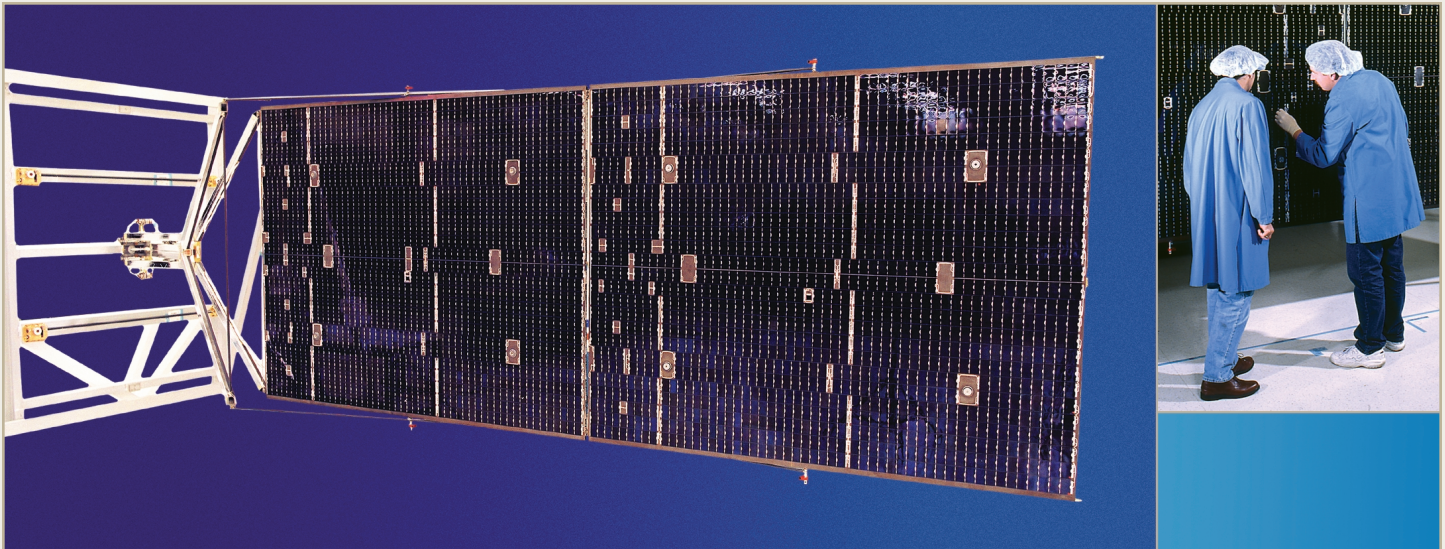


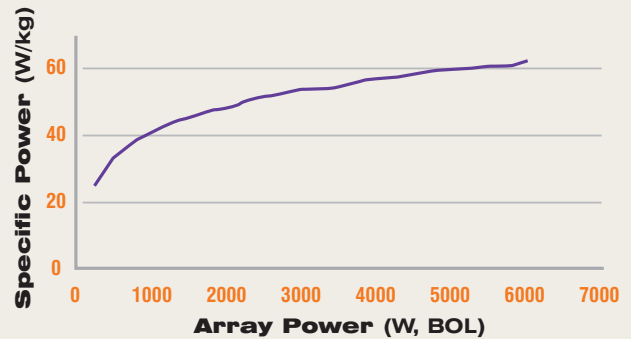
A Heritage-Technology Rigid Substrate Solar Array for Traditional Applications



Performance Features

- *GaInP/GaAs/Ge or Silicon Photovoltaics*
- *Carbon Fiber Structures*
- *Structural or Cable Synchronized Deployment with Non-Pyrotechnic Release*
- *Provides Stowed Power*
- *Long or Transverse Axis Deployment*
- *1/2 kW to 6 kW Power*
- *25 W/kg to 65 W/kg*
- *High Stiffness (0.3 Hz to 0.7 Hz)*
- *Mass Moment of Inertia (~300 kg-m²)*

Specific Power Performance



Applications

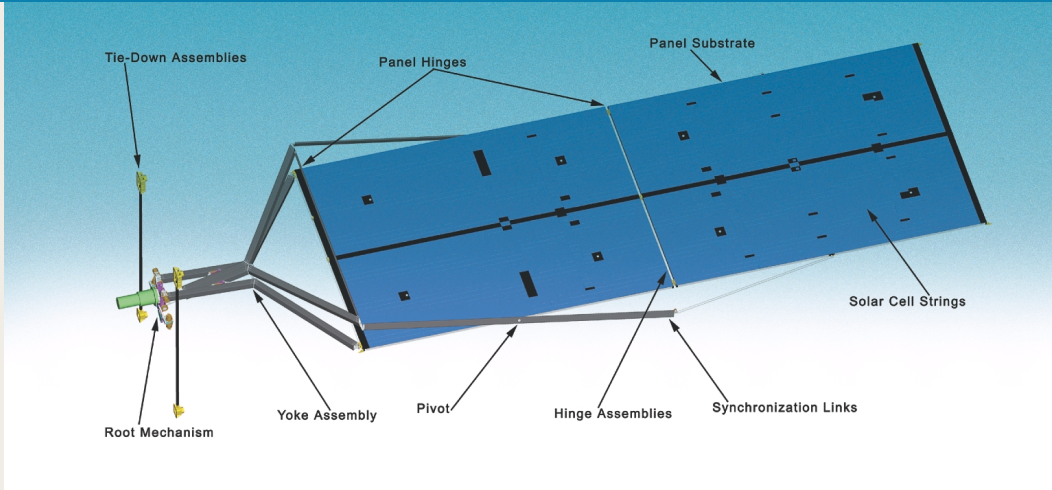
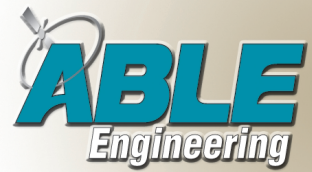
Program	Orbit	Customer	Cell Technology
MSTI-3(4)	LEO	Spectrum Astro	GaAs/Ge
Indostar	GEO	CTA Space (Orbital)	Silicon
BSAT-2A	GEO	Orbital Sciences	Silicon
BSAT-2B	GEO	Orbital Sciences	Silicon
NSTARc	GEO	Orbital Sciences	TJ GaInP/GaAs/Ge
GPS-IIF	MEO	Boeing Space Systems	TJ GaInP/GaAs/Ge
Swift	LEO	Spectrum Astro	DJ GaInP/GaAs/Ge
Galaxy XII	GEO	Orbital Sciences	TJ GaInP/GaAs/Ge
BSAT-2C	GEO	Orbital Sciences	DJ GaInP/GaAs/Ge
PAS2	GEO	Orbital Sciences	TJ GaInP/GaAs/Ge
PAS3	GEO	Orbital Sciences	TJ GaInP/GaAs/Ge
Deep Space 1	Interplanetary	JPL/BMDO	DJ GaInP/GaAs/Ge

Application Benefits

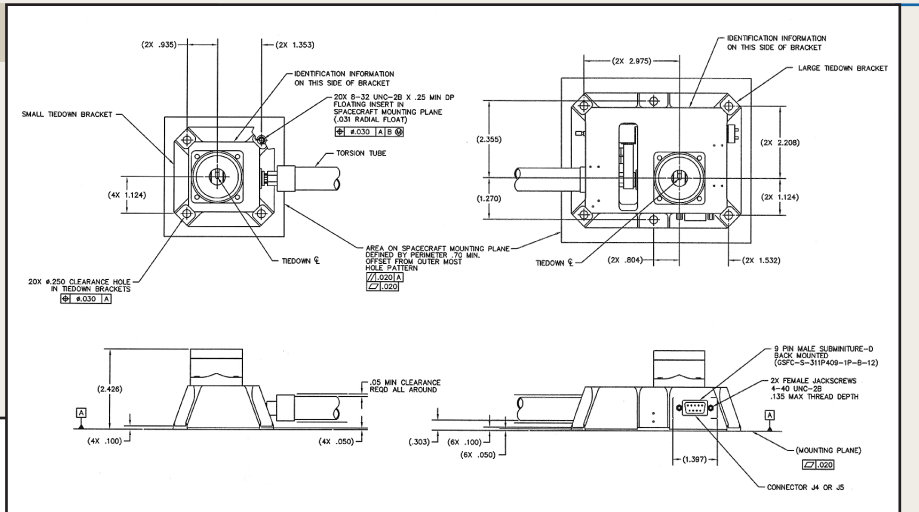
- *Low-Risk Heritage Technology*
- *Simple, Easily Adaptable Spacecraft Interfaces*
- *Low Non-Recurring Engineering*
- *Flight Heritage Components and Processes*
- *11-16 Month Delivery Time*
- *Streamlined Spacecraft Integration*

Able Engineering Company, Inc.
 805.685.2262 • www.aec-able.com
 7200 Hollister Ave.
 Goleta, CA 93117

PUMA Solar Arrays

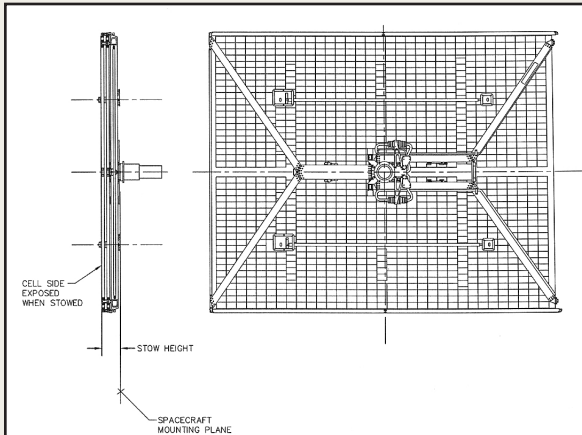


Typical Interfaces

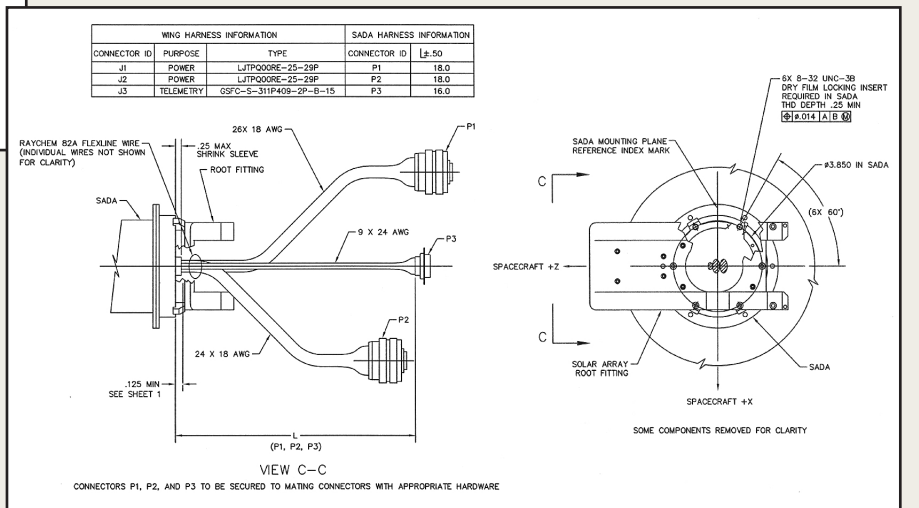


Wing Plan and Side Views

Outline Dimensions are Application Specific



Launch Restraint Assembly • Two Locations per Wing



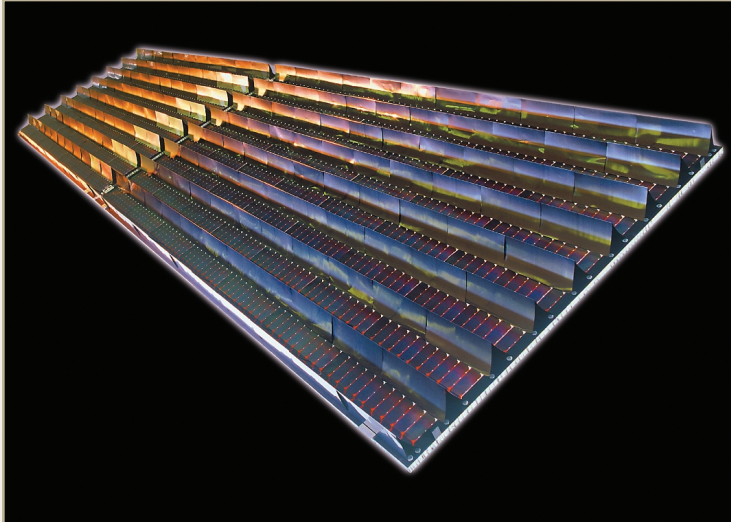
Typical Stow Height

Configuration	Max Dimension (Inches)
1 Panel	3.60
2 Panels	4.90
3 Panels	6.20
4 Panels	7.20

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SADA Mechanical and Electrical • Separate Power and Signal Interfaces

A Cost-Effective Advanced Array Technology to Maximize Delivered Value



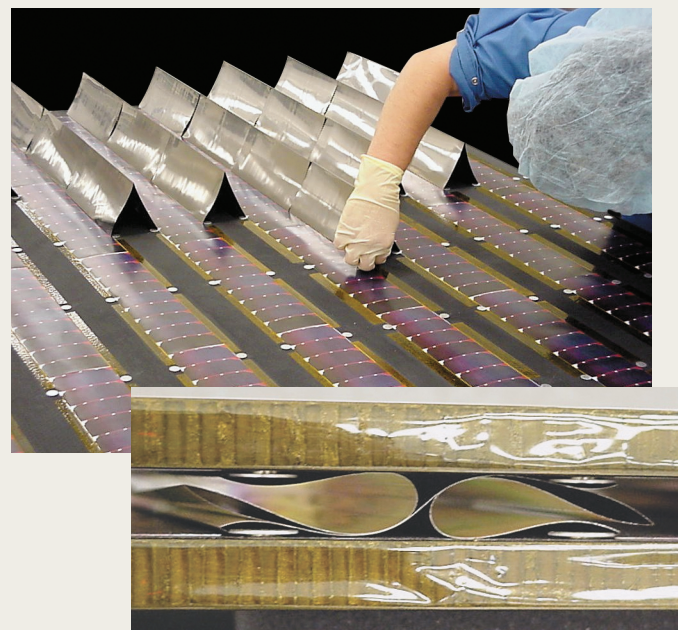
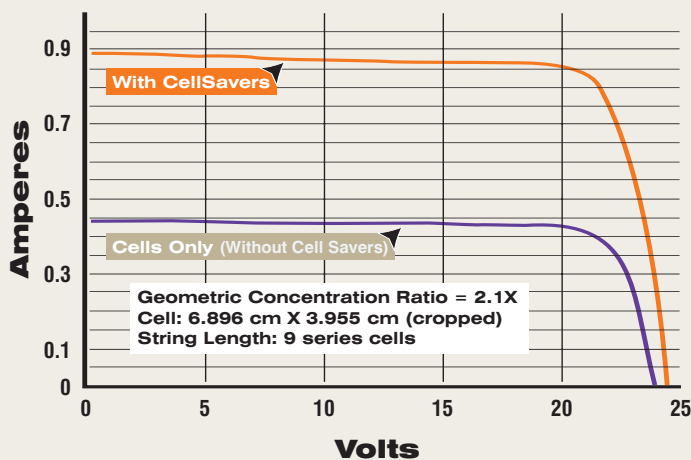
Performance Features

- ~2X Reflective On-Panel Concentrator
- One-piece All-metal Construction Redundantly Grounded to Panel
- Low Temperature Operation (75°C in GEO)
- Wide Solar Acceptance Angle ($\alpha \pm 12^\circ$, $\beta \pm 24^\circ$)
- Array Mass Reduction (~2%)
- Standard Large Area 3J Cells, Laydown and Interconnects
- Retrofits onto Existing Qualified Platforms

Applications Benefits

- ~25% Cost Savings at Array Level
- Low Risk Implementation
- Simple Self Deployment as Panels Open
- Utilize Standard ACS and Array Tracking
- Array-Level End-to-End Testing
- Maintains Heritage Qualification Basis

CellSaver LAPSS Data with Advanced Triple Junction Cells

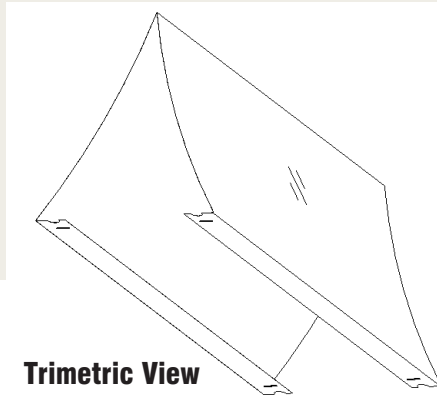


Stowed View

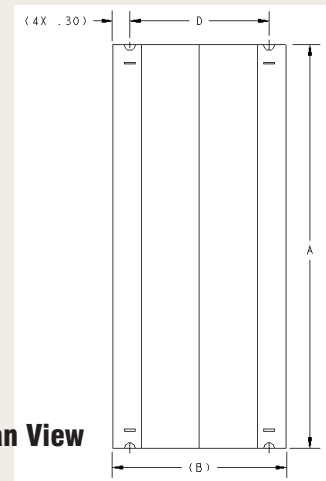
Typical Mechanical Interfaces



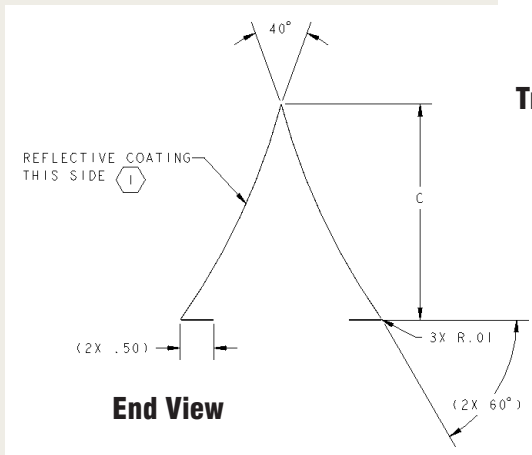
Typical Installed Fastening Tabs



Trimetric View



Plan View



End View

Interface Dimensions

Part No.	A	B	C	D
-1	6.942	3.000	3.200	2.400
-2	8.600	2.996	3.200	2.396
-3	9.260	2.996	3.200	2.396

Qualification

Test	Environment	Result
Electrical	AM0 LAPSS Test	Pass - >95% Efficiency
Off-Angle Performance	AM0 LAPSS Test	Cosine Fall-Off out to 12°(α) & 24°(β)
Random Vibration	52, 45 & 87 g-rms (x, y & z Respectively)	Pass - No shape or power impacts
GEO Thermal Cycles	-175°C to +120°C, 2,000 Cycles	Pass - < 1% Power Loss
LEO Thermal Vacuum Cycles	-115°C to +135°C, 1x10 ⁻⁵ Torr, 12 Cycles	Pass - No Measurable Power Loss
LEO Thermal Cycles	-100° C to +120° C, 80,000 Cycles	In Process - Pass Through 4000 Cycles
Long Term Stowage	11 mm Gap for 3 Years	In-Process - Pass Through 4 Months
Minimum Panel Gap	5 mm Gap for 1 Minute	Pass - No shape or power impacts
Deployment Cycles	50 Stow and Deploy Cycles	Pass - No shape or power impacts
Humidity	90% RH at 22°C, 60 Days	Pass - < 1% Power Loss
UV Radiation	10,000 Equivalent Sun Hours	In Process - Pass Through 1300 ESH
Electron Radiation	15 Year GEO Exposure	In Process - Pass Through 3 Years
Proton Radiation	15 Year GEO Exposure	In Process - Pass Through 3 Years