MICROLIGHT TYPE APPROVAL DATA SHEET (TADS)

NO: BM-8 ISSUE: 4

TYPE:		Hybred 44 XL
(1)	MANUFACTURER:	Medway Microlights, Burrows Lane, Middle Stoke, Rochester, Kent, ME 3 9RN
(2)	UK IMPORTER:	N/A
(3)	CERTIFICATION:	Section S Issue: Advance Copy March 1983 List of Section S Blue Papers and Additional CAA Requirements: NIL
(4)	DEFINITION OF BASIC STANDARD:	Design Specification Report 2 Issue 1

(5) COMPLIANCE WITH THE MICROLIGHT DEFINITION

(a) MTOW	342 kg
(b) No. Seats	2
(c) Maximum Wing Loading	19.61 kg/m ²
(d) Vso	22 kts IAS
(e) Permitted range of pilot weights	55 – 90 kg per seat.
(f) Typical Empty Weight (ZFW)	147 kg
(g) ZFW + 172 kg crew + 1 hr fuel (16 litres / 11.5 kg)	330.5 Kg
(h) ZFW + 86 kg pilot + full fuel (20 litres / 14.5 kg)	247.5 kg
(i) Max ZFW at initial permit issue	150 kg

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(6) POWER PLANTS				
Designation	44 XL	44 XL	44 XL	
Engine Type	Fuji Robin EC 44 PM Inverted	Fuji Robin EC 44 PM	Rotax 447 Inverted	
Reduction Gear	2.9:1 Toothed Belt	2.9:1 Toothed Belt	2.58:1	
Exhaust System	Nicklow	Nicklow	Southdown International (Raven Int)	
Intake System	-	-	-	
Propeller Type	Geoff Farr Wood Laminated	Newton Wood	Cyclone Pattern	
Propeller Dia x Pitch	58" x 33"	62" diameter x 36" pitch	61" x 32"	
Noise Type Cert No.	6M Issue 1	6M issue2	116M Issue 3	
AAN approving configuration	19117	19672	20884	

(7) MANDATORY LIMITATIONS:

(A) Max Take-Off Weight		342 kg		
(B) CG Limits		N/A		
(C) CG datum		(see annex C	C).	
(D) Cockpit Loadings		Front	Rear	Total
	Min	55 kg	90 kg	55 kg
	Max	90 kg 128 kg	90 kg -	180 kg 128 kg
(E) Never Exceed Speed		57 knots		
(F) Manoeuvring Speed		57 knots		
(G) Permitted Manoeuvres		30° Nose up Non Aeroba Normal acce Max bank ar	/ 30° nose do tic eleration limit ngle 60°	own s, +4 / 0g
(H) Fuel Contents (Max Useable)		4.5 Imp. Gal	lons	

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(I) Power Plant		See Table		
Engine	Fuji Robin EC 44 PM Inverted	Rotax 447 Inverted		
Max RPM	6500	6250		
MAX CHT	-	-		
MAX EGT	-	-		
Fuel Spec	4	-star /MOGAS lea	ided fuel to BS 404	0
Engine Oil Spec	Filtrate 2 Stroke	2 Stroke		
Gearbox oil spec				
Fuel/Oil Mix	32:1	50:1		
Coolant Temperature				
Oil Pressure	-	-		
Oil Temperature	-	-		
Fuel Pressure				

(8) INSTRUMENTS REQUIRED:

ASI	Altimeter	RPM	CHT / EGT	Compass	Coolant temp	Fuel Pressure	VSI	Slip ball
(to 100 mph min.)	Required			Required	k			

(9) CONTROL DEFLECTIONS: N/A

Elevator UP:	<u>±</u>	Tailplane trim UP:	Ŧ
Elevator DOWN:	±	Tailplane trim DOWN	±
Ailerons* UP:	±	Rudder LEFT:	±
Ailerons* Down:	<u>+</u>	Rudder RIGHT:	<u>+</u>

Pitch control Pitch - Control bar 22 ins rear from front Strut.

Ailerons Roll - Control Bar 28 ¹/₄" left and right of aircraft axis.

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(10) PILOT'S NOTES, MAINTENANCE MANUALS REFERENCES:

- 10.1 Manuals approved for use with this aircraft.
- (a) Owners Instruction Manual No. M1 Issue 1
- 10.2 The following placards are to be fitted:-
- (a) Flight Limitations Placard (to be visible to pilot) See Annex D.
- (b) Engine Limitations Placard (to be located near to engine instruments) See Annex D.
- (c) Fuel Limitations Placard (to be located near to filler cap)

A placard is to be fitted showing fuel capacity (litres), fuel type(s), fuel:oil ratio (if relevant) and if MTOW can be exceeded with full fuel and maximum cockpit weight, the fuel loads at MTOW for cockpit weights of 180kg / 170kg / 160kg etc. at 10kg intervals down to the maximum fuel load. An example is shown at Annex D.

(d). Switches See Annex D.

(11) MANDATORY MODIFICATIONS / SERVICE BULLETINS / AIRWORTHINESS DIRECTIVES ETC:

See Annex A for required modifications.

Annual Bettsometer test is to be carried out to 1150 grammes with wing sails fitted and tensioned to flight. Test must be to both upper and lower surfaces.

(12) MINIMUM PERFORMANCE AT MAX TAKE-OFF WEIGHT

Rate of Climb: 440 ft/min

Stall or Minimum Flying Speed: 22 kt

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Issue History

	Issue No.	Reason and signatory
1	28/06/85	Initial Issue W A Bevan
2	28/10/86	Addition of noise information W A Bevan
3	05/08/88	Alternative propeller with Fuji engine and addition of Rotax 447 engine W A Bevan
4	17/02/04	Addition of optional major modifications and MPD for upper torso restraint.

J C Barratt

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Illustration of Aircraft - 3 View



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ANNEX A – MANDATORY MODIFICATIONS

1. MPD 2001-006 Upper Torso Restraint for Passengers

ANNEX B - APPROVED OPTIONAL MODIFICATIONS

The installation of all optional modifications is to be inspected by a BMAA inspector and an entry made in the appropriate logbook(s). Note that other approved modifications may exist which are not listed here.

Mod	Description
MM3	Introduces a Newton 62" diameter x 36" pitch wooden propeller and Operator Handbook Supplement No. 1, Issue 1.
MM4	Introduce Rotax engine with Typhoon wing
MM13	Electric Carburettor Heater

ANNEX C

WEIGHING INFORMATION

1.	CG Datum:	N/A
2.	Weighing attitude:	N/A
3.	Mainwheel moment arm:	N/A
4.	Nosewheel moment arm:	N/A (units) (direction) of datum
5.	Fuel moment arm:	N/A (units) (direction) of datum
6.	Crew moment arm:	N/A (units) (direction) of datum
7.	Crew weights:	Minimum 55 kg / maximum 90 kg (maximum reducible, not below 86 kg, if required).
8.	Aft CG Limit:	N/A (units) (direction) of datum
9.	Fwd CG Limit:	N/A (units) (direction) of datum

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ANNEX D

EXAMPLE PLACARDS

(a) Flight Limitations Placard (to be visible to pilot)

[Type] [Engi	ine] [Registration]_
Never Exceed Speed:	57 kts IAS
Manoeuvring Speed :	57 kts IAS
Stall Speed:	22 kts IAS
Best climb speed:	45 kts IAS
Best glide speed:	40 kts IAS
Pitch Limits:	30° nose down, 30° nose up.
Bank angle limits:	+/ - 60°
Maximum Stall entry rate:	1 kt/s
Normal Acceleration Limits:	+4 / 0g
Empty Weight:	147 kg *
Max Take-Off Weight:	342 kg
Minimum Cockpit Weight:55 kg	
Maximum Cockpit Weight:	90 kg in each seat.*
Aerobatics and deliberate spinning	prohibited.

* This must match the most recent W&CG report for the aircraft.

(b) Engine Limitations Placard (to be located near to engine instruments)

A placard showing the limitations for all indicated engine parameters is to be mounted close to the engine instruments. This requirement need not be complied with for limitations shown as coloured markers (red for danger, amber for caution) on the instrument displays.

(c) Fuel Limitations Placard (to be located near to filler cap)

FU	EL
Capacity	_ Litres
(50:1 2 Stroke oil /	do not add oil)
Cockpit Weight (kg)	Max. Fuel Load (litres)
180	
170	
Or below	Full fuel
4-star MOGAS leade	ed fuel to BS 4040

(d) Switches

All switches are to be marked with function and sense (up=on, down=off).