40<sup>th</sup> Joint Propulsion Conference & Exhibit Ft. Lauderdale, Florida 11–14 July 2004

AIAA 2004-3391

# **GRYPHON: A FEASIBLE HORIZONTAL TAKEOFF NEXT GENERATION ARCHITECTURE CONCEPT**

Gerald G. Sadler\*, Joseph M. Roche\*\*, and William J. Taylor\*\*\* NASA-Glenn Research Center, 21000 Brookpark Rd., Cleveland, OH 44135

Andrew M. Crocker<sup>†</sup>, Adam M. Wuerl<sup>††</sup> Andrews Space, Inc., 505 Fifth Avenue South, Suite 300, Seattle, WA 98104

### Abstract

NASA's Next Generation Launch Technology (NGLT) program recently funded studies of the Gryphon launch architecture. Gryphon combines existing technologies with moderate propulsion advancements and a design for operations approach into a unique, aircraft-like architecture. A joint NASA/industry team was assembled by NGLT's Systems Analysis Project to assess the Gryphon concept and produce a conceptual design to specific performance, reliability, safety, and life cycle cost goals while meeting NASA and DoD requirements. The effort demonstrated the fundamental feasibility of the architecture. This paper describes the study, outlines the architecture's features and benefits, and highlights the critical issues identified for further study.

#### Acronyms

| ACES | Air Collection and Enrichment System  |
|------|---------------------------------------|
| CFD  | Computational Fluid Dynamics          |
| DRM  | Design Reference Mission              |
| FOM  | Figure of Merit                       |
| GTOW | Gross Takeoff Weight                  |
| HTHL | Horizontal Takeoff Horizontal Landing |
| IVHM | Integrated Vehicle Health Management  |
| L/D  | Lift-to-Drag Ratio                    |
| LEO  | Low Earth Orbit                       |
| LOM  | Loss of Mission                       |
| LOP  | Loss of Payload                       |
| LOV  | Loss of Vehicle                       |
| LOX  | Liquid Oxygen                         |

\* NASA Project Manager, Senior Member AIAA

\*\* System Engineer

\*\*\* System Engineer

<sup>†</sup> Lead Project Manager, Senior Member AIAA

<sup>††</sup> Engineer, Member AIAA

Copyright © 2004 by Andrews Space, Inc. and NASA Glenn Research Center. Published by the American Institute of Aeronautics and Astronautics, Inc. with permission.

| NGC  | Northrop Grumman Corporation       |
|------|------------------------------------|
| NGLT | Next Generation Launch Technology  |
| POD  | Point of Departure                 |
| R&D  | Research & Development             |
| RLV  | Reusable Launch Vehicle            |
| SAP  | Systems Analysis Project           |
| SBIR | Small Business Innovation Research |
| SLI  | Space Launch Initiative            |
| TAT  | Turn Around Time                   |
| TPS  | Thermal Protection System          |
| TSTO | Two Stage To Orbit                 |

## **Introduction to Gryphon**

Horizontal Takeoff, Horizontal Landing (HTHL) architectures typically either use highly advanced technologies (such as hypersonic propulsion, advanced materials, etc.) or are incapable of delivering substantial payloads to orbit. The Gryphon Two-Stage-to-Orbit (TSTO) HTHL architecture, however, meets NASA and DoD payload without significant requirements technology advancements. This is made possible by an in-flight propellant collection system, the Alchemist<sup>TM</sup> Air Collection and Enrichment System (ACES), developed by Andrews Space, Inc. Alchemist ACES generates liquid oxygen (LOX) through separation of atmospheric air, which allows Gryphon to take off without LOX on board, minimizing vehicle takeoff weight.<sup>1,2</sup> Studies have shown that ACES, previously proposed for hypersonic combined cycle reusable launch vehicles (RLVs), is a higher payoff, lowerrisk technology if LOX generation is performed while the vehicle cruises subsonically.<sup>3</sup> This enables RLVs that operate with existing airbreathing and rocket propulsion systems, creating a paradigm shift in space operations.

#### **Mission Operations**

Figure 1 shows a nominal Gryphon trajectory. Both vehicle stages use liquid hydrogen and oxygen engines for rocket-powered flight. The second stage,