

orthrop Grumman's family of space motors has supported the Department of Defense (DoD), NASA and commercial spacecraft manufacturers for decades, providing Earth escape, orbit transfer, orbit insertion and retro propulsion. Options include basic propulsion subsystems and fully integrated autonomous stages.

## Applications

The STAR™ family of motors and stages has served as upper stages for Northrop Grumman Pegasus and Minotaur launch vehicles as well as variants of the Titan, Atlas and Delta. NASA exploration spacecraft missions to the Moon, Venus, Mars and Pluto have highlighted the motors reliability and suitability for long duration space missions. Satellite orbit insertion applications include the GPS Block I/IA and Block II/IIA/IIR/IIM spacecraft, as well as numerous other DoD, civil and commercial communications, weather and remote sensing spacecraft.

## **Qualification and Production Data**

A large number of STAR motor variants ranging from a few pounds to over 10,000 pounds have been developed over the years and flown in support of over 2,500 missions. Designs can be modified for program-specific requirements.

## **Product Features and Benefits**

- Very high mass fraction
- Very high reliability
- Design flexibility
- Space storable propellants
- Spin- and three-axis stabilized stage options

| STAR™ Motor/Stages Features                                      | STAR 5C  | STAR 15G  | STAR 24   | STAR 27H  | STAR 37GV | STAR 48BV |
|--|----------|-----------|-----------|-----------|-----------|-----------|
| Total impulse, lb <sub>r</sub> -sec                              | 1,252    | 50,210    | 126,000   | 219,195   | 637,760   | 1,303,700 |
| Effective specific impulse, lb <sub>r</sub> -sec/lb <sub>m</sub> | 268.1    | 281.8     | 282.9     | 291.4     | 290       | 292       |
| Maximum thrust, lb <sub>f</sub>                                  | 455      | 2,800     | 4,420     | 5,250     | 15,250    | 17,490    |
| Burn/action time, sec  | 2.8/2.94 | 33.3/36.4 | 29.6/31.1 | 46.3/47.3 | 49.0/50.2 | 84.1/85.2 |
| Weight loaded, lb <sub>m</sub> (includes propellant)             | 9.86     | 206.6     | 481       | 810.9     | 2,390     | 4,780     |
| Propellant mass fraction   | 0.46     | 0.85      | 0.92      | 0.92      | 0.92      | 0.93      |
| Diameter, in.  | 4.77     | 15.04     | 24.5      | 27.3      | 35.2      | 49        |
| Length, in.  | 13.43    | 31.57     | 40.5      | 48.7      | 66.2      | 81.7      |



Example of a STAR motor satellite configuration

## For more information contact:

Mike Lara Phone: 410-392-1111 Mobile: 302-521-4208

Email: michael.lara@ngc.com

