

Business Airplanes

Revenues and deliveries soared last year; more new models introduced.

By Fred George

eneral aviation aircraft manufacturers set a billings record in 2005, topping \$15 billion for the first time and soaring more than 27 percent compared to 2004, according to GAMA. Unit deliveries jumped more than 20 percent in 2005 vs. the previous year, reaching 3,580 aircraft. In a prime sign of the industry's health and strength, 2005 piston-engine aircraft shipments were up by one-fifth over 2004. Cessna single-engine airplane sales surged in large part because of market enthusiasm for aircraft equipped with Garmin G1000 glass cockpit avionics. Cirrus and Columbia (nee Lancair) sales jumped, as well, with Piper, Raytheon and Mooney also showing impressive gains. Virtually all of the hot sellers are equipped with large-format LCD glass cockpits. It's tough to find new general aviation aircraft from major manufacturers that still have steam gauges.

Mooney, not content to rest on last year's success, introduced the M20TN Acclaim at Sun 'n Fun this year. The Acclaim will replace the 16-year-old TLS Bravo in the product lineup. Fitted with a 280-hp, twin turbo-normalized Continental TSIO-550-G, the Acclaim should be able to cruise at 230 KTAS at FL 250 and reclaim the title of fastest production single-engine aircraft, according to the Kerrville, Texas, firm. The Extra 400, one of the Acclaim's prime

competitors, has dropped out of this year's *Handbook* as the German firm focuses on developing the Extra 500 turboprop.

Moving up into cabin-class twins, many folks are still waiting for Adam to complete post-certification work and clean up IOUs for the turbocharged, pressurized push/pull Adam 500. Nearly a year after basic certification, buyers still are waiting for full operational approval, including pressurization, full IFR authorization and flight into known icing. However, FAA certification authorities say that Adam is making headway on many fronts, so 2006 could be the year that full-scale production deliveries begin.

In the single-engine turboprop sector, EADS Socata is making news with its TBM 850, a TBM 700 with an 850-shp Pratt & Whitney Canada engine that boosts its maximum cruise speed by more than 20 knots. The TBM 850 is limited to 700 hp for landing and takeoff, but once the gear and flaps are retracted, operators can push up the power lever to take advantage of the additional 150 hp. Pilatus also has a new offering, the PC-12/47, signifying its 4,740 kilogram or 10,450 pound MTOW. The boost in weight enables typically equipped aircraft to carry six passengers with full fuel. But aircraft registered in EASA countries are restricted to a 9,920-pound MTOW, so that costs operators three passengers with full

Ibis has elected to pull its Ae270 from the *Handbook*'s listing pending the development of a new wing for the aircraft. This will

enable the aircraft to carry more fuel and raise its MTOW, making it more competitive with the PC-12. The modified Ae 270 should reappear in the 2007 *Handbook*. We're still listing the Extra 500 in the single-engine turboprop section; however, the bargain leader in pressurized single-engine turboprops has yet to earn EASA or FAA type certification.

Evolutionary progress continues in the twin-turboprop class. Raytheon's King Air C90GT, powered by twin PT6A-135 engines, is making its debut in this year's Handbook. It replaces the C90B. Taking a lesson from Waco, Texas-based Blackhawk Modifications, Raytheon fitted twin PT6A-135 engines to the C90 and boosted its maximum cruise speed by more than 40 KTAS. The engines weigh virtually the same as the -21 turboprops they replace, so there's very little downside to the upgrade, other than slightly increased fuel consumption in cruise. But the C90GT climbs so much faster than the C90B its fuel economy actually is better on typical 300and 600-nm trips. And it's capable of slashing a half hour off 1,000-nm trips in exchange for an extra 5 percent fuel burn.

Piaggio's Avanti II should be certified by the end of this year or the beginning of 2007. The latest iteration has a Rockwell Collins Pro Line 21 glass cockpit that both increases capabilities and reduces empty weight. The addition of -66B engines will boost high-altitude cruise speed, but precise data won't be

available until all flight tests are complete.

Glance please at the VLJ section. Production deliveries of both the Eclipse 500 and Cessna Citation Mustang will begin later this year. The Adam 700 becomes the third VLJ to appear in the *Handbook*. FAA insiders say that Adam is making great strides toward restoring its credibility with the folks in the aircraft certification office, so Adam 700 development is getting back on track. Look for initial FAA type certification in the late 2007 to early 2008 time frame.

The VLJ section of the *Handbook* will continue to grow in coming years as Embraer along with upstarts such as Cirrus Design and Diamond Aircraft debut their own VLJs.

Conventional business jets continue to occupy the largest section of the Handbook. In 2004, deliveries increased by more than one-quarter. Manufacturers are investing heavily in product development because profit margins are strong, but competition is fierce. Raytheon, for instance, is developing the Premier IA, an updated version of the original design with improved brakes, better systems and upgraded avionics. It was designed to compete head-on with Cessna's CJ2, but Cessna then upped the ante with the CJ2+. It's fitted with highly flat-rated 2,490-pound-thrust Williams FJ44-3-24 turbofans that improve performance, and it's also equipped with much more capable Pro Line 21 avionics, as reported in B&CA's April 2006 issue (page 34). The Premier IA retains a significant speed advantage over the CJ2+, but the Cessna has considerably better runway performance. The CJ2+'s spectacular climb performance enables it to arrive within one to six minutes of the Premier IA on typical *B&CA* fixed distance missions.

Sino Swearingen now has earned basic type certification for the SJ30-2, but it's still working on post-certification IOUs prior to full-scale production deliveries. Flight into known icing approval is complete, so



Adam 700

the remaining items should not pose major obstacles to deliveries.

Cessna's Encore+ is new to this year's *Handbook*. The newest iteration of the Citation V is fitted with a fully integrated Collins Pro Line 21 avionics package that both offers more capabilities and reduces BOW by 100 pounds.

This year's *Handbook* welcomes aboard the Grob G180 SPn Utilijet that made a surprise debut at the 2005 Paris Air Show. The G180 is designed to compete head-on with turboprops that operate from unimproved runways. It offers impressive shortand soft-field takeoff performance with range and speed competitive with the CJ3. It can carry seven people with full fuel and it has the largest cabin in the light jet class by a wide margin.

Bombardier continues to refine its Learjet 45 series aircraft. The basic models have been dropped in favor of the 40XR and 45XR, both of which are fitted with Honeywell TFE731-20BR engines, which have robust hot-and-high performance. However, the engines still are limited to 3,500 pounds of thrust for takeoff, so sealevel runway performance is far from class

leading. But when departing hot-and-high airports, these aircraft are considerably more sprightly than any direct competitors. Bombardier's Learjet 60XR, fitted with Pro Line 21 avionics, is appearing for the first time in this year's *Handbook*. The cockpit improvements add new capabilities, but the Learjet 60 needs airframe improvements if it is to remain competitive in the future.

Gulfstream's G150, in contrast, has become considerably more competitive as final flight test data have been analyzed. Its runway performance now is very competitive with most other midsize jets and it can carry four passengers 2,950 miles. That provides coast-to-coast range under most probable wind conditions and onestop range on transatlantic routes. It's priced between the Learjet 60XR and Raytheon's new Hawker 850XP, which now sports winglets. The aero improvements boost the Hawker 850XP's range by 100 nm, but it still can't match the G150's range performance.

The Hawker 850XP remains Raytheon's top offering as Hawker 4000 type certification and production deliveries remain stalled pending systems development and completion of final function and reliability testing. Although the Hawker 4000 has had a provisional type certificate since December 2004, Raytheon officials are reluctant to predict when full FAA type certification will be earned.

The Falcon 50EX is making its final appearance in the *Purchase Planning Handbook*. Dassault is concentrating on developing its widebody aircraft, such as the Falcon 2000DX, which is making its first appearance in this year's *Handbook*. Similar to the Falcon 900DX, the 2000DX is a version of an existing aircraft that has a lower price tag and less fuel capacity. The 2000DX, which replaces the Falcon 2000, offers better runway performance, more range and better climb performance, plus



Embraer Legacy

it's fitted with the Primus Epic EASy cockpit.

Bombardier's Challenger 605, announced at the 2005 NBAA Convention, joins the *Handbook* this year, replacing the Challenger 604. While its performance is virtually identical to the older model, the Challenger 605 offers plenty of improvements for passengers, including larger windows, a slightly wider cabin and better furnishings. It's also 200 pounds lighter, so typically equipped aircraft will be able to carry five to six passengers with full fuel.

Airbus' A318 Elite becomes a new heavy-weight in the business aircraft tables. With up to 4,000-nm range with an optional auxiliary center tank, the A318 is intended to compete against veteran, purpose-built heavy iron jets from Bombardier, Dassault and Gulfstream. With 23,300 pounds of thrust on each side, the A318 Elite will offer excellent runway performance, comparing most favorably with purpose-built business jets. Lufthansa Technik is handling completions, so buyers can be confident they'll be built precisely and finished on



Cessna Mustang

time. So where are the performance numbers? Airbus has undertaken a thorough aerodynamic clean-up of its single-aisle aircraft and the first beneficiary is the A318 Elite. With modified winglets, fuselage-to-wing fairings and inlet duct changes, the A318 Elite will be noticeably more fuel efficient than legacy A318 aircraft.

The 71-ton A318, though, is no match for the largest aircraft in this year's *Handbook*, the BBJ3. Based on the next-generation Boeing 737-900ER airliner, the 94-ton BBJ3 will sport Aviation Partners-Boeing winglets and have a range of more than 4,700 nm with typical 19- to 24-passenger loads. Offering 28 feet more cabin length than the original BBJ, it's designed for folks who need one-third more cabin volume than the original aircraft and 11 percent more than that offered by the BBJ2.

For now, the surge in business aircraft sales seems secure. Market predictions from Honeywell, Rolls-Royce and other firms indicate strong growth through the decade. But storm clouds in the form of evertightening fuel supplies imperil future unrestricted growth. As we mentioned in the 2005 *Handbook*, "... fuel price increases threaten to create an economic microburst in the path of general aviation . . . " Folks flying these aircraft may be able to pay \$5.00 or more per gallon for fuel, but their customers on the ground already are feeling the pinch at the pump at \$3.00 per gallon. Consumer confidence ultimately will determine if the current growth trend in business aircraft will continue or if it will once again suffer through turbulent times. B&CA

How to Use the Airplane Charts

Manufacturer, Model

In some cases, the airplane manufacturer's name is abbreviated, but the company's full name and address can be found in the "Airframe Suppliers Directory" on page 168. The model name also is included in this group.

B&CA Equipped Price

Price estimates are first quarter, current year dollars for the next available delivery. Some aircraft have long lead times, thus the actual price will be higher than our published price. Note well, manufacturers may adjust prices without notification.

Piston-powered airplanes — Computed

retail price with at least the level of equipment specified in the *B&CA* Required Equipment List on page 117.

Turbine-powered airplanes — Average price of 10 of the last 12 commercial deliveries, if available. The aircraft serial numbers aren't necessarily consecutive because of variations in completion time and because some aircraft may be configured for non-commercial, special missions.

Characteristics

Seating Capacity — Crew + Typical Executive Seating/Maximum Seating. For example, 2+8/19 indicates that the aircraft requires two pilots, there

are eight seats in the typical executive configuration and the aircraft is certificated for up to 19 passenger seats.

A four-place single-engine aircraft is shown as 1+3/3, indicating that one pilot is required and there are three other seats available for passengers. We require two pilots for all turbofan airplanes, except the CJ1+, CJ2+, Mustang, Premier I, SJ30-2 and Eclipse 500 which have, or will have, a large percentage of single-pilot operators.

Four crewmembers are specified for ultra-long-range aircraft — three pilots and one flight attendant.

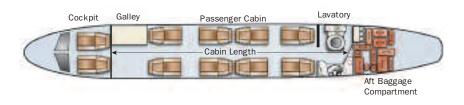
Each occupant of a turbine-powered airplane is assumed to weigh 200 pounds, thus allowing for stowed luggage and carryon items. In the case of piston-engine airplanes, we assume each occupant weighs 170 pounds. There is no luggage allowance for piston-engine airplanes.

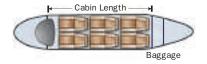
Wing Loading — MTOW divided by total wing area.

Power Loading — MTOW divided by total rated horsepower or total rated thrust.

FAR Part 36 Certificated Noise Levels — Flyover noise in A-weighted decibels (dBA) for small and turboprop aircraft. For turbofan-powered aircraft, we provide EPNdB (effective perceived noise levels) for takeoff, sideline and approach.

Cabin Length





Dimensions

External Length, Height and Span dimensions are provided for use in determining hangar and/or tiedown space requirements.

Internal Length, Height and Width are based on a completed interior, including insulation, upholstery, carpet, carpet padding and fixtures.

As shown in the Cabin Length illustration, for airplanes other than "cabinclass" models, the length is measured from the forward bulkhead ahead of the rudder pedals to the back of the rearmost passenger seat in its normal, upright position.

For "cabin-class" aircraft, we show the overall length of the passenger cabin, measured from the aft side of the forward cabin divider to the aftmost bulkhead of the cabin. The aftmost point is defined by the rear side of a baggage compartment that is accessible to passengers in flight or by the aft pressure bulkhead.

The overall length is reduced by the length of any permanent mounted system or structure that is installed in the fuselage ahead of the aft bulkhead. For example, some aircraft have full fuselage cross-section fuel tanks mounted ahead of the aft pressure bulkhead.

The second length number is the net length of the cabin that may be occupied by passengers. It's measured from the aft side of the forward cabin divider to an aft point defined by the rear of the cabin floor capable of supporting passenger seats, the rear wall of an aft galley or lavatory, an auxiliary pressure bulkhead or the front wall of the pressurized baggage compartment. Some aircraft have the same net and overall interior length because the manufacturer offers at least one interior configuration with the aft-most passenger seat located next to the front wall of the aft luggage compartment.

Interior height is measured at the center of the cross section. It may be based on an aisle that is dropped several inches below the main cabin floor that supports the passenger seats. Some aircraft have dropped aisles of varying depths, resulting in less available interior height in certain sections of the cabin.

Two width dimensions are shown for multiengine turbine airplanes — one at the widest part of the cabin and the other at floor level. The dimensions, however, are not completely indicative of the usable space in a specific aircraft because of individual variances in interior furnishings.



Gulfstream 450

Power

Number of engines, if greater than one, and the abbreviated name of the manufacturer:

CFE — ASE/GE joint venture

RR - Rolls-Royce

CFMI — CFM International

Cont — Teledyne Continental

Hon — Honeywell

IAE — International Aero Engines

Lyc — Textron Lycoming

P&WC — Pratt & Whitney Canada

Wms/RR — Williams/Rolls-Royce

Output — Takeoff rated horsepower for propeller-driven aircraft or pounds thrust for turbofan aircraft. If an engine is flat rated, enabling it to produce takeoff rated output at a higher than ISA (standard day) ambient temperature, the flat rating limit is shown as ISA+XX°C. Highly flat-rated engines, i.e., engines that can produce takeoff rated thrust at a much higher than standard ambient temperature, typically provide substantially improved highdensity altitude and high-altitude cruise performance.

Inspection Interval is the longest, scheduled hourly major maintenance interval for the engine, either "t" for TBO or "c" for compressor zone inspection. OC is shown only for engines that have "on condition" repair or replace parts maintenance.

Weights (lb)

Weight categories are listed as appropriate to each class of aircraft.

Max Ramp — Maximum ramp weight

Max Takeoff — Max takeoff weight as determined by structural limits.

Max Landing — Max landing weight as determined by structural limits.

Zero Fuel — Maximum zero fuel

weight, shown by "c" indicating the certificated MZFW or "b," a B&CAcomputed weight based on MTOW minus the weight of fuel required to fly 1.5 hours at high-speed cruise.

Max ramp, max takeoff and max landing weights may be the same for light aircraft that may only have a certificated max takeoff weight.

EOW/BOW — Empty operating weight is shown for piston-powered airplanes. Basic operating weight, which essentially is EOW plus required flightcrew, is shown for turbine-powered airplanes. EOW is based on the factory standard weight, plus items specified in the B&CA Required Equipment List, less fuel and oil. BOW, in contrast, is based on the average EOW weight of the last 10 commercial deliveries, plus 200 pounds for each required crewmember. We require four crewmembers — three flight crew and one cabin attendant for ultra-long-range aircraft.

There is no requirement to add in the weight of cabin stores, but some manufacturers choose to include galley stores and passenger supplies as part of the BOW buildup. Life vests, life rafts and appropriate deep-water survival equipment are included in the weight buildup of the 80,000-pound-plus, ultralong-range aircraft.

Max Payload — Zero fuel weight minus EOW or BOW, as appropriate. For piston-engine airplanes, max payload frequently is a computed value because it is based on the B&CA ("b") computed maximum ZFW.

Executive Payload — Based on 170 pounds per occupant for multiengine piston-engine airplanes and 200 pounds per occupant for turbineengine airplanes, as shown in the executive seating section of the "Characteristics" section. Pilots and passengers, however, are counted as occupants in piston-engine airplanes. Only passengers are counted as occupants in turbine-powered airplanes because the required crew is included in the BOW. If the executive payload exceeds the maximum payload, we use maximum payload. Executive payload is not computed for single-engine piston airplanes.

Max Fuel — Usable fuel weight based on 6.0 pounds per U.S. gallon for avgas or 6.7 pounds per U.S. gallon for jet fuel. Fuel capacity includes optional, auxiliary and long-range tanks, unless otherwise noted.

Available Payload With Max Fuel — Max ramp weight minus the tanks-full weight, not to exceed zero fuel weight minus EOW or BOW.

Available Fuel With Max Payload — Max ramp weight minus zero fuel weight, not to exceed maximum fuel capacity.

Available Fuel With Executive Payload — Available fuel weight based on max ramp weight minus BOW plus executive payload, up to the actual fuel capacity.

Limits

B&CA lists V speeds and other limits as appropriate to the class of airplane. These are the abbreviations used on the charts:

VNE — Never exceed speed (red line for piston-engine airplanes).

VNO — Normal operating speed (top of green arc for piston-engine airplanes).

Vmo — Maximum operating speed (red line for turbine-powered airplanes).

Mmo — Maximum operating Mach

number (red line for turbofan-powered airplanes and a few turboprop airplanes).

FL/Vmo — Transition altitude at which Vmo equals Mmo (large turboprop and turbofan aircraft).

Va — Maneuvering speed (except for certain large turboprop and all turbofan aircraft).

VDEC — Accelerate/stop decision speed (multiengine piston and light multiengine turboprop airplanes).

VMCA — Minimum control airspeed, airborne (multiengine piston and light multiengine turboprop airplanes).

Vso — Maximum stalling speed, landing configuration (single-engine airplanes) in KCAS.

Vx — Best angle-of-climb speed (single-engine airplanes).

VXSE — Best angle-of-climb speed, one-engine inoperative (multiengine piston and multiengine turboprop airplanes under 12,500 pounds).

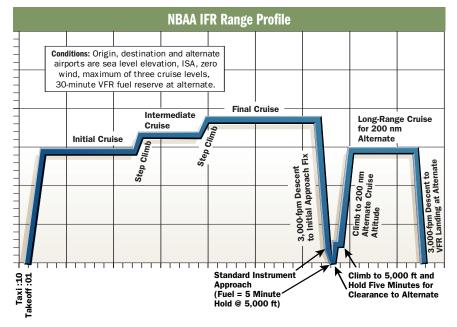
W — Best rate-of-climb speed (single-engine airplanes).

Wse — Best rate-of-climb speed, oneengine inoperative (multiengine piston and multiengine turboprop airplanes under 12,500 pounds).

V2 — Takeoff safety speed (large turboprops and turbofan airplanes).

VREF — Reference landing approach speed (large turboprops and turbofan airplanes, four passengers, NBAA IFR reserves; eight passengers for ultralong-range aircraft).

PSI — Cabin pressure differential (all pressurized airplanes).



Airport Performance

Approved Flight Manual takeoff runway performance is shown for sea-level, standard day and for 5,000 feet elevation/25°C day, density altitude. Allengine takeoff distance (TO) is shown for single-engine and multiengine piston, and turboprop airplanes with an MTOW of less than 12,500 pounds.

Accelerate/Stop distance (A/S) is shown for small multiengine piston and small turboprop airplanes. Takeoff field length (TOFL), the greater of the one-engine inoperative (OEI) takeoff distance or the accelerate/stop distance, is shown for FAR Part 23 Commuter Category and Part 25 airplanes. If the accelerate/stop and OEI accelerate/go distances are equal, the TOFL is the balanced field length.

Landing distance (LD) is shown for Part 23 Commuter Category and Part 25 Transport Category airplanes. The landing weight is BOW plus four passengers and NBAA IFR fuel reserves. We assume that 80,000-pound-plus ultralong-range aircraft will have eight passengers on board.

The V2 and VREF speeds are useful for reference when comparing the TOFL and LD numbers because they provide an indication of potential minimum-length runway performance when low RCR or runway gradient is a factor.

B&CA lists two additional numbers for large turboprop- and turbofan-powered airplanes. First, we publish the **Mission Weight**, which is the lower of:

- (1) The actual takeoff weight with four passengers (eight passengers for ULR aircraft) and full fuel when departing from a 5,000-foot/25°C airport.
- (2) The maximum allowable takeoff weight based when departing with the same passenger load and at the same density altitude.

For two-engine aircraft, the mission weight, when departing from a 5,000-foot, ISA+20°C airport, may be less than the MTOW because of Part 25 second-segment, OEI, climb performance requirements. Aircraft with highly flat-rated engines are less likely to have a mission weight that is performance limited when departing from hot-and-high airports.

For three- or four-engine aircraft, the mission weight usually is based on full tanks and the actual number of passengers, rather than being performance limited.

Second, we publish the **NBAA IFR Range** for the hot-and-high departure mission weight, assuming a transition into standard day, ISA flight conditions after take-

off. For purposes of computing NBAA IFR range, the aircraft is flown at the long-range cruise speed shown in the "Cruise" block or at the same speed as shown in the "Range" block.

Climb

The all-engine time to climb provides an indication of overall climb performance, especially if the aircraft has an allengine service ceiling well above our sample top-of-climb altitudes.

We provide the all-engine time to climb to one of three specific altitudes, based on type of aircraft departing at MTOW from a sea-level, standard-day airport:

(1) FL 100 (10,000 feet) for normally aspirated single-engine and multiengine piston aircraft, plus pressurized single-engine piston aircraft and un-pressurized turboprop aircraft.

(2) FL 250 for pressurized singleengine and multiengine turboprops.

(3) FL 370 for turbofan-powered aircraft. The data are published as time to climb in minutes/climb altitude. For example, if a non-pressurized twin-engine piston aircraft can depart from a sea-level airport at MTOW and climb to 10,000 feet in eight minutes, the time to climb is expressed as 8/FL 100.

We also publish the initial all-engine climb feet-per-nautical-mile gradient, plus initial engine-out climb rate and gradient, for single-engine and multiengine pistons and turboprops with MTOWs of 12,500 pounds or less.

The OEI climb rate for multiengine aircraft at MTOW is derived from the Airplane Flight Manual. OEI climb rate and gradient is based on landing gear retracted and wing flaps in the takeoff configuration used to compute the published takeoff distance.

The climb gradient for such airplanes is obtained by dividing the product of the climb rate (fpm) in the Airplane Flight Manual times 60 by the Vy or Vyse climb speed, as appropriate.

The OEI climb gradients we show for Part 23 Commuter Category and Part 25 Transport Category aircraft are the second-segment net climb performance numbers published in the AFMs. Please note:

The AFM net second-segment climb performance numbers are adjusted downward by 0.8 percent to compensate for variations in pilot technique and ambient conditions.

The OEI climb gradient is computed at the same flap configuration used to calculate the takeoff field length.



Bombardier Learjet 45XR

Ceilings (ft)

Maximum Certificated Altitude — Maximum allowable operating altitude determined by airworthiness authorities.

All-Engine Service Ceiling — Maximum altitude at which at least a 100-fpm rate of climb can be attained, assuming the aircraft departed a sea-level, standard-day airport at MTOW and climbed directly to altitude.

OEI (engine-out) Service Ceiling — Maximum altitude at which a 50-fpm rate of climb can be attained, assuming the aircraft departed a sea-level, standard-day airport at MTOW and climbed directly to altitude.

Sea-Level Cabin — Maximum cruise altitude at which a 14.7-psia, sea-level cabin altitude can be maintained in a pressurized airplane.

Cruise

Cruise performance is computed using EOW with four occupants or BOW with four passengers and one-half fuel load. Ultra-long-range aircraft carry eight passengers for purposes of computing cruise performance. Assume 170 pounds for each occupant of a pistonengine airplane and 200 pounds for each occupant of a turbine-powered aircraft.

Long Range — TAS, Fuel Flow in pounds/hour, flight level (FL) cruise Altitude and Specific Range for long-range cruise by the manufacturer.

Recommended (piston-engine airplanes) — **TAS, Fuel Flow** in pounds/hour, flight level (FL) cruise **Altitude** and **Specific Range** for normal cruise performance specified by the manufacturer.

High Speed — **TAS, Fuel Flow** in pounds/ hour, flight level (FL) cruise **Altitude** and **Specific Range** for short-range, highspeed performance specified by the manufacturer. Speed, fuel flow, specific range and altitude in each category are based on one mid-weight cruise point. They are not an average for the overall mission.

B&CA imposes a 12,000-foot maximum cabin altitude requirement on CAR3/ FAR Part 23 normally aspirated aircraft. Turbocharged airplanes are limited to FL 250, providing they are fitted with supplemental oxygen systems having sufficient capacity for all occupants for the duration of the mission.

Pressurized CAR 3/FAR Part 23 aircraft are limited to a maximum cabin altitude of 10,000 feet. For Part 23 Commuter Category and Part 25 aircraft, the maximum cabin altitude for computing cruise performance is 8,000 feet.

To conserve space, we use flight levels (FL) for all cruise altitudes, which is appropriate considering that we assume standard day ambient temperature and pressure conditions. Cruise performance is subject to *B&CA*'s verification.

NOTICE TO READERS

During recent years, the U.S. Federal Trade Commission has conducted investigations into the practice of certain industries in fixing and advertising list prices. It is the position of the FTC that it is deceptive to the public and against the law for list prices of any product to be specified or advertised in a trade area if the majority of sales are made at less than those prices.

B&CA is not in a position to know the prices for most of the sales in each trading area in the United States for each of the products in this issue. Therefore, the prices shown in the tables and text in the Purchase Planning Handbook are based on suggested list prices furnished to us by the manufacturers or distributors, or on prices estimated by the editors. It may be possible to purchase some items in your trading area at prices less than those reported in this issue of B&CA. Also, almost all manufacturers and distributors caution that prices are subject to change without notice.



Dassault 2000EX

Ranges

B&CA shows various paper missions for each aircraft that illustrate range versus payload tradeoffs, runway and cruise performance, plus fuel efficiency. Similar to the cruise profile calculations, *B&CA* limits the maximum altitude to 12,000 feet for normally aspirated, non-pressurized CAR 3/FAR Part 23 aircraft, 25,000 feet for turbocharged airplanes with supplemental oxygen, 10,000 feet cabin altitude for pressurized CAR 3/FAR Part 23 airplanes and 8,000 feet cabin altitude for Part 23 Commuter Category or Part 25 aircraft.

Seats-Full Range (single-engine piston airplanes) — Based on typical executive configuration with all seats filled with 170-pound occupants, with maximum available fuel less 45-minute IFR fuel reserves. We use the lower of seats full or maximum payload.

Tanks-Full Range (single-engine piston airplanes) — Based on one 170-pound pilot, full fuel less 45-minute IFR fuel reserves.

Executive Payload (multiengine piston airplanes and single-engine turboprops) — Based on typical executive configuration with all seats filled with 170-pound occupants, maximum available fuel less 45-minute IFR fuel reserves.

We use the lower of seats full or maximum payload.

Max Fuel With Available Payload (single-engine turboprops) — Based on BOW, plus full fuel and the maximum available payload up to maximum ramp weight. Range is based on arriving at destination with NBAA IFR fuel reserves, but only a 100-mile alternate is required.

Ferry (multiengine piston airplanes and single-engine turboprops) — Based on one 170-pound pilot, maximum fuel less 45-minute IFR fuel reserves.

Please note: None of the missions for piston-engine aircraft include fuel for diverting to an alternate. However, single-engine turboprops are required to have NBAA IFR fuel reserves, but only a 100-mile alternate is required.

NBAA IFR range format cruise profiles, having a 200-mile alternate, are used for Part 25 Transport Category turbine-powered aircraft. In the case of Part 23 turboprops, including those certificated in the Commuter Category, and Part 23 turbofan aircraft, only a 100-mile alternate is needed. The difference in alternate requirements should be kept in mind when comparing range performance of various classes of aircraft.

Max Payload With Available Fuel (multiengine turbine airplanes) — Based on aircraft loaded to maximum zero fuel weight with maximum available fuel up to maximum ramp weight, less NBAA IFR fuel reserves at destination.

Max Fuel With Available Payload (multiengine turbine airplanes) — Based on BOW plus full fuel and maximum available payload up to maximum ramp weight. Range based on NBAA IFR reserves at destination.

Full/Max Fuel With Four Passengers (multiengine turbine airplanes) — Based on BOW plus four 200-pound passengers



M28 Skytruck

and the lesser of full fuel or maximum available fuel up to maximum ramp weight. Ultra-long-range aircraft must have eight passengers on board.

Ferry (multiengine turbine airplanes) — Based on BOW, required crew and full fuel, arriving at destination with NBAA IFR fuel reserves.

We allow 2,000-foot increment step climbs above the initial cruise altitude to improve specific range performance. The altitude shown in the range section is the highest cruise altitude for the trip — not the initial cruise or mid-mission altitude.

The range profiles are in Nautical Miles, and the Average Speed is computed by dividing that distance by the total flight time or weight-off-wheels time en route. The Fuel Used or Trip Fuel includes the fuel consumed for start, taxi, takeoff, cruise, descent and landing approach, but not after-landing taxi or reserves.

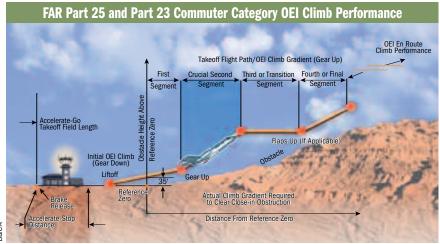
The **Specific Range** is obtained by dividing the distance flown by the total fuel burn. The altitude is the highest cruise altitude achieved on the specific mission profile shown.

Missions

Various paper missions are computed to illustrate the runway requirements, speeds, fuel burns and specific range, plus cruise altitudes. The mission ranges are chosen to be representative for the airplane category.

All fixed-distance missions are flown with four passengers on board, except for ultra-long-range airplanes, which have eight passengers on board. The pilot is counted as a passenger on board piston-engine airplanes. If an airplane cannot complete a specific fixed distance mission with the appropriate payload, *B&CA* shows a reduction of payload in the remarks section or marks the fields NP (Not Possible) at our option.

Runway performance is obtained from the Approved Airplane Flight



B&CA Required Equipment List

Jets ≥20,000 lb

Jets <20,000 lb

Turboprops >12,500 lb

Turboprops ≤12,500 lb

Single-Engine Turboprops

Multiengine Pistons, Turbocharged

Multiengine Pistons

Single-Engine Pistons, Pressurized

airplanes.

Flight Time (takeoff to touchdown, or weight-off-wheels, time) is shown for turbine airplanes. Some piston-engine manufacturers also include taxi time, resulting in a chock-to-chock, block time measurement. Fuel Used, though, is the actual block fuel burn for each type of aircraft, but it does not include fuel reserves. The cruise Altitude shown is that which is specified by the manufac-

Manual. Takeoff distance is listed for single-engine airplanes; acceler-

ate/stop distance is listed for piston-

twins and light turboprops; and take-

off field length, which often corre-

sponds to balanced field length, is used for Part 23 Commuter Category

and Part 25 large Transport Category

200 nm — Piston-engine airplanes.

turer for fixed-distance missions.

500 nm — Piston-engine airplanes.

300 nm — Turbine-engine airplanes, except ultra-long-range.

600 nm — Turbine-engine airplanes, except ultra-long-range.

1,000 nm — All turbine-engine airplanes.

3,000 nm — Ultra-long-range, turbine-engine airplanes.

6,000 nm — Ultra-long-range, turbine-engine airplanes.

Remarks

In this section, *B&CA* generally includes the base price, if it is available or applicable; the certification basis and year; and any notes about estimations, limitations or qualifications regarding specifications, performance or price.

All prices are in 2006 dollars, FOB at a U.S. delivery point, unless otherwise noted. The certification basis includes the regulation under which the airplane originally was type certificated, the year in which it originally was certificated and, if applicable, subsequent years during which the airplane was re-certificated.

"All data preliminary" indicates that actual aircraft weight, dimension and performance numbers may vary considerably after the model is certificated and delivery of completed aircraft begins. These aircraft are listed in italics.

General

The following abbreviations are used throughout the tables: "NA" means not available; "—" indicates the information is not applicable; and "NP" signifies that specific performance is not possible. **B&CA**

Siligle-Eligille Pistolis, i	riessuriz	eu							
Single-Engine Pistons, Turboch	arged								
Single-Engine Pistons									
Single-Lingille i istons									
POWERPLANT SYSTEMS									
Batt temp indicator (nicad only, for each battery)					•	•	•	•	•
Engine synchronization								•	•
Fire detection, each engine					•	•	•	•	•
Fire extinguishing, each engine							•		•
Propeller, reversible pitch					•	•	•		
Propellers, synchronized						•	•		
Thrust reversers/attenuators								•	_
AVIONICS									
ADF					•	•	•	•	_
Air data computer					_	•	•	•	_
Altitude alerter			_		•	•	•	•	_
Altitude encoder		•	÷	-	•	•	•	•	
Antennas, headsets, microphones		•	•	-	•	•		•	
		•	_	_	_	-			_
Automatic flight guidance, 2-axis, alt hold Automatic flight guidance, 3-axis, alt hold	_		•	•	•	•	•	•	
DME			_		÷	÷			
EFIS					_	÷	•	•	-
ELT			•		•	·	•	•	ì
Flight director			_		_	·	•	•	-
FMS, TSO C115 or GPS, TSO C129 IFR approach	•	•	•	•	•	•	•	•	
Glideslope receiver		•	Ť	ě	Ť	0	•	•	
HSI, slaved (or equivalent EFIS function)		•	•	•	•	•	•	•	
		•	•	•	•	•	•	•	
Radio altimeter					•	•	•	•	•
Radiotelephone							•	•	•
RMI (or equivalent function on EFIS display)					•				
RVSM certification								•	•
TAWS								•	•
TCAS I/II (FAR Part 25 airplanes only)								•	•
Transponder	•	•	•	•	•				•
VHF comm, 25-kHz spacing									•
VHF comm, 8.33-kHz spacing									•
VHF nav, 360-channel									_
Weather radar					•	•	•	•	_
GENERAL									
Air conditioning, vapor cycle (not required with APU)		•			•	•	•	•	_
Anti-skid brakes							•	•	_
APU (required for air-start engines, ACM air conditioning)							_		_
Cabin/cockpit dividers			_		_		•	•	_
Corrosion-proofing, internal		•	•	•	•	•	•	•	_
	•	•	•	•	•	•	•	•	_
Fire extinguisher, cabin					•	•	•	•	_
Fire extinguisher, cockpit		•	•	•	•	•	•	•	_
		•	•	-	•	•	•	•	•
		•	÷	•	÷	•	•	•	-
Headrests, air vents, all seats Lavatory			_		_		÷	÷	_
Lights, strobe/anti-collision beacon, navigation, landing/taxi	•	•	•		•	•	÷	÷	_
Lights, internally lighted instrument, cockpit flood, courtesy		•	Ť	·	Ť	·	÷	i	
Oxygen, supplemental, all seats			_	•	Ť	·	÷	i	ì
Refreshment center					_	·	÷	i	_
Seats, crew, articulating			•	•	•	ě	ě	ě	
Seats, passenger, reclining		ě	Ť	•	Ť	ě	ě	Ŏ	ì
Shoulder harness, all seats and crew with inertia reel		•	Ť	•	Ť	•	•	•	ì
Tables, cabin work					•	•	•	•	
CE AND RAIN PROTECTION									
Alternate static pressure source (not required with 2 DADC)	•	•	•	•	•	•			
Approval, flight into known icing		•	•	•	•	•	•	•	•
Ice protection plates						•	•		
Pitot heat	•	•	•	•	•	•	•	•	•
Static wicks	•	•	•	•	•	•	•	•	_
Windshield rain removal, mechanical or repellent coating						•	•	•	•
NSTRUMENTATION									
Angle-of-attack stall margin indicator								•	
EGT	•	•	•	•					
IVSI (or equivalent EFIS, DADC function)						•			•
Outside air temperature gauge	•	•	•	•	•	•	•	•	•
Primary flight instruments	•		•						

Dual required

Required

SINGLE-ENGINE PISTONS NORMALLY ASPIRATED

3&CA Equipped Price Characteristics Seating Wing Loading Power Loading Noise (dBA)	SR20 \$249,750	CE-182T	SR22-G2 SR22	Arrow PA-28R-200	Columbia 356 LC42-550FG
Characteristics Seating Wing Loading Power Loading		\$326,150	\$349,750	\$360.650	\$378,900
Wing Loading Power Loading	1+3/3	1+3/3	1+3/3	1+3/3	1+3/3
Power Loading	22.2	17.8	23.5	16.2	24.1
	15.00	13.48	10.97	13.75	10.97
Noise (dbA)					
	83.4	77.7	83.7	75.5	84.0
mensions External Length	26.0	29.0	26.0	24.7	25.2
t) Height	8.6	9.3	8.8	7.9	9.0
Span	35.5	36.0	38.3	35.4	36.1
Internal Length	8.0	7.2	8.0	7.3	7.9
Height	4.1	4.0	4.1	4.1	4.2
Width	4.1	3.5	4.1	3.5	4.2
ower Engine	Cont	Lyc	Cont	Lyc	Cont
Lingino	10-360-ES	IO-540-AB1A5	10-550-N	IO-360-C1C6	10-550-N
Output (hn)					
Output (hp)	200 hp	230 hp	310 hp	200 hp	310 hp
Inspection Interval	2,000t	2,000t	1,700t	2,000t	2,000t
leights (lb) Max Ramp	3,000	3,110	3,400	2,758	3,400
Max Takeoff	3,000	3,100	3,400	2,750	3,400
Max Landing	2,900	2,950	3,400	2,750	3,230
Zero Fuel	2,900c	2,985b	3,250b	2,645b	3,228c
EOW	2,070	1,944	2,250	1,787	2,400
Max Payload	830			858	
		1,041	1,000		828
Useful Load	930	1,166	1,150	971	1,000
Max Baggage	130	200	130	200	120
Max Fuel	336	522	486	432	588
Available Payload w/Max Fuel	594	644	664	539	412
Available Fuel w/Max Payload	100	126	150	113	172
mits VNE	200	175	201	186	235
Vno	165	140	178	148	179
VA	131	110	133	120	148
		·		·	
rport TO (SL elev./ISA temp.)	2,064	1,350	1,575	1,600	1,250
erformance TO (5,000' elev. @25°C)	3,486	2,708	2,717	3,200	1,800
Vso	56	50	59	55	57
Vx	81	65	78	72	80
VY	96	82	101	90	106
limb Time to Climb (min.)/Altitude	16/FL 100	15/FL 100	10/FL 100	17/FL 100	11/FL 100
Initial Gradient (ft/nm)	581	694	864	584	947
eiling (ft) Service	17,500	18,100	17,500	16,200	18,000
		123	162	117	154
3	140				
Range Fuel Flow	50	60	79	48	64
Altitude	FL 080	FL 100	FL 080	FL 080	FL 080
Specific Range	2.800	2.050	2.051	2.438	2.406
Rec'd TAS	152	135	172	131	183
Fuel Flow	63	68	91	55	97
Altitude	FL 080	FL 100	FL 080	FL 080	FL 080
Specific Range	2.413	1.985	1.890	2.382	1.887
High TAS	159	140	181	142	191
Speed Fuel Flow	70	77	100	70	104
•					
Altitude	FL 080	FL 060	FL 080	FL 080	FL 080
Specific Range	2.271	1.818	1.810	2.029	1.837
anges Seats Nautical Miles	584	735	841	327	580
Full Average Speed	140	120	162	136	153
Fuel Used	210	435	411	159	320
Specific Range/Altitude	2.781/FL 080	1.690/FL 060	2.046/FL 080	2.057/FL 080	1.813/FL 08
Tanks Nautical Miles	764	980	873	907	1,242
Full Average Speed	140	123	162	135	153
Fuel Used	275		427	396	
		471			515
Specific Range/Altitude	2.778/FL 080	2.081/FL 100	2.044/FL 080	2.290/FL 080	2.412/FL 08
issions 200 nm Runway	1,446	1,514	1,020	1,350	1,250
occupants) Block Time	1 + 18	1 + 29	1 + 06	1 + 28	1 + 10
Fuel Used	100	128	122	107	109
Specific Range/Altitude	2.000/FL 080	1.563/FL 060	1.639/FL 080	1.869/FL 080	1.835/FL 08
500 nm Runway	1,446	1,514	1,341	1,480	1,250
	3 + 17	3 + 37	2 + 45	3 + 41	2 + 48
	226	274	308	230	246
Block Time					
Block Time Fuel Used	2.212/FL 080	1.825/FL 060	1.623/FL 080	2.174/FL 080	2.033/FL 12
Block Time Fuel Used Specific Range/Altitude		\$258,500	\$334,700	\$292,400	\$378,900
Block Time Fuel Used Specific Range/Altitude emarks Suggested Base Price	\$236,700			CAD 2 40C0 /70	FAR 23
Block Time Fuel Used Specific Range/Altitude		FAR 23, 1996/01	FAR 23, 2000	CAR 3, 1960/78	
Block Time Fuel Used Specific Range/Altitude emarks Suggested Base Price	\$236,700	FAR 23, 1996/01 Includes G1000	FAR 23, 2000	CAR 3, 1960/78	Includes Avidy
Block Time Fuel Used Specific Range/Altitude Suggested Base Price	\$236,700		FAR 23, 2000	CAR 3, 1960/78	Includes Avidy Entegra.
Block Time Fuel Used Specific Range/Altitude smarks Suggested Base Price	\$236,700	Includes G1000 with TIS, XM wx	FAR 23, 2000	CAR 3, 1900/18	
Block Time Fuel Used Specific Range/Altitude marks Suggested Base Price	\$236,700	Includes G1000	FAR 23, 2000	CAR 3, 1900/78	
Block Time Fuel Used Specific Range/Altitude marks Suggested Base Price	\$236,700	Includes G1000 with TIS, XM wx	FAR 23, 2000	CAR 3, 1900/18	
Block Time Fuel Used Specific Range/Altitude marks Suggested Base Price	\$236,700	Includes G1000 with TIS, XM wx	FAR 23, 2000	CAR 3, 1960/18	

SINGLE-ENGINE PISTONS NORMALLY ASPIRATED

Manufacturer Model	Socata Trinidad GT TB-20 GT	Mooney Ovation2 GX M-20R	Cessna Stationair CE-206H	New Piper Piper 6X PA-32-201FT	Beechcraft Bonanza G36
3&CA Equipped Price	\$381,500	\$442,000	\$448,160	\$453,000	\$691,390
Characteristics Seating	1+3/4	1+3/4	1+5/5	1+4/5	1+4/5
Wing Loading	24.1	19.3	20.7	20.2	20.2
Power Loading	12.34	12.03	12.00	12.00	12.17
Noise (dBA)	74.0	72.6	84.5	83.9	76.7
imensions External Length	25.3	26.9	28.3	27.9	27.5
t) Height	9.4	8.3	9.3	8.5	8.6
Span	32.1	36.1	36.0	36.2	33.5
Internal Length	8.3	8.1	9.3	9.7	12.6
Height	3.9	3.7	4.1	3.5	4.2
Width	4.2	3.6	3.7	4.1	3.5
ower Engine	Lyc	Cont	Lyc	Lyc	Cont
_	IO-540-C4D5D	10-550-G	IO-540-AC1A5	IO-540-K1G5	IO-550-B
Output (hp)	250 hp	280 hp	300 hp	300 hp	300 hp
Inspection Interval	2,000t	2,000t	2,000t	2,000t	1,700t
/eights (lb) Max Ramp	3,097	3,374	3,614	3,615	3,663
Max Takeoff	3,086	3,368	3,600	3,600	3,650
Max Landing	3,086	3,200	3,600	3,600	3,650
Zero Fuel	2,959b	3,226b	3,448b	3,434b	3,509b
EOW	1,815	2,260	2,202	2,214	2,530
Max Payload	1,144	966	1,246	1,220	979
Useful Load	1,282	1,114	1,412	1,401	1,133
Max Baggage	1,282	120	200	200	670
Max Fuel	517	612	522	612	444
	765	502	522 890	789	689
Available Payload w/Max Fuel					
Available Fuel w/Max Payload	139	149	166	182	154
mits VNE	189	195	182	189	203
Vno	151	174	149	150	165
V _A	130	123	125	132	139
irport TO (SL elev./ISA temp)	2,083	1,620	1,860	2,028	1,913
erformance_T0 (5,000' elev. @25°C)	3,400	1,980	3,375	3,763	4,145
Vso	58	59	57	59	59
Vx	81	80	70	80	84
VY	95	105	86	90	100
limb Time to Climb (min.)/Altitude	12/FL 100	11/FL 100	15/FL 100	15/FL 100	14/FL 100
Initial Gradient (ft/nm)	783	714	691	967	730
eiling (ft) Service	20,000	20,000	15,700	17,200	18,500
ruise Long TAS	139	163	127	136	137
Range Fuel Flow	64	50	78	87	56
Altitude	FL 100	FL 120	FL 120	FL 120	FL 120
Specific Range	2.172	3.260	1.628	1.563	2.446
Rec'd TAS	151	186	135	144	163
Fuel Flow	73	84	88	99	80
Altitude	FL 100	FL 120	FL 080	FL 100	FL 100
Specific Range	2.068	2.214	1.534	1.455	2.038
High TAS	165	190	142	149	174
Speed Fuel Flow	85	95	100	111	94
Altitude	FL 100	FL 080	FL 060	FL 070	FL 080
Specific Range	1.941	2.000	1.420	1.342	1.851
anges Seats Nautical Miles	715	1,075	326	403	527
Full Average Speed	150	161	123	128	130
Fuel Used	355	438	326	386	240
Specific Range/Altitude	2.014/FL 100	2.454/FL 120	1.000/FL 060	1.044/FL 080	2.196/FL 040
Tanks Nautical Miles	935	1,465	595	800	930
Full Average Speed	149	173	127	132	140
Full Average Speed Fuel Used	469	558	456	547	403
Specific Range/Altitude	1.994/FL 100	2.625/FL 120	1.305/FL 120	1.463/FL 100	2.308/FL 080
lissions 200 nm Runway	1,560	1,230	1,860	1,500	1,750
l occupants) Block Time	1,500	1 + 13	1+33	1,500	1,730
Fuel Used	104	115	166	159	130
Specific Range/Altitude	1.923/FL 100	1.739/FL 050	1.205/FL 060	1.258/FL 100	1.538/FL 060
500 nm Runway		1,290			
	1,650	,	1,860	1,720	1,850
Block Time	3 + 11	2 + 58	3 + 36	3 + 50	2 + 54
Fuel Used	248	221	377	359	304
Specific Range/Altitude	2.016/FL 100	2.262/FL 100	1.326/FL 060	1.393/FL 100	1.645/FL 060
emarks Suggested Base Price	\$381,500	\$442,000	\$448,160	\$382,900	\$667,000
Certification Basis	FAR 23, 1981/84	CAR 3, 1955/94	FAR 23, 1998	CAR 3, 1965/96	CAR 3,
	Includes KFC225,	Includes Garmin	Includes G1000	Includes Avidyne	1956/69/83/0
	KMD 550,	G1000.	with TIS, XM wx	Entegra.	Includes Garmi
	WX 500; Garmin		and Stormscope.		G1000.
	430/530 opt'l.				

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SINGLE-ENGINE PISTONS TURBOCHARGED

Manufacti Model		Cessna Turbo Skylane CE-T182T	Socata Trinidad GT T/C TB-21 GT	Lancair Columbia 400 LC41-550-FG	Cessna Turbo Stationair CE-T206H	New Piper Piper 6XT PA-32-301XTC	Mooney Acclaim MO-20TN
	lipped Price	\$355,050	\$431,670	\$479,900	\$482,160	\$490,315	\$495,000
Character	9	1+3/3	1+3/4	1+3/4	1+5/5	1+4/5	1+3/5
	Wing Loading	17.8	24.1	25.5	20.7	20.2	19.3
	Power Loading	13.19	12.34	11.61	11.61	12.00	12.03
Dimension	Noise (dBA) s External Length	75.4	76.0	79.0	75.8	77.6	TBD
(ft)	Height	29.0 9.4	25.3 9.4	25.2 9.0	28.3 9.3	27.9 8.5	26.9 8.3
(11)	Span	36.0	32.1	36.1	36.0	36.2	36.4
	Internal Length	7.2	8.3	7.9	9.3	9.7	8.1
	Height	4.0	3.9	4.2	4.1	3.5	3.7
	Width	3.5	4.2	4.2	3.7	4.1	3.6
Power	Engine	Lyc	Lyc	Cont	Lyc	Lyc	Cont
		TIO-540-AK1A	TIO-540-AB1AD	TSI0-550-C	TIO-540-AJ1A	TIO-540-AH1A	TSI0-550-G
	Output (hp)	235 hp	250 hp	310 hp	310 hp	300 hp	280 hp
	Inspection Interval	2,000t	2,000t	2,000t	2,000t	2,000t	2,000t
Weights (3,112	3,086	3,600	3,617	3,615	3,374
	Max Takeoff	3,100	3,086	3,600	3,600	3,600	3,368
	Max_Landing	2,950	3,086	3,420	3,600	3,600	3,200
	Zero Fuel	2,971b	2,939b	3,390b	3,429b	3,420b	3,152b
	EOW May Dayload	2,042	1,990	2,500	2,319	2,274	2,353
	Max Payload	929	949	890 1 100	1,110	1,146	799 1 021
	Useful Load	1,070	1,096	1,100	1,298	1,341	1,021
	Max_Baggage Max_Fuel	200 522	143 517	120 588	200 522	200 612	120 612
٨٧٥	ilable Payload w/Max Fuel	522 548	517 579	588 512	522 776	612 729	612 409
	ailable Fuel w/Max Payload	141	147	210	190	195	222
Limits	VNE	175	189	230	182	189	195
21111110	VNO	140	151	181	149	150	174
	VA	110	130	158	125	132	127
Airport	TO (SL elev./ISA temp.)	1,350	1,922	1,300	1,740	1,888	1,620
	nce TO (5,000' elev. @25°C)	1,928	2,411	1,730	2,498	2,742	1,980
	Vso	50	59	59	57	59	60
	Vx	64	81	82	69	80	80
	VY	84	95	110	89	90	105
Climb	Time to Climb/Altitude	10/FL 100	10/FL 100	7/FL 100	11/FL 100	14/FL 100	7/FL 100
	Initial Gradient (ft/nm)	743	711	993	724	680	1,125
Ceilings (f		20,000	25,000	25,000	25,000	20,000	25,000
-	Service	20,000	25,000	25,000	27,000	20,000	25,000
Cruise	Long TAS	132	140	208	137	139	200
	Range Fuel Flow Altitude	62	67	78	85 FL 240	75 51, 120	81
	Specific Range	FL 200 2.129	FL 100 2.090	FL 250 2.667	FL 240 1.612	FL 120 1.853	FL 250 2.469
	Rec'd TAS	147	154	227	154	156	230
	Fuel Flow	72	74	120	98	102	144
	Altitude	FL 200	FL 100	FL 250	FL 240	FL 120	FL 250
	Specific Range	2.042	2.081	1.892	1.571	1.529	1.597
	High TAS	158	166	235	164	164	230
	Speed Fuel Flow	86	98	140	114	120	144
	Altitude	FL 200	FL 150	FL 250	FL 200	FL 120	FL 250
	Specific Range	1.837	1.694	1.679	1.439	1.367	1.597
Ranges	Seats Nautical Miles	542	484	831	202	361	520
	Full Average Speed	121	154	203	126	132	195
	Fuel Used	334	246	325	204	319	274
	Specific Range/Altitude	1.623/FL 060	1.967/FL 150	2.557/FL 100	0.990/FL 060	1.132/FL 100	1.898/FL 200
	Tanks Nautical Miles Full Average Speed	814	917 153	1,279	547 126	945 137	1,200
	Full Average Speed Fuel Used	132 468	461	203 493	136 448	547	198 551
	Specific Range/Altitude	1.739/FL 200	1.989/FL 100	2.594/FL 250	1.221/FL 200	1.728/FL 100	2.178/FL 250
Missions	200 nm Runway	1,860	1,560	1,300	1,740	1,500	1,340
(4 occupa		1 + 28	1 + 12	1+03	1 + 25	1 + 25	1 + 02
,	Fuel Used	143	111	155	188	159	135
	Specific Range/Altitude	1.399/FL 060	1.802/FL 100	1.290/FL 250	1.064/FL 060	1.258/FL 080	1.481/FL 120
	500 nm Runway	1,860	1,650	1,300	1,740	1,700	1,620
	Block Time	3 + 05	3 + 11	2 + 24	3 + 13	3 + 19	2 + 30
	Fuel Used	300	244	322	416	356	270
	Specific Range/Altitude	1.667/FL 200	2.049/FL 100	1.553/FL 250	1.202/FL 200	1.404/FL 140	1.852/FL 250
Remarks	Suggested Base Price	\$355,050	\$431,670	\$479,900	\$482,160	\$401,600	\$495,000
	Certification Basis	FAR 23, 2001	FAR 23, 1986	FAR 23	FAR 23, 1998	CAR 3 56/97	CAR 3,
			Includes KFC225.	,	Includes Garmin	Includes Avidyne	1955/89/pend.
		G1000.		Entegra.	G1000.	FlightMax	Includes Garmin
						Entegra.	G1000 . All data
							preliminary.

SINGLE-ENGINE PISTONS TURBOCHARGED | SINGLE-ENGINE PISTONS PRESSURIZED

Manufacture Model	er		New Piper Saratoga II TC PA-32R-301T
B&CA Equip	ped Price	e	\$581,600
Characterist		Seating	1+4/5
		Wing Loading	20.2
		Power Loading	12.00
		Noise (dBA)	76.6
Dimensions	Externa		27.9
(ft)		Height	8.5 36.2
	Internal	Span Length	9.7
	internar	Height	3.5
		Width	4.1
Power		Engine	Lyc
		Ü	TIO-540-AH1A
		Output (hp)	300 hp
		spection Interval	2,000t
Weights (lb))	Max Ramp	3,615
		Max Takeoff	3,600
		Max Landing	3,600
		Zero Fuel EOW	3,420b 2,486
		Max Payload	2,486 934
		Useful Load	1,129
		Max Baggage	200
		Max Fuel	612
Availa	able Payl	oad w/Max Fuel	517
		w/Max Payload	195
Limits		VNE	197
		Vno	154
		VA	134
Airport		elev./ISA temp.)	1,810
Performance	e 10 (5,0	00' elev. @25°C) Vso	2,680
		VSO	63 85
		V X V Y	95
Climb	Time to	o Climb/Altitude	9/FL 100
		Gradient (ft/nm)	753
Ceilings (ft)		Certificated	20,000
		Service	20,000
Cruise	Long	TAS	166
	Range	Fuel Flow	75
		Altitude	FL 200
	Rec'd	Specific Range TAS	2.213
	Nec u	Fuel Flow	99
		Altitude	FL 200
		Specific Range	1.859
	High	TAS	194
	Speed	Fuel Flow	138
		Altitude	FL 200
Donat	0	Specific Range	1.406
Ranges	Seats Full	Nautical Miles	354 142
	run	Average Speed Fuel Used	204
	Specific	Range/Altitude	1.735/FL 080
	Tanks	Nautical Miles	1,058
	Full	Average Speed	161
		Fuel Used	540
		Range/Altitude	1.959/FL 180
Missions	200 nm	Runway	1,670
(4 occupant	:s)	Block Time	1+07
	Specific	Fuel Used Range/Altitude	177
	500 nm	Runway	1.130/FL 200
	JOU IIII	Block Time	1,780 2 + 57
		Fuel Used	333
	Specific	Range/Altitude	1.502/FL 150
Remarks		ested Base Price	\$548,700
		ertification Basis	CAR 3, 56/97
			Seats-full range
			based on
			five occupants.

Manufactur Model	er		New Piper Malibu Mirage PA-46-350P	
B&CA Equip	ped Pric	e	\$970,000	
Characteris	tics	Seating	1+4/5	
		Wing Loading	24.8	
		Power Loading	12.40	
		Noise (dBA)	74.5	
Dimensions	Externa		28.9	
(ft)		Height	11.3	
	Internal	Span Length	43.0 12.4	
	IIILEIIIai	Height	3.9	
		Width	4.2	
Power		Engine	Lyc	
		6	TIO-540-AE2A	
		Output (hp)	350 hp	
	Ins	spection Interval	2,000t	
Weights (lb)	Max Ramp	4,358	
		Max Takeoff	4,340	
		Max Landing	4,123	
		Zero Fuel	4,123c	
		EOW May Payland	3,100	
		Max Payload	1,023	
		Useful Load Max Baggage	1,258 200	
		Max Fuel	720	
Avail	ahle Pavl	oad w/Max Fuel	538	
		w/Max Payload	235	
Limits		VNE	198	
		Vno	168	
		VA	133	
		PSI	5.5	
Airport		elev./ISA temp.)	2,090	
Performanc	e TO (5,0	000' elev. @25°C)	2,977	
		Vso	58	
		Vx Vy	81	
Climb Time	to Climb	o (min.)/Altitude	110 8/FL 100	
CIIIID TITLE		Gradient (ft/nm)	703	
Ceilings (ft)		Certificated	25,000	
	'	Service	25,000	
		Sea-level Cabin	12,300	
Cruise	Long	TAS	156	
	Range	Fuel Flow	66	
		Altitude	FL 250	
		Specific Range	2.364	
	Rec'd	TAS Fuel Flow	203	
		Altitude	108 FL 250	
		Specific Range	1.880	
	High	TAS	212	
	Speed	Fuel Flow	120	
	-,	Altitude	FL 250	
		Specific Range	1.767	
Ranges	Seats	Nautical Miles	212	
	Full	Average Speed	129	
		Fuel Used	149	
		Range/Altitude	1.423/FL 070	
	Tanks Full	Nautical Miles	1,370	
	ruii	Average Speed Fuel Used	158 670	
	Specific	Range/Altitude	2.045/FL 250	
Missions	200 nm		2,090	
(4 occupan		Block Time	1 + 06	
·		Fuel Used	170	
		Range/Altitude	1.176/FL 170	
	500 nm	Runway	2,090	
		Block Time	2 + 30	
	0 : :::	Fuel Used	352	
Domestic		Range/Altitude	1.420/FL 250	
Remarks		ested Base Price ertification Basis	\$970,000 FAR 23 1083/88	
	Ce	a cincación dasis	FAR 23, 1983/88	

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MULTIENGINE PISTONS NORMALLY ASPIRATED

Manufacture Model	er	Vulcanair SpA P 68C P 68C	Beechcraft Baron G58
B&CA Equip	ped Price	\$599,000	\$1,222,035
Characteris	tics Seating	1+5/6	1+4/5
	Wing Loading	20.7	27.6
	Power Loading	11.49	9.17
	Noise (dBA)	74.7	77.6
Dimensions	External Length	30.9	29.8
(ft)	Height	11.2	9.8
	Span	39.4	37.8
	Internal Length	13.3	12.6
	Height	3.9	4.2
	Width	3.8	3.5
Power	Engines	2 Lyc	2 Cont
	0	IO-360-A1B6	10-550-C
	Output (hp ea.) Inspection Interval	200 hp ea. 2,000t	300 hp ea.
Weights (lb)		4630	1,700t 5,524
weights (ib)	Max Takeoff	4,594	5,500
	Max Landing	4,365	5,400
	Zero Fuel	4,167c	5,215p
	EOW	2910	3,880
	Max Payload	1,257	1,335
	Useful Load	1,720	1,644
	Executive Payload	1,020	850
	Max Fuel	1,063	1,164
	Payload – Max Fuel	657	480
Av	ailable Fuel w/Max Payload	463	309
	ilable Fuel w/Exec. Payload	700	794
Limits	VNE	194	223
	Vno	154	195
	VA	132	165
Airport	TO (SL elev./ISA temp.)	1,260	2,300
Performance		4,000	4,000
	A/S (SL elev./ISA)	2,300	3,000
	A/S (5,000' elev. @ 25°C)	4,025	4,330
	VMCA	60	84
	VDEC VXSE	70 82	85
	VXSE	82 88	100 101
Climb Ti	me to Climb (min.)/Altitude	12/FL 100	10/FL 100
	nitial Engine-Out Rate (fpm)	217	390
	All-Engine Gradient (ft/nm)	920	988
	Engine-out Gradient (ft/nm)	147	232
Ceilings (ft)		_	_
	All-Engine Service	15,000	20,688
	Engine-Out Service	3,800	7,284
Cruise	Long TAS	144	163
	Range Fuel Flow	94	106
	Altitude	FL 80	FL 120
	Specific Range Rec'd TAS	1.532	1.538
	Rec'd TAS Fuel Flow	155	193
	Altitude	108 FL 80	150 FL 100
	Specific Range	1.435	1.287
	High TAS	162	200
	Speed Fuel Flow	116	190
	Altitude	FL 80	FL 80
	Specific Range	1.397	1.053
Ranges	Exec. Nautical Miles	575	993
	Payload Average Speed	145	154
	Trip Fuel	395	711
	Specific Range/Altitude	1.456/FL 80	1.397/FL 40
	Ferry Nautical Miles	855	1,563
	Average Speed	145	164
	Trip Fuel	561	1,081
Missions	Specific Range/Altitude 200 nm Runway	1.524/FL 80	1.446/FL 120
(4 occupant		1,450 1 ± 28	2,850 1 + 02
(- оссирані	Fuel Used	1 + 28 140	226
	Specific Range/Altitude	1.429/FL 80	0.885/FL 60
	500 nm Runway	1,500	2,900
	Block Time	3 + 25	2+31
	Fuel Used	375	531
	Specific Range/Altitude	1.333/FL 80	0.942/FL 60
Remarks	Suggested Base Price	\$599,000	\$1,186,000
	Certification Basis	FAR 23, 1976/80	CAR 3,
		Sagem glass	1957/69/83/05
		cockpit standard.	Inloudes Garmin
			G1000.

MULTIENGINE PISTONS TURBOCHARGED

Manufacture			Volennein Co.A	Din
Model	er		Vulcanair SpA P 68C-TC	Piper Seneca V
			P 68C-TC	PA-34-220T
B&CA Equip	ped Price		\$639,000	\$797,330
Characterist	ics	Seating	1+5/5	1+4/5
		Wing Loading	20.7	22.8
		Power Loading	10.94	10.80
Dimensions	Fortermed	Noise (dBA)	74.7	75.6
Dimensions (ft)	External	Length Height	31.33 11.2	28.6 9.9
(11)		Span	39.4	9.9 38.9
	Internal	Length	10.6	10.4
		Height	3.9	3.6
		Width	3.8	4.8
Power		Engines	2 Lyc	2 Cont
			TIO-360-C1A6D	TSI0-360-RB
		Output (hp ea.)	210 hp ea.	220 hp ea.
		spection Interval	2,000t	1,800t
Weights (lb)		Max Ramp	4,630	4,773
		Max Takeoff	4,594	4,750
		Max Landing	4,365	4,513
		Zero Fuel EOW	4,167c	4,479c
		Max Payload	2,976	3,391 1,088
		Useful Load	1,191 1,654	1,088 1,382
	F	xecutive Payload	1,020	1,362 850
	L	Max Fuel	1,146	732
	Pa	yload – Max Fuel	508	650
Av		w/Max Payload	463	294
		w/Exec. Payload	634	532
Limits		VNE	193	205
		Vno	153	166
		VA	126	140
		PSI		
Airport		elev./ISA temp.)	1,260	1,707
Performance		00' elev. @25°C)	2,200	2,435
		/S (SL elev./ISA)	2,300	2,510
	A/S (5,00	00' elev. @ 25°C) VMCA	3,000	3,117
		VMCA	63 NA	66 73
		VXSE	NA NA	83
		VYSE	NA NA	88
Climb Ti	me to Clim	b (min.)/Altitude	10/FL 100	11/FL 100
		e-Out Rate (fpm)	240	253
		Gradient (ft/nm)	1,400	996
Initial	Engine-out	Gradient (ft/nm)	NA	173
Ceilings (ft)		Certificated	20,000	25,000
		II-Engine Service	20,000	25,000
	En	gine-Out Service	11,500	16,500
		Sea Level Cabin TAS		
Cruise	Long Range	Fuel Flow	144	170
	Range	Altitude	100 FL 080	108 FL 230
		Specific Range	1.440	1.574
	Rec'd	TAS	155	200
		Fuel Flow	125	144
		Altitude	FL 080	FL 250
		Specific Range	1.240	1.389
	High	TAS	162	204
	Speed	Fuel Flow	150	156
		Altitude	FL 080	FL 250
		Specific Range	1.080	1.308
Ranges	Exec.	Nautical Miles	1,050	504
	Payload	Average Speed	145	156
	Crosifi	Trip Fuel	942	408
	Ferry	Range/Altitude Nautical Miles	1.115/FL 080 1,100	1.235/FL 130
	reny	Average Speed	1,100	865 159
		Trip Fuel	950	651
	Specific	Range/Altitude	1.158/FL 080	1.329/FL 140
Missions	200 nm	Runway	NA	1,940
(4 occupant		Block Time	1 + 28	1 + 11
		Fuel Used	260	225
	Specific	Range/Altitude	0.769/FL 080	0.889/FL 150
	500 nm	Runway	NA	2,040
		Block Time	3 + 25	2 + 44
		Fuel Used	485	484
		Range/Altitude	1.031/FL 080	1.033/FL 210
Remarks		ested Base Price	\$539,000	\$669,200
	C	ertification Basis	FAR 23, 1982	FAR 23,
			Data B&CA est. Sagem glass c'pit.	1971/80/97
			Jagem glass c pit.	

MULTIENGINE PISTONS PRESSURIZED

Manufacture Model	ır	Adam Aircraft A500 A500
B&CA Equip	ped Price	\$1,150,000
Characterist		1+5/5
	Wing Loading	41.2
	Power Loading	10.00
	Noise (dBA)	88.0
Dimensions	External Length	36.7
(ft)	Height	
(11)		9.5
	Span	44.0
	Internal Length	13.6
	Height	4.3
	Width	4.5
Power	Engines	2 Cont
		TSIO-550-E3B/E4B
	Output (hp ea.)	350 hp ea.
	Inspection Interval	1,600t
Weights (lb)	Max Ramp	7,050
worgines (iis)	Max Takeoff	7,000
	Max Landing	
		6,750
	Zero Fuel	6,500c
	EOW	4,250
	Max Payload	2,250
	Useful Load	2,800
	Executive Payload	1,020
	Max Fuel	1,320
	Payload – Max Fuel	1.480
Ava	ailable Fuel w/Max Payload	550
	lable Fuel w/Exec. Payload	
	VNF	1,780
Limits	· · · · ·	228
	Vno	166
	VA	161
	PSI	5.5
Airport	TO (SL elev./ISA temp.)	2,471
Performance	TO (5,000' elev. @25°C)	NA
	A/S (SL elev./ISA)	NA
	A/S (5,000' elev. @ 25°C)	NA
	VMCA	NA
	VDEC	81
	VXSE	90
	Vyse	105
	ne to Climb (min.)/Altitude	10/FL 100
	nitial Engine-Out Rate (fpm)	227
	All-Engine Gradient (ft/nm)	912
	Engine-out Gradient (ft/nm)	151
Ceilings (ft)	Certificated	25,000
	All-Engine Service	25,000
	Engine-Out Service	16,000
	Sea Level Cabin	12,400
Cruise	Long TAS	200
0.4.00	Range Fuel Flow	153
	Altitude	FL 250
	Specific Range	1.307
	Rec'd TAS	230
	Fuel Flow	209
	Altitude	FL 250
	Specific Range	1.100
	High TAS	230
	Speed Fuel Flow	209
	Altitude	FL 250
	Specific Range	1.100
Ranges	Exec. Nautical Miles	449
	Payload Average Speed	923
	Trip Fuel	408
	Specific Range/Altitude	1.100/FL 250
	Ferry Nautical Miles	1,541
	Average Speed	200
	Trip Fuel	1,223
	Specific Range/Altitude	1.260/FL 250
Missions	200 nm Runway	NA
(4 occupants	Block Time	1 + 02
	Fuel Used	210
	Specific Range/Altitude	0.952/FL 180
	500 nm Runway	NA
	Block Time	2 + 24
	Fuel Used	483
D '	Specific Range/Altitude	1.035/FL 250
Remarks	Suggested Base Price	NA
	Certification Basis	FAR 23 A 55
		All data preliminary.
		Í

SINGLE-ENGINE TURBOPROPS

Manufacture Model	er	Vulcanair SpA Mission VF-600w	Extra Aircraft EA 500
B&CA Equip		\$1,300,000	\$1,345,000
Characteris		1+9/9	1+5/5
	Wing Loading	32.2	30.5
	Power Loading Noise (dBA)	11.07 76.8	10.44 79.0
Dimensions	External Length	43.0	33.2
(ft)	Height	14.0	11.1
	Span	50.1	38.3
	Internal Length	16.6	13.6
	Height	3.7	4.1 4.6
Power	Width Engine	4.1 Walter	4.6 RR
	Liigino	M601F	250-B17F/2
	Output (shp)/Flat Rating	777/ISA+8°C	450/ISA+10°C
	Inspection Interval	3,000t	3,500t
Weights (lb)		8,653	4,696
	Max Takeoff Max Landing	8,598	4,696
	Zero Fuel	8,201 7,937c	4,409 4,319c
	BOW	5,181	3,186
	Max Payload	2,756	1,133
	Useful Load	3,472	1,510
	Executive Payload	1,800	1,000
	Max Fuel	2,204	1,032
	ailable Payload w/Max Fuel ailable Fuel w/Max Payload	1,268 716	478 377
	ilable Fuel w/Exec. Payload	1.672	510
Limits	VM0	190	219
	VA	143	158
	Psi	NA	5.5
Airport	TO (SL elev./ISA temp.)	2,020	1,400
Performanc		NA 61	1,900
	Vso Vx	61 95	61 90
	VX VY	125	100
Climb Ti	me to Climb (min.)/Altitude	NA/FL 100	15/FL 250
	Initial Gradient (ft/nm)	, NA	982
Ceilings (ft)		NA	25,000
	Service	NA	25,000
Cruise	Sea-level Cabin Long TAS	NA NA	12,300 210
Cruise	Long TAS Range Fuel Flow	NA NA	107
	Altitude	NA NA	FL 250
	Specific Range	NA NA	1.963
	High TAS	NA	230
	Speed Fuel Flow	NA	225
	Altitude Specific Range	NA NA	FL 140 1.022
NBAA IFR	Exec. Nautical Miles	NA NA	985
Ranges	Payload Average Speed	NA NA	200
(100 nm	(w/avail. fuel) Trip Fuel	NA	593
alternate)	Specific Range/Altitude	NA/NA	1.661/FL 250
	Max Fuel Nautical Miles	NA	1,600
	(w/avail. Average Speed	NA NA	205
	payload) Trip Fuel Specific Range/Altitude	NA NA/NA	905 1.768/FL 250
	Ferry Nautical Miles	NA) NA	1,625
	Average Speed	NA NA	207
	Trip Fuel	NA	905
Mississe	Specific Range/Altitude	NA/NA	1.796/FL 250
Missions	Flight Time	NA NA	1,200 1 + 30
(4 pax)	Fuel Used	NA NA	205
	Specific Range/Altitude	NA/NA	1.463/FL 250
	600 nm Runway	NA	1,250
	Flight Time	NA	2 + 56
	Fuel Used	NA NA (NA	400
	Specific Range/Altitude 1,000 nm Runway	NA/NA NA	1.500/FL 250 1 300
	Flight Time	NA NA	1,300 4 + 46
	Fuel Used	NA NA	650
	Specific Range/Altitude	NA/NA	1.538/FL 250
Remarks	Suggested Base Price	\$1,300,000	\$1,345,000
	Certification Basis	FAR/EASA 23	FAR 23 A 55
		pending	pending

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SINGLE-ENGINE TURBOPROPS

Manufactu Model		Cessna Caravan CE-208-675	Cessna Grand Caravan CE-208B	New Piper Aircraft Meridian PA-46-500T	Socata TBM 850 TBM 700 N	Pilatus PC-12 Series 10 PC-12/47
	ipped Price	\$1,693,400	\$1,744,125	\$1,924,730	\$2,799,850	\$3,350,000
Characteris	stics Seating Wing Loading	1+9/9 28.6	1+9/9 31.3	1+4/5 27.8	1+5/6 39.2	1+7/10 37.6
	Power Loading	11.85	12.96	10.18	10.56	8.71
	Noise (dBA)	79.0	82.7	76.8	76.2	74.3
Dimensions		37.6	41.6	29.5	34.9	47.3
(ft)	Height	14.8	15.5	11.3	14.3	14.0
	Span	52.1	52.1	43.0	41.6	53.3
	Internal Length	12.7	16.7	12.3	15.0	16.9
	Height	4.5	4.5	3.9	4.1	4.8
D	Width	5.3	5.3	4.1	4.0	5.0
Power	Engine	P&WC PT6A-114A	P&WC	P&WC	P&WC	P&WC
	Output (shp)/Flat Rating	675/ISA+27°C	PT6A-114A 675/ISA+27°C	PT6A-42A 500/ISA+55°C	PT6A-66D 700*/ISA+49°C	PT6A-67B 1,200/ISA+36°C
	Inspection Interval	3,600t	3,600t	3,600t	3,000t	3,500t
Weights (It		8,035	8,785	5,134	7,430	10,495
	Max Takeoff	8,000	8,750	5,092	7,394	10,450
	Max Landing	7,800	8,500	4,850	7,024	9,921
	Zero Fuel	7,467b	8,217b	4,850c	6,761b	9,039c
	BOW	4,852	5,258	3,404	4,885	6,565
	Max Payload	2,615	2,959	1,446	1,876	2,474
	Useful Load	3,183	3,527	1,730	2,545	3,930
	Executive Payload	1,800	1,800	800	1,000	1,400
Α.	Max Fuel vailable Payload w/Max Fuel	2,224 959	2,224	1,139 591	1,690 855	2,704
	vailable Fuel w/Max Payload	569	1,303 569	284	669	1,226 1,456
	ailable Fuel w/Exec. Payload	1,383	1,727	930	1,545	2,530
Limits	VMO	175	175	187	270	240
	VA	150	148	126	160	161
	Psı	<u> </u>	_	5.5	6.2	5.8
Airport	TO (SL elev./ISA temp.)	2,053	2,420	2,438	2,840	2,650
Performand		2,950	3,604	3,691	4,282	4,346
	Vso	61	61	61	65	67
	Vx	90	72	95	100	110
Climb T	VY Fime to Climb (min.)/Altitude	107	104	125	124	120
Cilino	Initial Gradient (ft/nm)	9/FL 100 625	12/FL 100 465	19/FL 250	18/FL 250	26/FL 250 660
Ceilings (ft		25,000	25,000	753 30,000	750 31,000	30,000
Ocinings (10	Service	25,000	23,700	30,000	31,000	30,000
	Sea-level Cabin			12,400	14,390	13,200
Cruise	Long TAS	154	154	178	255	202
	Range Fuel Flow	276	291	135	260	230
	Altitude	FL 100	FL 100	FL 280	FL 310	FL 300
	Specific Range	0.558	0.529	1.319	0.981	0.878
	High TAS Speed Fuel Flow	186	182	257	320	270
	Speed Fuel Flow Altitude	379 FL 100	379 FL 100	242 FL 280	446 FL 260	453 FL 220
	Specific Range	0.491	0.480	1.062	0.717	0.596
NBAA IFR	Exec. Nautical Miles	584	678	470	1,106	1,450
Ranges	Payload Average Speed	181	178	170	300	253
(100 nm	(w/avail. fuel) Trip Fuel	1,238	1,545	480	1,196	2,234
alternate)	Specific Range/Altitude	0.472/FL 100	0.439/FL 100	0.979/FL 280	0.925/FL 310	0.649/FL 260
	Max Fuel Nautical Miles	866	834	960	1,396	1,449
	(w/avail. Average Speed	183	179	245	255	253
	payload) Trip Fuel	1,812	1,788	965	1,677	2,233
	Specific Range/Altitude Ferry Nautical Miles	0.478/FL 100 1,131	0.466/FL 100 1,070	0.995/FL 280 1,100	0.832/FL 310 1,396	0.649/FL 260 1,487
	Average Speed	1,131	158	1,100	255	258
	Trip Fuel	2,014	2,006	965	1,677	2,244
	Specific Range/Altitude	0.562/FL 120	0.533/FL 120	1.140/FL 280	0.832/FL 310	0.663/FL 260
Missions	300 nm Runway	1,195	1,225	2,200	1,880	1,600
(4 pax)	Flight Time	1 + 41	1 + 43	1 + 20	1 + 10	1 + 14
	Fuel Used	651	667	379	434	540
	Specific Range/Altitude	0.461/FL 100	0.450/FL 100	0.792/FL 280	0.691/FL 260	0.556/FL 260
	600 nm Runway Flight Time	1,468	1,498	2,400	2,100	1,800
	Fuel Used	3 + 18	3 + 22	2 + 40	2 + 01	2 + 23
	Specific Range/Altitude	1,265 0.474/FL 100	1,295 0.463/FL 100	662 0.906/FL 280	1,125 0.533/FL 260	974 0.616/FL 260
	1,000 nm Runway	NP	NP	2,438*	2,450	2,050
	Flight Time	NP	NP	4 + 04	3 + 15	3 + 56
	Fuel Used	NP	NP	1,028	1,641	1,556
	Specific Range/Altitude	NP	NP	0.973/FL 280	0.609/FL 260	0.643/FL 260
Remarks	Suggested Base Price	\$1,440,000	\$1,474,000	\$1,895,000	\$2,576,930	\$2,792,000
	Certification Basis	FAR 23, 1984/98	FAR 23, 1986	FAR 23 A 52	FAR 23, 1990/03/06	FAR 23, 1996/05
		Utility category.	Utility category.	*1,000 nm mission	*850 shp max	EASA limits
				flown with three	continuous with flaps retracted: RVSM.	Max ramp weight
				passengers.	retracted; RVSIVI.	9,965 lb. MTOW 9,920 lb.
						IVITOW 9,920 ID.

MULTIENGINE TURBOPROPS 12,500 LB MTOW OR LESS

Manufacturer Model	Vulcanair SpA Viator AP68TP 600	Beechcraft King Air C90GT C90GT	Reims Cessna Caravan II RA406	Beechcraft King Air B200 B200	Piaggio Aero Industries Avanti II P180
B&CA Equipped Price	\$1,400,000	\$2,952,010	\$3,200,000	\$5,088,610	\$6,195,000
Characteristics Seating	1+8/9	1+5/12	1+8/13	1+7/15 41.3	1+7/9
Wing Loading Power Loading	33.0 10.08	34.4 9.18	38.9 9.36	7.35	67.0 7.12
Noise (dBA)	71.7	76.0	69.9	81.1	76.0
Dimensions External Length	37.0	35.5	39.1	43.8	47.3
ft) Height	11.9	14.3	13.2	14.8	13.0
Span Internal Length (OA/Net)	39.4 17.2/11.9	50.3 12.4/12.4	49.5 14.2/12.8	54.5 16.7/16.7	46.0 14.1/14.1
Height	4.1	4.8	4.3	4.8	5.8
Width (max)/Width (floor)	3.7/3.7	4.5/4.1	4.7/4.7	4.5/4.1	6.1/4.3
Power Engines	2 RR	2 P&WC	2 P&WC	2 P&WC	2 P&WC
0.1.1/1	250 B17C	PT6A-135A	PT6A-112	PT6A-42	PT6A-66B
Output (shp ea.)/Flat Rating Inspection Interval	328/ISA+25°C	550/ISA+30°C	500/ISA+22°C	850/ISA+26°C 3,600t	850/ISA+45°C
Weights (lb) Max Ramp	3,500t 6,669	3,600t 10,160	3,500t 9,435	12,590	3,600*t 12,150
Max Takeoff	6,613	10,100	9,360	12,500	12,100
Max Landing	6,283	9,600	9,360	12,500	11,500
Zero Fuel	5,621c	9,212b	8,500c	11,000c	9,800c
BOW	3,704	7,150	5,732	8,655	7,800
Max Payload Useful Load	1,917	2,062	2,768	2,345	2,000
Executive Payload	2,965 1,600	3,010 1,000	3,703 1,600	3,935 1,400	4,350 1,400
Max Fuel	1,501	2,573	3,183	3,645	2,802
Available Payload w/Max Fuel	1,464	437	520	290	1,548
Available Fuel w/Max Payload	1,048	948	935	1,590	2,350
Available Fuel w/Exec. Payload	1,365	2,010	2,103	2,535	2,802
Limits VMO	200	226	229	260	260
VA PSI	141	169 5.0	162 —	182 6.5	199 9.0
Airport TO (SL elev./ISA temp.)	2.034	2,392	2,964	2,579	3,262
Performance TO (5,000' elev. @25°C)	2,950	3,372	4,856	3,800	4,170
A/S (SL elev./ISA temp.)	2,887	3,382	4,363	3,411	4,250
A/S (5,000' elev. @ 25°C)	3,740	4,411	6,000	4,600	5,700
VMCA	79	80	90	86	99
Vdec Vxse	85 90	97 100	98 105	94 115	106 132
VXSE	105	100	111	121	140
Climb Time to Climb (min.)/Altitude	7/FL 100	17/FL 250	6/FL 100	16/FL 250	9/FL 250
Initial Engine-Out Rate (fpm)	NA	474	320	740	756
Initial All-Engine Gradient (ft/nm)	700	1,953	913	1,076	1,106
Initial Engine-Out Gradient (ft/nm)	NA NA	259	173	364	324
Ceilings (ft) Certificated All-Engine Service	25,000 25,000	30,000 30,000	30,000 30,000	35,000 35,000	41,000 41,000
Engine-Out Service	9,000	19,170	14,800	21,900	24,700
Sea-Level Cabin	_	11,065		15,293	24,000
Cruise Long TAS	169	208	180	228	311
Range Fuel Flow	261	332	416	378	387
Altitude	FL 100	FL 260	FL 100	FL 290	FL 390
Specific Range High TAS	0.648	0.627 270	0.433 231	0.603 289	0.804 392
Speed Fuel Flow	375	612	617	700	781
Altitude	FL 100	FL 200	FL 100	FL 220	FL 280
Specific Range	0.571	0.441	0.374	0.413	0.502
NBAA IFR Max Nautical Miles	324	178	427	406	960
Ranges Payload Average Speed	170	218	179	249	309
(100 nm (w/avail. fuel) Trip Fuel alternate) Specific Range/Altitude	1,048 0.309/FL 100	462 0.385/FL 260	1,009 0.423/FL 100	923 0.440/FL 330	1,470 0.653/FL 390
Max Fuel Nautical Miles	590	1,026	1,188	1,644	1,575
(w/avail. Average Speed	170	255	180	267	309
payload) Trip Fuel	1,100	2,074	2,767	3,024	2,170
Specific Range/Altitude	0.536/FL 100	0.495/FL 260	0.429/FL 100	0.544/FL 330	0.726/FL 410
Full Fuel Nautical Miles	620	831	1,201	1,331	1,600
(w/4 pax) Average Speed Trip Fuel	170 1,100	252 1,707	180 2,770	265 2,484	301 2,170
Specific Range/Altitude	0.564/FL 100	0.487/FL 260	0.434/FL 100	0.536/FL 330	0.737/FL 410
Ferry Nautical Miles	585	1,036	1,228	1,671	1,686
Average Speed	170	257	176	270	306
Trip Fuel	1,100	2,078	2,787	3,027	2,170
Specific Range/Altitude	0.532/FL 100	0.499/FL 260	0.441/FL 100	0.552/FL 330	0.777/FL 410
Missions 300 nm Runway 4 pax) Flight Time	NA 1 + 45	3,054 1 + 14	3,815 1 + 21	3,196 1 + 07	2,300 0 + 53
(4 pax) Flight Time Fuel Used	563	1 + 14 745	807	793	658
Specific Range/Altitude	0.533/FL 100	0.403/FL 210	0.372/FL 100	0.378/FL 250	0.456/FL 310
600 nm Runway	NA	3,251	4,075	3,264	2,410
Block Time	3 + 35	2 + 22	2 + 39	2 + 13	1 + 44
Fuel Used	1,100	1,347	1,609	1,327	1,092
Specific Range/Altitude	0.545/FL 100	0.445/FL 230	0.373/FL 100	0.452/FL 290	0.549/FL 350
1,000 nm Runway Flight Time	NP NP	3,368 3 + 54*	NP NP	3,364 3 + 39	2,540 3 + 01
Fuel Used	NP NP	2,075	NP NP	2,105	1,526
Specific Range/Altitude	NP/NP	0.482/FL 250	NP/NP	0.475/FL 290	0.655/FL 390
Remarks Suggested Base Price	\$1,300,000	\$2,950,000	NA NA	\$5,073,000	\$6,195,000
Certification Basis	FAR 23, 1986	CAR 3, 1959/81/91	FAR 23, 1986	FAR 23, 1973/80	FAR 23, 1990
	B&CA estimated	*Two passengers.		Pro Line 21	Includes Pro Line 21
	performance data.			RVSM approved.	avionics. All data
					preliminary.

MULTIENGINE TURBOPROPS >12,500 LB MTOW

Manus				
Manufacture Model			Polski Zaklady Lotnicze M28 Skytruck PZL M28 05	Beechcraft King Air 350 350
B&CA Equip		0	\$4,750,000	\$5,970,580
Characterist	ics	Seating Wing Loading	2+19/21 38.7	1+9/15* 48.4
		Power Loading	7.52	7.14
		Noise (dBA)	81.1	72.1
Dimensions	External	Length	43.2	46.7
(ft)		Height Span	16.1 72.3	14.3 57.9
	Internal	Length (OA/Net)	19.3/17.3	19.5/19.5
		Height	5.7	4.8
	Width (m	ax)/Width (floor)	5.7/NA	4.5/4.1
Power		Engines	2 P&WC PT6A-65B	2 P&WC PT6A-60A
0	utput (shn	ea.)/Flat Rating	1,100/ISA+23°C	1,050/ISA+10°C
		spection Interval	5,400t	3,600t
Weights (lb)		Max Ramp	16,584	15,100
		Max Takeoff	16,534	15,000
		Max Landing Zero Fuel	16,534 15,212c	15,000 12,500c
		BOW	9,695	9,850
		Max Payload	5,517	2,650
		Useful Load	6,889	5,250
	E	xecutive Payload	3,800	1,800
Δν	ailable Payl	Max Fuel oad w/Max Fuel	3,970 2,919	3,611 1,639
		w/Max Payload	1,372	2,600
Avai		w/Exec. Payload	3,089	3,450
Limits		Ммо	_	=
		Trans. Alt. FL		_
		VMO VA	192 132	260
		VA PSI	6.6	182 6.6
Airport		elev., ISA temp.)	1,066	3,300
Performance		00' elev. @25°C)	4,600	5,376
		/High WAT Limit	16,534	14,196
		NBAA IFR Range V2	1,050 92	1,556 109
		V Z VREF	73	100
	[anding Distance	682	2,390
	ne to Climl	o (min.)/Altitude	7/FL 10	15/FL 250
		ne-Out Rate (fpm)	650 395	552 304
Ceilings (ft)	a Liigiiie-Uu	t Gradient (ft/nm) Certificated	25,000	35,000
		II-Engine Service	30,000	35,000
		gine-Out Service	16,050	21,500
Cruise	Long	Sea-Level Cabin TAS		15,293
Juise	Long Range	Fuel Flow	152 550	235 362
		Altitude	FL 10	FL 330
		Specific Range	0.276	0.649
	High Speed	TAS	176	312
	Speed	Fuel Flow Altitude	720 FL 10	773 FL 240
		Specific Range	0.244	0.404
NBAA IFR	Max	Nautical Miles	205	899
Ranges	Payload	Average Speed	145	274
(100 nm alternate)	(w/avail.	fuel) Trip Fuel Range/Altitude	650 0.315/FL 10	1,897 0.474/FL 350
antomate)	Max Fuel		1,050	1,489
	(w/avail.		145	280
	payload)	Trip Fuel	3,250	2,951
		Range/Altitude	0.323/FL 10	0.505/FL 350
	Full Fuel (w/4 pax	Nautical Miles Nautical Miles Nautical Miles	1,150 145	1,540 285
	(/ - pax	Trip Fuel	3,275	2,958
		Range/Altitude	0.351/FL 10	0.521/FL 350
	Ferry	Nautical Miles	1,195	1,567
		Average Speed Trip Fuel	145 3,275	290 2,964
	Specific	: Range/Altitude	0.365/FL 10	0.529/FL 350
Missions	300 nm	Runway	825	2,564
(4 pax)		Flight Time	1 + 53	1+02
	Spooifie	Fuel Used	1,400 0.214/FL 10	880 0.341/FL 250
	600 nm	Range/Altitude Runway	0.214/FL 10 875	2,679
		Flight Time	3 + 45	2+01
		Fuel Used	2,750	1,467
		Range/Altitude	0.218/FL 10	0.409/FL 290
	1,000 nm	Runway Flight Time	1,000 6 + 27	2,804 3 ± 26
		Fuel Used	3,000	3 + 26 2,098
	Specific	: Range/Altitude	0.333/FL 10	0.477/FL 330
Remarks	Sugg	ested Base Price	\$4,750,000	\$5,954,000
	Ce	ertification Basis	FAR 25 pending	FAR 23, 1989
			Chelton EFIS	*may pay requires
*FAR 23 for	some Com	nmuter		*max pax requires two pilots. Pro Line
Category air				21 RVSM approved.

JETS LESS THAN 10,000 LB MTOW

Manufacture Model	er		Eclipse Aviation EA 500	Adam Aircraft A700	Cessna Citation Mustang CE-510
B&CA Equip		_	\$1,486,000	\$2,400,000	\$2,538,000
Characterist	ics	Seating Wing Loading	1+4/5 39.0	1+5/7 44.6	1+5/5 NA
		Power Loading	3.13	3.17	NA NA
		ΓO/Sideline/APR	NA/NA/NA	NA/NA/NA	NA/NA/NA
Dimensions	External	Length	33.1	40.9	40.6
(ft)		Height Span	11.0 37.4	9.6 44.0	13.4 43.2
	Internal	Length: OA/Net	7.5/5.2	11.9/9.0	9.8/9.8
		Height	4.2	4.2	4.5
Baggage	Internal	Vidth: Max/Floor Cu. ft/lb	4.7/3.0 26/260	4.5/NA NA/NA	4.6/NA 6/98
Daggage	External	Cu. ft/lb	NA/NA	30/NA	57/620
Power		Engines	2 P&WC	2 Wms Intl	2 P&WC
	Output /lb	ea.)/Flat Rating	PW610F	FJ33-4A 1.200/ISA+7°C	PW615F
		spection Interval	900/ISA+10°C 3,500t	3,500t	1,350/ISA+10°C 3.500t
Weights (lb)		Max Ramp	5,680	7,650	NA
		Max Takeoff	5,640	7,600	NA
		Max Landing Zero Fuel	5,360 4,640c	7,400 7,000c	NA NA c
		BOW	3,590	4,860	NA NA
		Max Payload	1,050	2,140	NA
	_	Useful Load xecutive Payload	2,090	2,790	NA 1.000
	E	Max Fuel	800 1.540	1,000 2.345	1,000 NA
		load w/Max Fuel	550	445	600
		l w/Max Payload	1,040	650	NA
Avai Limits	nable Fuel	w/Exec. Payload Ммо	1,290 0.640	1,790 0.700	NA 0.630
Lillits	Ti	rans. Alt. FL/VMo	FL 200/285	FL 280/260	FL 271/250
		PSI	8.3	8.5	8.3
Airport Performance		elev., ISA temp.) (5,000' @ 25 C)	2,155	3,400	3,120
remonitance		ligh Weight Limit	3,698 5,640	5,488 7,600	NA NA
		NBAA IFR RANGE	934	1,255	NA
\/-		@SL ISA, MTOW	94	102	NA
		x, NBAA IFR Res. ax, NBAA IFR Res.	83 2,040	96 2,316	NA NA
Climb		o Climb/Altitude	23/FL 370	25/FL 370	23/FL 370
		ne-Out Rate (fpm)	506	850	['] NA
FAR 25 E Ceilings (ft)	Engine-Out	Gradient (ft/nm) Certificated	323	500	NA 41,000
Cellings (IL)	Д	III-Engine Service	41,000 41,000	41,000 41,000	41,000 NA
		ngine-Out Service	25,000	29,000	NA
Cruise	Land	Sea-Level Cabin TAS	21,500	21,500	22,027
Cruise	Long Range	Fuel Flow	331 284	357 615	298 424
		Altitude	FL 410	FL 410	FL 410
	111.4	Specific Range	1.165	0.580	0.703
	High Speed	TAS Fuel Flow	375 472	387 851	340 633
	Ороси	Altitude	FL 310	FL 350	FL 350
NB 4 4		Specific Range	0.794	0.455	0.537
NBAA IFR Ranges	Max Payload	Nautical Miles Average Speed	637 306	912 344	726 300
(200 nm	(w/avail.		700	1,924	1,350
alternate)	Specific	Range/Altitude	0.910/FL 410	0.474/FL 410	0.538/FL 410
	Max Fuel		1,280	1,333	1,158
	(w/avail. payload)	Average Speed Trip Fuel	316 1,240	359 2,615	309 2,003
		c Range/Altitude	1.032/FL 410	0.510/FL 410	0.578/FL 410
	Four Pax	Nautical Miles	934	1,231	1,002
	(w/avail. fuel)	Average Speed Trip Fuel	316 950	353 2,465	307 1.768
		c Range/Altitude	0.983/FL 410	2,465 0.499/FL 410	0.567/FL 410
	Ferry	Nautical Miles	1,315	1,373	1,201
		Average Speed	316	365 2.641	316
	Specific	Trip Fuel c Range/Altitude	1,240 1.060/FL 410	2,641 0.520/FL 410	2,028 0.592/FL 410
NBAA IFR	300 nm	Runway	1,970	NA	NA NA
Missions		Flight Time	0 + 54	1 + 00	0 + 58
(4 pax)	Specific	Fuel Used c Range/Altitude	622 0.482/FL 180	623 0.482/FL 350	709 0.423/FL 350
	600 nm	Runway	0.482/FL 180 2,125	0.482/FL 350 NA	0.423/FL 350 NA
		Flight Time	1 + 43	1 + 54	1 + 55
	0===10	Fuel Used	899	965	1,173
	1,000 nm	Range/Altitude Runway	0.667/FL 290 2,155	0.622/FL 390 NA	0.512/FL 390 NA
	±,000 iiii	Flight Time	2,195 2 + 55	3 + 06	3 + 16
		Fuel Used	1,144	1,441	1,765
Remarks		c Range/Altitude ertification Basis	0.874/FL 370	0.694/FL 410	0.567/FL 410
Nemarks	C	er tilleation basis	FAR 23 pending; FAR 23 runway	FAR 23 pending All data	FAR 23 pending All data
			performance;	preliminary.	preliminary.
			1,000-nm mission		
			flown w/3 pax. All		
			data preliminary.		

JETS LESS THAN 20,000 LB MTOW

/lodel	er	Cessna CJ1+ CE-525	Cessna CJ2+ CE-525A	Beechcraft Premier IA Model 390	Cessna Citation Bravo CE-550	Sino Swearingen SJ30-2 SJ30-2
B&CA Equip		\$4,241,000	\$5,745,000	\$6,057,240	\$6,145,000	\$6,195,000
haracterist	ics Seating Wing Loading	1+7/7 44.6	1+8/9 47.4	1+6/7 50.6	2+7/8 45.8	1+5/6 73.2
	Power Loading	2.72	2.51	2.72	2.56	3.03
	(EPNdB): TO/Sideline/APR	73.5/85.2/88.5	75.5/86.1/89.7	78.3/87.9/92.0	73.7/85.2/91.2	78.5/86.2/91.8
imensions		42.6	47.7	46.0	47.3	46.8
t)	Height Span	13.8 46.9	14.0 49.8	15.3 44.5	15.0 51.7	14.2 42.3
	Internal Length: OA/Net	11.0/11.0	13.6/13.6	13.5/11.2	15.8/15.8	12.5/12.5
	Height	4.8	4.8	5.4	4.8	4.4
	Width: Max/Floor	4.8/3.1	4.8/3.1	5.5/3.7	4.8/3.2	4.8/2.8
aggage	Internal Cu. ft/lb	0/0	0/0	23/210	28/300	6/100
ower	External Cu. ft/lb Engines	45/725 2 Wms Intl	65/1,000 2 Wms Intl	54/550 2 Wms Intl	45/850 2 P&WC	53/500 2 Wms Intl
OWCI	Eligilies	FJ44-1AP	FJ44-3A-24	FJ44-2A	PW530A	FJ44-2A
	Output (lb. ea.)/Flat Rating	1,965/ISA+7°C	2,490/ISA+7°C	2,300/ISA+13°C	2,887/ISA+8°C	2,300/ISA+7°C
	Inspection Interval	3,500t	4,000t	3,500t	4,000t	3,500t
/eights (lb)		10,800	12,625	12,590	15,000	14,050
	Max Takeoff Max Landing	10,700 9,900	12,500 11,525	12,500	14,800	13,950 12,725
	Zero Fuel	9,900 8,400c	9,700c	11,600 10,000c	13,500 11,300c	10,500c
	BOW	6,965	7,925	8,550	9,500	8,650
	Max Payload	1,435	1,775	1,450	1,800	1,850
	Useful Load	3,835	4,700	4,040	5,500	5,400
	Executive Payload	1,400	1,600	1,200	1,400	1,000
Λ	Max Fuel	3,220	3,930	3,670	4,824	4,850
	ailable Payload w/Max Fuel ailable Fuel w/Max Payload	615 2,400	770 2,925	370 2,590	676 3,700	550 3,550
	ilable Fuel w/Exec. Payload	2,435	3,100	2,840	4,100	4,400
mits	MMO	0.710	0.737	0.800	0.700	0.830
	Trans. Alt. FL/VMo	FL 305/263	FL 291/278	FL 280/320	FL 279/275	FL 295/320
	PSI	8.5	8.9	8.4	8.9	12.0
rport	TOFL (SL elev., ISA temp.)	3,250	3,360	3,792	3,600	3,939
erformance	TOFL (5,000' @ 25°C) Hot/High Weight Limit	5,890 10,700	5,180 12,500	6,888 12,500	5,520 14,800	8,784 13,125
	NBAA IFR RANGE	1,195	1,597	1,178	1,508	1,915
	V2 @SL ISA, MTOW	111	116	118	121	110
VR	EF W/4 Pax, NBAA IFR Res.	101	101	112	102	100
inding Dista	ance w/4 Pax, NBAA IFR Res.	2,347	2,640	2,997	2,517	2,555
limb	Time to Climb/Altitude	21/FL 370	15/FL 370	17/FL 370	19/FL 370	16/FL 370
	R 25 Engine-Out Rate (fpm)	596	611	586	845	312
eilings (ft)	Engine-Out Gradient (ft/nm) Certificated	322 41,000	316 45,000	298 41,000	419 45,000	170 49,000
ciiiigs (it)	All-Engine Service	41,000	45,000	41,000	43,000	44,000
	Engine-Out Service	21,200	23,800	28,000	27,750	25,800
	Sea-Level Cabin	22,027	23,586	21,400	23,586	41,000
ruise	Long TAS	323	356	369	342	436
	Range Fuel Flow Altitude	532 FL 410	585 FL 450	662 FL 410	607 FL 450	684 FL 450
	Specific Range	0.607	0.609	0.557	0.563	0.637
	High TAS	384	413	451	399	475
	Speed Fuel Flow	858	1,096	1,203	1,135	1,188
	Altitude	FL 350	FL 350	FL 330	FL 350	FL 360
BAA IFR	Specific Range Max Nautical Miles	0.448	0.377	0.375	0.352	0.400
anges	Max Nautical Miles Payload Average Speed	779 344	995 368	787 390	907 336	1,445 428
200 nm	(w/avail. fuel) Trip Fuel	1,675	2,075	1,824	2,338	2,715
lternate)	Specific Range/Altitude	0.465/FL 410	0.480/FL 450	0.431/FL 410	0.388/FL 450	0.532/FL 450
	Max Fuel Nautical Miles	1,300	1,613	1,360	1,614	2,503
	(w/avail. Average Speed	353	379	408	345	415
	payload) Trip Fuel	2,569	3,157	2,934	3,603	4,133
	Specific Range/Altitude Four Pax Nautical Miles	0.506/FL 410 1,173	0.511/FL 450 1,574	0.464/FL 410 1,131	0.448/FL 450 1,495	0.606/FL 450 2,220
	(w/avail. Average Speed	352	378	402	344	422
	fuel) Trip Fuel	2,351	3,089	2,493	3,393	3,855
	Specific Range/Altitude	0.499/FL 410	0.510/FL 450	0.454/FL 410	0.441/FL 450	0.576/FL 450
	Ferry Nautical Miles	1,346	1,655	1,347	1,698	2,550
	Average Speed	359	386	410	347	425
	Trip Fuel Specific Range/Altitude	2,604 0.517/FL 410	3,187 0.519/FL 450	2,893 0.466/FL 410	3,633 0.467/FL 450	4,195 0.608/FL 470
BAA IFR	300 nm Runway	2,594	2,446	2,937	2,858	2,950
issions	Flight Time	0 + 53	0 + 49	0 + 48	0 + 53	0 + 46
pax)	Fuel Used	847	897	898	1,086	745
	Specific Range/Altitude	0.354/FL 350	0.334/FL 370	0.334/FL 370	0.276/FL 370	0.403/FL 410
	600 nm Runway	2,670	2,674	3,202	3,036	2,950
	Flight Time	1 + 41	1 + 35	1 + 33	1 + 42	1 + 28
	Fuel Used Specific Range/Altitude	1,371 0.438/FL 390	1,458 0.412/FL 410	1,432 0.419/FL 410	1,713 0.350/FL 410	1,205 0.498/FL 450
	1,000 nm Runway	3,050	0.412/FL 410 2,948	3,642	3,278	2,940
	Flight Time	2 + 51	2 + 36	2 + 30	2 + 49	2 + 22
	Fuel Used	2,035	2,158	2,229	2,495	1,865
	Specific Range/Altitude	0.491/FL 410	0.463/FL 430	0.449/FL 410	0.401/FL 430	0.536/FL 450
emarks	Certification Basis	FAR 23, 1992/00/05	FAR 23, 2000/05	FAR 23 Å 52, 2001 BOW includes 110-lb options.	FAR 25, 1978/97	FAR 23 Commute category All data prelimina

JETS LESS THAN 20,000 LB MTOW

Vlanufactur Vlodel	er	Cessna CJ3 CE-525B	Grob SPn Utilijet G180	Raytheon Aircraft Hawker 400XP Model 400A	Cessna Citation Encore+ CE-560
&CA Equi	pped Price	\$6,652,000	\$6,960,000	\$7,145,000	\$8,068,000
haracteris	tics Seating	2+6/8	1+8/9	2+7/9	2+7/11
	Wing Loading	47.2	51.6	67.6	52.2
NI-1-	Power Loading	2.46	2.46	2.75	2.48
	e (EPNdB): TO/Sideline/APR External Length	74.0/88.8/88.6 50.2	NA/NA/NA 48.6	89.0/93.7/91.7 48.4	70.0/89.8/90.5 48.9
mensions :)	Height	15.2	16.8	13.9	15.2
,	Span	53.3	48.9	43.5	54.8
	Internal Length: OA/Net	15.7/15.7	16.7/16.7	15.5/15.5	17.3/17.3
	Height	4.8	5.4	4.8	4.8
	Width: Max/Floor	4.8/3.1	5.0/4.1	4.9/3.7	4.8/3.2
aggage	Internal Cu. ft/lb External Cu. ft/lb	4/100	5/100	27/410	28/655
wer	External Cu. ft/lb Engines	65/1,000 2 Wms Intl	37/331 2 Wms Intl	26/450 2 P&WC	43/810 2 P&WC
JWCI	Liigiiles	FJ44-3A	2 Wills IIII FJ44-3A	JT15D-5	PW535B
	Output (lb. ea.)/Flat Rating	2,820/ISA+7°C	2,820/ISA+11°C	2,965/ISA+12°C	3,400/ISA+12°C
	Inspection Interval	4,000t	4,000t	3,600t	5,000t
eights (lb		14,070	14,000	16,500	17,030
	Max Takeoff	13,870	13,889	16,300	16,830
	Max_Landing	12,750	13,448	15,700	15,200
	Zero Fuel	10,510c	10,430c	13,000c	12,600c
	BOW Max Payload	8,700 1,810	7,939 2,491	10,950 2,050	10,500 2,100
	Useful Load	5,370	2,491 6,061	2,050 5,550	6,530
	Executive Payload	1,200	1,600	1,400	1,400
	Max Fuel	4,710	4,575	4,912	5,400
	ailable Payload w/Max Fuel	660	1,486	638	1,130
	ailable Fuel w/Max Payload	3,560	3,570	3,500	4,430
	ailable Fuel w/Exec. Payload	4,170	4,461	4,150	5,130
nits	Mmo	0.737	0.700	0.780	0.755
	Trans. Alt. FL/VMO	FL 293/278	FL 284/272	FL 260/320	FL 289/292
nort	TOFL (SL elev., ISA temp.)	8.9 3,180	8.3 3,000	9.1 3,906	8.9 3,590
rport erformanc		3,180 4,750	3,000 4,424	3,906 6,311	5,940
	Hot/High Weight Limit	13,870	13.314	15,830p	16,830
	NBAA IFR RANGE	1,761	1,967	1,197	1,694
	V2 @SL ISA, MTOW	114	100	116	116
	REF W/4 Pax, NBAA IFR Res.	98	87	105	99
	ance w/4 Pax, NBAA IFR Res.	2,411	2,560	2,960	2,426
mb	Time to Climb/Altitude	15/FL 370	13/FL 370	18/FL 370	13/FL 370
	AR 25 Engine-Out Rate (fpm)	800	750 450	305	858
FAR 25 Filings (ft)	Engine-Out Gradient (ft/nm) Certificated	425 45,000	450	158 45,000	444 45,000
iiiigs (it)	All-Engine Service	45,000 45,000	41,000 41,000	45,000 43,450	45,000 45,000
	Engine-Out Service	26,250	28,300	20,600	NA
	Sea-Level Cabin	23,586	21,300	24,000	23,586
uise	Long TAS	351	364	414	376
	Range Fuel Flow	618	671	938	804
	Altitude	FL 450	FL 410	FL 430	FL 450
	Specific Range	0.568	0.542	0.441	0.468
	High TAS Speed Fuel Flow	415 1,198	407 1,029	450 1,255	426 1,335
	Altitude	1,198 FL 350	1,029 FL 330	1,255 FL 390	1,335 FL 370
	Specific Range	0.346	0.396	0.359	0.319
BAA IFR	Max Nautical Miles	1,174	1,333	750	1,152
inges	Payload Average Speed	369	352	378	372
00 nm	(w/avail. fuel) Trip Fuel	2,576	2,849	2,133	3,039
ernate)	Specific Range/Altitude	0.456/FL 450	0.468/FL 410	0.352/FL 430	0.379/FL 450
	Max Fuel Nautical Miles	1,875	1,897	1,433	1,652
	(w/avail. Average Speed	378	355	394	381
	payload) Trip Fuel Specific Range/Altitude	3,880 0.483 /FL 450	3,872	3,750 0.382/FL/430	4,138
	Four Pax Nautical Miles	0.483/FL 450 1,740	0.490/FL 410 1,946	0.382/FL 430 1,351	0.399/FL 450 1,672
	(w/avail. Average Speed	377	351	393	384
	fuel) Trip Fuel	3,630	3,888	3,457	4,150
	Specific Range/Altitude	0.479/FL 450	0.501/FL 410	0.391/FL 430	0.403/FL 450
	Ferry Nautical Miles	1,916	2,028	1,561	1,727
	Average Speed	381	343	394	393
	Trip Fuel	3,900	3,903	3,786	4,190
AA IED	Specific Range/Altitude	0.491/FL 450	0.520/FL 410	0.412/FL 450	0.412/FL 450
AA IFR	300 nm Runway Flight Time	2,606	2,192	2,904	2,890
ssions pax)	Flight Time Fuel Used	0 + 49 968	0 + 49 784	0 + 46 1,098	0 + 49 1,076
pun)	Specific Range/Altitude	0.310/FL 370	0.383/FL 410	0.273/FL 370	0.279/FL 390
	600 nm Runway	2,612	2,265	3,154	2,956
	Flight Time	1 + 34	1 + 34	1 + 27	1 + 31
	Fuel Used	1,569	1,386	1,859	1,830
	Specific Range/Altitude	0.382/FL 410	0.433/FL 410	0.323/FL 410	0.328/FL 410
	1,000 nm Runway	2,754	2,451	3,570	3,047
	Flight Time	2 + 37	2 + 34	2 + 24	2 + 31
	Fuel Used	2,313	2,192	2,767	2,715
morte	Specific Range/Altitude	0.432/FL 430	0.456/FL 410	0.361/FL 430	0.368/FL 430
marks	Certification Basis	FAR 23, 2004 Commuter category	FAR 23 Commuter	FAR 25, 1981/85	FAR 25, 1988/94/00
		Commuter category	category pending All data preliminary.		pending All data preliminary.
			. III data prominiary.		data prominiary.

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Dimensions (ft)	cs Seating	Model 45 \$8,750,000	CE-560-XL	Model 45	Model 60		Hawker 850XP
Noise (Dimensions (ft)			\$10,716,800	\$11,149,000	\$12,900,000	G150 \$13,500,000	\$13,786,100
Dimensions (ft)		2+6/7	2+9/12	2+8/9	2+7/9	2+7/8	2+8/15
Dimensions (ft)	Wing Loading Power Loading	67.4 3.00	54.6 2.53	69.0 3.07	88.8 2.55	82.3 2.95	74.9 3.00
Dimensions (ft)	(EPNdB): TO/Sideline/APR	75.5/85.1/93.4	72.7/86.3/92.8	75.5/85.1/93.4	70.8/83.2/87.7	80.7/91.2/91.9	79.3/87.1/93.3
		55.6	51.8	57.6	58.7	56.8	51.1
	Height	14.1	17.2	14.1	14.6	19.1	18.1
	Span	47.8	56.3	47.8	43.8	55.6	54.3
	Internal Length (OA/Net) Height	17.7/17.7 4.9	18.5/18.5 5.7	19.8/19.8 4.9	17.7/15.8 5.7	17.67/17.67 5.75	21.3/21.3 5.7
	Width: Max/Floor	5.1/3.1	5.5/3.9	5.1/3.1	5.9/3.9	5.75/4.67	6.0/4.4
Baggage	Internal Cu. ft/lb	15/170	10/100	15/170	24/350	25/TBD	50/395
	External Cu. ft/lb	50/500	80/700	50/500	24/300	55/1,100	_/_
Power	Engines	2 HON	2 P&WC	2 HON	2 P&WC	2 HON	2 HON
,	Output (lb ea.)/Flat Rating	TFE731-20BR 3,500/ISA+25°C	PW545B 3,991/ISA+13°C	TFE731-20BR 3,500/ISA+25°C	PW305A 4,600/ISA+17°C	TFE731-40AR-200G	TFE731-5BR
(Inspection Interval	5,000/15A+25 C	5,000t	5,000c	6,000t	4,420/ISA+13°C 5,000c	4,660/ISA+10°C 4,200c
Weights (lb)	Max Ramp	21,250	20,400	21,750	23,750	26,250	28,120
	Max Takeoff	21,000	20,200	21,500	23,500	26,100	28,000
	Max Landing	19,200	18,700	19,200	19,500	21,700	23,350
	Zero Fuel	16,000c	15,100c	16,000c	17,000c	17,500c	18,450c
	BOW May Payload	13,715	12,800	13,890	14,985	15,100	16,330
	Max Payload Useful Load	2,285 7,535	2,300 7,600	2,110 7,860	2,015 8,765	2,400 11,150	2,120 11,790
	Executive Payload	1,200	1,800	1,600	1,400	1,400	1,600
	Max Fuel	5,375	6,740	6,062	7,910	10,300	10,000
Ava	ilable Payload w/Max Fuel	2,160	860	1,798	855	850	1,790
Ava	ilable Fuel w/Max Payload	5,250	5,300	5,750	6,750	8,750	9,670
	able Fuel w/Exec. Payload	5,375	5,800	6,062	7,365	9,750	10,000
Limits	Mmo	0.810	0.750	0.810	0.810	0.850	0.800
	Trans. Alt. FL/VM0 PSI	FL 270/330	FL 265/305	FL 270/330	FL 270/330	FL 293/330	FL 290/310
Airport	TOFL (SL elev., ISA temp.)	9.4 4,680	9.3 3,560	9.4 5,040	9.4 5,450	8.8 5,012	8.6 5,032
Performance		5,090	5,490	5,650	8,520	8,120	7,952
	Hot/High Weight Limit	19,760	20,200	20,662	23,371p	26,100	27,049
	NBAA IFR Range	1,617	1,742	1,869	2,228	2,977	2,676
	V2 @ SL ISA, MTOW	123	118	130	147	131	140
	FF W/4 Pax, NBAA IFR Res.	112	106	113	131	115	113
Climb	nce w/4 Pax, NBAA IFR Res. Time to Climb/Altitude	2,334 15/FL 370	2,739 15/FL 370	2,349 15/FL 370	3,061 13/FL 370	2,428 17/FL 370	2,245 19/FL 370
	R 25 Engine-Out Rate (fpm)	389	765	585	715	606	354
	ngine-Out Gradient (ft/nm)	190	389	270	292	278	152
Ceilings (ft)	Certificated	51,000	45,000	51,000	51,000	45,000	41,000
	All-Engine Service	45,200	45,000	44,700	42,400	TBD	39,000
	Engine-Out Service	28,400	28,600	27,900	24,500	TBD	18,800
Cruise	Sea-Level Cabin Long TAS	25,700 433	25,230 354	25,700 436	25,700 423	23,000 430	22,200 402
	Range Fuel Flow	950	862	982	1,134	1,187	1,168
	Altitude	FL 470	FL 450	FL 470	FL 430	FL 430	FL 390
	Specific Range	0.456	0.411	0.444	0.373	0.362	0.344
	High TAS	453	431	450	446	470	448
	Speed Fuel Flow	1,079	1,235	1,074	1,297	1,735	1,826
	Altitude	FL 470	FL 410	FL 470	FL 430	FL 370	FL 370
NBAA IFR	Specific Range Max Nautical Miles	0.420 1,367	0.349 1,130	0.419 1,557	0.344 1,742	0.271 2,276	0.245 2,394
	Payload Average Speed	421	387	423	413	420	390
(200 nm	(w/avail. fuel) Trip Fuel	3,754	3,607	4,254	5,261	7,294	7,959
alternate)	Specific Range/Altitude	0.364/FL 470	0.313/FL 450	0.366/FL 450	0.331/FL 410	0.312/FL 430	0.301/FL 410
	Max Fuel Nautical Miles	1,484	1,724	1,764	2,289	2,943	2,522
	(w/avail. Average Speed	422	396	424	415	422	391
	payload) Trip Fuel Specific Range/Altitude	4,014 0.370/FL 470	5,180 0.333/FL 450	4,712 0.374/FL 470	6,596 0.347/FL 410	8,939 0.329/FL 450	8,345 0.302/FL 410
	Four Pax Nautical Miles	1,617	1,722	1,869	2,269	2,950	2,642
	(w/avail. Average Speed	422	396	423	415	422	391
	fuel) Trip Fuel	4,087	5,175	4,764	6,548	8,942	8,395
	Specific Range/Altitude	0.396/FL 470	0.333/FL 450	0.392/FL 470	0.347/FL 410	0.330/FL 450	0.315/FL 410
	Ferry Nautical Miles	1,710	1,769	1,971	2,389	3,071	2,698
	Average Speed Trip Fuel	424 4,134	402 5,219	426 4,811	413 6,642	422 8,991	393 8,434
	Specific Range/Altitude	0.414/FL 490	0.339/FL 450	0.410/FL 490	0.360/FL 410	0.342/FL 450	0.320/FL 390
Missions	300 nm Runway	3,382	2,775	3,405	3,360	3,590	3,763
(4 pax)	Flight Time	0 + 44	0 + 46	0 + 44	0 + 44	0 + 49	0 + 48
	Fuel Used	1,201	1,245	1,206	1,194	1,227	1,472
	Specific Range/Altitude	0.250/FL 390	0.241/FL 390	0.249/FL 390	0.251/FL 390	0.244/FL 450	0.204/FL 370
	600 nm Runway	3,537	2,784	3,568	3,590	3,757	3,852
	Flight Time Fuel Used	1 + 23 1,966	1 + 29 2,094	1 + 23 1,977	1 + 25 1,940	1 + 30 1,977	1 + 28 2,720
	Specific Range/Altitude	0.305/FL 430	0.287/FL 410	0.303/FL 430	0.309/FL 430	0.303/FL 450	0.221/FL 370
	1,000 nm Runway	3,722	3,021	3,739	3,985	3,949	3,974
	Flight Time	2 + 15	2 + 26	2 + 15	2 + 19	2 + 27	2 + 22
	Fuel Used	3,101	3,210	3,123	3,083	3,014	4,387
	Specific Range/Altitude	0.322/FL 430	0.312/FL 430	0.320/FL 430	0.324/FL 430	0.332/FL 450	0.228/FL 370
Remarks	Certification Basis	FAR 25 A 77	FAR 25, 1998/04	FAR 25 A 77	FAR 25, 1981/92	FAR 25 A108,	CAR 4b, 1963;
		JAR 25 A 13,	RE100 APU	JAR 25 A 13	FAR 25 pending	2005	FAR 25 Amend
		1997/03 incorporates	standard.		Pro Line 21		1984/95/06 Pro Line 21 std.
		SB 40-11-1			std.		36-150 APU.

Manufacture Model		Cessna Citation Sovereign CE-680	Embraer Legacy Shuttle EMB-135LR	Bombardier Challenger 300 BD-100-1A10	Raytheon Aircraft Hawker 4000 Model 4000	Cessna Citation X CE-750	Dassault Falcon 50EX DA50
B&CA Equip		\$15,479,000	\$17,800,000	\$19,210,000	\$19,557,000	\$20,062,000	\$21,150,000
Characterist	tics Seating Wing Loading	2+9/12 58.2	2+16/37 80.0	2+8/13 74.4	2+8/14 70.6	2+8/11 68.5	2+9/19 79.8
	Power Loading	2.63	3.12	2.85	2.72	2.67	3.58
Noise	(EPNdB): TO/Sideline/APR	71.5/87.6/91.3	79.7/84.5/92.3	75.3/87.6/89.6	NA/NA/NA	73.2/83.8/90.3	83.8/95.2/93.1
Dimensions		63.5	86.4	68.6	69.2	72.3	60.8
(ft)	Height	20.3	22.2	20.0	19.6	19.3	22.9
	Span	63.3	65.8	63.8	61.8	63.9	61.8
	Internal Length (OA/Net) Height	25.3/25.3 5.7	53.4/42.4 6.0	28.6/16.5 6.1	29.5/25.0 6.0	23.9/23.9 5.7	28.0/23.5 5.9
	Width: Max/Floor	5.5/3.9	6.9/5.2	7.2/5.1	6.5/4.0	5.5/3.9	6.1/5.2
Baggage	Internal Cu. ft/lb	35/415	42/551	106/750	115/NA	variable/variable	-/-
	External Cu. ft/lb	100/1,000	325/2,205	_/_	-/-	82/775	115/2,200
Power	Engines	2 P&WC	2 RR	2 HON	2 P&WC	2 RR	3 HON
	0 1 1 (11) (5) 1 5 1 1	PW306C	AE 3007 A1/3	HTF 7000	PW308A	AE3007C1	TFE731-40
	Output (lb ea.)/Flat Rating Inspection Interval	5,770/ISA+18°C 6,000t	7,057/ISA+30°C OC	6,826/ISA+15°C OC	6,900/ISA+20°C 6,000t	6,764/ISA+15°C 4,500*t	3,700/ISA+17°C 6,000c
Weights (lb)		30,550	44,312	39.000	37,700	36,400	39,900
110.8.110 (1.0)	Max Takeoff	30,300	44,092	38,850	37,500	36,100	39,700
	Max Landing	27,100	40,785	33,750	33,500	31,800	35,715
	Zero Fuel	20,450c	35,274c	27,000c	25,000c	24,400c	25,570c
	BOW	18,150	25,353	23,500	22,475	22,100	22,250
	Max Payload	2,300	9,921	3,500	2,525	2,300	3,320
	Useful Load Executive Payload	12,400 1,800	18,959 3,200	15,500 1,600	15,225 1,600	14,300 1,600	17,650 1,800
	Max Fuel	1,800	11,321	1,600	1,600	12,931	15,520
Av	railable Payload w/Max Fuel	1,184	7,638	1,455	625	1,369	2,130
Av	ailable Fuel w/Max Payload	10,100	9,038	12,000	12,700	12,000	14,330
Ava	ilable Fuel w/Exec. Payload	10,600	11,321	13,900	13,625	12,700	15,520
Limits	Ммо	0.800	0.780	0.830	0.840	0.920	0.860
	Trans. Alt. FL/VMO	FL 298/305	FL 276/320	FL 290/320	FL 200/350	FL 307/350	FL 240/370
Airport	TOFL (SL elev., ISA temp.)	9.3 3,640	7.8 5,600	8.8 4,720	9.6 4,509	9.3 5,140	9.2 4,935
Performance		4,950	7,024	6,860	6,720	7,350	7,247
	Hot/High Weight Limit	30,000	44,092	38,345	37,500	34,980p	38,570
	NBAA IFR Range	2,903	2,049	3,276	3,362	2,980	3,382
	V2 @ SL ISA, MTOW	115	129	130	134	137	123
	REF W/4 Pax, NBAA IFR Res.	95	106	113	107	112	106
Climb	ance w/4 Pax, NBAA IFR Res. Time to Climb/Altitude	2,202 14/FL 370	2,057 22/FL 370	2,290 14/FL 370	2,490 13/FL 370	2,730 18/FL 370	2,159 17/FL 370
	R 25 Engine-Out Rate (fpm)	617	301	473	602	486	671
	Engine-Out Gradient (ft/nm)	322	140	219	270	213	322
Ceilings (ft)		47,000	37,000	45,000	45,000	51,000	49,000
	All-Engine Service	43,000	37,000	44,000	42,900	43,000	41,900
	Engine-Out Service	27,560	22,331	27,800	28,000	26,000	31,900
Cruise	Sea-Level Cabin Long TAS	25,230 387	19,493 424	23,500 459	25,240 430	25,230 470	23,000 417
Oluisc	Range Fuel Flow	1,133	1,931	1,577	1,453	1,529	1,586
	Altitude	FL 470	FL 370	FL 450	FL 450	FL 470	FL 430
	Specific Range	0.342	0.220	0.291	0.296	0.307	0.263
	High TAS	446	446	470	470	513	481
	Speed Fuel Flow Altitude	1,738 FL 390	2,122	1,794 FL 430	1,790	2,229	2,533
	Specific Range	0.257	FL 370 0.210	0.262	FL 410 0.263	FL 410 0.230	FL 390 0.190
NBAA IFR	Max Nautical Miles	2,405	1,167	2,544	2,775	2,703	2,868
Ranges	Payload Average Speed	404	393	446	422	462	410
(200 nm	(w/avail. fuel) Trip Fuel	8,327	6,662	9,935	10,727	9,973	12,308
alternate)	Specific Range/Altitude	0.289/FL 470	0.175/FL 370	0.256/FL 450	0.259/FL 450	0.271/FL 470	0.233/FL 450
	Max Fuel Nautical Miles (w/avail. Average Speed	2,847 406	1,658 402	3,291 449	3,400 422	3,070 462	3,249 410
	payload) Trip Fuel	9,564	9,039	12,220	12,683	11,055	13,590
	Specific Range/Altitude	0.298/FL 470	0.183/FL 370	0.269/FL 450	0.268/FL 450	0.278/FL 490	0.239/FL 470
	Four Pax Nautical Miles	2,881	1,893	3,291	3,341	3,125	3,350
	(w/avail. Average Speed	406	407	449	423	463	411
	fuel) Trip Fuel	9,576	9,436	12,220	12,503	11,078	13,590
	Specific Range/Altitude Ferry Nautical Miles	0.301/FL 470 2,960	0.201/FL 370 1,918	0.269/FL 450 3,335	0.267/FL 450 3,466	0.282/FL 490 3,221	0.247/FL 470 3,428
	Average Speed	407	408	451	423	463	412
	Trip Fuel	9,608	9,469	12,231	12,701	11,118	13,590
	Specific Range/Altitude	0.308/FL 470	0.203/FL 370	0.273/FL 450	0.273/FL 450	0.290/FL 490	0.252/FL 470
Missions	300 nm Runway	3,058	3,458	3,370	2,639	3,536	2,853
(4 pax)	Flight Time Fuel Used	0 + 45	0 + 53 1 916	0 + 47 1 570	0 + 48 1,520	0 + 41	0 + 47 1,607
	Specific Range/Altitude	1,488 0.202/FL 390	1,916 0.157/FL 370	1,570 0.191/FL 450	0.197/FL 450	1,837 0.163/FL 370	0.187/FL 450
	600 nm Runway	3,068	3,468	3,399	2,695	3,580	2,959
	Flight Time	1 + 27	1 + 35	1 + 26	1 + 26	1 + 16	1 + 27
	Fuel Used	2,399	3,288	2,554	2,542	2,855	2,705
	Specific Range/Altitude	0.250/FL 430	0.182/FL 370	0.235/FL 450	0.236/FL 450	0.210/FL 430	0.222/FL 450
	1,000 nm Runway	3,093	3,497	3,472	2,847	3,672	3,005
	Flight Time Fuel Used	2 + 22 3,750	2 + 32 5,147	2 + 18 3,889	2 + 17 3,940	2 + 03 4,469	2 + 21 4,232
	Specific Range/Altitude	0.267/FL 430	0.194/FL 370	0.257/FL 450	0.254/FL 450	0.224/FL 430	0.236/FL 450
Remarks	Certification Basis	FAR 25, 2004;	RBHA/FAR/JAR	FAR 25 A 98 and	FAR/JAR 25	FAR 25, 1996/02	FAR 25,
		JAR 25, 2005	25, 1999 Equipment level exceeds <i>B&CA</i>	JAR 25 Chg 15	2005	JAR 25, 1999/02 *Engine flight hour inspection.	1977/79/96 Optional 40,780-lb MTOW.
			MEL.				

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		Gulfstream 200 G200	Embraer Legacy Executive EMB-135BJ	Bombardier Challenger 850 CS CL-600-2B19	Dassault Falcon 2000DX DA2000EX	Bombardier Challenger 605	Dassault Falcon 2000EX DA2000EX
B&CA Equip		\$21,646,000	\$23,600,000	\$24,530,000	\$25,550,000	\$26,714,000	\$27,200,000
Characterist		2+10/18	2+13/16	2+27/50	2+8/19	2+9/19	2+8/19
	Wing Loading Power Loading	96.1 2.93	90.0 3.11	94.6 3.04	78.3 2.93	97.9 2.76	78.3 3.01
Noise	(EPNdB): TO/Sideline/APR	81.7/85.8/92.7	79.7/86.8/91.3	78.6/82.4/92.1	79.5/91.7/91.0	81.2/86.2/90.3	80.7/91.7/91.0
Dimensions		62.3	86.4	87.8	66.3	68.4	66.3
(ft)	Height	21.4	22.2	20.4	23.2	20.7	23.2
	Span	58.1	68.9	69.6	63.4	64.3	63.4
	Internal Length (OA/Net)	24.4/24.4	49.8/42.4	48.4/43.3	31.2/26.3	28.4/25.5	31.2/26.3
	Height Width: Max/Floor	6.3 7.2/5.7	6.0 6.9/5.2	6.1 8.2/7.2	6.2 7.7/6.3	6.1 8.2/7.2	6.2 7.7/6.3
Baggage	Internal Cu. ft/lb	25/367	240/1,000	296/2,700	131/1,600	115/900	131/1,600
888-	External Cu. ft/lb	125/1,980	—/—	—/—	—/—	—/—	—/—
Power	Engines	2 P&WC	2 RR	2 GE	2 P&WC	2 GE	2 P&WC
		PW306A	AE 3007 A1E	CF34-3B1	PW308C	CF34-3B	PW308C
	Output (lb ea.)/Flat Rating	6,040/ISA+17°C	7,987/ISA+22°C	8,729/ISA+8°C	7,000/ISA+15°C	8,729/ISA+15°C	7,000/ISA+15°C
Weights (lb)	Inspection Interval Max Ramp	6,000c 35,600	0C 49,758	0C 53,250	7,000c 41,200	0C 48,300	7,000c 42,400
weights (in)	Max Takeoff	35,450	49,604	53,000	41,200	48,200	42,200
	Max Landing	30,000	40,785	47,000	39,300	38,000	39,300
	Zero Fuel	24,000c	35,274c	44,000c	29,700c	32,000c	29,700c
	BOW	20,200	30,148	32,257	24,000	26,985	24,000
	Max Payload	3,800	5,126	11,743	5,700	5,015	5,700
	Useful Load	15,400	19,610	20,993	17,200	21,315	18,400
	Executive Payload Max Fuel	2,000	2,600	5,400	1,600	1,800	1,600
Δνα	ilable Payload w/Max Fuel	15,000 400	18,170 1,440	14,305 6,688	14,600 2,600	19,850 1,465	16,660 1,740
	ailable Fayload W/Max Fuel	11,600	14,484	9,250	2,600 11.500	16,300	1,740
	lable Fuel w/Exec. Payload	13,400	17,010	14,305	14,600	19,515	16,660
imits	Ммо	0.850	0.800	0.850	0.860	0.850	0.860
	Trans. Alt. FL/VMO	FL 245/360	FL 276/320	FL 254/335	FL 250/370	FL 222/348	FL 250/370
	PSI	8.8	8.4	8.3	9.3	9.2	9.3
Airport Performance	TOFL (SL elev., ISA temp.)	6,083	5,453	6,305	5,300 7,535	5,684	5,585
errormance	TOFL (5,000' @25°C) Hot/High Weight Limit	8,804 33,255p	7,385 49,119	7,433 47,112	7,535 39,283	9,123 47,535	8,120 41,343
	NBAA IFR Range	2,845	3,432	2,477	3,458	4,035	3,944
	V2 @ SL ISA, MTOW	150	139	146	135	147	136
VR	EF W/4 Pax, NBAA IFR Res.	121	113	122	113	117	113
	ance w/4 Pax, NBAA IFR Res.	2,590	2,306	2,366	2,640	2,364	2,640
Climb	Time to Climb/Altitude	19/FL 370	21/FL 370	26/FL 370	15/FL 370	21/FL 370	16/FL 370
	R 25 Engine-Out Rate (fpm)	395	619	572	643	680	643
FAR 25 E	Engine-Out Gradient (ft/nm) Certificated	158 45,000	267 41,000	235 41,000	286 47,000	278 41,000	286 47,000
Jennigs (It)	All-Engine Service	39,612	39,800	37,760	43,000	38,250	43,000
	Engine-Out Service	23,720	23,548	19,370	NA	20,000	NA NA
	Sea-Level Cabin	23,000	21,653	21,100	25,300	23,200	25,300
Cruise	Long TAS	430	424	426	418	437	422
	Range Fuel Flow	1,555	1,874	1,826	1,441	1,891	1,496
	Altitude Specific Range	FL 410 0.277	FL 410 0.226	FL 410 0.233	FL 430 0.290	FL 410 0.231	FL 430 0.282
	High TAS	470	447	460	482	470	482
	Speed Fuel Flow	2,065	2,240	2,170	2,306	2,437	2,351
	Altitude	FL 370	FL 390	FL 390	FL 390	FL 370	FL 390
	Specific Range	0.228	0.200	0.212	0.209	0.193	0.205
NBAA IFR	Max Nautical Miles	2,371	2,450	1,116	2,311	3,010	2,579
Ranges	Payload Average Speed	421	413	402 6.750	416	416	418
(200 nm alternate)	(w/avail. fuel) Trip Fuel Specific Range/Altitude	9,851 0.241/FL 410	12,300 0.199/FL 410	6,759 0.165/FL 370	9,567 0.242/FL 450	14,255 0.211/FL 410	10,767 0.240/FL 450
	Max Fuel Nautical Miles	3,444	3,361	2,186	3,270	3,975	3,823
	(w/avail. Average Speed	423	419	413	419	419	419
	payload) Trip Fuel	13,388	16,116	11,998	12,750	17,938	14,810
	Specific Range/Altitude	0.257/FL 430	0.209/FL 410	0.182/FL 390	0.256/FL 470	0.222/FL 410	0.258/FL 470
	Four Pax Nautical Miles (w/avail. Average Speed	3,312	3,403 419	2,477 415	3,425 419	4,035 419	3,912 419
	fuel) Average Speed Trip Fuel	423 12,974	16,138	12,213	419 12,750	17,962	14,810
	Specific Range/Altitude	0.255/FL 410	0.211/FL 410	0.203/FL 410	0.269/FL 470	0.225/FL 410	0.264/FL 470
	Ferry Nautical Miles	3,493	3,455	2,521	3,508	4,109	4,000
	Average Speed	423	419	415	418	419	419
	Trip Fuel	13,401	16,167	12,242	12,750	17,992	14,810
Missis	Specific Range/Altitude	0.261/FL 430	0.214/FL 410	0.206/FL 410	0.275/FL 470	0.228/FL 410	0.270/FL 470
Missions	300 nm Runway	3,850	3,455 0 ± 53	3,059 0 ± 49	3,200 0 ± 48	3,384	3,200
4 pax)	Flight Time Fuel Used	0 + 48 1,361	0 + 53 1,944	0 + 49 2,062	0 + 48 1,527	0 + 47 1,589	0 + 48 1,527
	Specific Range/Altitude	0.220/FL 430	0.154/FL 410	0.145/FL 410	0.196/FL 450	0.189/FL 410	0.196/FL 450
	600 nm Runway	3,900	3,497	3,285	3,223	3,412	3,223
	Flight Time	1 + 30	1 + 35	1 + 30	1 + 27	1 + 27	1 + 27
	Fuel Used	2,340	3,225	3,386	2,498	2,823	2,498
	Specific Range/Altitude	0.256/FL 430	0.186/FL 410	0.177/FL 410	0.240/FL 450	0.213/FL 410	0.240/FL 450
	1,000 nm Runway	3,965	3,560	3,597	3,367	3,468	3,367
	Flight Time Fuel Used	2 + 26 3,695	2 + 29 5,056	2 + 25 5,196	2 + 21 3,836	2 + 19 4,510	2 + 21 3,836
	Specific Range/Altitude	0.271/FL 430	0.198/FL 410	0.192/FL 410	0.261/FL 450	0.222/FL 410	0.261/FL 450
Remarks	Certification Basis	FAR 25 A82, 1998	RBHA/FAR/JAR	FAR/JAR 25, 1992	FAR/EASA 25	FAR 25,	FAR 25, 2003
	Co. amounton Duois	Optional 35,600-lb MTOW.	25, 1999 quipment level exceeds <i>B&CA</i> MEL.	Corporate shuttle; baggage volume is flexible. Opt'l ISA+15°C	pending All data preliminary.	1980/83/87/95 pending	7711 20, 2003

Manufacture Model		Bombardier Challenger 850 ER CL-600-2B19	Gulfstream Aero. Gulfstream 350 GIV-X	Bombardier Challenger 870 CS CL-600-2C10	Dassault Falcon 900DX DA900EX	Bombardier Challenger 890 CS CL-600-2D24	Gulfstream Aero. Gulfstream 450 GIV-X
B&CA Equip		\$28,950,000	\$29,500,000	\$29,650,000	\$32,750,000	\$33,950,000	\$34,170,000
Characteris	tics Seating Wing Loading	2+15/19 94.6	2+14/19 74.6	4+42/70 101.5	2+12/19 88.5	4+52/90 114.4	2+14/19 77.8
	Power Loading	3.04	2.56	2.96	3.11	3.22	2.67
Noise	(EPNdB): TO/Sideline/APR	78.6/82.2/92.1	74.1/89.7/92.3	82.7/89.4/92.6	79.7/90.5/92.3	83.4/89.4/92.6	75.8/89.5/92.5
Dimensions		87.8	89.3	106.1	66.3	118.9	89.3
(ft)	Height	20.4	25.2	24.8	24.8	24.8	25.2
• •	Span	69.6	77.8	76.3	63.4	81.6	77.8
	Internal Length (OA/Net)	48.4/40.2	45.1/37.0	68.1/59.2	39.0/33.2	80.9/71.8	45.1/37.0
	Height	6.1	6.2	6.2	6.2	6.2	6.2
D-dd-d-	Width: Max/Floor	8.2/7.2	7.3/5.5	8.3/7.2	7.7/6.3	8.3/7.2	7.3/5.5
Baggage	Internal Cu. ft/lb External Cu. ft/lb	202/900	169/2,000	439/4,305 —/—	127/2,866 —/—	486/5,005	169/2,000 —/—
Power	External Cu. ft/lb Engines	—/— 2 GE	/ 2 RR	/ 2 GE	/ 3 HON	—/— 2 GE	2 RR
1 OWCI	Liigilies	CF34-3B1	Tay Mk 611-8C	CF34-8C5	TFE731-60	CF34-8C5	Tay Mk 611-8C
	Output (lb ea.)/Flat Rating	8,729/ISA+8°C		12,670/ISA+15°C	5.000/ISA+17°C		13,850/ISA+15°C
	Inspection Interval	OC	12,000 or OCt	OC	6,000c	OC	12,000t or OC
Weights (lb)		53,250	71,300	75,250	46,900	85,000	74,300
	Max Takeoff	53,000	70,900	75,000	46,700	84,500	73,900
	Max_Landing	47,000	66,000	67,000	42,200	75,000	66,000
	Zero Fuel	44,000c	49,000c	62,300c	30,865c	70,600c	49,000c
	BOW Max Payload	34,167 9,833	42,700 6,300	44,907 17,393	25,800 5,065	48,262 22,338	43,000 6,000
	Useful Load	9,833 19,083	28,600	30,343	21,100	22,338 36,738	31,300
	Executive Payload	3,000	2,800	8,400	2,400	10,400	2,800
	Max Fuel	18,305	25,807	19,450	18,830	19,450	29,281
	ailable Payload w/Max Fuel	778	2,793	10,893	2,270	17,288	2,019
Av	ailable Fuel w/Max Payload	9,250	22,300	12,950	16,035	14,400	25,300
	ilable Fuel w/Exec. Payload	16,083	25,800	19,450	18,700	19,450	28,500
Limits	MMO	0.850	0.880	0.850	0.870	0.850	0.880
	Trans. Alt. FL/VM0 PSI	FL 254/335 8.3	FL 280/340 9.6	FL 254/335 8.6	FL 250/370 9.3	FL 254/335 8.6	FL 280/340 9.6
Airport	TOFL (SL elev., ISA temp.)	8.3 6,305	9.6 5,050	5,562	9.3 4,890	6,808	9.6 5,450
Performance		11,344	7,212	6,267	6,910	6,465	7,886
1 oriormano	Hot/High Weight Limit	53,000p	70,900	64,907	45,430	68.262	73,900
	NBAA IFR Range	3,091	3,890	2,502	4,088	2,356	4,385
	V2 @ SL ISA, MTOW	146	146	138	131	150	149
	REF W/4 Pax, NBAA IFR Res.	125	122	125	109	121	122
	ance w/4 Pax, NBAA IFR Res.	2,458	2,640	4,462	2,397	4,511	2,650
Climb	Time to Climb/Altitude R 25 Engine-Out Rate (fpm)	29/FL 370	15/FL 370	18/FL 370	17/FL 370	26/FL 370	16/FL 370
	Engine-Out Gradient (ft/nm)	501 206	790 323	425 185	796 365	410 164	712 285
Ceilings (ft)		41,000	45,000	41,000	51,000	41,000	45,000
• • • • • • • • • • • • • • • • • • •	All-Engine Service	37,760	43,200	38,100	40,600	37,800	42,400
	Engine-Out Service	19,370	26,000	21,400	31,600	18,100	25,000
	Sea-Level Cabin	21,100	26,700	21,100	25,300	21,100	26,700
Cruise	Long TAS	426	459	447	435	447	459
	Range Fuel Flow	1,968	2,510	2,661	1,738	2,819	2,585
	Altitude Specific Range	FL 410 0.216	FL 450 0.183	FL 410 0.168	FL 430 0.250	FL 410 0.159	FL 450 0.178
	High TAS	459	476	471	474	471	476
	Speed Fuel Flow	2,187	2,969	2,886	2,198	3,152	3,055
	Altitude	FL 410	FL 410	FL 410	FL 390	FL 410	FL 410
	Specific Range	0.210	0.160	0.163	0.216	0.149	0.156
NBAA IFR	Max Nautical Miles	1,116	3,008	1,138	3,405	1,118	3,463
Ranges	Payload Average Speed	402	450	418	421	418	452
(200 nm alternate)	(w/avail. fuel) Trip Fuel Specific Range/Altitude	6,759 0.165/FL 370	19,097 0.158/FL 450	9,270 0.123/FL 390	15,530 0.219/FL 450	10,089 0.111/FL 370	22,097 0.157/FL 450
aiternate)	Max Fuel Nautical Miles	3,096	3,726	2,157	3,912	1,823	4,281
	(w/avail. Average Speed	416	453	432	421	428	453
	payload) Trip Fuel	16,143	22,763	16,017	16,810	15,340	26,259
	Specific Range/Altitude	0.192/FL 390	0.164/FL 450	0.135/FL 390	0.233/FL 470	0.119/FL 390	0.163/FL 450
	Four Pax Nautical Miles	3,091	3,848	2,503	4,057	2,356	4,363
	(w/avail. Average Speed	416	453	435	422	435	453
	fuel) Trip Fuel	16,120	22,855	16,366	16,810	16,200	26,315
	Specific Range/Altitude Ferry Nautical Miles	0.192/FL 390 3,141	0.168/FL 450 3,898	0.153/FL 410 2.528	0.241/FL 470 4,143	0.145/FL 410 2,381	0.166/FL 450 4,419
	Average Speed	417	453	435	422	435	453
	Trip Fuel	16,172	22,891	16,392	16,810	16,228	26,351
	Specific Range/Altitude	0.194/FL 390	0.170/FL 450	0.154/FL 410	0.246/FL 470	0.147/FL 410	0.168/FL 450
Missions	300 nm Runway	3,430	3,207	3,704	2,700	4,196	3,214
(4 pax)	Flight Time	0 + 50	0 + 46	0 + 48	0 + 47	0 + 48	0 + 46
	Fuel Used	2,103	2,595	2,683	1,607	2,822	2,603
	Specific Range/Altitude 600 nm Runway	0.143/FL 410 3,643	0.116/FL 450 3,240	0.112/FL 410 3,744	0.187/FL 450 2,782	0.106/FL 410 4,196	0.115/FL 450 3,248
	Flight Time	1 + 31	1 + 25	1 + 29	1 + 27	1 + 29	1 + 25
	Fuel Used	3,480	4,102	4,450	2,675	4,668	4,116
	Specific Range/Altitude	0.172/FL 410	0.146/FL 450	0.135/FL 410	0.224/FL 450	0.129/FL 410	0.146/FL 450
	1,000 nm Runway	3,975	3,287	3,800	2,795	4,196	3,294
	Flight Time	2 + 25	2 + 18	2 + 23	2 + 20	2 + 23	2 + 18
	Fuel Used	5,389	6,158	6,836	4,150	7,174	6,180
Domestic	Specific Range/Altitude	0.186/FL 410	0.162/FL 450	0.146/FL 410	0.241/FL 450	0.139/FL 410	0.162/FL 450
Remarks	Certification Basis	FAR/JAR 25, 1992	FAR 25, 2004	FAR/JAR 25,	FAR 25,	FAR/JAR 25,	FAR 25, 2004
		Exec. config.; baggage volume is		1980/2002 Corporate shuttle;	1979/91/99	1980/2004 Optional 89,700-lb	
		flexible. Optional		baggage volume is		MTOW.	
		ISA+15°C engine		flexible.		Optional BEVS.	
		flat rating.					

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Manufactur Model	er	Dassault Falcon 900EX DA900EX	Bombardier Global 5000 BD-700-1A11	Gulfstream Aero. Gulfstream 500 GV-SP	Dassault Falcon 7X DA7X	Airbus Elite A318-112	Boeing BBJ3 737-900ER
B&CA Equip	pped Price	\$36,150,000	\$36,750,000	\$38,760,000	\$39,200,000	\$45,000,000	\$77,500,000
Characteris		2+12/19	3+8/17	2+16/19	2+12/19	4+18/18	4+19/215
	Wing Loading Power Loading	91.6 3.22	85.8 2.97	74.9 2.77	83.7 3.48	107.7 3.05	139.9 3.44
Noise	(EPNdB): TO/Sideline/APR	79.8/90.5/92.3	79.9/89.1/89.7	77.6/90.5/90.8	TBD/TBD/TBD	83.0/91.9/93.9	NA/NA/NA
Dimensions	External Length	66.3	96.8	96.4	76.1	103.2	138.2
(ft)	Height	24.8	25.5	25.8	25.6	42.1	41.2
	Span Internal Length (OA/Net)	63.4 39.0/33.2	94.0 42.5/37.3	93.5 50.1/42.6	82.6 46.0/39.1	111.8 NA/70.2	117.4 107.2/107.2
	Height	6.2	6.3	6.2	6.2	7.4	7.1
	Width: Max/Floor	7.7/6.3	8.2/6.9	7.3/5.5	7.7/6.3	12.2/11.6	11.6/10.7
Baggage	Internal Cu. ft/lb External Cu. ft/lb	127/2,866	195/1,000	226/2,500	140/NA	NA/NA	NA/NA
Power	External Cu. ft/lb Engines	/_ 3 HON	/ 2 RR	/ 2 RR	—/— 3 P&WC	NA/NA 2 CFMI	1,250/NA 2 CFMI
	2g65	TFE731-60	BR700-710A2-20	BR700-710C4-11	PW307A	CFM56-5B9/P	CFM56-7B27
	Output (lb ea.)/Flat Rating	5,000/ISA+17°C	14,750/ISA+20°C	15,385/ISA+15°C	6,100/ISA+17°C		27,300/ISA+15°C
Majahta (Ilh	Inspection Interval	6,000c	0C 87,950	0C 85,500	7,200c	0C	0C 188,200
Weights (lb) Max Ramp Max Takeoff	48,500 48,300	87,950 87,700	85,500 85,100	63,900 63,700	143,080 142,197	188,200 187,700
	Max Landing	42,000	78,600	75,300	60,500	126,765	157,300
	Zero Fuel	30,865c	56,000c	54,500c	39,100c	120,151c	149,300c
	BOW	26,029	50,830	48,000	33,100	93,700	110,500
	Max Payload Useful Load	4,836 22,471	5,170 37,120	6,500 37,500	6,000 30,800	26,451 49,380	38,800 77,700
	Executive Payload	2,471	1,600	3,200	30,800 2,400	49,380 3,600	77,700 3,800
	Max Fuel	21,000	35,733	34,939	28,900	45,761	63,007
	ailable Payload w/Max Fuel	1,471	1,387	2,561	1,900	3,619	14,693
	vailable Fuel w/Max Payload	17,635	31,950	31,000	24,800	22,929	38,900
Limits	nilable Fuel w/Exec. Payload Mmo	20,071 0.870	35,520 0.890	34,300 0.885	28,400 0.900	45,761 0.820	63,007 0.820
- IIIII	Trans. Alt. FL/VMo	FL 250/370	FL 303/340	FL 270/340	NA/370	FL 250/350	FL 260/340
	PSI	9.3	10.3	10.2	10.2	8.2	9.0
Airport	TOFL (SL elev., ISA temp.)	5,213	5,000	5,150	5,200	5,150	7,000
Performanc	e TOFL (5,000' @25°C) Hot/High Weight Limit	7,214 47,829	6,650 87,700	7,680 85,100	NA NA	6,370 142,200	12,840 174,857
	NBAA IFR Range	4,501	4,800	5,969	5,700	3,800	4,761
	V2@ SL ISA, MTOW	133	130	141	134	129	NA
	REF W/4 Pax, NBAA IFR Res.	110	107	111	104	NA	121
Landing Dist	tance w/4 Pax, NBAA IFR Res. Time to Climb/Altitude	2,411 18/FL 370	2,180 18/FL 370	2,220 16/FL 370	2,300 NA	NA NA	2,475 26/FL 370
	AR 25 Engine-Out Rate (fpm)	755	651	707	NA NA	NA NA	26/FL 370 NA
	Engine-Out Gradient (ft/nm)	340	300	301	NA	NA	NA
Ceilings (ft)		51,000	51,000	51,000	51,000	39,800	41,000
	All-Engine Service	40,100	44,600	44,000	NA	NA	35,000
	Engine-Out Service Sea-Level Cabin	31,400 25,300	20,600 26,500	27,700 29,200	NA 29,200	NA NA	NA 18,500
Cruise	Long TAS	436	470	459	NA NA	447	455
	Range Fuel Flow	1,809	2,836	2,425	NA	NA	5,278
	Altitude	FL 430	FL 450	FL 450	NA	NA	FL 370
	Specific Range High TAS	0.241 474	0.166 499	0.189 488	NA NA	NA 469	0.086 470
	Speed Fuel Flow	2,268	3,664	3,021	NA NA	NA NA	6,103
	Altitude	FL 390	FL 430	FL 430	NA	NA	FL 370
ND 4 4 IED	Specific Range	0.209	0.136	0.162	NA NA	NA	0.077
NBAA IFR Ranges	Max Nautical Miles Payload Average Speed	3,405 421	4,150 477	4,893 451	NA NA	NA NA	2,190 436
(200 nm	(w/avail. fuel) Trip Fuel	15,530	28,618	28,093	NA NA	NA NA	31,920
alternate)	Specific Range/Altitude	0.219/FL 450	0.145/FL 470	0.174/FL 490	NA	NA	0.069/FL 350
	Max Fuel Nautical Miles	4,404	4,858	5,788	NA	NA	4,324
	(w/avail. Average Speed payload) Trip Fuel	422 18,980	479 32,526	453 32,142	NA NA	NA NA	445 56,800
	Specific Range/Altitude	0.232/FL 470	0.149/FL 470	0.180/FL 490	NA NA	NA NA	0.076/FL 390
	Four Pax Nautical Miles	4,469	4,876	5,941	5,820	NA	4,736
	(w/avail. Average Speed	422	479	453	460	NA	446
	fuel) Trip Fuel Specific Range/Altitude	18,980 0.235/FL 470	32,535 0.150/FL 470	32,193 0.185/FL 490	26,491 0.220/FL 490	NA NA	57,440 0.082/FL 410
	Ferry Nautical Miles	4,562	4,921	6,009	0.220/FL 490 NA	NA NA	4,789
	Average Speed	422	479	453	NA	NA	446
	Trip Fuel	18,980	32,560	32,216	NA	NA	57,490
Missions	Specific Range/Altitude 300 nm Runway	0.240/FL 470	0.151/FL 470	0.187/FL 490	NA 2.760	NA NA	0.083/FL 410
Missions (4 pax)	300 nm Runway Flight Time	2,700 0 + 47	2,495 0 + 46	3,366 0 + 47	2,760 0 + 46	NA NA	NA 0 + 49
, , , , , ,	Fuel Used	1,614	2,755	2,380	1,902	NA	3,700
	Specific Range/Altitude	0.186/FL 450	0.109/FL 490	0.126/FL 450	0.158/FL 410	NA	0.081/FL 410
	600 nm Runway	2,784	2,584	3,388	2,754	NA NA	NA 1 + 20
	Flight Time Fuel Used	1 + 27 2,688	1 + 23 4,441	1 + 27 3,700	1 + 26 2,912	NA NA	1 + 29 7,300
	Specific Range/Altitude	0.223/FL 450	0.135/FL 490	0.162/FL 510	0.206/CL 490	NA NA	0.082/FL 410
	1,000 nm Runway	2,796	2,706	3,413	2,750	NA	4,065
	Flight Time	2 + 20	2 + 13	2 + 20	2 + 19	2 + 32	2 + 25
	Fuel Used	4,173	6,746	5,511 0.181/FL 510	4,374	NA NA	12,020 0.083/FL 410
Remarks	Specific Range/Altitude Certification Basis	0.240/FL 450 FAR 25	0.148/FL 490 FAR 25, 1998/04	0.181/FL 510 FAR 25, 2002	0.229/FL 490 FAR/EASA 25	NA FAR 25 2003	0.083/FL 410 FAR 25 A 77
	oor emounting basis	1979/91/96	EASA 25, 2004	1 AN 23, 2002	pending	*Also avail. with	pending
		Optional 49,000-lb	Optional 89,700-lb		All data	PW6124 engines	2008 dollars. All
		MTOW. Price	MTOW. Optional		preliminary.	and opt'l. 3,540-lb	data preliminary.
		includes EASy cockpit.	BEVS.			capacity ACT.	300 & 600 nm missions B&CA est.
		cockpit.					missions back est.

ULTRA-LONG-RANGE JETS

lanufacturer lodel		Gulfstream Aero. Gulfstream 550 GV-SP	Bombardier Global Express XRS BD-700-1A10	Airbus Corporate Jetliner A319-133	Boeing BBJ1 737-700IGW	Boeing BBJ2 737-800
&CA Equipped Price	Castina	\$46,665,000	\$47,750,000	\$55,000,000	\$57,500,000	\$69,500,000
haracteristics	Seating Wing Loading	4+16/19 80.1	4+8/17 95.9	4+48/134 126.3	4+19/149 127.5	4+19/189 129.9
	ower Loading	2.96	3.32	3.14	3.13	3.19
Noise (EPNdB): TO/	Sideline/APR	79.3/90.2/90.8	83.4/88.4/89.8	85.9/91.5/94.3	85.6/95.2/95.9	86.0/94.7/96.3
imensions External	Length	96.4	99.4	111.0	110.3	129.5
t)	Height	25.8	25.5	38.6	41.2	41.2
Internal Ler	Span ngth: OA/Net	93.5 50.1/42.6	94.0 48.4/43.2	111.8 78.0/78.0	117.4 79.2/79.2	117.4 98.3/98.3
iliterilai Lei	Height	6.2	6.3	7.4	7.1	7.1
Widt	h: Max/Floor	7.3/5.5	8.2/6.9	12.2/11.6	11.6/10.7	11.6/10.7
aggage Internal	Cu. ft/lb	226/2,500	195/1,000	NA/NA	NA/NA	NA/NA
External	Cu. ft/lb	_/_	/_	NA/4,687	150/NA	750/NA
ower	Engines	2 RR BR700-710C4-11	2 RR BR700-710A2-20	2 IAE V2527M-A5*	2 CFMI CFM56-7B27	2 CFMI CFM56-7B27
Output (lb ea.	/Flat Rating	15,385/ISA+15°C	14,750/ISA+20°C	26,500/ISA+30°C	27,300/ISA+15°C	27,300/ISA+15°0
	ction Interval	0C	0C	0C	0C	0C
/eights (lb)	Max Ramp	91,400	98,250	167,380	171,500	174,700
	Max Takeoff	91,000	98,000	166,447	171,000	174,200
	Max Landing	75,300	78,600	137,787	134,000	146,300
	Zero Fuel	54,500c	56,000c	128,969c	126,000c	138,300c
	BOW Max Payload	48,700 5.800	51,200 4,800	97,653 31,316	95,960 30,040	103,220 35,080
	Useful Load	5,800 42,700	4,800 47,050	69,727	75,540	71,480
Exec	utive Payload	3,200	1,600	9,600	3,800	3,800
	Max Fuel	40,994	44,642	61,017	71,737	69,982
Available Payload		1,706	2,408	8,710	3,803	1,498
Available Fuel w/		36,900	42,250	38,411	45,500	36,400
Available Fuel w/E		39,500	44,642	60,127	71,737	67,680
mits Trans	MMO	0.885 FL 270/340	0.890 FL 303/340	0.820 FL 250/350	0.820 FL 260/340	0.820 FL 260/340
irans	. Alt. FL/Vmo PSI	FL 270/340 10.2	FL 303/340 10.3	FL 250/350 8.3	FL 260/340 9.0	FL 260/340 9.0
irport TO (SL elev.; ISA)	5,910	6,190	5,905	5,950	6,985
erformance TO (5,000')		9,070	7,880	8,900	9,800	12,850
Hot/High	Weight Limit	91,000	94,513p	166,450	169,047	174,200
	A IFR Range	6,738	5,924	4,650	6,123	5,606
	, ISA, MTOW	147	136	152	141	152
VREF w / 4 Pax, NI		112	108	111	116	122
anding Distance w / 4 Pax, I limb Time to C	imb/Altitude	2,240 18/FL 370	2,196 20/FL 370	2,970 25/FL 370	2,330 25/FL 370	2,505 27/FL 370
FAR 25 Engine-O		594	473	770	NA NA	NA NA
FAR 25 Engine-Out Gra		242	209	NA NA	NA NA	NA NA
eilings (ft)	Certificated	51,000	51,000	41,000	41,000	41,000
AII-Er	ngine Service	42,700	42,400	37,200	38,000	37,700
	e-Out Service	25,820	18,000	22,000	20,500	20,000
	a-Level Cabin	29,200	30,125	22,000	18,500	18,500
ruise Long Range	TAS Fuel Flow	459 2,563	470 2,947	447 4,565	452 4,799	454 5,084
Range	Altitude	FL 450	FL 430	FL 370	FL 390	FL 390
Sr	pecific Range	0.179	0.159	0.098	0.094	0.089
High	TAS	488	499	469	470	470
Speed	Fuel Flow	3,021	3,652	5,800	5,710	5,937
0	Altitude	FL 430	FL 410	FL 390	FL 370	FL 370
	ecific Range autical Miles	0.162 5,767	0.137 5,575	0.081 2,625	0.082 3,167	0.079 2,227
	erage Speed	452	478	436	435	433
200-nm (w/avail. fue		33,993	38,955	33,161	39,380	29,820
	nge/Altitude	0.170/FL 490	0.143/FL 470	0.079/FL 410	0.080/FL 370	0.075/FL 370
Max Fuel N		6,698	6,020	5,085	6,016	5,620
(w/avail. A		454	479	435	442	445
payload)	Trip Fuel	38,202	41,420 0.145 (EL 400	55,817	66,730	64,740
Eight Pax N	nge/Altitude autical Miles	0.175/FL 490 6,708	0.145/FL 490 6,075	0.091/FL 410 5,200	0.090/FL 410 6,098	0.087/FL 410 5,580
(w/avail. A		453	479	439	441	445
fuel)	Trip Fuel	38,205	41,445	55,832	66,780	64,370
Specific Ra	nge/Altitude	0.176/FL 490	0.147/FL 490	0.093/FL 410	0.091/FL 410	0.087/FL 410
	autical Miles	6,862	6,185	5,250	6,166	5,667
A	erage Speed	451	480	434	441	445
Crosifia D-	Trip Fuel nge/Altitude	38,110 0.180/FL 510	41,493 0.149/FL 490	55,867 0.094/FL 410	66,820	64,770 0.087/FL 410
BAA IFR 1,000 nm	Runway	3,436	2,756	0.094/FL 410 3,300	0.092/FL 410 3,600	0.087/FL 410 3,670
lissions	Flight Time	2 + 20	2,750	2 + 25	2 + 27	2 + 26
Pax)	Fuel Used	5,599	6,676	9,975	10,610	11,300
Specific Ra	nge/Altitude	0.179/FL 490	0.150/FL 490	0.100/FL 410	0.094/FL 410	0.088/FL 410
3,000 nm	Runway	3,599	3,669	3,900	4,210	4,790
	Flight Time	6 + 42	6 + 20	6 + 54	6 + 55	6 + 51
Considi- D-	Fuel Used	15,474	18,988	28,600	30,280	32,370
Specific Ra 6,000 nm	nge/Altitude Runway	0.194/FL 490 5,277	0.158/FL 490 6.006	0.105/FL 410 NP	0.099/FL 410 5.865	0.093/FL 410 NP
8,000 nm	Flight Time	5,277 13 + 15	6,006 12 + 31	NP NP	5,865 13 + 36	NP NP
	Fuel Used	33,428	40,851	NP NP	65,500	NP NP
Specific Ra	nge/Altitude	0.179/FL 490	0.147/FL 490	NP	0.092/FL 410	NP
	ication Basis	FAR 25, 1997/03/02	FAR 25, 1998/03	FAR 25, 1999	FAR 25 A 77,	FAR 25 A 77,
		.,,, 52	JAR 25	*Also available with	1967/98	1967/98
			BEVS standard.	CFM56-5B7 engines	2007 dollars.	2007 dollars.
				with 27,000 lbf.		
				WILII 21,000 IDI.		