European Aviation Safety Agency

EASA

TYPE-CERTIFICATE DATA SHEET

AT-3 Series

Manufacturer: AERO Sp.z o.o

ul. Wał Miedzeszyński 844 03-942 Warszawa Poland

Variants:

AT-3 R100

Revision Record

Page	1	2	3	4	5	6
Issue	4	3	4	3	3	4

CONTENT

A: AT-3 R100

- A.I. General
- A.II. Certification Basis
- A.III. Technical Characteristics and Operational Limitations
- A.IV. Operating and Service Instructions
- A.V. Notes
- A.VI Revision record

<u>A:</u> <u>AT-3 R100</u>

A.I. General

Data Sheet No.: A.021 Issue: 04 Date: 24 July 2009

1. a) Type: AT-3 b) Variant: AT-3 R100

2. Airworthiness Category: Normal

3. Type Certificate Holder: AERO sp. z o.o

Wał Miedzeszyński 844 03-942 Warszawa

Poland

4. Manufacturer: AERO sp. z o.o

Wał Miedzeszyński 844

03-942 Warszawa

Poland

5. Certification Application Date: 24th September 2003

6. CAIB Certification Date: 12th February 2003

7. EASA Certification Date: 21 January 2005

A.II. Certification Basis

1. Reference Date for determining the applicable requirements: 11th March 2002

2. (Reserved)

3. (Reserved)

4. Certification Basis: As defined in CRI A-01, latest Issue

5. Airworthiness Requirements: EASA CS-VLA dated 14 November 2003

(Equivalent to JAR-VLA Issued 26 April 1990 including amendments up to VLA/92/1 dated 1

January 1992)

6. Requirements elected to comply: None

7. EASA Special Conditions: None

8. EASA Exemptions: None

9. EASA Equivalent Safety Findings: None

10. EASA Environmental Standards: ICAO, Annex 16, Volume 1, Chapter 10

A.III. Technical Characteristics and Operational Limitations

1. Type Design Definition: Master Drawing List, Document No. ATS3.02

dated 03.02.2003, Amendment 5 dated 16.05.2005

and subsequent

2. Description: Single engine, two-seater cantilever low wing

aeroplane, all metal construction, fixed tricycle

landing gear

3. Equipment: Equipment list, AFM, Document No. ATL3.03 or

ATL3.04, Section 2 and 6

4. Dimensions:

 $\begin{array}{ccc} \text{Span} & 7.55 \text{ m} \\ \text{Length} & 6.25 \text{ m} \\ \text{Height} & 2.23 \text{ m} \\ \text{Wing Area} & 9.30 \text{ m}^2 \end{array}$

5. Engine: Bombardier-Rotax 912S2 or S4

5.1 Engine Limits: Max take-off rotational speed 5800 r.p.m

Max continuous rotational speed 5500 r.p.m

For other engine limits refer to AFM, Document.

No. ATL3.03 or ATL3.04, Section 2

6. (Reserved)

7. Propeller

7.1 Type GT ELICHE GT-2/173/VVR-FW101SRTC

7.1.1 Diameter 1 730 mm

7.1.2 Kind Two-blade, wooden, constant pitch

7.1.3 Propeller configuration Tractor propeller

7.2 Type ELPROP 3-1-1P

7.2.1 Diameter 1 730 mm

7.2.2 Kind Three-blade ground adjustable propeller with

composite blades and metal hub

7.2.3 Propeller configuration Tractor propeller 7.2.4 Blade angle of incidence $21.5^{\circ} \pm 0.5^{\circ}$

8. Fluids:

8.1 Fuel: Minimum 95 Grade Unleaded Automotive Gasoline

or AVGAS 100 LL if other fuel is not available

8.2 Oil: Oils conforming to API classification marked SF or

SG

For more details see AFM, Document No. ATL3.03 or ATL3.04, Section 2

9. Fluid capacities:

9.1 Fuel: Total: 73.5 litres

Usable: 70.0 litres

9.2 Oil: Maximum: 3.5 litres

Minimum: 2.5 litres

10. Air Speeds:

Design Manoeuvring Speed V_A: 112 kt IAS

Flap Extended Speed V_{FE} : 85 kt IAS Maximum structural cruising speed V_{NO} 112 kt IAS

Never exceed speed V_{NE} : 127 kt IAS

11. Maximum Operating Altitude: Not defined

12. All Weather Capability: Day-VFR

13. Maximum Masses:

Take-off 582 kg Landing 582 kg

14. Centre of Gravity Range:

Forward limit up to 450 kg 0.229 m behind Datum

at 582 kg 0.305 m behind Datum varying linearly with mass in between

Rear limit: for all masses 0.394 m behind Datum

15. Datum: Wing Leading Edge

16. (reserved)

17. Levelling Means: Spirit Level on the cockpit side rail with canopy

open

18. Minimum Flight Crew: 1 (Pilot)

19. Maximum Passenger Seating Capacity:

20. (Reserved)

21. Baggage / Cargo Compartments

Location Max. allowable Load

Port Side Compartment 20 kg Starboard Side Compartment 10 kg

22. Wheels and Tyres

Nose Wheel Tyre Size 5.00 - 4 6ply Type III Main Wheel Tyre Size 380x150/15x6.00 - 5

A.IV. Operating and Service Instructions

Aeroplane Flight Manual (AFM) Document No. ATL3.03, Polish Language version

and ATL3.04, English Language version

Aeroplane Maintenance Manual (AMM) Document No.ATT3.02, Polish Language version

and ATT3.03, English Language version(incl.

Airworthiness Limitations)

A.V. Notes

- 1. This certification applies to:
 - Serial numbers AT3-008 and AT3-011 and subsequent,
 - Serial numbers from AT3-001 to AT3-005 and AT3-010 modernized according to the Remark No1 in the Master Drawing List, Document No ATS3.02 dated 03.02.2003, Amendment 5 dated 16.05.2005
- 2. Approved Noise Levels in accordance with:

ICAO Annex 16, Volume 1, Chapter 10:

66.6 dB(A)

A.VI Revision Record

- Issue 1 Initial issue.
- Issue 2 AT-3R100 airplanes S/N AT3-001 to 005 and 010 differ from airplanes S/N AT3-008 and 011 and subsequent only in standard equipment; that the part of optional equipment of these airplanes (S/N AT3-001 to 005 and 010) became standard equipment on airplanes S/N AT3-008 and 011 and subsequent. Accordingly, their Airplane Flight Manual have been changed. Additionally, the AFM of the airplanes S/N AT3-008 and 011 and subsequent took into account Imperial units.

Modification of the airplanes S/N AT3-001 to 005 and 010 to the standard of the airplanes S/N AT3-008 and 011 and subsequent includes an installation on these airplanes of optional equipment which constitutes standard equipment on airplanes S/N AT3-008 and 011 and subsequent. It also includes the change in the AFM.

- Issue 3 Addition of alternative ELPROP 3-1-1P three-blade ground adjustable propeller with composite blades and metal hub, approved as a major change by EASA approval EASA.A.C.01865 dated 11 October 2005.
- Issue 4 Clarification that the certification basis is CS-VLA, identical to JAR-VLA on the date of certification.