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Aug 2009

Satellites panel of 2009 Decadal Survey

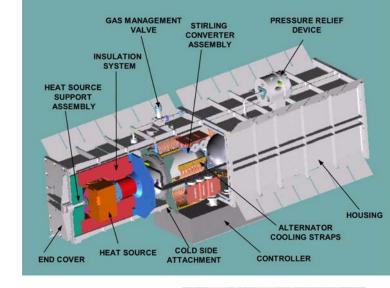
In this talk

- Review DSMCE study results (ended Feb 2009)
- Offer a few general comments about subsequent efforts for a Discovery proposal
- Offer lessons relevant to Satellites panel

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NASA wants to flight-test the Advanced Stirling Radioisotope Generator (ASRG) on a Discovery-class mission

- Pu²³⁸ is in very short supply!
 - ASRG requires 6x less Pu²³⁸ per W of electric power than conventional RTGs
- ASRG Characteristics:
 - ≥14 year lifetime requirement
 - Nominal power : 140 W electric (or is it 160 W?)
 - Mass ~ 20 kg
 - System efficiency: ~ 30 %
 - 2 GPHS ("Pu²³⁸ Bricks") modules
 - Uses 0.8 kg Pu²³⁸
- Reliability to be demonstrated by the end of 2009; Flight units ready by mid-2014 (or later)
- 9 DSMCE studies funded.



Lockheed Martin/Sunpower





Paired converters with interconnect sleeve assembly

How ASRGs enable IVO

- Juno shows that a Jupiter mission with a low data rate and little pointing flexibility fits
 NF
- High data rate needed to achieve lo science
 - Dynamic lo must be observed over many timescales and at high spatial resolution and at multiple wavelengths to make major advances over *Galileo*
 - High data rate requires high power at 5 AU; solar arrays would need to be extremely large
- Pointing flexibility needed to achieve lo science
 - Cannot keep solar arrays pointed at sun without expensive gimbals (and more power)
- Solar arrays degrade in radiation environment near lo
- Jitter from large solar arrays will degrade high-resolution imaging
- Radioisotope power less risky (e.g., tumbling or mis-pointed S/C).

Culann Patera volcanic center

IVO DSMCE Study Team

PI/Science team

- Alfred McEwen, UofA, PI
- Laszlo Keszthelyi (USGS), John Spencer (SwRI), Nick Thomas (UBern), Peter Wurz (UBern), Torrence Johnson (JPL), Phil Christensen (ASU), Karl-Heinz Glassmeier (IGEP), Zibi Turtle (APL), Krishan Khurana (UCLA), Julie Moses (LPI)

Instrument teams

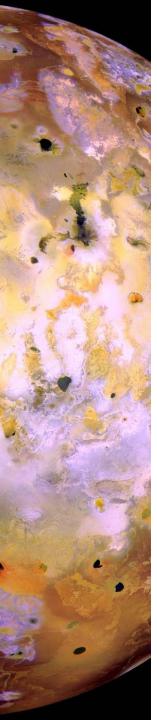
- US-built:
 - RCam: UA lead McEwen, Chris Shinohara, others
 - ThM: ASU lead P. Christensen
- Contributed:
 - INMS: U. Bern lead Peter Wurz
 - FGM: IGEP lead Karl-Heinz Glassmeier

Spacecraft team

 Tim Girard (Sierra Nevada Corp.), Chris Shinohara (UA), Roberto Furfaro (UA), Thomas Gardner (RMS), Dan Cheeseman (RMS)

JPL team

 Richard Beatty, Jan Ludwinski, Theresa Kowalkowski, Chen-wan Yen, Robin Evans, Insoo Jun, many others from Team X



An lo Mission is High Priority for Solar System Exploration

- 2002 Solar System Decadal Survey identified 4 broad themes:
 - The first billion years of Solar System history
 - Ancient voluminous and high-temperature volcanic and interior processes on the Earth, Moon, Mercury, and Mars are active on lo today.
 - Volatiles and organics: The stuff of life
 - Discerning how lo became devoid of water (and carbon?) but retained sulfur is fundamental to understanding the evolution of volatiles.
 - The origin and evolution of habitable worlds
 - Tidal heating determines the habitable zone in the Jovian system and analogous processes may apply to extrasolar planetary systems.
 - Processes: How planetary systems work
 - Hyperactive lo is an ideal target for studying how planets work.
 - Io is the Hawai'i of the Solar System.
 - lo Observer is mentioned in Decadal Survey for a New Frontiers
 Class mission in the next decade (2013-2023), the NRC study on
 choices for the next New Frontiers AO, and is included in NF-3 AO.
 - IVO can make significant contributions to all seven of the NF Io Observer Science goals

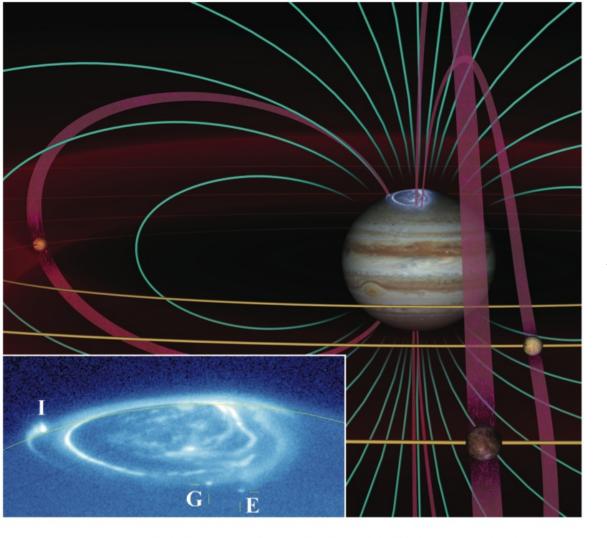
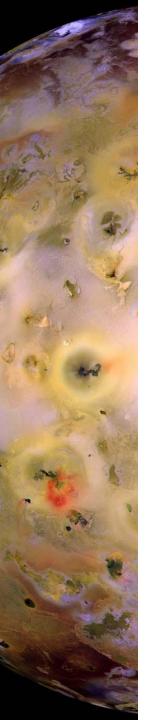


FIGURE 2.4 Electrical currents aligned with Jupiter's strong magnetic field couple the moons lo, Ganymede, and Europa with Jupiter's high-latitude ionosphere. Bright auroral emissions observed at the footpoints of the magnetic flux tubes linking the moons to the ionosphere (inset) are the signature of this coupling, the physics of which is only partially understood. (The letters superposed on the auroral image in the inset identify the emission features associated with the footpoints of lo, Ganymede, and Europa.) A probe to study the still unexplored polar regions of Jupiter's magnetosphere will answer basic questions about the nature of electrodynamic coupling between the Jovian atmosphere and magnetosphere and about auroral acceleration in a magnetospheric environment much different from Earth's. Hubble Space Telescope image courtesy of J.T. Clarke (Boston University) and NASA/Space Telescope Science Institute. Artist's rendering of the Jovian inner magnetosphere courtesy of J.R. Spencer (Lowell Observatory). Reprinted by permission from *Nature* 415:997-999 and cover, copyright 2002, Macmillan Publishers Ltd.

Jupiter Polar and Io Electrodynamics missions are a high priority in the Solar and Space Physics Decadal study

- Juno partially satisfies
 the Jupiter Polar
 objectives, but IVO would
 explore a different region
 of space and has a
 higher data rate for
 remote sensing of polar
 phenomena on Jupiter.
- IVO will observe auroral phenomena on lo as well as Jupiter and directly measure the species escaping from lo



IVO Mission Overview

- DSMCE study: \$450 M cost cap including launch vehicle (but not ASRGs or NEPA).
- Launch in 2015 on Atlas V 401.
- VEEGA trajectory and Jupiter Orbit Insertion (JOI) in 2021.
- Io flyby before JOI, then 6-month capture orbit about Jupiter inclined ~49°.
- 6 additional Io flybys over 10 months.
- Closest approach to Io from 1000 to 100 km, closer in extended mission.
 - Fly through active plumes late in late nominal and in extended mission.
- Collect 20 Gb science data near Io (100x total data from *Galileo* on each flyby).
- Playback of Io encounter data
 + Jupiter system science near apoapsis.
- Extended mission options
- End mission with Io impact

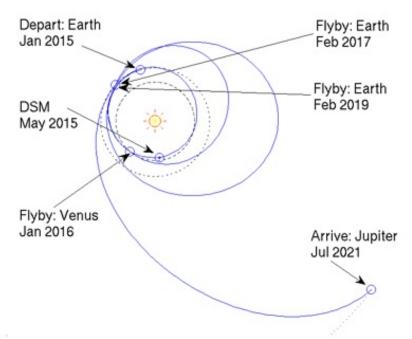


Table 5.1-3. Example Jupiter tour with 10 Io flybys.

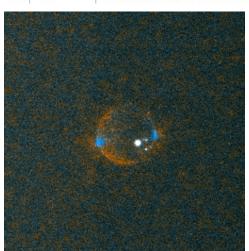
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17				lo-relative Quantities					Jupiter-relative Quantities			A.	$\overline{}$	
lo		Days				Phase	Á.,				Desc			
Flyby	Days	after last	Altitude	Lat	Lon	Angle	Speed	Vinf	Incl.	Apojove	Node	Perijove	Sun-Jup-	Jup-lo-
#	from JOI	lo	(km)	(deg)	(deg)	(deg)	(km/s)*	(km/s)	(deg)	(Rj)	(Rj)	(Rj)	lo (deg)	s/c (deg)
1	199.6	n/a	546	4.58	268.37	147.3	18.54	18.38	102.22	181.0	99.8	5.748	57.94	90.8
2	311.1	111.5	325	5.63	268.37	137.4	18.54	18.38	102.04	105.6	70.6	5.718	48.45	91.2
3	362.4	51.3	256	6.50	268.35	132.6	18.54	18.36	101.91	78.3	56.7	5.716	44.04	91.5
4	396.0	33.6	672	-6.50	88.43	50.4	18.47	18.36	78.16	103.0	69.4	5.728	41.13	88.4
5	445.5	49.5	355	2.54	251.10	108.1	18.52	18.35	103.09	81.2	59.1	5.745	36.82	109.0
6	480.9	35.4	998	-6.38	88.51	57.8	18.43	18.34	78.51	105.6	71.9	5.761	33.67	
7	532.2			-12.53	160.80	16.4	18.47	18.34	91.98		82.3		29.16	
8	594.1	61.9	291	3.92	257.07	101.5	18.52	18.35	101.96	86.8	65.1	5.791	23.68	102.3
9	633.1	38.9	433	-11.52	125.07	35.6	18.50	18.36	85.33	108.1	79.1	5.800	20.23	
10	686.1	53.1	412	-0.48	233.59	70.2	18.53	18.37	102.11	81.1	64.6	5.793	15.50	125.5
11	721.5	35.4	Impact	4										

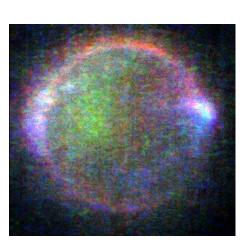


Table 5.1-4. Io eclipse opportunities inbound and outbound for each Io flyby in the example tour.

Enc #	In/Out	Enter eclipse	Exit eclipse	Eclipse Duration	Entry range (km)	Entry range (Rj)	Exit Range (km)	Exit Range (<u>Rj</u>)
1	In	7/13/2020 09:40:18	7/13/2020 11:56:33	2:16:15	1573574	22.01	1387711	19.41
1	Out	7/15/2020 04:08:49	7/15/2020 06:25:05	2:16:16	773088	10.81	908804	12.71
2	In	11/1/2020 21:41:32	11/1/2020 23:59:28	2:17:56	1455497	20.36	1269776	17.76
2	Out	11/3/2020 16:10:10	11/3/2020 18:28:07	2:17:57	824828	11.54	959899	13.43
3	In	12/23/2020 05:33:35	12/23/2020 07:52:03	2:18:28	1364752	19.09	1182472	16.54
3	Out	12/25/2020 00:02:20	12/25/2020 02:20:50	2:18:30	845559	11.83	977975	13.68
4	In	1/25/2021 20:39:17	1/25/2021 22:57:58	2:18:41	1297443	18.15	1118816	15.65
4	Out	1/27/2021 15:08:02	1/27/2021 17:26:40	2:18:38	877276	12.27	1015164	14.20
5	In	3/16/2020 10:03:30	3/16/2021 12:22:03	2:18:33	1291441	18.06	1111935	15.55
5	Out	3/18/2021 04:32:10	3/18/2021 06:50:43	2:18:33	899032	12.58	1034776	14.47
6	In	4/20/2021 19:37:20	4/20/2021 21:55:40	2:18:20	1232162	17.23	1057035	14.79
6	Out	4/22/2021 14:05:59	4/22/2021 16:24:18	2:18:19	932422	13.04	1073269	15.01

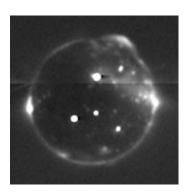






Visible colors in eclipse (Galileo)

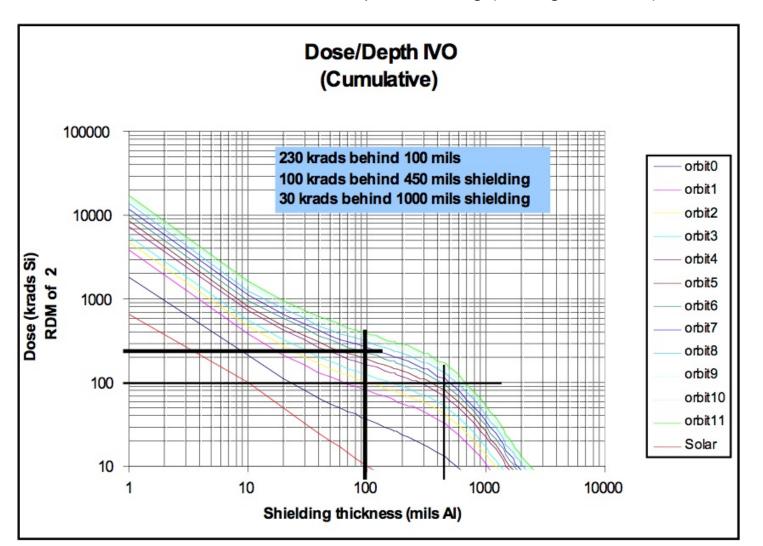
IVO: ~10 km/pixel images and125 km/pixel thermal mapping



New Horizons image

IVO Total Dose Environment

- Primary science achieved in 7 fast (~18.5 km/s) fly-bys
- Significant shielding typically available from S/C elements
 - 100 krad parts should suffice even without extra shielding
- IVO will have a radiation "vault" and spot shielding (200 kg allocated)



Jupiter Orbiters: Total Dose Comparisons (shielded by 100 mils (2.5 mm) Al and RDM = 1)

- **JEO: 2900 krads** (2009 JEO SDT report)
- 2007 Europa Explorer: 2600 krads (no "optional" lo flybys) (2009 JEO SDT)
- Galileo (all extended missions): 600 krads (Fieseler et al., IEEE Tr. Nuc. Sci 49, 2002)
- Juno: 500 krads (JPL team X)
- JGO: ~200 krads (85 krad behind 8 mm Al) (2009 JGO SDT report)
- IVO: 115 krads (JPL study for DSMCE)

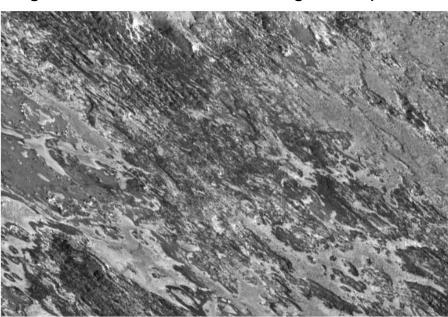
Science Payload

- Radiation-hard camera (RCam)
 - NAC (10 µrad/pixel), digital processing unit (DPU), 15 color bandpasses from 200-1100 nm
 - Monitor eruptions, Io Plasma Torus (IPT), measure peak lava temperatures, stereo images for topography, movies of plumes and lava lakes, eclipse imaging, optical navigation
- Thermal mapper (ThM)
 - 125 µrad/pixel, 10 bandpasses from 3-20 microns
 - Map and monitor temperatures, heat flow pattern related to tidal heating mechanisms, silicate emissions for composition
- Ion and Neutral Mass Spectrometer (INMS)
 - Determine composition of ions and neutrals in plumes, atmosphere, IPT
- Two fluxgate magnetometers (FGM)
 - Intrinsic and induced magnetic signatures of Io; explore new regions of Jovian magnetosphere

Narrow-Angle Radiation-hard Camera (RCam)

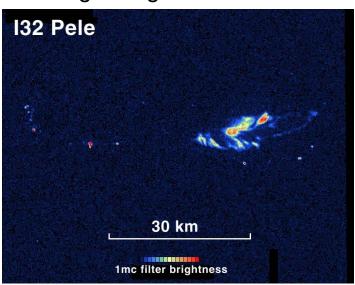
- Moderate resolution monitoring during approach to and departure from lo; high resolution and (limited) stereo near closest approach
 - NAC: 10 μrad/pixel gives 1 km/pixel from 100,000 km, 10 m/pixel from 1,000 km
 - New CMOS focal-plane system, pushbroom and framing modes
 - Excellent performance (< 2 e- read noise) after 1-2 Mrad total dose (Janesick et al. 2008)
 - Data readout extremely fast (240 Mb/s per ADC) to minimize radiation noise in images
 - Digital Processing Unit (new FPGA code)
- Near-simultaneous multispectral measurements for peak lava temperatures
 - 0.1 sec time differential could ruin the measurement because hot lava is so dynamic
 - Working on narrow (4 line) filters for nearly simultaneous color
- ASRG-induced jitter not a significant concern, unless 1 ASC is lost

Highest-resolution *Galileo* image: 6 m/pixel



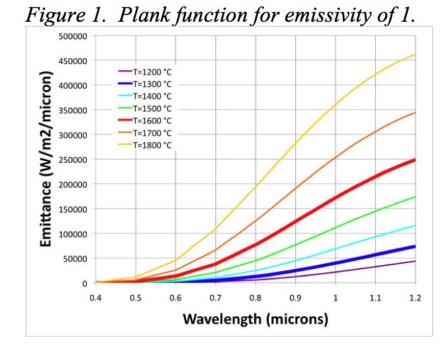
1 km (0.6 mile)

Lava glowing in the dark



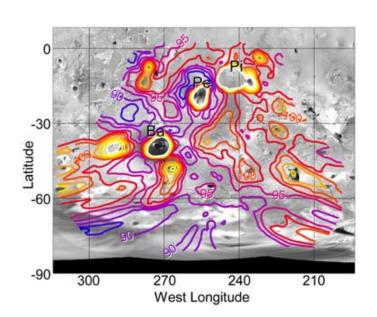
RCam Color Filters

- Spectral range of backside-thinned CMOS detector is ~200-1000 nm (QE > 0.1)
- Threshold Mission:
 - Blue-green (450-650 nm), red (650-800 nm), near-IR (>800 nm), and >950 nm for lava temperatures and color images
 - 16 sets of 4-line filters for nearly simultaneous measurements of lava temperatures via dTDI
 - UV (200-400 nm) for plumes
 - Na-D band to monitor Na cloud
- Baseline Mission:
 - Spectrally narrow filters for O, S+
 - Can usefully monitor Na, O escaping from Europa as well as Io; unique viewing geometry
 - Additional bands for silicate mineralogy and sulfur species
 - Methane bands for Jupiter
 - Room for 15 bandpasses with 64 dTDI lines on 2,000 x 2,000 array
 - Save at least 1000 x 2000 for clear framing mode



Thermal Mapper

- Threshold Mission: ~2, 5, 8, 20 microns to monitor volcanism and measure heat flow
 - lo surface Ts range from 70 to 1800 K!
 - Mercury T range is ~90 to 700 K
- Thermal Emission Imaging System (THEMIS)
 - 4.6 deg FOV, 250 urad/pixel, 1 km/pixel from 4,000 km, 10 bandpasses
 - New 640 x 480 detectors: 1 km/pixel from 8,000 km
- Baseline Mission: ~ 10 bandpasses
 - Optimize bandpasses for thermal mapping
 - Thermal emission compositional studies of silicate mineralogy
 - Lava expected to be glassy--problem for NIR but not thermal IR





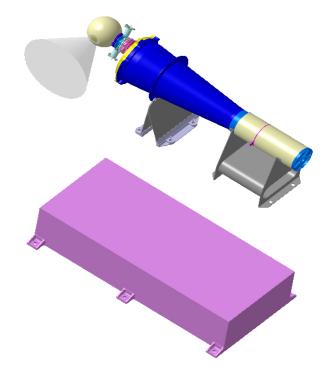
Ion and Neutral Mass Spectrometer (INMS)



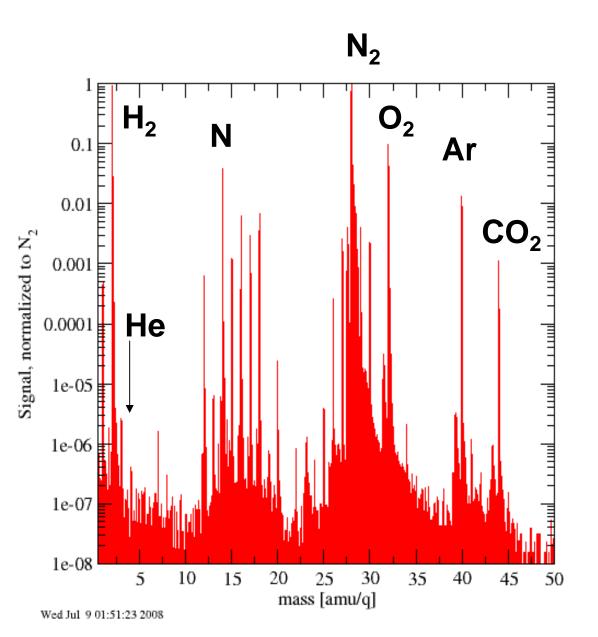
To be contributed by U. Bern—Peter Wurz and Swedish Institute of Space Physics--Martin Wieser and Stas Barabash

Mass range 1-300 amu; $M/\Delta M = 300 - 1000$, increases with mass

Mass spectra are recorded once every 5 seconds (flyby mode) that gives a direct science data rate of 19,200 bits/s.



P-BACE quick-look data



- Raw data--no background subtracted
- Dynamic range:
 6–7 orders of magnitude
- Mass range: 1–1000 amu/q

19

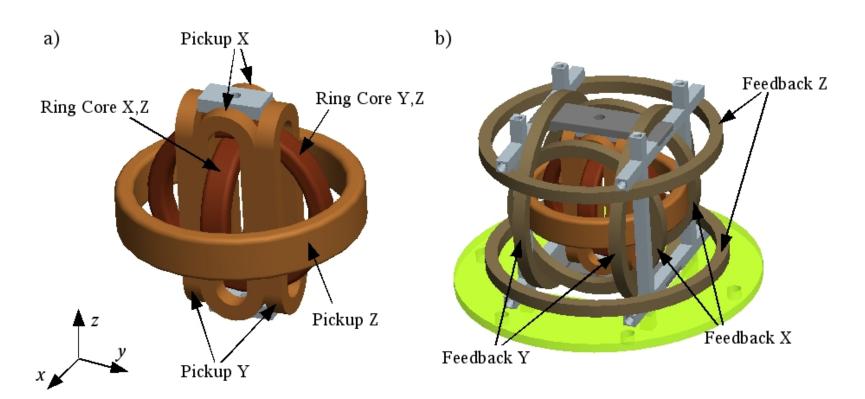


Fluxgate Magnetometer

Vector compensated fluxgate ringcore sensor

Contributed by Karl-Heinz Glassmeier, IGEP

Heritage: Rosetta, Venus Express, Themis, BepiColombo



IVO Science Objectives

There are 7 science goals for NF-3 *Io Observer*. *IVO*'s baseline mission addresses all of these and 1 more, prioritized into Groups A and B.

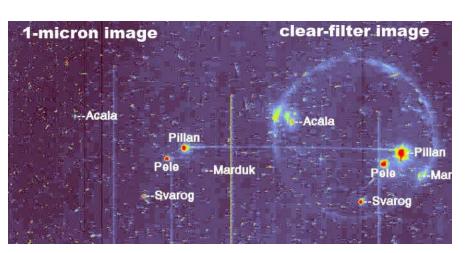
- A1. Understand the eruption mechanisms for lo's lavas and plumes and their implications for volcanic processes on Earth and the other terrestrial planets.
- A2. Determine lo's interior structure, especially the melt fraction of the mantle.
- A3. Determine the properties and mechanisms of lo's tidal heating and implications for the coupled orbital-thermal evolution of lo and Europa.
- B1. Investigate the processes that form lo's mountains and the implications for tectonics under high-heat-flow conditions that may have existed early in the history of other planets.
- B2. Understand lo's atmosphere and ionosphere, the dominant mechanisms of mass loss, and the connection to lo's volcanism.
- B3. Determine whether lo has a magnetic field and implications for the state of lo's large core.
- B4. Understand lo's surface chemistry, including volatiles and silicates, and implications for crustal differentiation and mass loss.
- B5. Improve our understanding of Jupiter system science, including meteorology, vertical structure, and auroral phenomena on Jupiter; composition and temporal variability of Europa's exosphere; Jovian magnetospheric processes; and small inner moons and rings of Jupiter.

- Understand the eruption
 mechanisms for Io's lavas and
 plumes and their implications for
 volcanic processes on Earth and the
 other terrestrial planets.
 - High-resolution repeat imaging at UV to thermal-IR wavelengths (RCam, ThM) and monitoring of escaping species (INMS).
 - Movies of dynamic phenomena such as plumes, vents, and lava lakes (RCam)
 - Requires high data rate



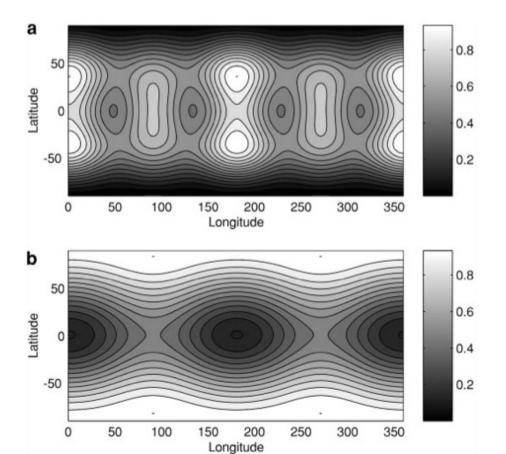
Lava fountain on Etna.

- Determine Io's interior structure, especially the melt fraction of the mantle.
 - Measure peak lava temperatures to estimate the temperature and melt fraction of the mantle (RCam)
 - Need near-simultaneous color, high SNR, and unsaturated data
 - Measure electromagnetic induction signal to test for a magma ocean (FGM).
 - Electrical conductivity of basaltic liquid is >10x that of solid
 - Plan 2 identical passes but at max and min magnetic latitude for induction responses of opposite sign.
 - Improved measurements of the shape of Io and search for nonsynchronous rotation (RCam).



Galileo eclipse image during the initial outburst (lava fountaining) from Pillan Patera, used to estimate lava temperatures on Io. IVO can provide such observations with much improved quality and more tightly constrained peak temperatures.

- Determine the properties and mechanisms of Io's tidal heating and implications for the coupled orbital-thermal evolution of Io and Europa.
 - Map and monitor global heat flow, especially polar regions (ThM).
 - Requires polar views

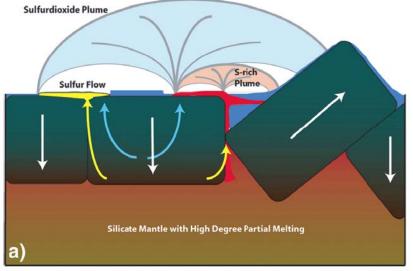


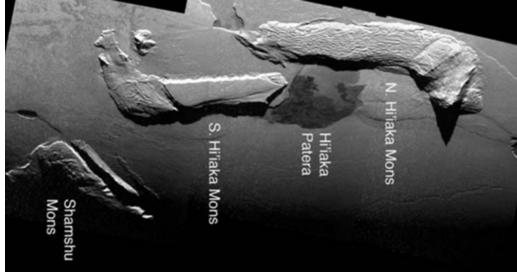
Models for heat flow patterns: asthenospheric (top) and deep mantle (bottom)

(Segatz et al., 1988; Ross et al., 1990; Tackley et al., 2001)

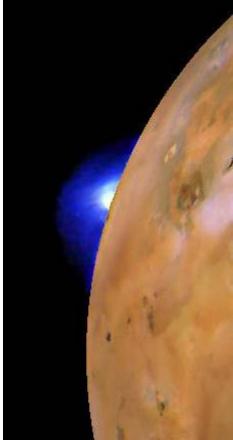
Objective A2 (melt fraction of the mantle) is also key to this objective.

- Investigate the processes that form Io's mountains and the implications for tectonics under high-heat-flow conditions that may have existed early in the history of other planets.
 - Image and measure topography of key tectonic structures (RCam).
 - WAC could provide pole-pole strip of stereo coverage from each flyby,
 25 m/pixel from 200 km range.
 - NAC can provide ~2 stereo image pairs per flyby as well as limb images.
 - Color imaging of volatiles associated with tectonic structures.



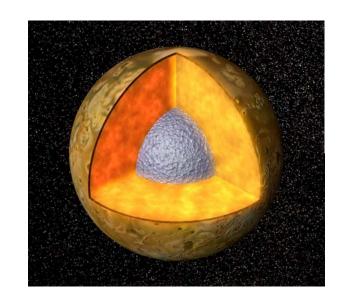


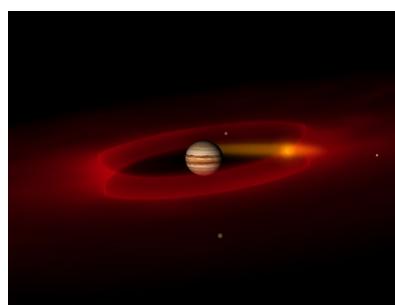
- Understand Io's atmosphere and ionosphere, the dominant mechanisms of mass loss, and the connection to Io's volcanism.
 - Determine composition and temporal and spatial variability of escaping species (INMS, RCam, FGM).
 - Directly sample gravitationally bound portions of volcanic plumes and atmosphere, late in the mission (INMS).
 - Consider two INMS instruments:
 - One mounted orthogonal to remote sensing, to acquire data at closest approach while imaging Io
 - One pointed 180 deg away from optical boresights, for flying through plumes.
 - Protect instruments, then observe plume source region at high resolution after flying through the plume.



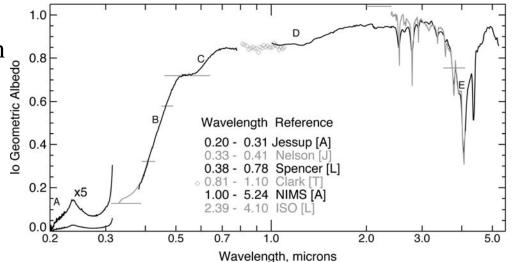
150-km high Loki plume imaged by *Voyager 1*

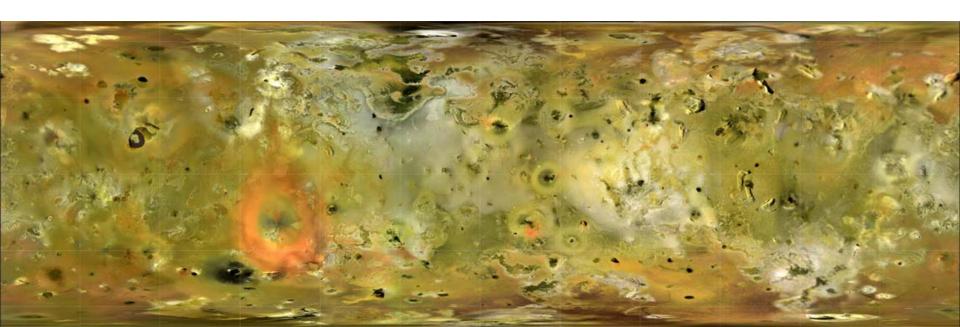
- Determine whether Io has a magnetic field and implications for the state of Io's large core.
 - Measure magnetic signatures with FGM
 - − Jupiter's field is ~2000 nT
 - Galileo found no internal dipole field at
 Io stronger than a few hundred nT, but
 core convection may be complex, leading
 to higher-order (quadrupolar or
 octupolar) fields
 - Target flybys to high magnetic latitudes and a variety of Io longitudes.
 - Combine with Galileo data (and future JEO data) to model higher-order magnetic fields.
 - Io's core may lack significant convection if the mantle is hot.



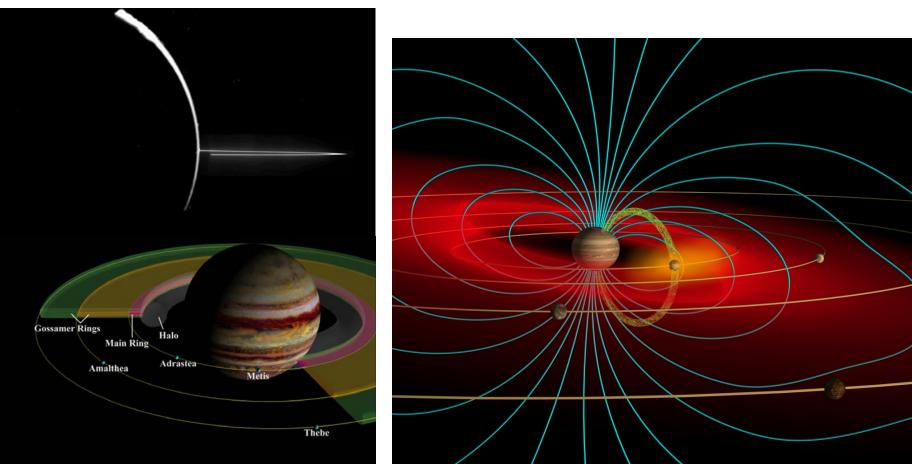


- Understand Io's surface chemistry, including volatiles and silicates, and implications for crustal differentiation and mass loss.
 - Image at UV to near-IR wavelengths
 (RCam), measure spectral thermal
 emission (ThM), and determine
 composition of escaping species (INMS).





- Improve our understanding of Jupiter system science, including meteorology, atmospheric structure, and auroral phenomena on Jupiter; composition and temporal variability of Europa's exosphere; Jovian magnetospheric processes; and small inner moons and rings of Jupiter.
 - Observations by full payload when outside Io encounter periods.
 - Strongly limited by data rate except initial orbit and extended mission



Technology Objectives

Long-term in-flight test of ASRG

- Extended mission for full 14-yr test
- Test microphonics via NAC imaging
- Make sure mission can continue if 1 of 4 ASCs fail
 - RCam NAC imaging jitter, less data volume

NASA Announced Jupiter/Europa Orbiter (JEO) for next Outer Planets Flagship Mission

- May increase attractiveness of IVO to NASA, due to risk reduction for JEO
- JEO may use ASRGs, because Pu²³⁸ is very scarce
- IVO would provide ASRG-based spacecraft design experience for Jupiter environment
- An IVO mission launched in ~2015 would provide several years of in-flight experience with ASRGs prior to JEO launch.
- IVO would provide ASRG experience in the Jupiter trapped radiation belts starting (nominally) after launch of JEO but well in advance of its arrival, providing experience with operations and mitigation of potential problems.
- New information about Jupiter radiation environment
 - Dosimeters to measure TID
 - Science instruments and star trackers provide further information

IVO Extended Mission

- Lifetime is the major concern for ASRGs, so a long-term life test is important to future exploration in the outer Solar System
 - An inflight life test of full 14-yr ASRG requirement would be best
 - IVO nominal mission is 8 years
 - Extended mission with continued lo flybys every 1-2 months would be limited to less than a few years by radiation damage.
 - Decadal Survey should urge NASA to give the ASRGs a long-term, inflight life test
- Extended mission option: pump out orbit to ~1 year period to ensure completion of 14-yr ASRG life test
 - Could provide another ~8 lo flybys
 - lo's sub-solar longitude at C/A will move ~37° per year
 - May be possible to target an outer irregular Moon, which probably has same origin as Trojans (possible captured KBOs)
 - Better measure of non-synchronous rotation of lo
- End mission by impact into lo or Jupiter
 - Planetary protection requirements



Io Science with JEO and IVO

• JEO SDT report describes 4 lo flybys, 3 with science observations

- lo is a secondary objective to JEO and could be descoped
- 85 krads/lo flyby x 4 = 340 krads before reaching Europa!

IVO will be designed to address key lo issues that JEO would or could be unable to accomplish

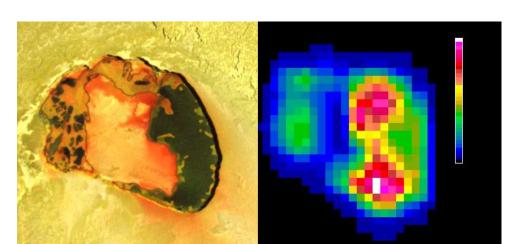
- Polar heat flow is key to distinguishing between deep and shallow tidal heating models
 - Galileo, JEO, NH, Earth-based telescopes: all equatorial views
- Measuring peak lava temperature is a requirement for IVO
- INMS will be designed for 19 km/s flybys at close range (~100 km) of lo.
 - Is JEO really going to fly through a plume and risk damage early in the mission?
- IVO orbit designed for repeat imaging at similar illumination angles
 - Random imaging leads to ambiguity over photometric or diurnal v. volcanic changes
- IVO will tweak orbits for optimal magnetic measurements
- IVO provides unique high-inclination remote sensing of lo plumes, atmosphere, and magnetospheric interactions
- In-situ measurements (INMS, magnetometer) in unexplored region via IVO orbit
- Spectral bandpasses of IVO remote sensing (15 for imaging, 10 for thermal mapper) will be chosen primarily for lo

• JEO will accomplish unique science even if IVO flies

- Unique instrumentation: UV spectrometer, etc.
- The more the better--lo is constantly changing.
 - Galileo saw new class of large polar eruptions only in its last year.

Typical Dataset from 1 Orbit

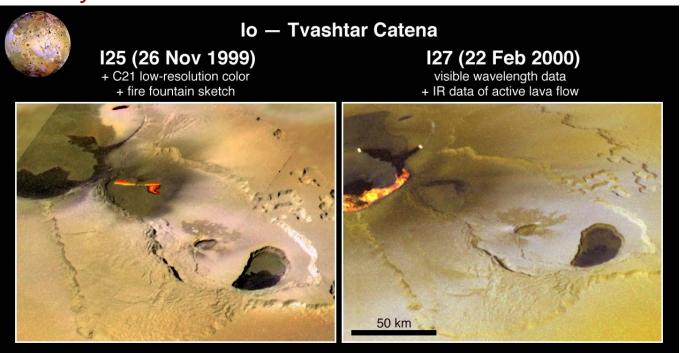
- Monitoring of Jupiter system.
- Imaging of entire illuminated hemisphere of Io at <1 km/pixel in 8 colors, key features at 10-100 m/pixel in 4 colors and <10 m/pixel clear filter.
- Imaging of high-temperature activity on night side in 4 colors at <100 m/pixel to measure liquid lava temperatures.
- Regional thermal mapping at 0.1-100 km/pixel.
- Two NAC stereo image pairs and two movies of active volcanism.
- Imaging of Io eclipses for ThM global heat flow mapping at <200 km/pixel and RCam emissions from 200-1000 nm at <16 km/pixel.
- Continuous FGM measurements.
- INMS data (hundreds of spectra) at Io closest approach, and other segments of time away from Io.
- Equivalent of ~5,000 Cassini ISS images (1,000 x 1,000 pixels)



Tupan Patera from Galileo imaging at 135 m/pixel and NIMS temperature map

IVO is exactly what NASA needs

- Exciting new target for Discovery
 - New Frontiers class science from Discovery program
- Test ASRG long-term and in Jupiter's trapped radiation
 - Given the great scarcity of Pu²³⁸ NASA is counting on ASRGs for many future missions
 - IVO provides a low-risk flight test during a 6-yr cruise to Jupiter, then an important test in a high radiation environment, then a longterm test to verify ASRGs can last 14 yrs.
 - A long-term ASRG test is important to future exploration at Saturn and beyond



Potential IVO Science Payload

- Narrow- and Wide-Angle Cameras
 - NAC (5 µrad/pixel) and WAC (220 µrad/pixel) with identical focal plane system (FPS) and digital processing unit (DPU),
 ~15 or more color bandpasses
- Thermal Mapper (ThM) (ASU)
 - 125 μrad/pixel, 10 bandpasses from 3-20 microns
- Two fluxgate magnetometers (FGM), ~3.6-m deployable boom
- Two Ion and Neutral Mass Spectrometers (INMS), 2 plasma instruments, and Langmuir probe, all with integrated electronics
- Energetic Particle Detector (EPD)
- 2-5 micron Near-IR Imager and Mapping Spectrometer
- Student-collaboration UV spectrometer or imager
- We have not yet demonstrated to ourselves that all of this can be accommodated in the Discovery mission, even with the increased cost cap.
 - But this payload likely would fit with a NF cost cap (and radioisotope power available)

Lessons relevant to Satellites panel

- DSMCE study was very helpful!
 - Only \$200 K (~\$2M for 9 studies)
 - It definitely made the difference for us in deciding to go forward with a Discovery proposal
 - There were pleasant surprises:
 - Radiation total dose only 15 krad per flyby (vs. 85 for JEO and 40 for GLL). Even JPL experts were surprised.
 - Cost estimate of 471 M was close to study goal of 450 M
 - Well within recently announced Discovery cost cap
- There were also significant omissions in Team X study
 - A first study such as this should use >30% margin on cost.
 - Needs to be 30% after Phase A study, but there will certainly be cost increases based on improved understanding.
- The NF Io Observer concept easily fits NF cost cap if radioisotope power is available
 - Probably doesn't fit with solar arrays, based on Juno example and need for more downlink and pointing flexibility and stability
 - I'm willing to make Team X study and/or DSMCE final report available to Satellites panel
 - There was also a summer student Team X study for a NF Io Observer