	FAN IMB ANGLE R (DEG)	LPT IMB ANGLE L (DEG)	LPT IMB ANGLE R (DEG)
92328	2	42	34	196	159		0.44								
92329				208	162										
92330				220	165.5										
92331				240	167.5										
92332	2	42	38	268	169.5				0.96						
92333				300	171.5										
92334				328	172										
92335				364	173										
92336	2	42	42	400	174						0.68				
92337				440	174.5										
92338				480	176										
92339				512	176.5										
92340	2	42	46	548	177										
92341				584	178										
92342				616	178.5										
92343				652	179										
92344	2	42	50	688	178.5							84			
92345				720	179.5										
92346				756	179.5										
92347				792	180										
92348	2	42	54	832	180									318	
92349				868	181										
92350				904	180.5										
92351				940	181.5										
92352	2	42	58	976	181								266		
92353				1016	181.5										
92354				1052	181.5										
92355				1096	183										
92356	2	43	2	1136	183										4
92357				1180	184										
92358				1220	184										
92359				1268	184										
92360	2	43	6	1312	184										
92361				1352	183										
92362				1396	184										
92363				1440	184										
92364	2	43	10	1484	183.5										
92365				1528	183										
92366				1576	183.5										
92367				1624	183										
92368	2	43	14	1668	182.5										
92369				1708	183										
92370				1748	183.5										
92371				1784	184.5										
92372	2	43	18	1816	185.5										
92373				1844	186.5										
92374				1868	187.5										

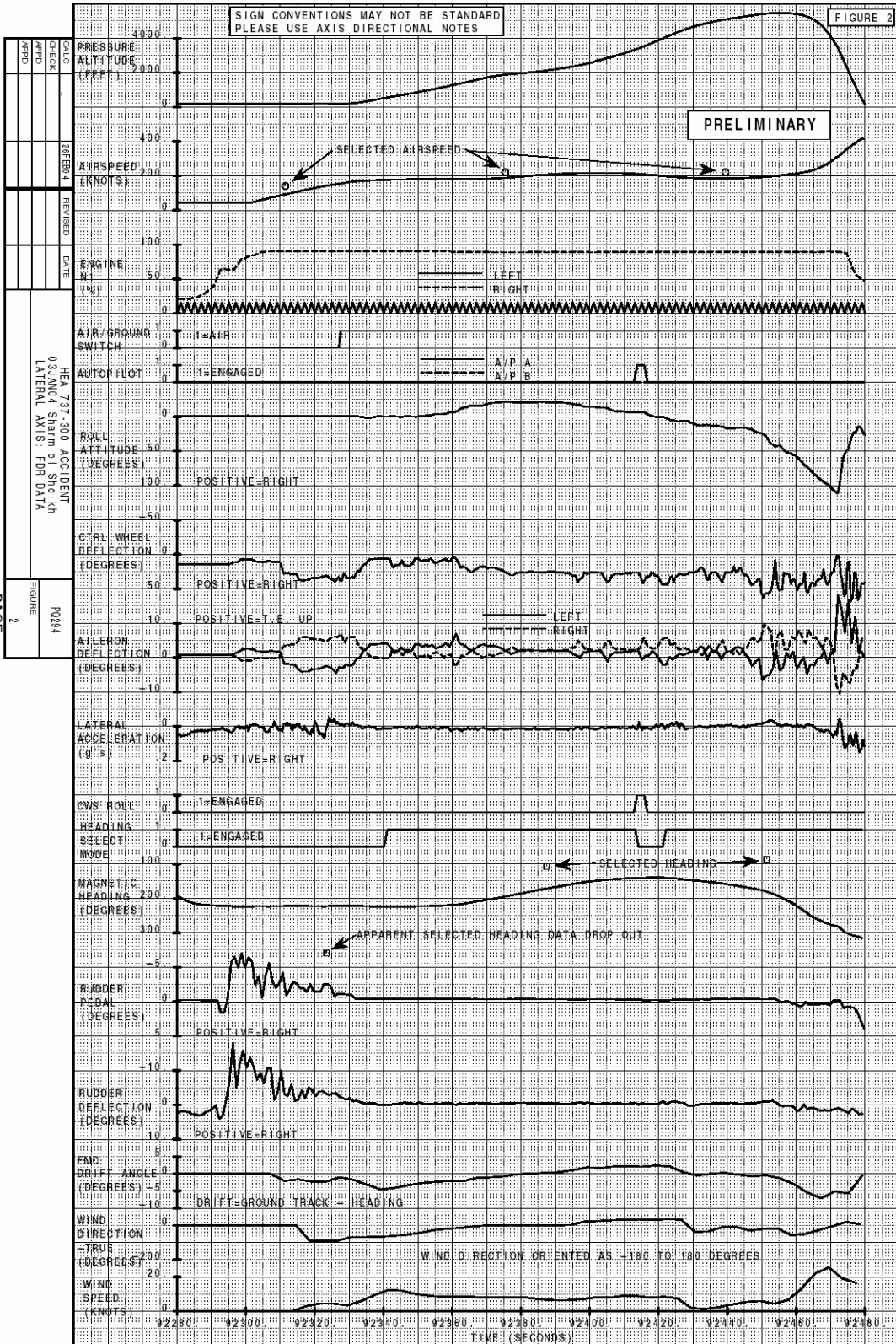
Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	CN1 TRACKED VIB L (SCALAR)	CN1 TRACKED VIB R (SCALAR)	CN2 TRACKED VIB L (SCALAR)	CN2 TRACKED VIB R (SCALAR)	TN1 TRACKED VIB L (SCALAR)	TN1 TRACKED VIB R (SCALAR)	FAN IMB ANGLE L (DEG)	FAN IMB ANGLE R (DEG)	LPT IMB ANGLE L (DEG)	LPT IMB ANGLE R (DEG)
92375				1892	188.5										
92376	2	43	22	1912	190	0.18									
92377				1932	191.5										
92378				1948	193										
92379				1964	194.5										
92380	2	43	26	1980	196.5			0.24							
92381				2000	198.5										
92382				2020	200.5										
92383				2040	202										
92384	2	43	30	2064	203.5					0.5					
92385				2084	205										
92386				2112	206										
92387				2136	207.5										
92388	2	43	34	2168	208.5										
92389				2196	209										
92390				2224	210.5										
92391				2252	212										
92392	2	43	38	2284	213.5		0.64								
92393				2320	214.5										
92394				2352	215.5										
92395				2392	215.5										
92396	2	43	42	2432	216				1						
92397				2472	216.5										
92398				2520	216.5										
92399				2572	217										
92400	2	43	46	2624	216.5						0.58				
92401				2676	216.5										
92402				2728	216										
92403				2784	216.5										
92404	2	43	50	2840	217										
92405				2892	217										
92406				2948	216.5										
92407				3004	216.5										
92408	2	43	54	3064	216							42			
92409				3124	216										
92410				3188	214.5										
92411				3252	214										
92412	2	43	58	3320	213.5									306	
92413				3392	212										
92414				3468	209.5										
92415				3544	209.5										
92416	2	44	2	3624	207								274		
92417				3712	206										
92418				3796	204.5										
92419				3880	203										
92420	2	44	6	3964	201										10
92421				4056	199										

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	CN1 TRACKED VIB L (SCALAR)	CN1 TRACKED VIB R (SCALAR)	CN2 TRACKED VIB L (SCALAR)	CN2 TRACKED VIB R (SCALAR)	TN1 TRACKED VIB L (SCALAR)	TN1 TRACKED VIB R (SCALAR)	FAN IMB ANGLE L (DEG)	FAN IMB ANGLE R (DEG)	LPT IMB ANGLE L (DEG)	LPT IMB ANGLE R (DEG)
92422				4136	196.5										
92423				4220	194.5										
92424	2	44	10	4308	195										
92425				4388	192										
92426				4460	190										
92427				4532	190										
92428	2	44	14	4600	188.5										
92429				4660	188										
92430				4720	187.5										
92431				4772	187										
92432	2	44	18	4824	186.5										
92433				4876	186										
92434				4920	185.5										
92435				4968	185.5										
92436	2	44	22	5008	185										
92437				5044	184.5										
92438				5076	185.5										
92439				5112	186										
92440	2	44	26	5144	186.5	0.24									
92441				5172	186										
92442				5204	186.5										
92443				5232	187										
92444	2	44	30	5260	187.5			0.62							
92445				5288	188.5										
92446				5320	189										
92447				5344	189.5										
92448	2	44	34	5372	191					0.9					
92449				5396	192										
92450				5420	193.5										
92451				5436	195										
92452	2	44	38	5452	196.5										
92453				5460	198.5										
92454				5464	200.5										
92455				5468	202.5										
92456	2	44	42	5460	205.5		0.7								
92457				5452	207.5										
92458				5432	209.5										
92459				5408	212										
92460	2	44	46	5380	215			0.92							
92461				5332	218.5										
92462				5276	222										
92463				5204	225.5										
92464	2	44	50	5096	230.5					0.58					
92465				4972	236.5										
92466				4816	244.5										
92467				4628	254										
92468	2	44	54	4388	264.5										

Time (seconds)	GMT HOURS (HOURS)	GMT MINUTES (MINUTES)	GMT SECONDS (SECONDS)	ALTITUDE (29 92) (FEET)	COMPUTED AIRSPD (KNOTS)	CN1 TRACKED VIB L (SCALAR)	CN1 TRACKED VIB R (SCALAR)	CN2 TRACKED VIB L (SCALAR)	CN2 TRACKED VIB R (SCALAR)	TN1 TRACKED VIB L (SCALAR)	TN1 TRACKED VIB R (SCALAR)	FAN IMB ANGLE L (DEG)	FAN IMB ANGLE R (DEG)	LPT IMB ANGLE L (DEG)	LPT IMB ANGLE R (DEG)
92469				4124	275.5										
92470				3820	289.5										
92471				3508	306.5										
92472	2	44	58	3068	317.5							166			
92473				2640	334										
92474				2216	352										
92475				1748	368.5										
92476	2	45	2	1320	382.5									334	
92477				904	395										
92478				524	410										
92479				180	416										
92480															

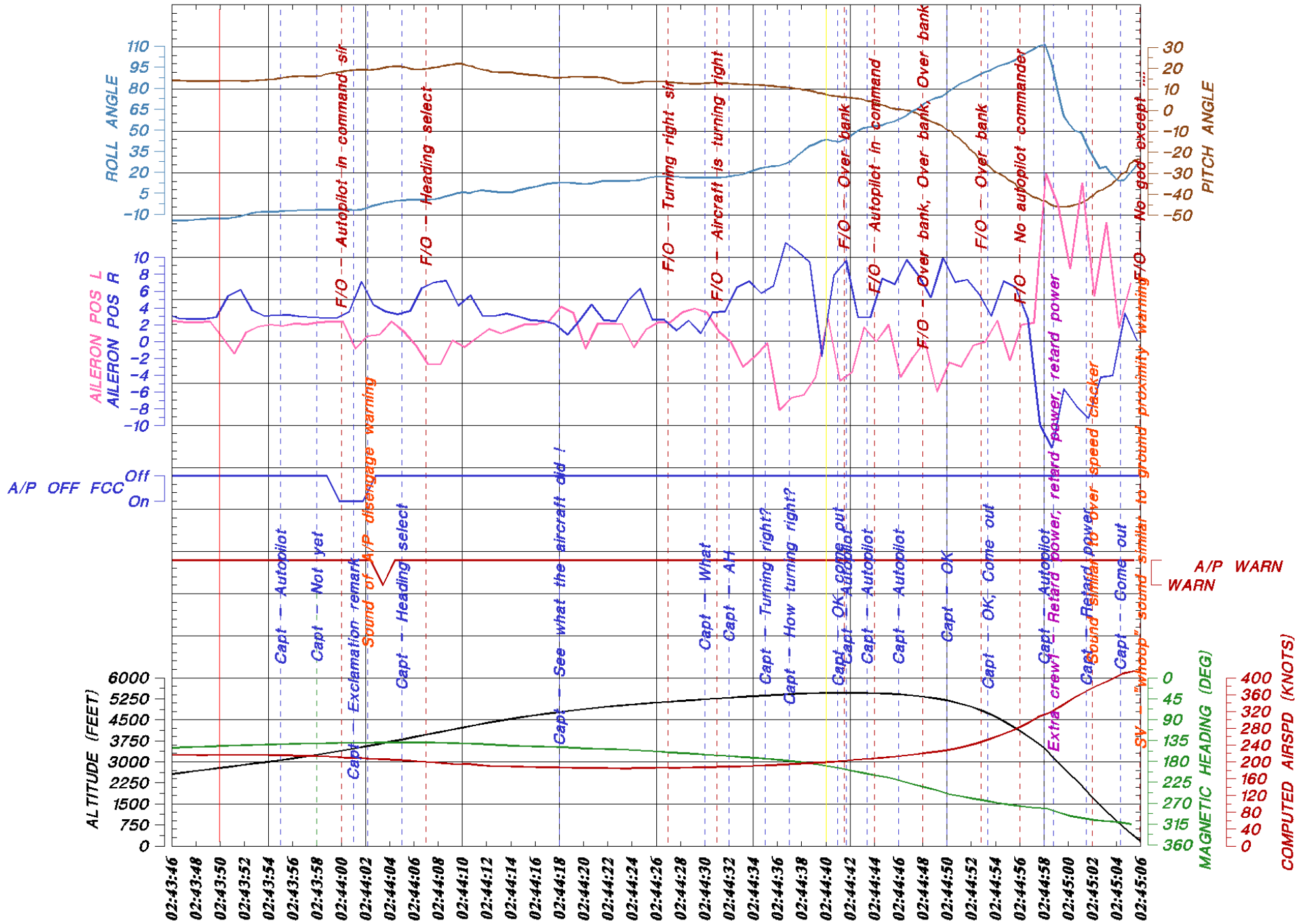
**Attachment 2, FDR Plots**





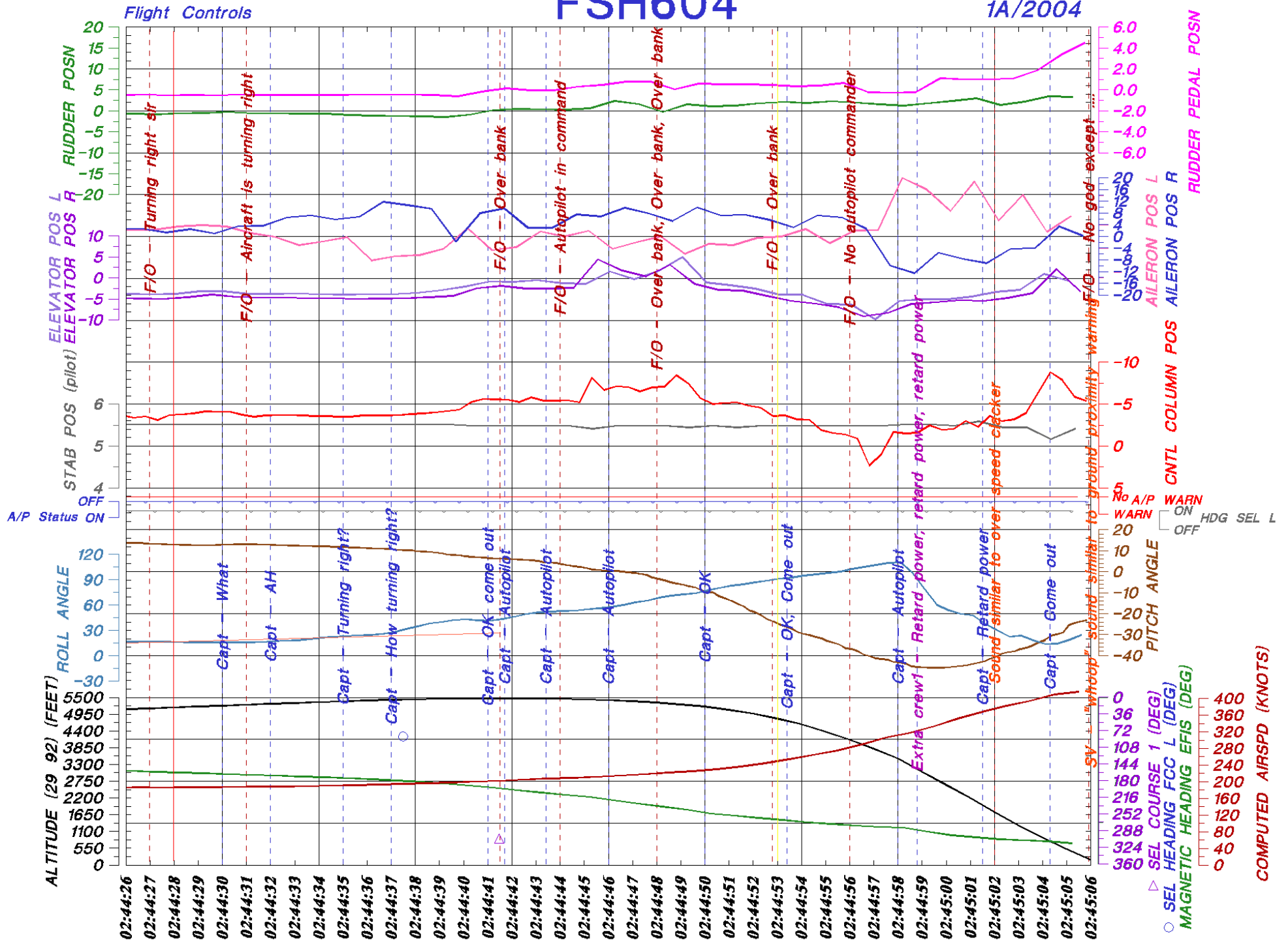
**Attachment 3, five plots represent FDR and CVR correlation**





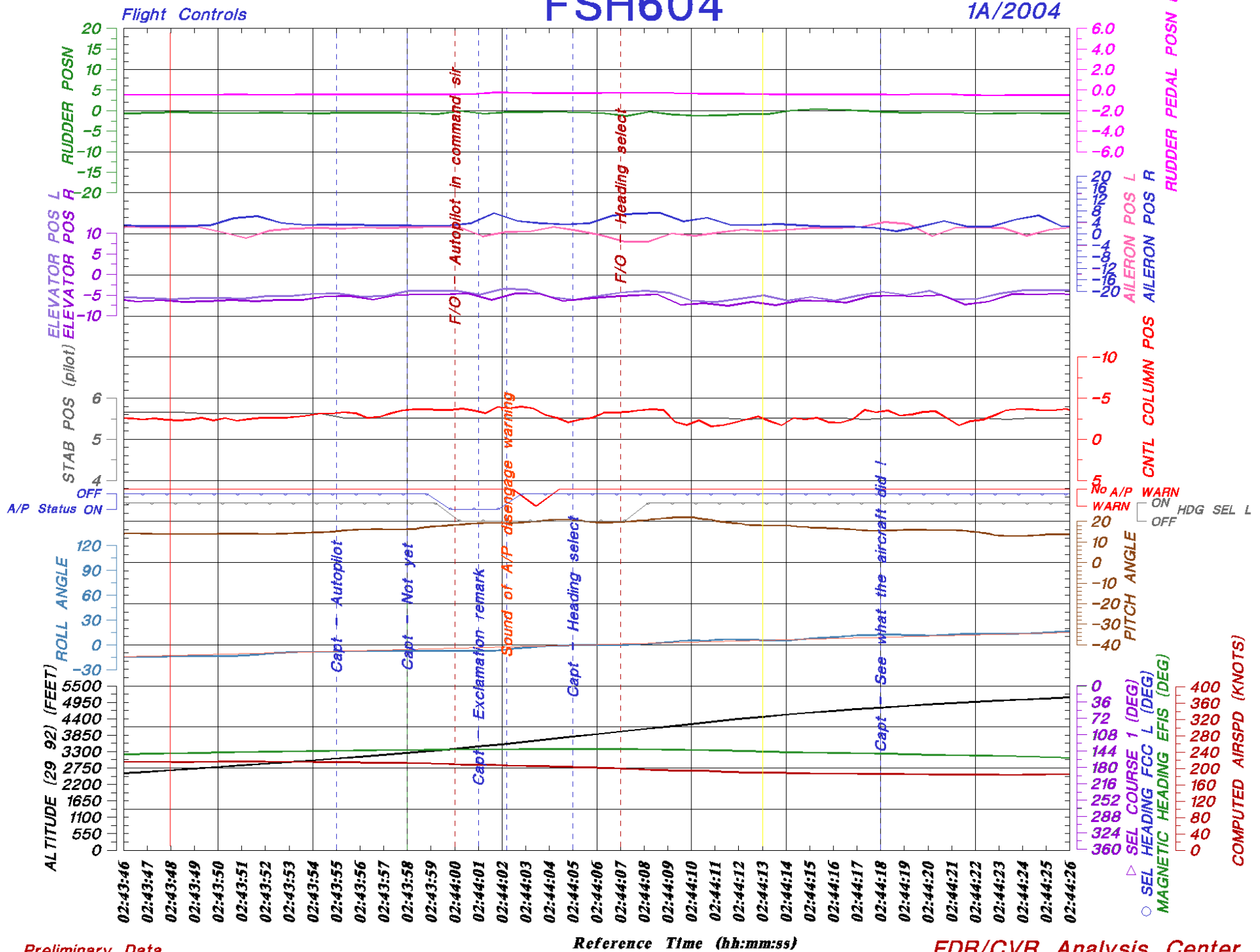
# FSH604

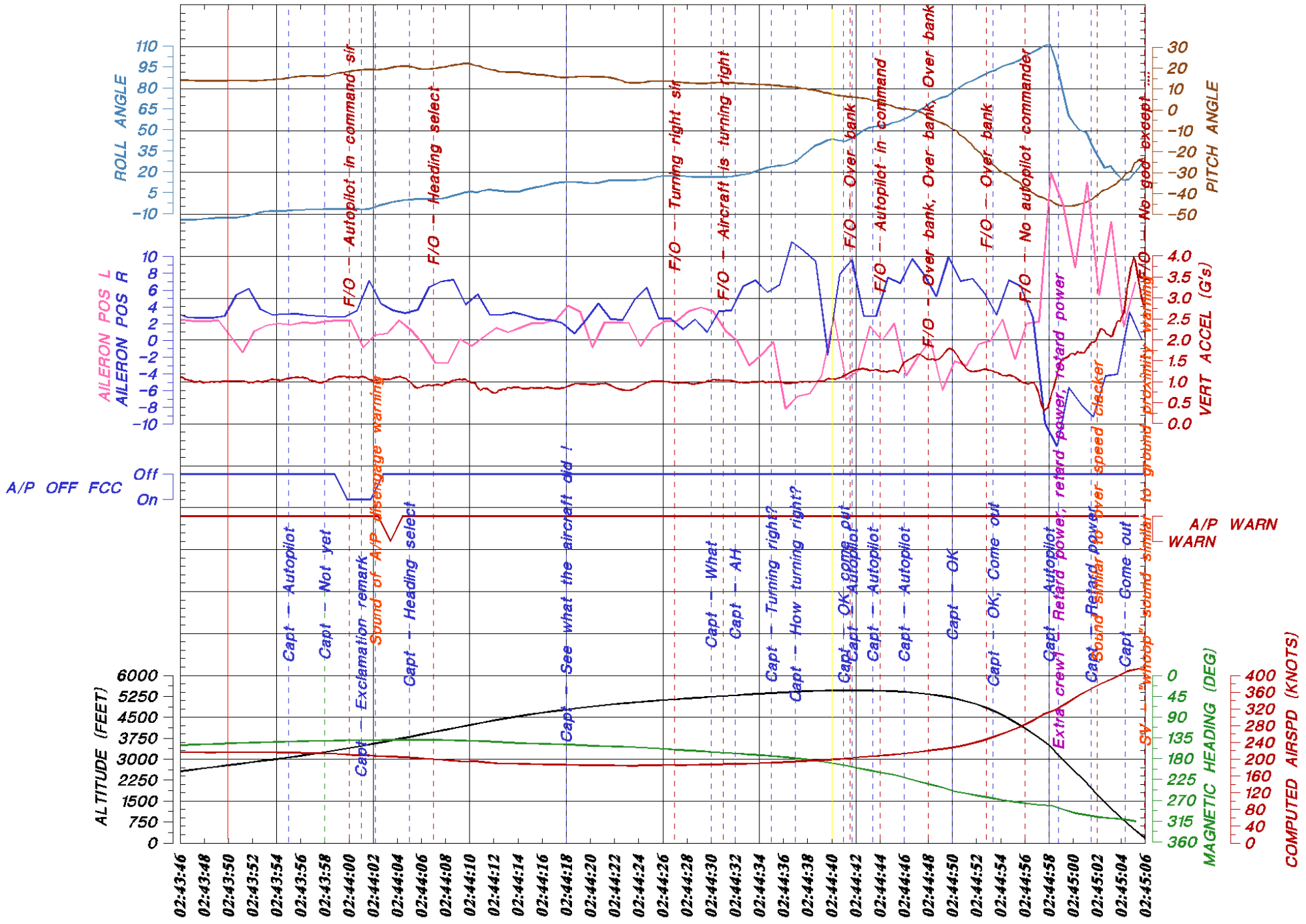
1A/2004

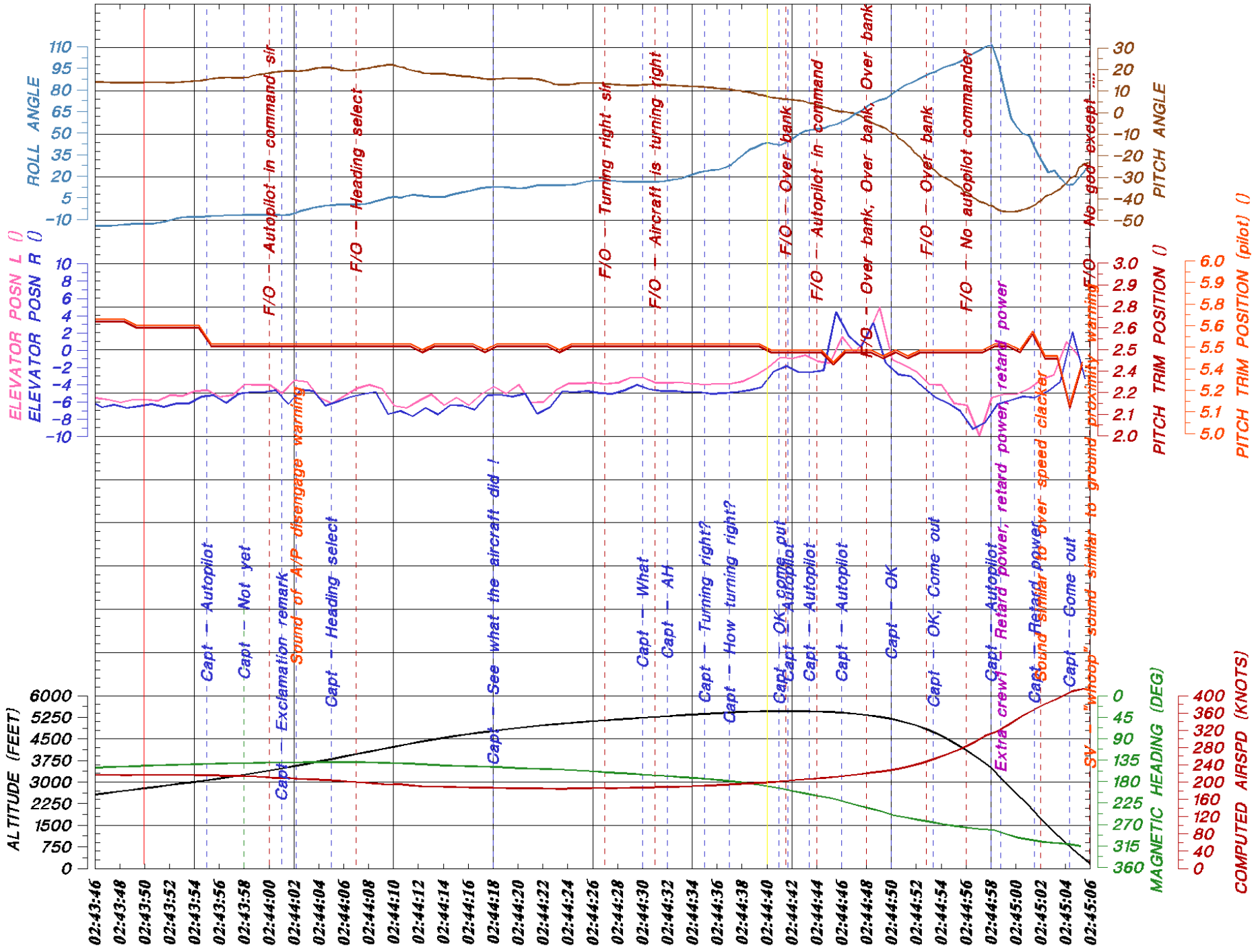


# FSH604

1A/2004







Attachment 4: Summaries of previous flight(s) by accident crew

Refer to 1.17.3.25, all departures from SSH (accident aircraft)

# **Exhibit C**

## **Cockpit Voice Recorder (CVR) Group Factual Report**

**Ministry of civil aviation  
Accidents Department  
Egypt, Cairo**

**October 14, 2004**

**Group Chairman's Factual Report – Cockpit Voice Recorder**

**ACCIDENT**

**Location:** Red Sea off Sharm el-Sheikh  
**Date:** January 3, 2004  
**Time:** 2:45:06 GMT  
**Operator:** Flash Airlines – Flight 604

The group convened at CVR/FDR laboratory at MCA headquarters - Cairo for retrieval of CVR conversation and aural sounds.

**SUMMARY**

On January 3, 2004, about 02:45:06 UTC, 04:45:06 Local time, Flash Airlines flight FSH604, a Boeing 737-300, Egyptian registration SU-ZCF, operated by Flash Airlines, crashed into the Red Sea shortly after takeoff from Sharm el-Sheikh International Airport (SSH) in South Sinai, Egypt. The flight was a passenger charter flight to Charles de Gaulle Airport (CDG), France with a stopover in Cairo international Airport (CAI) for refueling. Flight 604 departed from Sharm el-Sheikh airport with 2 pilots (Captain and First Officer), 1 observer, 4 cabin crew, 6 off-duty crew members and 135 passengers on board. The airplane was destroyed due to impact forces with the red sea with no survivals.

**Details of Investigation**

- The accident airplane's Cockpit Voice data recorder (CVR), Fairchild, Part no. 93-A100 – 80, serial no. 57994 was retrieved from the Red Sea on January 17, 2004 by the French Navy. The CVR was immersed in water and sealed in an ice chest and transported to MCA, accident investigation laboratory at Cairo.
- Readout of the CVR was accomplished using the laboratory's playback hardware and software as follow:



**Download Unit:**

A100 CVR play back Deck - Store 4DS

**Audio Analysis System:**

MPL 1024 , 12 Channel Microphone Mixer – Samson

Filter : PCAP II (Samson)

Amplifier : Samson - Servo-550 Studio Amplifier

**Software:**

Vegas 4 – Sound Forge 6 –PCAP II

- The recorder consisted of four channels of audio information.

Channel One:	First officir hot mic.
Channel Two:	Area Mic.
Channel Three:	Observer hot Mic.
Channel Four:	Captain hot Mic.
  
- After the initial retrieved sound task was completed another effort was undertaken with the assistance of BEA expert as follows:
  - The output signal from the tape deck playback machine was too low compared to the recording on the same conditions in BEA. This problem was solved by increasing the output level when the screw of the adjustable gain control was turned clockwise.
  
  - The sensitivity of the acquisition audio card of the PC was not good enough to capture correctly the audio signal coming from the tape deck player. This problem was solved by changing the value of the “Variable Signal Levels” on the hardware setting of the audio card, from the manufacture value +4 to -10. The gain was increased and the input signal amplified.
  
  - The speed of the tape was not correct with an interference of the power (115 V, 400 Hz) measured at 375 Hz. It was not possible to adjust properly the speed of the tape with the device installed. This problem is solved by reassembling the wave file with a correct ratio ( $400/375= 1.0665$ ).
  
  - Some high frequencies were missing when doing the spectrum analysis. This problem was solved by using a sampling rate of 32000 kHz instead of 22000 kHz.
  
  - The alignment of the head installed on tape deck player was checked, adjusted and was found satisfactory prior to playback the tape.

A new copy of the CVR was performed. This recorded copy is satisfactory.

- Due to the effect of aircraft power (115 V, 400 Hz) on the tape speed, the data had been retrieved at a sample rate 34128 HZ. Recording time of the Subject CVR measured found 31 min. and 13.7 sec. and the frequency was 402 HZ

## **Comments**

- Before start check list, below the line, Engine start, after start check list, and before Takeoff check list down to strobe lights are carried out.
- During flight control check at 02:37:40, two consecutive sounds had occurred, following at 02:37:41 the Captain had announced "turning to the right".
- Before the engine started, sound similar to Cockpit door operation was heard and no body other than the Captain, the First officer and the Extra crew1 was in the cockpit till the end of the CVR tape.
- At 02:42:43, the Captain requested for "Four Hundred Heading Select". One second later the First Officer acknowledged "Four Hundred Heading Select"
- At 02:42:484, the Captain had asked for "Level Change". One second later the First Officer repeated "Level Change".
- At 02:43:04 and at 02:43:11, the captain had announced "Left Turn". One second later the First Officer repeated " Left turn to establish Three Zero Six Sharm VOR"
- At 02:43:55, the Captain had asked for "Autopilot". At 02:44:00, The First officer announced "Autopilot in command" and at 02:44:02, the sound of autopilot disengages warning was heard.
- At 02:44:05, the Captain had asked for "Heading Select". At 02:44:07, the First Officer repeated "Heading Select"
- AT 02:44:27, the First Officer had announced "Turning right Sir" and again at 02:44:31, he confirmed "Aircraft is turning right".
- At time 2:44:35 Captain said "turning Right?"
- At time 2:44:37 Captain said "how turning right"
- At 02:44:41.7 and at 02:44:43.4, the Captain had asked for "Autopilot", and at 02:44:44 the First Officer replied "Autopilot in command"

- At 02:44:46, the Captain had asked for "Autopilot", and at 02:44:56, the First officer replied "No autopilot Commander" but again the Captain in command asked for "Autopilot".
- The phrase "Come out" was repeated three times by the captain at 02:44:41, 02:44:53.4 and 02: 45:04.3
- Extra crew 1 did not interfere during flight progress except at 02:44:58.8 when he had been announced "Retard Power, Retard Power, Retard Power"

**Transcript of a Fairchild A-100 cockpit voice recorder (CVR), serial no. 57994 installed on a B-737-500, SU-GZF, which was involved in a descent and collision into the Red Sea near Sharm on Jan, 2004**

UTC hh:mm:ss	Speaker	Content
02:13:53	ATC	Communication with Blue Panorama B757 (●●●) for 31 seconds
02:14:27	Extra crew1	طردوه ياعم مش عايزينه يقعد طول الليل هنا <i>They don't want him to stay here all night</i>
02:14:30	First officer	ممکن حضرتك علشان بيودوهم عند الهناجر <i>May be because they move them next to the hanger</i>
02:14:32	Extra crew1	لا قالوه حبيعتوه للغردقة * <i>They told him they will send him to Hurgada</i>
02:14:43	First officer	بص خلاص علشان انا شايف يعنى الترافك بدأ يقل فى اليومين دول <i>The traffic started to decrease</i>
02:14:47	Extra crew1	والله <i>Really</i>
02:14:48	First officer	آه مش <i>yes</i>
02:14:49	Extra crew1	انا افكرته على جذا <i>I thought it was still high</i>
02:14:50	First officer	لا احنا نازلين حاضرتك امبارح مثلاً الساعة خمسة ومن خمسة لغاية ستة المطار زى كدة بالضبط <i>No we are decreasing</i>
02:14:59	Extra crew1	ياه <i>Really</i>

UTC hh:mm:ss	Speaker	Content
02:15:02	First officer	ده بالعكس كله دلوقتي بيبتدى بقى يسافر خلاص كله قضى رأس السنة و الكريسماس <i>Every body is going back after Christmas &amp; new year</i>
02:15:07	Extra crew1	آه yes
02:15:21	Extra crew1	بوينج سبعة وخمسين <i>Boeing seven five seven</i>
02:16:02	First officer	بقول لحضرتك كابتن عصام يعنى استاذى يعنى <i>I am telling you sir captain Essam is my teacher</i>
02:16:10	Extra crew1	والله !!! <i>Really</i>
02:16:13	First officer	حضرتك كان مسمينى حتى "مازو" على اسم ابنه الصغير لو حضرتك تعرفه على اساس كنت ابتديت الطيران صغير <i>He even calls me like his youngest son</i>
02:16:24	Extra crew1	ابتديت ازاي <i>How did you start</i>
02:16:26	First officer	انا ابتديت حضرتك خلصت تمتاشر طبعا كوميرشيوال وقعدت حوالى سنة ونصف ابتديت قبل العشرين كده <i>I started by finishing commercial at eighteen and stayed for a year and half and started before twenty</i>
02:16:35	Extra crew1	آه yes
02:16:43	First officer	احسن حاجة فادتنى طبعا ان انا ابتديت على الميتين يعنى الميتين ده مدرسة <i>The best benefit was my starting on the two hundred</i>

UTC hh:mm:ss	Speaker	Content
02:16:46	Extra crew1	آه دراسة يعنى مش حظ <i>Yes studying not luck</i>
02:16:47	First officer	الحمد لله يعنى <i>Thank god</i>
02:16:52	Extra crew1	انا عايش بره <i>I live abroad</i>
02:16:54	First officer	* حضرتك <i>you sir</i>
02:16:55	Extra crew1	ياه ما عندناش النظام ده خالص لازم تعمل ألفين ساعة قبل ما حد يبصلك يعنى <i>You must have two thousand hours before anyone looks at you</i>
02:17:04	First officer	فين حضرتك <i>where sir</i>
02:17:05	Extra crew1	اشتغل مدرب شوية اشتغل رش شوية اشتغل bush pilot <i>Work as instructor a bit and a bit as bush pilot</i>
02:17:10	First officer	بس كلها إكسبيرينس عالية <i>But it is all high experience level</i>
02:17:12	Extra crew1	اكسبيرينس بس بتشتغل على طيارات صغيرة وسنجل إنجين ، ما بتخدش الإكسبيرينس اللي هو يعنى تقعد انت خمس سنين كده بتضيعهم أونطة يعنى بس انا زيك انا كنت دفعة تسعة وثمانين حتى كان عندي تمتناشر سنة حتى كان عندي يعنى كان لازم اجيب موافقة من بابا ومش عارف إيه <i>It is all experience but it is a waste of time</i>

UTC hh:mm:ss	Speaker	content
02:17:39	First officer	آه ما هوه بالضبط حصل معايا نفس الموضوع <i>Yes I passed through the same thing</i>
02:17:43		عدة اصوات منها فتح باب الكابينة Sound like cockpit door operation
02:18:10		صوت نقر على باب كابينة القيادة Knocking on cockpit door
02:18:11		أصوات sounds
02:18:13	Attendant	كابتن الركاب جت <i>Captain the passengers arrived</i>
02:18:14	Captain	اتفضلوا <i>let them in</i>
02:18:20	Extra crew2	السلام عليكم <i>Greeting</i>
02:18:21	Captain + extra crew1	وعليكم السلام ورحمة الله وبركاته <i>Response</i>
02:18:23	Extra crew2	انا حياتى جوه فى اوديت الفيران هنا <i>My life is in this rat room</i>
02:18:24		*صوت ضحك عالى Laughter



UTC hh:mm:ss	Speaker	Content
02:18:25	Captain	امشى اطلع بره <i>Go outside</i>
02:18:25		صوت ضحك laughter
02:18:26	First officer	انت طالع معانا <i>Are you coming with us</i>
02:18:27		(●●●) joking for 31 seconds
02:18:58	Captain	Before start check list
02:18:59	First officer	Flight deck preparation
02:19:00	Captain	Completed
02:19:01	First officer	light test
02:19:02	Captain	Checked
02:19:03	First officer	Oxygen
02:19:04	Captain	Push * hundred percent ( sound similar to oxygen mask test )
02:19:05	First officer	Yaw damper
02:19:06	Captain	On
02:19:07	First officer	Instrument transfer switches
02:19:08	Captain	Ok normal , I R S was *
02:19:12	First officer	Fuel

UTC hh:mm:ss	Speaker	Content
02:19:14	Captain	On
02:19:16	First officer	Galley power
02:19:17	Captain	On
02:19:18	First officer	Emergency Exit light
02:19:19	Captain	Armed
02:19:20	First officer	Passenger signs
02:19:21	Captain	set
02:19:22	First officer	Window heat
02:19:23	Captain	On
02:19:24	First officer	Hydraulics
02:19:26	Captain	Normal
02:19:28	First officer	Air condition & Pressurization
02:19:30	Captain	Packs on , bleeds on , set at Cairo
02:19:33	First officer	Auto pilot
02:19:34	Captain	Disengaged
02:19:35	First officer	Instruments
02:19:36	Captain	Cross Checked
02:19:37	First officer	Anti-skid
02:19:38	Captain	On

UTC hh:mm:ss	Speaker	Content
02:19:39	First officer	Auto brake
02:19:40	Captain	RTO
02:19:40	First officer	Speed brake
02:19:41	Captain	Down
02:19:42	First officer	Parking brake
02:19:43	Captain	Set
02:19:45	First officer	Stabilizer trim cut out switches
02:19:46	Captain	Normal
02:19:47	First officer	Wheel well fire warning
02:19:48	Captain	Checked
02:19:49	First officer	Radio radar and transponder
02:19:50	Captain	Set
02:19:51	First officer	Rudder and aileron trim
02:19:52	Captain	Neutral
02:19:53	First officer	Gear pins
02:19:55	Captain	Removed
02:19:56	First officer	Briefing for emergencies
02:19:58	Captain	*
02:19:59	First officer	Papers

UTC hh:mm:ss	Speaker	Content
02:20:01	Captain	Aboard
02:20:02	First officer	F M C / C D U
02:20:03	Captain	One three four , One three four , one four zero
02:20:06	First officer	N one and I A S ‘ bugs
02:20:07	Captain	None , ninety four set my sides
02:20:12	First officer	Flight director
02:20:13	Captain	Ok *
02:20:17	First officer	Before start check list complete down to the line
02:20:25	Extra crew1	طبعاً انتو منزلتوش من الاوتيل خالص <i>Of course you didn't leave the hotel</i>
02:20:27	Extra crew2	آه <i>yes</i>
02:20:29	Extra crew2	لا هانروح فين عريانيين <i>No where can we go without clothes</i>
02:20:33	Extra crew1	لا دول علشان شوناظهمم ضاعت <i>No that's because their bags are lost</i>
02:20:35	Captain	امبارح كنا جاينين ساعة الغسق شمس و two two <i>Yesterday we were coming at dusk and the sun was two two</i>
02:20:43	Extra crew2	اه <i>yes</i>

UTC hh:mm:ss	Speaker	Content
02:20:46	Captain	حسيت انه انا already شاييف الممر بالعافية هو بيقول in sight قولتله in sight ايه <i>I felt I could hardly see the runway and he was already saying in sight ..... what in sight</i>
02:20:53	Extra crew1	سن بأه يا كابتن <i>Age sir</i>
02:20:55	Captain + extra crew2	احنا * دا مش in sight بالنسبة لك او عى تقول in sight فى اللى انتا داخل عليه ده مش in sight خالص <i>This is not in sight never say in sight when you are entering like this</i>
02:20:59	Extra crew2	مش هو ده مش هو ده <i>This is not it</i>
02:21:00	Captain	مش باين لحد short انا يعنى انا بجيب الـ * اللى انا هو ده <i>It is not clear to the short</i>
02:21:05	First officer	ماهو الـ sunset ضارب مع الشمس مع haze <i>It is the sunset and the haze</i>
02:21:07	Captain	الشمس عمله haze مش ممكن <i>The sun is making haze</i>
02:21:07	First officer	عمله haze فظيع يعنى <i>It is making terrible haze</i>
02:21:10	Captain	لا عارف ارفع عنيا برة وبيقولى * in sight <i>I am unable to raise my eyes and he says in sight</i>
02:21:12		صوت ضحك * <i>Laughter</i>

UTC hh:mm:ss	Speaker	Content
02:21:13	Captain	فين in sight ده بيقولى اهو ياكابتن كابتن فى عينك <i>where in sight .....</i>
02:21:19	Captain	بقوله اذا كنت انا شايفه بالعافية ومحدده بالعافية تقولى in sight ازاي مستحيل تكون انتي شايفه <i>If I can hardly see it and he says in sight how ?</i>
02:21:26		*
02:21:27		ضحك Laughter
02:21:30	Captain	انتا عارف اصل ايه ال maneuver تبين ال in sight وخاصة فى الجزء بتاع ال short final <i>You know the maneuver shows in sight specially on short final</i>
02:21:34	First officer	بالذات ال correction بتاع ال heading <i>Specially heading correction</i>
02:21:37	Captain	بالضبط * <i>Exactly</i>
02:21:40	Captain	ده انا قولتله انا شايفه بالعافية انا اعدت ادور عليه علشان انزل عليه بالعافية ازاي يبقى in sight بالنسبة لك <i>I told him I searched for it to see it how in sight ?</i>
02:21:52	Extra crew2	in sight وخلص يا كومنندان مادققشى على الحاجات الصغيرة <i>Simply in sight</i>
02:21:52	Captain	صوت ضحك وانزل على الممر الثانى Laughter ..... <i>Then land on the other runway</i>

UTC hh:mm:ss	Speaker	Content
02:21:52		(●●●) conversation about the lost bags of the crew for 83 seconds
02:23:40		صوت مماثل لحركة باب غرفة القيادة sound similar to cockpit door operation
02:23:48	Captain	كام واحد كام راكب <i>How many Passengers?</i>
02:23:49	Station manager	ميه خمسة وتلاتين رأس One three five <i>heads</i>
02:23:51		(●●●) Joking + conversation of blue panorama eight three three amend their flight plan (For 150 seconds)
02:26:22	First officer	Sharm El Sheikh Flash Six Zero Four
02:26:29	ATC	Six Zero Four go ahead
02:26:31	First officer	weather Cairo أستأذن حضرتك لو فيه امكانية <i>Please weather Cairo</i>
02:26:34	ATC	ثواني <i>seconds</i>
02:27:35	First officer	ده option <i>This is option</i>
02:27:36	Extra crew1	هه <i>what</i>
02:27:37	First officer	فى option <i>There is option</i>

UTC hh:mm:ss	Speaker	Content
02:27:40		*
02:28:05	First officer	حضرتك طلبت level عالي ليه <i>Sir why did you request a high level?</i>
02:28:08	Captain	System كده هنفذه لانه هيقال من نسبة الـ consumption بتاعنا <i>For less consumption</i>
02:28:50	Extra crew1	عداد الـ center tank شغال <i>Is the center tank gauge operating?</i>
02:28:53	Captain	اه بس مشكوك فيه <i>Yes but not reliable</i>
02:28:57	Extra crew1	شغال يعنى هو zero فعلا <i>So it is zero</i>
02:28:58	Captain	اه <i>yes</i>
02:28:59	ATC	Flash Six Zero Four Sharm El Sheikh
02:29:02	First officer	Go ahead sir
02:29:03	ATC	Six Zero Four copy Cairo met condition time Zero Two double zero , Surface wind Two One Zero One Zero knots Visibility Six kilometers Clouds and Sky clear Temperature One Two ,dew point Zero One , QNH one zero one three
02:29:23	Captain	Clouds
02:29:24	First officer	And confirm dew point, Please



UTC hh:mm:ss	Speaker	Content
02:29:26	Captain	sky clear مافلوش <i>They didn't say sky clear</i>
02:29:27	ATC	Dew point Zero One
02:29:30	First officer	Roger, copied next call when ready إنشاء الله يافندم <i>God willing</i>
02:29:33	Captain	قالوه و sky clear و clouds عكس بعض <i>They said clouds and sky clear how , the two are opposite</i>
02:29:34	Extra crew1	اسأله عن ceiling كده <i>Ask him about ceiling?</i>
02:29:35	First officer	ازاي يعني <i>How?</i>
02:29:37	First officer	شوف بيقولك sky clear و cloud ازاي مش فاهم <i>See how sky clear and clouds I don't understand</i>
02:29:37	First officer	ماهو لخبطني فيها علشان كده ماعرفتش اكتب اللي بعده <i>He mixed me up I didn't know how to write it</i>
02:29:41	Extra crew1	مادكاش ceiling فعلا <i>He didn't give ceiling</i>
02:29:42	Captain	One Zero One Three
02:29:43	First officer	One zero one

UTC hh:mm:ss	Speaker	Content
02:29:44	Captain	هه
02:29:45	First officer	One zero one three
02:29:46	Captain	أه و المتين وعشرة وعشرة knots يبقى الـ runway <i>And two hundred and ten and ten knots and runway is</i>
02:29:50	First officer	Runway two three
02:29:53	Extra crew1	ماداش ( ceiling ) <i>He didn't give ceiling</i>
02:29:54	First officer	و الـ sky clear و clouds لا مافيش ceiling <i>No ceiling and ..... clouds and sky clear</i>
02:30:01	Extra crew1	ممکن تبقى مثلاً scattered <i>Maybe it is scattered</i>
02:30:02	First officer	ممکن يقصد scattered <i>Maybe he means scattered</i>
02:30:06		صوت خبط sound of knock
02:30:11	Extra crew1	بس برده لازم يبقى فيه ceiling <i>There should be ceiling</i>
02:30:14	First officer	اكيد <i>Definitely</i>

UTC hh:mm:ss	Speaker	content
02:30:14	Extra crew1	نعرف هنخرج منه إمتى <i>How can we know when will we clear it</i>
02:30:16	First officer	أه <i>yes</i>
02:30:16	Ground engineer	ياصباح الجمال <i>Good morning</i>
02:30:18	Captain	يا صباح الهنا يا باشمهندس <i>Good morning engineer</i>
02:30:21	Captain	شوفت ده <i>Did you see it ?</i>
02:30:22	Ground engineer	أه انا كان في امكاني اعمل بس لأ مش عاوز امد ايدى على حاجة دى <i>Yes I could do something but I don't want to touch this</i>
02:30:24	Captain	تخصصات كهربيا <i>Electrical specially</i>
02:30:27	Ground engineer	أه <i>yes</i>
02:30:29	Captain	زى ماكان بيحصل <i>Like what used to happen</i>
02:30:30	Ground engineer	أه <i>yes</i>

UTC hh:mm:ss	Speaker	Content
02:30:30	Captain	فى الطيارات اياها الثانية <i>In the other aircraft</i>
02:30:31	Ground engineer	ده صح <i>This is right</i>
02:30:32	Captain	Socket بس هز <i>Move socket</i>
02:30:33	Ground engineer	ايوه <i>yes</i>
02:30:36	Extra crew1	لازم عمرو عمل heavy landing <i>Probably Amr made a heavy landing</i>
02:30:37	Ground engineer	صوت ضحك <i>laughter</i>
02:30:39	Captain	راجل زى الفل <i>Good man</i>
02:30:41	Extra crew1	والله ما شاء الله <i>God's will</i>
02:30:48	Captain	لو نركز فى السن ده <i>If we concentrate at this age</i>
02:30:53	First officer + Extra crew1	عالطول ان شاء الله <i>Always god willing</i>

UTC hh:mm:ss	Speaker	Content
02:30:54	Captain	يافندم منترمش تحب تيجي معانا <i>Thank you would you like to come with us?</i>
02:30:56	Station manager	مين <i>Who?</i>
02:30:56	Ground engineer	نخطفه يا كابتن النهارده نخطفه <i>Lets steal him</i>
02:30:57	Station manager	ازاي بس اجي معاكم عندنا وارسو وعندنا * وعندنا * <i>How we have Warsaw and * and*</i>
02:31:01	Captain	بلا وارسو بلا حاجة <i>Forget Warsaw</i>
02:31:03	Station manager	لا النهاردة بالذات مش هاجي <i>No today I will not go with you</i>
02:31:05		صوت ضحك laughter
02:31:07	Station manager	مش قابل اجي يعنى عارف مش جاية مش قابلة <i>I can't make it , it can't be done</i>
02:31:10	Extra crew1	انتى نمت امبارح <i>Did you sleep last night</i>
02:31:11	Station manager	مين <i>Who?</i>

UTC hh:mm:ss	Speaker	content
02:31:11	Extra crew1	انتى <i>you</i>
02:31:12	Station manager	انا منمتش امبارح خالص انا هاخذها نوم انا لازم انام <i>I didn't sleep at all I must sleep</i>
02:31:16	Captain	طيب اتوكل على الله ، اسحبولنا الحاجة <i>Ok rely on god pull equipment away</i>
02:31:21		صوت sound similar to cockpit door operation
02:31:26	Attendant	كابتن one three five <i>captain one three five</i>
02:31:28	Captain	ثمانية و عشرين و بقولك ايه خمسين دقيقة ولا اقل ، إنشاء الله <i>Twenty eight and lets say fifty minutes , god willing One three five</i>
02:31:34	First officer	خمسين <i>fifty</i>
02:31:36	Captain	شكراً <i>thank you</i>
02:31:37	First officer	طب هو فين * <i>Ok where is he ?</i>
02:31:39	Captain	من هنا خمسين من هنا ستة و خمسين لكن إنشاء الله اقل إنشاء الله <i>From here fifty and from there fifty six but god willing less</i>

UTC hh:mm:ss	Speaker	Content
02:31:44	Attendant	أه أقفل الباب؟ <i>Yes close the door</i>
02:31:48	Attendant	بسرعة بسرعة علشان الكابتن بيقولى اقفل <i>Quickly the captain says close</i>
02:31:51		صوت قفل الباب <i>Sound of door closing</i>
02:31:52	First officer	ياكمومندان Startup <i>Startup commander</i>
02:31:53	Captain	اتفضل يا حبيبي <i>Please</i>
02:31:55	First officer	Sharm El Sheikh Tower Flash Six Zero four
02:32:00	ATC	Flash Six Zero Four Go ahead
02:32:02	First officer	On our stand, destination Cairo request startup clearance
02:32:05	ATC	Startup approved QNH One Zero One One , Runway Two Two Right
02:32:09	First officer	Startup approved for runway Two Two Right , Flash Six Zero Four thank you
02:32:13	First officer	Startup approved
02:32:19	Captain	Below the line
02:32:21	First officer	Doors
02:32:22	Captain	لسه <i>Not yet</i>
02:32:23	First officer	Air condition packs

UTC hh:mm:ss	Speaker	Content
02:32:24	Captain	Off
02:32:28	First officer	Start pressure
02:32:29	Captain	Sufficient
02:32:30	First officer	Anti collision light
02:32:31	Captain	On
02:32:31	First officer	Before start check list completed down to the after start
02:32:58	Extra crew3	يلا يا جماعة اتكلوا على الله <i>Come on fellows</i>
02:33:00	Attendant	Close two L Please
02:33:07		صوت خبطة (thump)
02:33:16	Captain	توكلنا على الله والحمد لله بسم الله الرحمن الرحيم <i>We rely on god , thank god , in the name of god</i>
02:33:20		اصوات خبطات Sounds
02:33:25	Attendant	Attention Cabin Crew doors in armed position and crosscheck
02:33:30		اصوات خبطات Sounds For 47 seconds (may be cockpit door , jump seat and unknown ratcheting sounds )
02:34:08	Captain	أيه ده بقى <i>What is this</i>



UTC hh:mm:ss	Speaker	content
02:34:09	First officer	بسم الله وتوكلنا على الله <i>In the name of god , we rely on god</i>
02:34:11	First officer	Duct pressure decrease start valve open
02:34:14	Captain	N two
02:34:25	Attendant	Ladies and Gentlemen, good morning on behalf of Captain Kheder and his crew members welcome you onboard Flash airlines, Boeing seven three seven three hundred Proceeding to Cairo, During our flight to Cairo we shall cover the distance at fifty minutes and altitude twenty seven thousand feet , you are kindly requested to fasten your seat belts and put the back of your seats in full up right position, and observe the no smoking sign during all the flight, thank you.
02:34:31	First officer	Oil pressure
02:34:48	First officer	Approaching forty six
02:34:50	First officer	Duct pressure normal start valve closed
02:34:51	Attendant	Cabin crew stand bye for demo.
02:35:06	Captain	number one توكلنا على الله <i>We rely on god</i>
02:35:08	First officer	Duct pressure decrease start valve open
02:35:10	Captain	N two
02:35:16	Captain	E G T ثلاثا عشر تسعنا عشر كده لما دور تاني <i>E G T thirteen, nineteen when it starts again</i>
02:35:21	First officer	Approach *
02:35:22	Captain	N one E G T ok Normal

UTC hh:mm:ss	Speaker	Content
02:35:27	First officer	Maximum motoring
02:35:30	First officer	Oil pressure
02:35:48	Captain	Approach forty six start cut out pressure normal Start valve closed start cut out
02:36:04	Captain	Stabilized
02:36:13	Captain	To the line
02:36:14	First officer	Electrical
02:36:16	Captain	On bus
02:36:17	First officer	Pitot heat
02:36:17	Captain	on
02:36:18	First officer	Anti-ice
02:36:19	Captain	on
02:36:19	First officer	Air condition and pressurization
02:36:21	Captain	Packs on , flight
02:36:23	First officer	Isolation valve
02:36:24	Captain	Auto
02:36:25	First officer	A P U
02:36:29	Captain	ندوره هناك فى الجو مش مشكلة ربنا يسهل <i>Start there in flight no problem with god's help</i>
02:36:30	First officer	Start levers

UTC hh:mm:ss	Speaker	Content
02:36:32		*
02:36:33	Captain	Idle detent
02:36:34	First officer	Ground equipment
02:36:36	Captain	Clear
02:35:36	First officer	After start check list completed
02:35:37	Captain	Taxiing
02:36:39	First officer	Sharm El Sheikh Flash six zero four Ready to taxi out
02:36:48	ATC	Six Zero Four Taxi right Delta Alpha Hold short Two Two Right
02:36:53	First officer	Roger to the right via Delta Alpha to holding point runway Two Two Right flash Six Zero Four
02:36:59	First officer	To the right ان شاء الله Delta Alpha يا كومنذار <i>Commander Delta Alpha god willing to the right</i>
02:37:02	Captain	ان شاء الله <i>God willing</i>
02:37:03	First officer	Holding point runway two two right and right side is clear
02:37:06		صوت sound
02:37:07	Captain	توكلنا على الله <i>We rely on god</i>
02:37:08	First officer	Shocks off zero two three *

UTC hh:mm:ss	Speaker	Content
02:37:09		صوت sound
02:37:09	Captain	هو ده مش شغال عادى <i>Is this not operating normally</i>
02:37:10		صوت sound
02:37:11		صوت ربما يكون ال- sound maybe parking brake release
02:37:14	First officer	One minute past for A P U
02:37:16	Captain	Off
02:37:18	First officer	A P U off sir
02:37:18		عدد ست اصوات متمائلين (six clicks)
02:37:23		صوت المحركات ( engine acceleration sound )
02:37:26	Captain	Flaps five
02:37:28		صوت عدد ثلاث خبطات ربما تكون صوت حركة ال-flap handle Three sounds similar to flap handle
02:37:30	Captain	Rudder right neutral left
02:37:34		صوت (high thump)
02:37:35	Captain	Neutral
02:37:37	First officer	Flight control checked
02:37:40		مجموعة أصوات متتالية Two consecutive sounds

UTC hh:mm:ss	Speaker	Content
02:37:41	Captain	Turning to the right
02:37:43	First officer	إن شاء الله via Delta <i>God willing via Delta commander</i>
02:37:44	Captain	مش هيه دي Delta <i>Is this Delta</i>
02:37:45	First officer	ان شاء الله <i>God willing</i>
02:37:49	First officer	Straight ahead
02:37:52		landing light صوت خبطة ربما تكون <i>Sound maybe landing light</i>
02:38:01	ATC	Flash Six Zero Four Ready to copy
02:38:03	First officer	Go ahead Sir
02:38:05	ATC	Flash Six Zero Four Destination Cairo as filed, climb initially flight level One Four Zero , One Six Seven Three on the Squawk
02:38:15	First officer	Our clear to destination Cairo via flight plan route One Four Zero initially, One Six Seven Three on the Squawk , Flash Six Zero Four and we have total Passengers One Three Five , <i>god willing</i> إن شاء الله
02:38:25	ATC	One Three Five and confirm Sierra Uniform Zulu Charlie Foxtrot
02:38:28	First officer	I do confirm
02:38:30	ATC	Continue taxi via Alpha line up Two Two Right advice ready for departure
02:38:34	First officer	Roger, next call ready <i>god willing</i> إن شاء الله

UTC hh:mm:ss	Speaker	Content
02:38:37	First officer	One four zero initially , one six seven three
02:38:44	Captain	Before takeoff
02:38:45	First officer	Recall
02:38:46	Captain	Checked
02:38:46	First officer	Flight Controls
02:38:47	Captain	Checked
02:38:48	First officer	Flaps
02:38:49	Captain	Five Green light
02:38:49	First officer	Stabilizer trim
02:38:51	Captain	Five units
02:38:52	First officer	Cockpit doors
02:38:54	Captain	Ok closed علشان الباب ده بيفتح Ok closed <i>because this door opens</i>
02:38:57	Extra crew1	عاوز ايه <i>what do you want</i>
02:38:57	Captain	أه علشان * ادى ليه * <i>Yes because * give why *</i>
02:38:58	Captain	لأ والله <i>No really</i>

UTC hh:mm:ss	Speaker	Content
02:39:01	First officer	Take off briefing
02:39:03	Captain	Standard briefing <i>god willing</i> انشاء الله
02:39:04	First officer	Before Check list completed down to the line <i>god willing</i> انشاء الله
02:39:12		(series of sounds) صوت خطبات
02:39:55	Captain	To the line
02:40:01	First officer	Engine start switches
02:40:02	Captain	On
02:40:02	First officer	Transponder
02:40:04	Captain	On
02:40:05	First officer	Before take off check list completed down to strobe lights
02:40:07	Captain	Completed <i>god willing</i> * إن شاء الله
02:40:36	Captain	Ready for departure حطها لى على الـ take off كده تسعين ونص <i>Set it on take off ninety and half ...ready for departure</i>
02:40:38	First officer	Flash Six Zero Four Ready for departure
02:40:46	ATC	Flash Six Zero Four Surface wind Two Eight Zero One Three knots left turn to intercept Radial Three Zero Six, clear for takeoff Two Two Right
02:40:55	First officer	Clear for takeoff runway Two Two Right with left turn to establish Three Zero Six Sharm VOR our Flash Six Zero Four clear for takeoff

UTC hh:mm:ss	Speaker	Content
02:41:01		One Thump (door knock )
02:41:02	Captain	مش كده left turn افتح لهم الباب <i>It is left turn.....open the door</i>
02:41:04	First officer	ان شاء الله <i>God willing</i>
02:41:09	Attendant	Cabin is Clear: المضيفة
02:41:12	Captain	شكراً <i>Thank you</i>
02:41:12	First officer	Final is clear
02:41:13		One thump
02:41:15		Four similar thumps may be landing lights
02:41:19	First officer	Left turn to establish radial Three Zero Six
02:41:29	Captain	Initially One Four Zero ?
02:41:30	First officer	إن شاء الله <i>God willing</i>
02:41:34	Captain	confirm initially One Four Zero
02:41:35	First officer	And Flash Six Zero Four Confirm to the left to establish Three Zero Six
02:41:40	Captain	Initial One Four Zero



UTC hh:mm:ss	Speaker	Content
02:41:43	ATC	إن شاء الله <i>God willing</i>
02:41:44	First officer	And initially One Four Zero
02:41:46	ATC	إن شاء الله <i>God willing</i>
02:41:48	Captain	توكلنا على الله <i>We rely on god</i>
02:41:59		Sound similar to increase of engine r.p.m
02:42:00	First officer	Stabilized sir N one
02:42:10	First officer	Takeoff power set speed building up, eighty knots, throttle hold
02:42:11	Captain	Eighty knots (one thump sound)
02:42:26	First officer	V one rotate
02:42:33		One thump sound similar to gear retraction
02:42:33.8	First officer	** Positive rate
02:42:34.6	Captain	Heading select
02:42:36	Captain	Gears up
02:42:36	First officer	Ok
02:42:43	Captain	Four Hundred Heading select
02:42:44	First officer	Four Hundred Heading select sir

UTC hh:mm:ss	Speaker	Content
02:42:48	Captain	Level Change
02:42:49	First officer	Level Change, MCP speed, N1 Armed sir
02:42:59	First officer	One Thousand
02:43:00	Captain	N one Speed Two twenty Flaps one
02:43:04	Captain	Left turn
02:43:05	ATC	Flash Six Zero Four airborne time Four Four when you ready to the left to intercept Three Zero Six radial report on course إن شاء الله <i>God willing</i>
02:43:11	Captain	Left turn
02:43:12	First officer	Roger when ready <i>god willing</i> إن شاء الله
02:43:18	First officer	Left turn to establish Three Zero Six Sharm V O R
02:43:19	MSR227	Sharm Egypt air two two seven <i>greeting</i> السلام عليكم
02:43:22	First officer	Speed available
02:43:23	Captain	Flaps up
02:43:23	ATC	Egypt air two two seven go ahead <i>greeting</i> وعليكم السلام ورحمة الله
02:43:26	MSR227	Maintaing flight level one two zero four three D M E in-bound to Sharm el Sheikh and request descent
02:43:34	ATC	Egypt air double two seven clear Sierra Hotel Mike V O R , visual approach runway two two right pilot discretion descend four thousand feet QNH one zero one one

UTC hh:mm:ss	Speaker	Content
02:43:35	First officer	Flaps up no light
02:43:37	Captain	After take off checklist
02:43:45	MSR227	هو حضرتك دلوقت الـ wind أد إيه <i>How much is the wind sir</i>
02:43:48	ATC	Indicated two eight zero one zero knots
02:43:53	MSR227	طب حضرتك ما نشغل runway zero four يا فندم <i>Can we use runway zero four sir</i>
02:43:55	Captain	Autopilot
02:43:56	MSR227	Right zero four
02:43:58	Captain	لسه <i>Not yet</i>
02:43:59	ATC	مفيش مشاكل Straight in ILS approach runway zero four left ان شاء الله report full establish QNH one zero one one <i>There is no problem Straight in ILS approach runway zero four left god willing report full establish QNH one zero one one</i>
02:44:00	First officer	Autopilot in command sir
02:44:01	Captain	اديله <i>Exclamation remark</i>
02:44:02		Sound of A/P disengage warning
02:44:05	Captain	Heading select
02:44:05	MSR227	Straight in approach runway zero four left, one zero one one , next call full establish Egypt air two two seven

UTC hh:mm:ss	Speaker	Content
02:44:07	First officer	Heading select
02:44:18	Captain	شوف الطياره عملت ايه <i>See what the aircraft did !</i>
02:44:27	First officer	Turning Right sir حضرتك
02:44:30	Captain	ايه <i>what</i>
02:44:31	First officer	Turning right الطياره <i>Aircraft is turning right</i>
02:44:32	Captain	أه <i>AH</i>
02:44:35	Captain	Turning right ?
02:44:37	Captain	Turning right ازای <i>How turning right</i>
02:44:41	Captain	Ok come out
02:44:41.5	First officer	Over bank
02:44:41.7	Captain	Autopilot
02:44:43.4	Captain	Autopilot
02:44:44	First officer	Autopilot in command
02:44:46	Captain	Autopilot

UTC hh:mm:ss	Speaker	Content
02:44:48	First Officer	tsk tsk
02:44:48	First Officer	Over bank, Over bank, Over bank
02:44:50	Captain	OK
02:44:52.8	First Officer	First Officer Over bank
02:44:53.4	Captain	OK
02:44:56	First Officer	Autopilot ماڤيش يا كوماندا <i>No autopilot commander</i>
02:44:58	Captain	Autopilot
02:44:58.8	Extra Crew 1	قل باور ، قل باور ، قل باور <i>Retard power , retard power , retard power</i>
02:45:01.5	Captain	<i>Retard power</i> قل باور
02:45:02		Sound similar to over speed clacker
02:45:04.3	Captain	Come out
02:45:05.9	First Officer	<i>No god except ....</i>
02:45:05	SV	“whoop” sound similar to ground proximity warning
02:45:06	End	End Of Recording

Exhibit C CVR Group Factual Report  
 Accident flight plan (copy of the flight plan referred to by ATC at  
 02:38:05 in the CVR transcript)



Aerodrome

FLIGHT PLAN PLAN DE VOL		مطار	
PRIORITY Priorite	ADDRESSEE(S) Destinataire(S)	برنامج رحلة	
FILING TIME Heure de depot	ORIGINATOR Expéditeur		
SPECIFIC IDENTIFICATION OF ADDRESSEE(S) AND/OR ORIGINATOR Identification Precise du(des) destinataire (s) et/ou de l'expéditeur			
3 MESSAGE TYPE Type de message	7 AIRCRAFT IDENTIFICATION Identification de l'aeronef	8 FLIGHT RULES Regles de vol	TYPE OF FLIGHT Type de vol
9 NUMBER Nombre	TYPE OF AIRCRAFT Type of aircraft	WAKE TURBULENCE CAT Cat de turbulence de de sillage	10 EQUIPMENT Equipement
13 DEPARTURE AERODROME Aerodrome de depart	TIME Heure		
15 CRUISING SPEED Vitesse croisiere	LEVEL Niveau	ROUTE Route	
16 DESTINATION AERODROME Aerodrome de destination	TOTAL EET Duree totale estimee HR. MIN	ALTN AERODROME Aerodrome de deplacement	2 ND ALTN AERODROME 2eme aerodrome de deplacement
18 OTHER INFORMATION Renseignements divers			
OPRI FLASH AIL REGI SURCF			
SUPPLEMENTARY INFORMATION (NOT TO BE TRANSMITTED IN FPL MESSAGES) Renseignements complementaires (a NE PAS TRANSMETTRE DANS LES MESSAGES DE PLAN DE VOL DEPOSE)			
19 ENDURANCE Autonomie HR. MIN	PERSONS ON BOARD Personnes a bord	EMERGENCY RADIO Radio de secours	
SURVIVAL EQUIPMENT/Equipement de survie		JACKET/CHEMISE de survie	
DINGHIES/Casos		LIGHT LAMPES	
AIRCRAFT COLOUR AND MARKINGS Couleur et marquages de l'aeronef		FLUORESCENT FLUORESC	
REMARKS Remarques		UHF VHF ELBA	
PILOT-IN-COMMAND Pilote commandant de bord		UHF VHF	
FILED BY/Depose par		Signature of Pilot-in command or delegated representative	
SPACE RESERVED FOR ADDITIONAL REQUIREMENTS			

## **Exhibit C      CVR Group Factual Report**

### **Spelling corrections**

Two spelling corrections should be made:

- The phrase "02:34:25 Attendant: "on behalf of Captain Kheder" (in page 269)
- should read "02:34:25 Attendant: "on behalf of Captain Khedr"
- The phrase "advice ready for departure" (in page 273) should read " advise ready for departure "

# **Exhibit D**

## **Airplane Performance Group Factual Report**



Ministry of civil aviation  
Accidents Department  
Egypt, Cairo

October14, 2004

**Group Chairman's Factual Report - Performance**

**A. ACCIDENT**

**Location:** Red Sea off Sharm el-Sheikh  
**Date:** January3, 2004  
**Time:** 2:45:06 GMT  
**Operator:** Flash Airlines – Flight 604

The group convened at MCA headquarters in Cairo from January15, 2004 for performance Factual Data collection

**B. SUMMARY**

On January 3, 2004, about 02:45:06 UTC, 04:45:06 Local time, Flash Airlines flight FSH604, a Boeing 737-300, Egyptian registration SU-ZCF, crashed into the Red Sea shortly after takeoff from Sharm el-Sheikh International Airport (SSH) in South Sinai, Egypt. The flight was a passenger charter flight to Charles de Gaulle Airport (CDG), France with a stopover in Cairo international Airport (CAI) for refueling. Flight 604 departed from Sharm el-Sheikh airport with 2 pilots (Captain and First Officer), 1 observer, 4 cabin crew, 6 off-duty crew members and 135 passengers on board. The airplane was destroyed due to impact forces with the Red Sea with no survivals.

The airplane had departed from Sharm el-Sheikh runway 22R and was air born at 02:42:33 UTC, approximately 2½ minutes prior to the crash, and had been cleared for a climbing left turn intercept the 306 radial from the Sharm el-Sheikh VOR station located just north of runway 22R. This climbing turn allows departing flights to gain sufficient altitude before proceeding over higher terrain located along the flight path to Cairo. Flight 604 was operating in Egyptian airspace as a charter flight operating under the provisions of Egyptian Civil Aviation Regulations Part 121

**C. DETAILS OF THE INVESTIGATION**

The purpose of the Aircraft Performance Group (ACPG) is to collect the factual information to determine and analyze the motion of the aircraft and the physical forces that produce that motion. In particular, the Group attempts to define the aircraft position and orientation throughout the flight, and determine its response to control inputs, system failures, external disturbances, or other factors that could affect its trajectory. The data the ACPG uses to obtain this information includes but is not limited to the following:

- Wreckage location and condition.
- Aircraft Surveillance Radar (ASR 12) Radar Data.
- Digital Flight Data Recorder (DFDR) data.
- Cockpit Voice Recorder (CVR) information.
- Weather information.
- Weight and Balance Data.
- Tests and Researches

**C.1 Wreckage Location and Condition:**

Refer to the Wreckage and Impact Factual Information

**C.2 Radar Data**

*Sharm el-Sheikh Radar*

- General Specifications:

ASR 12 Radar (Aircraft Surveillance Radar)

Secondary 250 nm

Primary 60 nm

15 Revolution Per minutes approximately (Scan time = 4.13 sec)

Radar site location: 2758.057n/ 03421.985e (Lat. 27.96762 Degree north, Long. 34.36642 Degree east)

Radar Elevation: 299.3 ft

- Radar data of accident flight

Ref Time 0 seconds at 02-44-00	Time	Flight Level	Target	Code	Target lat. Degree North	Target long. Degree East
27	02-44-27		275831n0342325e		27.971833	34.3875
29	02-44-29		275828n0342322e		27.971333	34.387
33	02-44-33		275816n0342306e		27.969333	34.384333
37	02-44-37		275808n0342257e		27.968	34.376167
41	02-44-41		275751n0342256e	airborn	27.9585	34.376
45	02-44-45	6	275751n0342256e	a	27.9585	34.376
49	02-44-49	10	275731n0342238e	a	27.955167	34.373
53	02-44-53	10	275721n0342231e	a	27.9535	34.371833
57	02-44-57	11	275711n0342221e	a	27.951833	34.370167
61	02-45-01	13	275700n0342209e	a	27.95	34.368167
65	02-45-05	15	275646n0342203e	a	27.941	34.367167
69	02-45-09	17	275621n0342208e	a	27.936833	34.368
73	02-45-13	17	275623n0342150e	a	27.937167	34.358333

77	02-45-17	18	275613n0342154e	a	27.9355	34.359
81	02-45-21	18	275605n0342154e	a	27.934167	34.359
85	02-45-25	20	275537n0342157e	a	27.922833	34.3595
89	02-45-29	21	275556n0342203e	a	27.926	34.367167
93	02-45-33	23	275509n0342211e	a	27.918167	34.3685
97	02-45-37	25	275501n0342219e	a	27.916833	34.369833
101	02-45-41	27	275442n0342220e	a	27.907	34.37
105	02-45-45	30	275431n0342237e	a	27.905167	34.372833
109	02-45-49	36	275412n0342243e	a	27.902	34.373833
113	02-45-53	36	275414n0342256e	a	27.902333	34.376
117	02-45-57	39	275353n0342307e	a	27.892167	34.3845
121	02-46-01	42	275340n0342315e	a	27.89	34.385833
125	02-46-05	44	275330n0342320e	a	27.888333	34.386667
129	02-46-09	47	275325n0342329e	a	27.8875	34.388167
133	02-46-13	50	275309n0342337e	a	27.884833	34.3895
137	02-46-17	50	275254n0342341e	a	27.875667	34.390167
141	02-46-21	51	275252n0342340e	a	27.875333	34.39
145	02-46-25	51	275228n0342346e	a	27.871333	34.391
149	02-46-29	53	275220n0342345e	a	27.87	34.390833
153	02-46-33	52	275202n0342336e	a	27.867	34.389333
157	02-46-37	51	275144n0342317e	a	27.857333	34.386167
159	02-46-39	46	275156n0342325e	a	27.859333	34.3875
161	02-46-41	46	275139n0342320e	a	27.8565	34.386667
165	02-46-45	46	275141n0342248e	a	27.856833	34.374667
167	02-46-47	46	275159n0342236e	n	27.859833	34.372667
169	02-46-49	46	275201n0342227e	n	27.866833	34.371167
173	02-46-53	46	275208n0342207e	n	27.868	34.367833
177	02-46-57	46	275222n0342153e	n	27.870333	34.358833
181	02-47-01	46	275231n0342143e	n	27.871833	34.357167
185	02-47-05	46	275242n0342115e	n	27.873667	34.3525
189	02-47-09	46	275255n0342100e	n ----	27.875833	34.35
				missing SSR code		
191	02-47-13		275307n0342037e	missing beacon	27.8845	34.3395
207	02-47-27		275319n0342032e	Disappear ance	27.8865	34.338667

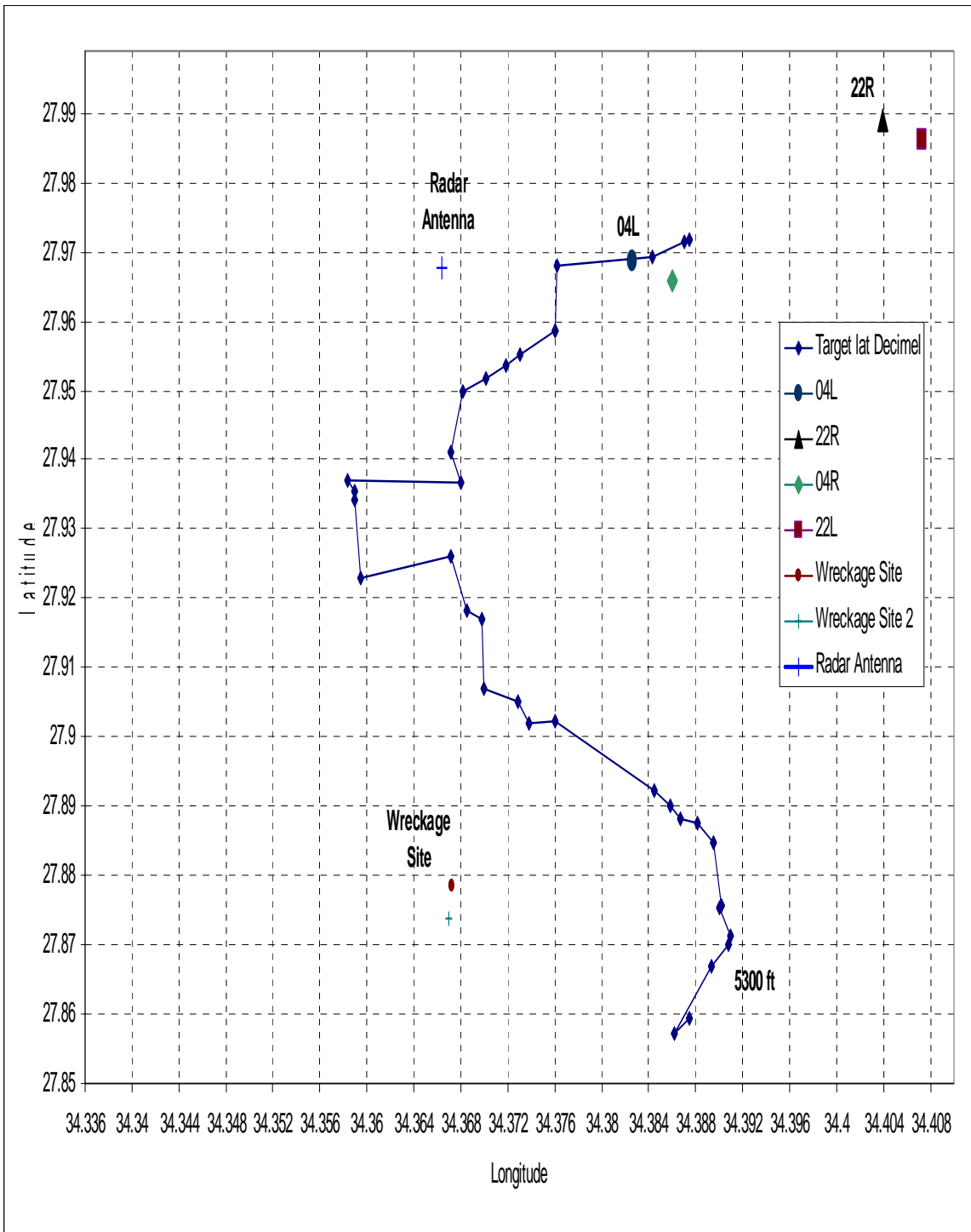


Figure C.2-1 Radar Data Plot, Sharm El Sheik Radar

*Hurgada Radar*

- General Specifications:

Radar site location: 2711.546N/03346.814E (Lat. 27.19243333 Degree north,  
Long. 33.78023 Degree east)

Radar Elevation: 176.344 ft

- Radar data of accident flight:

Ref Time	Time	Events & Altitude	Coordinates	Code	Target lat. Degree North	Target long. Degree East
0	seconds					
	at 02-44-00					
51	02 44 51	Initial Detection	275723N0342239E		34.37316667	27.95383333
53	02 44 53		275721N0342241E		34.3735	27.9535
57	02 44 57		275722N0342239E		34.37316667	27.95366667
61	02 45 01		275722N0342237E		34.37283333	27.95366667
65	02 45 05		275723N0342238E		34.373	27.95383333
69	02 45 09		275640N0342206E		34.36766667	27.94
72	02 45 12	1900ft	275616N0342159E	c	34.35983333	27.936
73	02 45 13	2000ft	275613N0342157E	c	34.3595	27.9355
77	02 45 17	2000ft	275605N0342150E	c	34.35833333	27.93416667
81	02 45 21	2100ft	275546N0342153E	c	34.35883333	27.92433333
85	02 45 25	2200ft	275538N0342159E	c	34.35983333	27.923
89	02 45 29	2300ft	275517N0342211E	c	34.3685	27.9195
93	02 45 33	2500ft	275506N0342213E	c	34.36883333	27.91766667
97	02 45 37	2700ft	275447N0342225E	c	34.37083333	27.90783333
101	02 45 41	2900ft	275434N0342231E	c	34.37183333	27.90566667
105	02 45 45	3200ft	275425N0342239E	c	34.37316667	27.90416667
109	02 45 49	3500ft	275407N0342246E	c	34.37433333	27.90116667
113	02 45 53	3800ft	275357N0342254E	c	34.37566667	27.89283333
117	02 45 57	4100ft	275345N0342304E	c	34.384	27.89083333
121	02 46 01	4300ft	275330N0342315E	a	34.38583333	27.88833333
125	02 46 05	4600ft	275328N0342318E	a	34.38633333	27.888
129	02 46 09	4900ft	275311N0342333E	a	34.38883333	27.88516667
133	02 46 13	5000ft	275257N0342341E	a	34.39016667	27.87616667
137	02 46 17	5100ft	275249N0342342E	a	34.39033333	27.87483333
141	02 46 21	5300ft	275232N0342353E	a	34.39216667	27.872
145	02 46 25	5300ft	275223N0342403E	a	34.4005	27.8705
148	02 46 28	Max. Alt.	275205N0342345E	a	34.39083333	27.8675

		5400ft				
149	02 46 29	5400ft	275206N0342357E	a	34.39283333	27.86766667
153	02 46 33	5300ft	275149N0342334E	a	34.389	27.85816667
157	02 46 37	5100ft	275143N0342317E	a	34.38616667	27.85716667
161	02 46 41	Descending 4600ft	275129N0342307E	a	34.3845	27.85483333
165	02 46 45	Still 4600ft	275136N0342254E	a	34.37566667	27.856
168	02 46 48	Still 4600ft	275123N0342234E	n	34.37233333	27.85383333
169	02 46 49	Still 4600ft	275125N0342235E	n	34.3725	27.85416667
173	02 46 53	Still 4600ft	275203N0342214E	n	34.369	27.86716667
177	02 46 57	Still 4600ft	275206N0342153E	n	34.35883333	27.86766667
181	02 47 01	Still 4600ft	275208N0342143E	n	34.35716667	27.868
185	02 47 05	Still 4600ft	275212N0342119E	n	34.35316667	27.86866667
188	02 47 08	Missing SSR&Still 4600ft	275213N0342105E	n	34.35083333	27.86883333

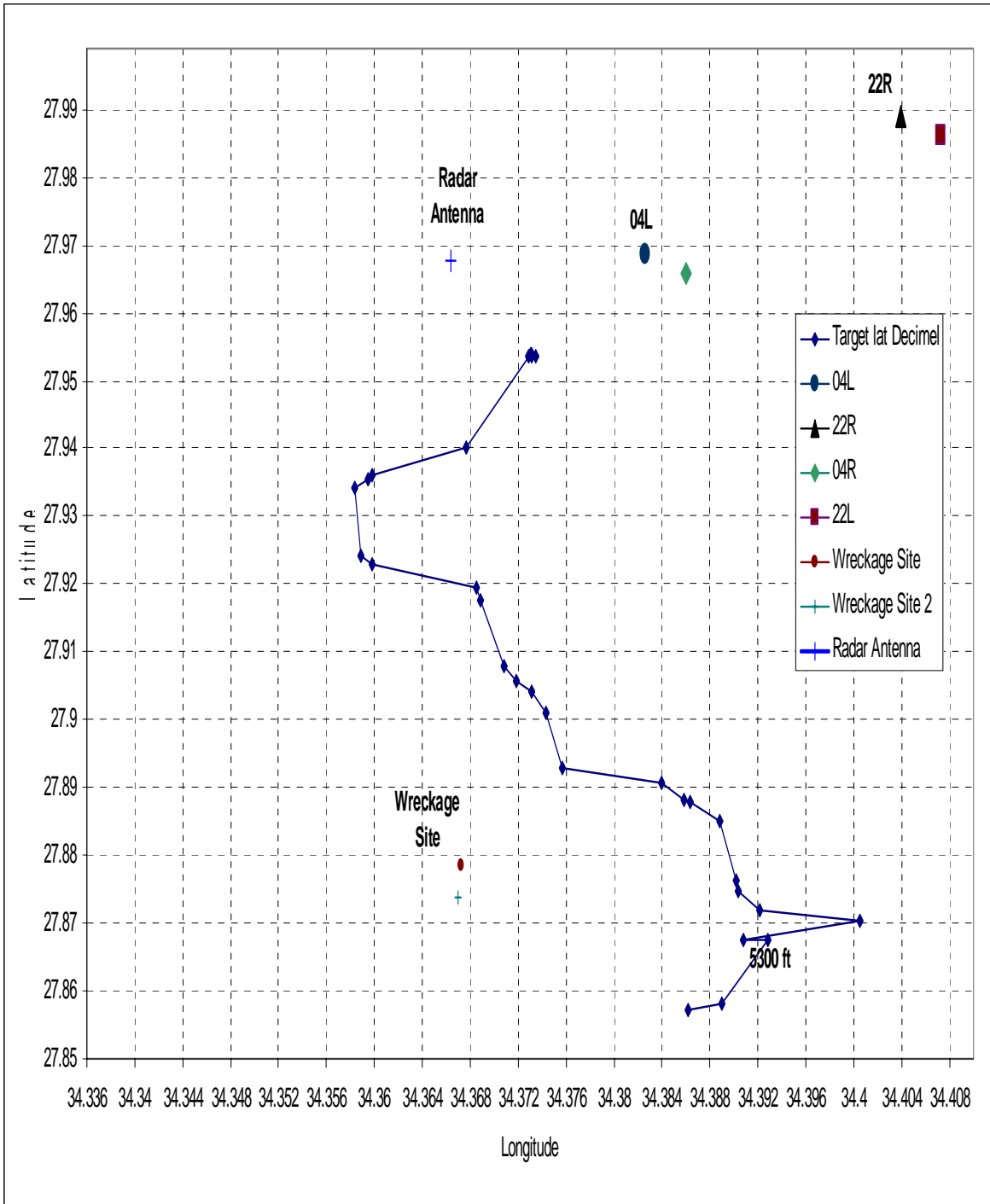


Figure C.2-2 Radar Data Plot, Hurgada Radar

## ASR 12 Radar (Aircraft Surveillance Radar) Specifications :

Secondary 250 nm

Primary 60 nm

15 Revolution Per minutes approximately (Scan time = 4.13 sec)

<u>Field</u>	<u>Valid Field Variables</u>	<u>Data Field Description</u>
1	A-Z, 0-9	Aircraft flight identifier or callsign
2	#, *, +, or blank	Special processing indicator: # = track is inhibited from CA processing, either with another specified track or with all other tracks * = track is inhibited from MSAW processing + = track is inhibited from both CA and MSAW processing blank = track is subject to both CA and MSAW processing
3	H, M, or L	Aircraft wake indicator: H = heavy M = medium L = light
4	000-999 or ••••	Cleared level: NNN= assigned altitude in hundreds of feet •••• = altitude unavailable or less than sea level
5	T, ↑, ↓, or blank	Cleared level qualifier: T = temporary altitude ↑ = vertical movement of track - climbing ↓ = vertical movement of track - descending blank= permanent cleared level
6	000-999 or ••••	Reported altitude: NNN= reported altitude in hundreds of feet 999 = altitude greater than 99,900 feet •••• = altitude unavailable, altitude less than sea level or altitude has not been updated for approximately 15 seconds
7	a, C, E, e, N, n, or blank	Altitude transition indicator:



- a = indicates altitude source is mode C, aircraft is below adapted transition level and altitude is in hundreds of feet above mean sea level
- C = indicates altitude source is mode C, aircraft is above adapted transition level and altitude is in flight levels
- E = indicates altitude source is manually entered, aircraft is above adapted transition level and altitude is in flight levels
- e = indicates altitude source is manually entered, aircraft is below adapted transition level and altitude is in hundreds of feet above mean sea level

<u>Field</u>	<u>Valid Field Variables</u>	<u>Data Field Description</u>
7 (Cont.)	a, C, E, e, N, n, or blank	Altitude transition indicator: N = indicates mode C altitude has not been updated for approximately 7.5 seconds and is considered unreliable, aircraft is above adapted transition level and altitude is in flight levels n = indicates mode C altitude has not been updated for approximately 7.5 seconds and is considered unreliable, aircraft is below adapted transition level and altitude is in hundreds of feet above mean sea level blank= no data is available or altitude data has not been manually entered
8	0000-7777 (octal)	Reported code
9	0000-9999	Track ground speed in knots
10	0000-7777 (octal)	Assigned code
11	A-Z, 0-9	Aircraft type (field is blank for manually created sim tracks)
12	A-Z, 0-9	Destination aerodrome or last adapted point on flight plan route (XXXX)
13	A-Z, 0-9	Scratch pad note entered by controlling operator (XXXXXX)

**C3. Digital Flight Data Recorder (DFDR) data.**

Refer to FDR Factual Report

**C4. Cockpit Voice Recorder (CVR) information.**

Refer to FDR Factual Report

**C5. Weather Information**

Sharm El Sheikh does not provide Automatic Terminal Information Service (ATIS).

The SSH weather at 0200Z was reported as:

270 degrees at 06 knots, ceiling and visibility OK (CAVOK), temperature 17 degrees Celsius, dew point minus 6 degree Celsius, altimeter 1011 HectoPascals (hPa), No significant change (NOSIG).

The SSH weather at 0300Z was reported as:

280 degrees at 08 knots, ceiling and visibility OK (CAVOK), temperature 17 degrees Celsius, dew point minus 6 degree Celsius, altimeter 1011 HectoPascals (hPa), No significant change (NOSIG).

## C6. Weight and Balance Data.

According to the Egyptian Civil Aviation Regulations, ECAR 91 Appendix H attachment 1 the aircraft has to be reweighed every three years. Furthermore, aircraft must be reweighed if the effect of modifications on the mass and balance is not accurately known. Flash Airlines aircraft was weighed last time on December 19, 2002 in Braathens SAFE, Stavanger, Norway and recalculated by Flash Airlines after the reinforced cockpit door modification installation on November 1<sup>st</sup>, 2003, and the results were as follows.

Empty Weight	:	70794 lbs
Moment	:	45921358.6 lb.in
% AMC	:	17.42%

The Flash Airlines weight and balance calculations provided to the flight crew contained the following information<sup>1</sup>:

	Weight (kilograms)	
Total Traffic Load	11,450 <sup>2</sup>	
Dry Operating Mass	33,200	
Actual Zero Fuel Mass	44,650	
Maximum Zero Fuel Mass	47,627	
Takeoff Fuel	7,000	
Actual Takeoff Mass	51,650	
Maximum Takeoff Mass (Certificate Limi	63,276	
Landing Mass	49,650	
Maximum Landing Mass (Certificate Limi	51,709	

Zero Fuel Mass Center of Gravity (CG)	20.0%	
Zero Fuel Mass CG Limits <sup>3</sup>	8.0% Forward	28.4% Aft
Takeoff Mass CG	18.0%	
Takeoff Mass CG Limits <sup>4</sup>	6.7% Forward	27.9% Aft

<sup>1</sup> See attached Flash Airlines Load and Trim Sheet.

<sup>2</sup> A review of the Load and Trim Sheet indicated a low 100-kilogram error. The total cargo weight plus passenger mass (Total Traffic Load) should be 11,550 kilograms. Correspondingly, the Zero Fuel Mass, Takeoff Mass, and Landing Mass will be low in error by the same 100-kilogram Mass.

<sup>3</sup> Estimated Zero Fuel Mass CG limits were derived from Flash Airlines Load and Trim sheet index chart based upon a Zero Fuel Mass of 44,650 kilograms.

Stabilizer Trim settings for takeoff were:

Flaps 1 or 5            4 ¾ Units  
Flaps 15                3 ¾ Units

According to the Flash Airlines Flight Operations Manual Chapter 6, Paragraph 6.1.8.3, Passenger and Baggage Masses, the following chart was published:

	Male	Female
All flights except	88kg	70kg
Holiday	83kg	69kg
Children	35kg	35kg

A review of the accident Load and Trim Sheet indicated a Passenger Mass of 9,450kg. If 350kg is removed for 10 children (10 x 35kg) the result is 9,100kg. Dividing the 125 adult passengers into the 9,100kg would give an average value of 72.8kg per adult passenger.

Using the table above, and assuming 50% Male and 50% Female adult passengers, the worst-case difference in weight calculation would be the following:

The average weight of male and female for all flights except would be  $88\text{kg} + 70\text{kg} / 2 = 79\text{kg}$  per adult passenger.

$$79\text{kg} \times 125 \text{ passengers} = 9,875\text{kg}$$

This represents an increase in weight of 775kg.

Using this value for Load and Trim calculations provided the following information:

Takeoff CG            18.2%MAC  
Zero Fuel Mass CG   20% MAC  
Takeoff Trim (flaps 5) 4 ¾ Units

These worst-case differences in values for passenger weight still fall within structural and calculated limitations for the airplane.

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<sup>4</sup> Estimated Takeoff Mass CG limits were derived from Flash Airlines Load and Trim sheet index chart based upon a Takeoff Mass of 51,650 kilograms.

DRY OPERATING MASS	33200	MAXIMUM MASSES FOR	ZERO FUEL	TAKEOFF	LANDING
Takeoff Fuel	7000		47027	51103	2090
OPERATING MASS	40200	Allowed Mass for Takeoff Lowest of a.b.c	44650	51700	40200
Notes: T/C PAx 135 PCs 136		Allowed Traffic Load		13500	
TOTAL PSGR OVB		Total Traffic Load		11450	
		UNDERLOAD		2059	

Dest	No of PSGR	TOTAL	FWD CARGO	AFT CARGO	Dest	No of PSGR	TOTAL	FWD CARGO 1	AFT CARGO 2
	AD(s) CH I					AD(s) CH I			
D	125 10 0	2100	700	1400					
E									
F									
G									
TOTAL		2100							

Passenger Mass	9450
TOTAL TRAFFIC LOAD	11450
Dry Operating Mass	33200
ZERO FUEL MASS Max	44650
Takeoff Fuel	7000
TAKEOFF MASS	51650
Trip Fuel	2090
LANDING MASS Max	49650

BALANCE ARM - INCHES	200	300	400	500	600	700	800	900	1000	1100
FWD CABIN	MID CABIN		AFT CABIN		FWD HOLD		AFT HOLD			

**ALL MASSES IN KILOGRAMS**

AREA	APS INDEX	PAYLOAD	UNIT
FWD HOLD	700	200 KG	
AFT HOLD	1400	200 KG	
FWD CABIN	45	4 PASS	
MID CABIN	45	NO INDEX CHANGE	
AFT CABIN	45	4 PASS	

FUEL INDEX	FLAPS 1 & 5	FLAPS 15	% MAC
6 1/4	5 3/4	6	6
5 1/4	5 1/4	5	8
5 1/2	5	4 3/4	10
5 1/4	4 1/2	4 1/2	12
5	4	4	14
4 1/2	3 3/4	3 3/4	16
4 1/2	3 1/4	3	18
4 1/4	3	2 3/4	20
4	2 3/4	2 1/2	22
3 3/4	2 1/2	2	24
3 3/4	2	1 3/4	26
3 1/4	1 3/4	1 3/4	28
3	1 3/4	1 3/4	30

Prepared by	AMR
Approved by	

## **C7. Tests and Research**

The FDR records the movements of the pilot's controls (e.g. control column, control wheel position and rudder pedals), the movement of the control surfaces (e.g. elevator, aileron and rudder) as well as motion of the airplane (e.g. pitch and roll attitude and heading angle). The performance evaluation was conducted to determine if the control surfaces were responding normally to the pilot's controls and if the airplane was responding normally to movement of the control surfaces.

In order to accomplish this work, Boeing's 737-300 aerodynamic simulation model was used to recreate the accident flight. The simulation calculates the response of the airplane to movement of the flight control surfaces – for example, it can calculate the roll rate resulting from a 10 degree deflection of the ailerons. The simulation has been verified by comparison against actual flight test data and was used for the design and certification of the 737-300 airplane. In addition, the simulation is the basis for 737-300 crew training simulators used around the world. It should be noted that the 737-300 simulation model is essentially a computer program that represents a nominal airplane with nominal engines. Small differences between the simulation and individual airplane's behavior are common and expected due to differences in control surface rigging, engine wear, and other normal tolerances.

### **Performance Evaluation**

FDR data are recorded at relatively low sample rates and are recorded from different sources, some of which have inherent biases. Because of these issues, a kinematic consistency (KINCON) process was used to supplement the FDR data and calculate additional parameters to be used in the performance analysis. Kinematic consistency analysis is a general practice for processing flight data (either flight test data or FDR data) to ensure consistency of position, speed, and acceleration data.

### **C7.1 Baseline Simulation**

A baseline simulation recreation of the accident flight was started just as the airplane turned onto the runway and the throttles were advanced, and the simulation was stopped at the end of the FDR data. Because the simulation can calculate the response of the airplane to control inputs, a set of control input time histories (column, wheel, and rudder movements) can be determined that results in the simulation following the same path as the accident airplane. It is important to note that this process does not use the control or surface position data recorded on the FDR, only the path information (e.g. accelerations, attitude and altitude).

Comparisons between the recorded FDR data and the simulation time history data are provided for longitudinal and lateral/directional data in Figures Figure C7-1 and Figure C7-2 respectively.

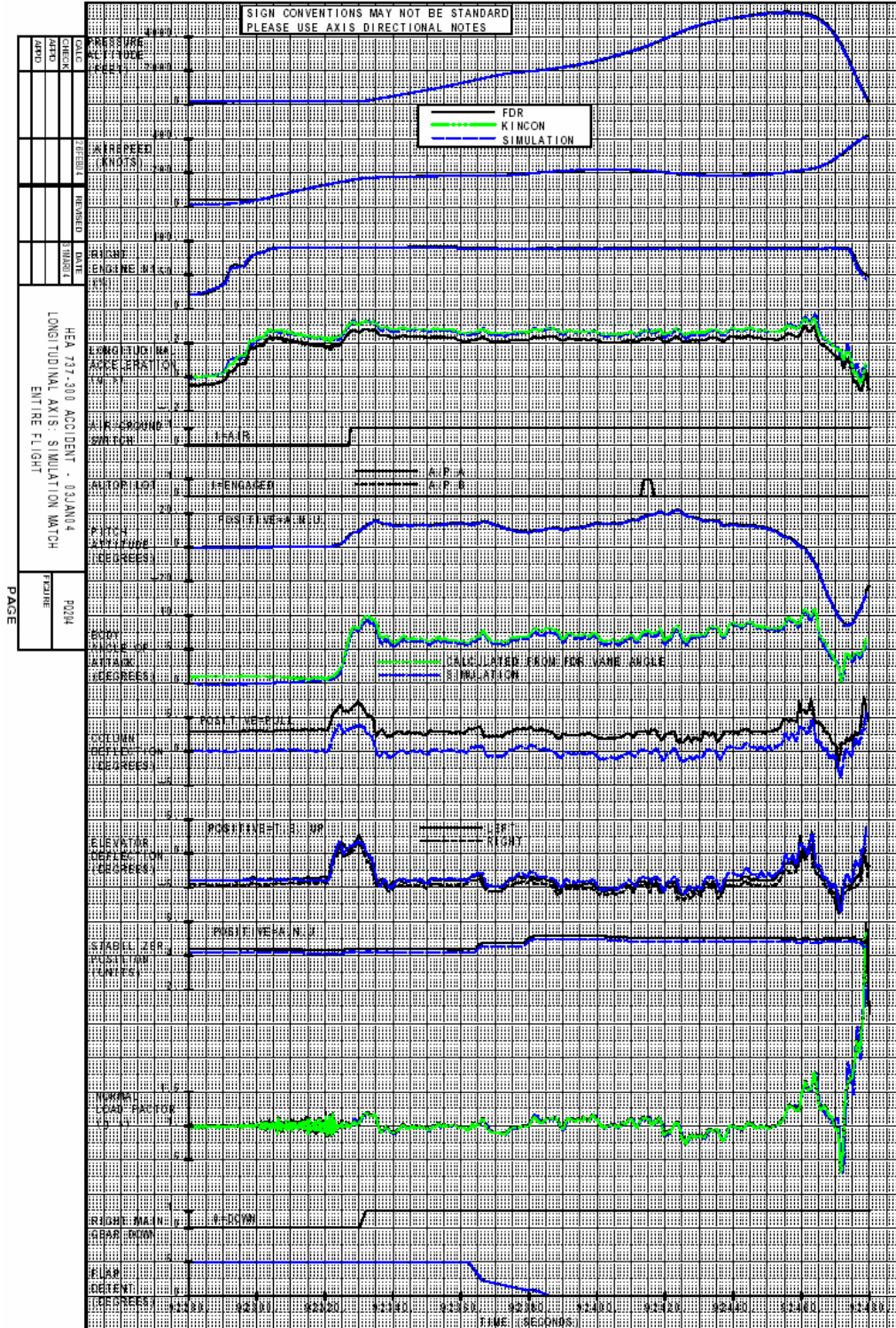


Figure C7-1 – FDR and Simulation Match Data – Longitudinal Axis





An examination of the baseline simulation revealed that the path of the accident airplane is consistent with the recorded motion of the control surfaces. Specifically, the extreme bank attitude that occurs towards the end of the flight is consistent with recorded motion of the ailerons.

The simulation also revealed that the motion of the control surfaces is consistent with the recorded motion of the control inputs, with the exception of control wheel

#### C7.2 Hypothetical Faults resulting in a rolling moment

Several hypothetical airplane system faults were examined to determine if any could have resulted in the right roll behavior recorded on the FDR. These faults included:

- Uncommanded deployment of the #1 slat
- Uncommanded spoiler deflection to full travel (hardover)
- A spoiler disconnected from its actuator (spoiler float)
- Flap asymmetry
- Thrust asymmetry
- Unrecorded rudder motion

The hypothetical faults listed above are similar in that they each create a rolling moment unrelated to the position of the ailerons that will cause the airplane to bank. That is to say, if one of these faults had occurred, the path of the airplane would have differed from that predicted by the recorded position of the ailerons.

#### Multi-Purpose Engineering Cab Simulator

Additional tests were conducted at Boeing's multi-purpose engineering cab simulator or M-Cab. The M-Cab is similar to a flight crew training simulator in that it consists of a realistic flight deck mounted on a movable base. The M-Cab includes a visual system providing out-the-window views to the flight crew. Because the M-Cab is used to simulate the flight deck of many different Boeing models, actual flight instruments are not used. Instead, a large LCD display is programmed to simulate the flight instrument displays. Examples of the M-Cab's flight instrument displays for the 737-300 are shown in section 1.6.2.

Major differences between the M-Cab and a typical flight crew training simulator are listed below.

- The M-Cab can simulate different model airplanes including 707, 727, 737, 747, 757, 767, and 777.
- The M-Cab can be reprogrammed to simulate a wide variety of hypothetical aircraft system faults.
- The M-Cab can be "backdriven" to reproduce recorded data, such as the simulation match to the accident flight discussed in section 1.16.2. In addition, the backdrive can be interrupted at any point with a transition to normal simulator operation at the current flight conditions. This capability (known as "breakout" allows pilots in the simulator to attempt to recover the airplane from various points in the accident profile.
- The operation of the M-Cab is recorded at a high sample rate

The M-Cab was used to recreate the accident flight as well as to study a number of hypothetical airplane system faults.

## Tests conducted in the M-Cab

The M-Cab was used to examine some of the faults mentioned in section 1.16.3, as well as a number of other hypothetical faults affecting the lateral control system or the autopilot system. M-Cab tests included:

- Backdrive of FDR data
- Backdrive with breakout at 02:44:44
- Backdrive with breakout at 02:44:56
- Spoiler float
- Uncommanded aileron trim to full authority
- Uncommanded aileron trim to half authority
- Autopilot servo actuator hardover without force limiter engaged
- Autopilot servo actuator hardover with force limiter engaged
- Autopilot servo actuator hardover with pressure regulator and relief valve inoperative

The tests in the M-Cab were conducted with an out-the-window scene equivalent to that available to the accident pilots with the following exceptions:

- 1) The visibility conditions simulated (ceiling and visibility unlimited at night with no moon) were those reported at the airport at the time of the accident. Actual visibility conditions on the flight deck at the time of the accident are unknown.
- 2) The ground in the vicinity of Sharm el-Sheikh was depicted through the use of satellite photography taken during daylight hours. It did not represent the nighttime scene of street lights, building lights, etc. against an otherwise dark landscape.

# **Exhibit E**

## **Site and Wreckage Group Factual Report**

## Site and Wreckage Group Report

### 1. Summary of the Accident

On 3 January 2004, Flash Airlines flight FSH604, a Boeing 737-300 registered as SU-ZCF, operating as a chartered flight from Sharm el-Sheikh, Egypt to Paris, France, via Cairo departed from Sharm el-Sheikh airport (SSH) at approximately 02:40 UTC. The airplane crashed into the Red Sea approximately 6 nautical miles southwest of the airport at approximately 02:44 UTC.



### 2. Scope of Site and Wreckage Group Field Notes

The scope of this report is the recovery operations that took place from 3 January 2004 through 28 January 2004 in the Red Sea off Sharm el-Sheikh, Egypt and initial inspection for the recovered parts. Recovery operations initially consisted of the recovery of floating wreckage elements only. Recovery of the underwater wreckage (including FDR and CVR) began when the first ship equipped with a suitable Remote Operated Vehicle (ROV), arrived at the accident scene on 11 January 2004.

This report provides a summary of the recovery operations and documents the wreckage that was identified and recovered.

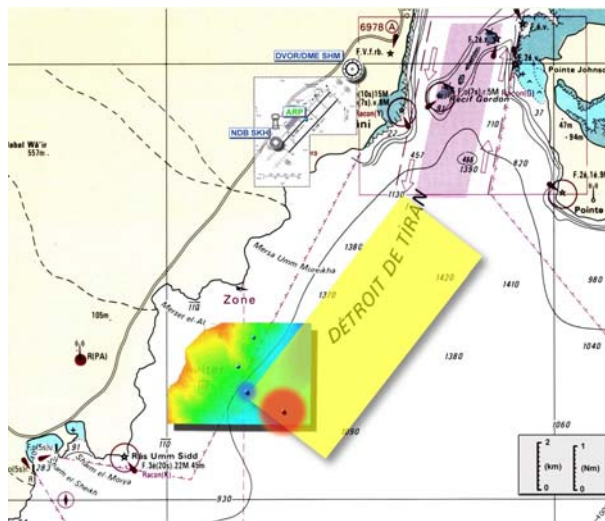
### 3. Recovery Operations

#### 3.1 Survival aspects

The initial search for possible survivors and the recovery of bodies were priorities for the rescue and investigation teams. Rescue teams were on site minutes after the accident. They searched for survivors but due to the high energy impact of the aircraft with the sea surface, the depth of the water in this area, their efforts were unsuccessful in recovering any survivors.

Efforts were made to locate human remains by use of deep sea cameras and robots but were also not successful due to the location of the wreckage and the depth of more than 1000 meters.

### 3.2 Floating Wreckage



The floating wreckage which was recovered shortly after the crash was stored in a hangar in Sharm el-Sheikh airport. On 11 January 2004, the Site and Recovery Group met in the hangar for wreckage inspection. The wreckage was then identified (as much as possible), inspected, segregated (aircraft parts or personal effects). Later, the personal effects were transferred to the Egyptian Legal Authority in Sharm el-Sheikh. A database for the floating wreckage was created (including wreckage pictures).

### 3.3 Underwater Wreckage

Because of the depth of the Red Sea in the area where the accident occurred (approximately 1000 meters), specialized recovery resources were required for the submerged wreckage. The French vessels "Ile de Batz" and "Janus II" were contracted to conduct the underwater wreckage survey and recovery. Both vessels were equipped with deep water recovery capabilities consisting of submersible Remotely Operated Vehicles (ROV). The necessary support equipment to accurately locate and map the airplane wreckage was provided by the French Navy. An oceanographic vessel, the "Beautemps-Beaupré" was sent to the accident site to undertake a bathymetry (depth mapping) of the seabed and a survey of tidal currents.



### 3.4 FDR / CVR Recovery

The initial focus of the underwater recovery operation was finding and retrieving the protected recorders, the Cockpit Voice Recorder (CVR) and Flight Data Recorder (FDR) and mapping the searched areas. Each recorder is equipped with an acoustic transmitter, called a “pinger” that transmits a detection signal that can be used to locate the box. Based on the initial determination of pinger locations, the ROV from Ile de- Batz, Scorpio, began a visual search using its cameras to find the recorders. To refine the location of the pingers, a network of sonobuoys (GIB, GPS Intelligent Buoys), (see Exhibit E Attachment 4 for detailed description of this operation), was employed in a cooperative effort between the French and Egyptian Navies. This method produced a new pinger position accurate to within 10 meters and the ROV was moved to the new location. A visual search of a grid created around the new pinger location resulted in discovery of the FDR on 16 January 2004. The FDR was recovered by the ROV and taken onboard the Ile de Batz. Custody of the recorder was transferred to the Investigator in Charge (IIC) at the port of Sharm El Sheikh.

The pinger of the second recorder (CVR) was initially identified approximately 800 meters north of the first pinger. However, it was decided to continue the visual search using grids in the area where the first recorder was found. This search was successful and resulted in finding of the CVR on 17 January 2004 (approximately 24 hours after the FDR). It was also taken onboard the Ile de Batz and custody was transferred to the Investigator in Charge (IIC) at the port of Sharm El Sheikh.

FDR underwater Location: N27 52.3605, E34 22.0165.

CVR underwater Location: N27 52.3467, E34 22.0207.

The recorders were both sent to Cairo for read out and analysis.

The focus of the recovery operation then changed to detailed mapping of the wreckage and recovery of selected airplane equipment. In addition, the recovery operation included recovery of any equipment deemed important to the investigation based on the review of the FDR and CVR in Cairo.

### 3.5 Wreckage Mapping

During the structured search for the recorders, the position (latitude and longitude) and description of surveyed wreckage was recorded. Following recovery of the FDR and CVR, additional grids were defined for ROV operations. These grids were used to systematically survey and document the entire wreckage area. The positions of large pieces, such as the three landing gears and the cores of the two engines were identified.

Data from both ships involved in mapping and recovery were consolidated into a single listing of all surveyed wreckage, which is included herein as Exhibit E Attachment 5.

The distribution of wreckage is included within a rectangle of approximately 275 by 440 meters defined by the following corner point coordinates:

North corner:	N 27°52,559	E 34°21,933
East corner:	N 27°52,410	E 34°22,126
South corner:	N 27°52,294	E 34°22,022
West corner:	N 27°52,450	E 34°21,817

Multiple surveys of the area confirmed the containment of the wreckage within these established boundaries.

### 3.6 Recovered Wreckage

The investigation team developed a strategy for wreckage recovery based on the review of the FDR and CVR undertaken in Cairo. Flight control actuation components and flight deck systems were considered as a priority.

A system was developed for recording the description, external dimensions and the location, in latitude and longitude coordinates, of all recovered wreckage pieces. A database of recovered floating wreckage is included herein as Exhibit E Attachment 5. Another database documenting all wreckage recovered by Ile de Batz and Janus II is included as Exhibit E Attachment 5. Both databases reference digital images of all floating and recovered wreckage.

Recovered wreckage was stored aboard the ships in sea water until taken ashore and loaded onto trucks. All of the recovered wreckage is stored in a hangar at Sharm El Sheikh Airport and is under the control of the investigative authorities.

## 4. Partial list of the Recovered Wreckage

- Parts of the horizontal stabilizer central section structure (called "Texas Star"), elements of the elevator structure and components of the elevator control system, including both elevator PCU's (Power Control Unit), both autopilot actuators, the feel and centering unit including the feel actuator.
- Horizontal stabilizer jackscrew and actuator gearbox.



- Vertical stabilizer structure with rudder control system components, including the main rudder PCU and standby rudder PCU, the feel and centering mechanism and with the trim actuator.
- Aileron PCU, spoiler mixer and TBD spoiler actuators.

## 5. Initial observations

- The two engines were found approximately 24 meters apart
- The left and right main landing gear assemblies were found in between the two engines
- The recovered thrust reverser actuator was found retracted
- The recovered leading edge flap actuator was found retracted
- The recovered trailing edge flap jackscrew indicates that flaps were retracted
- The stabilizer jackscrew was measured at 7.5 inches between the flat of the ball nut and the flat of the end stop which corresponds to a stabilizer leading edge position between 2 and 3 degrees down or a trim unit setting between 5 and 6 pilot units.<sup>1</sup>

## 6. Wreckage Data bases and Photos

The full data base and photos of the wreckage are on a CD, which is available at the Egyptian Civil Aviation Ministry (MCA). This CD contains:

- a. A folder with three Excel files for wreckage complete data base.
  - i. Floating Wreckage data base.
  - ii. Recovered Wreckage data base.
  - iii. Underwater Surveyed Wreckage data base.
- b. A folder for photos with four sub-folders
  - i. Floating Wreckage Photos: 104 photos.
  - ii. Recovered Wreckage Photos: 98 photos.
  - iii. Underwater Surveyed Wreckage Photos: 330 photos.
  - iv. Wreckage Recovery Process Photos: 25 photos

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<sup>1</sup> B737-300 Aircraft Maintenance Manual 27-41-00

# **Exhibit E**

# **Attachment 1**

**Water Depth at Sharm el-Sheikh**

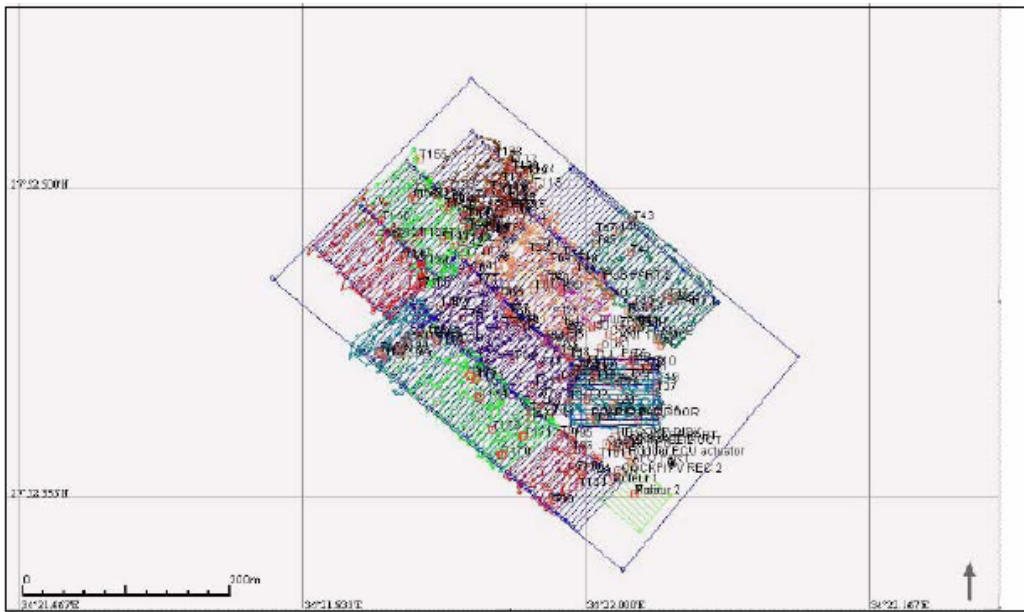


# **Exhibit E**

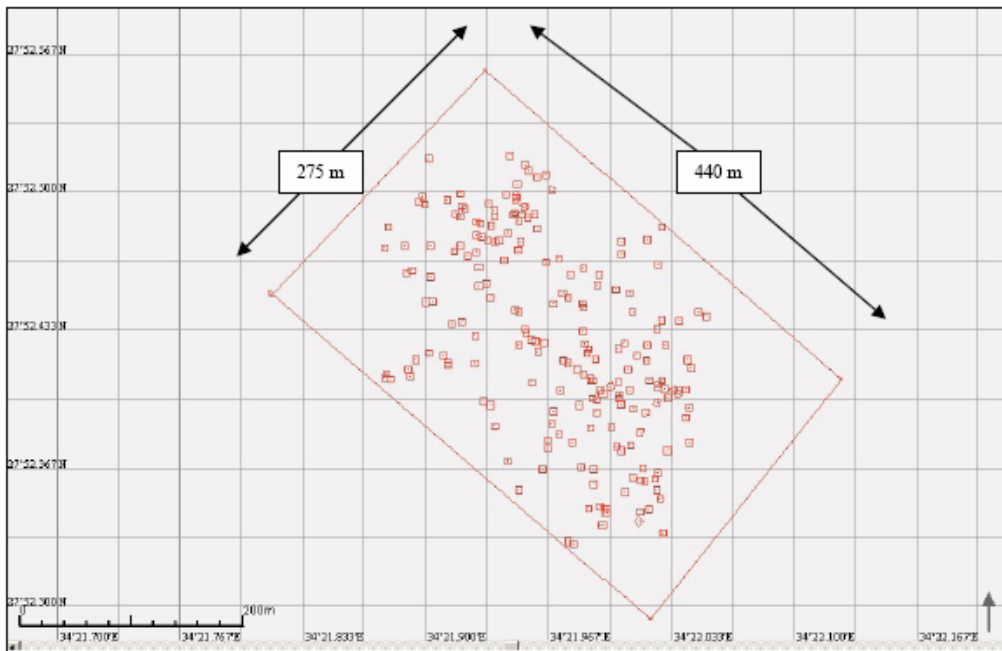
# **Attachment 2**

**Search Areas**

# Search Areas



Total Search Areas with ROV Search Lines

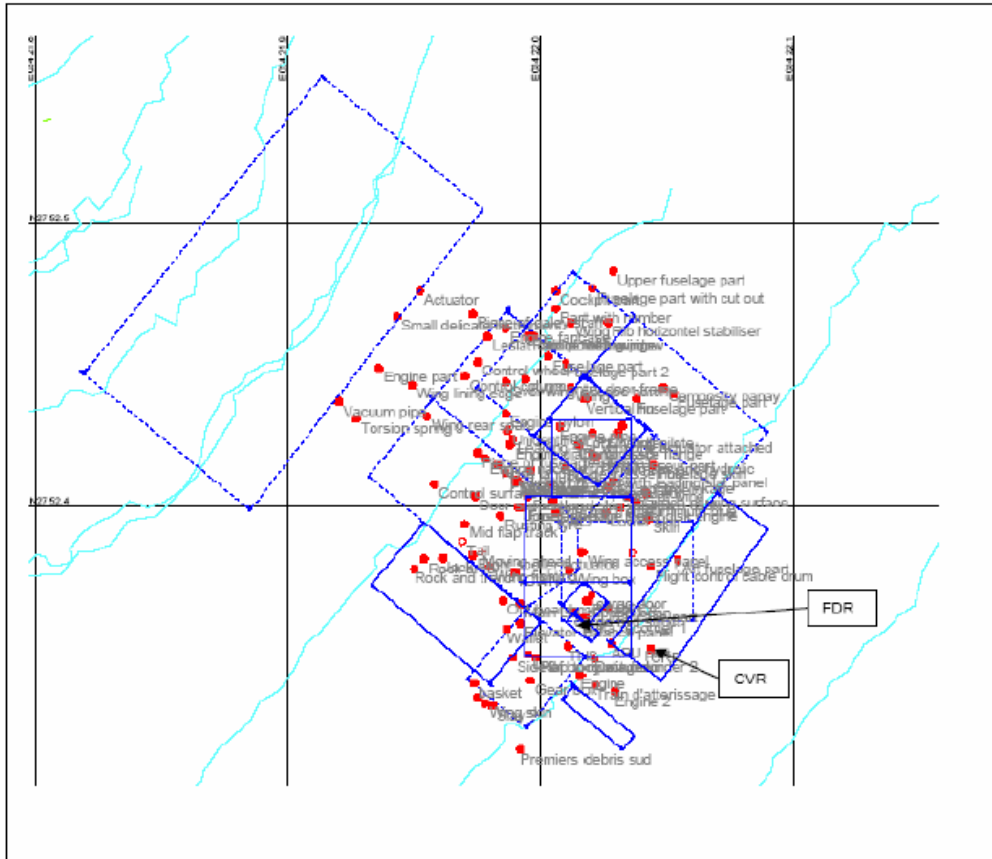


# **Exhibit E**

## **Attachment 3**

**FDR and CVR Locations**

# FDR and CVR Locations



# **Exhibit E**

## **Attachment 4**

**Use of a GIB System For  
Recorders Recovery**



## Use Of A GIB System For Recorders Recovery

A flight recorder immersed under water can be located by the signals (1 bip/second with 37,5 kHz ( $\pm 1$  kHz)) transmitted by the ULB beacon (pinger) attached to the recorder. This pinger starts as soon as it is in contact with water and is designed to transmit this signal for at least thirty days.

The French Navy used an acoustic detector assembled on a pole called "Helle" which tracks signals on frequencies ranging from 7 to 50 kHz. This detector has two reception antennae, one omni-directional and the other directional. It was connected to an audio system that controlled the frequencies and was coupled with a GPS positioning system.

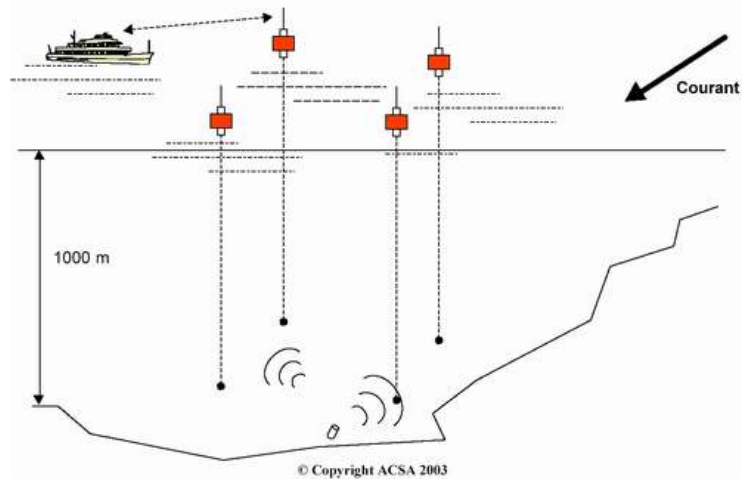
The first stage in the search consisted of checking signal transmissions and defining an general area using the omni-directional antenna. The seafloor being uncharted at that time, locating the beacons was complicated by possible reflections from the transmitted sound waves and possible secondary echoes. The next stage consisted of taking successive bearings using the directional antenna so to get a more precise fix.

This acoustic search determined two possible positions for the beacons: one to the south with a position considered as nominal since it could be picked up from all bearings, but which was transmitting more weakly than the one identified further north. The measurements and calculations performed gave an estimated depth of around one thousand meters.

To confirm these results, the USBL (ultra short base line - acoustic positioning) of the *Ile de Batz* (the first recovery ship on site) was later temporarily modified (in coordination with its manufacturer Sonardyne) and adapted to the reception of the signals transmitted by the southern pinger. These results confirmed the presence of a transmission source beneath the *Ile de Batz* which had been positioned directly above the estimated position.

To narrow the search area, the French Navy contracted ACSA to supply a GIB system (GPS Intelligent Buoys). They adapted a network of four acoustic receivers, combined with GPS information, to conduct a search at a depth of around one thousand meters .

The hydrophones, immersed 450 meters down around the initial identified position, drifted with the current while permanently transmitting information on their position and any signals received. An algorithm integrated all data to determine the recorder's fixed position.



The ROV started searching for the recorders using its cameras based on an initial determination of the position of its beacon. This position was then refined by the ACSA system. That produced a theoretical position with a precision of plus or minus ten meters over one hundred meters.

Squares of twenty by twenty meters were systematically searched by the ROV.

The FDR was discovered on 16<sup>th</sup> January 2004 approximately twelve meters from the computed position.

On the basis of the initial analysis of wreckage distribution, it was decided to define a zone to the south of the position of the FDR. The CVR was found on 17<sup>th</sup> January 2004 in a nearby traced square.

# **Exhibit E**

## **Attachment 5**

**Wreckage Database  
(Floating, Recovered, Surveyed)**

FSH604 Floating Wreckage Database

Ident. Tag No.	Exam Date	Item Description			ATA 2 digit	L/C/R	Length (in)	Width (in)	Description
		Nomenclature	Part No. "_"=unreadable "?"=uncertain digit	Serial No.					
FW1	10-Jan-04	Inboard Spoiler Panel	65-46452-62A	MA4836	27		48	20	
FW2	10-Jan-04	Fuselage Frame Segment	65C27018-1		53		28	20	Fuselage frame segment that includes ground stud GD03004D
FW3	10-Jan-04	Fuselage Frame Segment	69-35352-14		53		10	20	Fuselage frame segment with handwritten notation "400"
FW4	10-Jan-04	Spoiler Panel Fragment	65-46451-70A	MA15971	27		52	11	
FW5	10-Jan-04	Outbd Foreflap Section			27		39	11	Leading edge crushed
FW6	10-Jan-04	Aft flap segment	65-4_870-132		27		22	10	
FW7	10-Jan-04	TE Lower panel	65C25559-1?6		57		40	30	Rib P/N 65-52126-26
FW8	10-Jan-04	Outbd Spoiler	65-46451-70A	MA15970	27		26	21	
FW9	10-Jan-04	Inbd Spoiler			27		58	19	Bulb Seal P/N -60754-23
FW10	10-Jan-04	Aft flap segment	65-47870-15? Or -16?		27		33	16	
FW11	10-Jan-04	Aft outbd flap segment	65-46435-281	18_	27		35	14	
FW12	10-Jan-04	Aft flap segment	65-46435-282	1890	27		24	15	
FW13	10-Jan-04	Inbd flap segment	-47870-154		27		30	17	
FW14	10-Jan-04	Outbd foreflap segment			27		20	8	
FW15	10-Jan-04	Spoiler panel segment			27	L?			Bulb seal P/N 10__0754-23?8 or -28?8 Actuator rod end shows signs of corrosion on a portion of the fracture surface
FW16	10-Jan-04	#3 Spoiler	65-46451-708	MA15952	27	L			Spoiler position determined by position transmitter fitting on inbd leading edge lower surface
FW17	10-Jan-04	Inbd foreflap segment	65-46430-134 (rib)		27				
FW18	10-Jan-04	Aft flap segment			27		39	17	
FW19	10-Jan-04	Aft flap segment			27				Possibly outboard
FW20	10-Jan-04	Outbd aft flap segment			27				
FW21	10-Jan-04	Spoiler			27				
FW22	10-Jan-04	Inbd spoiler segment			27				

FSH604 Floating Wreckage Database

Ident. Tag No.	Item Description				ATA 2 digit	L/C/R	Length (in)	Width (in)	Description
	Exam Date	Nomenclature	Part No. "_"=unreadable "?"=uncertain digit	Serial No.					
FW23	10-Jan-04	#6 spoiler segment			27				Segment of wing web stuck in spoiler direction of travel of wing piece forward and up relative to spoiler
FW24	10-Jan-04	Spoiler fragment	65-46451-708	MA15973	27				
FW25	10-Jan-04	RH lower fin fairing	65-48249-24		55	R			
FW26	10-Jan-04	Outbd aft flap			27	L	84	18	
FW27	10-Jan-04	Elevator or aileron fragment with trim tab			27		31	22	
FW28	10-Jan-04	Aft flap fragment	7870-90 (LE rib)		27		32	15	
FW29	10-Jan-04	Foreflap			27		36	12	
FW30	10-Jan-04	LH elevator upper surface	65C25746-147		27	L	20	14	
FW31	10-Jan-04	Inbd aft flap segment			27		24	12	
FW32	10-Jan-04	Trim tab segment	65C25797-18	135	27		17	6	
FW33	10-Jan-04	Graphite trim tab			27		20	6	
FW34	10-Jan-04	Fixed TE wing upper panel			57		22	9	
FW35	10-Jan-04	Trailing edge structure			57		18	14	
FW36	10-Jan-04	Elevator segment			27		40	20	
FW37	10-Jan-04	Access Panel #910BL			57		28	14	
FW38	10-Jan-04	RH elevator trim tab	65C26384-26A	402347D	27		30	6	
FW39	10-Jan-04	Elevator TE segment			27		33	18	
FW40	10-Jan-04	Rudder fragment			27		33	17	
FW41	10-Jan-04	Elevator or aileron TE segment			27		29	25	
FW42	10-Jan-04	Elevator or aileron TE segment			27		24	19	
FW43	10-Jan-04	Wing LE lower access panel	65C26278-21		57		11	14	
FW44	10-Jan-04	Elevator TE panel			27		22	16	

FSH604 Floating Wreckage Database

Ident. Tag No.	Exam Date	Item Description			ATA 2 digit	L/C/R	Length (in)	Width (in)	Description
		Nomenclature	Part No. "_"=unreadable "?"=uncertain digit	Serial No.					
FW45	10-Jan-04	Rudder Fragments (many)			27				This item number describes a collection of many fragments, most about 12'x12' or less
FW46	10-Jan-04	TE Panel?	65C27482-44		57				
FW47	10-Jan-04	Wing body fairing fragment			53		22	21	
FW48	10-Jan-04	Slide bottle	64236-3 (Air Cruisers)		25				"ALT 749 855"
FW49	10-Jan-04	Slide bottle	D17851-31 (Air Cruisers)		25				
FW50	10-Jan-04	Slide bottle	630120 (BF Goodrich)		25				Structural Composites P/N 1270274
FW51	10-Jan-04	Slide bottle	D17977-3 (Air Cruisers)		25				"ALT 210A-6011" Structural Composites P/N 1270274
FW52	10-Jan-04	Oxy Bottle	801307 and 0B50087		25				
FW53	10-Jan-04	Escape Slide (fwd)	10-61323-478	2206	25				Air Cruisers P/N D31591-478 Serial No. 2206
FW54	10-Jan-04	Life Vests (qty 13)			25				3 crew unfired squib 5 pax unfired squib 1 pax one squib fired, one unfired 4 pax without squib
FW55	10-Jan-04	Escape Slide (aft)	10-61323-?	726A	25				Air Cruisers P/N 61621-46

FSH604 Recovered Wreckage Database

Ident. Tag No.	Item Description			Length (in)	Width (in)	Height (in)	Description
	Nomenclature	Part No. "xx"=unreadable or uncertain digit(s)	Serial No.				
RW1	Horizontal Stabilizer Jackscrew Actuator Gearbox	Forging 65-49964-6		28	10.5	9.5	Screw endstop spline exposed Ballscrew fractured at 0.75 in. from spline shoulder. Safety rod failed at 1.5 in. from spline shoulder.
RW2	Thrust Reverser Actuator	DR MO6118, WO9013550, 81205, 315A808-x, 315A1810-3		28.5	10	5	Ports with "RET" and "EXT"
RW3	Structure			8	5	2	
RW4	Flap Transmission	69-73301-1	8592	30.5	6.5	4.5	Dimension from nut flat to end stop of screw is 21 7/8 in. Dimension from end stop flat to end of part is 2 in.
RW5	Cable Quadrant with Cable	4308xx	0748	6.5	6	3.5	Attached cable is 1/8 inch diameter is 24 inches long
RW6	Scavenge Pump Filter Module			9	3.5	6	Port text: "REAR SCAV IN", "FRONT SCAV IN", "TGB AGB SCAV IN"
RW7	Thrust Reverser Cowl Opening Actuator	1FA1401221		21	5	2	Dimension from shoulder of actuator to end of rod is 11.5 in. "Locked" text on rod
RW8	Hydraulic Component			7	6	6	ball bearing for shaft
RW9	Structure			15	8.5	2.5	
RW10	Hydraulic Component	65C26859x	SC144x	7	2	3	
RW11	Electric Part			4	3	3	
RW12	Hydraulic Actuator			16	5	5	Hydraulic ports with "Extend" and "Retract"
RW13	Hydraulic Actuator			11.5	6	4	
RW14	Engine Start Pad with Gear	104471-0	27494	8.5	8.5	7	
RW15	Horizontal Stabilizer center section rear beam			195	93	48	

FSH604 Recovered Wreckage Database

Ident. Tag No.	Nomenclature	Item Description		Length (in)	Width (in)	Height (in)	Description
		Part No.	Serial No.				
RW15	Left Elevator PCU	"xx"=unreadable or uncertain digit(s) 65-44761	10759A				
RW15	Right Elevator PCU	65-44761-21	0765A				
RW15	Elevator Feel Unit	65-44503-xx	771				
RW15	A/P Actuator - Lower	158300-101	5190				
RW15	A/P Actuator - Upper	158300-101	5173				
RW15	Elevator PCU Input Rod	65-455147-1					
RW15	Left Elevator Position Trasmmitter	69-73373-2, Boeing: S250N104-5	87887				
RW15	Right Elevator Position Trasmmitter	Boeing: S250N104-4	23315				
RW15	Mach Trim Actuator	81205 / 10-61369-7	A1163				
RW15	Mach Trim Trasducer	xxxxxxx	xxxxxx				
RW15	Elevator Balance Panels	65-C-26393-5					
RW16	Tube			32	12	5	
RW17	Electric Motor			5	5	5	Simmond Precision 400Hz Phase 3 High Speed Amps 12 Duty Cycle Intermittxx
RW18	Aileron PCU	65-44828-4 E4	8920	12	9	5	1.75 in. from sleeve endface to rod end flange face. PCU rod at other end sheared in endcap
RW19	Hydarulic Actuator	65-44552-4	952	14	4	4	End gland flat to far side of jam nut is 0.5 inch



FSH604 Recovered Wreckage Database

Ident. Tag No.	Item Description			Length (in)	Width (in)	Height (in)	Description
	Nomenclature	Part No. "xx"=unreadable or uncertain digit(s)	Serial No.				
RW20	Spoiler Mixer	65-50856, 65-46358-1, 69-40296-4, 65-50xx6, 65-46369-4, 65-51633-6, 65-46359-14		14	16	5	
RW21	Fuel system part	66503-4034-33, 66503- 4034-352, 66503 4455- 056, 66503-4414-022	5624, 4294	11	6.5	4.5	
RW22	Flap Angle Gearbox	65-50585-15 Rev x		9	14	4	
RW23	Torque Tube with Splines			23	4	4.5	
RW24	Hydraulic Actuator Rod End With attached structure	69-73485-108, 65C26796-16revA, 65C36641-30revE		10	4	5	
RW25	Horizontal Stabilizer Jackscrew	Assy 65-51524-16		32.5	19	7	Dimension from the flat of the ball nut to the flat of the endstop is 7.5 inches.
RW26	Structure			15.5	8	7	
RW27	Force Transducer - Autopilot	10-61072-7 M	3284	4	2.5	2.5	
RW28	Flap transmission	xx27501-3	10902A	3.5	4	3.5	
RW29	Speedbrake Mechanism		80477	9	6	3.5	
RW30	Hydraulic Transfer Valve			10	2.5	2.5	
RW31	Electrical component	311 13646 01	9212	5	3	2	
RW32	Fuel Timer	074327119M71607	GOS20184	7.5	6.5	3	3 tubes attached, the longest of which is 41 inch.
RW33	Spoiler Valve Manifold	65-44565-5	Wx9027307	7.5	7.5	3.5	
RW34	Section of vertical stabilizer With components			93	40	45	
RW34	Main Rudder PCU	65C37053-9	892x				Includes Jetpipe servo valve 75130-A3099 S/N 411171

FSH604 Recovered Wreckage Database

Ident. Tag No.	Item Description			Length (in)	Width (in)	Height (in)	Description
	Nomenclature	Part No. "xx"=unreadable or uncertain digit(s)	Serial No.				
RW34	Rudder Pressure Reducer	Teijin Seiki 1704600-x	10xx				SCD No. 10-62255-xx, Includes Eaton Hydraulic Pressure Transducer Boeing PN10-62254-1 Ser.No. 146451 Date of MFG 01/99. Includes Parker Solenoid valve P/N 881600-001 S/N 30708 SCD BAC 10-60811-13.
RW34	Feel and Centering Unit	Assy 65-51251-5					Assy date: MAY 11 1992, Bracket P/N 65C25410-5, Control Rod from F&C unit to input rod: Assy 69-37285-8 02/18/91
RW34	Actuator, rudder trim	10-62025-3 revU	C1412				MPC Products Corp. MFR 19710/U26B 81205 D/C 9218 FT 04-29-92
RW34	Standby Rudder PCU	Assy 1150	6005x				
RW35	Blade seal	65-48248-5, 1060754-770		29	15	4	42 in. long seal folded on itself
RW36	Flap Leading Edge	65-46430-129	1650	30	18	7	Flap leading edge with tube and roller assembly
RW37	Column cable quadrant	65-52995-11, 65-535924, Assy 6x-5359xx, 65C31007-xx		19	12	6	
RW38	First Officer's control wheel			12	8	3	
RW39	A4 Power Amplifier	641-8592-001		9	7	3	
RW40	Recognition Light	30-0906104MOD	601	9	7	6	
RW41	APU Turbine Disc			15	14	3	
RW42	Bellcrank with rod and flex cable	315A1897-5		26	10	5	
RW43	Control Surface with broken actuator	65C26633-27		21	13	9	

FSH604 Recovered Wreckage Database

Ident. Tag No.	Item Description			Length (in)	Width (in)	Height (in)	Description
	Nomenclature	Part No. "xx"=unreadable or uncertain digit(s)	Serial No.				
RW44	Crank Assembly	69-20427-1, 69-20235-2, 65-25844-7, 65-25820-9		18.5	6	4	
RW45	Spoiler Actuator	65-44561-x	7048	23	24	8	
RW46	Drum	65-44065		9	7.5	1	
RW47	OUTBD Gnd Spoiler	65C26864-3	E-0376	23	19	8	
RW48	Spoiler Actuator Valve	65-44645		8	8	4	
RW49	Spoiler Actuator	65-44561-15	10275	43	10.5	14	
RW50	VOR / DME Indicator	N/A	N/A	4	3.5	4	
RW51	Cockpit Temperature Selector	N/A	N/A	5	2	2	
RW52	Frist Aid Kit	N/A	N/A	10	10	2.5	
RW53	Portable cylinder Pressure indicator.	N/A	N/A	2	1.5	1.75	
RW54	Clamp	2703-300.A	N/A	4.5	4	0.75	
RW55	Passenger Oxygen Mask	250054	N/A	5.5	5	4	
RW56	Wing Piece of Structure	N/A	N/A	55.5	14.5	10	

FSH604 Surveyed Wreckage Database  
(Janus II)

T#	Latitude	Longitude	Description	Janus II photo reference	Recovered Wreckage No.
n/a	52.4270	21.9890	Pile of electrical wires beside T54	2004-01-19-200844.JPG	
n/a	52.4160	21.9390	not ident.	2004-01-20-120103.JPG	
T1	52.4090	21.9915	Mid flap		
T2	52.4090	21.9900	MLG door mecanisme		
T3	52.4100	21.9900	Passager seat frame		
T4	52.4150	22.0440	Fuselage skin		
T5	52.4090	22.0280	Seat frame		
T6	52.4041	22.0103	Fuselage skin		
T7	52.4055	22.0258	Fuselage skin		
T8	52.4047	22.0293	Mechanism		
T9	52.4040	22.0369	Safety, life jacket and fuselage	2004-01-19-073927.JPG	
T10	52.4047	22.0409	Piece of wing surface		
T11	52.4025	22.0367	Aluminium with blue paint		
T12	52.4043	22.0343	Piece of wing		
T13	52.4070	22.0260	Piece of wing		
T14	52.4084	22.0044	Frame		
T15	52.4060	21.9998	Piece of passanger seat		
T16	52.4040	21.9951	Fuselage skin / windows		
T17	52.4022	22.0050	Windows frame		
T18	52.3975	22.0057	PSU		
T19	52.3960	22.0425	Skin		
T20	52.3983	22.0253	Lower skin		
T21	52.4002	22.0045	Fuselage skin		
T22	52.4025	21.9963	Seat frame		
T23	52.3997	21.9934	Fuselage Skin		
T24	52.4004	22.0312	Metal Disk (engine)		
T25	52.3954	22.0124	Composite piece. Belt and tissue		
T26	52.3937	22.0193	Metal Piece		
T27	52.3910	22.0410	Fuselage and windows		
T28	52.3936	21.9933	spoiler actuator attached to portion of the wing spar	2004-01-19-094158.JPG, 2004-01-20-170624.JPG, 2004-01-20-170615.JPG	
T29	52.3840	22.0161	Wing access panel		
T30	52.3750	22.0060	Composity panel		
T31	52.3861	21.9899	Rear part of fuselage		
T32	52.3865	22.0006	Pylon		
T33	52.3750	22.0310	Lower body skin		
T34	52.3788	22.0431	flt. cont. cable drum	2004-01-19-112045.JPG	
T35	52.4380	22.0280	Fuselage skin		
T36	52.4400	22.0520	Fuselage skin with "Cut here" indicated		
T37	52.4420	22.0480	Pile of debris	2004-01-19-132940.JPG, 2004-01-19-133012.JPG	
T38	52.4260	22.0300	Composite panel fixed te		
T39	52.4190	22.0420	skin with letters		
T40	52.4420	22.0120	Wing	2004-01-19-160043.JPG, 2004-01-19-155924.JPG	
T41	52.4650	22.0260	RIB horizontal stabilizer		
T42	52.4530	22.0030	Fuselage section with "FLASH" text	2004-01-19-162335.JPG, 2004-01-19-163724.JPG, 2004-01-19-163717.JPG	

FSH604 Surveyed Wreckage Database  
(Janus II)

T#	Latitude	Longitude	Description	Janus II photo reference	Recovered Wreckage No.
T43	52.4830	22.0280	Upper fuselage part		
T44	52.4550	21.9940	Forward entry door frame - 1L		
T45	52.4700	22.0060	Part with number		
T46	52.4770	22.0200	Fuselage part with a door cutout		
T47	52.4760	22.0060	Fuselage part "Brew handle must be in down position during taxi, take off,		
T48	52.4600	21.9950	Leading edge slat with part of wing	2004-01-19-193417.JPG	
T49	52.4120	21.9860	Lower wing scan with leading slat panel		
T50	52.4244	22.0042	Skin		
T51	52.4191	21.9929	Skin		
T52	52.4240	21.9890	Leading edge slat with one actuator attached	2004-01-19-195521.JPG	
T53	52.4146	21.9826	Nose landing gear assembly		
T54	52.4266	21.9869	Main Equipment Center skin door	2004-01-19-201051.JPG, 2004-01-19-201214.JPG	
T55	52.4220	21.9884	Engine diagonal brace		
T56	52.4329	21.9858	Engine pylon		
T57	52.4440	21.9860	Over wing escape hatch		
T58	52.4280	21.9600	Passenger seat recline actuator		
T59	52.4490	21.9780	No identify		
T60	52.4459	21.9856	not ident.	2004-01-19-230150.JPG, 2004-01-19-230124.JPG	
T61	52.4460	21.9700	control column	2004-01-19-232047.JPG	
T62	52.4510	21.9750	control wheel	2004-01-19-233054.JPG	
T63	52.4630	21.9860	Engin fancase		
T64	52.4600	21.9790	leading edge slat and portion of wing	2004-01-20-000743.JPG, 2004-01-20-000254.JPG	
T65	52.4420	21.9510	Engine fan case		
T66	52.4320	21.9550	Wing rear spar		
T67	52.4680	21.9730	passenger seat frame with spring	2004-01-20-010121.JPG, 2004-01-20-010033.JPG, 2004-01-20-010020.JPG, 2004-01-20-010020.JPG, 2004-01-20-005839.JPG, 2004-01-20-005834.JPG, 2004-01-20-005723.JPG, 2004-01-20-005721.JPG	
T68	52.4660	21.9660	Wing spar piece		
T69	52.4760	21.9520	spoiler actuator	2004-01-20-023738.JPG, 2004-01-20-023718.JPG, 2004-01-20-023627.JPG, 2004-01-20-023611.JPG, 2004-01-20-023523.JPG, 2004-01-20-023601.JPG	
T70	52.4545	21.9292	Eng VSV HPC		
T71	52.4673	21.9429	Small delicate instrument		
T72	52.4373	21.9200	Flap angle gearbox?		
T73	52.4468	21.9006	Wing center section structure		
T74	52.4490	21.9360	Engine part ?		
T75	52.4307	21.9273	Torsion spring		
T76	52.4432	21.9490	Wing leading edge Flap FSS394		

FSH604 Surveyed Wreckage Database  
(Janus II)

T#	Latitude	Longitude	Description	Janus II photo reference	Recovered Wreckage No.
T77	52.4337	21.9544	Wing rear spar station 286 and linkage		
T78	52.4173	21.9272	Cable drum and support	2004-01-20-114025.JPG, 2004-01-20-113958.JPG	
T79	52.4260	21.9510	Internal handle Passenger / service		
T80	52.4286	21.9579	Structural and skin		
T81	52.4273	21.9644	wires and some panel	2004-01-20-121606.JPG, 2004-01-20-121514.JPG	
T82	52.4229	21.9614	Outside passenger door - Left		
T83	52.4188	21.9751	Pieces of fuselage skin with cockpit window cutout		
T84	52.4080	21.9580	control surface with broken actuator	2004-01-20-131900.JPG	
T85	52.4175	21.9780	Engine Nacelle with pneumatic and hydraulic		
T86	52.4041	21.9738	Door support and skin 2x2m		
T87	52.3880	21.9690	Horizontal stabilizer center section with part of the left stab, elev. & tab	2004-01-20-141831.JPG, 2004-01-20-141650.JPG, 2004-01-20-141859.JPG, 2004-01-20-141908.JPG, 2004-01-20-143558.JPG, 2004-01-20-144151.JPG, 2004-01-20-142138.JPG, 2004-01-20-142144.JPG, 2004-01-20-142035.JPG, 2004-01-20-142301.JPG, 2004-01-20-143410.JPG, 2004-01-20-142215.JPG, 2004-01-20-141924.JPG	RW15
	52.3880	21.9690	Hydraulic tube ~1m (Raised with RW15)		RW16
T88	52.4100	21.9900	trailing edge flap control linkage	2004-01-20-155813.JPG, 2004-01-20-161009.JPG	
T89	52.3970	21.9840	Brusting Tyre		
T90	52.4000	21.9910	Uper Fuselage skin		
T91	52.3940	21.9700	Mid Flap Track		
T92	52.3830	21.9730	Flight spoiler actuator valve	2004-01-20-171655.JPG	
T93	52.3790	21.9800	Wing fitting		
T94	52.3670	21.9850	Outboard Mid Flap		
T95	52.3660	21.9920	Main LG Support Beam		
T96	52.3590	21.9920	Elevator balance panel	2004-01-20-184651.JPG	
T97	52.3470	21.9890	Side of body Wing skin		
T98	52.3310	21.9780	Wing skin		
T99	52.3300	21.9810	slide (?) + ??	2004-01-20-193955.JPG	
T100	52.3480	21.9950	Lug		
T101	52.3551	22.0078	No identify		
T102	52.3450	21.9980	Hydraulic		
T103	52.3390	21.9960	Gear box		
T104	52.3470	21.9980	Flap Torque Tube		
T105	52.4877	21.9560	Floor pannel with structure		
T106	52.4890	21.9477	ELEC WIRING		
T107	52.4899	21.9487	PERSO EFFECT		

FSH604 Surveyed Wreckage Database  
(Janus II)

T#	Latitude	Longitude	Description	Janus II photo reference	Recovered Wreckage No.
T108	52.4861	21.9510	Human remain		
T109	52.4766	21.9402			
T110	52.4758	21.9382	small electronic box		
T111	52.4803	21.9452	unknow small part		
T112	52.4890	21.9530	wiring and insulation		
T113	52.5008	21.9692	Valve		
T114	52.4820	21.9610	Stil ring		
T115	52.4892	21.9597	control wheel stering force sensor (recovered)		RW27
T116	52.4985	21.9495			
T117	52.4965	21.9492	Engine insulation		
T118	52.4974	21.9497	Electric Motor		
T119	52.4928	21.9538	Engine case		
T120	52.4785	21.9309	floor panel with structure		
T121	52.4769	21.9339	elec motor		
T122	52.4838	21.9362	Bracket		
T123	52.4930	21.9540	belly skin and stucture		
T124	52.5083	21.9658	personal effect		
T125	52.4879	21.9380	miscelaneous structure		
T126	52.4910	21.9378	side of body structure with wiring		
T127	52.5036	21.9503	personal effect		
T128	52.5102	21.9564	Crank arm		
T129	52.5070	21.9610	sit & personal effect		
T130	52.4987	21.9439	electric motor		
T131	52.4845	21.9300	wing structure		
T132	52.5131	21.9545	bleed air duct		
T133	52.4943	21.9346	unknow electrical part		
T134	52.4856	21.9281	unknow linkage		
T135	52.4790	21.9281	miscellanious metal structure		
T136	52.4932	21.9200	oxygen bottle		
T137	52.4993	21.9191	hydraulic activator		
T138	52.5176	21.9464	hydraulic tube		
T139	52.4977	21.8986	oxygen bottle		
T140	52.4635	21.9294	part of wheel mecanism ( recovered)		RW28
T141	52.4557	21.9332	control command base		
T142	52.4688	21.9230	personal effect		
T143	52.4710	21.9280	Speed bracke lever		RW29
T144	52.4713	21.9157	T/R cowl opening actuator		
T145	52.4740	21.9190	engine part fuel pump		
T146	52.4880	21.9190	Engine part Link		
T147	52.4620	21.8930	Engine part oil pressure switch		
T148	52.4920	21.9220	Oxygen bottle		
T149	52.4895	21.9166	Engine part gear box		
T150	52.4960	21.9120	Engine part Gear box		
T151	52.4740	21.8890	Engine part Compressor Disk		
T152	52.4730	21.8780	Toilet system AC motor		
T153	52.4950	21.8970	Transfer valve		RW30
T154	52.4940	21.9000	Landing gear component		
T155	52.5160	21.9020	? Electronic		RW31
T156	52.4830	22.0250	Engine part Fuel Timer		RW32
T157	52.4740	21.9030	Engine part		
T158	52.4610	21.8900	Engine part pressure switch (T147)		

FSH604 Surveyed Wreckage Database  
(Janus II)

T#	Latitude	Longitude	Description	Janus II photo reference	Recovered Wreckage No.
T159	52.4590	21.9030	Engine part TIR Cowl hold open actuator		
T160	52.4470	21.9040	Landing gear support		
T161	52.4290	21.8930	Debris structure		
T162	52.4090	21.8810	Hydraulic component		
T163	52.4110	21.8930	Hydraulic component		
T164	52.4370	21.9160	Structure		
T165	52.4100	21.8930	Structure		
T166	52.4200	21.9030	Coupler		
T167	52.4200	21.9040	Spoiler valve manifold		RW33
T168	52.4170	21.9130	Flight spoiler		
T169	52.4180	21.9100	Hydraulic fuse		
T170	52.3560	21.9510	Engine part Disk		
T171	52.3660	21.9640	Electric wires		
T172	52.3800	21.9670	Electronic Box		RW39
T173	52.3700	21.9450	Engine part		
T174	52.3870	21.9380	Engine part		
T175	52.3970	21.9360	Unidentified		
T176	52.3990	21.9320	LV Cover		
T177	52.3760	21.9670	Push Pull cable		RW42
T178	52.4480	21.9940	Electronic Box		RW40



FSH604 Surveyed Wreckage Database  
(Ile de Batz)

T#	Time	Latitude	Longitude	Description	Date	Recovered Wreckage No.
	9:54:02	52.4192	22.0207	skin	12-Jan-04	
	10:00:04	52.4165	22.0190	white skin 1.5x1m	12-Jan-04	
	10:02:21	52.4185	22.0182	STA600 left side escape hatch 4.5m skin	12-Jan-04	
	10:30:08	52.4205	22.0172	skin	12-Jan-04	
	10:41:50	52.4205	22.0183	skin, maybe lap splice, no paint	12-Jan-04	
	10:45:57	52.4214	22.0190	stringers & skin	12-Jan-04	
	11:01:17	52.4249	22.0285	skin section	12-Jan-04	
	11:05:23	52.4185	22.0215	engine case with stator vane	12-Jan-04	
	11:12:42			window frame	12-Jan-04	
	11:13:40	52.4085	22.0108	Possible wing skin 6in.x3ft.	12-Jan-04	
	11:49:58	52.4361	22.0348	fuselage piece 1x2m	12-Jan-04	
	11:56:30	52.4237	22.0233	Fuselage skin 3x4m	12-Jan-04	
	12:35:05	52.4410	22.0462	belly skin 1x1m, dark paint	12-Jan-04	
	13:04:04	52.4086	22.0011	butt splice	12-Jan-04	
	13:52:20	52.4142	22.0096	fuselage skin with 1.5 window frames	12-Jan-04	
	14:25:00	52.4212	22.0100	two pieces of skin, 1x1m, 1x2m	12-Jan-04	
	14:31:36	52.4187	22.0126	737 Airplane Flight Manual (AFM)	12-Jan-04	
	15:20:25	52.4217	22.0149	ring/strip of cap sealed fasteners with adjacent wing?	12-Jan-04	
	15:40:38	52.4384	22.0369	fuselage skin 4x2m, white	12-Jan-04	
	15:49:40	52.4444	22.0364	Instrument panel?	12-Jan-04	
	15:53:12	52.4388	22.0309	fuselage skin, 7 stringers x 2 frames @ lap, no structure attached, dark & light paint	12-Jan-04	
	16:03:49	52.4306	22.0189	fuselage skin 4x2m, possibly part of logo arrow above windows	12-Jan-04	
	16:05:19	52.4259	22.0152	ballscrew	12-Jan-04	
	16:25:23	52.4175	22.0063	ball of loose tangled wires	12-Jan-04	
	16:35:40	52.4305	22.0197	skin fragment, sect 43, ~STA 460	12-Jan-04	
	16:50:50	52.4429	22.0312	skin 2x1m	12-Jan-04	
	17:11:32	52.4067	21.9965	portion of floor beam & seat track	12-Jan-04	
	17:13:00	52.4104	21.9967	wing lower surface	12-Jan-04	
	13:40:07	xxx	xxx	fuselage skin fragment, 1 or 2 windows with possible door cutout	13-Jan-04	
	5:57:00	xxx	xxx	magnetic tape(?)	14-Jan-04	
	6:18:00	xxx	xxx	skin	14-Jan-04	
	10:04:00	xxx	xxx	VHF antenna	14-Jan-04	
	10:23:00	xxx	xxx	fuselage skin	14-Jan-04	
	11:10:00	xxx	xxx	compressor part	14-Jan-04	
	12:54:00	xxx	xxx	white box	14-Jan-04	
	15:20:46	xxx	xxx	compressor flange	14-Jan-04	
	15:42:39	xxx	xxx	fuselage part	14-Jan-04	
	17:13:14	52.4129	21.9963	wing lower skin, 4 access panels, 3mx1m, +front spar +leading edge, reg.mark "SU-Z", ~STA600	14-Jan-04	
	17:40:50	52.4416	22.0194	front spar of vertical stabilizer skin, 2-3m long spar, ref SRM 55-30-10	14-Jan-04	
	17:55:12	52.4726	22.0062	skin 0.5mx20cm	14-Jan-04	
	17:58:15	52.4247	22.0048	Metal duct, 1mx10cm	14-Jan-04	
	18:07:20	52.4157	21.9993	Frame and skin, 1m	14-Jan-04	

FSH604 Surveyed Wreckage Database  
(Ile de Batz)

T#	Time	Latitude	Longitude	Description	Date	Recovered Wreckage No.
	18:16:05	52.4161	21.9972	skin, 1x2m composite	14-Jan-04	
	18:19:20	52.4204	21.9962	skin, white, 1mx30cm	14-Jan-04	
	19:20:38	52.4321	22.0352	skin and stringers, 1x4m, white paint	14-Jan-04	
	19:53:01	52.4516	22.0116	skin with three windows, external paint scheme identifies this as ~STA500, 3x3m	14-Jan-04	
	20:06:08	52.4419	22.0128	concrete block with cable through center, used by French Navy for depth measurement	14-Jan-04	
	20:26:36	52.4324	22.0322	skin, 1.5x1.5m, window frame, white paint	14-Jan-04	
	20:30:39	52.4292	22.0363	skin, no paint, 0.5x0.5m with light insulation	14-Jan-04	
	20:53:50	52.4332	22.0250	skin, 1x0.5m, partial blue letter?	14-Jan-04	
	20:56:15	52.4379	22.0194	spar with elliptical holes, vertical stab skin	14-Jan-04	
	21:22:14	52.4476	21.9976	skin, 2x3m, doublers, chem mill waffle pattern	14-Jan-04	
	21:31:55	52.4411	22.0143	concrete block, French Navy Bathymetry device	14-Jan-04	
	22:02:21	52.4233	22.0360	Emergency light battery tray	14-Jan-04	
	22:41:02	52.4241	22.0306	possible LRU handle 4x1.5in., black	14-Jan-04	
	23:23:16	52.4248	22.0221	possible LRU handle	14-Jan-04	
	23:29:07	52.4200	22.0304	white exterior 2x1m	14-Jan-04	
	23:54:09	52.4207	22.0215	fuselage skin 1x2m	14-Jan-04	
	3:23:00	52.3645	22.0266	Fan case fragment	16-Jan-04	
	3:39:00	52.3664	22.0179	HP compressor disk	16-Jan-04	
	3:46:00	52.3664	22.0179	Front engine mount	16-Jan-04	
	4:03:00	52.3782	22.0105	Wing Box Fragment	16-Jan-04	
	16:50:30	52.3585	22.0230	Fuselage Skin White/Blue	16-Jan-04	
	16:54:32	52.3600	22.0186	Flight Data Recorder (FDR)	16-Jan-04	FDR
	5:48:00	52.3621	22.0121	Box Structure w/Blue skin	17-Jan-04	
	5:53:10	52.3650	22.0080	Fuselage Skin, 1x1m	17-Jan-04	
	5:57:41	52.3660	22.0150	Floor Section, 2x3m	17-Jan-04	
	6:26:13	52.3590	22.0200	Cargo Door Section, >1x2m	17-Jan-04	
	6:57:10	52.3590	22.0170	Floor Frames, Side of Body Center Section, 2x0.5m	17-Jan-04	
	7:19:20	52.3700	22.0220	Nose tire	17-Jan-04	
	7:22:29	52.3710	22.0226	Fuselage skin, 1x1.5m	17-Jan-04	
	7:30:12	52.3670	22.0290	Section of entry door, "Automatic Slide Armed", 1x0.5m	17-Jan-04	
	7:34:34	52.3610	22.0290	Nose wheel hub	17-Jan-04	
	7:42:20	52.3690	22.0250	Flat bulkhead/pressure deck, 1x1.5m	17-Jan-04	
	7:55:50	52.3545	22.0150	Part of fin/torque tube, possible rudder mechanism attached, 2x0.5m	17-Jan-04	
	8:12:45	52.3612	22.0149	Vertical fin trailing edge beam lower structure(?), >1x1m	17-Jan-04	
	8:22:29	52.3522	22.0289	Empennage/APU firewall section, 1x1.5m	17-Jan-04	
	8:44:57	52.3524	22.0167	Skin APU/Floor Beam, wing spar side of body	17-Jan-04	

FSH604 Surveyed Wreckage Database  
(Ile de Batz)

T#	Time	Latitude	Longitude	Description	Date	Recovered Wreckage No.
	9:08:08	52.3585	22.0194	Galley parts, cargo liner, floor beam, blue skin (large pile mixed debris), 2X2m	17-Jan-04	
	9:28:28	52.3577	22.0121	Vertical, right side lower by logo, access door 9529 (Standby Rudder PCU door), 1x2m	17-Jan-04	
	10:03:00	52.3587	21.9861	Elevator control surface with balance panel, graphite, "65C26393-5" & "69-41307-20"	17-Jan-04	
	10:19:04	52.3649	21.9909	Main Landing Gear Beam - Right	17-Jan-04	
	11:41:36	52.3638	21.9930	Wing skin, 1x2ft.	17-Jan-04	
	12:22:39	52.3526	22.0263	Skin with vortex generators and APU firewall	17-Jan-04	
	13:07:44	52.3659	22.0181	Thrust reverser cascade vanes	17-Jan-04	
	13:56:35	52.3644	22.0224	APU oil fill access door, P/N 65-76712-509, 1x1m	17-Jan-04	
	14:17:10	52.3639	22.0236	Panel, honeycomb w/ white paint & blade seal, 1x3m	17-Jan-04	
	14:43:39	52.3759	22.0158	Section of tire, MLG(?)	17-Jan-04	
	15:04:08	52.3734	22.0262	Tailcone with strobe position light, 1x1m	17-Jan-04	
	15:04:08	52.3734	22.0262	Skin with text "sta... do not plug", static port @ STA 420	17-Jan-04	
	15:57:34	52.3510	22.0268	APU fragment	17-Jan-04	
	16:09:58	52.3507	22.0256	Thrust reverser cowl fragment, 0.25x0.1m	17-Jan-04	
	16:23:30	52.3557	22.0128	Thrust reverser block door	17-Jan-04	
	16:54:30	52.3618	22.0102	Aft flap actuating mechanism pull cable	17-Jan-04	
	16:58:33	52.3608	22.0123	Engine Starter Casing	17-Jan-04	
	17:05:10	52.3608	22.0169	Flap carriage spindle (?)	17-Jan-04	
	17:26:10	52.3571	22.0135	Wing spar 1.5x0.3m	17-Jan-04	
	18:28:09	52.3454	22.0160	Cockpit Voice Recorder (CVR)	17-Jan-04	CVR
	18:28:09	52.3454	22.0160	Nose landing gear retract actuator, extended (corresponding to gear-up)	17-Jan-04	
	18:28:09	52.3454	22.0160	Toothed gear and support, gear diameter ~6in.	17-Jan-04	
	16:06:47	52.3369	22.0153	Engine Core, combustion chamber to exhaust, engine axis vertical with fuel nozzles at bottom and crushed exhaust at the top	18-Jan-04	
	17:32:00	52.3403	22.0222	Left and Right main landing gear assemblies	18-Jan-04	
	17:52:54	52.3342	22.0176	Flap support w/ transmission	18-Jan-04	
	18:38:36	52.3340	22.0279	Engine Core, combustion chamber to exhaust, engine axis vertical with fuel nozzles at bottom and exhaust at the top	18-Jan-04	
	18:38:36	52.3340	22.0279	two wheels (MLG?...viewed from engine)	18-Jan-04	

FSH604 Surveyed Wreckage Database  
(Ile de Batz)

T#	Time	Latitude	Longitude	Description	Date	Recovered Wreckage No.
	18:38:36	52.3340	22.0279	Main Engine Control (beside engine) P/N 66503-6063-215, S/N WYG80008	18-Jan-04	
	19:17:34	52.3377	22.0298	Main Landing Gear beam	18-Jan-04	
	23:00:00	52.4185	21.9335	Fuselage upper skin just above entry door	20-Jan-04	
	5:10:00	52.4600	21.9970	Fuselage skin at least 5 passenger windows and the "FLASH" logo	21-Jan-04	
	5:43:00	51.8541	25.5599	skin panel	21-Jan-04	
	6:32:00	52.4436	22.0179	Low pressure compressor case	21-Jan-04	
	0:11:46	52.3814	22.0543	skin, aft crown w/ blue lettering from "FLASH AIRLINES", 1x4m	22-Jan-04	
	5:18:00	52.3616	22.0444	Tire	22-Jan-04	
	6:30:00	52.3483	22.0271	Wing panels	22-Jan-04	
	6:38:00	52.3519	22.0266	APU shroud	22-Jan-04	
	9:13:20	52.3505	22.0192	Hydraulic Actuator	22-Jan-04	RW13
	9:22:53	52.3403	22.0227	Flap track with transmission	22-Jan-04	
	9:22:53	52.3403	22.0227	hydraulic endcap	22-Jan-04	RW
	9:22:53	52.3403	22.0227	hydraulic valve	22-Jan-04	RW8
	9:22:53	52.3403	22.0227	flap track and flap ball screw with transmission	22-Jan-04	
	9:22:53	52.3403	22.0227	flap ballscrew without transmission	22-Jan-04	
	9:22:53	52.3403	22.0227	Thrust reverser actuator	22-Jan-04	RW2
	9:22:53	52.3403	22.0227	Engine start pad with gear	22-Jan-04	RW14
	10:16:16	52.3387	22.0246	Outboard mid flap carriage	22-Jan-04	
	16:14:05	52.3517	22.0109	Horizontal stabilizer trim motor	22-Jan-04	RW1
	19:21:00	52.3603	22.0019	Outboard flap jackscrew	22-Jan-04	RW4
	20:05:08	52.3529	22.0090	MLG tire, Inbd flap track, Engine Pylon, MLG uplock hook, inbd flap track cam roller, & other MLG wheel well components	22-Jan-04	
	20:51:51	52.3725	21.9828	Outboard mid flap (same as T94?)	22-Jan-04	
	21:15:12	52.3838	21.9678	Hydraulic component - unknown	22-Jan-04	
	21:42:46	52.3958	21.9157	MLG brake hydraulic actuator	22-Jan-04	RW10
	21:58:40	52.3941	21.9494	Hyd valve - motor	22-Jan-04	RW11
	22:45:30	52.3709	21.9895	MLG support beam and some flap structure	22-Jan-04	
	23:01:00	52.3669	21.9943	Hydraulic Actuator with Ext/Ret labeling	22-Jan-04	RW12
	23:25:30	52.3600	21.9905	Fire wall (APU or Engine)	22-Jan-04	
	23:28:50	52.3540	21.9924	Pylon attach fitting & engine firewall	22-Jan-04	
	23:32:26	52.3554	21.9963	Engine gearbox (hyd or fuel) & wing skin	22-Jan-04	
	23:40:21	52.3554	22.0016	Quadrant with cable attached	22-Jan-04	RW5
	23:58:20	52.3646	21.9870	Wing skin, structure, & engine fire wall	22-Jan-04	
	0:02:00	52.3644	21.9875	Balance panel (elev & stab structure?)	23-Jan-04	
	0:08:10	52.3694	21.9840	MLG beam & inbd flap spindle	23-Jan-04	

FSH604 Surveyed Wreckage Database  
(Ile de Batz)

T#	Time	Latitude	Longitude	Description	Date	Recovered Wreckage No.
	0:14:00	52.3732	21.9777	Hose - unknown	23-Jan-04	
	0:35:00	52.3730	21.9770	Landing gear lock actuator	23-Jan-04	
	3:40:00	52.3522	221.9859	Plug door - small	23-Jan-04	
	4:36:11	52.3804	21.9704	Wing skin, 2mx10cm.	23-Jan-04	
	6:30:00	52.3538	21.9770	Thrust reverser blocker door	23-Jan-04	
	8:58:00	52.3623	21.9514	Engine disk	23-Jan-04	
	9:21:40	52.3383	21.9811	Fuselage skin & escape slide	23-Jan-04	
	10:30:00	xxx	xxx	unintentional recovery	23-Jan-04	RW3
	10:30:00	xxx	xxx	unintentional recovery	23-Jan-04	RW9
	12:00:00	xxx	xxx	Engine T/R cown opening actuator	23-Jan-04	RW6
	12:00:00	xxx	xxx	Enigne oil lubricating unit with MCD intact	23-Jan-04	RW7
	6:00:00	52.3580	22.0163	Vertical stabilizer section, Aft spar with lugs still attached to fuselage frame to just above standby PCU. Aft spar with structure to rudder hinge, including front spar of rudder surface.	24-Jan-04	RW34
	6:00:00	52.3580	22.0163	Blade seal ~42 inch (Raised with RW34)		RW35
	6:00:00	52.3580	22.0163	Flap leading edge with tube (Raised with RW34)		RW36
	6:37:20	52.3538	22.0257	Structure (2m) and hydraulic component with spline shaft input	24-Jan-04	
	14:40:00	52.3461	22.0233	Parts of an engine gearbox	24-Jan-04	
	14:47:00	52.3435	22.0220	Actuator electric motor	24-Jan-04	RW17
	17:06:00	52.4098	22.0097	Pile of cabin interior parts (O2 masks, reading lights, etc.)	24-Jan-04	
	18:15:00	52.4088	22.0418	Structural element, possibly balance panel or balance weights.	24-Jan-04	
	19:04:40	52.3682	22.0006	Hydraulic actuator with separate control valve attached.	24-Jan-04	
	19:16:40	52.3635	22.0210	Side of body & cargo floor structure	24-Jan-04	
	19:21:04	52.3662	22.0279	Flap actuator with spindle attached	24-Jan-04	
	19:21:04	52.3662	22.0279	Passenger seat & dense debris	24-Jan-04	
	20:00:03	52.3653	22.0164	Large fuselage section, including belly skin and cargo compartment	24-Jan-04	
	20:05:45	52.3605	22.0207	Door with door lock actuator (P/N 65C255442-5)	24-Jan-04	
	20:30:00	52.3564	21.9926	Leading edge flap actuator with valve module attached.	24-Jan-04	RW19
	20:35:45	52.3579	21.9930	Flap attach structure	24-Jan-04	
	20:53:54	52.3617	22.0140	Spoiler mixer	24-Jan-04	RW20
	20:53:54	52.3617	22.0140	lateral override mechanism	24-Jan-04	
	20:53:54	52.3617	22.0140	Aileron PCU	24-Jan-04	RW18
	22:00:00	52.3525	22.0185	Landing gear brake and wheel tire assembly	24-Jan-04	
	22:04:20	52.3522	22.0115	Landing gear brake components and landing gear actuator (nose wheel steering?)	24-Jan-04	
	22:42:20	52.3585	22.0215	Significant structural element (?)	24-Jan-04	
	22:48:40	52.3660	22.0287	Structural fitting	24-Jan-04	

FSH604 Surveyed Wreckage Database  
(Ile de Batz)

T#	Time	Latitude	Longitude	Description	Date	Recovered Wreckage No.
	23:15:30	52.3548	22.0165	Landing gear actuator	24-Jan-04	
	23:18:50	52.3554	22.0148	Part of engine fuel system	24-Jan-04	RW21
	0:57:30	52.3544	22.0076	Flap angle gearbox	25-Jan-04	RW22
	1:15:30	52.3545	22.0155	White drive shaft	25-Jan-04	RW23
	1:43:10	52.3519	22.0227	Fractured actuator rod attached to structure	25-Jan-04	RW24
	1:49:20	52.3526	22.0179	Jackscrew of horizontal stabilizer	25-Jan-04	RW25
	20:53:54	52.3617	22.0140	Center section structural joint recovered with RW20	25-Jan-04	RW26

# **Exhibit E**

# **Attachment 6**

**Selected Wreckage Photos**

# Floating Wreckage











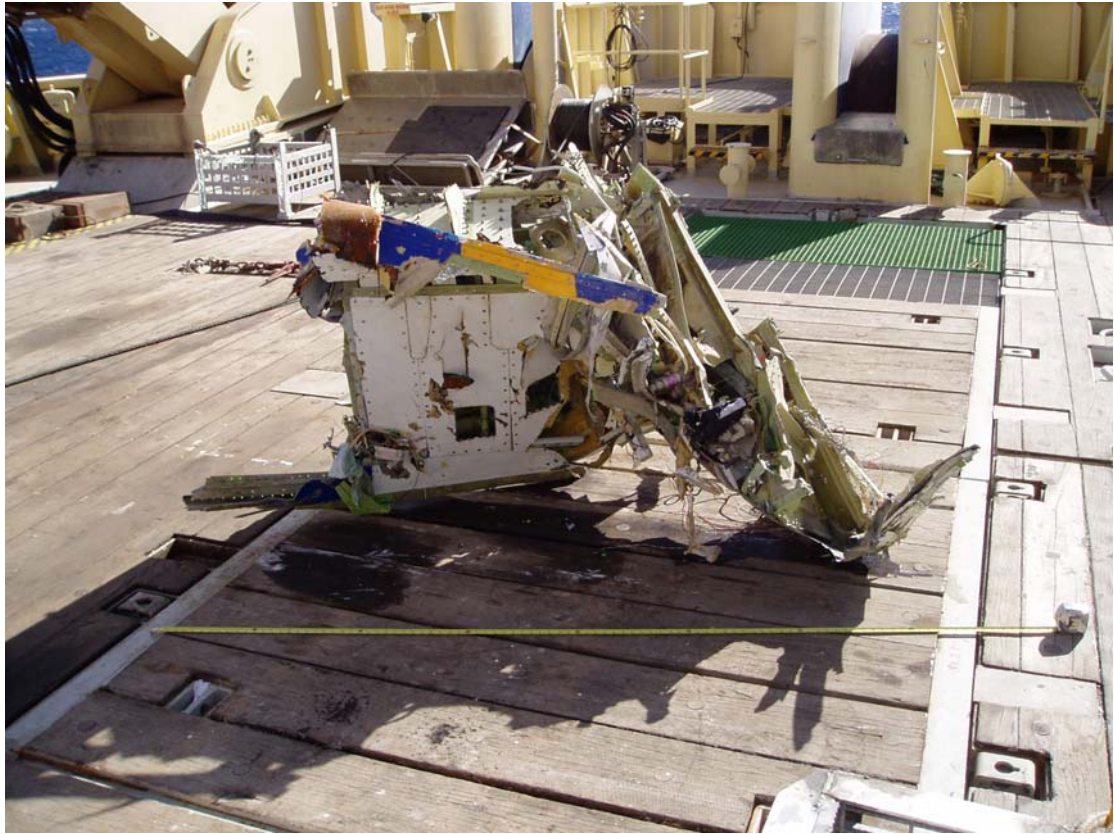


**Underwater Recovered Wreckage**















# **Exhibit F**

## **Operations Group Field Report**

January 22, 2004

## Group Chairman's Field Report

# OPERATIONS

### 1. ACCIDENT

Operator: Flash Airlines  
Location: Sharm-El-Sheikh, Egypt  
Date: January 3, 2004  
Time: 0246 UTC<sup>1</sup>  
Airplane: Boeing B-737-300, SU-ZCF, Serial Number 26283

### 2. SUMMARY

On January 3, 2004, about 02:45:06 UTC, 04:45:06 Local time, Flash Airlines flight FSH604, a Boeing 737-300, Egyptian registration SU-ZCF, crashed into the Red Sea shortly after takeoff from Sharm el-Sheikh International Airport (SSH) in South Sinai, Egypt. The flight was a passenger charter flight to Charles de Gaulle Airport (CDG), France with a stopover in Cairo international Airport (CAI) for refueling. Flight 604 departed from Sharm el-Sheikh airport with 2 pilots (Captain and First Officer), 1 observer, 4 cabin crew, 6 off-duty crew members and 135 passengers on board. The airplane was destroyed due to impact forces with the Red Sea with no survivals.

The airplane had departed from Sharm el-Sheikh runway 22R and was air born at 02:42:33 UTC, approximately 2½ minutes prior to the crash, and had been cleared for a climbing left turn intercept the 306 radial from the Sharm el-Sheikh VOR station located just north of runway 22R. This climbing turn allows departing flights to gain sufficient altitude before proceeding over higher terrain located along the flight path to Cairo. Flight 604 was operating in Egyptian airspace as a charter flight operating under the provisions of Egyptian Civil Aviation Regulations Part 121

### 3. DETAILS OF THE INVESTIGATION

The Operations group convened at 1100 on January 14, 2004 at the offices of the Ministry of Civil Aviation. An interview was conducted with the Chief Pilot of Flash Airlines regarding the pilot and co-pilot qualifications. Pilot training records were reviewed and information was collected to include medical and flying licenses and total flying time. A member of the operations group participated in the interview of the ground engineer that flew

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<sup>1</sup> All times are Universal Coordinated Time based on a 24-hour clock, unless otherwise noted. Actual time of accident is approximate, to be determined by the correlation of the Flight Data Recorder (FDR) and Air Traffic Control (ATC) transcripts.

on the airplane prior to the accident flight. A review of the weight and balance of the flight was conducted. Activities were concluded on January 22, 2003.

### **3.1 AIRPORT INFORMATION**

According to the Aeronautical Information Publication (AIP), Sharm El Sheikh International Airport was located 23 kilometers northeast of the city. The elevation of the airport was 143 feet mean sea level. The airport had two paved parallel runways; 04L-22R and 04R-22L. Both runways were 3081 meters in length and 45 meters in width. Runways 04R and 04L had CAT 1 Approach Lighting System and runways 22R and 22L had Simple Approach Lighting System. Neither runway had runway centerline lights.

According the AIP Flight procedures, there was no standard departures and standard arrival routes or any other systematic procedures established within. Sharm El Sheikh approach airspace, heading, flight level, speed and or holding instructions shall be specified in approach control clearances to arriving and departing flights as appropriate to meet the requirements of traffic conditions.

### **3.2 FLIGHT CREW INFORMATION**

Both flight crewmembers were certificated under Egyptian Civil Aviation Supervisory Sector Authority (ECASSA).

#### **3.2.1 Captain Khedr Abdalla Saad Said**

- Date of birth: February 26, 1950
- Date of hire with Flash Airlines: February 16, 2003
- Airline Transport Pilot Egyptian Certificate Number 561 (issued December 15, 1984)
  - Airplane Multiengine Land
  - Airplane Single Engine Land/Commercial Pilot
- Limitations: None
- Type Ratings: ATR-42, B-737/300/400/500 (issued May 27, 2003), DHC-5 Buffalo, C-130, Gornhorya.
  - Medical: First Class (issued November 19, 2003)
  - Limitations: None
  - Initial Ground School Training:
    - Written Test: April 9, 2003
    - Oral Test: May 22, 2003
  - Initial Simulator Training B-737-300/400/500: April 28 - May 12, 2003
  - Initial Proficiency Check B-737-300/400/500: May 12, 2003
  - Last Proficiency Check B-737-300/400/500: May 12, 2003
  - Last Line Check: July 23, 2003
  - Last Recurrent Training: December 16, 2003

- **FLIGHT TIMES:**

Total flight time (hrs/min) <sup>2</sup> :	7,443:45
Total flight time on B-737:	474:15
Total flight time PIC:	5,473:35
Military Instructor Flight time:	1,967:55
Total flight time last 24 hours <sup>3</sup> :	7:15
Total flying time last 30 days:	83:51
Total flying Time 90 days:	244:43

### 3.2.2 First Officer Amr Mahmoud Shafie

- Date of birth: January 1, 1979
- Date of hire with Flash Airlines: May 22, 2002
- Egyptian Commercial Pilot License Number 3284 (issued April 12, 1997), Commercial Pilot License issued by the Federal Aviation Administration (FAA) Certificate Number 2546582 (issued July 31, 1996)
  - Airplane Multiengine Land
  - Airplane Single Engine Land/Commercial Pilot
  - Instrument Airplane
  - Private Privileges
- Limitations: None
- Type Ratings: CESSNA (ISSUED April, 12, 1997) I, B737-200 (ISSUED July, 22,1998) II, B737-300/400/500 (ISSUED July, 18, 2002) II
  - Medical: First Class (issued May 5, 2003)
  - Limitations: None
  - Initial Ground School Training:
    - Written Test: June 10, 2002
    - Oral Test: May 22, 2002
  - Initial Simulator Training\_B-737-300/400/500: June 22 – June 30, 2002
  - Initial Proficiency Check B-737-300/400/500: June 30, 2002
  - Last Proficiency Check B-737-300/400/500: May 15, 2003
  - Last Line Check: July 11, 2002

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<sup>2</sup> Times are calculated for the captain up until December 31, 2003.

<sup>3</sup> Times do not include the accident flight.

- Last Recurrent Training: December 12, 2003
- FLIGHT TIMES:

Total flight time (hrs/min) <sup>4</sup> :	788:53
Total flight time B-737:	242:28
Total flying time last 24 hours <sup>5</sup> :	7:15
Total flying time last 30 days:	43:45
Total flying Time 90 days:	61:10

### 3.3 WEIGHT AND BALANCE

The Flash Airlines weight and balance calculations provided to the flight crew contained the following information<sup>6</sup>:

	Weight (kilograms)
Total Traffic Load	11,450 <sup>7</sup>
Dry Operating Mass	33,200
Actual Zero Fuel Mass	44,650
Maximum Zero Fuel Mass	47,627
Takeoff Fuel	7,000
Actual Takeoff Mass	51,650
Maximum Takeoff Mass (Certificate Limit)	63,276
Landing Mass	49,650
Maximum Landing Mass (Certificate Limit)	51,709

Zero Fuel Mass Center of Gravity (CG)	20.0%	
Zero Fuel Mass CG Limits <sup>8</sup>	8.0% Forward	28.4% Aft
Takeoff Mass CG	18.0%	
Takeoff Mass CG Limits <sup>9</sup>	6.7% Forward	27.9% Aft

<sup>4</sup> Times are calculated for the first officer up until December 31, 2003.

<sup>5</sup> Times do not include the accident flight.

<sup>6</sup> See attached Flash Airlines Load and Trim Sheet.

<sup>7</sup> A review of the Load and Trim Sheet indicated a low 100-kilogram error. The total cargo weight plus passenger mass (Total Traffic Load) should be 11,550 kilograms. Correspondingly, the Zero Fuel Mass, Takeoff Mass, and Landing Mass will be low in error by the same 100-kilogram Mass.

<sup>8</sup> Estimated Zero Fuel Mass CG limits were derived from Flash Airlines Load and Trim sheet index chart based upon a Zero Fuel Mass of 44,650 kilograms.

<sup>9</sup> Estimated Takeoff Mass CG limits were derived from Flash Airlines Load and Trim sheet index chart based upon a Takeoff Mass of 51,650 kilograms.





According to the Director, the prevailing winds at SSH require the use of runway 04L 70%-80% of the year. On the date of the accident, runway 04L was being used. However, sometime during the day prior to the accident, the runway was changed to 22R.

There was not an inspection of the runway after notification of the accident, however, it was stated that the landing airplane after the accident did not report debris on the runway. There is a daily runway inspection performed at SSH.

### **3.5 METEOROLOGY**

Sharm El-Sheikh does not provide Automatic Terminal Information Service (ATIS).

The SSH weather at 0200Z was reported as:

270 degrees at 06 knots, Ceiling and visibility OK (CAVOK) temperature 17 degrees Celsius, dewpoint minus 6 degree Celsius, altimeter 1011 hectoPascals (hPa), No significant change (NOSIG).<sup>11</sup>

The SSH weather at 0300Z was reported as:

280 degrees at 08 knots, Ceiling and visibility OK (CAVOK) temperature 17 degrees Celsius, dewpoint minus 6 degree Celsius, altimeter 1011 hectoPascals (hPa), No significant change (NOSIG).

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<sup>11</sup> See attached weather reports for SSH.