#### Standard Aircraft Characteristics

## NAVY MODEL T-2B AIRCRAFT

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PUBLISHED BY DIRECTION OF THE COMMANDER OF THE NAVAL AIR SYSTEMS COMMAND

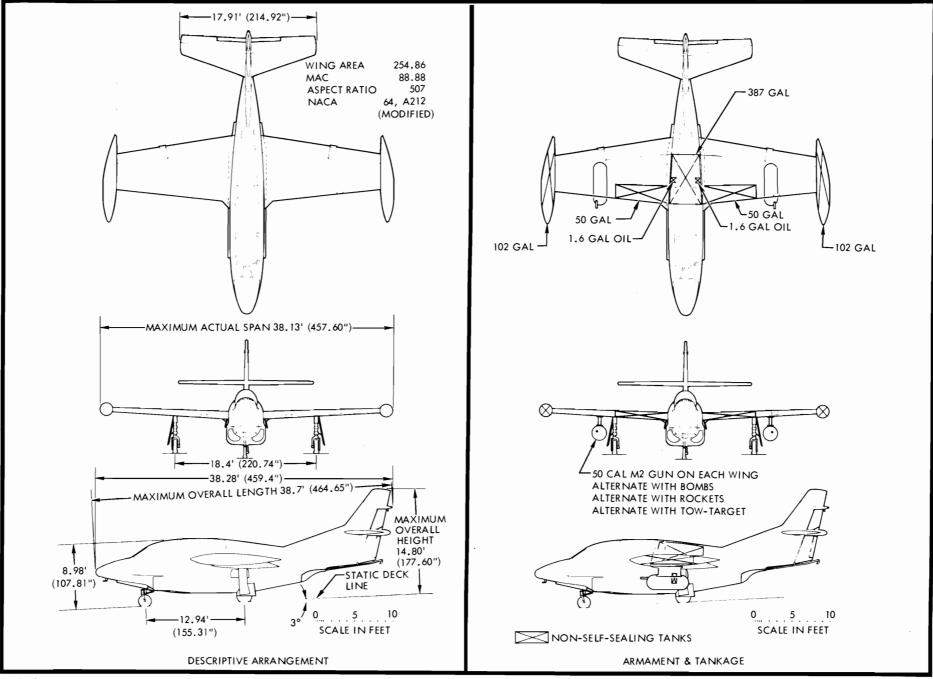
**JANUARY 1970** 



# STANDARD AIRCRAFT CHARACTERISTICS T-2B TRAINER

NORTH AMERICAN AVIATION, INC.

T-21



T-2B

NAVAIR 00-110 AT2-2				SERVIC	
POWER PLANT	MISSION AND DESCRIPTION	WEIGHTS			
NO. & MODEL (2) J60-P-6 MFR. Pratt & Whitney Aircraft ENG. SPEC. NO. N-2444	This airplane is a follow-on production configuration of the Model YT-2B (T2J-2) Prototype Trainer which was a development of a dual	LOADING Empty (\$)	LBS 8220	LF	
TYPE	engine configuration of the T-2A (T2J-1) Airplane. The primary purpose of this airplane is to provide performance and versatility so as to logically and systematically present the complexities of modern service aircraft.	Deisgn	9915	6.5	
DIAMETER (APPROX.) 22.00 in. TAIL PIPE NOZZLE (TYPE) Fixed Covergent	The T–2B Airplane incorporates in its design an NACA low–drag,	Basic	13, 284	4.85	
RATING	laminar-flow airfoil section, a power-operated, clam-shell type canopy, speed brakes, a catapult hook and an arresting gear. The wing incorporates	Combat	not applicab	le	
LBS THRUST MAX RPM ALT MIN  MIL 3000 16,700 SL 30	single-slotted flaps. Flight controls, excluding the mechanical rudder system, are power operated (elevator power-boosted and aileron irrevers-	Max. Take-Off	14,048	4.59	
NOR. 2570 15,750 SL - MAX (S) 2905 - SL 30	ible system). A large baggage compartment is located in the nose section.	Max. Landing	13,661	(field)	
T.O. 2905 - SL 30  (S) Installed rating, limited by full power	A low level escape system is also featured in the T-2B Airplane which provides safe ejection throughout the flight envelope down to zero altitude	Max. Landing	11,958	(carrier)	
lever position.	and a speed of 75 knots.	(S) SD-524-2-1			
ORDNANCE	Flaps and landing gear may be extended up to a speed of 165 knots.  The maximum permissible dive speed above 7500 feet is Mach number  0.85. The maximum structural design speed is 485 knots.	FUEL AND OIL			
NO. CAL. ROUNDS LOCATION 2 .50 100 each Wings			FUEL	-	
	Mfg. Model: NA-288, 291, and 294	LOCATION	TANKS	GAL.	
Removable gun package, P/N 200-89045, one gun package location under each wing (W.S. 134.5) Gun camera, (1) type KB-9A location on gun sight.	First Flight Date: 21 May 1965	Fuselage Wing Leading Edge Tip-Tanks	1 2 2	387 100 204	
BOMBS		Grade	J	IP-4 or JP-5 MIL-J-5624D	
External, one location under each wing (W.S. 134.5)			OIL "	VIIL-J-3024D	
NO. TYPE 2 MK 86 practice bombs		Engine Tank Specification	2	3.2 MIL-L-23699	
or 2 MK 15 practice bombs or 2 A/A37B-3 practice racks and 12 MK76		ELEC	ELECTRONICS		
Mod 4 or 5 practice bombs or 2 A/A 37B-3 practice racks and 12 MK			, I KO IVIC	<u> </u>	
106-3 practice bombs  ROCKETS		IFF/SIF Identification UHF Command	n	A N/APX-64 A N/ARC-52X	
External, one location under each wing (W.S. 134.5)		TACAN Equipment Receiving Set		A N/AR N-52(V) A N/ARR-40	
	DIMENSIONS	Direction Finder Interphone		A N/ARA-25A C-2379/AIC	
NO. SIZE TYPE 2 2.25 in. Practice Rocket		Provisions for: Fire Control System		AN/AWG-6	
14 2.75 in. Mighty Mouse Rocket 2 2.75 in. Single-Mounted	Wing Area 254.86 Sq. Ft. Span 38.13 Ft. Length 38.28 Ft.	The conner system		(MK6 MOD 4 + AN/APG-30A)	
Fire control system MK-6 MOD 4 including MK-8 MOD 9 gunsight. Guns can be carried in lieu of, but not in combination with, bombs or rockets. Provisions for aerial tow target.	Height 14.80 Ft. Tread 18.40 Ft.				

T-2B

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TAKE-OFF LOADING COND	ITION		BASIC TRAINER (Tiptanks)	BASIC TRAINER & ROCKETS	BASIC TRAINER & GUN PACKAGES	BASIC TRAINE
TAKE-OFF WEIGHT		lb.	13, 284	13,771	13,771	14,048
Fuel Internal/External		lb.	3166/1326	3166/1326	3166/1326	3166/1326
Payload		lb.	0	252	60	300
Wing loading		lb./sq.ft.	52.0	54.0	54.0	55.2
Stall speed - power-off		kn.	99.5	101.5	101.5	102.5
Take-off run at S.L. – calm	(A)	ft.	1460	1550	1550	1630
Take-off run at S.L. 25 kn. wing	(A)	ft.	920	1000	1000	1040
Take-off to clear 50 ft clam	(A)	ft.	2300	2450	2450	2540
Max. speed/altitude	(B)	kn./ft.	472/4000	464/6000	439/15000	434/15000
Rate of climb at S.L.	(B)	fpm.	6100	5800	5550	5300
Time: S.L. to 20,000 ft.	(B) (C)	min.	4.5	4.8	5.2	5.4
Time: S.L. to 30,000 ft.	(B) (C)	min.	8.1	8.7	10.2	10.8
Service ceiling (100 fpm)	(B)	ft.	42,600	41,400	39,950	39,250
Combatrange	(D)	n.mi.	966 (1)	903 (1)	758 (I)	720 (I)
Average cruising speed		kn.	360	356	352	345
Cruising altitude (avg)		ft.	44,000	43,000	41,500	40,600
Combat radius		n.mi.	498 (11)	337 (111)	235 (IV)	165 (V)
Average curising speed out/inbound		kn./kn.	360/360	356/356	352/352	238/263
Mission time		hr.	3.21	2.53	2.19	1.90
COMBAT LOADING CONDI	TION					
COMBAT WEIGHT	(E)	lb.	11,487	11,722	11,914	11,951
Engine power			max. power	max. power	max. power	max. power
Fuel		lb.	2695	2695	2695	2695
Combat speed/combat altitude	(B)	kn./ft.	462/25000	465/5000	437/25000	432/5000
Rate of climb/combat altitude	(B)	fpm/ft.	3200/25000	5920/5000	2350/25000	5280/5000
Combat ceiling (500 fpm)	(B)	ft.	42,800	41,900	40,100	39,750
Rate of climb at S.L.	(B)	fpm.	7150	6900	6450	6380
Max. speed at S.L.	(B)	kn.	472	464	432	427
Max. speed/altitude	(B)	kn./ft.	472/S.L.	465/8000	442/15000	436/15000
LANDING WEIGHT		lb.	9660	10091	10123	10428
Fuel		lb.	788	812	854	872
Stall speed-power-off/with approach pow		kn/kn.	85/85	86.5/86.5	87/87	88/88

Landing Distance-grd, roll/over 50 foot obst.

NOTES: (A) Take-off Power
(B) Maximum Power
(C) Allows for Weight Reduction During Ground Operations and Climb

ft/ft.

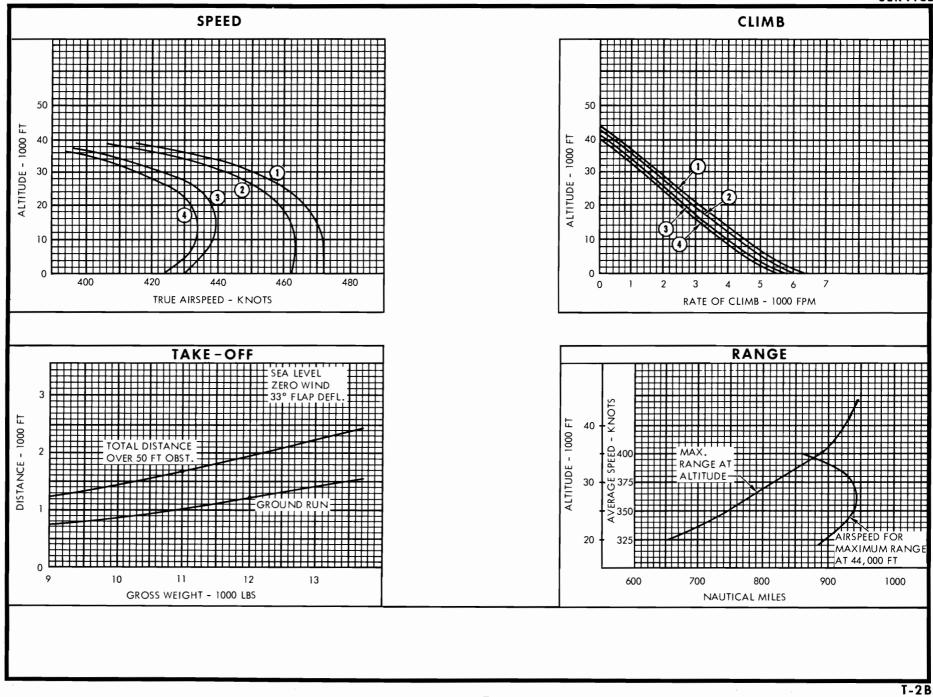
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(D) Ranges are Based on Flight Test Fuel Consumption Data
(E) Represents Combat Training Missions (Less Pay Load & 40% Fuel)

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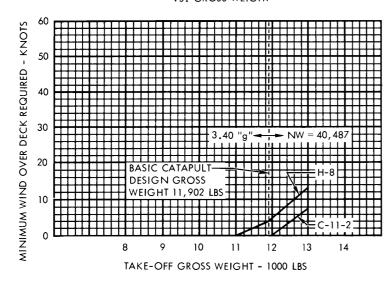
External Fuel is Carried in Fixed Tip Tanks

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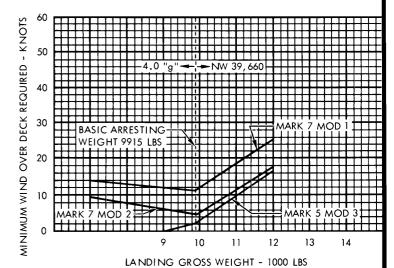


#### CARRIER SUITABILITY

MINIMUM WIND OVER DECK REQUIRED FOR CATAPULTING VS. GROSS WEIGHT



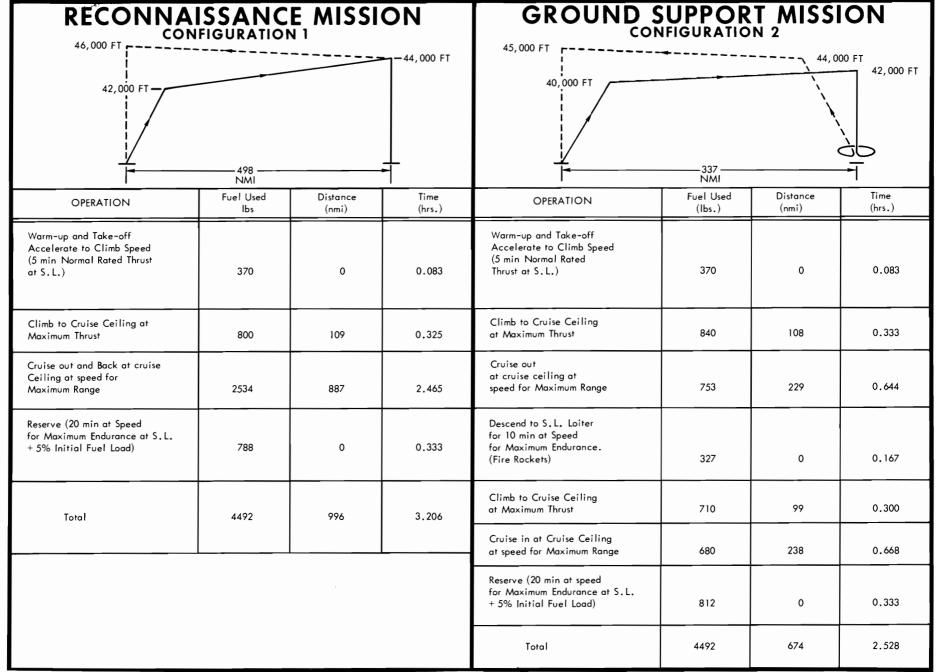
MINIMUM WIND OVER DECK REQUIRED FOR LANDING VS. GROSS WEIGHT



NOTES:

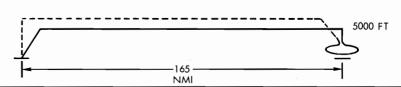
 No Wind Over Deck is Required for C-7, C-11 and C-13 Catapults.

 Approach Speeds are Based on Approach Speeds Recommended from Ship Board Carrier Suitability Trails for T-2A (T2J-1) Airplane.



### GENERAL PURPOSE AND ESCORT MISSION CONFIGURATION 3

### LOW ALTITUDE ATTACK MISSION CONFIGURATION 4



OPERATION	Fuel Used (lbs.)	Distance (nmi)	Time (hrs).	OPERATION	Fuel Used (Ibs.)	Distance (nmi)	Time (hrs.)
Warm-up and Take-off Accelerate to Climb Speed (5 min Normal Rated Thrust at S.L.)	370	0	0.083	Warm-up and Take-off Accelerate to Climb Speed (5 min. Normal Rated Thrust at S.L.)	370	0	0.083
Climb to Cruise Ceiling at Maximum Thrust	890	102	0.347	Climb to 5000 feet at Maximum Thrust	85	4	0.017
Cruise out at Cruise Ceiling at speed for Maximum Range	511	133	0.378	Cruise out at 5000 feet at speed for Maximum Range	1463	161	0.676
Descended to 35,000 feet for 20 min combat (max fuel flow) expend ammo	631	0	0.333	Descended to S.L. Loiter for 10 min at speed for maximum endurance. Drop Bombs	344	0	0.167
Climb to Cruise Ceiling at Maximum Thrust	760	95	0.316	Climb to 5000 feet at Maximum Thrust	70	4	0.013
Cruise back at Cruise Ceiling at speed for Maximum Range	476	140	0.398	Cruise back at 5000 at speed for Maximum Range	1288	161	0.613
Reserve (20 min at Speed for Maximum Endurance at S.L. + 5% Initial Fuel Load)	854	0	0.333	Reserve (20 min at speed for Maximum Endurance at S.L. + 5% Initial Fuel Load)	872	0	0.333
Total	4492	470	2.188	Total	4492	330	1.902



 $\pm \rm U.S.$  Government printing office: 1970-393-871/S-16608