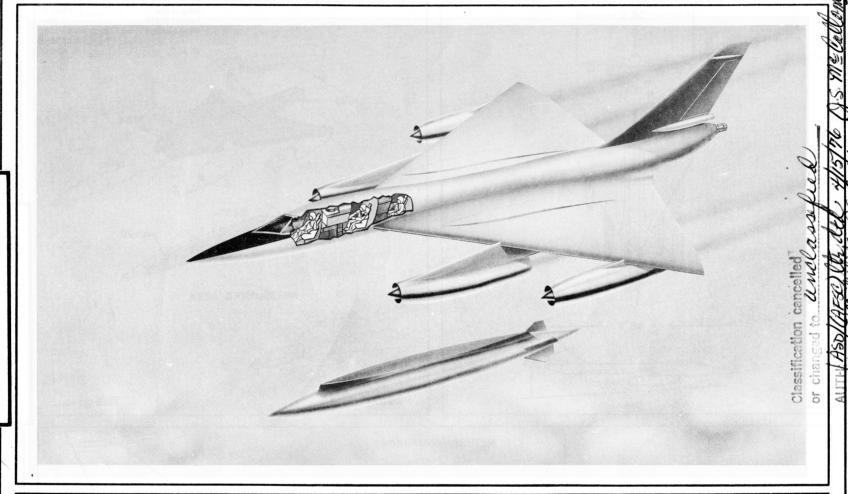
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SERVICE



Standard Aircraft Characteristics

BY AUTHORITY OF THE SECRETARY

OF THE AIR FORCESification cancelled HUSTLER

B/RB-58A

FOUR J79-GE-5

GENERAL ELECTRIC

Signature and Grade

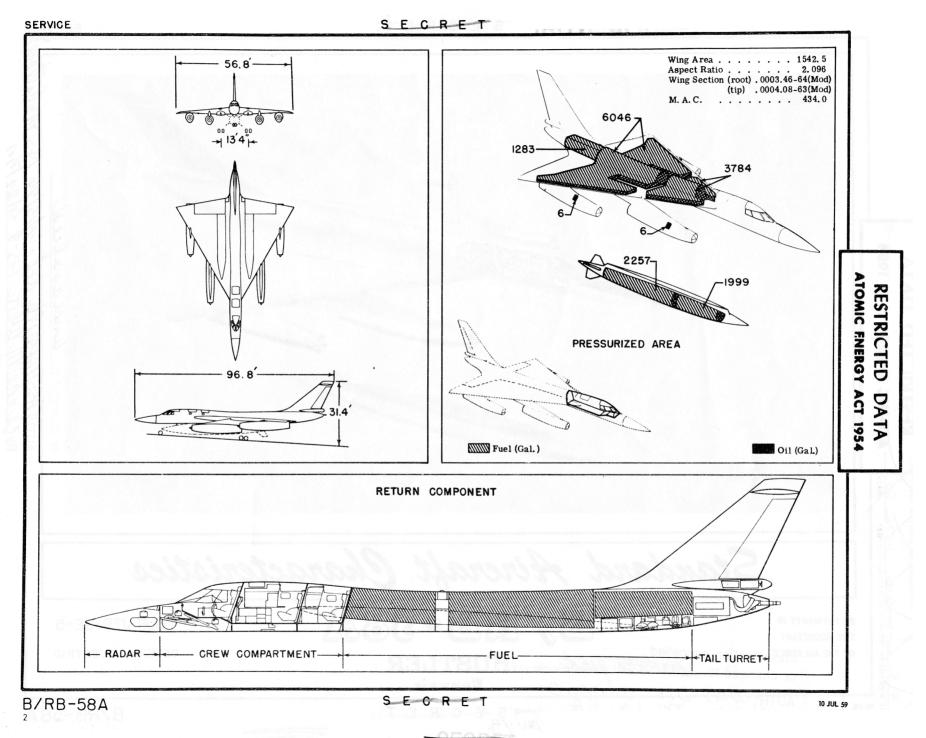
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Convair

B/RB-58A

STOMIC ENERGY RESTRICTED







WEIGHTS

Loading	Basic Airplane	With MB-1C Pod
	Lb. L.F.	Lb. L.F.
Empty	51,061(c)	. 53,581(c)
Basic	51,501(c)	. 54,021(c)
Design	. 158,0002.00	. 158,0002.00
Combat	.*107,2503.00	
Max T.O	. ±90,0003.00	. 158,0002.00
Max Inflight	.±125,1472.00	**158,0002.00
Max Land	. #95,0001.33	. #95,0001.33
Calculated *For De	gio Mission I I imited by	nace #I imited by mon

(c) Calculated *For Basic Mission ±Limited by space #Limited by gear strength **Refueled ±Limited by c.g.

FUEL

Location	Baci	c Airnlane	With MB-1C Pod
Location			
			No. Tanks Gal.
Fus/Wg, Fwd	1	3784	1 3784
Fus/Wg, Aft	1	6046	1 6046
Fus, Tail	1	1283	1 1283
Pod, Fwd. (drop)			1 1999
Pod, Aft (drop) .			1 2257
00Y.81	Cotal	11,113	15,369
Grade			
Specification			MIL-F-5624B
		OIL	No. Tanks Gal.
Nacelle			4 24
			Synthetic
specification			MIL-L-7808

ELECTRONICS

	The state of the s
Navigation-Bombing System	Air-to-Air IFF System APX-48
Defensive ECM System	Station Keeping-Rendezvous Equip .
Radar Warning System	Indirect Bomb Damage Assessment.
Active Defense System MD-7	Inflight Printer
Civil Navigation Aids System ARN-50	
UHF Communications System ARC-57	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
Interphone Integral Part of ARC-57	NOTE: Air Force Model Desig-
Long Range Communication Set . ARC-68	nations do not apply
Air-to-Ground IFF System APX-47	10.0

DIMENSIONS

Wing										
Spar	ı.									56.81
Inci	de	nc	е							3 ⁰ 0'
Dihe	edi	ral	(Oυ	tb	d.	of	,		
										2°14'
Swe										
Length										96.8
Height										31.4
Tread										13.3

WARHEAD

Model						. W-39
Designation						Class C
Weight	•	•	•	•	•	6230 lbs

G U N S

No. Type Size Rds. ea. Location 1 M-61 20mm 580 Tail, tur

Note: Capacity for 1040 rds.

Mission and Description

Navy Equivalent: None

Mfr's. Model: 4

The principal mission of the B/RB-58A is to deliver a warhead in a free fall bomb pod, over a distant target, while traveling at supersonic speed, and the performance of electronic reconnaissance missions.

The crew consists of pilot, navigator-bombardier and a defense systems operator. The entire crew compartment is air conditioned; each crew station is equipped with an ejection seat.

Special features are the tailless design with a "delta" wing planform and the droppable pod housing the warhead and fuel for the outbound portion of the flight.

Other features include automatic pilot, power operated controls, braking parachute, and provisions for single-point ground and air refueling.

Development

Design Initiated													.17	Feb. 51
First Flight		•		•	•								.11	Nov. 56

POWER PLANT

	600.00	
	No. & Model	
1	Mfr	General Electric
	Engine Spec No	E714C
	Туре	Axial
	Length	
	Diameter	
	Weight (Dry)	
	Tail Pipe	Auto. Var. Area
	Augmentation	Afterburner

ENGINE RATINGS

S.L. Static	LB	RPM	Min	
Max:	*15,600	7460		
Mil:	10,000	7460		
Nor:	9700	7460	cont.	

* With Afterburner



ATOMIC ENERGY RESTRICTED ACT 1954 DATA

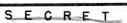
3	CO. 1869a. 1 - 1 - 2 - 2	I	II (9)33	ш	IV
CONDITIONS	bu ved dadah mangun bung	BASIC MISSION	MAXIMUM RADIUS	500 N. MI. COMBAT ZONE @ M = 2.0	FERRY
REFUELED MISSION	stern Oliver er seda den	2.200.01	KADIOS	COMBAT ZONE @ M = 2.0	RANGE
TAKE-OFF WEIGHT	(lb)	158,000	158,000	158,000	155,329
Fuel at 6.5 lb/gal (JP-4)	(lb)	96,338	96,338	96,338	99,897
Payload (Warhead)	(lb)	6230	6230	6230	None
Take-off ground run at SL	(ft)	5870	5870	5870	5650
Take-off to clear 50 ft	(ft)	9030	9030	9030	8700
Rate of climb at SL	(fpm)	14,780	14,780	14,780	15,100
Rate of climb at SL (one engine out)	(fpm)	9250	9250	9250	9450
EFUELED WEIGHT	(1ь)	158,000	158,000	158,000	155,329
Transferred fuel	(lb)	64,100	64,100	64,100	64,000
Refuel radius	(n.mi)	2500	2500	2500	2500
Wing loading	(lb/sq ft)	102.3	102.3	102.3	100.7
Service ceiling (100 fpm)	(ft)	47,900	47,900	47,900	48,300
Service ceiling (100 fpm) Service ceiling (one engine out)	(ft)	38,300	38,300	38,300	38,700
COMBAT RANGE	(ft)				8416
OMBAT RADIUS	(n.mi)	3910	4225	3200	
Combat zone radius	(n.mi)	100		500	
Combat zone speed 2	(kn)	1147	Paning	1147	
Target altitude	(ft)	52,700	38,800	53,000	Mace Dr.
Target speed (2)	(kn)	1147	552	1147	- Lebertô
Average cruise speed outside combat zone	(kn)	544	539	544	533
Initial cruise altitude	(ft)	30,000	30,000	30,000	30,000
Final cruise altitude	(ft)	50,350	50,150	50,700	49,500
Total mission time	(hr)	14.7	16,18	11.3	16,28
OMBAT WEIGHT	(lb)	107,250	111,030	105,050	62,150
Combat Altitude	(ft)	58,500	42,700	58,850	49,500
Combat speed (2)	(kn)	1147	1147	1147	1147
Combat speed Combat climb Combat ceiling (500 fpm) Service ceiling (100 fpm) Service ceiling (one engine out) Maximum rate of climb at SL Maximum speed at service ceiling Basic speed at 35,000 ft (2)	(fpm)	1000	15,600	1,000	17,800
Combat ceiling (500 fpm)	(ft)	59,000	58,300	59,350	66,400
Service ceiling (100 fpm)	(ft)	59,400	58,700	59,750	67,000
Service ceiling (one engine out)	(ft)	48,750	48,000	49,250	58,000
Maximum rate of climb at SL	(fpm)	38,000	36,800	38,300	64,000
Maximum speed at service ceiling (2)	(kn)	1147	1147	1147	1147
Basic speed at 35,000 ft	(kn)	1131	1131	1131	1131
ANDING WEIGHT	(lb)	58,800	59,400	57,900	62,150
Ground roll at SL	(ft)	2700	2730	2660	2845
Ground roll (parachute)	(ft)	2055	2080	2010	2200
Total distance from 50 ft	(ft)	5060	5100	5015	5225
Total distance from 50 ft (parachute)	(ft)	4415	4450	4365	4580

NOTES ① Max. Power ② High speeds restricted by engine and airframe structural limit

3 Data based on airplane carrying Free Fall Bomb Pod (MB-1C) with W-39 Warhead.

PERFORMANCE BASIS:

(a) Data source: Contractor's estimated data



		1 1	
Q	1		/
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Loading	and	Performa	nce-Type	ical Mis	sion
		V	VI	VII	VIII
C O N DITION NON-REFUELED MISSION		BASIC MISSION	MAXIMUM RADIUS	500 N. MI. COMBAT ZONE @ M = 2.0	FERRY RANGE
TAKE-OFF WEIGHT Fuel at 6.5 lb/gal (JP-4) Payload (Warhead) Wing loading Take-off ground run at SL Take-off to clear 50 ft Rate of climb at SL Rate of climb at SL (one engine out) Time: SL to 20,000 ft Time: SL to 30,000 ft Service ceiling (100 fpm) Service ceiling (one engine out) COMBAT RANGE COMBAT RADIUS Combat zone radius	(f (f) (f) (f) (f) (g) (m) (g) (m) (f) (f) (f) (f) (f) (f) (f) (f) (f) (f	96,338 6230 eq ft) 102.3 t) 5870	158,000 96,338 6230 102,3 5870 9030 14,780 9250 6,9 13.7 47,900 38,300	158,000 96,338 6230 102,3 5870 9030 14,780 9250 6,9 13.7 47,900 38,300	155,329 99,897 None 100.7 5650 8700 15,100 9450 6,7 13.5 48,300 38,700
Combat zone radius Combat zone speed Target Altitude Target speed Average cruise speed outside combat zone Initial cruise altitude Final cruise altitude Total mission time	(i (i (k (k (i	n) 1147 t) 55,900 n) 1147 t) 544 t) 30,000 t) 50,600	43,200 552 539 30,000 50,300 10,13	1147 56,100 1147 544 30,000 50,900 5,24	 533 30,000 49,700 10.58
COMBAT WEIGHT Combat altitude Combat speed Combat ceiling Combat ceiling (500 fpm) Service ceiling (100 fpm) Service ceiling (one engine out) Maximum rate of climb at SL Maximum speed at service ceiling Basic speed at 35,000 ft		b) 85,500 t) 63,100 n) 1147 m) 1000 t) 63,600 t) 64,000 t) 53,600 m) 47,800 n) 1147 n) 1131	89,000 48,000 1147 15,600 63,000 63,400 52,900 45,800 1147 1131	84,450 63,500 1147 1000 64,000 64,400 53,900 48,200 1147 1131	61,900 49,700 1147 18,000 66,600 67,200 58,200 64,200 1147 1131
LANDING WEIGHT Ground roll at SL Ground roll (parachute) Total distance from 50 ft Total distance from 50 ft (parachute)	(b) 58,200 (t) 2680 (t) 2020 (t) 5040 (t) 4380	59,000 2710 2060 5075 4425	57,500 2645 1990 5000 4345	61,900 2825 2185 5205 4565

NOTES

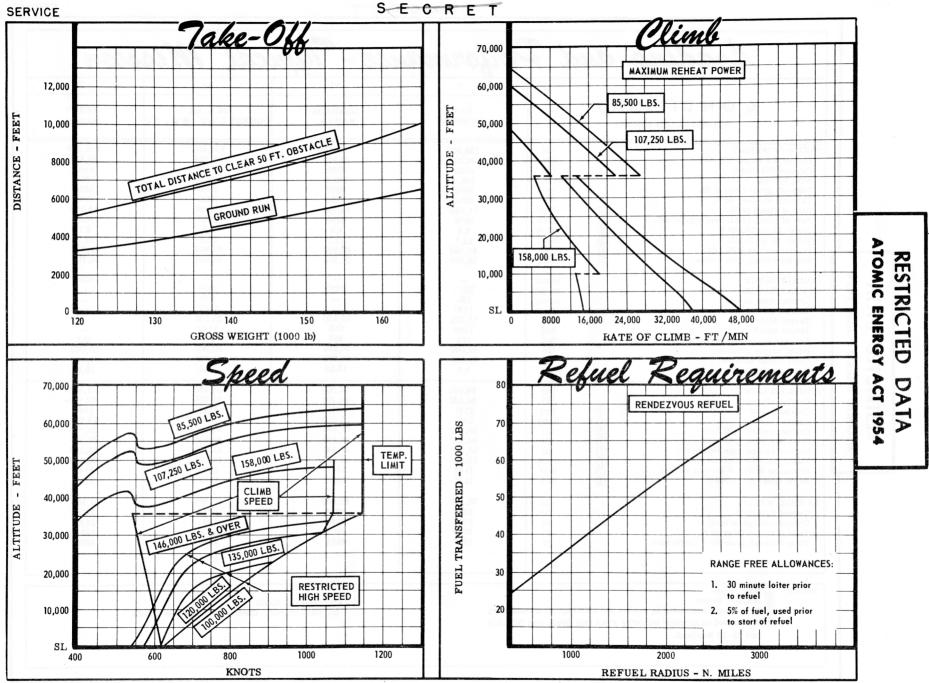
Max power
 Military Power
 High speeds restricted by engine and airframe structural limit

(4) Data based on airplane carrying Free Fall Bomb Pod (MB-1C) with W-39 Warhead.

PERFORMANCE BASIS:

(a) Data source: Contractor's Estimated Data



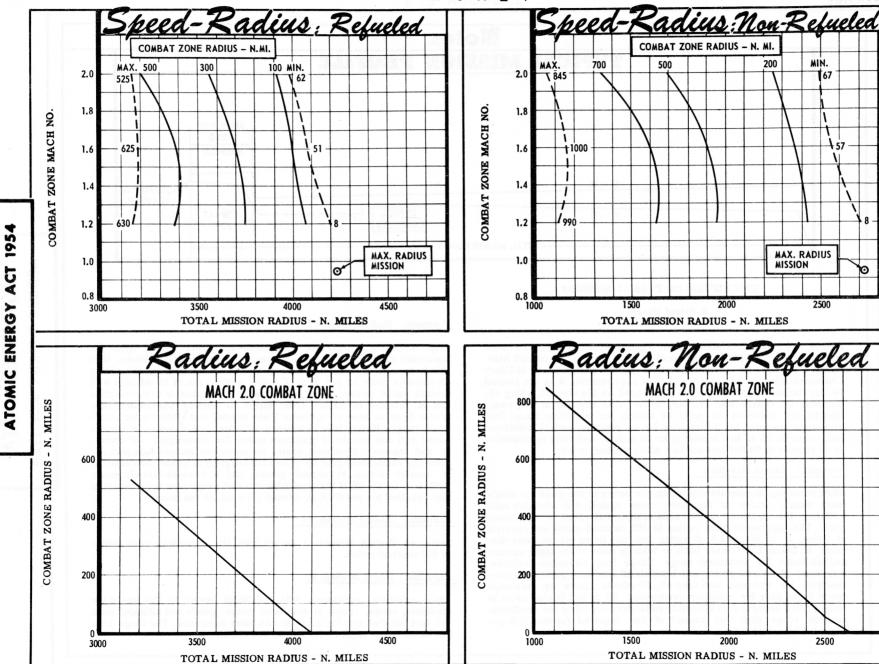


B/RB-58A

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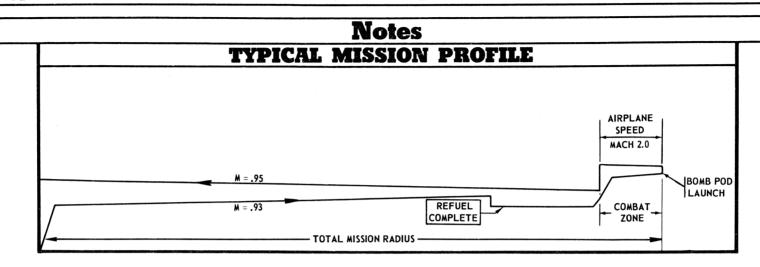
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SECTION

B/RB-58 A





DESCRIPTION OF TYPICAL MISSIONS

Refueled Missions

Radius and range data shown on the Typical Missions pages are based on refueling the B-58 2500 nautical miles from the takeoff base. (2500 n.mi. is the maximum refuel radius permitted by Military Characteristics in meeting the radius requirement; it is not related to any tanker capability.) Data are based on the aircraft taking off, climbing on course with Military power, then cruising at long range speeds and altitudes to point of hookup for refueling. Range free allowances include: (a) 30 minutes for rendezvous, and (b) additional fuel equal to 5% of fuel burned prior to hookup. Refueling is conducted at an altitude of 30,000 feet and at a Mach number of 0.8.

Formula: Radius Mission I

After refueling, accelerate with maximum power to the speed for maximum range, cruise at maximum range speeds and altitudes until initiating the maximum power acceleration and climb to supersonic zone. The supersonic zone radius is 100 nautical miles and consists of flying in at Mach 2.0, dropping pod, and flying out the same distance at Mach 2.0. Cruise back to base is made at the Mach number and altitudes for maximum range. Range free allowance includes: 5 minutes of normal power and 1 minute of maximum power fuel consumption for warm-up and take-off, 30-minute cruise on maximum range flight path for refueling rendezvous; 5% of fuel burned prior to refuel, and a reserve fuel allowance sufficient to fly 8% of the creditable mission range plus that amount of fuel required for one GCA goaround (80 n.mi. at sea level).

Formula: Radius Mission II

After refueling, accelerate with maximum power to the speed for maximum range, cruise at maximum range speeds and altitudes, then climb on course with maximum power so as to reach cruise ceiling 15 minutes prior to pod drop. Conduct a bomb run of 15 minutes with normal rated power, drop pod, conduct 2 minutes of evasive action and 8 minutes of escape from target at normal rated power; then cruise back to base at optimum Mach number and altitudes for maximum range. Range free allowance includes: five minutes of normal power and one minute of maximum power fuel consumption for warmup and take-off, 30-minute cruise on maximum range flight path for refueling rendezvous plus 5% of fuel burned prior to refuel, two minutes of evasive action at normal power, and a reserve fuel allowance sufficient to fly 8% of the creditable mission range plus that amount of fuel required for one GCA go-around (80 n.mi. at sea level).

Formula: Radius Mission III

Same as Radius Mission I except that the supersonic zone radius is 500 nautical miles.

Formula: Range Mission IV

After refueling, accelerate with maximum power to the speed for maximum range, cruise at maximum range speeds and altitudes until the fuel allotted for the mission has been consumed. The range free allowance is the came as for the Radius Mission I.





Non-Refueled Missions

Formula: Radius Mission V

Takeoff, climb on course with Military power to the altitude for maximum range, cruise at maximum range speeds and altitudes until the maximum power acceleration and climb to supersonic zone is initiated. The supersonic zone radius is 200 nautical miles and consists of flying in at Mach 2.0, dropping pod, and flying out the same distance at Mach 2.0. Cruise back to base is made at the Mach number and altitudes for maximum range. Range free allowance includes 5 minutes of normal power and 1 minute of maximum power fuel consumption for warm-up and take-off and a reserve fuel allowance sufficient to fly 8% of the creditable mission range plus that amount of fuel required for one GCA go-around (80 n.mi. at sea level).

Formula: Radius Mission VI

Takeoff, climb on course with Military power to the altitude for maximum range, cruise at speeds and altitudes for maximum range, and then climb on course with maximum power in order to reach cruise ceiling 15 minutes prior to pod drop. Conduct a bomb run of 15 minutes with normal rated power, drop pod, conduct 2-minute evasive action,

and 8-minute escape from target at normal rated power; then cruise back to base at the Mach number and altitudes for maximum range. Range free allowance includes five minutes of normal power and one minute of maximum power fuel consumption for warm-up and take-off, the two minutes of evasive action at normal power, and a reserve fuel allowance sufficient to fly 8% of the creditable mission range plus that amount of fuel required for one GCA go-around (80 n.mi. at sea level).

Formula: Radius Mission VII

Same as Radius Mission V except that the supersonic zone radius is 500 nautical miles.

Formula: Range Mission VIII

Takeoff, climb on course with Military power to the altitude for maximum range, and cruise at maximum range speeds and altitudes until the fuel allotted for the mission has been consumed. The range free allowance is the same as for Radius Mission V.

SUPPLEMENTAL DATA

The curves on page 10 complete the coverage of the mission performance capabilities of the B-58. The upper plot shows target coverage capabilities of the B-58 flying a Mach 2.0 zone at altitude; the lower plot shows target coverage for the B-58 flying a sea level zone.

Three basic types of mission are depicted:

- non-refuel, radius missions with radius zones and with pretarget zones.
- (2) refuel, radius missions with radius zones and with pretarget zones.
- (3) refuel, 2000 n.mi. post-strike stage missions with pretarget zones.

For both the high altitude and the sea level missions, the plots show target distance vs. zone radius for missions with a radius zone and vs. zone range for missions with a pre-target zone.

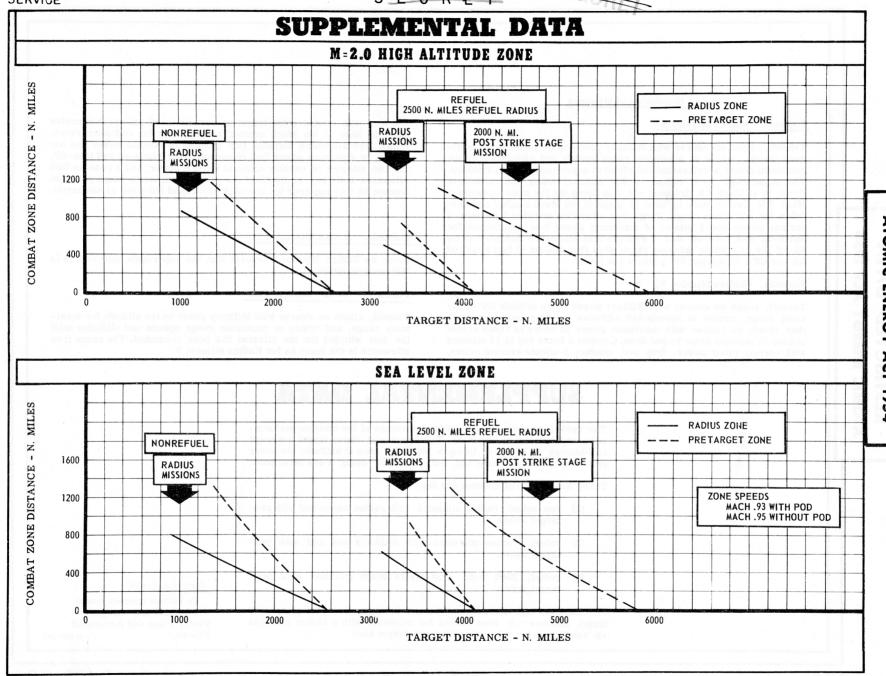
Revision Basis:

To reflect current model designation and changes to Power Plant and Armament Blocks.

(15 NOV 58)

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B/RB-58 A

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