

# TYPE-CERTIFICATE DATA SHEET

NO. EASA.A.573

For Type
Virus SW 121

# **Type Certificate Holder**

Pipistrel Vertical Solutions d.o.o. Vipavska cesta 2, 5270 Ajdovščina Slovenia, Europe

# For models:

- A) Virus SW 121
- B) Virus SW 128 (Velis Electro)



TE.CERT.00048-001 © European Union Aviation Safety Agency, 2020. All rights reserved. ISO9001 Certified. Page 1 of 12

Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

Intentionally left blank



TE.CERT.00048-001 © European Union Aviation Safety Agency, 2020. All rights reserved. ISO9001 Certified. Page 2 of 12

Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

SECTION	I A: MODEL A DESIGNATION	4
A.I.	General	4
A.II.	EASA Certification Basis	4
A.III.	Technical Characteristics and Operational Limitations	5
A.IV.	Operating and Service Instructions	7
A.V.	Notes	7
SECTION	I B: MODEL B DESIGNATION	8
B.I.	General	8
B.II.	EASA Certification Basis	8
B.III.	Technical Characteristics and Operational Limitations	9
B.IV.	Operating and Service Instructions	11
B.V.	Notes	
SECTION	I ADMINISTRATIVE	12
I.	Acronyms & Abbreviations	12
II.	Type Certificate Holder Record	12
III. (	Change Record	12

#### **SECTION A:** MODEL A DESIGNATION

# A.I. General

1. Type/ Model/ Variant

1.1 Type: Virus SW 1211.2 Model: Virus SW 121

2. Airworthiness Category: Normal

3. Manufacturer: Pipistrel d.o.o. Ajdovščina

Goriška cesta 50a 5270 Ajdovščina

**SLOVENIA** 

4. EASA Type Certification Application Date: 16.07.20105. EASA Type Certification Date: 18.04.2016

#### A.II. EASA Certification Basis

1. Reference Date for determining the applicable requirements: 29.07.2013

2. Airworthiness Requirements: Certification Specifications and

Acceptable Means of Compliance for Light Sport Aeroplanes CS-LSA, Amendment 1 from 29 July 2013.

3. Special Conditions: SC-ELA.2015-01 (CRI F-102),

Noise Requirements (CRI N-01)

SC-OLSA-div-01 (CRI O-18) (see note

3)

4. Exemptions:none5. (Reserved) Deviations:none6. Equivalent Safety Findings:none

7. Environmental Protection: see TCDSN EASA.A.573.

## A.III. <u>Technical Characteristics and Operational Limitations</u>

1. Type Design Definition: Master document list No. MDL-121-01-00-001 revision A00 or

later approved revision

2. Description: Single engine, two-seat, high wing cantilever composite

construction aircraft with T-tail empennage configuration

and fixed tricycle landing gear.

3. Equipment: Minimum equipment see Pilot Operating Handbook POH-

121-00-40-001, Section 6.4

4. Dimensions

 Length
 6.45 m 21.15 ft 

 Span
 10.70 m 35.6 ft 

 Height
 2.06 m 6.75 ft 

 Wing Area
  $9.51 \text{ m}^2$   $102.4 \text{ ft}^2$ 

5. Engine

5.1. Model: Rotax 912 S35.2 Type Certificate: EASA.E.121

5.3 Limitations: Maximum Power Rating: 73.5 kW / 5800 RPM max 5 min

Maximum Continuous Power: 69 kW / 5500 RPM

5.4. Muffler model Akrapovic iS, drawing number 121-78-00-000

6. Load factors: +4G/-2G

7. Propeller

7.1 Model: MTV-33-1-A/170-200

7.2 Type Certificate: EASA.P.048

7.3 Number of blades: 2

7.4 Diameter: 1700 mm7.5 Rotation direction: clockwise

8. Fluids

8.1 Fuel

Refer to Pilot Operating Handbook POH-121-00-40-001, Section 2.7

8.2 Oil

Refer to Pilot Operating Handbook POH-121-00-40-001, Section 2.8

8.3 Coolant

Refer to Pilot Operating Handbook POH-121-00-40-001, Section 2.8



TE.CERT.00048-001© European Union Aviation Safety Agency, 2020. All rights reserved. ISO9001 Certified. Page 5 of 12 Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

9. Fluid capacities

9.1 Fuel Total: 100 liters

Usable: 99 liters

9.2 Oil Maximum oil capacity: 3.5 liters

Minimum oil required: marked on dipstick

9.3 Coolant system 2.3 liters (approximately)

10. Air Speeds V<sub>NE</sub>: 163 KTAS (see note 1)

V<sub>NO</sub>: 120 KIAS (see note 2)

V<sub>A</sub>: 100 KIASV<sub>FE</sub>: 81 KIASV<sub>AE</sub>: 100 KIAS

11. Flight Envelope Maximum operating altitude 18,000 ft MSL

12. Approved Operations

Capability

VFR day operations; Night VFR operations (see note 3)

13. Maximum Masses Maximum takeoff - 600 kg / 1323 lbs

Maximum landing - 600 kg / 1323 lbs Maximum zero fuel - 555 kg / 1221 lbs

14. Centre of Gravity Range Forward CG limit – 25% MAC / 267 mm

Aft CG limit - 35% MAC / 357 mm

15. Reference datum

The wing's leading edge at the root of the wing

16. Control surface deflections Refer to AMM

17. Levelling Means Refer to section 6.2 of the POH

18. Minimum Flight Crew One (1) pilot

19. Maximum Passenger

**Seating Capacity** 

One (1) passenger

20. Baggage/ Cargo Location – port side, aft of the door

21. Wheels and Tyres Main wheel – 4.00" x 6", Tyre PN: 5050010

Nose wheel – 4.00" x 4", Tyre PN: 5050007

22. Lifetime limitations Refer to AMM

# A.IV. Operating and Service Instructions

1. Aircraft Flight Manual POH-121-00-40-001 A02 or later approved issue

2. Aircraft Maintenance Manual AMM-121-01-00-001 A00 or later approved issue

3. Structural Repair Manual Refer to AMM

4. Weight and Balance Manual Refer to POH

5. Illustrated Parts Catalogue IPC-121-00-50-001\_A00 or later approved issue

# A.V. Notes

Note 1: VNE is reduced from 163 KIAS at sea level by 2.2 KIAS for every 1000 ft.

Note 2: VNO decreases by 0.5 KIAS for every 1000 ft above FL100.

Note 3: When Night VFR kit PN 1159663 or 1159679 or 1159680 is installed.

#### SECTION B: MODEL B DESIGNATION

# B.I. General

1. Type/ Model/ Variant

1.1 Type: Virus SW 121

1.2 Model: Virus SW 128 (Velis Electro)

2. Airworthiness Category: Normal

3. Manufacturer: Pipistrel d.o.o. Ajdovščina

Goriška cesta 50a 5270 Ajdovščina

**SLOVENIA** 

4. EASA Type Certification Application Date: 24.10.2017

5. EASA Type Certification Date: 15.05.2020

#### **B.II. EASA Certification Basis**

1. Reference Date for determining the applicable requirements: 24th October 2017

2. Airworthiness Requirements (note 1) Certification Specifications and

Acceptable Means of Compliance for Light Sport Aeroplanes CS-LSA, Amendment 1 from 29 July 2013; Certification Specifications and Acceptable Means of Compliance for Airborne Communications, Navigation and Surveillance CS ACNS issue 2 dated

26th April 2019 (subparts A, B, D)

3. Special Conditions: SC-LSA-F2480-01 - LSA Propulsion

Lithium Batteries;

SC-LSA-15-01 - Electric Powerplant Installation for CS LSA aeroplanes; SC-ELA.2015-01 - Lithium battery

installations;

4. Exemptions: none5. (Reserved) Deviations: nonete 1)6. Equivalent Safety Findings: none

7. Environmental Protection: see TCDSN EASA.A.573.



## **B.III.** <u>Technical Characteristics and Operational Limitations</u>

1. Type Design Definition: Master Drawing List No. DWG-128-02-40-001 latest approved

revision

2. Description: Electric engine, two-seat, high wing cantilever composite

construction aircraft with T-tail empennage configuration, fixed tricycle landing gear and three-bladed composite fixed pitch

propeller.

3. Equipment: For equipment list refer to POH-128-00-40-001 Pilot's Operating

Handbook, Section 2

4. Dimensions

 Length
 6.47 m 21.22 ft 

 Span
 10.71 m 35.13 ft 

 Height
 2.08 m 6.82 ft 

 Wing Area
  $9.51 \text{ m}^2$   $102.4 \text{ ft}^2$ 

5. Load factors: +4G/-2G

6. Engine

6.1. Type/Model: Pipistrel electric engine E-811 / 268MVLC

6.2 Type Certificate: EASA.E.234

6.3 Limitations: Maximum Take-off Power MTOP: 57.6 kW / 2500 RPM max 90 s

Maximum Continuous Power: 49.2 kW / 2350 RPM

7. Propeller (note 4)

7.1 Type/Model: Pipistrel P-812 / 164-F3A

7.2 Number of blades: 3

7.4 Diameter: 1640 mm7.5 Rotation direction: clockwise

7.6 Pitch: 18° @615mm from axis

7.7 Weight: 4,88 kg

7.8 Control system: N/A (fixed pitch)

7.9 Max speed: 2500 RPM7.10 Max driving power: 57.6 kW7.11 Max driving torque: 220 Nm

7.13 Designation system: **Type**: P-812; **Diameter in cm**: 164; **Pitch**: F: fixed, G: ground

adjustable, V: variable, C: Constant speed; Number of blades: 3;

Blade type: A.



TE.CERT.00048-001©European Union Aviation Safety Agency,2020.All rights reserved.ISO9001 Certified.Page 9 of 12 Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

8. Energy Storage System (ESS)

Two (2) propulsion Lithium batteries connected in parallel.

Type: Pipistrel PB345V124E-L

Rated capacity at 23°C: 11.0 kWh (each)

Nominal voltage: 345 VDC Cooling system: Liquid

Battery management system (BMS): Integral

9. Fluids

9.1 Coolant: Refer to POH-128-00-40-001 Pilot's Operating Handbook, Section 2

10. Fluid capacities

10.1 Coolant system - for engine cooling system: 0.9 liters (approximately)

- for battery cooling system: 5.4 liters (approximately)

11. Air Speeds V<sub>NE</sub>: 108 KIAS

 $V_{NO}$ : 98 KIAS  $V_{A}$ : 100 KIAS  $V_{FE}$ : 81 KIAS

12. Flight Envelope Maximum operating altitude 12.000 ft MSL

13. Approved Operations

Capability

VFR day operations

14. Maximum Masses Maximum takeoff - 600 kg / 1323 lbs

Maximum landing - 600 kg / 1323 lbs

15. Centre of Gravity Range Forward CG limit – 25.2% MAC / 269 mm

Aft CG limit - 32.6% MAC / 336 mm

16. Reference datum

The wing's leading edge at the root of the wing

17. Control surface deflections Refer to AMM-128-00-60-001 Aircraft Maintenance Manual

latest approved issue

18. Levelling Means Refer to section 6.2 of the Pilot's Operating Handbook

19. Minimum Flight Crew One (1) pilot

TCDS No.: EASA.A.573 Virus SW 121 Date: 10 June 2020

Issue: 05

20. Maximum Passenger

**Seating Capacity** 

One (1) passenger

21. Wheels and Tyres Main wheel – 4.00" x 6", Tyre PN: 5050010

Nose wheel - 4.00" x 4", Tyre PN: 5050035

For further options and details refer to EL-128-00-30-002 Equipment List latest approved issue

22. Lifetime limitations for the airframe: Refer to section 4 of the AMM-128-00-60-001

Aircraft Maintenance Manual;

for the propeller: Refer to section 4 of the PIM-812-61-00-001

Propeller Instruction Manual;

B.IV. Operating and Service Instructions

1. Aircraft Flight Manual POH-128-00-40-001 Pilot's Operating Handbook

latest approved issue

2. Aircraft Maintenance Manual AMM-128-00-60-001 Aircraft Maintenance Manual

latest approved issue

3. Structural Repair Manual Refer to AMM-128-00-60-001 Aircraft Maintenance Manual

4. Weight and Balance Manual Refer to POH-128-00-40-001 Pilot's Operating Handbook

5. Propeller Instructions Manual Refer to PIM-812-61-00-001 Propeller Instruction Manual

5. Illustrated Parts Catalogue IPC-128-00-50-001 Illustrated Part Catalogue latest approved issue

#### **B.V.** Notes

Note 1: Requirements 4, 5, 6.1, 6.2, 6.4, 6.7, 6.10, 6.11, 7.1, 7.3, 7.4 of ASTM F2840-11, as far as the engine and its parts are concerned, are covered through the corresponding certification basis in the engine TCDS EASA.E.234.

Note 2: The propeller is certified as part of the aircraft and therefore is only certified for installation on SW128. For propeller Operating and Service Instructions see: PIM-812-61-00-001 Propeller Instruction Manual



# **SECTION ADMINISTRATIVE**

# I. Acronyms & Abbreviations

AMM Aircraft maintenance manual

CS-LSA Certification specification for light sport aeroplanes

EASA European Union Aviation Safety Agency

ESS Energy Storage System
IPC Illustrated parts catalogue
KIAS Indicated airspeed in knots
KTAS True airspeed in knots
MAC Mean aerodynamic chord

MSL Mean sea level

MDL Master document list

POH Pilot's operating handbook RPM Revolutions per minute

VFR Visual flight rules

# II. Type Certificate Holder Record

Pipistrel Vertical Solutions d.o.o. Vipavska cesta 2, 5270 Ajdovščina Slovenia, Europe

#### III. Change Record

Issue	Date	Changes	TC Issue No. & Date
Issue 01	18/04/2016	Initial Issue	18/04/2016
Issue 02	22/09/2017	Update for major change Night VFR operations	
Issue 03	12/03/2018	Corrected in section A.IV the reference to the Maintenance	
		Manual	
Issue 04	15/10/2018	Change of Type Certification Holder, Removed reference to	
		CRI A-01 from section A.II (2)	
Issue 05	10/06/2020	Model Virus SW 128 added	

-END-



TE.CERT.00048-001 © European Union Aviation Safety Agency, 2020. All rights reserved. ISO9001 Certified. Page 12 of

Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.