NAVAIR 00-110AH53-2

## Standard Aircraft Characteristics

# NAVY MODEL CH-53D AIRCRAFT

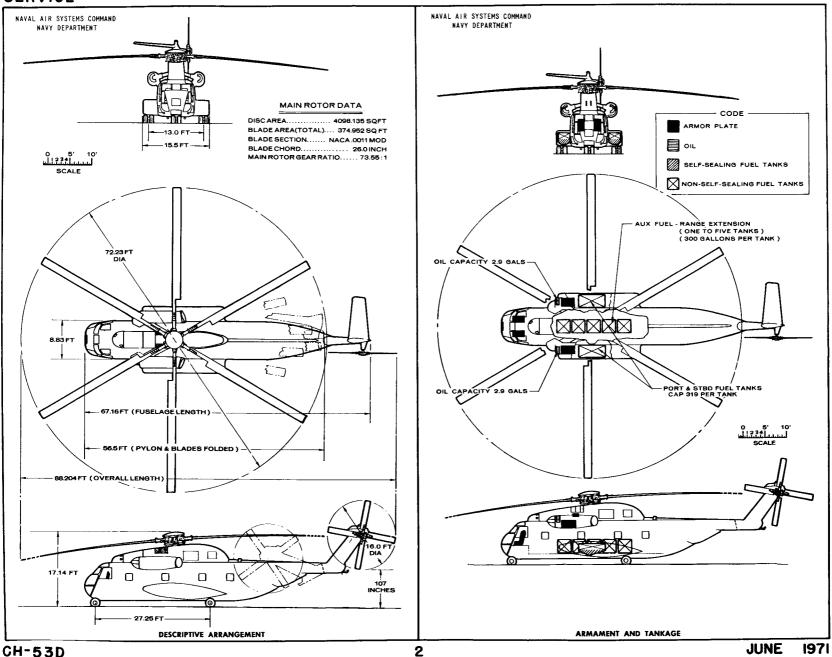
PUBLISHED BY DIRECTION OF THE COMMANDER OF THE NAVAL AIR SYSTEMS COMMAND



# STANDARD AIRCRAFT CHARACTERISTICS

**CH-53D** 

SIKORSKY



#### **POWER PLANT**

No. and Model: (2)T64-GE-413 Manufacturer: General Electric

Engine Spec. No.: E1159(19 March 1969) Type: Axial

Length:

78.8 inches

Diameter:

23.8 inches

Gear Ratio (Eng/Rotor) 73.55 to 1

#### **RATINGS**

S. L. STATIC	SHP	RPM	MIN.
Max.	*3925	13600	10
Military	3695	13600	30
Normal	3230	13600	Cont.

\* See Note Performance Summary page for transmission ratings.

#### **ELECTRONICS**

AN/ARC-51A Radio Set (UHF) AN/ARN-52 Tacan Navigation Set AN/APX-64 IFF Transponder Set AN/ARC-94 Radio Set AN/ARN-59 ADF AN/ARC-54 or 131 Radio Set (VHF) AN/AIC-14 Interphone System AN/APN-154 Radar Beacon Set AN/APN-171 Radar Altimeter ID-351 or 387/ARN Course Indicator ID-663A/U or B/U or C/U Bearing Distance Heading Indicator

#### MISSION AND DESCRIPTION

Primarily designed as an assault transport, the helicopter is employed in the movement of cargo and equipment and in the transportation of troops. When appropriately equipped, it may be used in the recovery of downed aircraft and personnel, performing mine countermeasure missions, and towing of vehicles

The twin turbine engine helicopter uses a single main rotor and a single anti-torque tail rotor. The blades are all metal construction. Main rotor blades are equipped with Sikorsky BIM to eliminate mandatory blade retirement. Conventional helicopter controls are provided for both pilot and copilot. The mechanical controls are augmented by two parallel and independent hydraulic servo systems. An automatic flight control system (AFCS) is also provided. Landing gear is retractable. Main rotor blades and tail pylon fold for stowage aboard an aircraft carrier. To facilitate cargo loading, the aircraft is equipped with a hydraulically operated rear ramp,

two cargo winches, roller conveyors, and tie-down facilities.

#### **DEVELOPMENT**

First Flight:		
First Service Use:		
Production Status:	In	Production

#### **DIMENSIONS**

Main Rotor Dia.	
Length (blades & pylon folded)	
Height (blades & pylon folded)	
No. of Blades, main	
Blade Area (each)	62.5 sq. ft.
Disc Area	4098.1 sq. ft.
Main Wheel Tread	13'-0"
Max. Width (Main Blades and Pylon Folded)	)15'-6"

### **WEIGHTS**

LOADING	POUNDS	L.F.
Empty (A)	23628	
Basic	23634	
Design	33500	3.0
**Design Alternate	42000	2.39
*Combat	34958	•
*Take-off	36693	

(A) Actual

Crew (Normal)

For Basic Mission ECP 6144 Part II

#### **FUEL AND OIL**

LOCATION	NO. TANKS		GAL.
L. Sponson* R. Sponson*	1		319 319
Cabin**	5	TOTAL	<u>1500</u> 2138
Specification *Bottom third self-	sealing	JP-4 MIL-J	., JP-5 -5624D
**Aux. tanks for rar	nge extension	n	
Nacelles		2 (to	t. 5.8)

#### **ACCOMMODATIONS**

Cargo (basic mission) 8000 l	b.
Cargo (overload mission) 12742 l	b.
Troops	
Litters 24	
Cabin Size Clearance:  Length	

BAT RANGE FERRY RANGE SION III MISSION IV	RETRIEVAL (TROPICAL DAY)* MISSION V
93 41513	28693
14538	4338
0/0 0/0	0/9289**
10.13	7.00
0/1740 560/820	2110/2325
0/7200 1650/2850	7900/8900
0/2460 1855/2150	2320/2710
50 13300	17500
157	172
S.L. 127/8000	169/3000
886	
135	
8000	
	. 106
	138
	3000
6.62	1.55
1	6.62

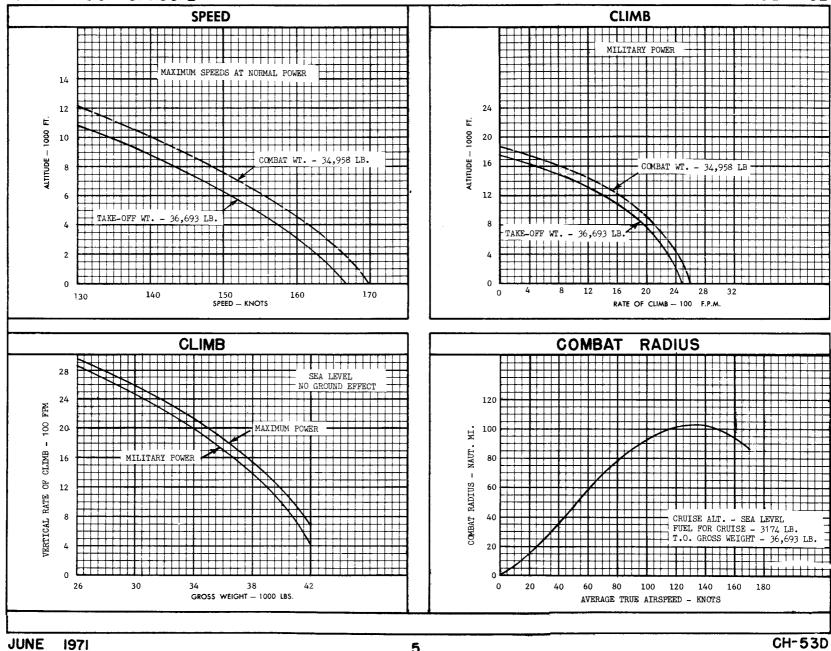
- NOTES: (A) Normal power
  - (B) Military power
  - (C) Maximum power
  - Tropical Day: 91.5°F. at 3000 ft. cruise altitude.
  - Inbound payload is carried externally ( $\Delta f = 35 \text{ sq. ft.}$ )

#### Performance Basis:

- (1) ICAO Standard conditions (except Mission V), no wind, no ground effect.
- (2) Calculated data based on Navy flight tests on CH-53A helicopter.
- (3) Range and radius based on General Electric specification fuel consumption data using fuel grade JP-5.
- (4) Fuel consumption data are increased 5% above engine specification values.
- (5) Transmission ratings are 7560 HP dual engine and 3780 HP single engine operation.
- (6) Aircraft red line airspeed is 170 knots IAS.
- (7) Weight data based on "Actual Weight and Balance Report", SER-65575 dated 10 March 1970.
- (8) Performance reference: Sikorsky Report, SER-65583, "Substantiating Data for Standard Aircraft Characteristics and Performance Charts for CH-53D Helicopter."



#### SERVICE



#### NOTES

#### BASIC ASSAULT

Warm-Up & Take-Off: 2 min. at S.L., NRP

Cruise Out: At 150 kts. airspeed at S.L. to remote base

Hover Over Remote Base: At S.L. out of ground effect, for 10 min.

Land At Remote Base: Discharge 8000 lbs. and accept 4000 lbs. for return

Warm-Up & Take-Off: 2 min. at S.L., NRP

Cruise Back: At 150 kts. airspeed at S.L.

Reserve: 10% of initial fuel load

RADIUS

#### OVERLOAD ASSAULT

Warm-Up & Take-Off: 2 min. at S.L., NRP

Cruise Out: At 150 kts. airspeed at S.L. to remote base

Hover Over Remote Base: At S.L. out of ground effect for 10 min.

Land At Remote Base: Discharge 12,742 lbs. and accept 4000 lbs. for return

Warm-Up & Take-Off: 2 min. at S.L., NRP Cruise Back: At 150 kts. air-

speed at S.L.
Reserve: 10% of initial fuel

RADIUS

#### DUD RETRIEVAL (TROPICAL DAY)

Warm-Up & Take-Off: 5 min. at S.L., 90°F, NRP Climb: On course to 3000 ft. 91.5°F with Mil power

Cruise Out: At best range speeds to remote base

Hover Over Base: Out of ground effect, at 3000 ft., 91.5°F for 10 min. Pick up external maximum payload (0.G.E. hover, △ f=35 ft²)

Cruise Back: At best range speeds 3000 ft., 91.5°F Descend: To S.L. (no fuel used, no distance gained)

Reserve: 10% of initial fuel load

#### COMBAT RANGE

Warm-Up & Take-Off: 5 min. at S.L., NRP Cruise Out: At S.L. at best range speeds until reserve fuel remains

Reserve: 10% of initial fuel load



#### FERRY RANGE

Warm-Up & Take-Off: 5 min. at S.L., NRF Climb: On course to 8000 ft.

with Mil power Cruise Cut: At best range speeds until reserve fuel

remains
Descend: To S.L. (no fuel used,
no distance gained)
Reserve: 102 of initial fuel

Reserve: 10% of initial fuel load

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